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Introduction

# ntroduction

- Final Goal Assessment of forecasts and interpretability for different machine learning algorithms, including the probabilistic models
- **Method** Use a dataset for which the laws that interconnects the different features are known from general principles
- Dataset CNEOS asteroids dataset for more than 3500 asteroids
- Theoretical laws Celestial mechanics
- Algorithms involved probabilistic models GLASSO, mgm, minforest, mmod
- Algorithms involved others Random forest, Support Vector Machines, Quadratic Discriminant Analysis, Logistic Regression

## Celestial mechanics

$$\mathbf{F}_1 = \mathcal{G} \cdot \frac{m_1 m_2}{r^3} \mathbf{r} = m_1 \ddot{\mathbf{r}}_1 \tag{1}$$

$$\mathbf{F}_2 = -\mathcal{G} \cdot \frac{m_1 m_2}{r^3} \mathbf{r} = m_1 \ddot{\mathbf{r}}_2 \tag{2}$$

If we consider the motion of the second item with respect to the first one

$$\ddot{\mathbf{r}} = \ddot{\mathbf{r}}_2 - \ddot{\mathbf{r}}_1 \quad \mu = \mathcal{G}(m_1 + m_2) \tag{3}$$

$$\frac{d^2\mathbf{r}}{dt^2} + \mu \frac{\mathbf{r}}{r^3} = 0 \tag{4}$$

 $\mathbf{r} \times \ddot{\mathbf{r}} = 0 \implies \mathbf{r}$  and  $\dot{\mathbf{r}}$  lies in the same plane

With polar coordinates 
$$\hat{\mathbf{r}}$$
 and  $\hat{\boldsymbol{\theta}}$ 

$$\mathbf{r} = r\hat{\mathbf{r}} \tag{5}$$

$$\dot{\mathbf{r}} = \dot{r}\hat{\mathbf{r}} + r\dot{\theta}\hat{\boldsymbol{\theta}} \tag{6}$$

$$\ddot{\mathbf{r}} = \left(\ddot{r} - r\dot{\theta}^2\right)\hat{\mathbf{r}} + \left[\frac{1}{r}\frac{d}{dt}\left(r^2\dot{\theta}\right)\right]\hat{\boldsymbol{\theta}} \tag{7}$$

$$\mathbf{h} = r^2 \dot{\theta} \hat{\mathbf{z}} \tag{8}$$

$$h = r^2 \dot{\theta} \tag{9}$$



Figure 1: [4]

$$\delta A \approx \frac{1}{2}r(r+dr)\sin(\delta\theta) \approx \frac{1}{2}r^2\delta\theta$$
 (10)

$$\frac{dA}{dt} = \frac{1}{2}r^2\frac{d\theta}{dt} = \frac{1}{2}h\tag{11}$$

h is constant  $\implies 2^{th}$  Kepler law

# Celestial mechanics [4]: 1<sup>th</sup> Kepler law

Using the substitution  $u = \frac{1}{r} h = r^2 \dot{\theta}$ 

$$\dot{r} = -\frac{1}{u}\frac{du}{d\theta}\dot{\theta} = -h\frac{du}{d\theta} \tag{12}$$

$$\ddot{r} = -h\frac{d^2u}{d\theta^2}\dot{\theta} = -h^2u^2\frac{d^2u}{d\theta^2}$$
 (13)

$$\frac{d^2u}{d\theta^2} + u = \frac{\mu}{h^2} \tag{14}$$

$$u = \frac{\mu}{h^2} \left[ 1 + e \cos(\theta - \phi) \right] \tag{15}$$

$$r = \frac{p}{1 + e\cos(\theta - \phi)} \qquad (16)$$

e is eccentricity

- circle: e = 0 p = a
- ellipse: 0 < e < 1 $p = a(1 - e^2)$
- parabola: e = 1 p = 2q
- hyperbola: e > 1 $p = a(e^2 - 1)$

a is the semi-major axis of the conic

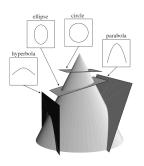


Figure 2: [4]

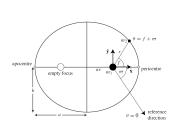


Figure 3: [4]

$$b^2 = a^2(1 - e^2) \tag{17}$$

$$r = \frac{a(1 - e^2)}{1 + e \cdot \cos(\theta - \phi)} \tag{18}$$

Area swept in one orbital period T

$$A = \pi ab$$

We know that: hT/2  $h^2 = \mu a(1 - e^2)$ 

Therefore

$$T^2 = \frac{4\pi^2}{\mu} a^3 \tag{19}$$

$$\frac{m_c + m}{m_c + m'} = \left(\frac{a}{a'}\right) \left(\frac{T'}{T}\right)^2 \tag{20}$$

But since  $m, m' \ll m_c$ 

$$(a)^3 \approx (T')^2 \tag{21}$$

And therefore

$$T' \approx a'^{3/2} \tag{22}$$

The mass of the asteroid is not involve

## Mean motion

$$n = \frac{2\pi}{T} \tag{23}$$

$$v_{perihelion} = na\sqrt{\frac{1+e}{1-e}} \tag{24}$$

$$v_{aphelion} = na\sqrt{\frac{1-e}{1+e}} \tag{25}$$

# $\begin{array}{cccc} & & & & & & \\ & & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ \end{array}$ circumsribed circle

Figure 4: [4]

## Mean anomaly

$$M = n(t - \tau) \tag{26}$$

- M = f = 0  $t = \tau$  Perihelion
- $M = f = \pi$   $t = \tau + T/2$ Aphelion

$$M = E - e \sin E \tag{27}$$

Jupiter Tisserard invariant

$$T_P = \frac{a_p}{a} + 2\cos I \sqrt{\frac{a}{a_p}(1 - e^2)}$$
 (28)

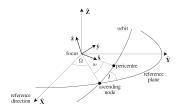


Figure 5: [4]

I: inclination of the orbit

 $\Omega$ : longitude of the ascending node

$$\Phi = \frac{L}{4\pi r^2} \tag{29}$$

$$m = -2.5 \log_{10} \Phi + C \tag{30}$$

$$m_1 - m_2 = -2.5 \log_{10} \frac{\Phi_1}{\Phi_2} \tag{31}$$

$$M - m = -2.5 \log_{10} \frac{\Phi \cdot d^2}{\Phi \cdot 10^2}$$
 (32)

$$M = m + 5 - 5\log_{10}d \tag{33}$$

Where  $\Phi$  is the flux for a sphere of radius r, m the relative magnitude and M the Absolute magnitude

$$\Phi = \frac{L}{4\pi r^2} \tag{34}$$

$$m = -2.5 \log_{10} \Phi + C \tag{35}$$

$$m_1 - m_2 = -2.5 \log_{10} \frac{\Phi_1}{\Phi_2} \tag{36}$$

$$M - m = -2.5 \log_{10} \frac{\Phi \cdot d^2}{\Phi \cdot 10^2} \tag{37}$$

$$M = m + 5 - 5\log_{10}d \tag{38}$$

Where  $\Phi$  is the flux for a sphere of radius r, m the relative magnitude and M the Absolute magnitude

## **Amors**

Earth-approaching NEAs with orbits exterior to Earth's but interior to Mars' (named after asteroid (1221) Amor)



$$\begin{array}{c} a > 1.0 \ \mathrm{AU} \\ 1.017 \ \mathrm{AU} < q < 1.3 \ \mathrm{AU} \end{array}$$

# **Apollos**

Earth-crossing NEAs with semi-major axes larger than Earth's (named after asteroid (1862) Apollo)



$$\begin{array}{l} a > 1.0 \text{ AU} \\ q < 1.017 \text{ AU} \end{array}$$

## **Atens**

Earth-crossing NEAs with semi-major axes smaller than Earth's (named after asteroid (2062) Aten)



 $\begin{array}{c} a < 1.0 \text{ AU} \\ Q > 0.983 \text{ AU} \end{array}$ 

## **Atiras**

NEAs whose orbits are contained entirely within the orbit of the Earth (named after asteroid (163693) Atira)



 $\begin{array}{c} a < 1.0 \ \mathrm{AU} \\ Q < 0.983 \ \mathrm{AU} \end{array}$ 

(q = perihelion distance, Q = aphelion distance, a = semi-major axis)

• Potentially Hazardous Asteroids: MOID < 0.05 au M < 22.0 NEAs whose Minimum Orbit Intersection Distance (MOID) with the Earth is 0.05 au or less and whose absolute magnitude (M) is 22.0 or brighter

- The asteroid dataset was retrieved from Kaggle [2], which reports into a more machine readable form the dataset of The Center for Near-Earth Object Studies (CNEOS) [3], a NASA research centre.
- 3552 Asteroids
- Among the 40 the features, the ones connected only to the other name of the asteroid, or connected only to the name of the orbit and the one connected with the orbiting planet ( since for all it was the Earth) were discarted
- The proportion hazardous/not hazardous was set 1:5
- The continuous measures were standardised and demeaned

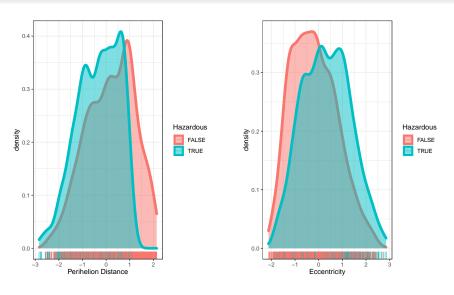
Introduction

Features	Туре
Neo Reference ID	not used
Absolute Magnitude	Continuous
Est Dia in KM (min)	Continuous
Est Dia in KM (max)	Continuous
Close Approach Date	Continuous
Epoch Date Close Approach	Continuous
Relative_Velocity	Continuous
Miss_Dist	Continuous
Min_Orbit_Intersection	Continuous
Jupiter_Tisserand_Invariant	Continuous
Epoch_Osculation	Continuous
Eccentricity	Continuous

Introduction

Features	Туре
Semi Major Axis	Continuous
Inclination	Continuous
Asc Node Longitude	Continuous
Orbital Period	Continuous
Perihelion Distance	Continuous
Perihelion Arg	Continuous
Perihelion Time	Continuous
Mean_Anomaly	Continuous
Mean_Motion	Continuous
Hazardous	Categorical (Binary)

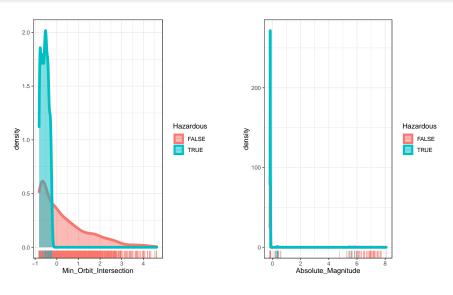




a) Perihelion Distance

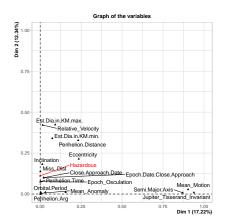
b) Eccentricity

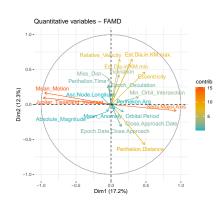




c) Min orbit intersection

d) Absolute magnitude





- [1] https://cneos.jpl.nasa.gov/about/neo\_groups.html.
- [2] https://www.kaggle.com/shrutimehta/nasa-asteroids-classification.
- [3] https://cneos.jpl.nasa.gov/.
- [4] Carl D Murray and Stanley F Dermott. *Solar system dynamics*. Cambridge university press, 1999.