A Design Study Approach to Classical Control

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Homework B.13

The objective of this problem is to design an observer that estimates the state of the system and to use the estimated state in the controller designed in Homework B.12.

- (a) Modify the simulink diagram from HW B.12 as described in the description of HW A.13 to add a plot of the true state x, and estimated state \hat{x} , and the observation error $x \hat{x}$.
- (b) For the sake of understanding the function of the observer, for this problem we will use exact parameters, without an input disturbance. Modify pendulum_dynamics.m so that the parameters known to the controller are the actual plant parameters (uncertainty parameter $\alpha = 0$).
- (d) In the control block, add an observer to estimate the state \hat{x} , and use the estimate of the state in your feedback controller. Tune the poles of the controller and observer to obtain good performance.
- (e) As motivation for the next chapter, add an input disturbance to the system of 0.05 and observe that there is steady state error in the response even though there is an integrator. This is caused by a steady state error in the observation error. In the next chapter we will show how to remove the steady state error in the observation error.

Solution

Matlab code used to design the observer based controller is shown below:

```
1 % inverted Pendulum parameter file
 2 clear all
 4 % system parameters known to controller
 5 \text{ P.m1} = 0.25; \% \text{ kg}
 6 \text{ P.m2} = 1;
7 \text{ P.ell} = 0.5; \% \text{ m}
 8 \text{ P.b} = 0.05; \% \text{ N m}
9 P.g = 9.8; % m/s^2
11 % initial conditions
12 P.z0 = 0;
13 P.zdot0 = 0;
14 P.theta0 = 0;
15 P.thetadot0 = 0;
17 % input constraint
18 P.F_max = 5;
_{20} % sample rate for controller
P.Ts = 0.01;
23 % gain for dirty derivative
_{24} P.sigma = 0.05;
  % state space design
28 \text{ P.A} = [...]
       0, 0, 1, 0; ...
       0, 0, 0, 1; ...
       0, -P.m1*P.g/P.m2, -P.b/P.m2, 0;...
       0, (P.m1+P.m2)*P.g/P.m2/P.ell, P.b/P.m2/P.ell, 0;...
33 ];
34 P.B = [0; 0; 1/P.m2; -1/P.m2/P.ell];
35 P.C = [...
       1, 0, 0, 0; ...
       0, 1, 0, 0; ...
37
       ];
40 % form augmented system
```

```
41 Cout = [1,0,0,0];
42 A1 = [P.A, zeros(4,1); -Cout, 0];
43 B1 = [P.B; 0];
45 % tunning parameters
46 tr_z = 1.5; % rise time for position
47 tr_theta = .5; % rise time for angle
48 zeta_z = 0.707; % damping ratio position
49 zeta_th = 0.707; % damping ratio angle
integrator_pole = -10;
52 % compute gains
          = 2.2/tr_theta; % natural frequency for angle
53 wn_th
           = 2.2/tr_z; % natural frequency for position
54 WN Z
55 des_char_poly = conv(conv([1,2*zeta_z*wn_z,wn_z^2],...
                        [1,2*zeta_th*wn_th,wn_th^2]),...
56
                        poly(integrator_pole));
57
58 des_poles = roots(des_char_poly);
60 % is the system controllable?
if rank(ctrb(A1,B1))\neq 5,
      disp('System Not Controllable');
63 else % if so, compute gains
      K1 = place(A1,B1,des_poles);
      P.K = K1(1:4);
      P.ki = K1(5);
66
67 end
68
69 % observer design
volume{70} wn_th_obs = 10 \times volume{10},
71 wn z obs
              = 10*wn z;
72 des_obsv_char_poly = conv([1,2*zeta_z*wn_z_obs,wn_z_obs^2],...
                         [1,2*zeta_th*wn_th_obs,wn_th_obs^2]);
74 des_obsv_poles = roots(des_obsv_char_poly);
76 % is the system observable?
77 if rank(obsv(P.A,P.C))\neq 4,
      disp('System Not Observable');
79 else % if so, compute gains
      P.L = place(P.A', P.C', des_obsv_poles)';
81 end
```

Matlab code for the observer based control is shown below:

```
1 function out=pendulum_ctrl(in,P)
       z_r
              = in(1);
              = in(2);
       z_m
3
       theta_m = in(3);
              = in(4);
5
       % implement observer
7
       persistent xhat
                           % estimated state (for observer)
       persistent F
9
       if t<P.Ts,</pre>
10
           xhat = [0;0;0;0];
11
           F = 0;
12
       end
13
       N = 10;
14
       for i=1:N,
           xhat = xhat + ...
16
               P.Ts/N*(P.A*xhat+P.B*F...
17
                        +P.L*([z_m;theta_m]-P.C*xhat));
18
19
       end
       zhat = xhat(1)
20
21
       % integrator
22
       error = z_r - zhat;
       persistent integrator
24
25
       persistent error_d1
       % reset persistent variables at start of simulation
26
       if t<P.Ts==1,</pre>
           integrator = 0;
28
           error d1 = 0;
29
       end
30
       integrator = integrator + (P.Ts/2) * (error+error_d1);
31
       error_d1 = error;
32
33
       % compute the state feedback controller
34
       F_{unsat} = -P.K*xhat - P.ki*integrator;
35
       F = sat(F_unsat, P.F_max);
36
37
       % integrator anti-windup
       if P.ki\neq 0,
39
          integrator = integrator + P.Ts/P.ki*(F-F_unsat);
40
       end
41
43
       out = [F; xhat];
44
45 end
```

See the wiki for the complete solution.