

¹ Measurement and Identification of the Nonlinear Dynamics of a
² Jointed Structure Using Full-Field Data; Part II - Nonlinear
³ System Identification

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19 Abstract

The dynamic responses of assembled structures are greatly affected by the mechanical joints, which are often the cause of nonlinear behavior. To better understand and, in the future, tailor the nonlinearities, accurate methods are needed to characterize the dynamic properties of jointed structures. In this paper, the nonlinear characteristics of a jointed beam is studied with the help of multiple identification methods, including the Hilbert Transform method, Peak Finding and Fitting method, Dynamic Mode Decomposition method, State-Space Spectral Submanifold, and Wavelet-Bounded Empirical Mode Decomposition method. The nonlinearities are identified by the responses that are measured via accelerometers in a series of experiments that consist of hammer testing, shaker ringdown testing, and response/force-control stepped sine testing. In addition to accelerometers, two high-speed cameras are used to capture the motion of the whole structure during the shaker ringdown testing. Digital Image Correlation (DIC) is then adopted to obtain the displacement responses and used to determine the mode shapes of the jointed beam. The accuracy of the DIC data is validated by the comparison between the identification results of acceleration and displacement signals. As enabled by full-field data, the energy-dependent characteristics

35 of the structure are also presented. The setup of the different experiments is described in
36 detail in Part I of this research. The focus of this paper is to compare nonlinear system
37 identification methods applied to different measurement techniques and to exploit the use
38 of high spatial resolution data.

39 *Keywords:* Jointed Structures, Nonlinear System Identification, Digital Image Correlation
40 (DIC), Amplitude-Dependent Characteristics, Energy-Dependent Characteristics

41 **1. Introduction**

42 Mechanical joints are of crucial importance in the dynamics of assembled structures
43 [1]. Significant nonlinearities can be caused by friction, wear, and non-idealized boundary
44 conditions [2, 3, 4]. With slip and clapping behaviors happening between the interfaces
45 [5], the complexity of the joints makes the modeling highly challenging. Therefore, it is
46 extremely important to characterize the behaviors of jointed structures experimentally with
47 high accuracy, which is a pressing need in several industries such as aerospace [6], automotive
48 [7], and naval [8].

49 Among several tools available, backbone curves are a common method for describing
50 nonlinear systems [9, 10] as they characterize the amplitude-dependent natural frequency
51 and damping for single mode oscillations from free decay data [11]. The interactions that
52 may occur in the system can be elucidated, enabling the investigation of modal energy
53 exchange due to the nonlinearities, which cannot be analyzed by conventional linear system
54 identification methods. Moreover, the backbone curve makes it possible to estimate or
55 update the nonlinear characteristics within a model based on the experimental response [12,
56 13, 14, 15]. Even more so than frequency, damping is critical in predicting and understanding
57 the behaviors of a structure [16, 17]. Often, the damping nonlinearity in a jointed structure
58 changes the damping capacity of a structure by orders of magnitude across a range of
59 response amplitudes while the stiffness nonlinearity changes the natural frequencies by a few
60 percent [18]. The amplitude-damping curve is a general way to demonstrate the dissipation
61 of systems [10], especially for jointed structures [1]. Together, the knowledge of amplitude-
62 dependent damping and amplitude-dependent frequency characteristics can be used to make
63 predictions for the steady-state forced harmonic response of a structure. The key to this
64 is the single-nonlinear-mode theory, for which simple, explicit approximations are available
65 [19, 20, 21].

66 There are many different methods for calculating the amplitude-dependent frequency and
67 damping curves, hereafter also referred to as the frequency and damping backbone curves.
68 The Hilbert Transform (HT) method [22, 23] is generally considered to be a reliable time-
69 domain identification method [9] and relatively accurate compared with other identification
70 methods [24]. The Peak Finding and Fitting (PFF) method [25] is another time-domain
71 method, which is recently proposed based on the traditional zero-crossing (ZC) method
72 [10]. One limitation of the above two methods is that they are mono-modal, i.e., they
73 are formulated to analyze one mode and one point at a time and cannot identify modal

interactions. One approach to address this shortcoming is the use of frequency-domain methods, such as the Dynamic Mode Decomposition (DMD) method, which finds coherent spatial-temporal pattern from the dataset [26]. When the dimension of acquisition is less than the rank of the system, the modified version of DMD, i.e., the Hankel DMD [27, 28], can be used to analyze the system. For analyzing vibrations of mildly nonlinear systems, the window based Hankel DMD can be used to obtain the time-varying modal parameters. A fundamentally different approach from time-domain and frequency-domain methods is the identification of the nonlinear normal modes of a system. Nonlinear normal modes (NNMs) [29] are a popular and useful tool for the analysis of vibrations in nonlinear, conservative systems. Similarly, Spectral Submanifolds (SSMs) [30] represent the extension of classical nonlinear normal modes to forced-damped nonlinear structures. SSMs can be exploited to gain precious insights on the dynamical behavior [31, 32] and to construct reduced-order models [33]. A methodology was proposed to analytically compute both the shape of SSMs and their corresponding backbone curves according to a data-assimilating model that was fitted on experimental vibration signals [34]. This means that a data-driven polynomial state space model can be constructed based on SSMs. Besides the aforementioned methods, time-frequency analysis is another powerful way to extract information about frequency and damping in structural dynamics. Empirical Mode Decomposition (EMD) is a technique that decomposes a signal into a finite set of intrinsic mode functions (IMFs) [35]. Wavelet-Bounded EMD (WBEMD) is an improvement over EMD, improving the separation of IMFs, which represent different characteristic time scales. In this manner, the WBEMD method is able to identify not just frequency and damping information, but also the portions of the time record that the information is extracted from, allowing an analyst to visualize how the system changes over time across a large range of frequencies [36]. These system identification methods are described more in detail in Sect. 2.

Nonlinearities are usually characterized by amplitude-dependent curves [9, 37], i.e., the frequency/damping is plotted as the function of the amplitude. The shortcoming of this representation is that the characterization of the nonlinearity is related to the amplitude of a certain accelerometer, thus a specific point/location. In the view of reaching a more global metric, attention has been paid to the relationship between the energy stored in a mode and the corresponding frequency characteristics (hereafter referred to as the frequency-energy relationship) [38]. While frequency-energy plots can be easily computed in analytical or numerical studies [29, 39], their experimental determination remains a challenge. The energy of a physical structure is difficult to calculate based on the information of either dense or sparse accelerometers.

With the development of high-speed cameras, the motion of the whole structure can be recorded by images with high spatial resolution. Digital Image Correlation (DIC) [40, 41] can then process the recorded images and extract displacements of the structure. Although they have lower accuracy, high-speed cameras perform non-contact measurements and allow higher spatial resolution than accelerometers. DIC has recently been applied to study jointed structures [5, 42] by measuring the slip and separation behaviors in the interfaces. Additionally, DIC has advantages over other non-contact measurement techniques, chiefly

116 laser Doppler vibrometer (LDV). For instance, DIC is often a more affordable approach
117 than LDV and the latter is also more time consuming, since a LDV has to scan across the
118 surface one point at a time. The disadvantage of current DIC methods, though, is the large
119 post-processing time to extract the large amount of data from videos of the experiment as
120 well as limitations in memory for data storage.

121 The number of testing methods for extracting amplitude-dependent characteristics from
122 nonlinear structures has been growing in the last two decades [38]. Hammer excitation
123 or imposed initial conditions are used to trigger decaying oscillations [24], but they typ-
124 ically lack in isolating single mode responses. Another family of methods, such as force
125 appropriation [43, 44] or control-based continuation [45], exploits mono-harmonic forcing
126 from a shaker as well as the phase-quadrature criterion to isolate resonant vibrations and
127 to extract amplitude-dependent properties. However, these methods can be excessively
128 time-consuming and they can suffer from structure-shaker interactions. Resonance decay,
129 or shaker ringdown testing [46, 47], has been introduced to solve these issues. Here, a
130 high-amplitude resonant vibration is isolated via phase-quadrature using the shaker and,
131 afterwards, the shaker is decoupled from the structure, which shows single-mode decaying
132 responses. Although decoupling the shaker from the structure is not a trivial operation, the
133 extraction of amplitude-dependent properties is faster with respect to other shaker-based
134 methods.

135 This research consists of two parts. The first [48] focuses on experimental investigations,
136 describing the tools, methods, and challenges for extracting the data from a benchmark
137 system in vibrations of joint-assembled structures. Both accelerometers and high-speed
138 cameras are adopted to record the motion of the system over time. In this work, the spatially
139 dense data from [48] is analyzed in order to assess the efficacy of using DIC for nonlinear
140 system identification. Multiple nonlinear system identification methods are utilized and
141 compared with each other in order to establish their performances and to highlight pros and
142 cons from a practitioner viewpoint. Both free decay vibration and forced-response testing
143 are analyzed for the extraction of backbone and damping curves and, due to the availability
144 of full-field data and energy estimations, their interconnections are investigated with novel
145 metrics. Moreover, this research shows the agreement of data from the accelerometers and
146 DIC and sheds further light on amplitude-dependent properties by inspecting the shape
147 variations and the energy of the structure.

148 The paper is organized as follows. In Sect. 2, the different methods adopted are summa-
149 rized. Section 3 reports on the analysis of experiments, first focusing on accelerometer data
150 and then on DIC data, which comes from hammer testing, shaker ringdown testing, and
151 response/force-control stepped sine testing. Additionally, an energy criterion is established
152 to connect amplitude-dependent properties from forced-response testing with those obtained
153 via free-decay response. Section 4 presents a critical comparison of the adopted methods,
154 and Sect. 5 concludes the paper.

155 **2. Nonlinear System Identification Methods**

156 In this section, the five methods utilized throughout this research are briefly reviewed,
157 including the Hilbert Transform (HT) method [22, 23], Peak Finding and Fitting (PFF)
158 method [25], State-Space Spectral Submanifold (SS-SSM) method [34], Dynamic Mode De-
159 composition (DMD) method [27, 28], and Wavelet-Bounded Empirical Mode Decomposition
160 (WBEMD) method [36]. Among these methods, the HT and PFF are time-domain ap-
161 proaches, the DMD belongs to frequency-domain method, the WBEMD is a time-frequency
162 analysis method [9], and the SS-SSM is a data-driven modeling technique. Each of these
163 methods are applied to analyze the data from different experiments in Sect. 3.

164 *2.1. The Hilbert Transform Method*

165 Generally, the HT method is regarded as one of the most popular and reliable nonlinear
166 system identification methods. It was proposed by Feldman [22, 23] and was applied to
167 identify different nonlinear systems [49, 50, 51]. The sum of the signal and its HT composes
168 the analytical signal. When the analytical signal is written in polar coordinates, the in-
169 stantaneous amplitude and instantaneous frequency can be calculated. However, smoothing
170 is needed for the identified characteristics [52]. There are a few ways to accomplish the
171 smoothing, including filtering the identification results and mirroring the original signal be-
172 fore applying the HT method. According to [24], Empirical Modal Decomposition (EMD)
173 [35] is an effective way that can be combined with the HT method to provide the smoothed
174 results for both instantaneous amplitude and frequency. Based on the local time scale of
175 the signal, EMD can be used to decompose the signal into intrinsic mode functions [35], the
176 first of which generally contains the highest frequency components. By excluding the high
177 frequency components, EMD is able to smooth the results identified by the HT method.
178 This means that EMD serves as a lowpass filter, but, importantly, it does not introduce a
179 phase shift [24]. After smoothing, the damping ratio can be calculated based on the two
180 smoothed quantities.

181 *2.2. The Peak Finding and Fitting Method*

182 The PFF method [25] is based on the zero-crossing (ZC) method [10], which calcu-
183 lates instantaneous frequency according to the time instants where the signal crosses zero.
184 The identification accuracy of PFF was demonstrated by several cases in [24]. In the PFF
185 method, the local maxima (and minima) of the signal are first searched. With the combi-
186 nation of the local maxima (and minima) and the two points near them, the signal near the
187 local maxima (and minima) can be approximated by a polynomial. Therefore, improved
188 estimations of the true values of the peaks (and valleys) are obtained from the fitted polyno-
189 mial, which can overcome errors due to under-sampling. Then, the PFF method calculates
190 instantaneous frequency as the reciprocal of the difference between two sequential time in-
191 stants at which the signal reaches these peaks (and valleys). The PFF method is able to
192 provide the result of the instantaneous frequency that achieves similar accuracy as the ZC

method. Furthermore, by improving the extraction of the peaks, the instantaneous amplitude of the response can be obtained with higher accuracy by the PFF method. Therefore, the damping ratio, which is calculated by the instantaneous frequency and instantaneous amplitude, has less noise compared with that obtained by the ZC method [25]. Moreover, unlike the HT method, the instantaneous frequency and instantaneous amplitude calculated by the PFF method do not need to be smoothed, thus saving computational time.

2.3. The State-Space Spectral Submanifold Method

A spectral submanifold (SSM) is referred to as the smoothest nonlinear continuation of a linearized mode shape of an equilibrium [30]. Thus, SSMs are calculated in order to analyze the reduced dynamics of the system, giving a nonlinear extension of the linear dynamics of the modal subspace. Moreover, the amplitude-dependent frequency curve along SSMs has been shown to approximate the backbone curve of forced frequency responses [11]. With respect to methods developed in the framework of conservative nonlinear normal modes [29], the theory of SSMs relaxes some critical assumptions on the purely parasitic effect of damping. Thus, the SS-SSM data-driven method developed in [34] puts no restriction on the type and magnitude of the damping; however, it does require that the signal being analyzed have a dynamic response that is sufficiently close to a single-mode (or two-dimensional) SSM of the underlying system. The data are simultaneously assimilated into a nonlinear discrete-time state space model from which the parametrization and the reduced SSM dynamics are analytically computed, giving a nonlinear extension of the linear dynamics of the modal subspace. The reduced-order model of the SSM is then exploited to derive amplitude-dependent properties. This approach was demonstrated to reproduce backbone curves with high accuracy according to the results of both numerical and experimental data [34, 53].

2.4. The Dynamic Mode Decomposition Method

For the fluid dynamics community, DMD is one of the most promising system identification tools since it allows for high dimensional measurements. This algorithm finds coherent spatial-temporal patterns from a densely measured dataset [26]. In this paper, since dense spatial data is provided due to the combination of high-speed cameras and DIC, DMD is used to comprehend the underlying dynamics of the structure. As the rank of the structural vibration is much less than the dimension of data acquisition, this type of system identification technique is well suited. Also, for the cases when only one accelerometer is used, this technique is still able to provide accurate information about the modal parameters of the system. In the present context, the modified version of DMD, known as Hankel DMD [27, 28], is used to analyze the system. The formulation of Hankel DMD is similar to the formulation of the Eigensystem Realization Algorithm [28, 54] that is typically used for system identification of linear dynamical systems. Hence, in this study, for analyzing the vibration of the jointed structure that is a mildly nonlinear system, window-based Hankel DMD is used to capture the time-varying modal parameters of the structure. It is assumed that in a small time window, this mildly nonlinear structure approximately behaves like a

232 linear dynamical system. Amplitude information is obtained via filtering at frequencies of
233 interest.

234 *2.5. The Wavelet-Bounded Empirical Mode Decomposition Method*

235 By exploiting EMD, a signal is decomposed into a finite set of nearly orthogonal, monochro-
236 matic intrinsic mode functions (IMFs) [35, 55]. Theoretically, every IMF should be a physical
237 and mathematical representation of a single characteristic time scale contained in the origi-
238 nal oscillatory signal. However, in practice, EMD yields spurious, non-physical IMFs, which
239 need to be eliminated before starting the dynamical analysis [56]. Another issue with the
240 regular EMD is the problem of mode mixing, where one IMF is comprised of components at
241 different frequencies and, thus, is not a fair representation of any single time scale extracted
242 from the original signal. The WBEMD augments the regular EMD by using a masking sig-
243 nal, which provides a quantitative measure of the isolation of an IMF around a characteristic
244 frequency [36]. This measure is used as the objective function of a minimization problem.
245 In the WBEMD algorithm, after applying EMD, the IMF is transformed into the maximum
246 wavelet domain [57] where a bounding function is fitted over the IMF. This procedure is
247 able to extract optimally-separated IMFs, each representing distinct time scales present in
248 the signal. Once frequency information is obtained, ad-hoc filtering is applied to retrieve
249 amplitude and damping characteristics of the signal.

250 **3. Analysis of Experiments**

251 Using the nonlinear system identification methods described in the previous section, the
252 experiments on the Half Brake-Reuß Beam (HBRB) described in [48] are analyzed. These
253 experiments include hammer testing, shaker ringdown testing, and response/force-control
254 stepped sine testing. **Figure 1** summarizes the experiments, along with the pre-processing
255 techniques and the identification method used to extract nonlinear information. The HBRB
256 is modified from a normal BRB [1], which is a prismatic 304 stainless steel beam joined
257 in the middle through a three-bolt lap joint (**Fig. 2**), to be half of the nominal thickness.
258 This modification was chosen in order to reduce the first three natural frequencies of the
259 system such that high-speed videography could easily measure the response of the first three
260 modes without sacrificing spatial resolution. The nonlinear characteristics extracted from
261 the hammer testing and shaker ringdown testing are compared with the backbone curves
262 extracted directly from the response/force-control stepped sine testing [48].

263 During the shaker ringdown testing, apart from accelerometers, two high-speed cameras
264 are used to capture images of the HBRB. DIC [40, 41] is applied to the images, providing
265 the displacement responses of the beam. The specific set of measurements used from each
266 experiment are further highlighted in the flowchart of **Fig. 1**. During this study, each bolt is
267 tightened to 10 N·m and the first mode is mainly considered (except where noted otherwise).
268 Further details of all of the experiments in this section can be found in Part I of this research
269 [48].

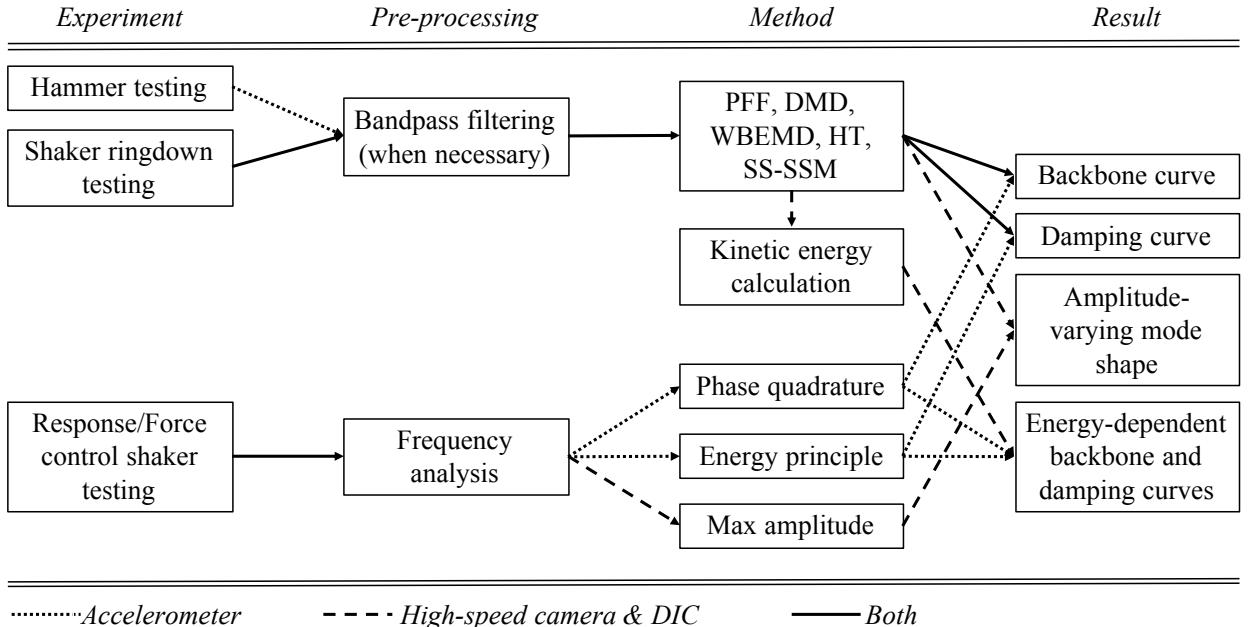


Figure 1: Flowchart that represents the experiments and the methods discussed in this paper, also including the necessary pre-processing and the identification results. Different measurements, from accelerometer or speed camera, are highlighted using different linestyles.

270 3.1. Free Decay Response by Accelerometers

271 In this section, the identification results from free decay responses of accelerometer mea-
 272 surements are discussed. Specifically, the amplitude-dependent properties extracted via
 273 hammer testing and shaker ringdown using the methods presented in Sect. 2 are discussed.

274 3.1.1. Hammer Testing

275 During the hammer testing of the HBRB, free boundary conditions are achieved by
 276 hanging the beam with the combination of fishing lines and bungees [48, 15]. The constraints
 277 (at the node points of the first mode) are applied since the beam should be confined within a
 278 limited space, and it is demonstrated in Part I that the specific arrangements of constraints
 279 used here do not significantly affect the characteristics of the beam [48]. The HBRB is
 280 excited by the hammer and is measured by three accelerometers (**Fig. 2**) with a sampling
 281 frequency of 6400 Hz. As an example, a free decay response obtained by the accelerometer
 282 nearest to the impact location (indicated as Accel 3 in **Fig. 2**) is shown in **Fig. 3 (a)**.

283 With the time history response, the methods described in **Sect. 2** are applied to obtain
 284 backbone curves and amplitude-damping curves of the HBRB, with the exception of the
 285 SS-SSM method. This method is excluded since it assumes that the data are dominated by
 286 a single-mode SSM, which is not the case in hammer testing.

287 The backbone curves extracted in this section are defined as the amplitude-frequency
 288 relations of principal modes, or components, of the decaying signals. Since the HT and

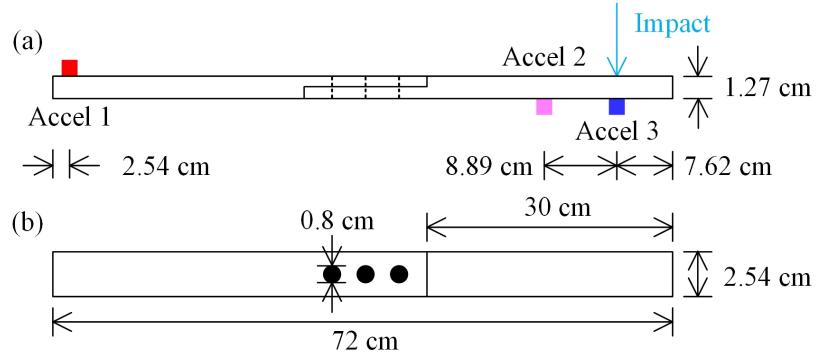


Figure 2: The setup of the hammer testing for the HBRB, showing from the (a) top and (b) side views.

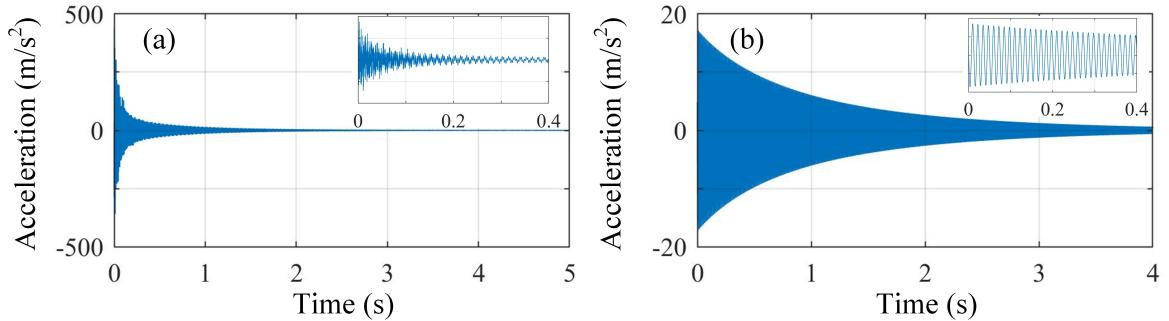


Figure 3: The free decay response measured by Accel 3, showing (a) raw response (with zoomed response from 0 s to 0.4 s) and (b) filtered response (with zoomed response from 0 s to 0.4 s).

289 PFF methods can only be applied to a signal with one frequency component, the free decay
 290 response has to be filtered [58] before these two methods are applied. The effects of applying
 291 a third-order butterworth bandpass filter whose frequency band is from 75.27 Hz to 85.27
 292 Hz are shown in **Fig. 3 (b)**. The acceleration response is truncated at 4 s to obtain the
 293 identification results with a high signal-to-noise ratio. Additionally, for the HT method, the
 294 signal is further truncated from 0.5 s to 3.9 s so that the end effects can be reduced.

295 As shown in **Fig. 4**, all of the different methods show great agreement in the identification
 296 of both the backbone curve and amplitude-damping curve for the first mode. This means
 297 that the instantaneous frequency, amplitude, and damping ratio are well identified by these
 298 methods. The change in damping (from 0.13% to 0.25%) as the response amplitude increases
 299 from 0.4 m/s^2 to 17 m/s^2 is significantly greater than the change in frequency (from 80.3
 300 Hz to 80 Hz), which is typical in the context of jointed structures.

301 The spikes observed in the damping curve obtained by the HT method are caused by
 302 the calculation of the damping ratio, which is related not only to the amplitude, but also
 303 to its derivative [25]. Thus, although the signal was already truncated before using the HT
 304 method, there are still errors at the two ends in the damping ratio, which is one of the
 305 drawbacks of the HT method.

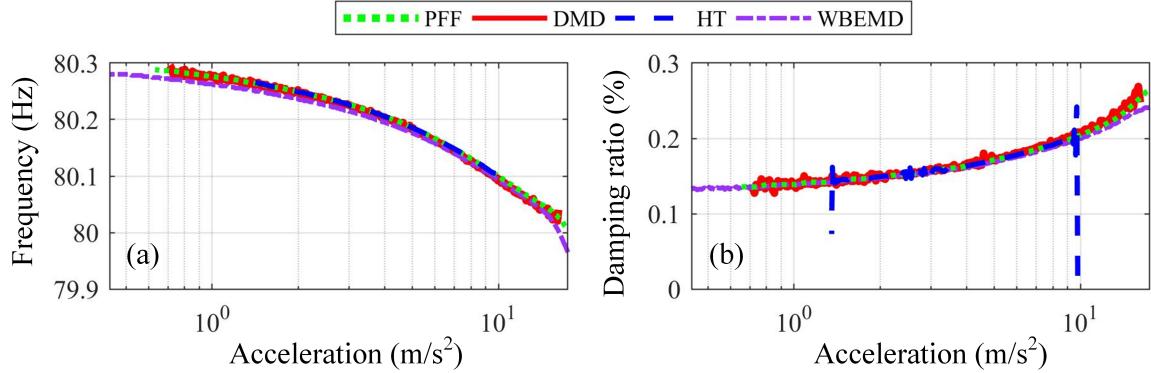


Figure 4: The identification results of the hammer testing by the PFF (green dotted line), HT (blue dashed line), DMD (red solid line), and WBEMD (purple dash-dotted line) methods, showing (a) backbone curve and (b) amplitude-damping curve.

In comparing the PFF, DMD, and WBEMD methods, the PFF and DMD method both overlay the HT method, while the WBEMD method results in slightly lower predictions, both for frequency and damping. This is justified by the fact that WBEMD does not pre-filter the signals, in contrast with the other methods. Lastly, both the PFF and WBEMD show significantly less noise, especially in measurements of damping, than the DMD and HT methods.

3.1.2. Shaker Ringdown Testing

With the experimental setup similar to that of the hammer testing, shaker ringdown testing for the first mode of the HBRB is performed. The shaker is attached to the same location as the hammer impact point in **Fig. 2** and it is used to obtain approximately single-mode responses [59]. After isolating a resonant frequency response (reaching a 90-degree phase lag between force and driven point) or NNM, the shaker is physically detached from the HBRB [60], and the system decays to the equilibrium along the related SSM [30], as illustrated in **Fig. 5**. See [48] for a more detailed description of the experiment. Hence, the backbone curve computed is the SSM free decay instantaneous frequency-amplitude relation. Apart from the three accelerometers, two high-speed cameras are used to measure the displacement of the beam. DIC [40, 41] is applied to the captured images to acquire the displacement responses. The analysis of the displacement responses is presented in Sect. 3.3, while this section focuses on the acceleration measured by the accelerometer located on the shaker excitation point.

The Short-time Fourier transform (STFT) analysis is conducted on the response of the shaker ringdown testing. As it can be seen from **Fig. 5** (b), the energy is only in the first mode (below 100 Hz) and it stays in the first mode, except at the moment when the shaker is decoupled from the beam; however, the effect shown by the STFT analysis (**Fig. 5(b)**), does not persist past 30.2 seconds, which is where the signal is truncated for analysis. All of the methods described in **Sect. 2** are applied to the signal from time 30.2 s

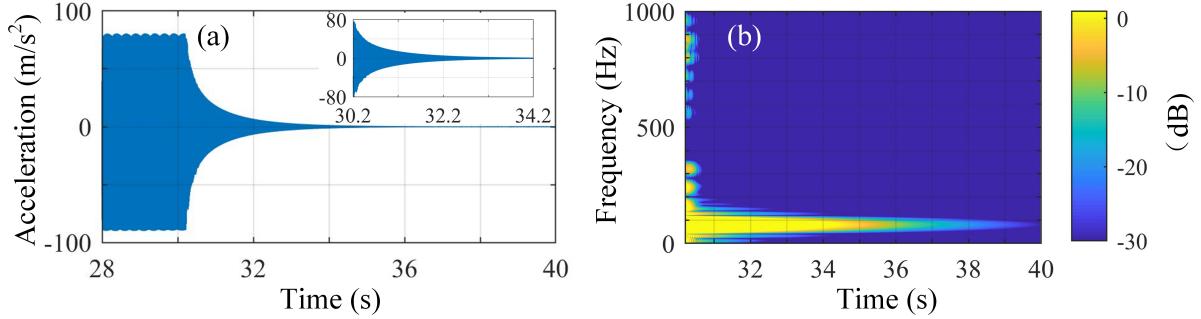


Figure 5: The acceleration response measured by Accel 3, showing (a) raw response from 28 s to 40 s (with zoomed response from 30.2 s to 34.2 s) and (b) STFT analysis of the response from 30.2 s to 34.2 s.

to 40 s (i.e., from the moment that the shaker is decoupled from the system until the noise floor is reached). It is shown in **Fig. 6** that there is little difference in the identification results, which means that the instantaneous frequency, amplitude, and damping ratio are all identified with similar accuracy by these methods, with several exceptions discussed in what follows. The instantaneous damping ratio rises with the amplitude increasing, from about 0.1% to more than 0.6% (**Fig. 6 (b)**). Thus, a strong nonlinearity in the damping of the HBRB is observed. The instantaneous frequency undergoes small variations (0.5 Hz, or 0.6%) in this amplitude regime. Unlike the impact hammer tests, which were limited in terms of the amount of energy that they could put into a single mode and could not drive the system to a high enough amplitude to make this observation, the shaker ringdown experiments exhibit a softening-hardening trend as the amplitude increases (**Fig. 6 (a)**). As shown in the next sections, this trend is not a characteristic of this specific accelerometer, but rather of the whole structure.

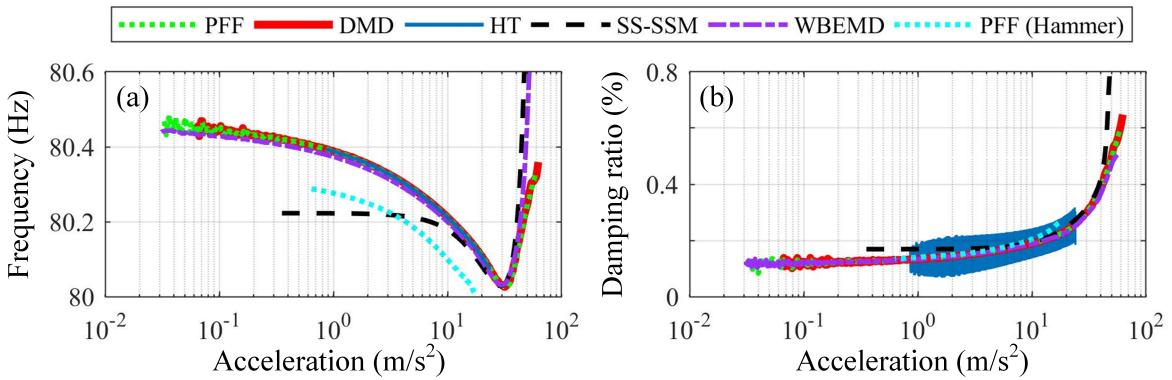


Figure 6: The identification results of the shaker ringdown testing by the PFF (green dotted line), DMD (red thick line), HT (blue thin line), WBEMD (purple dash-dotted line), and SS-SSM (black dashed line) methods, showing (a) backbone curve and (b) amplitude-damping curve. The cyan dotted lines are the identification results of the hammer testing using the PFF method.

One challenge for analysis is that most methods (e.g., the PFF method) are only capable of dealing with a signal that contains exactly one frequency component. Thus, although

347 the free decay response of the shaker ringdown testing can be approximately regarded as
348 the signal with single frequency, the response is still filtered by the third-order butterworth
349 bandpass filter (that is used in the hammer testing) before the PFF method is applied.
350 The identification results of the HT method have much fewer data points than the other
351 methods, since the curves have been truncated to avoid the end effects [52]. Although the
352 identified frequency is smoothed [35] before being used to calculate the damping ratio, the
353 corresponding damping ratio shows the most noise among all of the methods. The results
354 obtained by the WBEMD method slightly differ from those of the others at high amplitudes.

355 The SS-SSM method, with respect to the other signal processing ones, analyzes the dy-
356 namics of a discrete-time state space model fit on the signals shown in **Fig. 5** (a). The results
357 in **Fig. 6** have been obtained using a 6-dimensional system with polynomial nonlinearities
358 up to order 5. The state space is made of measurement delays (using the typical settings
359 of autoregressive-moving-average models [34]), and the SSM expansion is then carried out
360 up to order 19. There is a discrepancy in the frequency of the SSM model and those of
361 the other backbones for low amplitudes since the SSM model has a well-defined linearized
362 frequency that, along with the model nonlinearities, is optimized in order to minimize the
363 model prediction errors.

364 Although **Fig. 4** and **Fig. 6** both show the results from free decay responses, these two
365 tests have their own characteristics. To further compare the results from the two tests,
366 the backbone and amplitude-damping curves obtained by the PFF method are presented
367 in **Fig. 6**. This figure shows that the highest amplitude in the identification results of the
368 shaker ringdown testing (80 m/s^2) is significantly larger than that of the hammer testing
369 (20 m/s^2). This shows that the shaker ringdown tests are able to acquire a wider range of
370 amplitudes than the hammer testing, providing more information about the variation of the
371 amplitude-dependent frequency and damping ratio. This is due to the broadband nature
372 of the hammer excitation in contrast with the possibility to isolate a single mode without
373 appreciably exciting the other modes with shaker ringdown testing. Specifically, the initial
374 acceleration in the hammer testing reaches 500 m/s^2 (**Fig. 3** (a)), while the acceleration of
375 the shaker ringdown testing approaches only 100 m/s^2 (**Fig. 5** (a)). However, the response
376 containing only the first mode has the initial acceleration of 20 m/s^2 in the hammer testing
377 (**Fig. 3** (c)), and 80 m/s^2 in the shaker ringdown testing (the zoom in **Fig. 5** (a)). Thus,
378 most of the energy is dissipated along a single mode or SSM in shaker ringdown testing, while
379 hammer testing allows the interplay of multi-modal nonlinear dissipation. As evidenced in
380 **Fig. 6** by the discrepancies between the amplitude properties, other frequency components
381 present in the signal influence the amplitude curves of the first mode, even though proper
382 filtering is performed. Therefore, if the aim of testing is to study modes in isolation, shaker
383 ringdown testing should be preferred due its capability to isolate single modes and to obtain
384 larger amplitude ranges.

385 3.2. Forced-response Testing

386 The forced-response testing aims at validating the previous free decay results. The
387 instantaneous frequency-amplitude curve of decaying vibrations has been often observed

388 to serve as a backbone for the frequency response for small forcing [46, 47, 43, 61, 62]
 389 (in particular in [63, 64, 21] for non-conservative systems), and this relation is formally
 390 established in [11]. Specifically, the backbone curve of a single mode SSM is the leading
 391 order approximation of frequencies and amplitudes at which forced response maxima occur
 392 for different forcing values.

393 In this research, both response-amplitude-control and force-amplitude-control stepped
 394 sine testing are conducted on the HBRB for extracting forced responses. These are measured
 395 by accelerometers to identify the backbone and amplitude-damping curves. The difference
 396 between the two types of testing is that, in the response-amplitude-control testing, the re-
 397 sponse amplitude of the driven point is controlled to be constant. By contrast, the amplitude
 398 of the shaker force is controlled to remain fixed in the force-amplitude-control testing. Since
 399 the first mode of the HBRB is concerned, the excitation frequency is swept from 79 Hz to
 400 81 Hz with the step of 0.05 Hz. The response-amplitude-control testing is conducted with
 401 amplitudes of 0.5 g, 2 g, 4 g, 5 g, 6 g, 7 g, and 8 g, and the force-amplitude-control testing
 402 contains the force levels of 0.2 N, 0.5 N, 1.0 N, and 1.5 N. More detailed description of
 403 the forced-response testing can be found in Part I [48]. **Figure 7** (a) shows the measured
 404 response amplitude for the two types of forced testing, depicting the response-control testing
 405 in red and in green the force-control testing. In what follows, the principal harmonics in
 406 force and acceleration signals are only considered as higher order harmonics have negligible
 influence for the system under analysis, cf. **Sect. 3.4.1**.

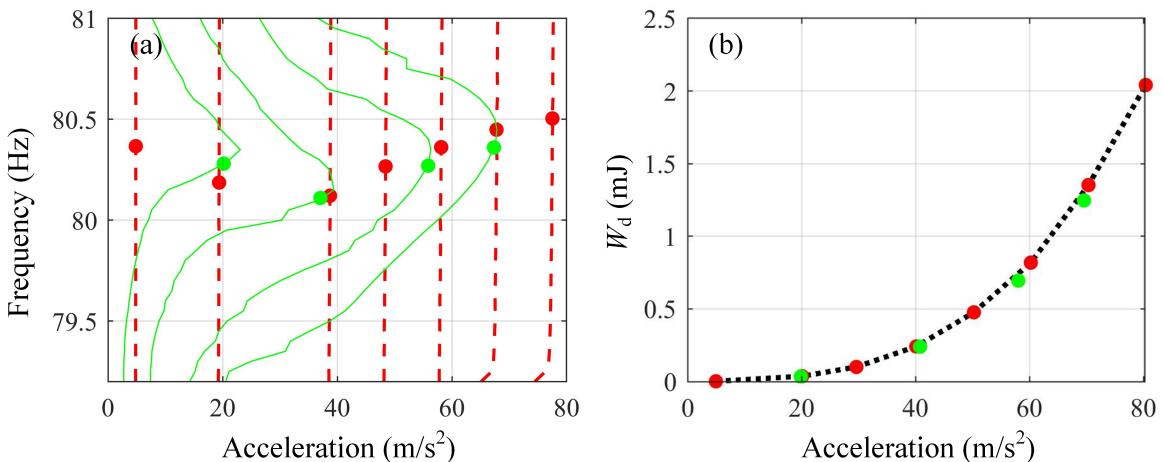


Figure 7: The results of the forced-response testing, showing (a) measured response amplitude in the response-control testing (red dashed line) and force-control testing (green dashed line) with the respective resonance frequency indicated with circles, and (b) dissipated energy over one oscillation cycle at resonance frequencies shown with consistent colors. Here, the black line connects red markers.

407
 408 The extraction of maximal or resonant responses is performed by selecting the responses
 409 satisfying the quadrature phase-lag criterion [46, 44, 61]. Here, the displacement response of
 410 the driven point has approximately a 90° phase lag with respect to the forcing signal. The
 411 identification of the backbone curve is then straightforward for both tests and is indicated by

412 circles in **Fig. 7** (a). The identification of the damping ratio, however, is not as immediate.
 413 For a periodic trajectory of a mechanical system, the energy principle states that the work
 414 done by non-conservative forces must be zero in one oscillation cycle [63, 44]. In other
 415 words, the input energy provided by the shaker W_a is dissipated, mainly due to friction
 416 occurring at the interface. **Figure 7** (b) shows the energy dissipated W_d in one cycle of
 417 resonant oscillation. By denoting the first harmonic component of the acceleration of the
 418 driven point response $\ddot{q}(t) = -A \sin(\Omega t)$, the forcing amplitude F , and the resonant forcing
 419 frequency Ω , it holds that

$$W_a = \int_0^{2\pi/\Omega} \dot{q}(t) F \cos(\Omega t) dt = \frac{\pi A F}{\Omega^2}, \quad (1)$$

420 and the dissipated energy over one oscillation cycle is

$$W_d = \int_0^{2\pi/\Omega} 2m^* \zeta_A^* \Omega \dot{q}^2(t) dt = 2m^* \zeta_A^* \frac{\pi A^2}{\Omega^2}, \quad (2)$$

421 where m^* is the equivalent modal mass. An equivalent, amplitude-dependent, modal damp-
 422 ing ratio ζ_A^* can be determined from the equality $W_a = W_d$

$$\zeta_A^* = \frac{F}{2m^* A}. \quad (3)$$

423 The modal mass is estimated by calibration with the results from the free decay response,
 424 which results in $m^* = 1.55$ kg. Another computation of the dissipated energy is introduced
 425 later in **Sect. 3.4.3** exploiting full-field data from DIC.

426 The comparison between amplitude properties extracted from forced testing and the ones
 427 from free decay response are shown in **Fig. 8**. The results of the response-control and force-
 428 control testing are close to each other. Additionally, the results of the two forced-response
 429 testing are in good accordance with those of the shaker ringdown testing, especially in the
 430 curves of damping. Although there is some discrepancy in the identification of frequency,
 431 the trend of the backbones obtained by the forced-response testing is the same as that
 432 obtained by the shaker ringdown testing. The partial discrepancy could be justified by the
 433 low resolution (0.05 Hz) of forced testing, some control limitations at low amplitudes, and
 434 the influence of the metallic stinger used to attach the beam to the shaker. The latter results
 435 in an overall stiffness increase, consequently shifting the frequencies as shown in **Fig. 8** (a).
 436 Moreover, from Eqs. 1-3, the ratio between the external forcing amplitudes F and those of
 437 the inertial forces $m^* A$ is equal to $2\zeta_A^*$, which stays small, thereby justifying the comparison
 438 between the results obtained via free decay response and forced response.

439 Although the response-control and force-control stepped sine testing are both capable of
 440 identifying nonlinear characteristics with high accuracy, they have several drawbacks. First,
 441 the identification is generally sparse as forced experiments require a considerable amount of
 442 time. Additionally, at least for the system under investigation, the experimental equipment
 443 showed limitations in reaching low amplitude levels of excitation.

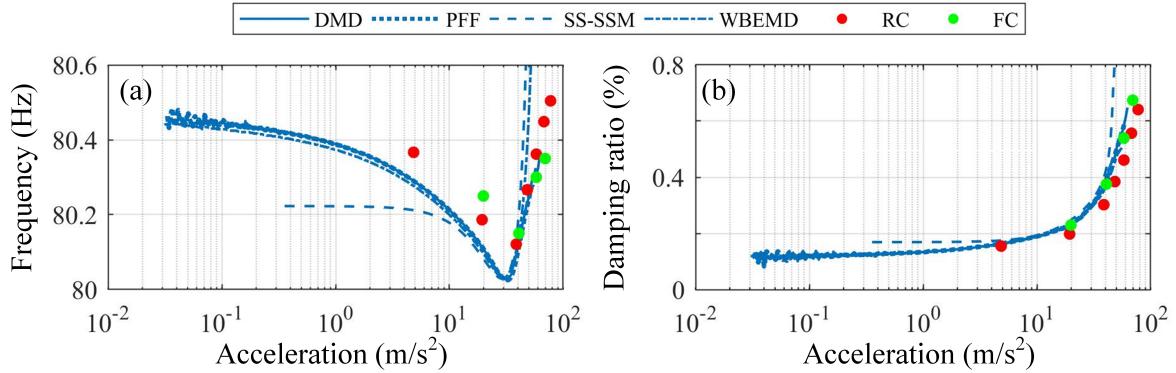


Figure 8: The identification results of the response-control testing (red circles), force-control testing (green circles), and shaker ringdown testing (blue lines) using different identification methods, showing (a) backbone curve and (b) amplitude-damping curve.

444 3.3. Free Decay Response by DIC

445 Following the analysis of the accelerometer data in the previous section, this section
 446 focuses on the measurements from the two high-speed cameras, which are used to capture
 447 the displacement of the HBRB during the shaker ringdown testing. DIC [40, 41] is applied to
 448 the recorded images to calculate the displacement responses. Since there are multiple points
 449 in the images, the raw response and the filtered response (filtered by the same butterworth
 450 filter that is used in the hammer testing) of one of the points are selected to be shown
 451 in **Fig. 9**. The raw response mainly contains the first mode of the HBRB. Although the
 452 decaying signals shown in **Fig. 5** (a) and **Fig. 9** (a) comes from the same experiment,
 453 the presence of other frequency components is more evident in **Fig. 9**. In particular, the
 454 oscillation due to sway mode of the beam, at low frequency, is clearly visible, especially at the
 455 initial time instances of the measurement. The reason why it is more visible in the DIC data
 456 of **Fig. 9** (a) than in the accelerations in **Fig. 5** (a) is that DIC processes displacements,
 457 which are more evident at low frequencies.

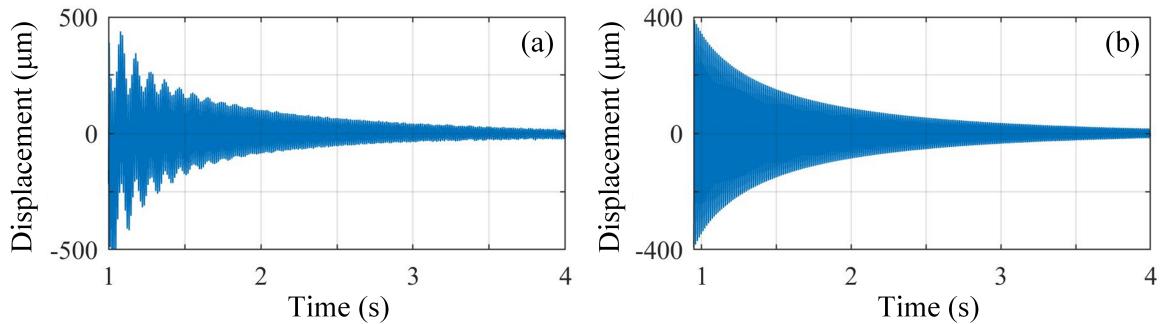


Figure 9: The displacement response of one of the points of the HBRB in the shaker ringdown testing, showing (a) raw response obtained by DIC and (b) filtered response.

458 Since all of the methods have already been compared when dealing with the acceleration

459 signal in the hammer testing and shaker ringdown testing in Sect. 3.1, only the PFF
 460 and DMD methods are applied here to the displacement responses obtained via DIC. An
 461 important distinction between the two methods is that the PFF method is applicable to
 462 the response signal of one single point, while the DMD method is based on the Hankel
 463 matrix and can thus be applied to the response signals of all of the points together. In
 464 other words, given a large number of displacement responses, the PFF method will provide
 465 multiple curves, each of which corresponds to one of the points, and the DMD method will
 466 result in only one curve that contains the information of all of the points.

467 Note that the time series obtained from DIC have different signal-to-noise ratios across
 468 the length of the beam. In particular, time series whose locations lie close to the nodes
 469 of the mode shape are predominantly affected by noise. Additionally, during a free decay,
 470 signal-to-noise ratios worsen as time increases so that the time series have to be truncated
 471 to preserve a high signal-to-noise ratio.

472 The traditional way to present the backbone curve of a structure is to plot the fre-
 473 quency as a function of the response amplitude. **Figure 10** shows the identification results
 474 of the PFF method regarding the backbone curves and amplitude-damping curves of the
 475 measurement points on the beam.

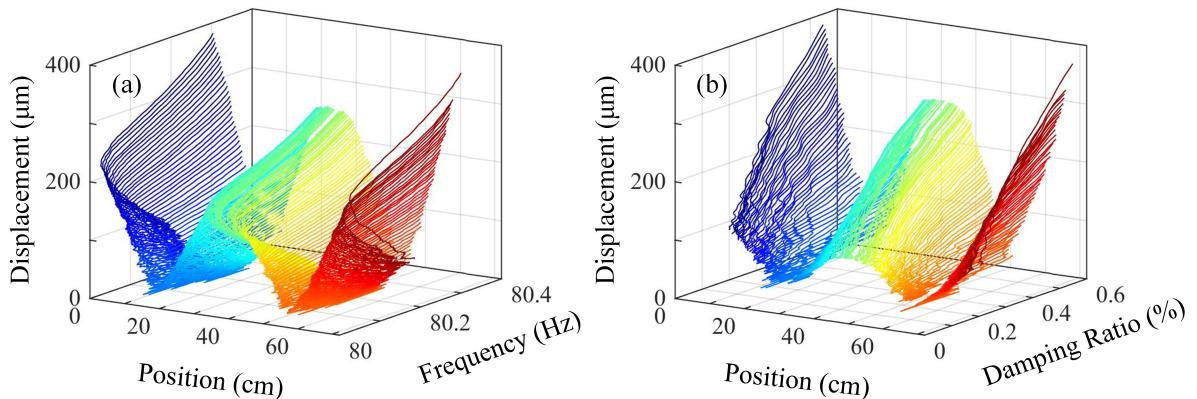


Figure 10: The identification results of the DIC data using the PFF method, which correspond to the measurement points at different positions, showing (a) backbone curves and (b) amplitude-damping curves.

476 It can be seen from **Fig. 10** that although the range of the frequency/damping ratio
 477 is the same for each point, the range of the displacement amplitude differs. Since all of
 478 the points are in different positions on the beam, the amplitude of each point is different
 479 from each other. Thus, it may not be particularly meaningful to express the frequency
 480 or damping ratio as a function of the displacement amplitude. Another way to normalize
 481 the response is to plot the frequency and damping ratios as a function of time, i.e., the
 482 instantaneous frequency and instantaneous damping ratio. The results of both the PFF
 483 and DMD methods are shown in **Fig. 11**, whose x-axes appear reversed when compared to
 484 amplitude-based plots, cf. **Fig. 6**.

485 Regarding the displacement responses obtained by DIC, both the PFF method and the

486 DMD method bring about similar results of the instantaneous frequency and damping ratio.
 487 As the PFF method is applied to the response signal of each measurement point, it provides
 488 a large number of curves whose variability is recast to noise effects. By contrast, because the
 489 DMD method combines the response signals of all of the points together and calculates one
 490 curve for each amplitude-dependent characteristic, the results of the DMD method can be
 491 regarded as the average of those obtained by the PFF method. Moreover, the instantaneous
 492 frequency and damping ratios from DIC data are consistent with those from accelerometer,
 493 as show in **Fig. 11** which includes the curves obtained via DMD from accelerometer data.
 494 However, acceleration signals have longer time histories than those obtained by the DIC data
 495 due the memory of the cameras. In particular, videography-based techniques are limited by
 496 the trade-off between spatial resolution and sampling frequency, which both determine the
 497 maximum length of the time signals possible based on available camera memory.

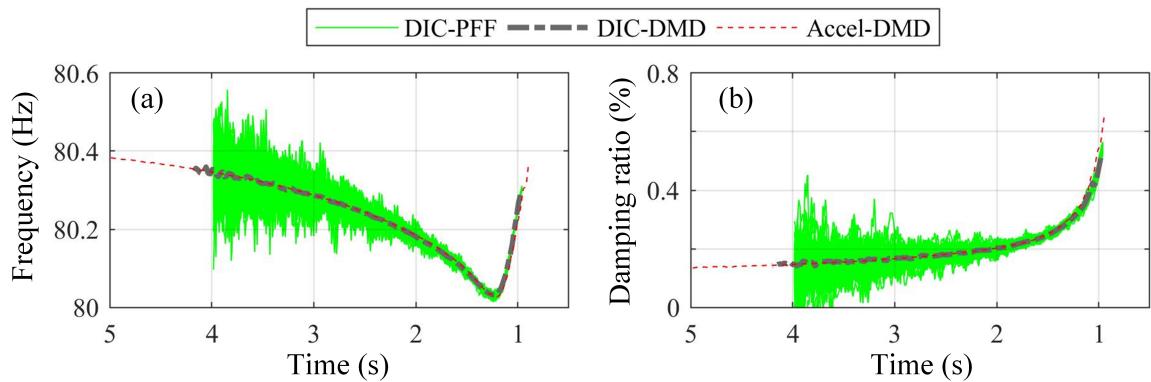


Figure 11: The results obtained by the DIC data and acceleration data using PFF (green solid lines) and DMD (grey dashed-dotted line for DIC data and red dashed line for acceleration data) methods, showing (a) instantaneous frequency and (b) instantaneous damping ratio.

498 3.4. Additional Information Brought by DIC

499 Unlike the accelerometers that are sparsely distributed on the beam, the cameras are
 500 able to measure the displacement of an entire face of the HBRB, so that all of the points
 501 on the beam are recorded simultaneously. After the images are processed by DIC, the full-
 502 field information of the beam is available (i.e., the displacement responses of the points are
 503 obtained). Hence, DIC data allows the detailed analysis of the beam's mode shape and also
 504 the evaluation of its energy. Note that a similar degree of information cannot be achieved
 505 by accelerometers since it is difficult to place dense accelerometers on the structure without
 506 changing its characteristics at the same time [15].

507 3.4.1. Measurements of Mode Shapes from Ringdown Testing

508 In shaker ringdown testing, the motion of the beam during the free decay can be assumed
 509 as an estimation of the mode shape of the considered mode, since the beam is released from
 510 the quadrature between force and acceleration of the driven point [48]. The dominance of the

511 fundamental harmonic during the decay is shown in **Fig. 5** (b) and the motion synchronism
512 is further discussed and verified both at the end of this subsection and in **Fig. 13**. The mode
513 shape is calculated as the deflection of the beam at each maximum of the cycle of motion.
514 In this way, the time-varying (and thus, amplitude-varying) mode shape is obtained. The
515 number of measurement points depends on the spatial resolution chosen for DIC [40, 41]
516 [48]. The ultimate result consists of two lines describing the mode shape of the upper and
517 lower part of the beam. These lines are displayed on a sketch of the beam in **Fig. 12** (a).

518 In **Fig. 12**, the shape of the first mode over 3.5 s of free decay is shown. This time
519 duration is approximately 277 cycles of vibration; for visualization purposes, only 30 time
520 instants are presented in **Fig. 12**. The abscissa is the position of the measurement points and
521 the ordinate (Y axis) shows the amplitude of the responses of the corresponding measurement
522 points. Each line consists of 206 DIC data points. It can be seen from **Fig. 12** that the
523 amplitudes of the mode shapes reach zero at around 17 cm and 55 cm, which are the two
524 positions of the node points of the first mode. Additionally, there is a discontinuity of the
525 curvature of the mode shape at 29.7 cm for the lower line (blue) and at 42.3 cm for the upper
526 line (red). These locations correspond to the left and right edges of the interface between
527 the two half beams, shown in **Fig. 12** (a). When the amplitude of motion increases, the
528 described discontinuity intensifies, due to the geometry of the beam. The mode shapes in
529 **Fig. 12** refer only to the maxima of the cycle of motion, i.e., a deflection downwards of
530 the center of the beam and upwards of the extremities. By contrast, the nodes of the mode
531 shape preserve their location when motion amplitude changes.

532 The discontinuity in the mode shape is due to the opening of the interface on the lower
533 edge (**Fig. 12** (c)), and closing of the interface on the upper edge (**Fig. 12** (d)). This
534 behavior is also visible in **Fig. 12** (b), which reconstructs the maximum modal displacement
535 of the entire beam from the two lines of that come from the DIC processing (**Fig. 12** (a)),
536 and is consistent with other measurements of a similar lap joint [65, 66]. The difference
537 between the upper and lower lines shows that the maximum and minimum differences occur
538 at the locations of the discontinuities (29.7 cm and 42.3 cm (**Fig. 12** (e))), clearly reflecting
539 the gap opening at the interface. However, the deflection ratio between upper and lower
540 lines, depicted in **Fig. 12** (f), does not reveal significant variations over time.

541 Full-field data can also be used to assess whether the decaying vibrations shows synchro-
542 nism or not. According to the original definition of a conservative nonlinear normal mode by
543 Rosenberg [67], a periodic motion is synchronous if all maxima and minima of the position
544 time series are achieved at the same time. Due to the presence of damping, synchronous
545 motion is not guaranteed, but resonant motions in weakly damped structures have been still
546 observed to be (approximately at least) synchronous [46, 47]. In **Fig. 13** (a), a time-extract
547 of the full displacement field for the lower line of the beam is illustrated, where the abscissa
548 displays time, cf. **Fig. 7** (b), the ordinate displays the position along the beam, and the
549 color scale depicts the displacement magnitude. The horizontal white lines that appear from
550 the colormap correspond to the nodes of the mode shape, while the vertical white lines show
551 the times at which all of the responses are equal to zero. At least for this time portion, the
552 decaying motion may already appear synchronous. This fact has also been verified in detail

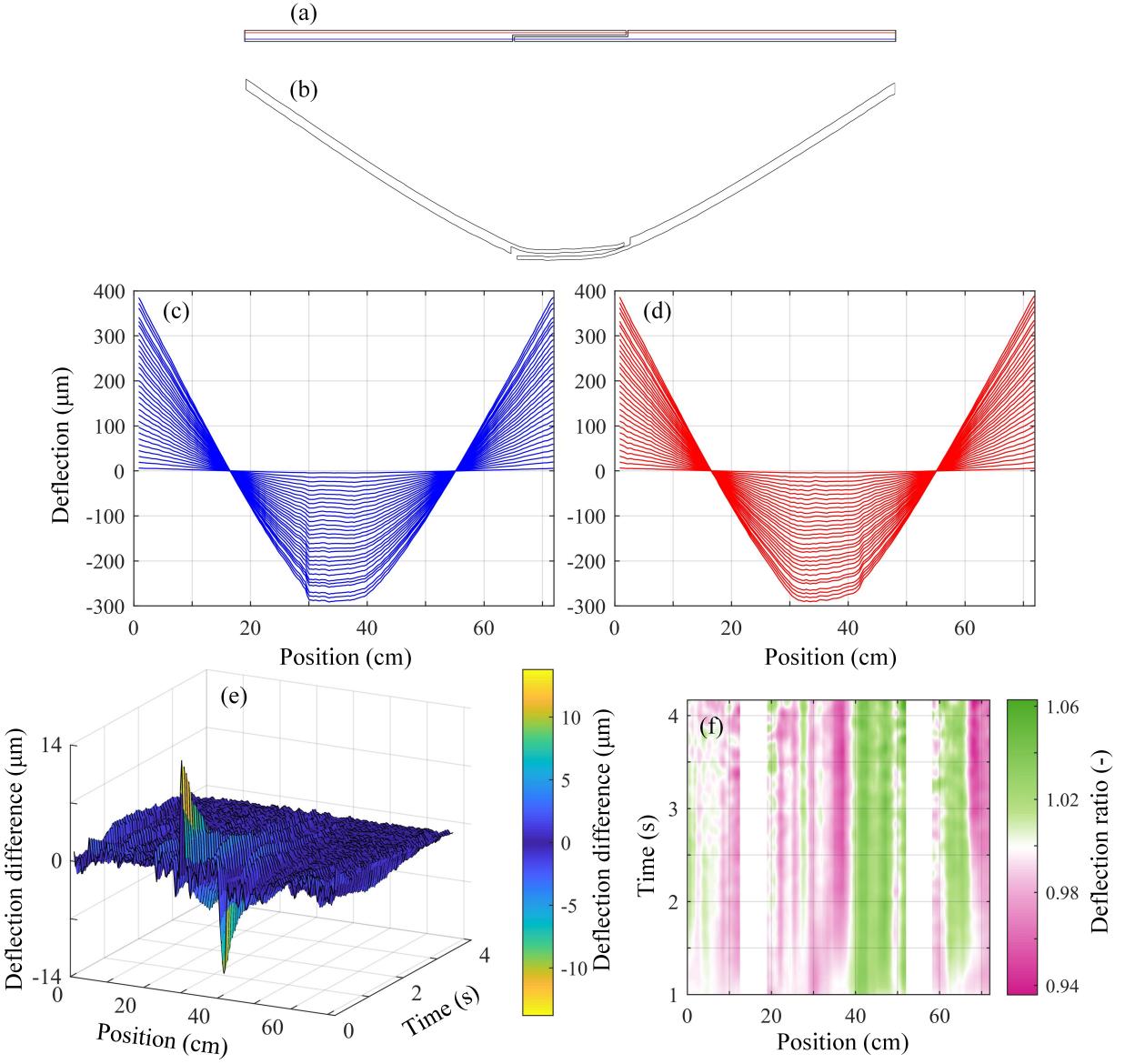


Figure 12: Amplitude-dependent mode shape of the 1st mode of the HBRB from shaker ringdown testing, showing (a) sketch of the HBRB with the two lines identified with DIC, (b) maximum deflection shape, (c) DIC lower line, (d) DIC upper line, (e) difference, and (f) ratio between upper and lower line.

exploiting the PFF method, which computes the time at which a local extremum (either maximum or minimum) occurs for every displacement time series. Let $t_{ext}(l, k)$ be the time at which the k -th extremum occurs at the location l and, by calling Ω the location domain, a metric to describe the synchronous behavior of the system is defined as

$$\Delta S(k) = \max_{l \in \Omega} |t_{ext}(l, k) - \bar{t}_{ext}(k)|, \quad \bar{t}_{ext}(k) = \frac{1}{|\Omega|} \sum_{l \in \Omega} t_{ext}(l, k). \quad (4)$$

557 The quantity $\Delta S(k)$ acts as a metric for assessing the degree of synchronism in the structure
 558 and it is plotted in **Fig. 13** (b) once normalized by the sampling time $t_{samp} = 0.0002$ s.
 559 For computing $\Delta S(k)$, the points close to mode shape nodes have been excluded from the
 560 calculation domain Ω since their displacement time series are dominated by noise. Moreover,
 561 the increasing variability of ΔS over time is due to the deterioration of the signal-to-noise
 562 ratio for decreasing motion amplitudes. This analysis leads to the conclusion that the time
 563 at which an extremum is detected for a particular location differs by less than one sampling
 564 interval from the average time at which the same extremum is detected at any other location
 565 across the beam. A similar result holds for the times at which the displacement field is zero.
 566 Hence, the decaying vibration can be approximated as synchronous and this evaluation has
 567 crucial consequences for the evaluation of the energy and the mode shape of the beam, as
 568 discussed in **Sect. 3.4.3**.

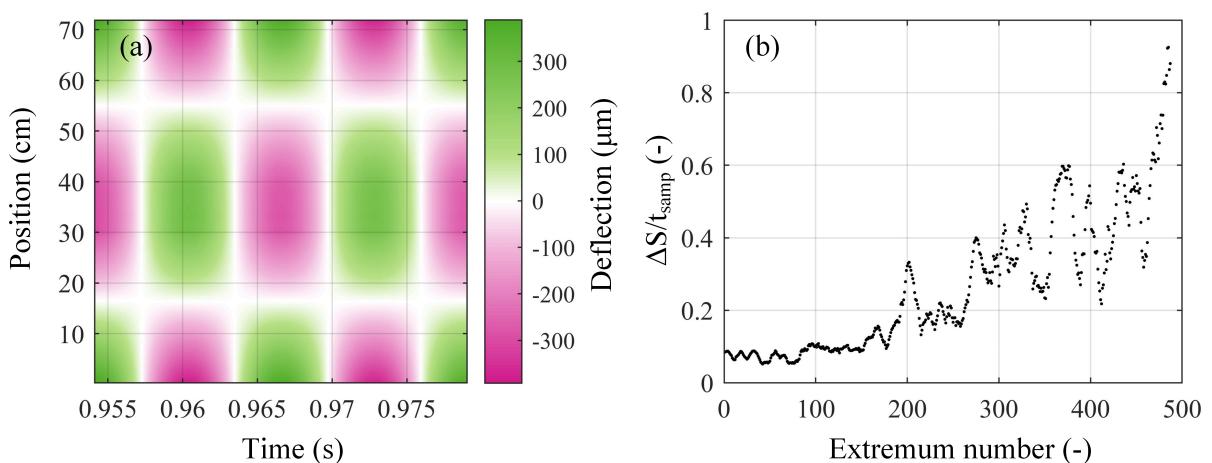


Figure 13: Assessment of motion synchronism, showing (a) full-field motion of the beam lower line for a selected time interval and (b) variability of the peak time identified via the PFF method.

569 *3.4.2. Measurements of Mode Shapes from Response-Amplitude-Controlled Fixed Sine Test-*
 570 *ing*

571 Further information of the beam's mode shape can be extracted from response-amplitude-
 572 controlled fixed sine testing. The mode shape can be estimated from the time series, which
 573 has a controlled frequency, unlike the decaying vibrations measured in the ringdown tests.
 574 As the fixed sine testing is conducted to ensure that the system achieves a steady state
 575 response, multiple forcing cycles of the steady state response are able to be measured for
 576 each excitation amplitude. Tests are conducted for the response levels 0.5 g, 2 g, 4 g, 8 g,
 577 and 10 g, and the shaker is tuned for each amplitude to excite the first mode, as described
 578 in [48]. The same considerations of **Sect. 3.4.1** about the synchronism of the motion apply
 579 here. Moreover, **Fig. 14** shows the response spectrum of the beam extremity for the highest
 580 level of the excitation, where the higher harmonics are larger. The peaks below 8 Hz are
 581 rigid body modes, and the peak at 146 Hz is due to the fan internal to each of the high-speed

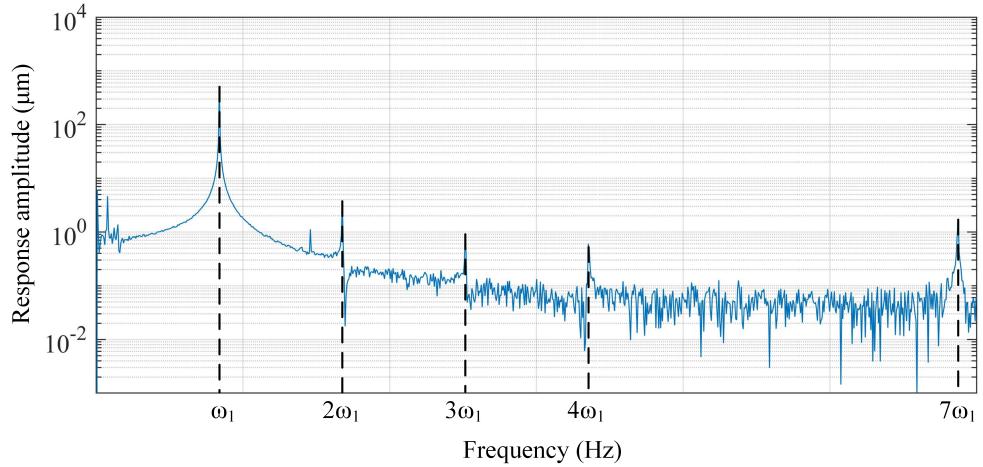


Figure 14: Response spectrum of response-amplitude-controlled fixed sine test at 10 g.

582 cameras [48]. Although several harmonics of the first mode are present, their amplitude is
 583 several orders of magnitude lower than the fundamental harmonic. In order to smooth the
 584 results, the data are filtered by a third-order bandpass butterworth filter centered at the
 585 first natural frequency with a bandwidth of 40% of the first natural frequency. The results
 586 of the mode shape estimation are shown in **Fig. 15** for each amplitude level, both for the
 587 upper and lower lines. Two additional plots shed further light on the amplitude-dependent
 588 properties of the mode shape. The first one, **Fig. 16** (a), shows the upper line mode shape
 589 of a central portion of the beam, normalized with respect to the left tip displacement. From
 590 the lowest to the highest excitation amplitude, the displacements of the central part of the

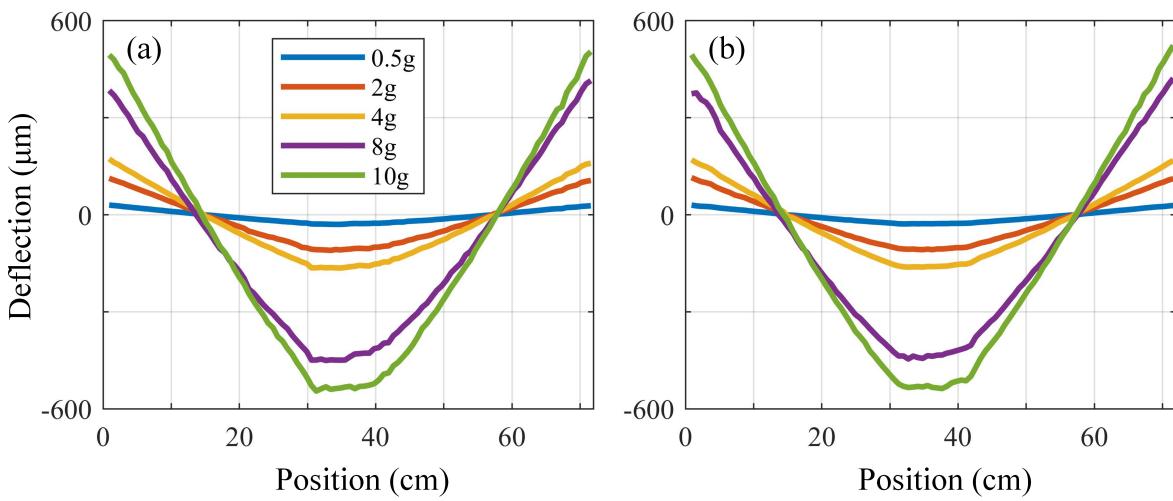


Figure 15: Amplitude-dependent mode shape of the 1st mode of the HBRB from response-amplitude-controlled testing, showing (a) DIC lower line and (b) DIC upper line. Each line is the mode shape calculated from the maximum displacement for an excitation level.

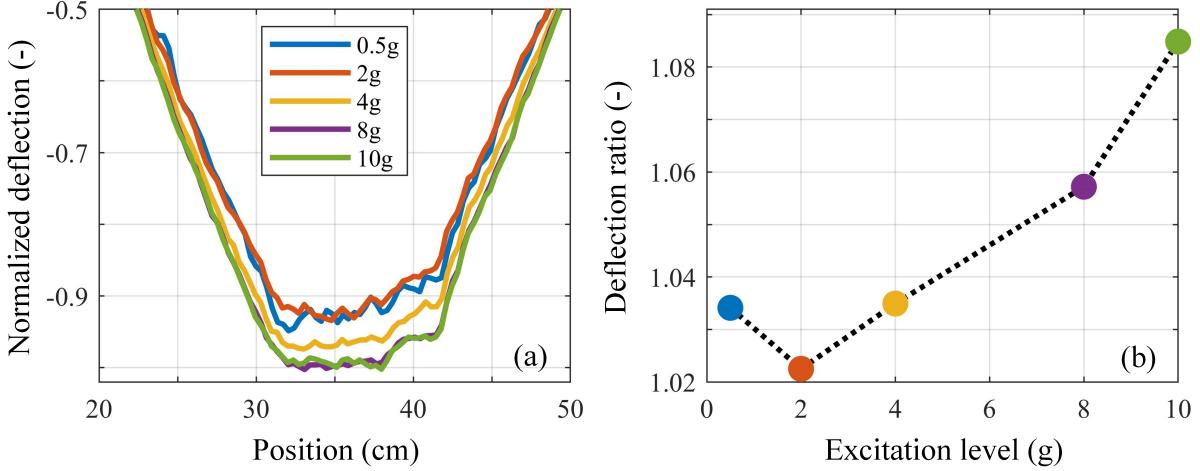


Figure 16: Amplitude-dependent mode shape of the 1st mode of the HBRB from response-amplitude-controlled testing of DIC upper line, showing (a) normalized deflection and (b) deflection ratio between the middle of the beam to the tip displacement.

beam increase approximately by 7%, with respect to the tip displacement, but decrease for the 2 g level. This suggests a softening behavior of the modal stiffness. The overall trend resembles that of the backbone curve, but the variation in this case is greater with respect to the frequency change in the backbone curve, cf. **Fig. 8**, since this is a local phenomenon while the natural frequency depends on global characteristics. This behavior is illustrated more compactly in **Fig. 16** (b) with the deflection ratio between the displacement of the middle of the beam to the tip displacement. Precisely, the tip displacement is computed as the average between the displacements of the ten leftmost points from DIC, and the middle displacement is the average among the ten displacements around the beam center. While the locations of the mode shape nodes remain unchanged, the principal variation of the mode shape is concentrated in the middle of the beam, suggesting that further studies should connect this macroscopic observation with microscopic phenomena of the contact patch.

3.4.3. Energy of the Beam

The total energy of a structure consists of kinetic energy and potential (strain) energy. As shown in the STFT analysis (**Fig. 5** (b)), the free decay response obtained from the shaker ringdown testing only contains the first mode of the HBRB. The potential energy is assumed to be zero at the equilibrium position, corresponding to the configuration when all displacements are zero. As verified previously, the vibration is synchronous and, during the decay, every point passes through zero at the same time. At these time instances, the total energy of the structure is therefore equal to the kinetic energy.

As DIC provides responses of a large number of points, the HBRB can be discretized into lumped mass elements. The filtered displacement responses (**Fig. 9** (b)) can be transferred into velocity responses by numerical differentiation. Thus, with the full-field information of

614 the structure and the mass of the whole beam ($M = 1.796$ kg as measured), the energy E
 615 at time instant t can be approximated by

$$E(t) = \sum_{i=1}^N \frac{1}{2} m v_i^2(t), \quad (5)$$

616 where N is the number of the measurement points, v_i is the velocity of the i th point, and
 617 $m = M/N$ is the mass of each measurement point.

618 Here, the total number of the measurement points is $N = 206$, which is determined during
 619 the processing of the DIC. The energy trend, computed using all measurement points, over
 620 the shaker ringdown testing is shown in **Fig. 17** (a) with a black line, and the red dots depict
 621 energy peaks. In order to study the influence of the number of the measurement points on
 622 the calculation of E , N is varied from 5 to 206 over an equally spaced grid. **Figure 17**
 623 (b) shows the energy of the beam, at the time instant of the first peak, as a function of N ,
 624 which converges for $N > 100$. In other words, the number of points N needed to have an
 625 accurate estimation of the energy is higher than 100. This spatial resolution is not achievable
 626 with contact sensors such as accelerometers or strain gauges, but only with a non-contact
 627 measurement approach, such as videography and DIC.

628 The relationships between energy and displacement/acceleration amplitudes for the ac-
 629 celerometer located on the shaker (see Accel 3 in **Fig. 2** (a)) is shown in **Fig. 17** (b). A
 630 clear, quadratic trend, as defined in Eq. (5), is observed. The accuracy of the energy cal-
 631 culation by DIC is validated in **Fig. 17** (d) by means of the dissipated energy. The red
 632 and green dots are the work done by the forcing for the response-control (red) and force-
 633 control (green) testing procedures for different amplitudes, cf. **Fig. 7** (b). These quantities
 634 are computed using the measurement of the shaker force and the acceleration of the driven
 635 point. As argued in **Sect. 3.2**, this work is equal to that done by the non-conservative
 636 forces over one oscillation cycle. For the shaker ringdown, the peaks in the kinetic energy
 637 correspond to the total energy of the system, since the motion is synchronous. Hence, the
 638 difference between the energy of the k -th peak and that of the $(k+2)$ -th peak (see the inset
 639 in **Fig. 17** (a)) is the work done by the non-conservative forces in one oscillation cycle. The
 640 dissipative work computed from forced testing and from shaker ringdown is expected to be
 641 comparable from theoretical speculations. Indeed, **Fig. 17** (d) shows that there is a very
 642 good agreement with the two. Here, the blue line shows the dissipative work, computed
 643 using the kinetic energy decay obtained from full-field data, plotted against the acceleration
 644 amplitude level. It agrees well with the energy measurements obtained in forced testing
 645 using a single accelerometer.

646 The energy computed here could serve for model validation or tuning, but it is also useful
 647 to construct frequency-energy plots, typically adopted in the NNM literature [9] to charac-
 648 terize the dynamical feature of nonlinear responses. Frequency-energy plots are particularly
 649 meaningful for conservative systems, where each NNM orbit lies on a specific energy level
 650 [29]. However, they are relevant also for weakly forced, damped structures as their resonant
 651 periodic motions typically show small deviations from their conservative counterpart, which

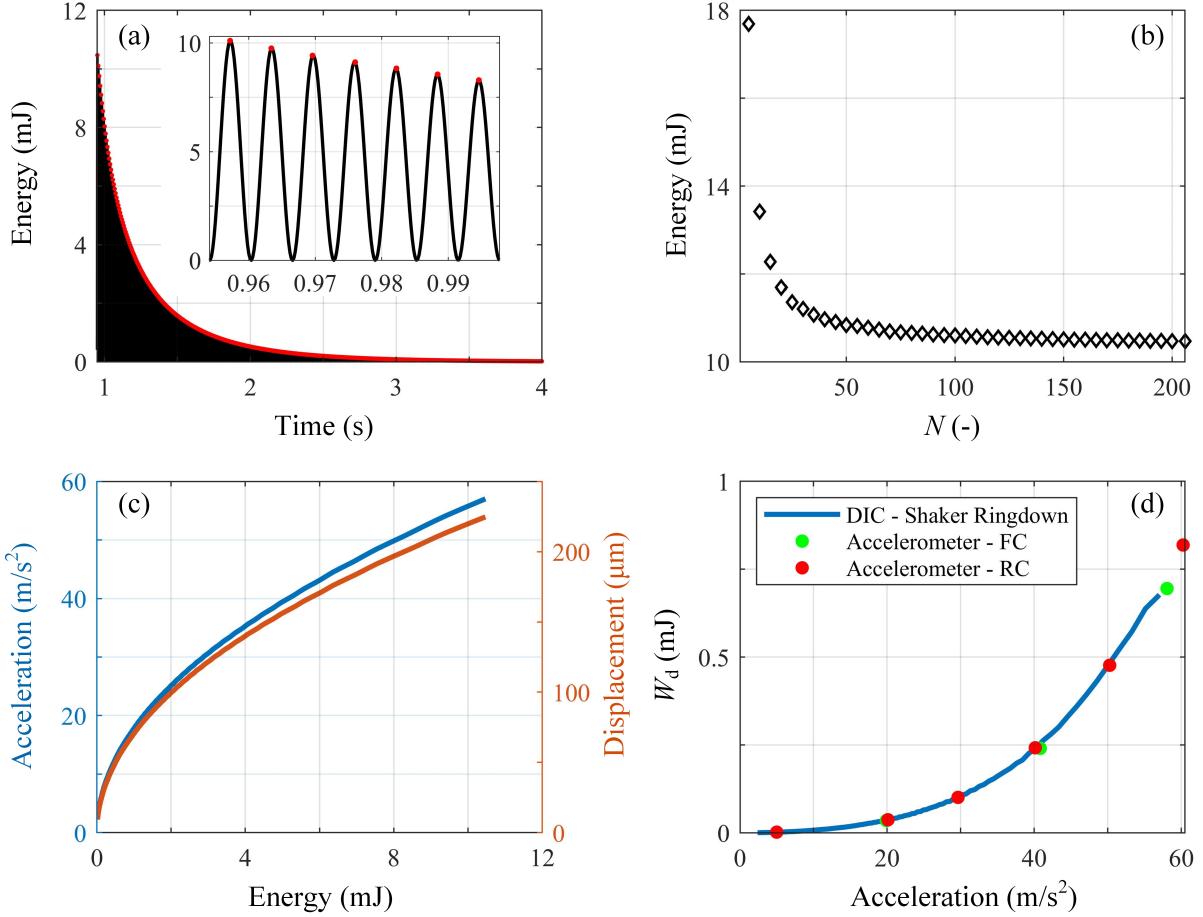


Figure 17: The energy of the HBRB where (a) shows the trend of the kinetic energy over time (with the inset showing the first seven peaks) and (b) the dependence of the energy on the number of the measurement points. Plot (c) shows the relation between energy and displacement/acceleration amplitudes at the location of Accel 3 in **Fig. 2**. The last plot in (d) shows the dissipated energy over one oscillation cycle computed via accelerometer measurements from forced testing (red and green dots, cf. **Sect. 3.2**) and the same quantity extracted using full-field data from shaker ringdown.

can be neglected [46, 47, 61]. Experimentally, the construction of frequency-energy plots is not practical since measuring the energy of a structure accurately is impractical with accelerometers, and thus the frequency-amplitude (typically acceleration amplitude) backbone plots are often reported instead, which is a flawed metric [19, 20]. In this study, however, the calculation of energy is feasible due to the full-field information from DIC.

Figure 18 shows the nonlinear characteristics of the system using the full-field DIC data. Here, all of the 206 DIC points are considered to ensure a sufficiently converged result, since the use of $N = 206$ instead of a smaller value, such as 100, comes at no significant computational cost increase. As an alternative to the displacement or acceleration amplitude used previously (see **Figs. 8** or **11**), the damping and frequency are now combined with the energy calculations, as shown in **Fig. 18** (a)-(d). Specifically, **Fig. 18** (c) and (d)

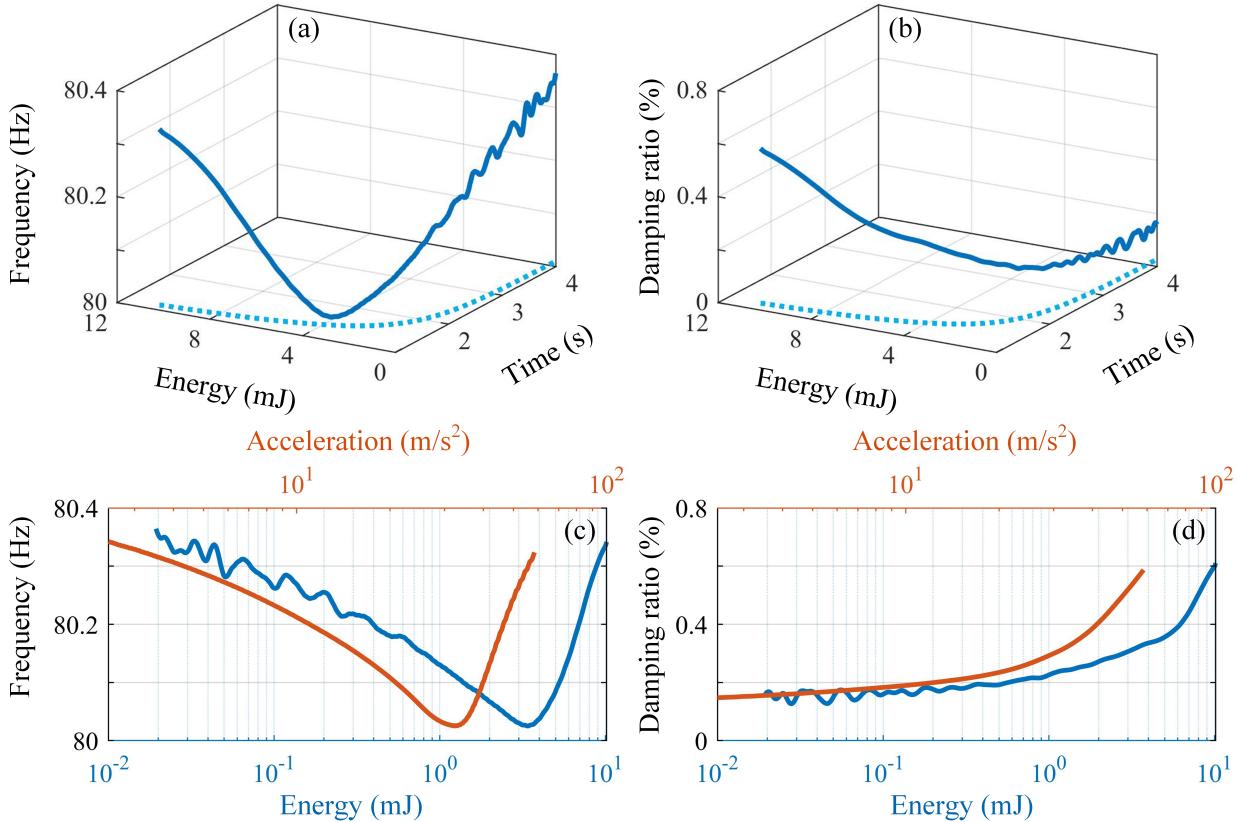


Figure 18: The relationship between the time, energy, amplitude, and nonlinear characteristics of the HBRB, including (a) relationship between time, energy, and frequency, (b) relationship between time, energy, and damping ratio, (c) energy-dependent and amplitude-dependent frequencies, (d) energy-dependent and amplitude-dependent damping ratios. The latter amplitudes refers to the acceleration amplitudes measured by Accel 3 in **Fig. 2**.

superimpose amplitude-dependence and energy-dependence. Clearly, the two properties show consistent values of frequency and damping whose amplitudes are related by a scaling factor and a quadratic relationship. Amplitude-dependent property plots act as proxies for the energy-dependent property plots in this case, but the latter may provide a more global perspective on the system under analysis. However, the energy-dependent plots in **Fig. 18** (c) and (d) show an increased variability towards small amplitudes, due to the degradation of the signal-to-noise ratio in DIC measurements. By contrast, amplitude-dependent curve shows a smoother behavior since the accelerometer data, at the same amplitudes, is not subject to these noise issues. It is important to note that if the system was not vibrating synchronously, the amplitude dependent plots may have a less clear relationship with the energy plots depending on which point is used to measure the amplitude.

674 **4. Comparison of the Methods from a Practitioner Viewpoint**

675 Nonlinear system identification, as shown in this paper, requires a toolbox approach
676 in which different techniques have to be deployed depending on the system under analysis
677 and the testing procedure. Force-response testing tends to provide robust information since
678 steady states can be properly isolated, but the shaker unavoidably influences the system
679 dynamics. Hammer excitation is easy to perform, but it generates a multi-modal component.
680 This is less than ideal if the identification targets a particular mode of the structure, as much
681 of the energy is distributed across modes that are not of interest. In this context, shaker
682 ringdown should be the preferred test method as it combines the advantages from both
683 shaker testing (i.e., single frequency and high level of excitation at that frequency) and
684 hammer excitation (i.e., transient free decay data). However, decoupling the shaker from
685 the structure is often not a straightforward task [48].

686 Excluding SS-SSM, all of the methods used in this paper are signal processing techniques
687 suited for time-frequency analyses. The extraction of frequency patterns in a time series
688 varies according to the theoretical foundations of the method, but a common requirement of
689 time-frequency methods is the need to filter the transient data before extracting the nonlinear
690 characteristics. By contrast, for SS-SSM, the frequency-amplitude relation is intrinsic in the
691 identified nonlinear model of the system. Encouragingly, all of the techniques showed good
692 agreement in the results for shaker ringdown and hammer testing.

693 Among all of the methods, DMD is the only one that was originally born for the analysis
694 of multi-point measurements of the dynamical system. The other techniques are more suited
695 for single-point measurements, and hence required suitable rearrangements in order to be
696 applied to multiple measurements simultaneously. Moreover, the HT, PFF, SS-SSM, and
697 WBEMD methods have been developed and extensively used for studying transient regimes
698 in mechanical systems. By contrast, DMD, broadly known in the fluid dynamics commu-
699 nity for unfolding the dynamics of ergodic attractors, has presented challenges for tackling
700 transient phenomena [68]. However, the results of DMD show a very good agreement with
701 the other techniques in the context of this research.

702 Every technique deployed in this paper presents either computational or implementation
703 challenges. From the complexity viewpoint, all of the techniques have drawbacks. The HT
704 and PFF methods require filtering that needs to be manually tuned for the best results [58]
705 in order to retain the core harmonic information of the time series [25]. Moreover, the HT
706 method suffers from the issue of end effect in the estimation of the damping curve, even
707 though appropriate signal pre-processing is performed, cf. **Fig. 4** (b) and see [25]. For the
708 SS-SSM method, model fitting is fundamental for obtaining adequate results [34], which
709 requires a computational routine to determine the coefficients that describe the SSM and its
710 dynamics [31]. For DMD, the results depend on the selection of time scales (e.g., sampling,
711 windowing) and of the number of dynamic modes. Finally, WBEMD needs an optimization
712 in order to identify the intrinsic mode functions, and hence it requires some application
713 dependent parameters [57]. Overall, DMD, HT, and PFF may seem to be less complex
714 in terms of implementation for the context of the paper. However, SS-SSM may also be

715 exploited to construct data-driven reduced models, and WBEMD can efficiently uncover
716 multi-modal transient phenomena.

717 5. Conclusions

718 This paper compared the performance of multiple nonlinear identification methods based
719 on the experimental data from a jointed structure. The backbone curve and amplitude-
720 damping curve were extracted from the signals using Hilbert Transform (HT), Peak Finding
721 and Fitting (PFF), Dynamic Mode Decomposition (DMD), State-Space Spectral Submani-
722 fold (SS-SSM), and Wavelet-Bounded Empirical Mode Decomposition (WBEMD). The re-
723 spondes were measured by accelerometers in the experiments including hammer testing,
724 shaker ringdown testing, and response/force-control stepped sine testing. During the shaker
725 ringdown testing, apart from the accelerometers, two high-speed cameras recorded the im-
726 ages of the jointed beam. Then, Digital Image Correlation (DIC) was adopted to extract
727 the displacement responses from the images to reconstruct the mode shapes of the whole
728 beam. The main conclusions of this paper are summarized as follows:

- 729 • Backbone and damping curves obtained from hammer testing and shaker ringdown
730 testing showed a similar trend. However, the amplitude range in the shaker ringdown
731 testing was wider than that in the hammer testing despite the peak amplitude of the
732 acceleration of the tip of the beam being smaller in the shaker tests. This happens
733 because the hammer testing excites multiple modes, while the shaker ringdown testing
734 is able to excite one single mode of the structure. Especially at high amplitudes,
735 the amplitude-dependent properties obtained via hammer testing deviated from those
736 obtained via shaker ringdown testing, although proper filtering was applied. This
737 observation proves that other modes can indeed influence the amplitude-dependent
738 properties of a specific mode. Therefore, hammer testing should be carefully handled
739 when aiming to extract single-mode nonlinear information.
- 740 • Amplitude-dependent properties computed from shaker ringdown testing and from
741 forced-response testing presented good agreement, as expected from theoretical
742 results. In this paper, an energy criterion was used to extract backbone and damping
743 curves from response/forced-control testing. Although it requires precise tuning and an
744 effective decoupling mechanism, shaker ringdown testing is more efficient with respect
745 to forced testing, where the acquisition of dense datasets can necessitate extensive
746 experimental campaigns.
- 747 • Other than amplitude-dependent properties, which show good agreement with those
748 coming from accelerometer data, full-field data from two high-speed cameras are able
749 to reconstruct the mode shape of the beam. The deformation induced by the lap joint
750 is visible and the variation of the mode shape with the amplitude of motion can be
751 assessed as well as motion synchronism. Moreover, full-field data allows the definition
752 of the kinetic energy of the beam, which is able to determine amplitude-dependent

753 properties. The value of the kinetic energy also showed good convergence properties
754 as the number of measurement points was increased. Moreover, the dissipated energy
755 computed from the decay of kinetic energy during a shaker ringdown test is consistent
756 with the dissipated energy computed in forced testing. The metric of energy, which
757 cannot be achieved with traditional accelerometers alone due to the need for a spatially
758 dense data set for adequate convergence, can be useful for practices such as model
759 validation and model updating. However, especially when dealing with long structures
760 in free vibrations, the cameras are limited in memory, and thus they are only able to
761 record a short period of time.

- 762 • All of the methods used in the paper agreed well in the extraction of amplitude-
763 dependent properties from free vibrations. Other than showing performances, this
764 paper also critically compared these methods, pointing out benefits and potential
765 challenges in order to enrich the toolbox of practitioners in structural dynamics.
- 766 • In terms of the frequency-amplitude and damping-amplitude curves, the beam un-
767 der investigation showed limited frequency change, i.e., 0.5 Hz, while the damping
768 ratio rose from about 0.1% at low amplitude to more than 0.6% at high amplitudes.
769 Additionally, the mode shape presented substantial amplitude dependence, especially
770 in the area of the contact patch. The deflection ratio, which tries to quantify this
771 variation, exhibited a similar trend to that of the backbone curve, but with a much
772 wider range amounting to 7%. Thus, the strongest nonlinearities for the jointed beam
773 studied were observed in the damping behavior and the variation of the mode shape.
774 Further studies can help to uncover the connection of these macroscopic phenomena
775 with interactions occurring in the contact patch.
- 776 • The relationship between the energy and nonlinear frequency and damping ratio of
777 the beam have been obtained due to the full-field DIC data. The energy-dependent
778 characteristics provide the information of the whole beam instead of a few points
779 on the structure. Accelerometers, as a traditional way of measurement, make the
780 measurement of the beam's energy prohibitively difficult since it is not practical to
781 place a large number (i.e., 100s) of accelerometers on this specific structure, let alone
782 the effect of the added mass. By contrast, high-speed cameras provide a convenient way
783 of measuring the structure without contact. Additionally, responses of the structure
784 can be obtained with a high spatial resolution, making it possible to accomplish the
785 identification of the whole structure.

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794 **Data Access**

795 The data used in the paper is available at [https://github.com/mattiacenedese/](https://github.com/mattiacenedese/BRBtesting)
796 [BRBtesting](#).

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