

A Micromechanic-based approach towards the Response of Fiber-reinforced Composite Laminates under ballistic impact and blast loading

Rahul Singh
Mathscapes Research

Thursday 21st June, 2018

Composites material are a significant part of a wide variety of applications such as bulletproof vests of soldiers on a battlefield to the passenger jet airliners. The dynamic and multitudinous forces of nature acting on these composites push the materials towards complex deformations, answers to which involves extensive work using numerical techniques, in order to save the expenditure involved in testing and experimentation. In the present study, an attempt has been made to solve this challenging problem to predict the response of composite laminates under ballistic impact and blast loading conditions by implementing a progressive damage based micromechanical model for unidirectional composite laminates and fiber metal laminates.

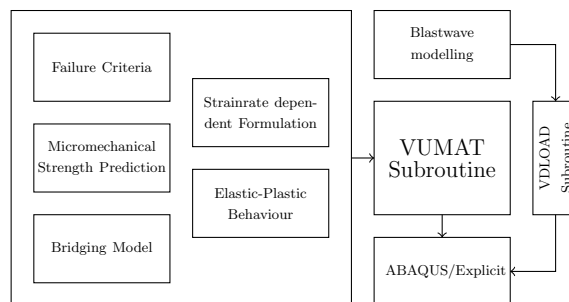


Figure 1: Simulation Schematic

The material response is studied by integrating the concepts of continuum damage mechanics (CDM), bridging model[1], elasticity and plasticity[2], micromechanics based strength prediction[3], strain-rate dependent formulation[4] and a string of failure criteria into the user coded VUMAT subroutine which is used in ABAQUS/Explicit for simulation. In addition to this, the layers of laminates are bonded using surface-based cohesive behav-

ior whereas the interaction between the projectile and the plate is defined using the general contact algorithm. The model consisting of a target plate is having C3D8R elements and the projectile has meshed with R3D4 elements. The mesh is checked for the quality and errors using the metrics such as geometric deviation factor and aspect ratio.

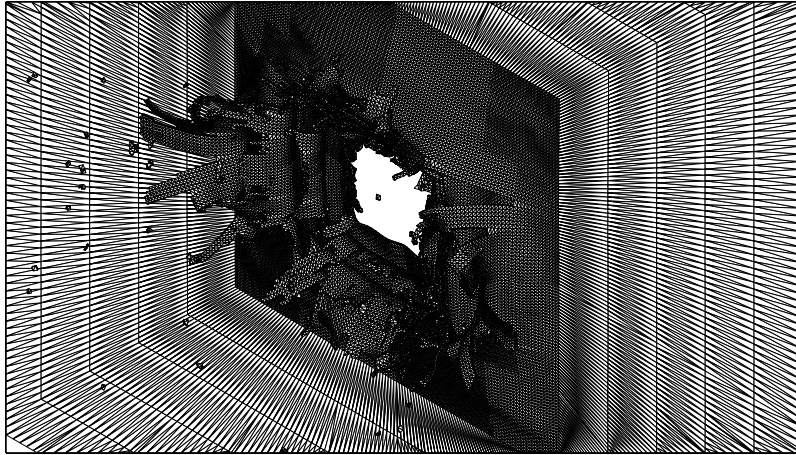


Figure 2: Back face deformation of 8 layered cross ply laminate under blast loading ($I = 642MPa$)

The blast wave is modeled using the VDLOAD subroutine which contains modified Friedlander equation while considering a near-field blast approach to model time decay and a decoupled approach in modeling spatial decay. The results are obtained using materials Carbon/Epoxy and Glass/Epoxy for simulation. The results of the simulation are validated by the work of performed by previous researchers[5][6] and was found in good agreement.

References

- [1] Jianqiao Ye and Daxu Zhang. Prediction of failure envelopes and stress-strain curves of fiber composite laminates under triaxial loads. *Journal of Composite Materials*, 46(19-20):2417–2430, 2012.
- [2] Wei Zhang, Bohong Gu, and Baozhong Sun. Thermal-mechanical coupling modeling of 3d braided composite under impact compression loading and high temperature field. *Composites Science and Technology*, 140:73–88, 2017.
- [3] Zheng Ming Huang. Micromechanical strength formulae of unidirectional composites. *Materials Letters*, 40(4):164–169, 1999.

- [4] S.h. Xin and H.m. Wen. A progressive damage model for fiber reinforced plastic composites subjected to impact loading. *International Journal of Impact Engineering*, 75:40–52, 2015.
- [5] Aswani Kumar Bandaru and Suhail Ahmad. Modeling of progressive damage for composites under ballistic impact. *Composites Part B: Engineering*, 93:75–87, 2016.
- [6] Ercan Sevkati, Benjamin Liaw, Feridun Delale, and Basavaraju B. Raju. A combined experimental and numerical approach to study ballistic impact response of s2-glass fiber/toughened epoxy composite beams. *Composites Science and Technology*, 69(7-8):965–982, 2009.

Full paper access. This is a shortened version of full research work, and is available on request. Contact at <https://mathscapes.xyz/contact>.

Usage and permission. This paper or any portion thereof may not be reproduced or used in any manner whatsoever without the express written permission of Mathscapes Research except for the use of brief quotations in a review.