Lesson 10

Timoshenko Beam

Minh-Chien Trinh, PhD

Division of Mechanical System Engineering Jeonbuk National University

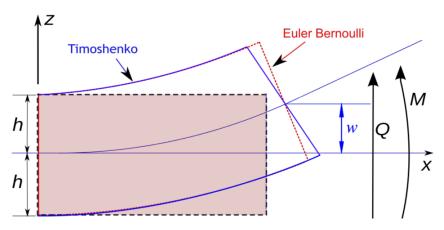
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Content

- 1. Timoshenko Beam
 - Static Analysis
 - Example 1 Python code
 - Free Vibrations
 - Example 2 Python code
 - Buckling Analysis
 - Example 3 Python code



The Timoshenko beam theory accounts for transverse shear deformation and rotational inertia effects. Thus, it is capable of modeling thin and thick beams.



- The deformed cross-section planes remain plane but not normal to the middle axis.
- \bullet The beam lays in the x-z plane, the displacement field is

$$u_1(x,z,t) = z\theta_y(x,t), \quad u_3(x,z,t) = w(x,t)$$

 u_1 , u_3 are the axial and transverse displacements of the 3D fibers of the beam.

w, θ_y denote the kinematic parameters of the theory as constant transverse displacement and rotation of cross-section plane about a normal to the middle axis x.

The normal and transverse shear strain are

$$\varepsilon_{x} = \frac{\partial u_{1}}{\partial x} = z \frac{\partial \theta_{y}}{\partial x}; \qquad \gamma_{xz} = \frac{\partial u_{1}}{\partial z} + \frac{\partial u_{3}}{\partial x} = \theta_{y} + \frac{\partial w}{\partial x}$$



The strain energy considers both bending and shear contributions,

$$U = \frac{1}{2} \int_{V} \sigma_{x} \varepsilon_{x} dV + \frac{1}{2} \int_{V} \tau_{xz} \gamma_{xz} dV = \frac{1}{2} \int_{V} E \varepsilon_{x}^{2} dV + \frac{1}{2} \int_{V} kG \gamma_{xz}^{2} dV = \frac{1}{2} \int_{-a}^{a} EI_{y} \left(\frac{\partial \theta_{y}}{\partial x} \right)^{2} dx + \frac{1}{2} \int_{-a}^{a} kGA \left(\frac{\partial w}{\partial x} + \theta_{y} \right)^{2} dx$$

 $au_{xz} = kG\gamma_{xz}$ is the transverse shear stress, k = 5/6 is the shear correction factor.

$$G = \frac{E}{2(1+v)}$$
 is the shear modulus.

t Each node of this 2-node element has 2 DOFs (i.e., w and θ_v)

$$\mathbf{u}^{eT} = \begin{bmatrix} w_1 & w_2 & \theta_{y1} & \theta_{y2} \end{bmatrix} = \begin{bmatrix} \mathbf{w}^{eT} & \mathbf{\theta}_y^{eT} \end{bmatrix} \iff \mathbf{u}^e = \begin{bmatrix} \mathbf{w}^e \\ \mathbf{\theta}_y^e \end{bmatrix}$$

In opposition to Bernoulli beam, here the interpolation of displacements is independent for both w and θ_v .

$$w = \mathbf{N}\mathbf{w}^e$$
; $\theta_y = \mathbf{N}\mathbf{\theta}_y^e$

with the shape functions
$$\mathbf{N} = \begin{bmatrix} \frac{1}{2}(1-\xi) & \frac{1}{2}(1+\xi) \end{bmatrix}$$
 in natural coordinates $\xi \in [-1,1]$

The strain energy then becomes

$$U = \frac{1}{2} \mathbf{\theta}_{y}^{eT} \int_{-a}^{a} EI_{y} \mathbf{N'}^{T} \mathbf{N'} dx \mathbf{\theta}_{y}^{e} + \frac{1}{2} \int_{-a}^{a} kGA \left(\mathbf{N'} \mathbf{w}^{e} + \mathbf{N} \mathbf{\theta}_{y}^{e} \right)^{T} \left(\mathbf{N'} \mathbf{w}^{e} + \mathbf{N} \mathbf{\theta}_{y}^{e} \right) dx \quad with \quad \mathbf{N'} = \frac{d\mathbf{N}}{dx}$$



❖ The coordinate transformation is applied to have the integrals in natural coordinates as

$$U = U_b + U_s = \frac{1}{2} \mathbf{\theta}_y^{eT} \int_{-1}^{1} \frac{EI_y}{a^2} \mathbf{N'}^T \mathbf{N'} a d\xi \mathbf{\theta}_y^e + \frac{1}{2} \int_{-1}^{1} kGA \left(\frac{1}{a} \mathbf{N'} \mathbf{w}^e + \mathbf{N} \mathbf{\theta}_y^e \right)^T \left(\frac{1}{a} \mathbf{N'} \mathbf{w}^e + \mathbf{N} \mathbf{\theta}_y^e \right) a d\xi$$

 $U_{\scriptscriptstyle h}$ is the bending part, $U_{\scriptscriptstyle s}$ is the shear part.

❖ The bending part can be directly computed, the shear part should be reordered.

$$U_{s} = \frac{1}{2} \int_{-1}^{1} kGA \left[\left[\frac{1}{a} \mathbf{N'} \quad \mathbf{N} \right] \left[\frac{\mathbf{w}^{e}}{\mathbf{\theta}_{y}^{e}} \right] \right]^{T} \left(\left[\frac{1}{a} \mathbf{N'} \quad \mathbf{N} \right] \left[\frac{\mathbf{w}^{e}}{\mathbf{\theta}_{y}^{e}} \right] \right) a d\xi$$

$$= \frac{1}{2} \int_{-1}^{1} kGA \left[\mathbf{w}^{eT} \quad \mathbf{\theta}_{y}^{eT} \right] \left[\frac{1}{a} \mathbf{N'}^{T} \right] \left[\frac{1}{a} \mathbf{N'} \quad \mathbf{N} \right] \left[\frac{\mathbf{w}^{e}}{\mathbf{\theta}_{y}^{e}} \right] a d\xi$$

$$= \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} kGA \left[\frac{1}{a} \mathbf{N'}^{T} \right] \left[\frac{1}{a} \mathbf{N'} \quad \mathbf{N} \right] a d\xi \mathbf{u}^{e}$$

$$= \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} kGA \left[\frac{1}{a^{2}} \mathbf{N'}^{T} \mathbf{N'} \quad \frac{1}{a} \mathbf{N'}^{T} \mathbf{N} \right] a d\xi \mathbf{u}^{e}$$

$$= \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} kGA \left[\frac{1}{a^{2}} \mathbf{N'}^{T} \mathbf{N'} \quad \mathbf{N}^{T} \mathbf{N} \right] a d\xi \mathbf{u}^{e}$$



Finally, the strain energy becomes

$$U = \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} \left(\frac{EI_{y}}{a} \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbf{N'}^{T} \mathbf{N'} \end{bmatrix} + \frac{kGA}{a} \begin{bmatrix} \mathbf{N'}^{T} \mathbf{N'} & a\mathbf{N'}^{T} \mathbf{N} \\ a\mathbf{N}^{T} \mathbf{N'} & a^{2} \mathbf{N}^{T} \mathbf{N} \end{bmatrix} \right) d\xi \mathbf{u}^{e}$$

❖ Thus, the stiffness matrix for a generic element (size 4 x 4) is

$$\mathbf{K}^{e} = \int_{-1}^{1} \left(\frac{EI_{y}}{a} \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbf{N'}^{T} \mathbf{N'} \end{bmatrix} + \frac{kGA}{a} \begin{bmatrix} \mathbf{N'}^{T} \mathbf{N'} & a\mathbf{N'}^{T} \mathbf{N} \\ a\mathbf{N}^{T} \mathbf{N'} & a^{2} \mathbf{N}^{T} \mathbf{N} \end{bmatrix} \right) d\xi$$

- ❖ Due to shear locking in thin beams, it is strongly recommended to use exact integration with 2 Gauss points for the bending stiffness and use reduce integration with 1 Gauss points for the shear stiffness.
- The external work is

$$W_{E} = \int_{-a}^{a} p w dx = \int_{-a}^{a} p \mathbf{N} \mathbf{w}^{e} dx = \mathbf{w}^{eT} \int_{-a}^{a} p \mathbf{N}^{T} dx = \mathbf{u}^{eT} \int_{-a}^{a} \begin{bmatrix} p \mathbf{N}^{T} \\ \mathbf{0} \end{bmatrix} dx = \mathbf{u}^{eT} \int_{-1}^{1} \begin{bmatrix} p \mathbf{N}^{T} \\ \mathbf{0} \end{bmatrix} a d\xi$$

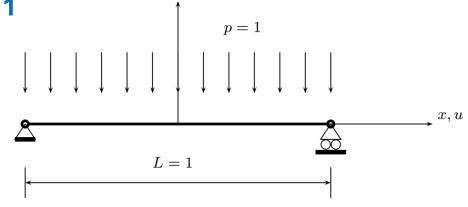
The force vector is given by

$$\mathbf{f}^e = \int_{-1}^1 \begin{bmatrix} p \, \mathbf{N}^T \\ \mathbf{0} \end{bmatrix} a \, d\xi$$



1. Timoshenko Beam - Example 1

A simply-supported (SS) Timoshenko beam, under uniform load p = 1, beam width b = 1, thus $I = h^3/12$. $E = 18^8$, v = 0.3



Exact solutions based on the first order shear deformation theory of SS Timoshenko beam:

$$w(x) = \frac{PL^4}{24D} \left(\frac{x}{L} - \frac{2x^3}{L^3} + \frac{x^4}{L^4} \right) + \frac{PL^2}{2S} \left(\frac{x}{L} - \frac{x^3}{L^3} \right)$$

with the shear stiffness S=kGA and the flexural stiffness $D=\frac{Eh^3}{12\left(1-v^2\right)}$ $w_{\rm max}=\frac{5}{284}\frac{pL^4}{EI}$

$$w(x) = \frac{PL^4}{24D} \left(6\frac{x}{L} - \frac{4x^3}{L^3} + \frac{x^4}{L^4} \right) + \frac{PL^2}{2S} \left(2\frac{x}{L} - \frac{x^2}{L^2} \right)$$

$$w_{\text{max}} = \frac{1}{8} \frac{pL^4}{EL}$$

1. Timoshenko Beam – Example 1 - Python

Python code: https://github.com/mctrinh/fem-class



1. Timoshenko Beam - Free Vibrations

The kinematic energy has two parts (i.e., translation and rotations related)

$$K = \frac{1}{2} \int_{-a}^{a} \rho A \dot{w}^{2} dx + \frac{1}{2} \int_{-a}^{a} \rho I_{y} \dot{\theta}_{y}^{2} dx$$

❖ After coordinate transformation and linear interpolation

$$K = \frac{1}{2}\dot{\mathbf{w}}^{eT} \int_{-1}^{1} \rho A \mathbf{N}^{T} \mathbf{N} \, ad\xi \dot{\mathbf{w}}^{e} + \frac{1}{2}\dot{\mathbf{\theta}}_{y}^{eT} \int_{-1}^{1} \rho I_{y} \mathbf{N}^{T} \mathbf{N} ad\xi \dot{\mathbf{\theta}}_{y}^{e}$$

Collecting the terms of the displacement vector leads to

$$K = \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} \left(\rho A a \begin{bmatrix} \mathbf{N}^{T} \mathbf{N} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} + \rho I_{y} a \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ \mathbf{0} & \mathbf{N}^{T} \mathbf{N} \end{bmatrix} \right) d\xi \mathbf{u}^{e}$$

The element mass matrix is

$$\mathbf{M}^{e} = \int_{-1}^{1} \begin{bmatrix} \rho A a \mathbf{N}^{T} \mathbf{N} & \mathbf{0} \\ \mathbf{0} & \rho I_{v} a \mathbf{N}^{T} \mathbf{N} \end{bmatrix} d\xi$$

The element stiffness matrix is like that in the static analysis.

1. Timoshenko Beam – Example 2

- **...** Consider a thin Timoshenko cantilever beam with (L = 1, h = 0.001).
- Results are presented in the nondimensional natural frequencies as

$$\overline{\omega} = \omega L^2 \sqrt{\frac{\rho A}{EI_y}}$$

Python codes: https://github.com/mctrinh/fem-class

1. Timoshenko Beam - Buckling Analysis

The work (energy) due to the applied compression load is

$$W_g = \frac{1}{2} \int_{-a}^{a} P\left(\frac{\partial w}{\partial x}\right)^2 dx = \frac{1}{2} \mathbf{w}^{eT} \int_{-1}^{1} \frac{P}{a^2} \mathbf{N}^{\prime T} \mathbf{N}^{\prime} a d\xi \mathbf{w}^{e}$$

The relation is written in terms of the displacement vector

$$W_g = \frac{1}{2} \mathbf{u}^{eT} \int_{-1}^{1} \frac{P}{a} \begin{bmatrix} \mathbf{N'}^T \mathbf{N'} & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} d\xi \mathbf{u}^e$$

The geometric stiffness matrix is

$$\mathbf{K}_g = \int_{-1}^{1} \frac{P}{a} \begin{bmatrix} \mathbf{N}'^T \mathbf{N}' & \mathbf{0} \\ \mathbf{0} & \mathbf{0} \end{bmatrix} d\xi$$

The buckling analysis of Timoshenko beams considers the solution of the eigen-problem

$$\left[\mathbf{K} - \lambda \mathbf{K}_{g}\right] \mathbf{X} = \mathbf{0}$$

 λ is the critical buckling load, X is the buckling modes.

The exact solution for simply-supported (SS) and clamped (CC) Timoshenko is

$$P_{cr} = \frac{\pi^2 EI}{L_{\it eff}^2} \Bigg[1 + \frac{\pi^2 EI}{L_{\it eff}^2 kGA} \Bigg]^{-1} \quad \textit{with} \quad \quad L_{\it eff} = L \quad \textit{for SS (pinned-pinned) beams,} \\ L_{\it eff} = L/2 \quad \textit{for CC (fixed-fixed) beams.}$$



1. Timoshenko Beam - Problem 3

Python code: https://github.com/mctrinh/fem-class



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