

A Design Study Approach to Classical Control

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Homework B.14

- (a) Modify your solution from HW [B.13](#) so that the uncertainty parameter in `pendulum_dynamics.m` is $\alpha = 0.2$, representing 20% inaccuracy in the knowledge of the system parameters, and so that the input disturbance is 0.5. Also, add noise to the output channels z_m and θ_m with standard deviation of 0.001. Before adding the disturbance observer, run the simulation and note that the controller is not robust to the large input disturbance.
- (b) Add a disturbance observer to the controller, and verify that the steady state error in the estimator has been removed. Tune the system to get good response.

Solution

Matlab code used to design the observer based controller is shown below:

```
1 % inverted Pendulum parameter file
2 clear all
3
4 % system parameters known to controller
5 P.m1 = 0.25; % kg
6 P.m2 = 1;    % kg
7 P.e11 = 0.5; % m
8 P.b = 0.05; % N m
```

```

9  P.g = 9.8; % m/s^2
10
11 % initial conditions
12 P.z0 = 0;
13 P.zdot0 = 0;
14 P.theta0 = 0;
15 P.thetadot0 = 0;
16
17 % input constraint
18 P.F_max = 5;
19
20 % sample rate for controller
21 P.Ts = 0.01;
22
23 % gain for dirty derivative
24 P.sigma = 0.05;
25
26
27 % state space design
28 P.A = [...
29     0, 0, 1, 0;...
30     0, 0, 0, 1;...
31     0, -P.m1*P.g/P.m2, -P.b/P.m2, 0;...
32     0, (P.m1+P.m2)*P.g/P.m2/P.e11, P.b/P.m2/P.e11, 0;...
33 ];
34 P.B = [0; 0; 1/P.m2; -1/P.m2/P.e11];
35 P.C = [...
36     1, 0, 0, 0;...
37     0, 1, 0, 0;...
38 ];
39
40 % form augmented system
41 Cout = [1,0,0,0];
42 A1 = [P.A, zeros(4,1); -Cout, 0];
43 B1 = [P.B; 0];
44
45 % tuning parameters
46 tr_z = 1.5; % rise time for position
47 tr_theta = .5; % rise time for angle
48 zeta_z = 0.707; % damping ratio position
49 zeta_th = 0.707; % damping ratio angle
50 integrator_pole = -10;
51
52 % compute gains
53 wn_th = 2.2/tr_theta; % natural frequency for angle

```

```

54 wn_z      = 2.2/tr_z; % natural frequency for position
55 des_char_poly = conv(conv([1,2*zeta_z*wn_z,wn_z^2],...
56                          [1,2*zeta_th*wn_th,wn_th^2]),...
57                          poly(integrator_pole));
58 des_poles = roots(des_char_poly);
59
60 % is the system controllable?
61 if rank(ctrb(A1,B1))≠5,
62     disp('System Not Controllable');
63 else % if so, compute gains
64     K1 = place(A1,B1,des_poles);
65     P.K = K1(1:4);
66     P.ki = K1(5);
67 end
68
69 % observer design
70 % form augmented system for disturbance observer
71 A2 = [P.A, P.B; zeros(1,4), zeros(1,1)];
72 C2 = [P.C, zeros(2,1)];
73 % pick observer poles
74 wn_th_obs = 10*wn_th;
75 wn_z_obs = 10*wn_z;
76 des_obsv_char_poly = conv([1,2*zeta_z*wn_z_obs,wn_z_obs^2],...
77                           [1,2*zeta_th*wn_th_obs,wn_th_obs^2]);
78 des_obsv_poles = roots(des_obsv_char_poly);
79 dist_obsv_pole = -1;
80
81 % is the system observable?
82 if rank(observ(A2,C2))≠5,
83     disp('System Not Observable');
84 else % if so, compute gains
85     L2 = place(A2', C2', [des_obsv_poles;dist_obsv_pole]);
86     P.L = L2(1:4,:);
87     P.Ld = L2(5,:);
88 end

```

Matlab code for the observer based control is shown below:

```

1 function out=pendulum_ctrl(in,P)
2     z_r      = in(1);
3     z_m      = in(2);
4     theta_m = in(3);
5     t        = in(4);
6

```

```

7      % implement observer
8      persistent xhat      % estimated state (for observer)
9      persistent dhat      % estimate disturbance
10     persistent F          % delayed input (for observer)
11     if t<P.Ts,
12         xhat = [0;0;0;0];
13         dhat = 0;
14         F     = 0;
15     end
16     N = 10;
17     for i=1:N,
18         xhat = xhat + ...
19             P.Ts/N*(P.A*xhat+P.B*(F+dhat)...
20                 +P.L*([z_m;theta_m]-P.C*xhat));
21         dhat = dhat + P.Ts/N*P.Ld*([z_m;theta_m]-P.C*xhat);
22     end
23     zhat = xhat(1);
24
25
26     % integrator
27     error = z_r - zhat;
28     persistent integrator
29     persistent error_d1
30     % reset persistent variables at start of simulation
31     if t<P.Ts==1,
32         integrator = 0;
33         error_d1   = 0;
34     end
35     integrator = integrator + (P.Ts/2)*(error+error_d1);
36     error_d1 = error;
37
38     % compute the state feedback controller
39     F_unsat = -P.K*xhat - P.ki*integrator - dhat;
40     F = sat( F_unsat, P.F_max);
41
42     % integrator anti-windup
43     if P.ki≠0,
44         integrator = integrator + P.Ts/P.ki*(F-F_unsat);
45     end
46
47     out = [F; xhat];
48
49 end
50
51 %

```

```
52 % saturation function
53 function out = sat(in,limit)
54     if in > limit, out = limit;
55     elseif in < -limit, out = -limit;
56     else out = in;
57     end
58 end
```

See the wiki for the complete solution.