Power system Stability

Power System Stability

- In order to operate as an interconnected system all of the generators (and other synchronous machines) must remain in synchronism with one another.
- Loss of synchronism results in a condition in which no net power can be transferred between the machines.
- A system is said to be transiently unstable if following a disturbance one or more of the generators lose synchronism.

Types of Disturbances

- Loss of generation
- □ Line switching
- □ Faults
- Sudden load changes

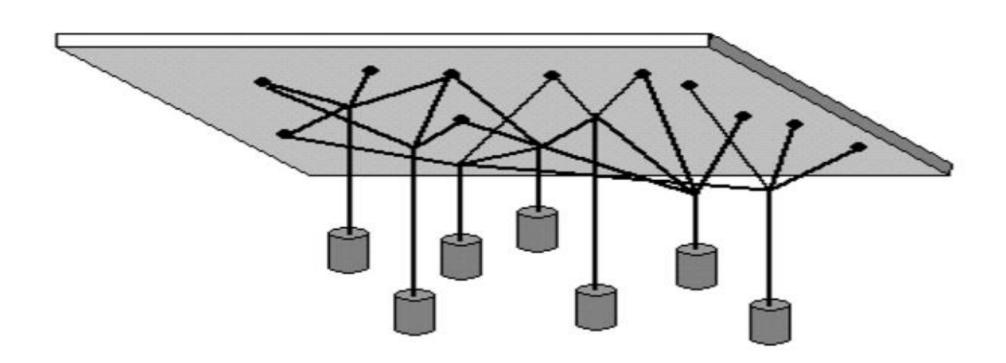
Definition

Power system stability refers to the ability of the synchronous machines to move from one steadystate operating point following a disturbance to another steady-state operating point, without losing synchronism.

Power System Stability

- Stability problems are generally divided into two major categories:
 - Steady state stability.
 - Transient stability.
- Steady state stability refers to the ability of the power system to regain synchronism after small and slow disturbance like gradual power changes.
- Transient stability refers to the ability of the power system to regain synchronism after large and sudden disturbance like the occurrence of a fault.

Mechanical Analogy of PSTS



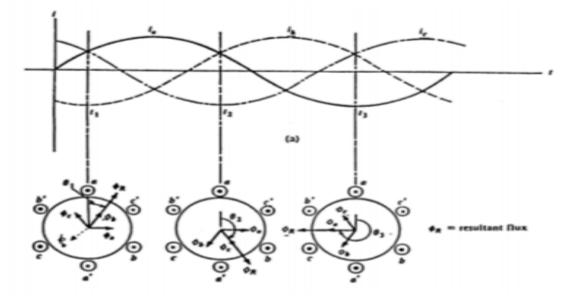
Power System Stability

- Number of masses representing synchronous machines are interconnected by a network of elastic strings representing transmission lines.
- Assume the system is initially at rest with the net force on each string is below its break point.
- If one of the strings is cut, representing the loss of a transmission line, the masses will undergo transient oscillations and the forces on the strings will fluctuate.
- The system will either settle down to a new steady state operating point or additional strings will break and the system might collapse, completely.

Power System Transient Stability

- In order to operate as an interconnected system all of the generators must remain in synchronism with one another
 - synchronism requires that the rotors turn at exactly the same speed
- Loss of synchronism results in a condition in which no net power can be transferred between the machines
- A system is said to be transiently unstable if following a disturbance one or more of the generators lose synchronism

Under normal operating conditions, the relative position of the rotor axis and the resultant magnetic field is fixed. This angle is called the power angle.



- During any disturbance, rotor will accelerate or decelerate with respect to the synchronously rotating air gap mmf.
- The equation describing the relative motion is called the swing equation.

Consider a synchronous generator developing an electromagnetic torque T_e and running at the synchronous speed. If T_m is the mechanical torque, then under steady state operation with losses being neglected:

$$T_m = T_e$$

 \Box A deviation from steady state due to a disturbance results in an accelerating (T_m > T_e) or decelerating (T_m < T_e) torque T_\alpha on the rotor: $T_\alpha = T_m - T_e$

- If J is the moment of inertia of the rotor, from law's of rotation we have: $J\frac{d^2\theta_m}{dt^2}=T_m-T_e$
- Where θ_{m} is the angular displacement of the rotor with respect to the stationary reference axis of the stator.
- Since we are interested in the rotor speed relative to the synchronous speed, the angular reference is chosen relative to a synchronously rotating frame moving with constant angular velocity ω_{sm}:

$$\theta_m = \omega_{sm}t + \delta_m$$

Where δ_m is the rotor position before disturbance.

Derivative of the previous equation results in:

$$\omega_m = \frac{d\theta_m}{dt} = \omega_{sm} + \frac{d\delta_m}{dt}$$

□ And the rotor acceleration is:
$$\frac{d^2\theta_m}{dt^2} = \frac{d^2\delta_m}{dt^2}$$

$$\int \frac{d^2\delta_m}{dt^2} = T_m - T_e$$

 \square Multiplying the previous equation with ω_{m} :

$$J\omega_{\scriptscriptstyle m}\,\frac{d^2\delta_{\scriptscriptstyle m}}{dt^2} = \omega_{\scriptscriptstyle m}T_{\scriptscriptstyle m} - \omega_{\scriptscriptstyle m}T_{\scriptscriptstyle e} \qquad \qquad \Longrightarrow \qquad J\omega_{\scriptscriptstyle m}\,\frac{d^2\delta_{\scriptscriptstyle m}}{dt^2} = P_{\scriptscriptstyle m} - P_{\scriptscriptstyle e}$$

$$\therefore M \frac{d^2 \delta_m}{dt^2} = P_m - P_e \quad \text{Where M is called the inertia constant}$$

An other constant which is often used because its range of values for particular types of rotating machines is quite narrow is called normalized inertia constant H. It is related to M as follows:

 $H = \frac{1}{2} \cdot \frac{M\omega_{sm}}{S_{rated}}$

□ So the swing equation will be:

$$M \frac{d^2 \delta_m}{dt^2} = \left(\frac{2HS_{rated}}{\omega_{sm}}\right) \frac{d^2 \delta_m}{dt^2}$$

$$\frac{2H}{\omega_{sm}} \frac{d^2 \delta_m}{dt^2} = P_{m(p.u.)} - P_{e(p.u.)}$$

 \square Depending on the unit of the angle δ , the previous equation can be either:

$$\frac{H}{\pi f} \frac{d^2 \delta}{dt^2} = P_{m(p.u.)} - P_{e(p.u.)}$$

When δ is in electrical radian

OR

$$\frac{H}{180.f} \frac{d^2 \delta}{dt^2} = P_{m(p.u.)} - P_{e(p.u.)}$$

When δ is in electrical degree

- Consider a system that consists of a single machine connected to an infinite bus through an external reactance.
- The electrical power output may be expressed as follows:

$$P_e = P_{\text{max}} \sin \delta$$

Substituting the expression for the electrical power output in the swing equation will result in:

$$\frac{2H}{\omega_{s}} \frac{d^{2}\delta}{dt^{2}} + P_{\text{max}} \sin \delta = P_{m}$$

Example 1

A 500 MVA, 20 kV, 60 Hz, four pole synchronous generator is connected to an infinite bus through a purely reactive network. The generator has an inertia constant H = 6 MJ/MVA and is delivering power of 1.0 per unit to the infinite bus at steady state. The maximum power that can be delivered is 2.5 per unit. A fault occurs that reduces the generator output power to zero, find:

- a) The angular acceleration
- ы The speed in rev/min at the end of 15 cycles.
- c) The change in the power angle at the end of the 15 cycles.

Example 1, Solution

At steady state
$$\longrightarrow$$
 $P_m = P_e = 1.0$

Under fault conditions the accelerating torque is found as follows:

$$\alpha = \frac{d^2 \delta}{dt^2} = \frac{\pi . f}{H} (P_m - P_e) \qquad \text{or} \qquad \alpha = \frac{180. f}{H} (P_m - P_e)$$

$$\alpha = \frac{180.60}{6} (1-0) = 1800 \text{ elec. degrees/s}^2$$

$$\alpha = \frac{2}{P} (1800) = 900 \text{ mech. degrees/s}^2$$

$$\alpha = \frac{2}{P}(1800) = 900 \text{ mech. degrees/s}^2$$

Example 1, Solution

A 15-cycle interval is equivalent to a time interval of:

$$t = (15)(1/60) = 0.25 \,\mathrm{s}$$

The synchronous speed of the machine is found as follows

$$\omega_{sm} = \frac{120.f}{P} = \frac{120.60}{4} = 1800 \,\text{rpm}$$

$$\omega_m = \omega_{sm} + \frac{d\delta_m}{dt}$$
 $\omega_m = \omega_{sm} + \alpha t = 1800 + (150)(0.25) = 1837.5 \text{ rpm}$

Example 1, Solution

The machine is initially operating at δ_0 which can be found as follows:

$$P_o = P_{\text{max}} \sin \delta_o$$

Therefore the initial angle is:

$$\delta_o = \sin^{-1}(P_o/P_{\text{max}}) = \sin^{-1}(1/2.5) = 23.6^\circ$$

Integrating the following equation $\omega_m = \omega_{sm} + \alpha t$

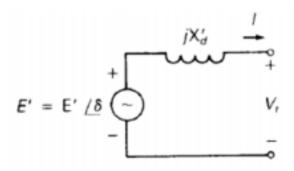
$$\omega_m = \omega_{sm} + \alpha t$$

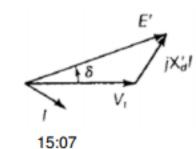
$$\delta = \delta_o + \frac{1}{2}\alpha t^2 = 79.83^o$$

Simplified Synchronous Machine Model and

System Equivalent

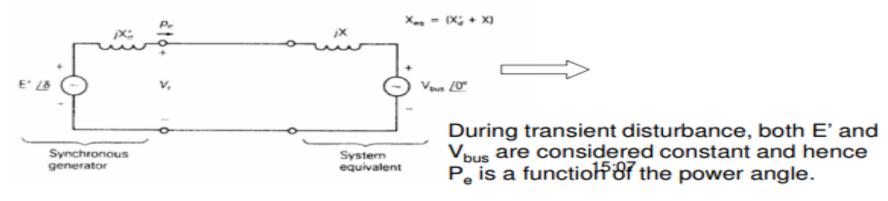
- The simplified model of a synchronous machine along with its phasor diagram is shown below:
- □ The model is based on the following assumptions:
 - The machine is operating under balanced three-phase positive sequence conditions.
 - Machine excitation is constant
 - Machines losses and saturation are neglected.





Simplified Synchronous Machine Model and System Equivalent

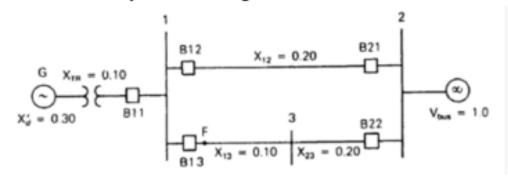
- Each generator in the model is connected to a system consisting of transmission lines, transformers, loads and other machines.
- The system can be represented by an "infinite bus" behind a system of reactance.
- An infinite bus is an ideal voltage source that maintains constant voltage magnitude, constant phase and constant frequency.
 Typical system is shown in the following figure:
 P_e = \frac{EV_{but}}{X_{ex}} \sin \delta
- Typical system is shown in the following figure:



Example 2

The system shown below is a three phase, 60 Hz synchronous generator connected through a transformer and parallel transmission lines to an infinite bus. All reactances are given in per unit on a common system base. If the infinite bus receives 1.0 per unit real power at 0.95 p.f. lagging determine:

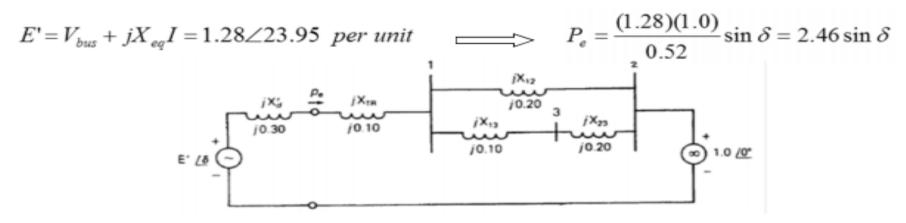
- a) The internal voltage of the generator.
- The equation for the electrical power delivered by the generator versus its power angle.



The equivalent reactance diagram is shown below:

$$\begin{split} X_{eq} &= X'_{d} + X_{TR} + X_{12} \big\| \big(X_{13} + X_{23} \big) = 0.520 \quad per \ unit \\ I &= \frac{P}{V_{bus}(p.f.)} \angle - \cos^{-1}(p.f.) = \frac{1}{1(0.95)} \angle - \cos^{-1}(0.95) = 1.053 \angle - 18.2 \quad per \ unit \end{split}$$

The machine internal voltage is:



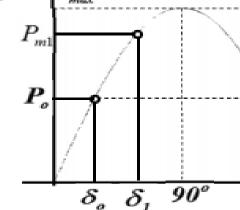
Transient Stability

Consider a power angle curve of a generator as shown in the Figure. Under steady-state operation:



Mechnical input = Electrical output

$$P_m = P_o$$
 at $\delta = \delta_o < 90^\circ$ sychronous Speed



$$At \quad t = t$$

If the generator mechanical input power is increased to P_{m1} to supply P_{e1}

$$At \quad t = t_{o+}$$

Due to the rotor inertia, the rotor angle δ can not adjust instantaneously.

$$\mathcal{S} = \mathcal{S}_o$$

$$Mechnical\ Input = P_{m1} & & Output = P_o$$

$$P_{m1} \ > \ P_{o1}$$

The rotor starts accelerating and the rotor angle δ starts increasing.

At
$$\delta = \delta_1$$

Mechnical Input (P_{m1}) = Electrical output (P_{e1})

The rotor can not stop at $\delta = \delta_1$ due to its inertia and δ continuous to increase up to $\delta = \delta_2 > \delta_1$.

There are two cases for δ swing:

Case 1:
$$\delta_2 < \delta_m$$

 $Mechnical\ Input(P_{m1}) < Electrical\ output(P_{e1})$

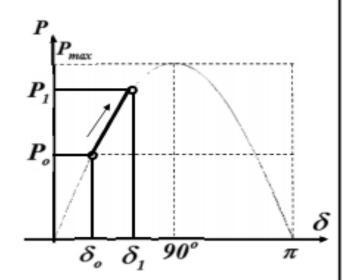
The rotor starts decelerating and the rotor angle δ starts decreasing.

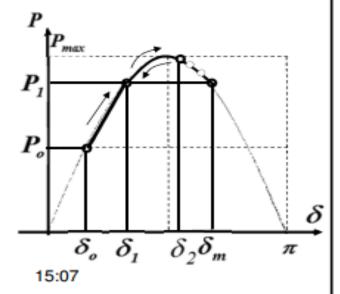
The rotor angle will oscillate around δ_i and the system losses will damp this oscillation.

Mechnical Input(P_{m1}) = Electrical output(P_{e1}) Steady State and Synchronios Speed

NOTE

The maximum swing of the rotor angle beyond δ_i depends on the amount of sudden increase in input.



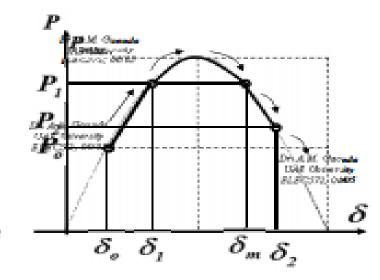


Mechnical Input $(P_{ij})^{ij} > Electrical output(P_{ij})$

The rotor starts accelerating and the rotor angle starts increasing.

Mechinical Input $(P_n) >> Electrical Summet (P_n)$

The rotor angle will further increase beyond δ_2 and the system becomes unstable.



NOTES

1 the changes in the input are generally small, therefore can not can be transient stability problem.

2, The other cases, the system are subject to transient changes (such as faults and switching or disconnecting tie lines which may cause transient stability problem. Equal Area Criteria is used to study the transient stability problem.

Equal-Area for Transient Stability Assessment:

- In order to predict whether a particular system is stable after a disturbance it is necessary to solve the dynamic equation describing the behavior of the rotor angle immediately following an imbalance or disturbance to the system.
- The system is said to be unstable if the angle between any two machines tends to increase without limit.
- On the other hand, if under disturbance effects, the angles between every possible pair reach maximum value and decrease thereafter, the system is stable.

Reminder "Stability"

A power system is stable if it reaches another stable operating point after a disturbance

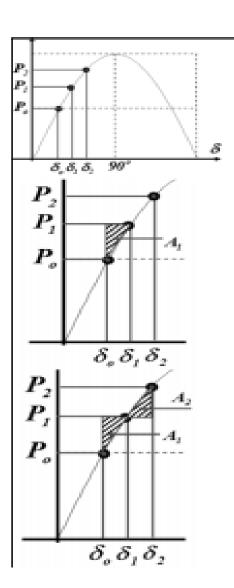
A simple method for determining stability is known as the equal-area method.

Consider a power angle curve of a system working under steady-state condition where: $Mechnical\ input = Electrical\ output$

$$\delta = \delta_o$$

15:07

sychronous Speed



If the mechanical input for the generator is increased.

 $Mechnical\ Input = P_{m1}$

At
$$\delta = \delta_o$$

Mechnical Input > Electrical Output

The rotor starts accelerating and the rotor angle starts increasing from $\delta = \delta_o$ to $\delta = \delta_I$. During the time taken by the load angle to increase from δ_o to δ_I , the rotor absorbs kinetic energy equal the area defined as AI in the power angle curve.

At
$$\delta = \delta_i$$

Mechnical Input = Electrical Output

The rotor has gained some energy that will drive its speed slightly beyond the synchronous speed. The rotor angle ($\delta = \delta_1$) continues to increase until the rotor releases the kinetic energy stored in it ($\delta = \delta_2$). This energy is defined by area A25.07

At
$$\delta = \delta$$

Mechnical Input < Electrical Output

The rotor starts decelerating and the rotor angle (δ) starts decreasing to ($\delta = \delta_I$). Area 1 must equal to Area 2.

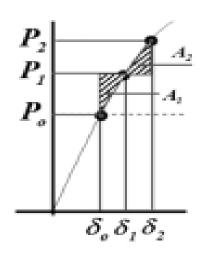
The area A1 is:

$$\begin{split} A_I &= \int_{\delta_o}^{\delta_I} P_I \quad d\delta - \int_{\delta_o}^{\delta_I} P_{\max} \sin(\delta) \ d\delta \\ A_I &= P_I (\delta_I - \delta_o) - \left[-P_{\max} \cos(\delta) \right]_{\delta_o}^{\delta_I} \\ A_1 &= P_1 (\delta_1 - \delta_o) - P_{\max} (\cos \delta_o - \cos \delta_1) \end{split}$$

The area A2 is:

$$A_2 = \int_{\delta_1}^{\delta_2} P_{max} \sin(\delta) d\delta - \int_{\delta_l}^{\delta_2} P_l d\delta$$

$$A_2 = P_{max} (\cos \delta_1 - \cos \delta_2) - P_1(\delta_2 - \delta_1)$$



For stability, The area A1 (energy stored) = The area A2 (energy released)

$$P_1(\delta_1 - \delta_o) - P_{\max}(\cos \delta_o - \cos \delta_1) = P_{\max}(\cos \delta_1 - \cos \delta_2) - P_1(\delta_2 - \delta_1)$$

$$P_{1}\delta_{1}-P_{1}\delta_{o}-P_{\max}\cos\delta_{o}+P_{\max}\cos\delta_{1}= P_{\max}\cos\delta_{1}-P_{\max}^{-15:07}\cos\delta_{2}-P_{1}\delta_{2}+P_{1}\delta_{1}$$

$$P_{\text{diff}}(\cos \delta_o - \cos \delta_{2000}) = P_{\text{diff}}(\delta_2 - \delta_o)$$

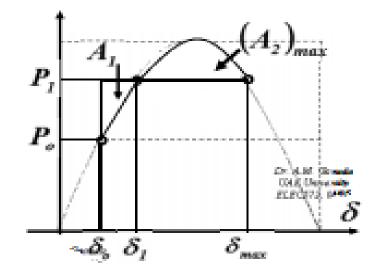
The angles are in radian.

NOTE:

For a given (δ_o), the maximum value of the swing is given by:

$$\mathcal{S}_{\text{womax}}^{\text{de}} = \pi - \delta_1$$

If the load angle (δ) does not go beyond (δ_{max}), then the rotor will swing back to requilibrium (steady-state condition) at ($\delta = \delta_i$).



The equal area condition is satisfied. The system remains within the stability

Example 3

Consider the system shown below:

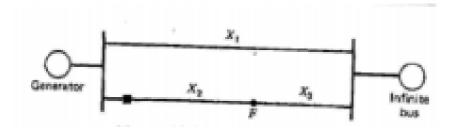
$$X_1=0.4$$
, $X_2=0.2$, $X_3=0.2$, $X_d'=0.2$

E (generator internally generated voltage)=1.2

$$V$$
 (at the load) = 1.0

$$P_{\rm m} = 1.5$$

Suppose the system is working at steady state at δ_1 when the breaker due to a fault at point F opens. Find δ_1 , δ_2 (the power angle under the new condition) and δ_4 (the maximum power angle when the breaker opens. Is the system stable or not?



At
$$t = 0^{-}$$

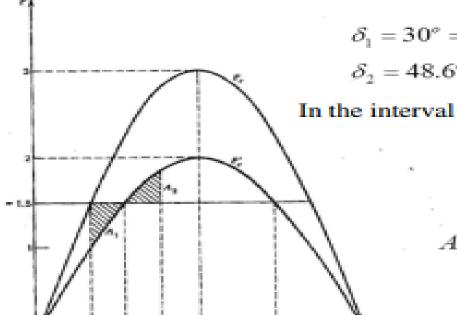
$$X_{tot} = \frac{(0.2 + 0.2)(0.4)}{.2 + 0.2 + 0.4} = 0.2$$

$$P_e = \frac{E \times V}{X} \sin \delta_1 = P_m = 1.5 \quad \Longrightarrow \quad P_e = \frac{1.2 \times 1.0}{0.2 + 0.2} \sin \delta_1 = 3.0 \times \sin \delta_1 = 1.5$$

$$\Longrightarrow$$
 $\delta_1 = 30^\circ$

When the breaker opens:

$$P'_e = \frac{1.2 \times 1.0}{0.2 + 0.4} \sin \delta_2 = 2.0 \times \sin \delta_2 = 1.5 \implies \delta_2 = 48.6^\circ \implies \delta_4 = 131.4^\circ$$



nge the angles to radian

$$\delta_1 = 30^\circ = 0.524 \, \text{rad}$$

$$\delta_4 = 131.4^\circ = 2.293 \, \text{rad}$$

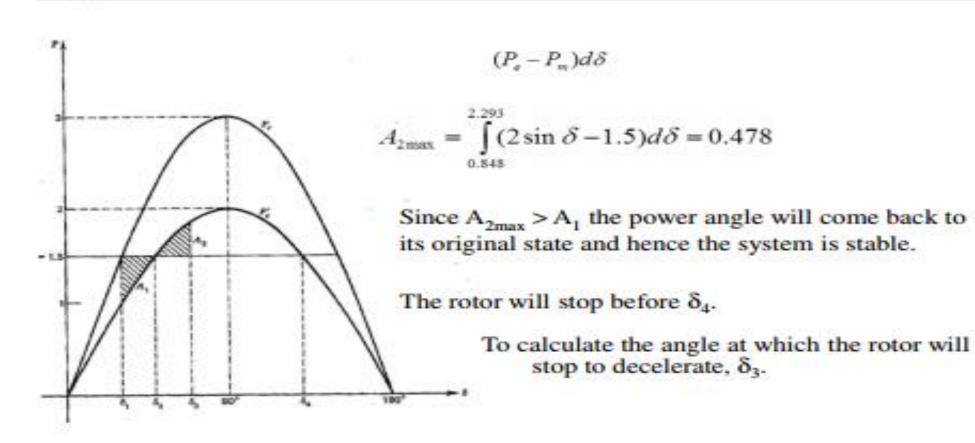
$$\delta_2 = 48.6^{\circ} = 0.848 \, rad$$

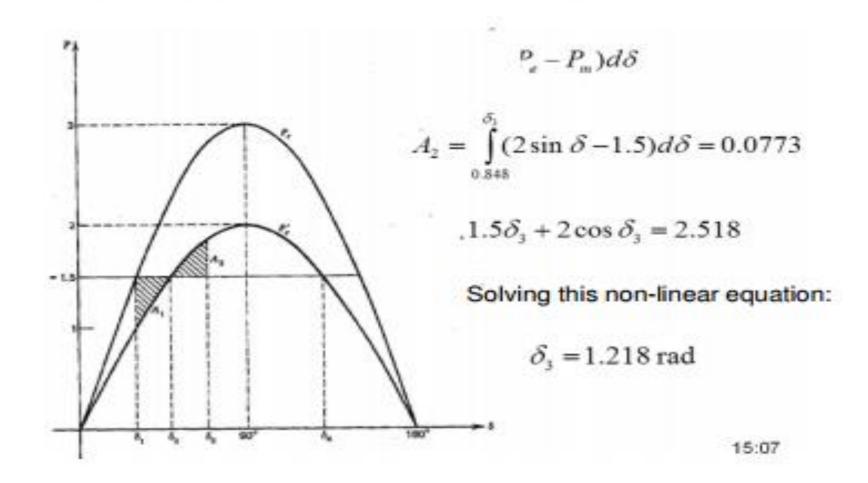
In the interval δ_1 - δ_2 , $P_m > P_e$ and the rotor is accelerating.

$$(P_m - P_e)d\delta$$

$$A_1 = \int_{0.524}^{0.848} (1.5 - 2\sin \delta) d\delta = 0.0773$$

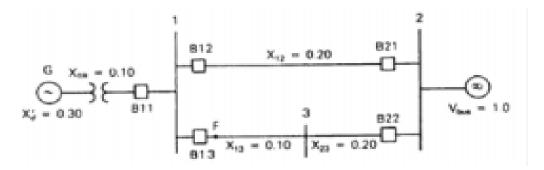
The question is, will A_2 will be large enough before reaching δ_4 15:07





Example 4

The synchronous generator shown below is initially operating in the steady state condition given in example 2 when a temporary three phase-to-ground fault occurs near bus-1. Three cycles later the fault extinguishes by itself. Due to a relay mis-operation all circuit breakers remain closed. Determine whether stability is maintained or not.



Example 4, solution

Plots of P_e and P_m are shown below:

$$P_e = \frac{(1.28)(1.0)}{0.52} \sin \delta = 2.46 \sin \delta$$

The initial torque angle is: $\delta_o = 23.95^\circ = 0.4179 \ radian$

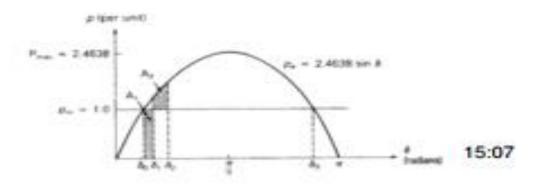
$$\delta_o = 23.95^\circ = 0.4179 \ radian$$

When the short circuit occurs, P_e drops to zero and the swing equation will be:

$$\frac{2H}{\omega_{syn}} \frac{d^2 \delta(t)}{dt^2} = P_{n\varphi,u}.$$

Integrating twice

$$\delta(t) = \frac{\omega_{sym} P_{mp.u.}}{4H} t^2 + \delta_0$$

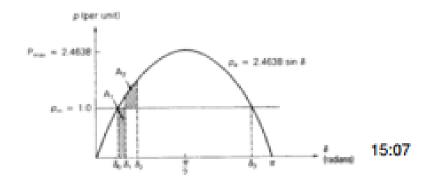


Example 4, solution

At t = 3 cycles = 0.05 second
$$\delta_1(0.05s) = \frac{\omega_{zyn}P_{sup.ss}}{4H}t^2 + \delta_0 = 0.4964 = 28.44^0$$

The acceleration area A1 is equal to: $A_1 = \int_{\delta_0}^{\delta_1} P_n d\delta = \int_{\delta_0}^{\delta_1} 1.0 d\delta = 0.4964 - 0.4179 = 0.0785$

At t=0.05 s the fault extinguishes and P_e instantaneously increases from zero to sinusoidal curve and the torque angle continues to increase until the decelerating A2 equals A1.

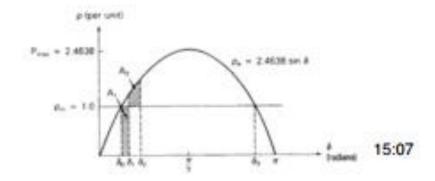


Example 4, solution

$$A_2 = \int_{\delta_1}^{\delta_2} (P_{\text{max}} \sin \delta - P_m) d\delta = \int_{0.4964}^{\delta_2} (2.4638 \sin \delta - 1) d\delta = A_1 = 0.0785$$

$$2.4638\cos \delta_2 + \delta_2 = 0.0785$$
 $\delta_2 = 0.7003 = 40.12^\circ$

Sine the torque angle is less than the maximum angle (180-23.95=156.05), the system is stable

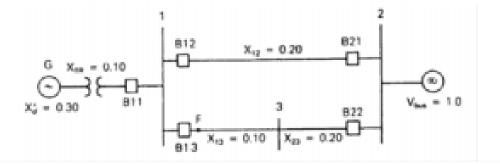


Critical Clearing Time

- In the previous example the fault took 3 cycles before it cleared itself. What will happen if the fault was not cleared?
- This will lead to the following question: what is the longest allowed time for the fault after which the system will lose its stability?
- This time is called critical clearing time and will be discussed in the following example.

Example 5

For the question in example 4, assume the short circuit was not cleared after 3 cycles, calculate the critical clearing time.

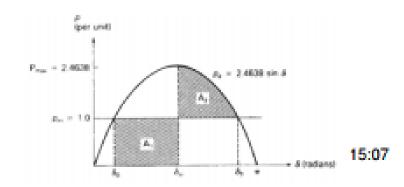


Example 5, solution

- At the critical clearing angle the fault is extinguished.
- □ The power angle is then increases to a maximum of 180-23.95=156.05 = 2.7236 radian which gives the maximum decelerating area.
- Equating the accelerating and decelerating areas will result in:

$$A_1 = \int_{\delta_0}^{\delta_\sigma} P_m d\delta = A_2 = \int_{\delta_\sigma}^{\delta_2} (P_{\max} \sin \delta - P_m) d\delta \qquad \Longrightarrow \qquad \int_{0.4179}^{\delta_\sigma} 1.0 d\delta = A_2 = \int_{\delta_\sigma}^{2.7236} (2.4638 \sin \delta - 1.0) d\delta$$

$$\delta_{cc} = 1.5489 = 88.74^{\circ}$$



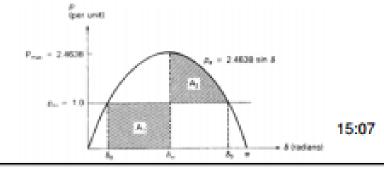
Example 5, solution

From the solution to the swing equation given in the previous example:

$$\delta(t) = \frac{\omega_{\text{sym}} P_{\text{nsp.u.}}}{4H} t^2 + \delta_0 \qquad \longrightarrow \qquad t = \sqrt{\frac{4H}{\omega_{\text{sym}} P_{\text{nsp.u.}}}} (\delta(t) - \delta_0)$$

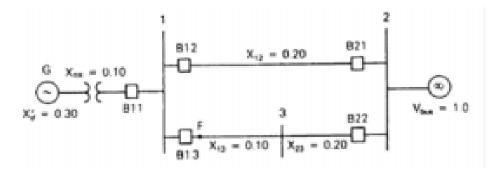
$$t_{cr} = \sqrt{\frac{4H}{\omega_{syn}P_{mp.u.}}} (\delta_{cr} - \delta_0) = 0.1897 s = 11.38 cycles$$

So if the fault is cleared before the critical clearing time (11.38 cycles), stability is maintained.



Example 6

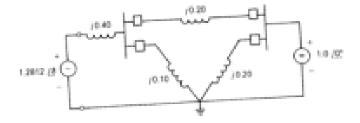
The synchronous generator shown below is initially operating in the steady state condition given in example 2 when a permanent three phase-to-ground fault occurs on bus 1-3 at bus-3. The fault is cleared by opening the circuit breaker at the ends of line 1-3 and line 2-3. These circuit breakers then remain open. Calculate the critical clearing angle.



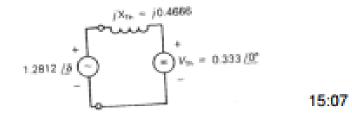
Example 6, solution

From example 2:
$$P_{e1} = \frac{(1.28)(1.0)}{0.52} \sin \delta = 2.46 \sin \delta$$

The faulted network is shown below:



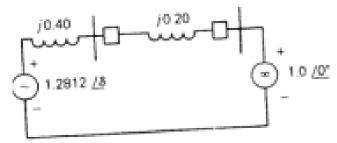
The Thevenin equivalent of the faulted network is:



Example 6, solution

So:
$$P_{e2} = \frac{(1.28)(0.333)}{0.4666} \sin \delta = 0.915 \sin \delta$$

The post-fault network is shown below:

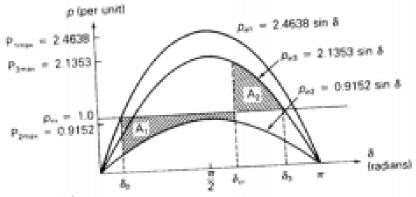


So, the post-fault electrical power delivered is:

$$P_{e3} = \frac{(1.28)(1.0)}{0.6} \sin \delta = 2.1353 \sin \delta$$

Example 6, solution

The p- δ curves as well as the accelerating are A1 and decelerating area A2 corresponding to critical clearing are shown below:



$$A_1 = \int_{\delta_\sigma}^{\delta_\sigma} (P_m - P_{2\max} \sin \delta) d\delta = A_2 = \int_{\delta_\sigma}^{\sigma_3} (P_{3\max} \sin \delta - P_m) d\delta$$

$$\int_{0.4179}^{\delta_{cr}} (1 - 0.9152 \sin \delta) d\delta = \int_{\delta_{cr}}^{2.6542} (2.1353 \sin \delta - 1) d\delta \qquad \longrightarrow \delta_{cr} = 1.98 \text{ rad} = 113.5^{\circ}$$