# Requirements Traceability Matrix

Company C

**TERMA Case** 

Document ID: 4

March 11. 2020

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Version number	Contributors	Date	Comment
1.0	HY, JM, AM, LU, TSL, MT, VNV, MM, JCJ, JAN	11-03-2020	Initial document
2.0	HY, JM, AM, LU, TSL, MT, VNV, MM, JCJ, JAN	18-03-2020	Updated requirements, Breakdown of requirements

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# 1. Scope

The scope of this document is the the requirements traceability matrix regarding the system requirements specified in the System Requirements Specification (SRS) document in relation to the original requirements from the Technical Requirements Document for the Updated Reconnaissance Pod (TRD) document.

### 2. Referenced documents

Ref.	Doc. No.	Title
[SOW]	1034832-SO	Statement of Work for the Updated Reconnaissance Pod
[TRD]	1034832-DC	Technical Requirements Document for the Updated Reconnaissance Pod

# 3. Requirements Traceability Matrix

Project r	name:	Updated Reconnaissance Pod	Busines Area:	SS				
Project r	manager:		Business Analysts lead:					
QA lead	:		Target implementatio n date:					
Req.id.	Category or functional activity	Requirement description	Use case referen ce	Design docum ent referen ce	Code or module referen ce	Test case referen ce	User accepta nce validatio n	Comments
SLR-143	System capability	The pod shall be able to acquire images of targets and areas on the ground.		R01				
SR-01	System capability	The imagery sensors shall be mounted facing down.		R02				
SLR-183	System capability	The pod shall be able to acquire electro-optical images with a footprint of 600x600 m and a ground resolution distance of less than 10 cm while flying at an altitude of 10 kft at a ground speed of 400 knots.		R03				

System capability	The pod will have a wide glass window under the camera chamber to capture a footprint of 600x600m.	R04	
System capability	The pod will read the data stream from the aircraft and calculate image capturing intervals from speed and altitude.	R05	
System capability	The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots without image overlap.	R06	
System capability	The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots with an image overlap of 55% in the line of flight.	R07	
System capability	The pod shall be able to georeference the imagery with an absolute precision better than 1 m (1 standard deviation).	R08	
System capability	The pod will have a Georeference handler to get absolute precision of image georeference.	R09	
System capability	The pod shall support air-to-air mode, where forward motion-compensation is disabled.	R10	
System capability	The pod shall be able to adjust the image acquisition to account for the terrain height, given a digital elevation model of the Earth.	R11	
External interface requirement	The pod shall run on the 115V 400Hz AC power available from the aircraft	R12	
External interface requirement	The pod shall have a power connector that will be connected to the aircraft interface.	R13	
External interface requirement	The pod shall obtain live flight information from the aircraft via its MIL-STD 1553 bus.	R14	
External interface requirement	The pod will have MIL-STD 1553 Interface Handler	R15	
External interface requirement	The pod shall react properly to the "power-on" signal available as a 28 V discrete signal from the aircraft.	R16	
	System capability  System capability  System capability  System capability  System capability  System capability  External interface requirement  External interface requirement  External interface requirement  External interface requirement	System capability  The pod will read the data stream from the aircraft and calculate image capturing intervals from speed and altitude.  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots without image overlap.  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots without image overlap.  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots with an image overlap of 55% in the line of flight.  The pod shall be able to georeference the imagery with an absolute precision better than 1 m (1 standard deviation).  The pod will have a Georeference handler to get absolute precision of image georeference.  The pod shall support air-to-air mode, where forward motion-compensation is disabled.  The pod shall be able to adjust the image acquisition to account for the terrain height, given a digital elevation model of the Earth.  External interface requirement  The pod shall run on the 115V 400Hz AC power available from the aircraft  The pod shall have a power connector that will be connected to the aircraft interface.  The pod shall batin live flight interface requirement  External interface  The pod will have MIL-STD 1553 literface Handler  The pod shall react properly to the "power-on" signal available as a 28 V	System capability  The pod will read the data stream from the aircraft and calculate image capturing intervals from speed and altitude.  System capability  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots without image overlap.  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots without image overlap.  The pod shall be able to acquire electro-optical images of an area with a size 6 km wide and 60 km long in a single flyover at an altitude of 15 kft at a ground speed of 350 knots with an image overlap of 55% in the line of flight.  The pod shall be able to georeference the imagery with an absolute precision better than 1 m (1 standard deviation).  R08  The pod will have a Georeference handler to get absolute precision of image georeference.  The pod shall base of the correction of image georeference.  The pod shall support air-to-air mode, where forward motion-compensation is disabled.  The pod shall be able to adjust the image acquisition to account for the terrain height, given a digital elevation system capability  The pod shall have a power connector that will be connected to the aircraft are requirement  The pod shall have a power connector that will be connected to the aircraft interface.  External interface  The pod shall have a power connector that will be connected to the aircraft interface.  External interface  The pod shall lotain live flight information from the aircraft via its requirement  Interface Handler  The pod shall race through the power of signal available as a 28 V

SR-07	External interface requirement	The pod will have an Analog Interface Handler to react to "power-on" signals.	R17		
SLR-179	External interface requirement	The pod shall react properly to the "zeroize" signal available as a 28 V discrete signal from the aircraft.	R18		
SR-08	External interface requirement	The pod will have an Analog Interface Handler to react to "power-on" signals.	R19		
SLR-180	External interface requirement	The pod shall output live sensor data as RS-170 standard video.	R20		
SR-09	External interface requirement	The Pod interface connector will include live sensor data stream as RS-170 standard video.	R21		
SLR-177	Internal interface requirement	The pod shall have a power consumption less than 6700 Watt.	R22		
SLR-168	Internal data requirement	The pod shall include relevant flight information as metadata in the imagery.	R23		
SR-10	Internal data requirement	The Recording software will read data streams from the aircraft interface and stamp the metadata on image capture.	R24		
SLR-155	Safety requirement	The pod shall be safe to operate and maintain, meaning that all safety risks shall have a Risk Assessment Code less than "Medium" according to MIL-STD 882.	R25		
SR-11	Safety requirement	The components and connection of the URP will be able to operate under the vibrations and disturbances during flight.	R26		
SLR-173	Safety requirement	The transportation dolly shall be safe to operate and maintain, meaning that all safety risks shall have a Risk Assessment Code less than "Medium" according to MIL-STD 882.	R27		
SLR-153	Security and privacy requirement	The pod shall be able to destroy all stored data in accordance with AEDP-03, Sanitization Level #2 "Purge" upon receiving a zeroize command.	R28		
SR-12	Security and privacy requirement	The pod will have Analog Interface Handler to check for the zeroize command to destroy all stored data.	R29		
SLR-181	System environment requirement	The pod shall ensure that the temperature around the camera does not change at a rate higher than +/- 3 degrees Celsius per hour in order to	R30		

		avoid condensation when climbing from 0 to 10,000 ft with a climb rate of 50,000 ft/min.			
SR-13	System environment requirement	A temperature sensor will be installed in the camera chamber.	R31		
SR-14	System environment requirement	The sensor readings will be used to direct the airflow to the camera chamber to maintain Temperature.	R32		
SLR-162	System environment requirement	The pod shall be able to sustain the shock and vibration loads present during flight, mounting and transportation. This requirement will be handled by Terma through Finite Element modeling.	R33		
SLR-163	System environment requirement	The pod shall be able to operate under the climatic conditions ranging from "A2 - Hot Dry" to "C2 - Cold" for deployment on aircraft defined in AECTP-230.	R34		
SR-15	System environment requirement	The Design of the pod will include a controlled airflow inside, that will maintain the operating temperature.	R35		
SLR-174	System environment requirement	The transportation dolly shall be able to operate under the climatic conditions ranging from "A2 - Hot Dry" to "C2 - Cold" defined in AECTP-230.	R36		
SR-16	System environment requirement	The cushioning material on the transportation dolly will be able to operate and maintain properties under the climatic conditions ranging from "A2 - Hot Dry" to "C2 - Cold".	R37		
SR-17	System environment requirement	The transportation dolly wheels should be able to operate under the climatic conditions ranging from "A2 - Hot Dry" to "C2 - Cold".	R38		
SR-18	System environment requirement	The transportation dolly lift will be able to operate under the climatic conditions ranging from "A2 - Hot Dry" to "C2 - Cold".	R39		
SLR-149	Computer resource requirement	The pod shall be able to store up to 10,000 images on-board.	R40		
SLR-160	Computer resource requirement	The pod should be able to store up to 100,000 images on-board.	R41		
SR-19	Computer resource requirement	The pod will include Flash Storage Unit able to store up to 100,000 images.	R42		

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Design and construction constraints	The pod shall be center-mountable on the Royal Danish Air Force (RDAF) F-16 AM/BM fighter aircrafts in version M6.5.	R43			
Design and construction constraints	The transportation dolly will be able to fit under RDAF F-16 with the URP mounted on the plane, or carried on the dolly.	R44			
Design and construction constraints	The mounting interface on the URP shall support a weight of 700 pounds.	R45			
Design and construction constraints	The pod shall have a mass less than 700 pounds in total.	R46			
Design and construction constraints	The pod shall have a geometric cross-section of 0.40 m2 or less as seen from the front.	R47			
Design and construction constraints	The pod should have a geometric cross-section of 0.25m2 or less as seen from the front.	R48			
Design and construction constraints	The pod Should have a geometric cross-section of 0.25 m2 or less as seen from the front.	R49			
Design and construction constraints	The pod shall be equipped with an electro-optical sensor.	R50			
Design and construction constraints	The pod shall be equipped with at least one of the following sensors: - XTS-365-18+IR - CA-265-12+IR	R51			
Logistics-related requirement	The transportation dolly shall be able carry the weight of the SLR.	R52			
Logistics-related requirement	The transportation dolly will be able to carry at least 700 pounds.	R53			
Logistics-related requirement	The wheels should be able to support at least 700 pounds and absorb road bumps.	R54			
Logistics-related requirement	The transportation dolly loaded with a SLR shall be operable by a single person.	R55			
Logistics-related requirement	The transportation dolly will have an interface on the back left side.	R56			
Logistics-related requirement	The transportation dolly interface will have the lift controls and the release button.	R57			
	construction constraints  Design and construction constraints  Logistics-related requirement  Logistics-related requirement	Design and construction	Design and construction F-16 AM/BM fighter aircrafts in version M6.5.  The transportation dolly will be able to fit under RDAF F-16 with the URP mounted on the plane, or carried on the construction the dolly.  Design and construction The mounting interface on the URP shall support a weight of 700 pounds.  Design and construction The pod shall have a mass less than 700 pounds in total.  Design and construction The pod shall have a geometric cross-section of 0.40 m2 or less as seen from the front.  Design and The pod should have a geometric construction cross-section of 0.25m2 or less as seen from the front.  Design and The pod Should have a geometric construction cross-section of 0.25 m2 or less as seen from the front.  Design and The pod Should have a geometric construction cross-section of 0.25 m2 or less as seen from the front.  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The transportation dolly will have an interface on the back left side.  The	Design and construction F-16 AM/BM fighter aircrafts in version constraints M6.5.  The transportation dolly will be able to fit under RDAF F-16 with the URP construction mounted on the plane, or carried on the constraints dolly.  Design and construction The mounting interface on the URP construction shall support a weight of 700 pounds.  Design and construction The pod shall have a mass less than constraints 700 pounds in total.  Design and The pod shall have a geometric construction constraints seen from the front.  Design and The pod should have a geometric construction cross-section of 0.40 m2 or less as constraints seen from the front.  Design and The pod should have a geometric cross-section of 0.25m2 or less as constraints seen from the front.  Design and The pod Should have a geometric cross-section of 0.25m2 or less as seen from the front.  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SLR-172	Logistics-related requirement	The transportation dolly shall be able to lift and lower a SLR for mount and dismounting on a F-16 (both when the aircraft is empty and when it is fully loaded).	R58		
SR-26	Logistics-related requirement	The transportation dolly will be mounted with a hydraulic lift able to lift at least 700 pounds.	R59		
SR-27	Logistics-related requirement	The transportation dolly will draw power from the transportation truck.	R60		
SR-28	Logistics-related requirement	The transportation dolly will be equipped with a scissor lift.	R61		

# 4. Relations

