

Hazardous asteroids forecast via Markov random fields

Project for the exam: Probabilistic Modelling (DSE)

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Introduction

- **Final Goal** Assessment of forecasts and interpretability for different machine learning algorithms, including the probabilistic models
- **Method** Use a dataset for which the laws that interconnects the different features are known from general principles
- **Dataset** CNEOS asteroids dataset for more than 3500 asteroids
- **Theoretical laws** Celestial mechanics
- **Algorithms involved - probabilistic models** GLASSO, mgm, minforest, mmod
- **Algorithms involved - others** Random forest, Support Vector Machines, Quadratic Discriminant Analysis, Logistic Regression

Celestial mechanics

$$\mathbf{F}_1 = \mathcal{G} \cdot \frac{m_1 m_2}{r^3} \mathbf{r} = m_1 \ddot{\mathbf{r}}_1 \quad (1)$$

$$\mathbf{F}_2 = -\mathcal{G} \cdot \frac{m_1 m_2}{r^3} \mathbf{r} = m_1 \ddot{\mathbf{r}}_2 \quad (2)$$

If we consider the motion of the second item with respect to the first one

$$\ddot{\mathbf{r}} = \ddot{\mathbf{r}}_2 - \ddot{\mathbf{r}}_1 \quad \mu = \mathcal{G}(m_1 + m_2) \quad (3)$$

$$\frac{d^2 \mathbf{r}}{dt^2} + \mu \frac{\mathbf{r}}{r^3} = 0 \quad (4)$$

$\mathbf{r} \times \ddot{\mathbf{r}} = 0 \implies \mathbf{r}$ and $\dot{\mathbf{r}}$ lies in the same plane

Celestial mechanics

With polar coordinates $\hat{\mathbf{r}}$ and $\hat{\boldsymbol{\theta}}$

$$\mathbf{r} = r\hat{\mathbf{r}} \quad (5)$$

$$\dot{\mathbf{r}} = \dot{r}\hat{\mathbf{r}} + r\dot{\theta}\hat{\boldsymbol{\theta}} \quad (6)$$

$$\ddot{\mathbf{r}} = \left(\ddot{r} - r\dot{\theta}^2\right)\hat{\mathbf{r}} + \left[\frac{1}{r}\frac{d}{dt}\left(r^2\dot{\theta}\right)\right]\hat{\boldsymbol{\theta}} \quad (7)$$

$$\mathbf{h} = r^2\dot{\theta}\hat{\mathbf{z}} \quad (8)$$

$$h = r^2\dot{\theta} \quad (9)$$

Celestial mechanics [1]: 2th Kepler law

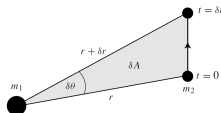


Figure 1: [1]

$$\delta A \approx \frac{1}{2}r(r + dr) \sin(\delta\theta) \approx \frac{1}{2}r^2 \delta\theta \quad (10)$$

$$\frac{dA}{dt} = \frac{1}{2}r^2 \frac{d\theta}{dt} = \frac{1}{2}h \quad (11)$$

h is constant \implies 2th Kepler law

Celestial mechanics [1]: 1th Kepler law

Using the substitution $u = \frac{1}{r}$ $h = r^2 \dot{\theta}$

$$\dot{r} = -\frac{1}{u} \frac{du}{d\theta} \dot{\theta} = -h \frac{du}{d\theta} \quad (12)$$

$$\ddot{r} = -h \frac{d^2 u}{d\theta^2} \dot{\theta} = -h^2 u^2 \frac{d^2 u}{d\theta^2} \quad (13)$$

$$\frac{d^2 u}{d\theta^2} + u = \frac{\mu}{h^2} \quad (14)$$

$$u = \frac{\mu}{h^2} [1 + e \cos(\theta - \phi)] \quad (15)$$

Celestial mechanics [1]: 1th Kepler law

$$r = \frac{p}{1 + e \cos(\theta - \phi)} \quad (16)$$

e is **eccentricity**

- circle: $e = 0$ $p = a$
- ellipse: $0 < e < 1$ $p = b$
- parabola: $e = 1$ $p = 2q$
- hyperbola: $e > 1$
 $p = a(e^2 - 1)$

b is the **semi-major axis** of the conic

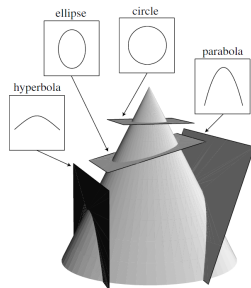


Figure 2: [1]

Celestial mechanics [1]: 3th Kepler law

$$b^2 = a^2(1 - e^2) \quad (17)$$

$$r = \frac{a(1 - e^2)}{1 + e \cdot \cos(\theta - \phi)} \quad (18)$$

Area swept in one **orbital period** $T \implies A = \pi ab$ We know that:

$$hT/2 \quad h^2 = \mu a(1 - e^2)$$

Therefore

$$T^2 = \frac{4\pi^2}{\mu} a^3 \quad (19)$$

3th Kepler law

$$\frac{m_c + m}{m_c + m'} = \left(\frac{a}{a'}\right) \left(\frac{T'}{T}\right)^2 \quad (20)$$

But since $m, m' \ll m_c$

$$(a')^3 \approx (T')^2 \quad (21)$$

And therefore

$$T' \approx a'^{3/2} \quad (22)$$

Orbital parameters

Mean motion

$$n = \frac{2\pi}{T} \quad (23)$$

$$v_{perihelion} = na\sqrt{\frac{1+e}{1-e}} \quad (24)$$

$$v_{aphelion} = na\sqrt{\frac{1-e}{1+e}} \quad (25)$$

Orbital parameters

Mean anomaly

$$M = n(t - \tau) \quad (26)$$

- $M = f = 0 \quad t = \tau$ Perihelion
- $M = f = \pi \quad t = \tau + T/2$ Aphelion

$$M = E - e \sin E \quad (27)$$

Jupiter Tisserard invariant

$$T_P = \frac{a_p}{a} + 2 \cos I \sqrt{\frac{a}{a_p} (1 - e^2)} \quad (28)$$

References I

- [1] Carl D Murray and Stanley F Dermott. *Solar system dynamics*. Cambridge university press, 1999.