

A Design Study Approach to Classical Control

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Homework C.13

The objective of this problem is to design an observer that estimates the state of the system and to use the estimated state in the controller designed in Homework [C.12](#).

- (a) Modify the simulink diagram from HW [C.12](#) as described in the description of HW [A.13](#) to add a plot of the true state x , and estimated state \hat{x} , and the observation error $x - \hat{x}$.
- (b) For the sake of understanding the function of the observer, for this problem we will use exact parameters, without an input disturbance. Modify `satellite_dynamics.m` so that the parameters known to the controller are the actual plant parameters (uncertainty parameter $\alpha = 0$).
- (d) In the control block, add an observer to estimate the state \hat{x} , and use the estimate of the state in your feedback controller. Tune the poles of the controller and observer to obtain good performance.
- (e) As motivation for the next chapter, add an input disturbance to the system of 1 and observe that there is steady state error in the observation error. In the next chapter we will show how to remove the steady state error in the observation error.

Solution

Matlab code used to design the observer based controller is shown below:

```
1 clear all
2
3 % initial conditions
4 P.theta0 = 0;
5 P.phi0 = 0;
6 P.thetadot0 = 0;
7 P.phidot0 = 0;
8
9 % system parameters known to controller
10 P.Js = 5; % kg m^2
11 P.Jp = 1; % kg m^2
12 P.k = 0.15; % N m
13 P.b = 0.05; % N m s
14
15 % maximum torque
16 P.taumax = 5; %Nm
17
18 % sample rate for controller
19 P.Ts = 0.01;
20
21 % gain for dirty derivative
22 P.sigma = 0.05;
23
24 % state space design
25 P.A = [...
26     0, 0, 1, 0;...
27     0, 0, 0, 1;...
28     -P.k/P.Js, P.k/P.Js, -P.b/P.Js, P.b/P.Js;...
29     P.k/P.Jp, -P.k/P.Jp, P.b/P.Jp, -P.b/P.Jp;...
30 ];
31 ];
32 P.B = [0; 0; 1/P.Js; 0];
33 P.C = [...
34     1, 0, 0, 0;...
35     0, 1, 0, 0;...
36 ];
37
38 % form augmented system
39 Cout = [0,1,0,0];
40 A1 = [P.A, zeros(4,1); -Cout, 0];
```

```

41 B1 = [P.B; 0];
42
43 % tuning parameters
44 wn_th = 0.6;
45 zeta_th = 0.707;
46 wn_phi = 1.1;
47 zeta_phi = 0.707;
48 integrator_pole = -1;
49
50 % gain selection
51 ol_char_poly = charpoly(P.A);
52 des_char_poly = conv(conv([1,2*zeta_th*wn_th,wn_th^2],...
53     [1,2*zeta_phi*wn_phi,wn_phi^2]),...
54     poly(integrator_pole));
55 des_poles = roots(des_char_poly);
56
57 % is the system controllable?
58 if rank(ctrb(A1,B1))≠5,
59     disp('System Not Controllable');
60 else % if so, compute gains
61     K1 = place(A1,B1,des_poles);
62     P.K = K1(1:4);
63     P.ki = K1(5);
64 end
65
66 % observer design
67 % pick observer poles
68 wn_th_obs = 10*wn_th;
69 wn_phi_obs = 10*wn_phi;
70 des_obsv_char_poly = conv(...
71     [1,2*zeta_phi*wn_phi_obs,wn_phi_obs^2],...
72     [1,2*zeta_th*wn_th_obs,wn_th_obs^2]);
73 des_obsv_poles = roots(des_obsv_char_poly);
74
75 % is the system observable?
76 if rank(observ(P.A,P.C))≠4,
77     disp('System Not Observable');
78 else % if so, compute gains
79     P.L = place(P.A', P.C', des_obsv_poles)';
80 end

```

Matlab code for the observer based control is shown below:

```

1 function out=satellite_ctrl(in,P)

```

```

2   phi_r      = in(1);
3   theta_m    = in(2);
4   phi_m      = in(3);
5   t          = in(4);
6
7   % implement observer
8   persistent xhat      % estimated state (for observer)
9   persistent tau
10  if t<P.Ts,
11      xhat = [0;0;0;0];
12      tau  = 0;
13  end
14  N = 10;
15  for i=1:N,
16      xhat = xhat + ...
17          P.Ts/N*(P.A*xhat+P.B*tau...
18              +P.L*([theta_m;phi_m]-P.C*xhat));
19  end
20  phihat = xhat(2);
21
22  % integrator
23  error = phi_r - phihat;
24  persistent integrator
25  persistent error_d1
26  % reset persistent variables at start of simulation
27  if t<P.Ts==1,
28      integrator = 0;
29      error_d1   = 0;
30  end
31  integrator = integrator + (P.Ts/2)*(error+error_d1);
32  error_d1 = error;
33
34  % compute the state feedback controller
35  tau_unsat = -P.K*xhat - P.ki*integrator;
36  tau = sat( tau_unsat, P.taumax);
37
38  % integrator anti-windup
39  if P.ki≠0,
40      integrator = integrator + P.Ts/P.ki*(tau-tau_unsat);
41  end
42
43  out = [tau; xhat];
44
45  end
46

```

```
47 %  
48 % saturation function  
49 function out = sat(in,limit)  
50     if in > limit, out = limit;  
51     elseif in < -limit, out = -limit;  
52     else out = in;  
53     end  
54 end
```

See the wiki for the complete solution.