A Design Study Approach to Classical Control

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Updated: May 18, 2016

Homework A.11

The objective of this problem is to implement state feedback controller using the full state. Start with the simulation files developed in Homework A.10.

- (a) Select the closed loop poles as the roots of the equations $s^2 + 2\zeta\omega_n + \omega_n^2 = 0$ where ω_n , and ζ were found in Homework A.8.
- (b) Add the state space matrices A, B, C, D derived in Homework A.6 to your param file.
- (c) Verify that the state space system is controllable by checking that rank($\mathcal{C}_{A,B}$) = n.
- (d) Find the feedback gain K so that the eigenvalues of (A BK) are equal to desired closed loop poles. Find the reference gain k_r so that the DC-gain from θ_r to θ is equal to one. Note that $K = (k_p, k_d)$ where k_p and k_d are the proportional and derivative gains found in Homework A.8. Why?
- (e) Modify the control code to implement the state feedback controller. To construct the state $x = (\theta, \dot{\theta})^{\top}$ use a digital differentiator to estimate $\dot{\theta}$.

Solution

From HW A.6, the state space equations for the single link robot arm are given by

$$\dot{x} = \begin{pmatrix} 0 & 1.0000 \\ 0 & -0.667 \end{pmatrix} x + \begin{pmatrix} 0 \\ 66.667 \end{pmatrix} u$$
$$y = \begin{pmatrix} 1 & 0 \end{pmatrix} x.$$

Step 1. The controllability matrix is therefore

$$C_{A,B} = [B, AB] = \begin{pmatrix} 0 & 66.6667 \\ 66.6667 & -44.44440 \end{pmatrix}.$$

The determinant is $det(\mathcal{C}_{A,B}) = -4444 \neq 0$, therefore the system is controllable.

Step 2. The open loop characteristic polynomial is

$$\Delta_{ol}(s) = \det(sI - A) = \det\begin{pmatrix} s & -1\\ 0 & s + 0.667 \end{pmatrix} = s^2 + 0.667s,$$

which implies that

$$\mathbf{a}_A = (-0.667, 0)$$

$$\mathcal{A}_A = \begin{pmatrix} 1 & -0.667 \\ 0 & 1 \end{pmatrix}$$

Step 3. The desired closed loop polynomial is

$$\Delta_{cl}^d(s) = s^2 + 2\zeta\omega_n s + \omega_n^2 = s^2 + 6.3621s + 20.2445$$

which implies that

$$\alpha = (6.3621, 20.2445).$$

Step 4. The gains are therefore given as

$$K = (\boldsymbol{\alpha} - \mathbf{a}_A) \mathcal{A}_A^{-1} \mathcal{C}_{A,B}^{-1}$$

= (0.3037 0.0854)

$$k_r = \frac{-1}{C(A - BK)^{-1}B}$$
$$= 0.3037.$$

Alternatively, we could have used the following Matlab script

```
1 % state space model
_{2} A = [0, 1; 0, -3*P.b/P.m/(P.ell^{2})];
_{3} B = [0; 3/P.m/(P.ell^2)];
4 C = [1, 0];
6 % gains for pole locations
7 charpoly = [1,2*zeta*wn,wn^2];
8 des_poles = roots(charpoly);
10 % is the system controllable?
if rank (ctrb (A, B)) \neq 2,
12
       disp('System Not Controllable');
13 else
       P.K = place(A, B, des_poles);
14
       P.kr = -1/(C*inv(A-B*P.K)*B);
15
16 end
```

The Matlab code for the controller is given by

```
function tau=arm_ctrl(in,P)
       theta_c = in(1);
       theta = in(2);
3
               = in(3);
4
5
       % use a digital differentiator to find thetadot
       persistent thetadot
7
       persistent theta_d1
       % reset persistent variables at start of simulation
9
       if t<P.Ts,
                       = 0;
           thetadot
11
           theta_d1
                       = 0;
12
       end
13
       thetadot = (2*P.tau-P.Ts)/(2*P.tau+P.Ts)*thetadot...
14
           + 2/(2*P.tau+P.Ts)*(theta-theta_d1);
15
       theta_d1 = theta;
16
17
       % construct the state
18
       x = [theta; thetadot];
19
       % compute equilibrium torque tau_e
20
       tau_e = P.m*P.q*(P.el1/2)*cos(theta);
21
       % compute the state feedback controller
22
       tau_tilde = - P.K*x + P.kr*theta_c;
23
```

```
% compute total torque
tau = sat( tau_e + tau_tilde, P.tau_max);
e end
```