

BOEING PROPRIETARY

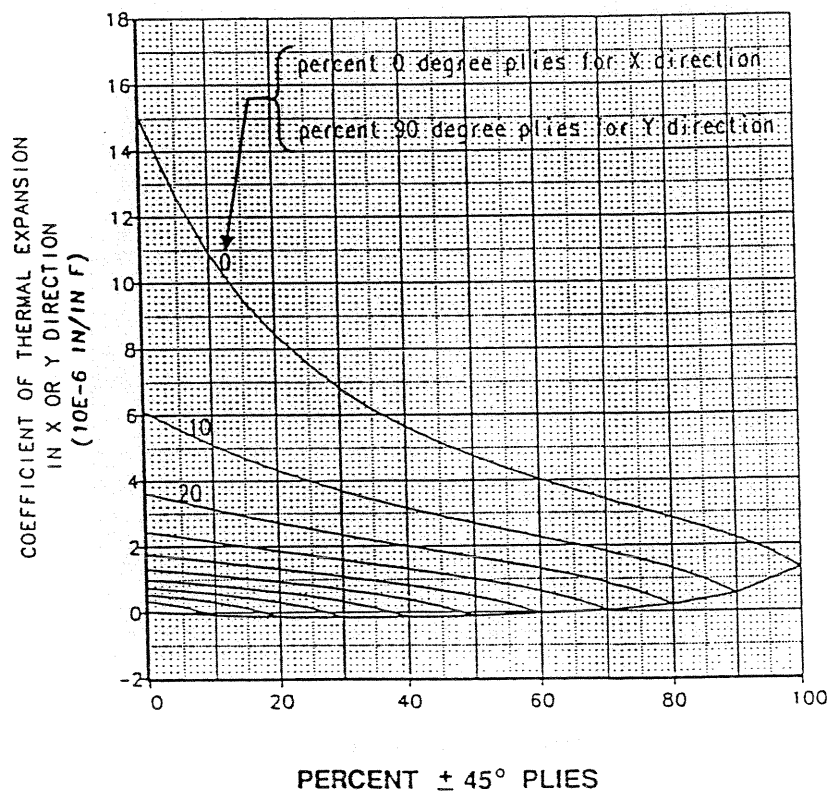
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE II, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 35 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



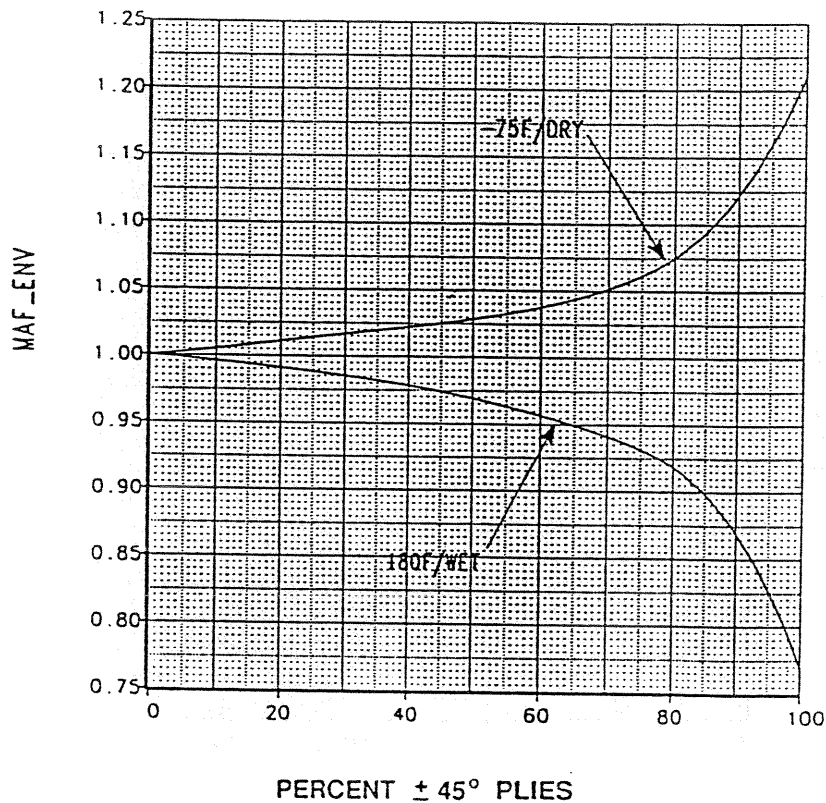
DG-100-527

DG-100-527

Figure 2.2.1-4 Nominal α_x Coefficient of Thermal Expansion

BOEING PROPRIETARY
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE II, CLASS 1 or 3, GRADE 145 or 190	
Resin Content:	35 (% WT)
Process Specification:	BAC 5317-1
(0/±45/90) LAMINATES	



$$E_x \text{ ENVIRONMENT} = E_x \text{ NOMINAL} \times \text{MAF_ENV}$$

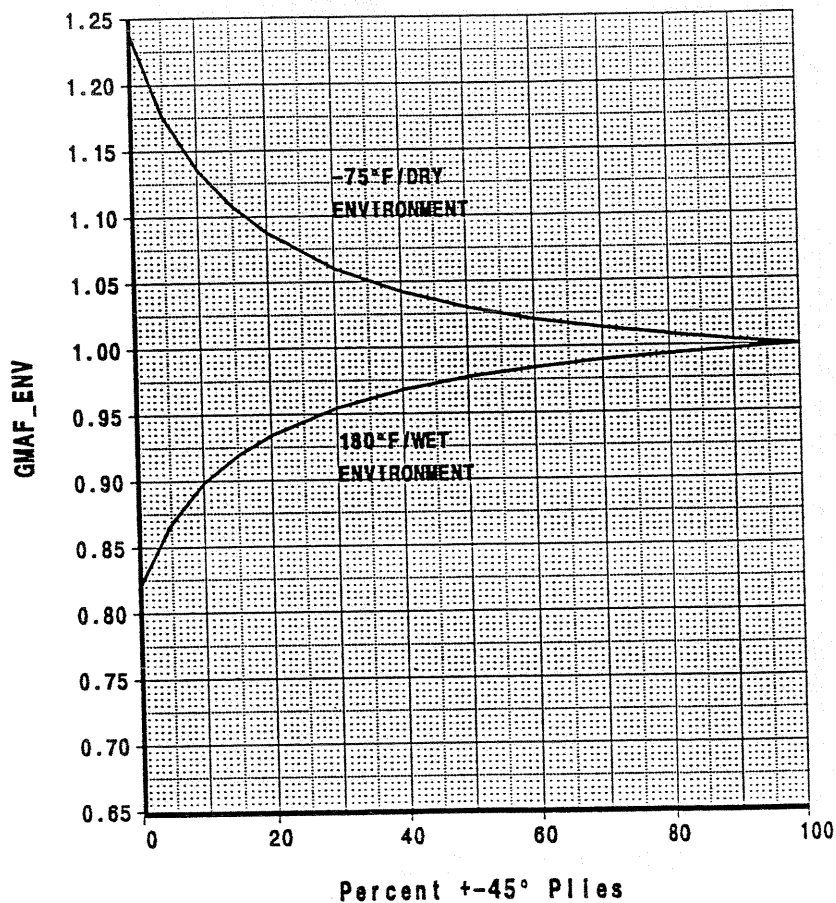
DG-100-528

Figure 2.2.1-5 E_x Environmental Adjustment Factor

BOEING PROPRIETARY

THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE II, CLASS 1 or 3, GRADE 145 or 190	
Resin Content:	35 (% WT)
Process Specification:	BAC 5317-1
(0/45/90) LAMINATES	



$$G_{XY\text{ENVIRONMENT}} = G_{XY\text{NOMINAL}} \times \text{GMAF_ENV}$$

DG-100-529

Figure 2.2.1-6 G_{xy} Environmental Adjustment Factors

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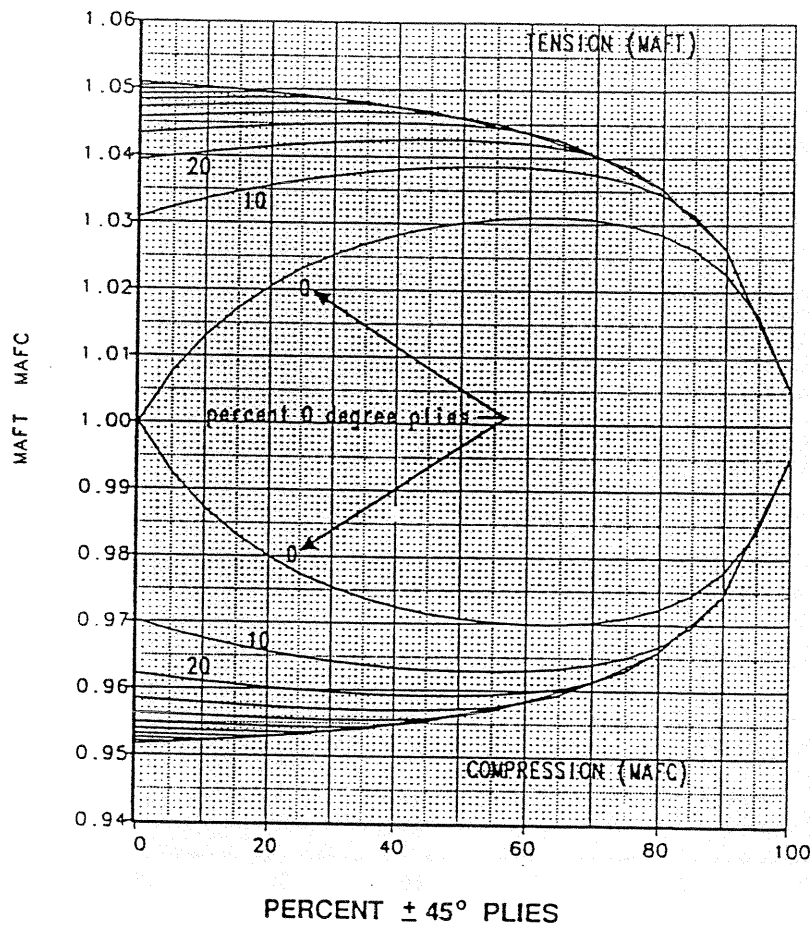
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE II, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 35 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0±45/90) LAMINATES



$$E_{x \text{ TENSION}} = E_{x \text{ NOMINAL}} \times \text{MAFT}$$

$$E_{x \text{ COMPRESSION}} = E_{x \text{ NOMINAL}} \times \text{MAFC}$$

DG-100-530

Figure 2.2.1-7 E_x Tension/Compression Adjustment Factors

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2.2.2 BMS 8-212, Type III, Class 1 or 3, Grade 145 or 190

This specification applies to 350°F cure epoxy resin impregnated BMS 9-8 Type I carbon fiber (average modulus range 32 to 35 msi) unidirectional tape.

- Type III: 37 percent by weight resin content.
Class 1: Unidirectional tape for manual layup.
Class 3: Unidirectional tape for use with automated tape laying equipment or manual layup.

Contents:

Table 2.2.2-1 Ply Properties

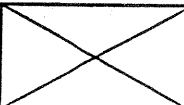
- Figure 2.2.2-1 Nominal E_x Modulus Plot
Figure 2.2.2-2 Nominal G_{xy} Modulus Plot
Figure 2.2.2-3 Nominal ν_{xy} Poisson's Ratio Plot
Figure 2.1.1-4 Nominal α_x Coefficient of Thermal Expansion Plot

Figure 2.2.2-5 E_x Environmental Adjustment Factors Plot
Figure 2.2.2-6 G_{xy} Environmental Adjustment Factors Plot
Figure 2.2.2-7 E_x Tension/Compression Adjustment Factors Plot

BOEING PROPRIETARY

THE BOEING COMPANY

Table 2.2.2-1 Ply Properties

TABLE: 2.2.2-1		PLY PROPERTIES						
Prepreg Material Specification: BMS 8-212, TYPE III CLASS 1 or 3, GRADE 145 or 190								
Resin Content:		37 (% WT)						
Cured Ply Thickness:		0.0059 (In) - GRADE 145				0.0080 (In) - GRADE 190		
Density:		0.056 lb/in³						
Process Specification:		BAC 5317-1						
PROPERTY		UNIT	ENVIRONMENTAL CONDITION					
			-75° F		70° F		180° F ⑤	
			DRY	WET	DRY	WET	DRY	WET
MODULUS ②	E ₁ ①	Msi	17.1					
	E ₂	Msi			1.28			
	G ₁₂	Msi	0.82		0.66			0.54
	G ₁₃	Msi	0.82		0.66			0.54
	G ₂₃	Msi	0.82		0.66			0.54
POISSON'S RATIO	ν ₁₂	----	0.34					
COEFFICIENTS OF LINEAR THERMAL EXPANSION ③	α ₁	In/in ° F	.02 x 10 ⁻⁶			.02 x 10 ⁻⁶		
	α ₂	In/in ° F	15 x 10 ⁻⁶			15 x 10 ⁻⁶		
COEFFECIENTS OF LINEAR MOISTURE EXPANSION ④	β ₁	In/in %M						
	β ₂	In/in %M						
THERMAL CONDUCTIVITY	κ ₁	BTU/(hr ft °F)						
	κ ₂	BTU/(hr ft °F)						
	κ ₃	BTU/(hr ft °F)						

① E₁ is the average of tension and compression moduli. For special analyses use tension/compression modulus adjustment factors to determine tension or compression E₁.

② Modulus values are secant values at a strain level of 4000 μ in/in. For special analysis use modulus versus strain curves to determine secant values at the desired strain level.

③ Values are for expansion between -75° F and 70° F, and 70° F and 180° F.

④ %M = Percent absorbed moisture by weight.

⑤ 180° F values are to be used for 160° F environment.

DG-100-531

BOEING PROPRIETARY

THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 37 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES

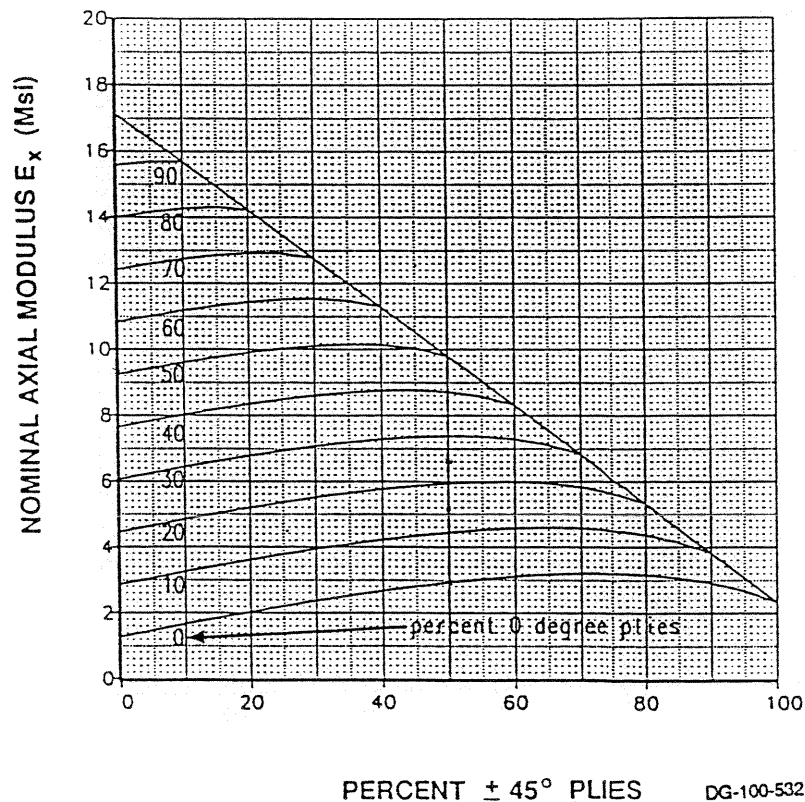
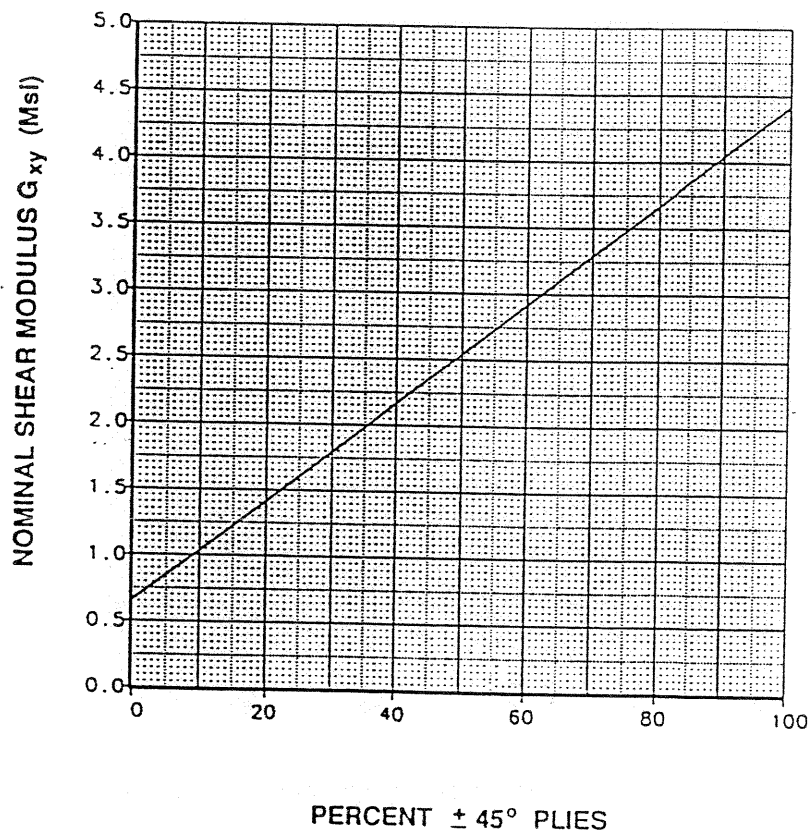


Figure 2.2.2-1 Nominal E_x Modulus

BOEING PROPRIETARY
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190	
Resin Content:	37 (% WT)
Process Specification:	BAC 5317-1
70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES	



DG-100-533

Figure 2.2.2-2 Nominal G_{xy} Modulus

BOEING PROPRIETARY

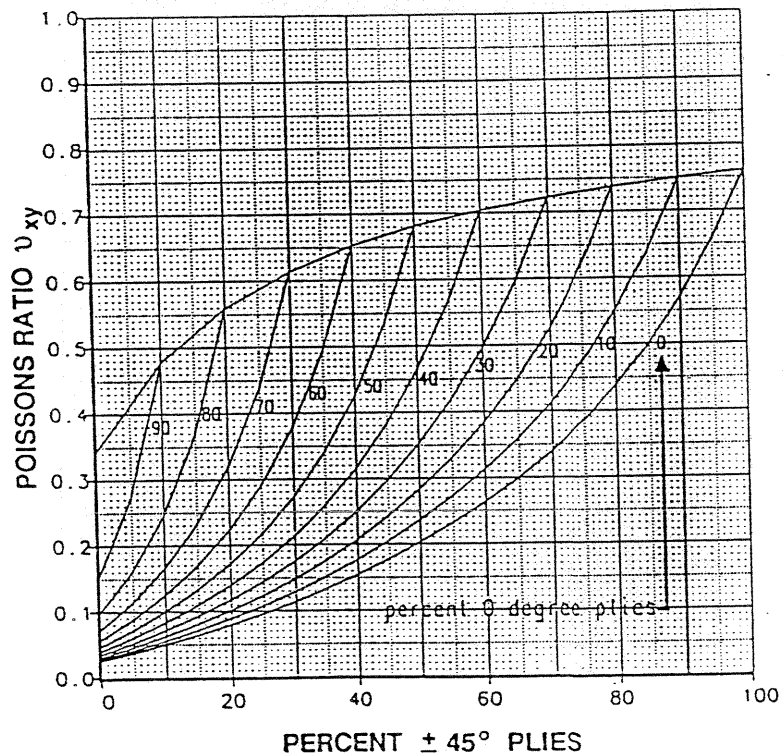
THE BOEING COMPANY

Prepreg Material Specification: **BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190**

Resin Content: **37 (% WT)**

Process Specification: **BAC 5317-1**

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



DG-100-534

Figure 2.2.2-3 Nominal v_{xy} Poisson's Ratio

BOEING PROPRIETARY

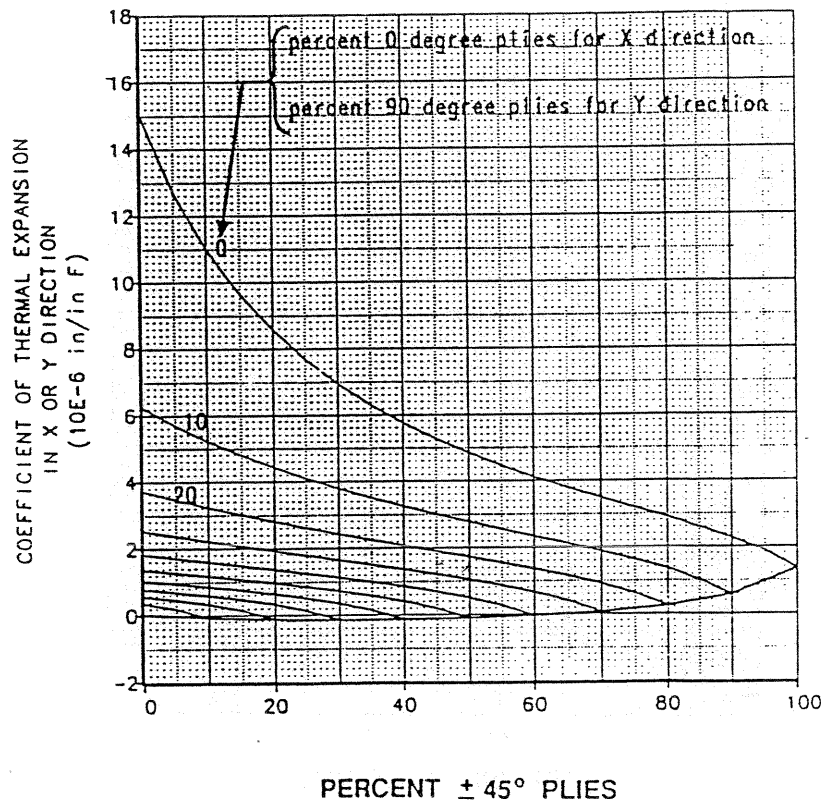
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 37 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



DG-100-535

Figure 2.2.2-4 Nominal α_x Coefficient of Thermal Expansion

BOEING PROPRIETARY

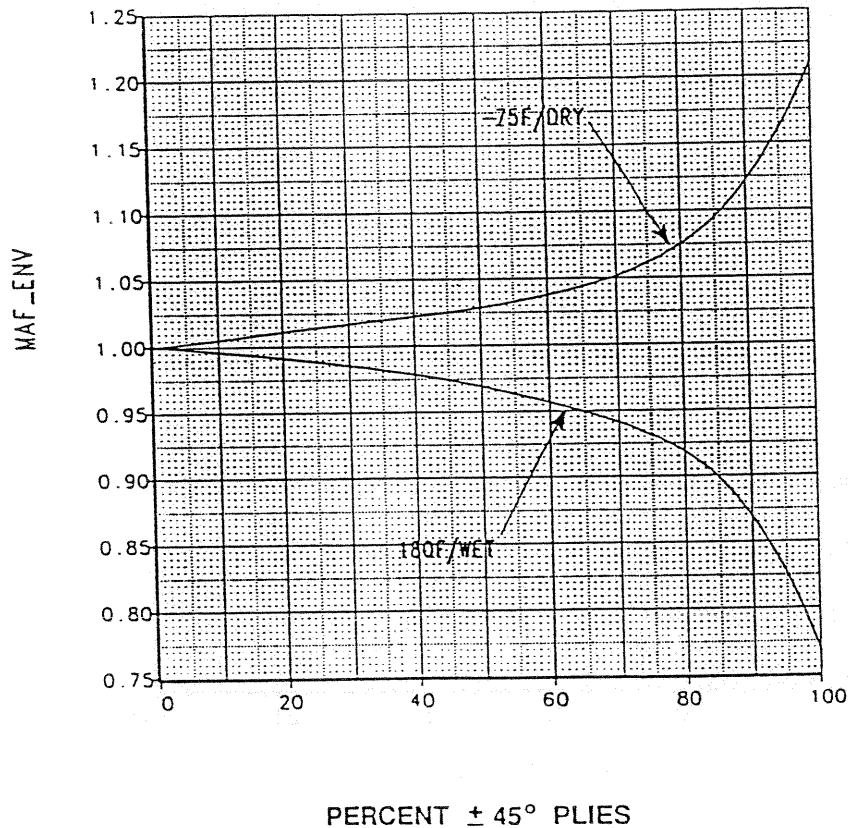
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 37 (% WT)

Process Specification: BAC 5317-1

(0/±45/90) LAMINATES



$$E_x \text{ ENVIRONMENT} = E_x \text{ NOMINAL} \times \text{MAF_ENV}$$

DG-100-536

Figure 2.2.2-5 E_x Environmental Adjustment Factors

BOEING PROPRIETARY

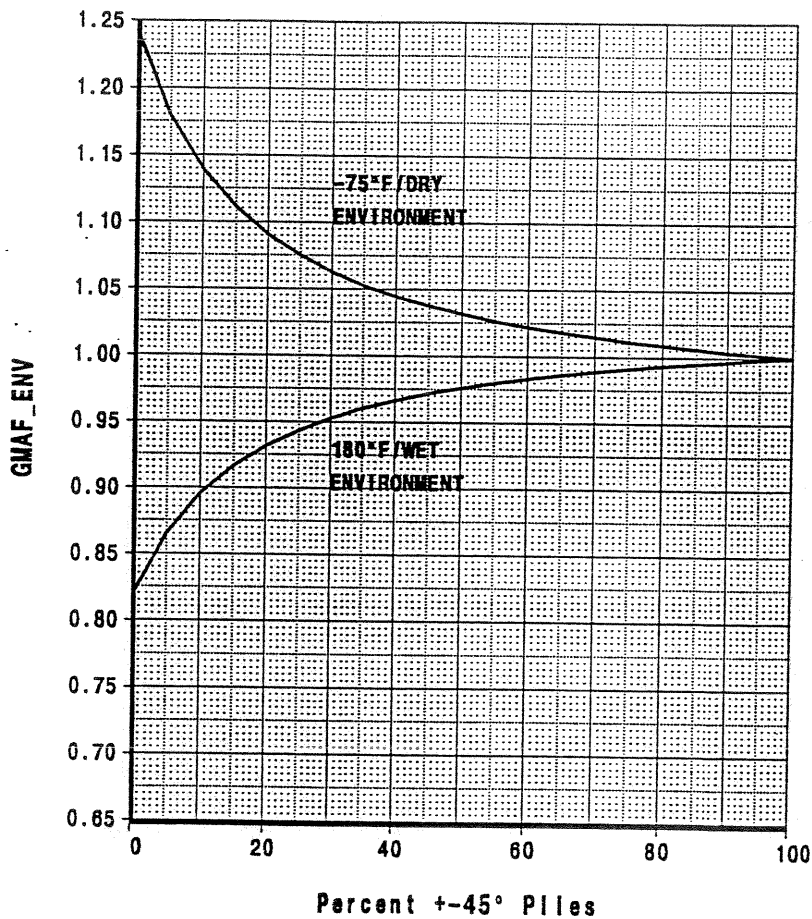
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 37 (% WT)

Process Specification: BAC 5317-1

(0/±45/90) LAMINATES



$$G_{XY\text{ENVIRONMENT}} = G_{XY\text{NOMINAL}} \times \text{GMAF_ENV}$$

DG-100-537

Figure 2.2.2-6 G_{xy} Environmental Adjustment Factors

BOEING PROPRIETARY

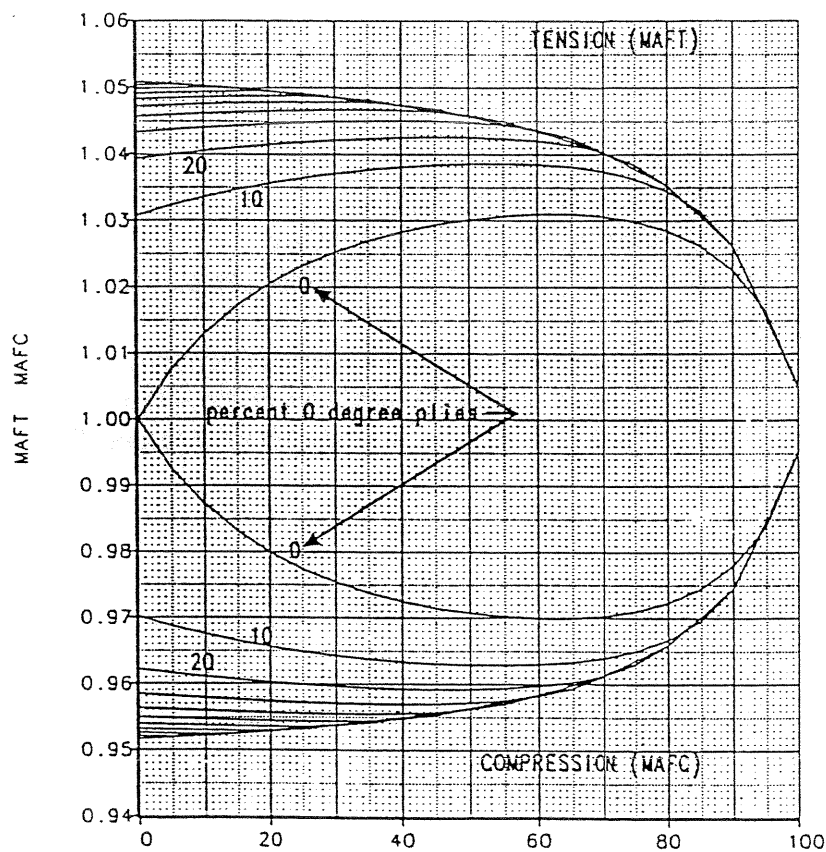
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE III, CLASS 1 or 3, GRADE 145 or 190

Resin Content: 37 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



PERCENT ± 45° PLIES

$$E_x \text{ TENSION} = E_x \text{ NOMINAL} \times \text{MAFT}$$

$$E_x \text{ COMPRESSION} = E_x \text{ NOMINAL} \times \text{MAFC}$$

DG-100-538

Figure 2.2.2-7 E_x Tension/Compression Adjustment Factors

BOEING PROPRIETARY

THE BOEING COMPANY

2.2.3 BMS 8-212, Type IV, Class 2, Style 3K-70-PW

This specification applies to 350°F cure epoxy resin impregnated BMS 9-8 Type I carbon fiber (average modulus range 32 to 35 msi) woven fabric.

Type IV: 40 percent by weight resin content.

Class 2: Woven Fabric Prepreg.

Contents:

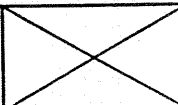
Table 2.2.3-1 Ply Properties

Figure 2.2.3-1	Nominal E_x Modulus Plot
Figure 2.2.3-2	Nominal G_{xy} Modulus Plot
Figure 2.2.3-3	Nominal ν_{xy} Poisson's Ratio Plot
Figure 2.1.1-4	Nominal α_x Coefficient of Thermal Expansion Plot
Figure 2.2.3-5	E_x Environmental Adjustment Factors Plot
Figure 2.2.3-6	G_{xy} Environmental Adjustment Factors Plot
Figure 2.2.3-7	E_x Tension/Compression Adjustment Factors Plot

BOEING PROPRIETARY

THE BOEING COMPANY

Table 2.2.3-1 Ply Properties

TABLE: 2.2.3-1		PLY PROPERTIES						
Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW								
Resin Content:		40 (% WT)						
Cured Ply Thickness:		0.0083 (in)						
Density:		0.055 lb/in³						
Process Specification:		BAC 5317-1						
PROPERTY		UNIT	ENVIRONMENTAL CONDITION					
			-75° F		70° F		180° F ⑤	
			DRY	WET	DRY	WET	DRY	WET
MODULUS ②	E ₁ ①	Msi	8.30					
	E ₂	Msi	8.30					
	G ₁₂	Msi	0.87		0.70			0.45
	G ₁₃	Msi	0.87		0.70			0.45
	G ₂₃	Msi	0.87		0.70			0.45
POISSON'S RATIO	ν ₁₂	----	0.06					
COEFFICIENTS OF LINEAR THERMAL EXPANSION ③	α ₁	in/in ° F	1.6 x 10 ⁻⁶			1.6 x 10 ⁻⁶		
	α ₂	in/in ° F	1.6 x 10 ⁻⁶			1.6 x 10 ⁻⁶		
COEFECIENTS OF LINEAR MOISTURE EXPANSION ④	β ₁	in/in %M						
	β ₂	in/in %M						
THERMAL CONDUCTIVITY	κ ₁	BTU/(hr ft °F)						
	κ ₂	BTU/(hr ft °F)						
	κ ₃	BTU/(hr ft °F)						

① E₁ is the average of tension and compression moduli. For special analyses use tension/compression modulus adjustment factors to determine tension or compression E₁.

② Modulus values are secant values at a strain level of 4000 μ in/in. For special analysis use modulus versus strain curves to determine secant values at the desired strain level.

③ Values are for expansion between -75° F and 70° F, and 70° F and 180° F.

④ %M = Percent absorbed moisture by weight.

⑤ 180°F values are to be used for 160°F environment.

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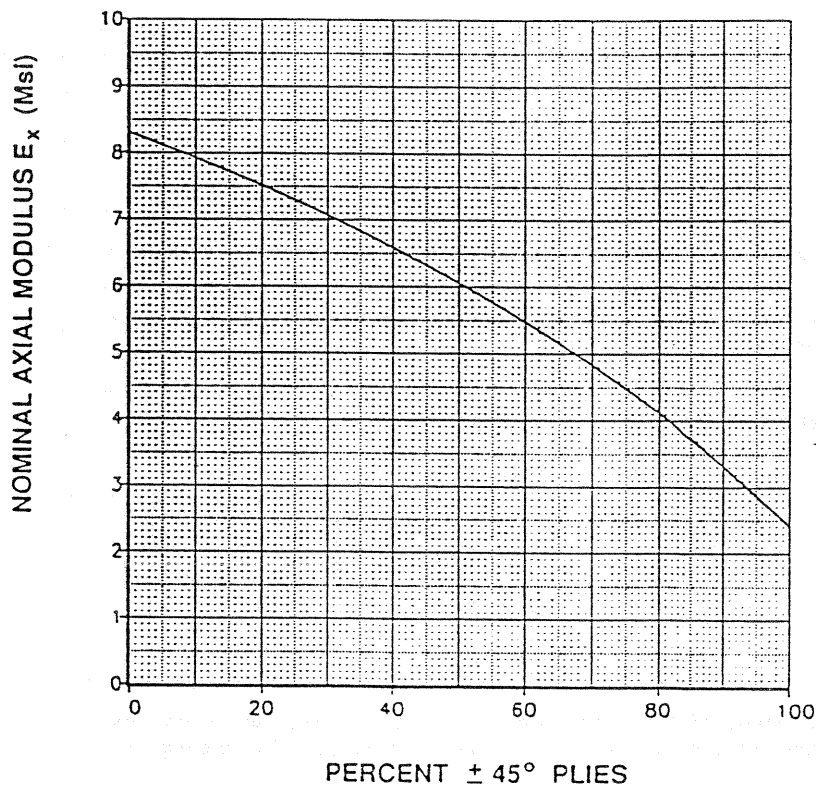
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW

Resin Content: 40 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



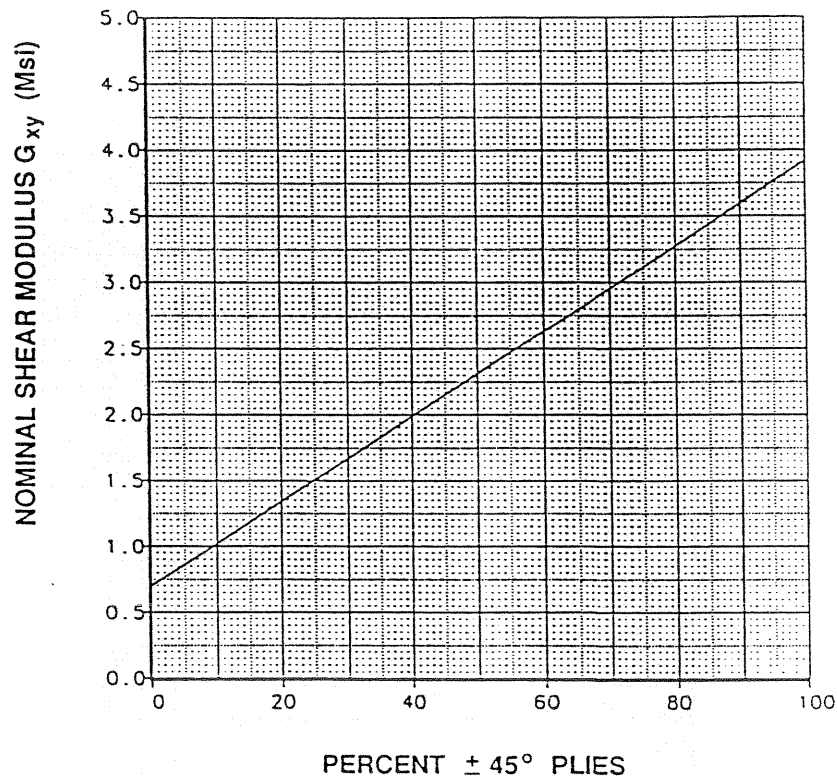
DG-100-540

Figure 2.2.3-1 Nominal E_x Modulus

BOEING PROPRIETARY

THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW	
Resin Content:	40 (% WT)
Process Specification:	BAC 5317-1
70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES	



DG-100-541

Figure 2.2.3-2 Nominal G_{xy} Modulus

BOEING PROPRIETARY

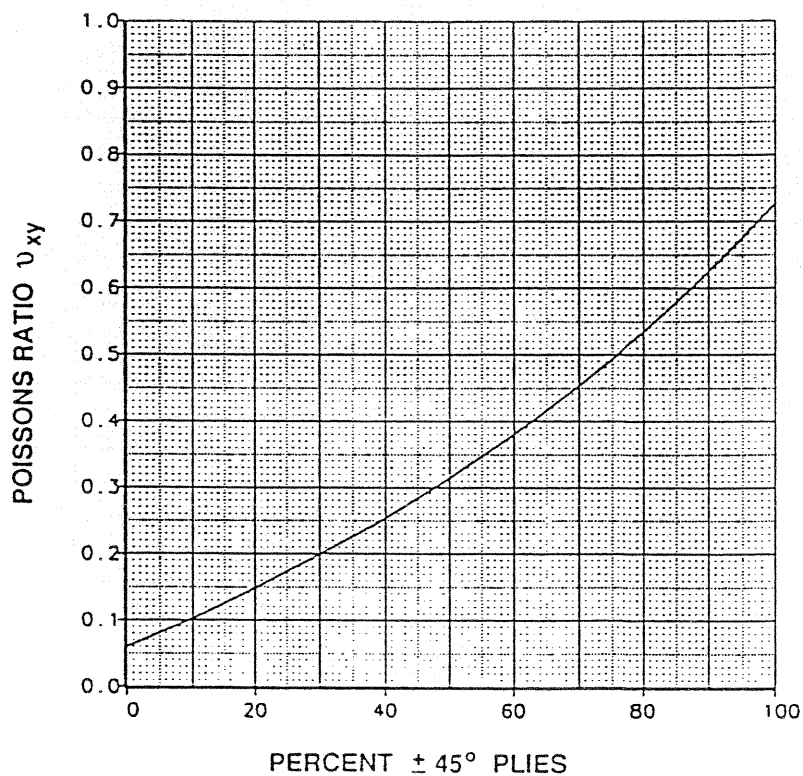
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW

Resin Content: 40 (% WT)

Process Specification: BAC 5317-1

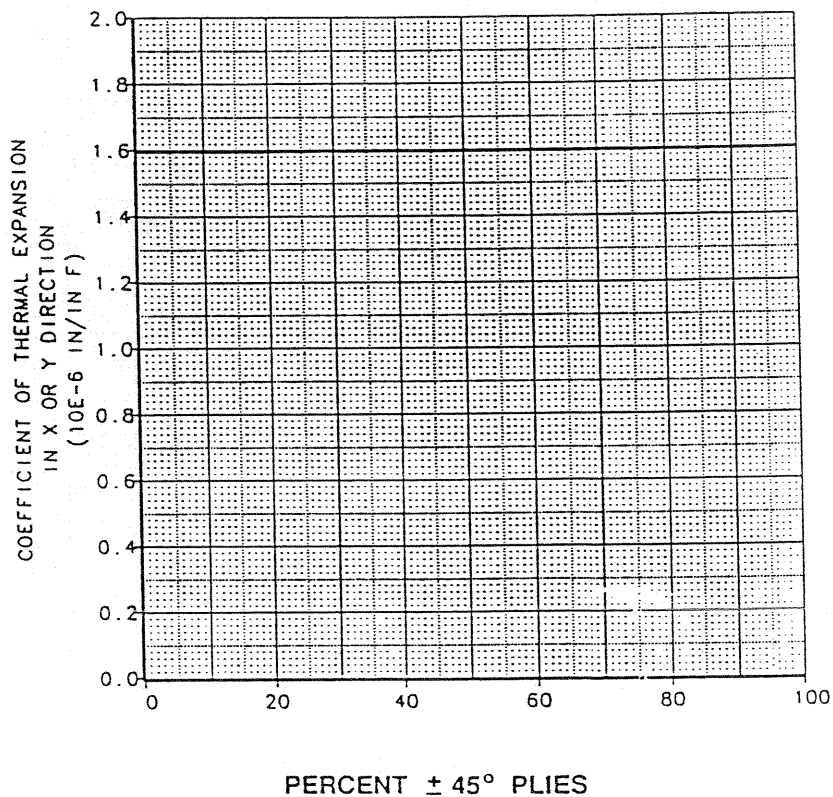
70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



DG-100-542

Figure 2.2.3-3 Nominal v_{xy} Poisson's Ratio

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW	
Resin Content:	40 (% WT)
Process Specification:	BAC 5317-1
70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES	



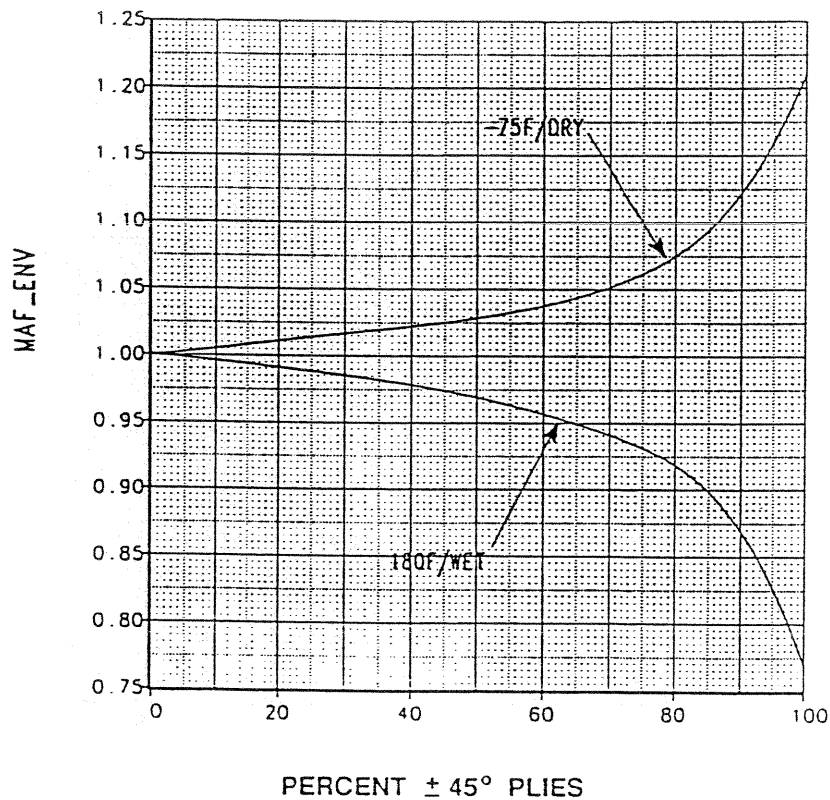
DG-100-543

Figure 2.2.3-4 Nominal α_x Coefficient of Thermal Expansion

BOEING PROPRIETARY

THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW	
Resin Content:	40 (% WT)
Process Specification:	BAC 5317-1
(0/±45/90) LAMINATES	



$$E_{x \text{ ENVIRONMENT}} = E_{x \text{ NOMINAL}} \times \text{MAF_ENV}$$

DG-100-544

Figure 2.2.3-5 E_x Environmental Adjustment Factors

BOEING PROPRIETARY

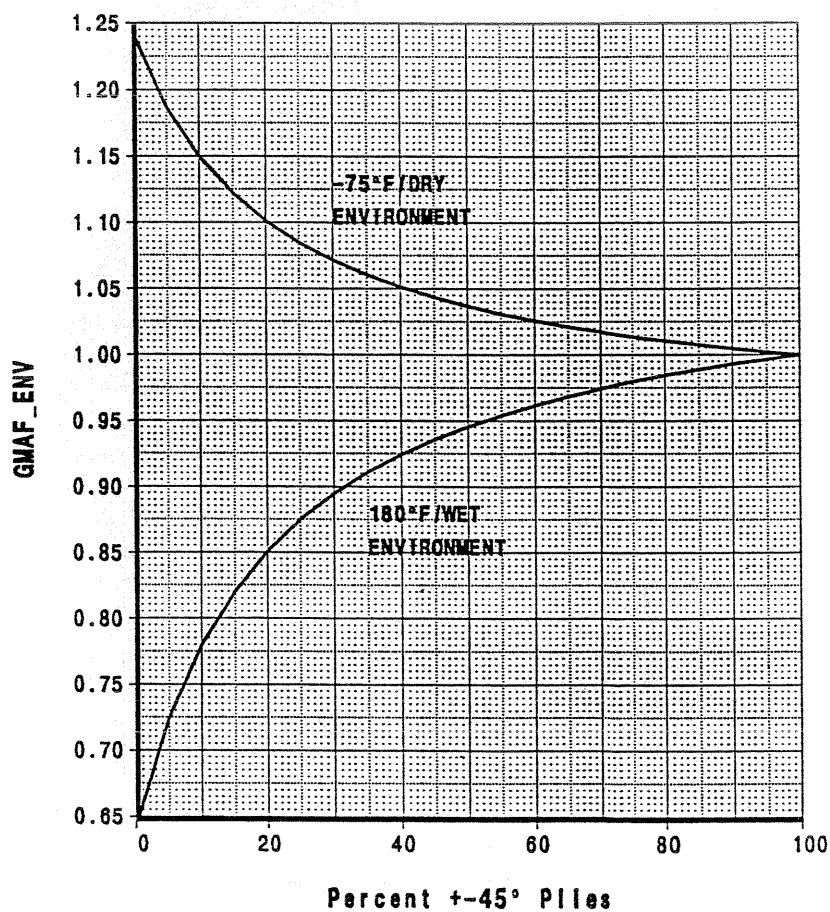
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW

Resin Content: 40 (% WT)

Process Specification: BAC 5317-1

(0/±45/90) LAMINATES



$$G_{XY\text{ENVIRONMENT}} = G_{XY\text{NOMINAL}} \times \text{GMAF_ENV}$$

DG-100-545

Figure 2.2.3-6 G_{xy} Environmental Adjustment Factors

BOEING PROPRIETARY

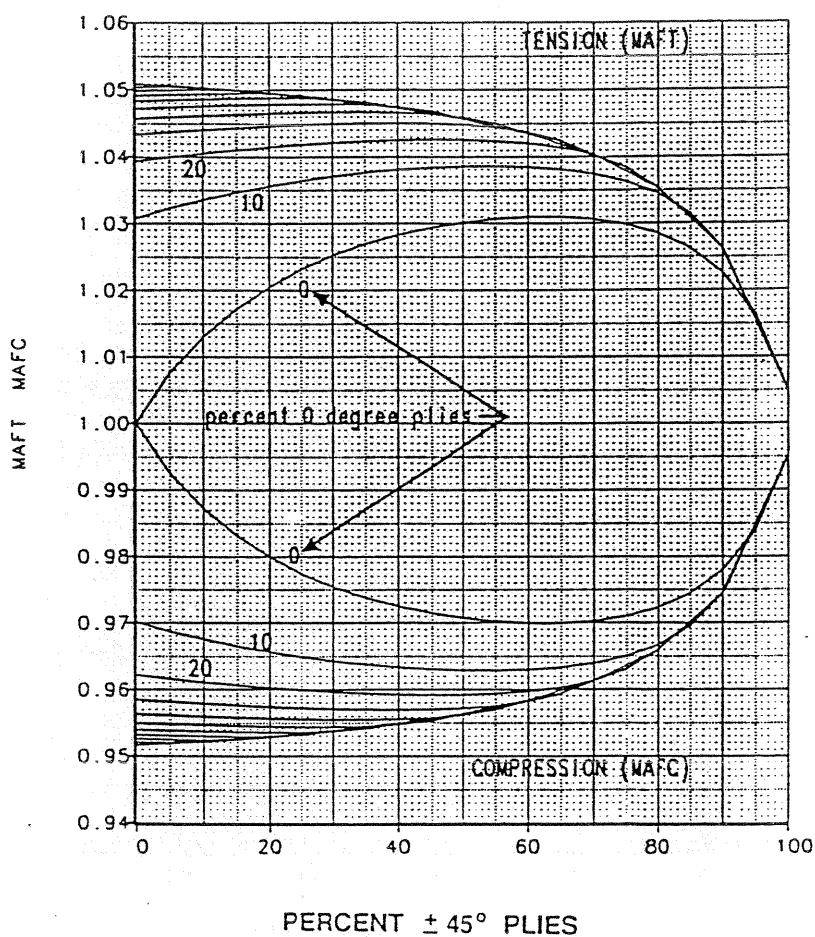
THE BOEING COMPANY

Prepreg Material Specification: BMS 8-212, TYPE IV, CLASS 2, STYLE 3K-70-PW

Resin Content: 40 (% WT)

Process Specification: BAC 5317-1

70° F/DRY ENVIRONMENT, (0/±45/90) LAMINATES



$$E_{x \text{ TENSION}} = E_{x \text{ NOMINAL}} \times \text{MAFT}$$

$$E_{x \text{ COMPRESSION}} = E_{x \text{ NOMINAL}} \times \text{MAFC}$$

DG-100-546

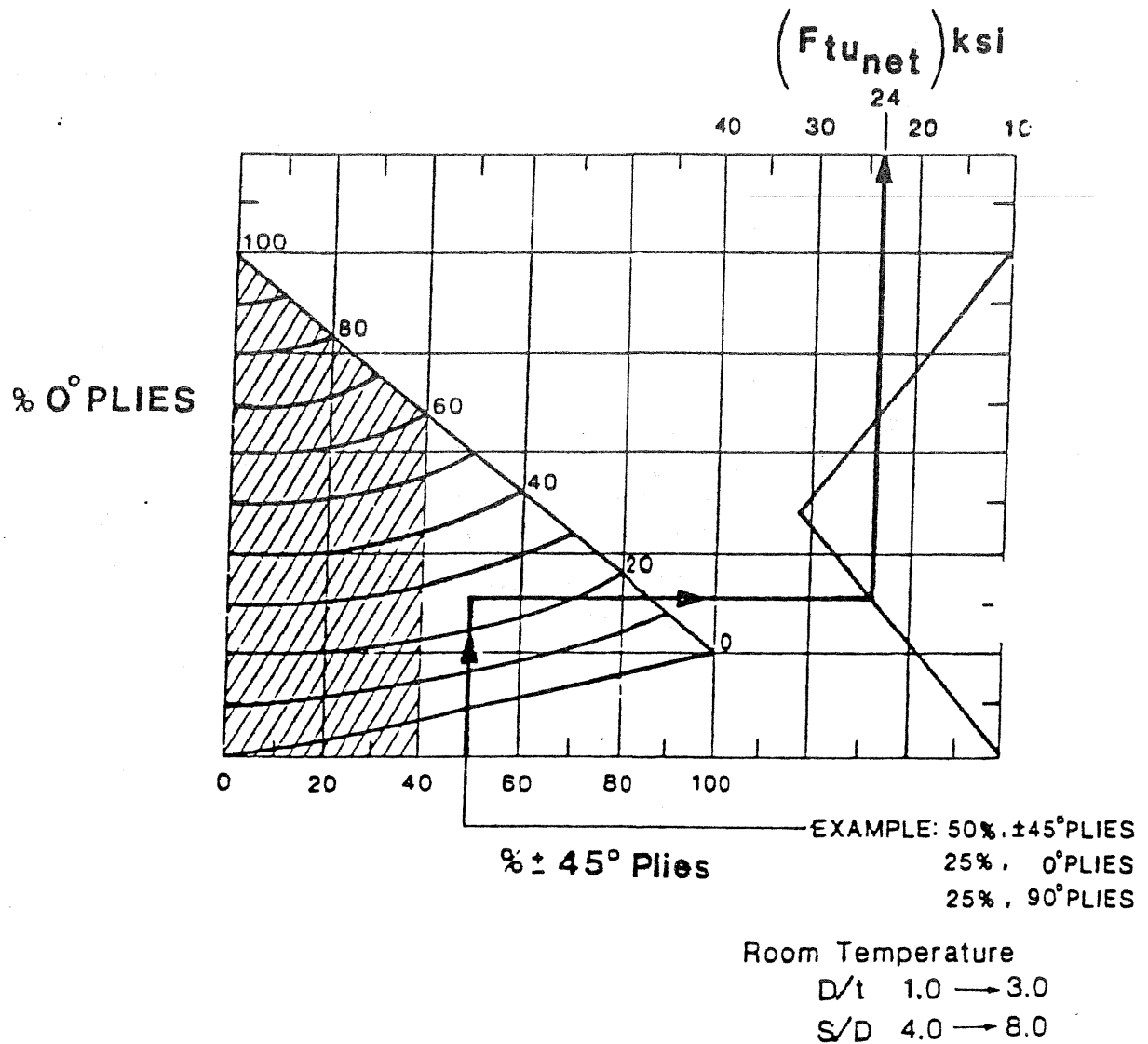
Figure 2.2.3-7 E_x Tension/Compression Adjustment Factors

BOLTED JOINT TENSION STRENGTH

TAPE & FABRIC

Graphite/Epoxy BMS 8-212 Class 1&2 per BAC 5562

CAUTION: DATA MAY NOT BE VALID FOR LAMINATES WITH BUTT
SPICES NEAR FASTENER(S)



See Figure 4.3.3-2 For Restriction On Minimum
Percentage Of ± 45° Plies For Bearing Strength

From Reference 1, Page 4.1 and Reference 3 and Boeing Data

FIGURE 4.3.3-1

Boeing D6-44714

4-25

Rev. E

POISSON'S RATIO

TAPE

GRAPHITE/EPOXY, BMS 8-212 Type II CLASS 1, PER BAC 5562

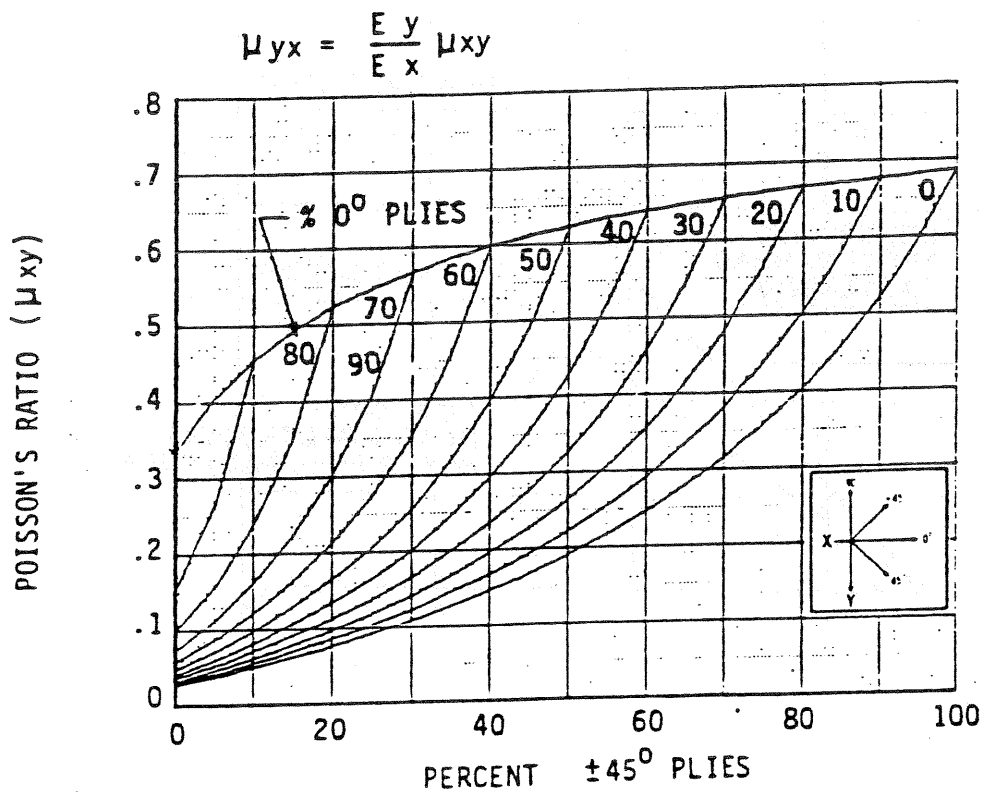
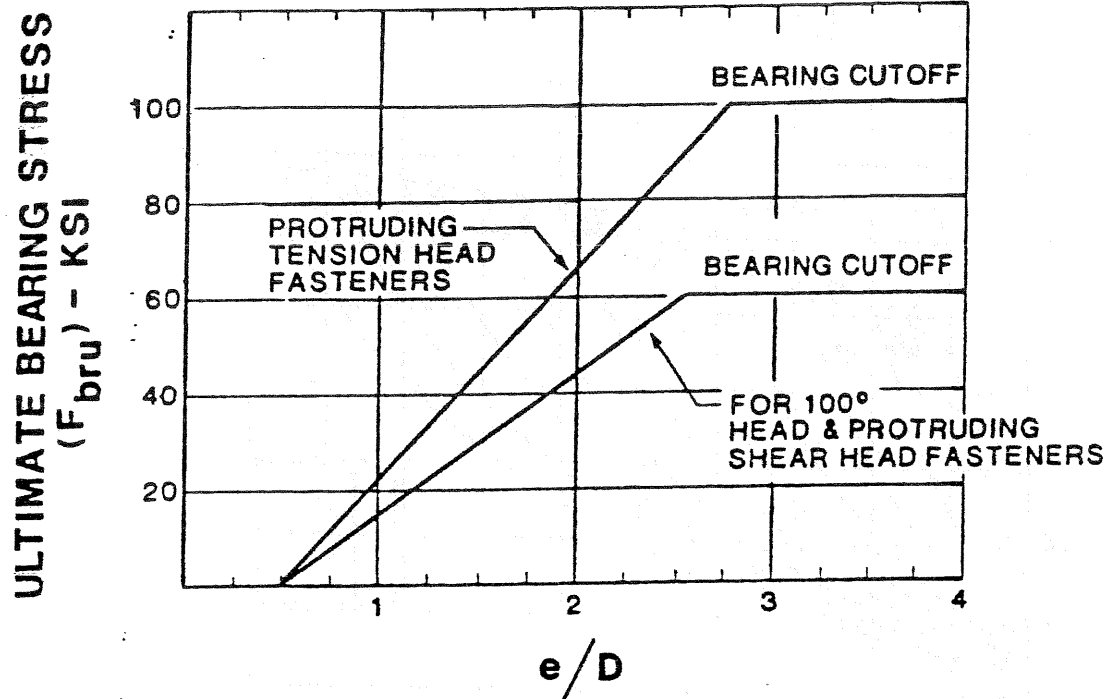


FIGURE 4.3.5-1
BOEING D6-44714

BOLTED JOINT BEARING STRENGTH

TAPE AND FABRIC

Graphite/Epoxy BMS 8-212 Class 1 & 2 per BAC 5562



ROOM TEMPERATURE

THIS DATA IS VALID FOR LAMINATES CONTAINING A MINIMUM OF 40% \pm 45% PLIES, AND FOR A COUNTERSINK DEPTH LESS THAN 2/3 THE LAMINATE THICKNESS.

From Reference 1, Page 4.2, Reference 2, and Boeing Data

Figure 4.3.3 - 2

Boeing D6-44714

4-26

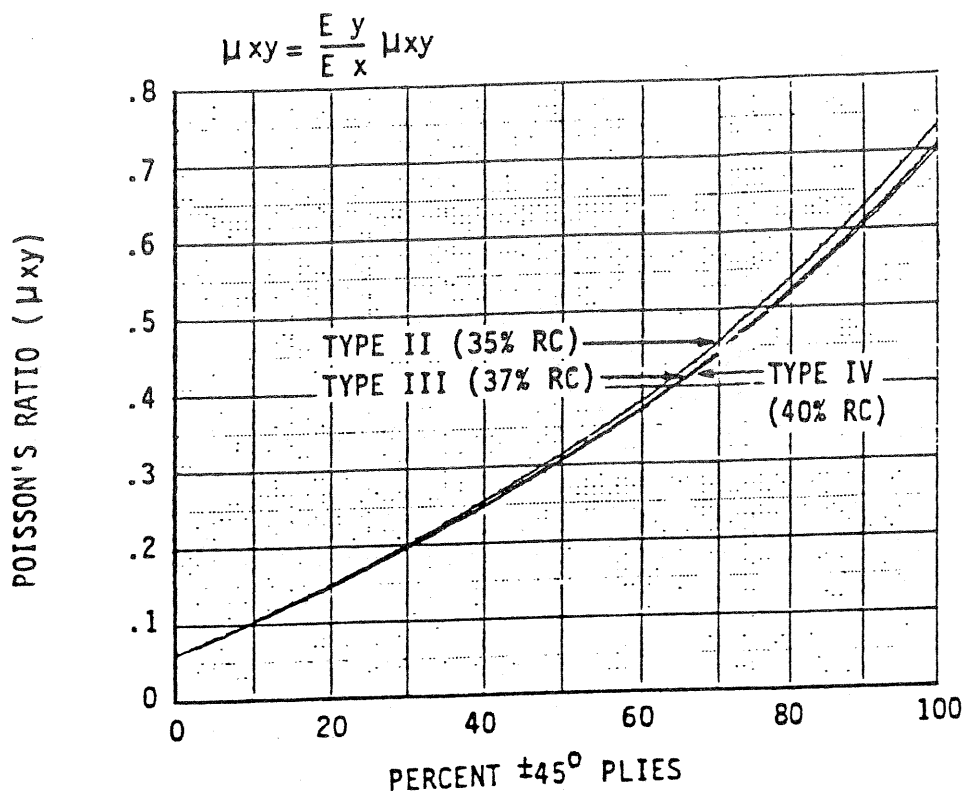
Rev. D

POISSON'S RATIO

FABRIC

GRAPHITE/EPOXY

5 8-212 CLASS 2, PER BAC 5562



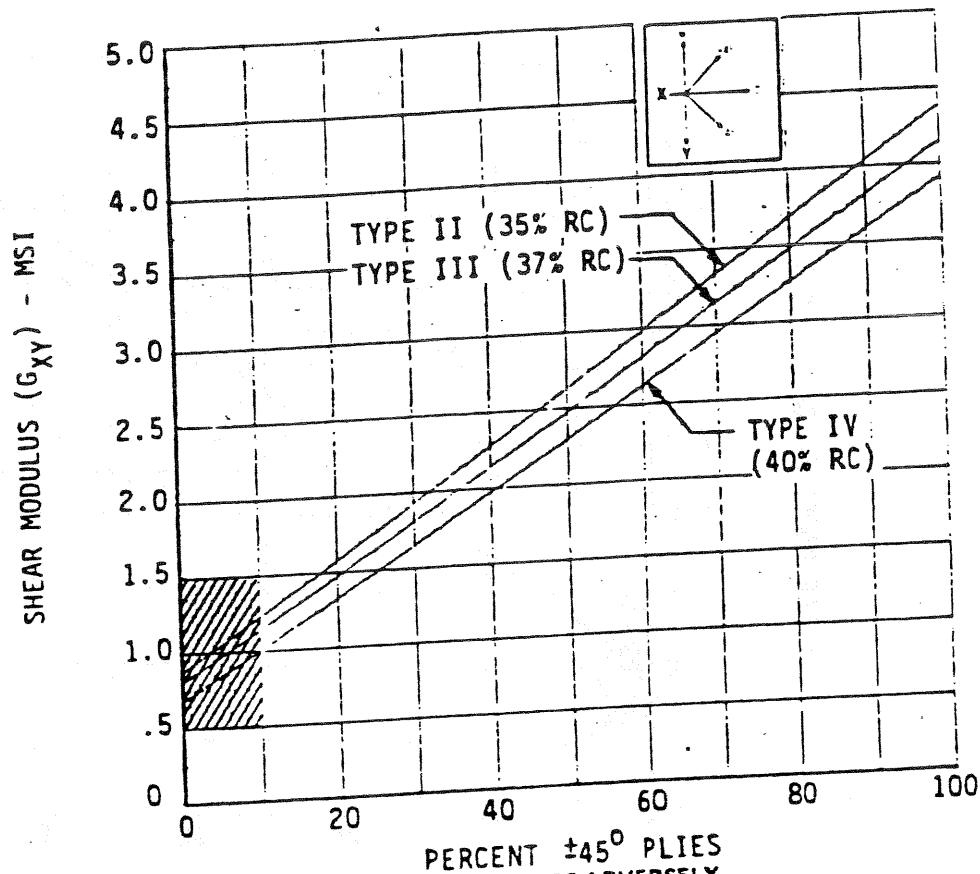
RC=NOMINAL RESIN CONTENT PER BMS 8-212

FIGURE 4.3.5-3
BOEING D6-44714

IN-PLANE SHEAR MODULUS AND ALLOWABLE STRAINS

FABRIC

GRAPHITE/EPOXY, BMS 8-212 CLASS 2, PER BAC 5562



CAUTION: PROPERTIES IN THE CROSS-HATCHED REGION MAY BE ADVERSELY AFFECTED BY TEMPERATURE AND HUMIDITY.

RC= NOMINAL RESIN CONTENT PER BMS 8-212

LIMIT & ULTIMATE STRAIN (IN/IN)	LIMIT STRAIN	ULTIMATE STRAIN	
		"A" BASIS	"B" BASIS
LAMINATE SHEAR E_{xy}	.0053	.0108	.0117
SANDWICH SHEAR E_{xy}	①	.0070	.0076

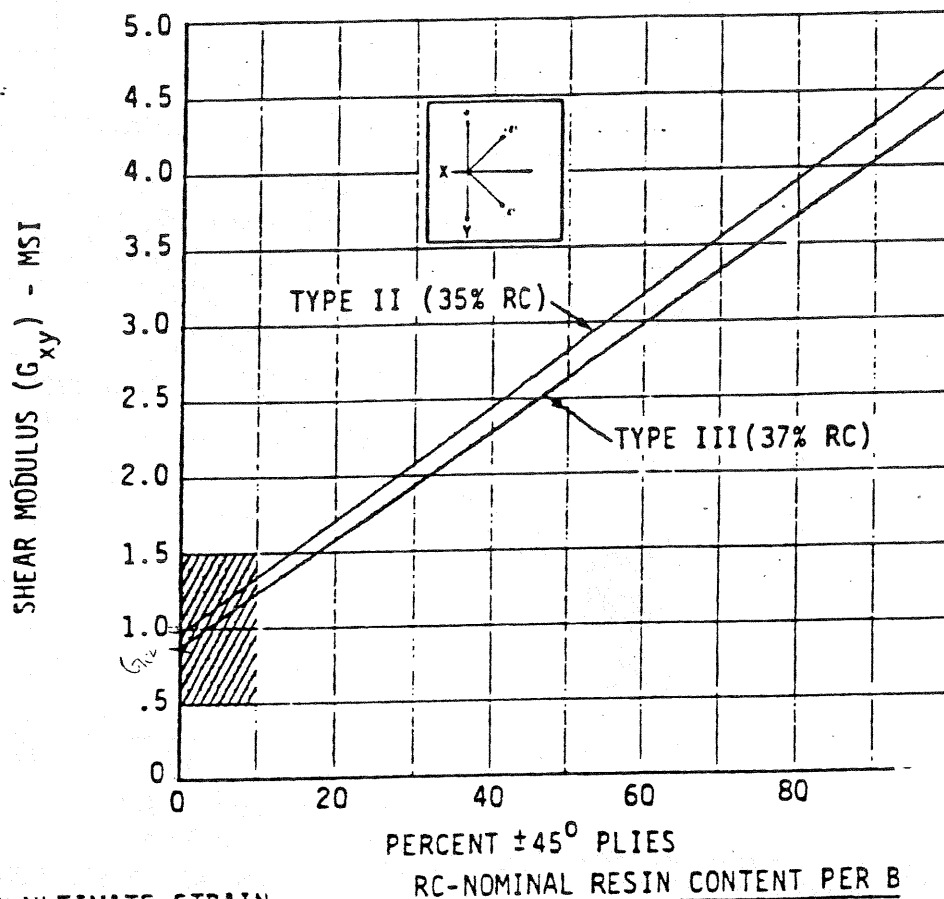
① ULTIMATE STRAIN $\div 1.5$

FIGURE 4.3.1-5
BOEING D6-44714

IN-PLANE SHEAR MODULUS AND ALLOWABLE STRAINS

TAPE

GRAPHITE/EPOXY, BMS 8-212 CLASS 1, PER BAC 5562



LIMIT & ULTIMATE STRAIN
(IN/IN)

	LIMIT STRAIN	ULTIMATE STRAIN	
		"A" BASIS	"B" BASIS
SHEAR ϵ_{xy}	.0053	.0133	.0144

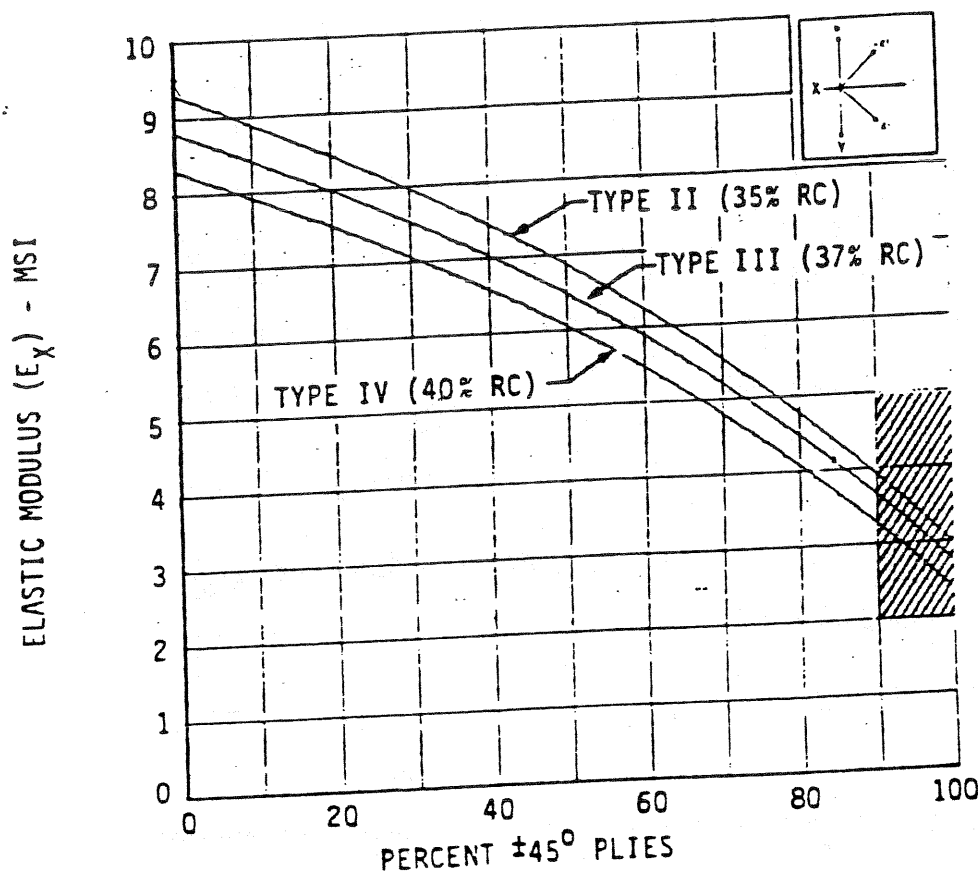
CAUTION: PROPERTIES IN THE CROSS-HATCHED REGION MAY BE ADVERSELY AFFECTED BY TEMPERATURE AND HUMIDITY.

FIGURE 4.3.1-4
BOEING D6-44714

ELASTIC MODULUS AND ALLOWABLE STRAINS

FABRIC

GRAPHITE/EPOXY, BMS 8-212 CLASS 2, PER BAC 5562



CAUTION: PROPERTIES IN THE CROSS-HATCHED REGION MAY BE ADVERSELY AFFECTED BY TEMPERATURE HUMIDITY.

RC-NOM RESIN CONTENT PER BMS 8-212

LIMIT & ULTIMATE STRAIN IN/IN)	LIMIT STRAIN	ULTIMATE STRAIN	
		"A" BASIS	"B" BASIS
LAMINATE TENSION E_t	.0035	.0059	.0064
LAMINATE COMPRESSION E_c	.0027	.0054	.0059
SANDWICH TENSION E_t	①	.0043	.0046
SANDWICH COMPRESSION E_c	①	.0035	.0038

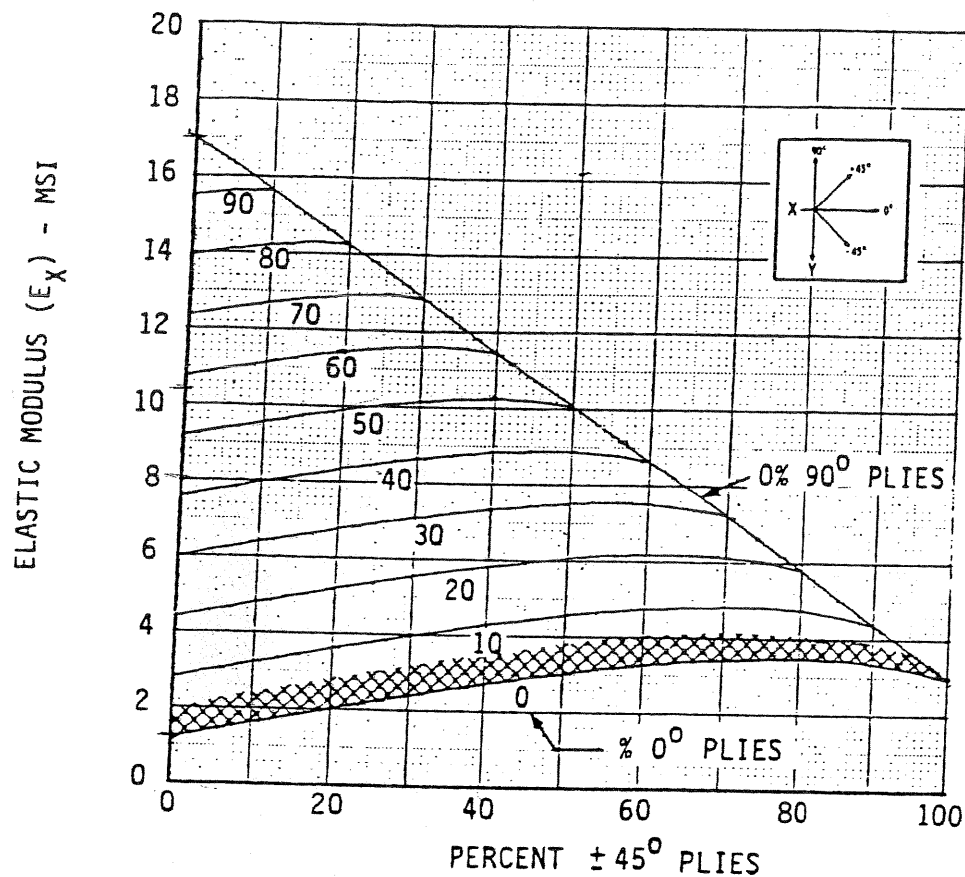
① ULTIMATE STRAIN $\div 1.5$

FIGURE 4.3.1-3
BOEING D6-44714

ELASTIC MODULUS AND ALLOWABLE STRAINS

TAPE

GRAPHITE/EPOXY, BMS 8-212 TYPE III CLASS 1, PER BAC 5562



CAUTION: PROPERTIES IN THE CROSS-HATCHED REGION MAY BE ADVERSELY AFFECTED BY TEMPERATURE AND HUMIDITY.

LIMIT & ULTIMATE STRAIN (IN/IN)

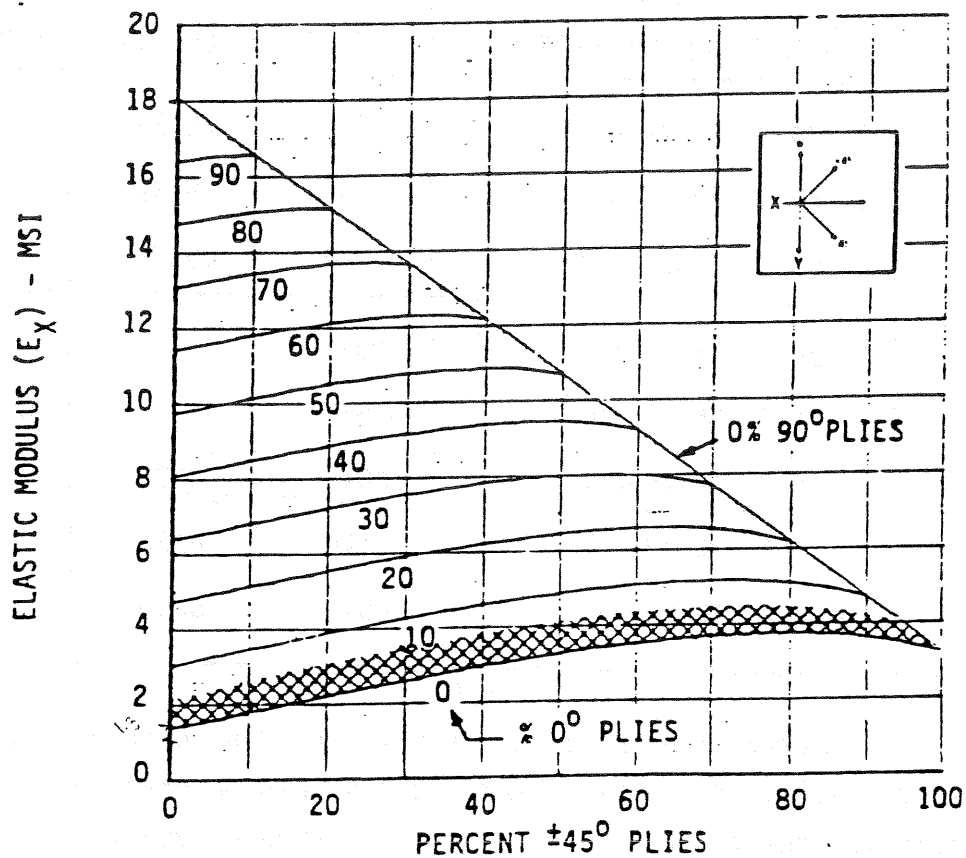
	LIMIT STRAIN	ULTIMATE STRAIN	
		"A" BASIS	"B" BASIS
TENSION ϵ_t	.0035	.0082	.0090
COMPRESSION ϵ_c	.0027	.0067	.0072

FIGURE 4.3.1-2
BOEING D6-44714

ELASTIC MODULUS AND ALLOWABLE STRAINS

TAPE

GRAPHITE/EPOXY, BMS 8-212 TYPE II CLASS 1, PER BAC 5562



CAUTION: PROPERTIES IN THE CROSS-HATCHED REGION MAY BE ADVERSELY AFFECTED BY TEMPERATURE AND HUMIDITY.

LIMIT & ULTIMATE STRAIN
(IN/IN)

	LIMIT STRAIN	ULTIMATE STRAIN	
		"A" BASIS	"B" BASIS
TENSION ϵ_t	.0035	.0082	.0090
COMPRESSION ϵ_c	.0027	.0067	.0072

FIGURE 4.3.1-1
BOEING D6-44714

4.3 Mechanical Properties (con't)

All properties presented are based on laminate thickness less than .30 inches. No change in these properties is anticipated when tests are completed on laminate thicknesses greater than .30 inch. Laminate thickness should be calculated using ply thickness values shown below. Adjustments for variations in measured fiber volume or part thickness should not be made.

CURED THICKNESS TABLE (BMS 8-212)		NOMINAL CURED THICKNESS-INCHES*		
FORM	BMS 8-212 DESIGNATION	TYPE II	TYPE III	TYPE IV
Tape	Class 1 Grade 95	.0037	.0039	-
	Class 1 Grade 145	.0056	.0059	-
	Class 1 Grade 190	.0074	.0078	-
Fabric	Class 2 3K-70-P	.0075	-	-
	Class 2 3K-70-PW	.0075	-	.0083
	Class 2 3K-70-CSW	.0073	.0075	-
	Class 2 3K-135-8H	.0144	.0150	-
	Class 2 1K-50-5H	.0048	.0050	-

TABLE 4.3-1

Because Type I material properties are not included in this section, the thickness values for Type II materials should be used when making stress calculations for Type I materials. Although Type I materials have different thicknesses than Type II materials after cure, they contain the same amount of graphite fiber per ply resulting in the same allowable per ply. Therefore, by using Type II allowables and unit thicknesses for stress calculations when using Type I material, the proper allowable stress and margin of safety will be obtained.

s Analysis. (See Table 4.2-1 for preliminary weight data).