A Design Study Approach to Classical Control

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Homework E.12

- (a) Modify the state feedback solution developed in Homework E.11 to add an integrator with anti-windup to the feedback loop for z.
- (b) Add a constant input disturbance of 1 Newtons to the input of the plant and allow the plant parameters to vary up to 20%.
- (c) Tune the integrator pole (and other gains if necessary) to get good tracking performance.

Solution

Step 1. The original state space equations are

$$\dot{x} = \begin{pmatrix} 0 & 0 & 1.0000 & 0\\ 0 & 0 & 0 & 1.0000\\ 0 & -18.1923 & 0 & 0\\ -9.8000 & 0 & 0 & 0 \end{pmatrix} x + \begin{pmatrix} 0\\ 0\\ 2.6519\\ 0 \end{pmatrix} u$$

$$y_r = \begin{pmatrix} 0 & 1 & 0 & 0 \end{pmatrix} x,$$

therefore the augmented system is

$$A_{1} = \begin{pmatrix} A & \mathbf{0} \\ -C & \mathbf{0} \end{pmatrix} = \begin{pmatrix} 0 & 0 & 1.0000 & 0 & 0 \\ 0 & 0 & 0 & 1.0000 & 0 \\ 0 & -18.1923 & 0 & 0 & 0 & 0 \\ -9.8000 & 0 & 0 & 0 & 0 & 0 \\ 0 & -1.0000 & 0 & 0 & 0 & 0 \end{pmatrix}$$

$$B_{1} = \begin{pmatrix} B \\ \mathbf{0} \end{pmatrix} = \begin{pmatrix} 0 \\ 0 \\ 2.6519 \\ 0 \\ 0 \end{pmatrix}$$

Step 2. Following the design in HW E.11, we use the tuning parameters $t_{r_z} = 1.2$, $t_{r_{\theta}} = 0.5$, $\zeta_z = 0.707$, $\zeta_{\theta} = 0.707$. In addition, we will add an integrator pole at $p_I = -5$. The new controllability matrix

$$C_{A_1,B_1} = [B_1, A_1B_1, A_1^2B_1] = \begin{pmatrix} 0 & 2.6519 & 0 & 0 & 0\\ 0 & 0 & 0 & -25.9890 & 0\\ 2.6519 & 0 & 0 & 0 & 472.7979\\ 0 & 0 & -25.9890 & 0 & 0\\ 0 & 0 & 0 & 0 & 25.9890 \end{pmatrix}.$$

The determinant is $det(\mathcal{C}_{A_1,B_1}) = -1.2345e + 05 \neq 0$, therefore the system is controllable.

The open loop characteristic polynomial

$$\Delta_{ol}(s) = \det(sI - A_1) = \det\begin{pmatrix} s & 0 & -1.0000 & 0 & 0\\ 0 & s & 0 & -1.0000 & 0\\ 0 & 18.1923 & s & 0 & 0\\ 9.8000 & 0 & 0 & s & 0\\ 0 & 1.0000 & 0 & 0 & s \end{pmatrix}$$
$$= s^5 - 178.2842s,$$

which implies that

$$\mathbf{A}_{A_1} = \begin{pmatrix} 0 & 0 & 0 & -178.2842 & 0 \end{pmatrix}$$

$$\mathcal{A}_{A_1} = \begin{pmatrix} 1 & 0 & 0 & 0 & -178.2842 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 1 \end{pmatrix}.$$

The desired closed loop polynomial

$$\Delta_{cl}^{d}(s) = (s^{2} + 2\zeta_{z}\omega_{n_{z}}s + \omega_{n_{z}}^{2})(s^{2} + 2\zeta_{\theta}\omega_{n_{\theta}}s + \omega_{\theta_{z}}^{2})(s+5)$$

$$= s^{5} + 13.8139s^{4} + 82.9192s^{3} + 265.3469s^{2} + 420.5664s + 325.3556,$$

which implies that

$$\alpha = (13.8139, 82.9192, 265.3469, 420.5664, 325.3556).$$

The augmented gains are therefore given as

$$K_1 = (\boldsymbol{\alpha} - \mathbf{a}_{A_1}) \mathcal{A}_{A_1}^{-1} \mathcal{C}_{A_1, B_1}^{-1}$$

= $\begin{pmatrix} 31.2675 & -23.0425 & 5.2090 & -10.2100 & 12.5190 \end{pmatrix}$

Step 3. The feedback gains are therefore given by

$$K = K_1(1:2) = (31.2675 -23.0425 5.2090 -10.2100)$$

 $k_I = K_1(3) = 12.5190.$

Alternatively, we could have used the following Python script

```
State Space
12 # tuning parameters
                     # rise time for position
13 \text{ tr}_z = 1.2
t_{14} tr_theta = 0.5
                    # rise time for angle
15 zeta_z = 0.707 # damping ratio position
16 zeta_th = 0.707 # damping ratio angle
integrator_pole = -5.0
18
20 # State Space Equations
21 \# xdot = A*x + B*u
22 \# y = C \star x
A = \text{np.array}([[0.0, 0.0, 1.0, 0.0],
                  [0.0, 0.0, 0.0, 1.0],
                  [0.0, -P.g, 0.0, 0.0],
25
                  [-P.m1*P.g/((P.m2*P.length**2)/3.0+P.m1*(P.length/2.0)**2), 0.0, 0
26
27
B = np.array([[0.0],
                 [0.0],
29
                 [0.0],
30
                 [P.length / (P.m2 * P.length ** 2 / 3.0 + P.m1 * | P.length ** 2 / 4.
31
  C = np.array([[1.0, 0.0, 0.0, 0.0],
33
                 [0.0, 1.0, 0.0, 0.0]]
34
35
36 # form augmented system
37 A1 = np.matrix([[0.0, 0.0, 1.0, 0.0, 0.0],
                  [0.0, 0.0, 0.0, 1.0, 0.0],
38
                  [0.0, -P.g, 0.0, 0.0, 0.0],
39
                  [-P.m1*P.g/((P.m2*P.length**2)/3.0+P.m1*(P.length/2.0)**2), 0.0, 0
40
                  [-1.0, 0.0, 0.0, 0.0, 0.0]
41
42
43 B1 = np.matrix([[0.0],
                  [0.0],
44
45
                  [0.0],
46
                  [P.length/(P.m2*P.length**2/3.0+P.m1*P.length**2/4.0)],
                  [0.0]])
47
48
49 # gain calculation
50 wn_th = 2.2/tr_theta # natural frequency for angle
wn_z = 2.2/tr_z # natural frequency for position
52 des_char_poly = np.convolve(
      np.convolve([1, 2*zeta_z*wn_z, wn_z**2],
53
                   [1, 2*zeta_th*wn_th, wn_th**2]),
54
```

```
np.poly(integrator_pole))
56 des_poles = np.roots(des_char_poly)
57
58 # Compute the gains if the system is controllable
if np.linalg.matrix_rank(cnt.ctrb(A1, B1)) != 5:
      print("The system is not controllable")
60
61 else:
      K1 = cnt.acker(A1, B1, des_poles)
62
       K = np.matrix([K1.item(0), K1.item(1), K1.item(2), K1.item(3)])
63
       ki = K1.item(4)
64
65
66 print('K: ', K)
67 print('ki: ', ki)
```

Python code that implements the associated controller listed below.

```
1 import numpy as np
2 import ballbeamParam as P
3 import ballbeamParamHW12 as P12
5 class ballbeamController:
       def __init__(self):
           self.integrator = 0.0 # integrator
           self.error_d1 = 0.0 # error signal delayed by 1 sample
8
           self.K = P12.K # state feedback gain
           self.ki = P12.ki # Integral gain
10
           self.limit = P.Fmax # Maximum force
11
           self.Ts = P.Ts # sample rate of controller
12
13
       def update(self, z_r, x):
14
           z = x.item(0)
15
           # integrate error
16
           error = z_r - z
17
           self.integrateError(error)
           # Construct the linearized state
19
           x_{tilde} = x - np.array([[P.ze], [0], [0], [0]])
^{20}
           zr\_tilde = z\_r - P.ze
21
           # equilibrium force
23
           F_e = P.m1*P.q*P.ze/P.length + P.m2*P.q/2.0
^{24}
           # Compute the state feedback controller
^{25}
26
           F_tilde = -self.K @ x_tilde - self.ki*self.integrator
           F_unsat = F_e + F_tilde
27
28
           F = self.saturate(F_unsat)
```

```
self.integratorAntiWindup(F, F_unsat)
29
           return F
30
31
       def differentiateZ(self, z):
^{32}
           self.z_dot = self.beta*self.z_dot + (1-self.beta)*((z + self.z_d1) / self.z_d1)
33
           self.z_d1 = z
34
35
       def differentiateTheta(self, theta):
36
           self.theta_dot = self.beta*self.theta_dot + (1-self.beta)*((theta - self.
37
           self.theta_d1 = theta
38
39
       def integrateError(self, error):
40
           self.integrator = self.integrator + (self.Ts/2.0)*(error + self.error_d1)
41
           self.error_d1 = error
42
43
       def integratorAntiWindup(self, F, F_unsat):
44
           if self.ki != 0.0:
45
               self.integrator = self.integrator + P.Ts/self.ki*(F-F_unsat)
46
47
       def saturate(self,u):
48
           if abs(u) > self.limit:
               u = self.limit*np.sign(u)
50
           return u
```

The complete simulation files are contained on the wiki associated with this book.