

Shop Visit History Report

	IIN: Z3312	2		Serial N	umber:	161824492	3700			Part Number:	3822341-5		
Item No	Part No	Part TSN	Part CSN	Part TSO	Part CSO	Part TSR	Part CSR	FitmenV Removal	Date	Parent Serial Number	Parent TSN	Parent CSN	Visit Remarks
	3822341-5	0	0	0	0	0	0	FITMENT	17-0ct-2017	P2273	7184	4819	NEW-ARC 9001544725-80005

^{***} End Of Report***



Rev.No.:

Shop Visit History Report

IIN: 20000

Serial Number: P2273

Part Number :

3800550-1

Item No	Part No	Part TSN	Part CSN	Part TSO	Part CSO	Fitment/ Removal	Date	A/C Regn	Position	TSN	CSN	Visit No.	Visit Remarks
1	3800550-1	0	0	0	0	FITMENT	28-Jun-2013	swu	APU	0	0	20131009001	
2	3800550-1	7184	4819	7184	4819	REMOVAL	16-Sep-2017	swu	APU	19991	2377	20171101001	OVERHAUL
3	3800550-1	7184	4819	0	0	FITMENT	17-0ct-2017	-		0	0	20171101001	OVERHAULED BY HONEYWELL W/O 5011527375
4	3800550-1	7186	4825	2	6	REMOVAL	17-0ct-2017	-		2	6	20171101002	DMM HRS/CYCS RELEASED AFTER APU FINAL TESTING
5	3800550-1	7186	4825	2	6	FITMENT	15-Nov-2017	SNB	APU	11340	1341	20171101002	-

*** End Of Report***



Approving Civil Aviation Authority/Country: A A / United States	^{2.} AUTH	3. Form Tra-cking Number: 20170003733110Y14 9001544725-80005				
FAA/United States	FAA Form	8130-3, AIRWORTHINES	S APPROVAL TA	G		
4. Organization Name and Address	: Honeywell International Inc. 111 S. 34th Street Phoenix, Arizona 85072	Production Approval PT1222NM	Honeywell Internat Units 2-4, Chevron Hemel Hempstead UNITED KINGDON	, Eaton Road , HP2 7UB	5. Work Order/ContracVInvoid 4705219014-000010 Page 1 of ;1	
6.ltem: 17. Description:		8. Part Number:	9. Quantity:	10. Serial Numb	per:	11. Status/
001 IMPELLER, COMPRESS	OR, SECOND STAGE	3822341-5		161824492370	00	Work: NEW
13a. Certifies the items identified abo	ove were manufactured in conform	mity to:		Other re	egulation specified	in Block 12

13c. Approval/Authorization No.:

ODA-602216-NM

05/0CT/2017

13e. Date (dd/mmm/yyyy):

User/ Installer Responsibilities

14b. Authorized Signature:

14d. Name rinted):

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where tre user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of the country specified in 81 ock 1, it is essential that the user/installer ensures that his/her airworthiness authority of

14c. Approval/CertificateNo.:

14e. Date():

13d. Name (Typed or Printed):

Aggrey Kanyanga

Honeywell

LIFE LIMITED PART LOG

..._,,,.

ASSEMBLY NAME:	PART NUMBER:							
LIFE LIMITED PART NAME: IMPELLER, COMPRESSOR SECOND ST AGE	PART NUMBER: 3822341-5 SERIAL NUMBER: 1618244923700							
MANUFACTURED PER FAR PART 2I UNDER PRODUCTION CERTIFICATE PC 413 DATE INSTALLED REMOVED ENGINE SERIAL NUMBER	SIGNATURE OR ACCEPTANCE STAMP: TIME ON PART THIS INSTALLATION CYCLES HOURS	: <u>O.wic</u> l Bag_nali TOTAL TIME ON PART CYCLES HOURS	SIGNATURE/FAA NUMBER					
? .;;J_:;;)7':5	0 0.0	0 0.0	Colof x (6)					

GTCP 331--500[8] PERFORMANCE DATA SHEET

	0001		MANCE DATA	OHIEE!	
NOTIFICATION NUMBER	330005518			SERVICE ORDER	<u> </u>
APUMODEL <u>G</u>	TCP331-500[!3j	PART NUMBER	: <u>380 055</u> 0-1	SERIAL NUMBER	: <u>P-2273</u>
OPERATION TEXT	LAB TEST			MANUAL	EM 49-26-57
OPERATION NO	0220			REVISION	20
QUANTITY		UNITS	NO LOAD	103°F ECS MODE	103°F MES MODE
BAROMETRIC PRESSURE		PSIA	14.59	14.59	I 14.59
FUEL INLET PRESSURE		PSIG	25.5	20.8	I 20.3
OIL PUfy1P DISCHARGE PRESSURE		PSIG	64.6	63.5	I 63.4
OIL PUMP DISCHARGE TEMPERATUR	E	OF	160	163	I 162
GEARBOX PRESSURE		PSIA	14.96	14.89	14.86
COMPRESSOR INLET TEMPERATURE	MEASURED	OF	96.5	99.2	98.5
APU INLET TEMP RATURE (T2) (ARIN	NC)	OF	101.5	103.7	101.8
TURBINE DISCHARGE	UPPER (EGT 1)	OF	717.4	1063.9	I 1087.6
TEMPERATURE (UNIT RAKES)	LOWER (EGT 2): OF	725.3	1076.7	1112.1
EXHAUST GAS TOTALTEMPERATURE	(LA <u>MEASURED</u>	<u>°E</u>	+	<u>1075.6</u>	1103.4
	CORRECTED	°F		1100.6	1097.3
ORIFICE INLET AIR PRESSURE		"Hg		75.6	86.5
ORIFICE INLET TEMPERATURE		OF		410.5	420.1
ORIFICE DELTA PRESSURE		"H20		59.92	24.48
IGV POSITION (IGVPOS)		DEGREES		12.5	5.0
IGV PERFORMANCE ADJ (IGVPERADJ	J)	DEGREES		5.0	
BLEED AIRFLOW	ACTUAL	LBS/MIN		474.86	320.04
	CORRECTED			470.8	299.2
RESTRICTED AIRF.LOW (DISC. CORR	FLOW)	LBS/MIN		166.88	106.02
BLEED AIR TOTAL PRESSURE	INDICATED	PSIA		54.5	58.2
	CORRECTED			54.2	54.3
BLEED AIR TOTAL TEMPERATURE	INDICATED	OF		420.7	430.9
SEEES AND TOTAL TERM ENGINE	CORRECTED			424.6	427.7
UNIT VIBRATION	ACCESSORY		0.23	0.14	0.15
CIVIT VIBRATION	TURBINE	IN/SEC	0.34	0.14	0.13
TURBINE WHEEL SPEED	TORBINE	RPM	39041	39041	
	APPLIED		///////////////////////////////////////		39038
SHAFT LOAD		SHP		179.3	179.4
FUEL CONSUMPTION	CORRECTED			180.7	180.7
I UEL CONSUIVIPTION	INDICATED	LBS/HR	336.8	626	653
	CORR -CTED	LBS/HR		628.00	627.24
TEST TECHNICIAN	HONEYWELL (S) TEST 09		QUALITY CONTRO	DL	COC PO
DATE	Stamp &Sig			•	Stamp & Sign
DATE	16-0ct-201	7	DATE	1	<u>Z OCT 2017</u>

FORM 331-500B-T-04 R19 (DDMMYY 131016)



TURBINE ACCEPTANCE TAG ,AJIIQ_ TRACEABILITY INPUT LIFE LIMITED PARTS RECORD/TRACE INPUT_PAGE

TSN: 7184:03

PART NO: 3800550-1 **SERIAL NO:** P-2273

CUSTOMER: SINGAPORE AIRLINES LTD

MODEL: GTCP331-5008

SERIES: 62 **DATE:** 17-Oct-17

S/0 NO: 5011527375

CSN: 4819 CSR: NA

.-NOTIFICATION NO: 330005518

TSR: NA CSR: NA
TSO: 0:00 CSO: 0

DESCRIPTION	PART NO	SERIAL NO	LOT NO	TSN	CSN	LIFE LIMITS /CYCLES\	REMARKS
D/COMP IMPELLER	3822612-1	13-182449-08I 88		7184:03	4819	27000	OVERHAULED
IST STG IMPELLER	3822611-1	13-182449-08502		7184:03	4819	27000	OVERHAULED
2ND STG IMPELLER	3822341-5	1618244923700		00	00	27000	NEW
I ST T-WHEEL ASSY	3842151-3	12-168522-01537					OVERHAULED
DISKPART NO	3842152-1	12-100322-01337	-	7184:03	4819	27000	
2ND T-WHEEL ASSY	3842155-4	12-168522-01912					OVERHAULED
DISK PART NO	3842156-1	12-100022-01912	•	7184:03	4819	27000	
3RD T-WHEEL ASSY	3842160-6	12-156101-09303	13(#1	7184:03	4819	27000	OVERHAULED

APU ACCESSORIES RECORD/NOC

DESCRIPTION	PART NO.	SERIAL NO.	<u>SERJES</u>	CHG NOS	REMARKS
STARTERCTRL VLV	3283076-5	1578	1	-	REPAIRED
AIR TURB STARTER	3505814-3	1541	1	-	OVERHAULED UNIT TRANSFERRED FROM SINGAPORE AIRLINES LTD APU S/N: P-2290
STARTER MOTOR	2704442-5	1234	2	-	OVERHAULED
FUEL CLUSTER	3879008-1	CUA11725	-	•	VISUAL INSPECTED/ TESTED
IGV ACTUATOR	3883499-3	1552	-	-	VISUAL INSPECTED/ TESTED
PNEU CLUSTER	3884863-7	P-1525	1	-	REPAIRED
CHECK VALVE	3202854-1	1401	1	-	REPAIRED
SURGE CTRL VALVE	3290814-5	3290814/888	1	-	OVERHAULED
LUBE CLUSTER	4131000-6	1668	2	-	REPAIRED
AIR OIL COOLER	160488-2	1664	2	-	CLEANED & FLUSHED/ VISUAL INSPECTED/ TESTED
TEMP CTRL VALVE	160536-1	1508	2	-	VISUAL INSPECTED / TESTED
IGNITION UNIT	3876195-9	J17340052	-	-	NEW
IGNITION SYS ASSY	3888275-10	-		-	REPAIRED (ALTERNATE OUTGOING S/N:J17340052)
G/BOXASSY	3805034-11	P-1401	-	-	MODIFIED/ REPAIRED
DRIVEN COMP	3804011-10	P-1398	-	-	OVERHAULED
COMP MODULE	3826980-13	P-1394	-	-	OVERHAULED
TURBMODULE	3844517-7	P-1393	-	-	OVERHAULED
OIL LEVEL SENSOR	3876211-3	-		-	VISUAL INSPECTED/ TESTED (ALTERNATE OUI GOING S/N: 97424-DT1471)
DMM	304643-2	GE12119	1		VISUAL INSPECTED

REMARKS: APU HOURS AND CYCLES ARE BASED AT THE POINT OF ASSEMBLY AND OBTAINED FROM THE DMM.

INSPECTOR SIGNATURE & STAMP:

FORM: GTCP331-5008-G16-R.7 (ddmmyy 290615)

Rev.No.:

Available hours/cycles for APU life limited parts

APU Description	AUXILIARY POWER UNIT				
Part Number	3800550-1	Serial number	P2273	IIN	Z0000
Aircraft Reg		Position		Date Installed	17-0ct-2017
TSN / CSN	7184/4819	TSLV / CSLV	0/0		
Remarks	OVERHAULED BY HONEYV	VELL W/O 5011527375			

Item No (ATA Chapter	Position	IIN	Part Number	Part Description	Serial Number	TSO# LL(HR)	CSO# LL(CYC)	TSN	CSN	Avai Hours	lable Cycles
	49-00-00		Z0000	3800550-1	AUXILIARY POWER UNIT	P2273	0	0	7184	4819	NA	NA
2	49-23-00		Z3311	3822611-1	IMPELLER COMPRESSOR 1ST STAGE	13-182449-08502		27000 •	7184	4819		22181
3	49-23-00		Z3312	3822341-5	IMPELLER, COMPRESSOR 2ND STG	1618244923700		27000 •	0	- 0		27000
4	49-23-00		Z3313	3822612-1	IMPELLER COMPRESSOR LOAD	13-182449-08188		27000 •	7184	4819		22181
5	49-25-00		Z3511	3842151-3	ROTOR ASSEMBLY, 1ST STAGE	12-168522-01537		21000-	7184	4819		22181
6	49-25-00		Z3512	3842155-4	ROTOR ASSEMBLY, 2ND STAGE	12-168522-01912		27000•	7184	4819		22181
7	49-25-00		Z3513	3842160-6	ROTOR ASSEMBLY, 3RD STAGE	12-156101-09303		27000 •	7184	4819		22181

*** End Of Report***



Rev. No:

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APU shop visit on log report

	Number	3800550-1	Serial Numb	er P2273	Desc	cription	AUXILIA	RY POWER UN	IIT	
	/ CSN raft Regn	7184 / 4819	TSLV / CSLV Owner	- <i>I</i> - SIA	TSC)/CSO	0/0			
Date	e fitted Removed	17-0ct-2017	Visit Number Visit Reason	20171	101001 Job HAULED BY HONEYWEL	Number L W/O 5011527	HW-501 ² 7375	1527375		
Item No	IIN	ATA Chapter	Part Description Position	Part No	Serial No	TSN/ CSN	TSO / CSO	(SL) Availab Hrs/ Cyc	le (HL) Hrs/ Cyc	Remarks
	23311	49-23-00	IMPELLER COMPRESSOR 1ST STAGE	3822611-1	13-182449-08502	7184 / 4819	01 0		/ 22181	
2	233j2	49-23-00	IMPELLER, COMPRESSOR 2ND STG	. 3822341-5	1618244923700	<u>61</u>	8 <u>1</u>		27000	
3	23313	49-23-00	IMPELLER COMPRESSOR LOAD	3822612-1	13-182449-08188	7184 / 4819	<i>01</i> 0		22181	
4	23511	49-25-C0	ROTOR ASSEMBLY, 1ST STAGE	3842151-3	12-168522-01537	7184 / 4819	<i>01</i> 0		22181	
5	23512	49-25-C0	ROTOR ASSEMBLY, 2ND STAGE	3842155-4	12-168522-01912	7184 / 4819	0/ 0		. <i>l</i> 22181	
6	23513	49-25-C0	ROTOR ASSEMBLY, 3RD STAGE	3842160-6	12-156101-09303	7184 / 4819	0/ 0		22181	
(*)- F	Revised Life L	imit								

*** End Of Report***



Rev.No.:

Available hours/cycles for APU life limited parts

APU Description	AUXILIARY POWER UN	Т			
Part Number	3800550-1	Serial number	P2273	IIN	Z0000
Aircraft Reg	SNB	Position	APU	Date Installed	15-Nov-2017
TSN / CSN	10878 / 7124	TSLV / CSLV	3694 / 2305		
Remarks	AT REMOVAL DATED 23	3 NO√ 2019			

Item №	ATA Chapter	Position	IIN	Part Number	Part Description	Serial Number	TSO# LL(HR)	CSO# LL(CYC)	TSN	CSN	Available Hours Cycles
	49-00-00	APU	20000	3800550-1	AUXILIARY POWER UNIT	P2273	3694	2305	10878	7124	NA NA 1
2	49-23-00		23311	3822611-1	IMPELLER COMPRESSOR 1ST STAGE	 13-182449-08502		27000'	10878	7124	19876
3	49-23-00		23312	3822341-5	IMPELLER, COMPRESSOR 2ND STG	1618244923700		27000'	3694	2305	24695
4	49-23-00		23313	3822612-1	IMPELLER COMPRESSOR LOAD	13-182449-08188		27000'	10878	7124	19876
5	49-25-00		23511	3842151-3	ROTOR ASSEMBLY, 1ST STAGE	12-168522-01537		27000.	10878	7124	19876
6	49-25-00		23512	3842155-4	ROTOR ASSEMBLY. 2ND STAGE	12-168522-01912		27000'	10878	7124	19876
7	49-25-00		23513	3842160-6	ROTOR ASSEMBLY. 3RD STAGE	12-156101-09303		27000'	10878	7124	19876





Note: #-TSO/CSO only applicable to Module and Non-Life Limited Parts.
- Life Calculation on Revised Life Limit - Life Calculation on Soft Life Limit - Life Calculation on Hard Life Limit