

Iowa State University
Aerospace Engineering

AER E 322 Lab 01

Practice Experiment and Data Analysis

Matthew Mehrrens, Peter Mikolitis, and Natsuki Oda

February 3, 2023

Aerospace Structures Laboratory Report
Lab 01 Practice Experiment and Data Analysis

Section 4 Group 2

Matthew Mehrrens, Peter Mikolitis, and Natsuki Oda

AER E 322

February 3, 2023

Spring 2023

Contents

1	Pre-Lab	2
1.1	Introduction	2
1.2	Objectives	2
1.3	Hypothesis	3
1.3.1	Test 1	3
1.3.2	Test 2	3
1.3.3	Test 3	3
2	Lab Work	4
2.1	Variables	4
2.1.1	Independent Variables	4
2.1.2	Dependent Variables	5
2.2	Work Assignments	5
2.3	Materials, Apparatus, and Procedures	5
2.4	Data	5
3	Conclusion	7
3.1	Analysis	7
3.2	Conclusion	7

Aerospace Structures Laboratory Report
Lab 01 Practice Experiment and Data Analysis

Section 4 Group 2

Matthew Mehrrens, Peter Mikolitis, and Natsuki Oda

AER E 322

February 3, 2023

Spring 2023

Chapter 1

Pre-Lab

1.1. Introduction

Aircraft wings undergo oscillations and other random forces while in flight. This lab replicates and analyzes some of the forces and oscillations a wing will experience in flight while also serving as an introduction to the PASCO tool kits and data processing. To simulate the wing, we used a cantilevered aluminum beam, and to generate and measure the oscillations, we used a PASCO tool kit—specifically the PASCO wave driver, displacement sensor and motion sensor. There were three rounds of testing; each additional round of testing introduced a new variable into the beam movement that changed the shape of the data. The data was collected using the PASCO tool kit and software provided. After the lab, we analyzed and processed the data in Python to how each variable effected the oscillation of the beam.

1.2. Objectives

During this lab, our primary objectives were to:

1. Learn how to record data under dynamic conditions and analyze or post-process the data.
2. Observe approximately how a common aerospace structural material might respond to oscillatory forces.

Aerospace Structures Laboratory Report
Lab 01 Practice Experiment and Data Analysis

Section 4 Group 2

Matthew Mehrrens, Peter Mikolitis, and Natsuki Oda

AER E 322

February 3, 2023

Spring 2023

3. Gain familiarity with the PASCO tool kit and the PASCO Capstone software.

1.3. Hypothesis

1.3.1. Test 1

We predict this test will provide the cleanest data of the three tests. Since the only force acting on the beam should be from the wave driver, we expect the displacement graph to show a uniform and steady wave—matching the oscillations of the wave driver. The data from this test should closely match the oscillations of an airplane wing in very steady flight.

1.3.2. Test 2

This test adds a spring-loaded weight to the cantilevered beam. Due to the oscillations of the spring loaded weight, we expect to see sudden highs and lows in the data corresponding with when the spring-loaded weight is in compression or tension respectively. The data from this test should demonstrate the oscillations of the wing in steady flight if there is an additional oscillatory or vibrational force simultaneously acting on the wing.

1.3.3. Test 3

This test is similar to Test 2 (see section [1.3.2](#)) except a third significant force has been introduced. Due to the addition of arbitrary impulses being applied by hand to the free end of the beam, we expect the data to show large peaks and dips in the data correlated with the timing of the impulses. The data from this test should demonstrate the oscillations of real flight as described in section [1.3.2](#) but also how the wing might react during periods of high turbulence where sudden, large impulses may act on the wing.

Chapter 2

Lab Work

2.1. Variables

2.1.1. Independent Variables

Wave Driver Frequency

The frequency of the wave generated by the PASCO wave driver.

Wave Driver Amplitude

The maximum voltage the wave driver will use when oscillating, proportional to the displacement of the wave driver arm.

Sampling Frequency

The time intervals at which the PASCO motion sensor or displacement sensor poll the position or displacement of the beam. Sensor with a higher sampling frequency will collect more data in the same amount of time.

Smoothing Span

The number of elements after a given element used to calculate a rolling mean while data processing. A small smoothing span will better preserve the shape of the data; whereas, a

Aerospace Structures Laboratory Report
Lab 01 Practice Experiment and Data Analysis

Section 4 Group 2

Matthew Mehrtens, Peter Mikolitis, and Natsuki Oda

AER E 322

February 3, 2023

Spring 2023

larger smoothing span will better reduce noise and outliers.

Time

Each test was run for exactly 15 s, measured in 0.01 s intervals.

2.1.2. Dependent Variables

Displacement/Position

The change in location of the cantilevered beam from its equilibrium position, measured closer to the free end along a perpendicular vertical axis.

Best Fit Curve

The line of best fit or fit curve is a normalized curve matching the raw or smoothed data and depends on the shape and magnitude of the displacement data.

2.2. Work Assignments

Refer to Table [2.1](#) for the distribution of work during this lab.

2.3. Materials, Apparatus, and Procedures

2.4. Data

Aerospace Structures Laboratory Report
Lab 01 Practice Experiment and Data Analysis

Section 4 Group 2

Matthew Mehrtens, Peter Mikolitis, and Natsuki Oda

AER E 322

February 3, 2023

Spring 2023

Task	Matthew	Peter	Natsuki
<i>Lab Work</i>			
Date Recording		X	
Exp. Setup	X		
Exp. Work			X
Exp. Clean-Up	X	X	X
<i>Data Processing</i>			
Data Import	X		
Smoothing	X	X	
Line of Best Fit	X	X	X
<i>Report</i>			
Introduction			X
Objectives		X	
Hypothesis	X		
Variables		X	
Materials		X	
Apparatus		X	
Procedures	X		
Data	X		
Analysis	X	X	X
Conclusion			X
References			X
Appendix			X
Revisions	X	X	X
Editing	X		

Table 2.1: Work assignments for lab 01.

Chapter 3

Conclusion

3.1. Analysis

3.2. Conclusion