

Preamble

Theory box

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Formula box

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Lab/examples box

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1 Fluids as energy carriers

1.1 Fluid state variables and properties

Formulas

1.1.1 State variables

Density

$$\rho \triangleq \frac{m}{V} \left[\frac{kg}{m^3} \right] \quad (1)$$

Specific volume

$$v \triangleq \frac{V}{m} = \frac{1}{\rho} \left[\frac{m^3}{kg} \right] \quad (2)$$

1.1.2 Viscosity

Kinematic viscosity

$$\nu \triangleq \frac{\eta}{\rho} \left[\frac{m^2}{s} \right] \quad (3)$$

Dynamic viscosity

$$\eta \triangleq \nu \cdot \rho \left[Pa \cdot s = \frac{Ns}{m^2} = \frac{kg}{m \cdot s} \right] \quad (4)$$

1.1.3 Real and ideal fluid

Real fluid

variable density ($\Delta\rho \neq 0$)
friction ($\eta > 0, \nu > 0$)

Ideal fluid

incompressible ($\Delta\rho = 0$)
frictionless ($\eta = 0, \nu = 0$)

1.1.4 Compressibility

Mach number

$$M \triangleq \frac{u}{c} \quad (5)$$

where:

- M is the Mach number [-]
- $M \lesssim 0.3$: incompressible flow
- u is the flow velocity [m/s]
- c is the speed of sound in the fluid [m/s]

and:

- $c_w^{20^\circ} = 1484$ m/s
- $c_a^{20^\circ} = 343$ m/s

1.2 Laminar and turbulent flow

Reynolds number

$$Re = \frac{v \cdot L}{\nu} = \frac{\rho \cdot v \cdot L}{\eta} [-] \quad (6)$$

where:

- v is the mean flow velocity [m/s]
- L is the characteristic length [m]

Re values

- $Re < 2000$: laminar flow
- $Re \simeq 2300$: critical point
- $2000 < Re < 4000$: transitional regime
- $Re \geq 4000$: turbulent flow

1.3 Pressure and velocity

Pressure

1.3.1 Total pressure

In addition to the static pressure p_{stat} , there is also the dynamic pressure p_{dyn} and the total pressure p_{tot} :

$$p_{\text{tot}} = p_{\text{stat}} + p_{\text{dyn}} \quad (7)$$

1.3.2 Absolute pressure

Absolute pressure p_{abs} refers to the pressure in a vacuum $p_{\text{vacuum}} = 0 Pa$ while relative pressure p_{rel} can refer to any chosen reference pressure p_{ref} .

$$p_{\text{abs}} = p_{\text{rel}} - p_{\text{ref}} \quad (8)$$

1.3.3 Velocity

Velocity is a vector quantity:

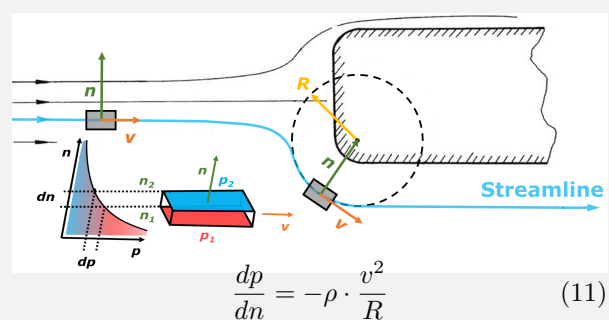
$$\vec{v} = (v_x v_y v_z) \quad (9)$$

The magnitude is given by:

$$v = \sqrt{v_x^2 + v_y^2 + v_z^2} \quad (10)$$

1.4 Curvature pressure formula

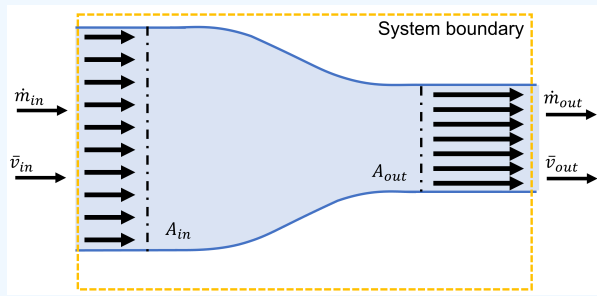
Deflection motion of a fluid element around a blunt body



2 Mass conservation

2.1 Continuity equation / Mass conservation

Continuity equation



2.1.1 Steady mass-flow

$$\dot{m}_{in} = \dot{m}_{out} \quad (12)$$

2.1.2 Incompressible fluid

$$\dot{m} = \rho \dot{V} \implies \dot{V}_{in} = \dot{V}_{out} \quad (13)$$

2.1.3 Streamline theory

$$\dot{V} = \bar{v} A \implies \bar{v}_{in} A_{in} = \bar{v}_{out} A_{out} \quad (14)$$

3 Energy conservation

3.1 Fluid mechanical energy conservation

Derivation of the Bernoulli equation

$$\dot{m}_1 \left(\frac{p_1}{\rho} + \frac{v_1^2}{2} + gz_1 \right) = \dot{m}_2 \left(\frac{p_2}{\rho} + \frac{v_2^2}{2} + gz_2 \right) \quad (15)$$

This derivation is based on the assumption that the system has:

- steady flow
- ideal fluid
- adiabatic process
- no work in or out of the system
- 1D streamline flow

3.1.1 Energy flow

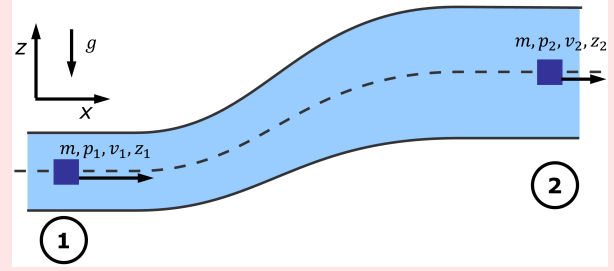
$$\begin{aligned} \frac{dE}{dt} = & \underbrace{\sum P + \sum \dot{Q}}_{\text{Energy flow across system boundary}} \\ & + \underbrace{\sum_{in} \left[\dot{m}^{\swarrow} \cdot \left(h^{\swarrow} + \frac{v_1^2}{2} + gz_1 \right) \right]}_{\text{Energy transfer mass in}} \\ & - \underbrace{\sum_{out} \left[\dot{m}^{\nearrow} \cdot \left(h^{\nearrow} + \frac{v_2^2}{2} + gz_2 \right) \right]}_{\text{Energy transfer mass out}} \end{aligned} \quad (16)$$

3.1.2 Outflow formula according to Torricelli

$$gz_1 = \frac{v_2^2}{2} \implies v_2 = \sqrt{2g\Delta z} \quad (17)$$

3.2 Bernoulli equation

Specific energy equation



$$\frac{p_1}{\rho} + \frac{v_1^2}{2} + gz_1 = \frac{p_2}{\rho} + \frac{v_2^2}{2} + gz_2 = \text{const.} \left[\frac{J}{kg} \right] \quad (18)$$

3.2.1 Alternative forms

Pressure equation

$$p_1 + \frac{\rho v_1^2}{2} + \rho gz_1 = p_2 + \frac{\rho v_2^2}{2} + \rho gz_2 = \text{const.} [Pa] \quad (19)$$

Height equation

$$\frac{p_1}{\rho g} + \frac{v_1^2}{2g} + z_1 = \frac{p_2}{\rho g} + \frac{v_2^2}{2g} + z_2 = \text{const.} [m] \quad (20)$$

True energy equation

The Bernoulli equation states that the sum of these energies is constant along a streamline.

3.2.2 Pressure energy

$$E_p = m \cdot \frac{p}{\rho} [J] \quad (21)$$

3.2.3 Kinetic energy

$$E_{kin} = m \cdot \frac{v^2}{2} [J] \quad (22)$$

3.2.4 Potential energy

$$E_{pot} = m \cdot g \cdot z [J] \quad (23)$$

3.2.5 Energy conservation

$$E_{p,1} + E_{kin,1} + E_{pot,1} = E_{p,2} + E_{kin,2} + E_{pot,2}$$

$$m \left(\frac{p_1}{\rho} + \frac{v_1^2}{2} + gz_1 \right) = m \left(\frac{p_2}{\rho} + \frac{v_2^2}{2} + gz_2 \right) \quad (24)$$

3.3 Hydrostatics

Fundamental law of hydrostatics

$$p = p_0 + \rho gh = \text{const.} [Pa] \quad (25)$$

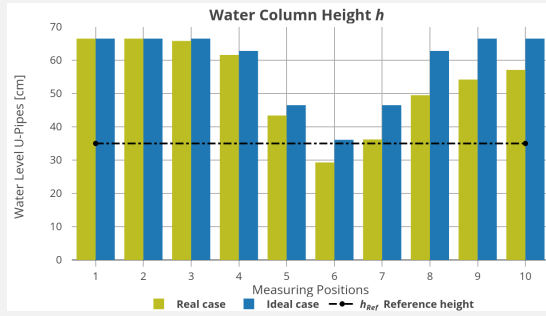
derived from:

$$p = p_0 + \frac{F_g}{A} = p_0 + \frac{mg}{A} = p_0 + \frac{\rho h Ag}{A} \quad (26)$$

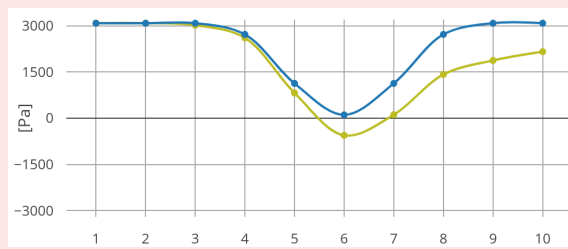
3.4 Venturi effect experiment

Venturi effect

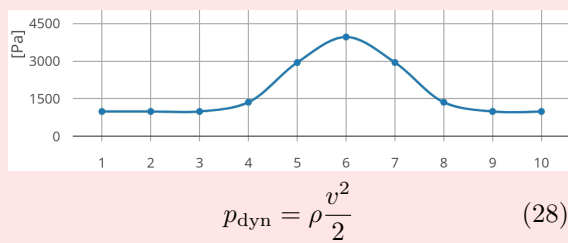
Height – pressure difference at $\dot{V} = 6 \text{ l/s}$



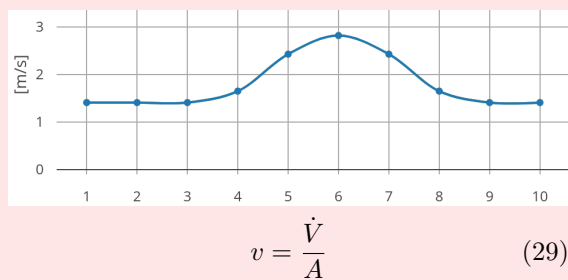
Relative static pressure p_{rel}



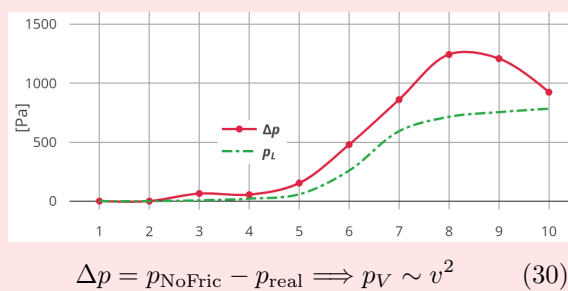
Dynamic pressure p_{dyn}



Dynamic pressure v

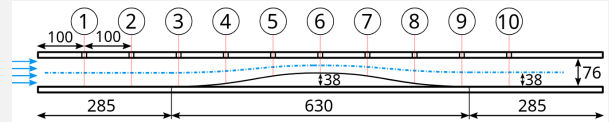


Pressure difference Δp

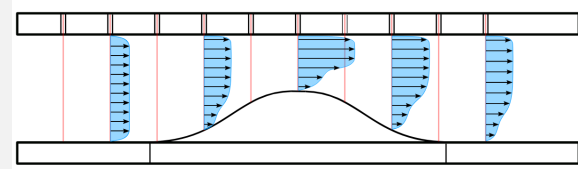


Venturi effect

Measurement points

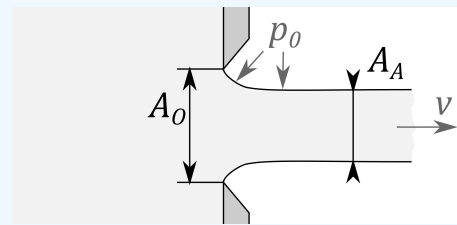


Measurement shear flow



3.5 Contraction coefficient

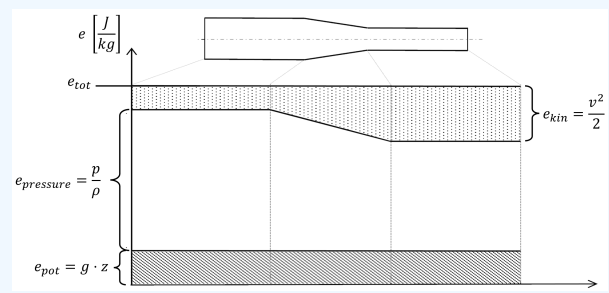
Outflow contraction coefficient α



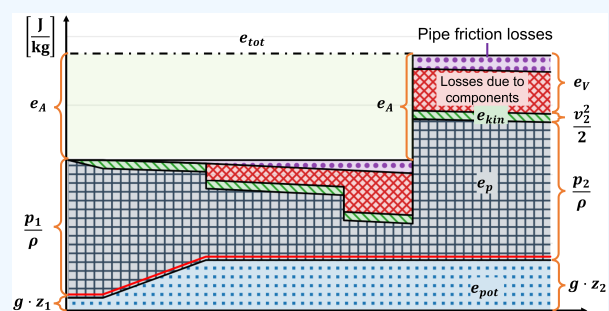
$$\alpha = \frac{A_{actual}}{A_{opening}} = \frac{\pi}{2 + \pi} \approx 0.611[-] \quad (31)$$

3.6 Energy line diagram

Ideal fluid energy line diagram



Extended energy line diagram



3.7 Extended Bernoulli equation

Extension of the Bernoulli equation

$$\frac{p_1}{\rho} + \frac{v_1^2}{2} + gz_1 + e_A = \frac{p_2}{\rho} + \frac{v_2^2}{2} + gz_2 + e_V \left[\frac{J}{kg} \right]$$

$$E_{p,1} + K_1 + U_1 + E_A = E_{p,2} + K_2 + U_2 + E_V [J] \quad (32)$$

3.7.1 Additional terms

Work term e_A

$$e_A = \frac{p_A}{\rho} = gz_A = \frac{E_A}{m} = \frac{P_A}{\dot{m}} \left[\frac{J}{kg} \right] \quad (33)$$

where:

e_A : work term [J/kg] E_A : energy difference [J]
 p_A : pressure diff [Pa] P_A : power difference [W]
 z_A : height difference [m]

If energy is added to the fluid along a streamline from point 1 to point 2 (eg. a pump), the total energy at point 2 becomes higher than at point 1.

Sign convention

$e_A > 0$: work is done on the fluid
 → energy is added to the fluid (eg. pump);

$e_A < 0$: work is done by the fluid
 → energy is extracted from the fluid (eg. turbine).

Pump and turbine work Y

In the pressure equation, the pressure p_A increase (or decrease with a turbine) can be read directly at the working term, hence:

$$e_A = \frac{E_A}{m} = Y = H \cdot g = \frac{p_A}{\rho} \left[\frac{J}{kg} \right] \quad (34)$$

The hydraulic power P_{hyd} is then given by:

$$P_{hyd} = \dot{m} \cdot Y = \dot{V} \cdot \rho \cdot Y = \rho \cdot \dot{V} \cdot g \cdot H [W] \quad (35)$$

Specific loss term e_V

$$e_V = \frac{p_V}{\rho} = gz_V = \frac{E_V}{m} = \frac{P_V}{\dot{m}} \left[\frac{J}{kg} \right] \quad (36)$$

where:

e_V : loss term [J/kg] E_V : energy loss [J]
 p_V : pressure diff [Pa] P_V : power loss [W]
 z_V : height loss [m]

The effects of a viscous fluid along a streamline from point 1 to point 2 are taken into account by e_V .

Pressure loss Δp_V

$$\Delta p_V = e_V \cdot \rho = \frac{E_V \cdot \rho}{m} = g \cdot z_V \cdot \rho = \zeta \cdot \rho \cdot \frac{v^2}{2} [Pa] \quad (37)$$

3.8 Loss behavior in turbulent flows

Zeta value

$$\zeta = \frac{2 \cdot \Delta p_V}{\rho \cdot v^2} \quad (38)$$

Total pressure loss

If multiple losses occur in a system due to sequentially connected hydraulic components, the total loss $\Delta p_{V,tot}$ is given by the sum of the individual losses:

$$\Delta p_{V,tot} = \sum_i \Delta p_{V,i} = \sum_i \zeta_i \cdot \rho \cdot \frac{v_i^2}{2} [Pa] \quad (39)$$

$$\Delta p_{V,tot} = \rho \cdot \frac{v^2}{2} \cdot \sum_i \zeta_i = \rho \cdot \frac{v^2}{2} \cdot \zeta_{tot} [Pa] \quad (40)$$