

Estimated weight

Arms 1 : 0.045 kg ($\times 2$)
 Arms 2 : 0.046 kg ($\times 2$)
 Motors : 0.046 kg ($\times 4$)
 Propellers : 0.015 kg ($\times 4$)
 Body : 0.612 kg

Tie downs : 0.035 kg ($\times 2$)
 Wing Mount : 0.171 kg
 Controller : 0.011 kg
 Battery : 0.270 kg

Total Weight ≈ 1.56 kg

Weights

$$F_{\text{arms}_1} = 0.045 \text{ kg} \cdot 9.81 \text{ m/s}^2 = 0.441 \text{ N}$$

$$F_{\text{arms}_2} = 0.046 \text{ kg} \cdot 9.81 \text{ m/s}^2 = 0.451 \text{ N}$$

$$F_{\text{tie downs}} = 0.035 \text{ kg} \cdot 9.81 \text{ m/s}^2 = 0.343 \text{ N}$$

$$F_{\text{body}} = (0.612 + 0.171 + 0.011 + 0.270) \text{ kg} \cdot 9.81 \text{ m/s}^2 = 10.438 \text{ N}$$

$$F_{\text{motors}} = (0.046 + 0.015) \text{ kg} \cdot 9.81 \text{ m/s}^2 = 0.598 \text{ N}$$

$$F_{\text{Weight}} = 2 \cdot F_{\text{arms}_1} + 2 \cdot F_{\text{arms}_2} + F_{\text{body}} + 4 \cdot F_{\text{motors}} \cdot 2 \cdot F_{\text{tie downs}} = 13.506 \text{ N}$$

Thrust

$$F_{\text{thrust}_{\max}} = 958 \text{ gram-Force} = 9394.77 \text{ mN} = 9.395 \text{ N}$$

1 gram-Force
 $=$
 9.80665 mN

$$F_{\text{net}_{\max}} = (4 \cdot F_{\text{thrust}_{\max}}) - F_{\text{weight}} = 24.074 \text{ N} \therefore \text{Max Payload } \approx 2.454 \text{ kg}$$

$\text{(Net force available)} \quad (\text{@ MAX THRUST}) \approx 5.410 \text{ lbs}$

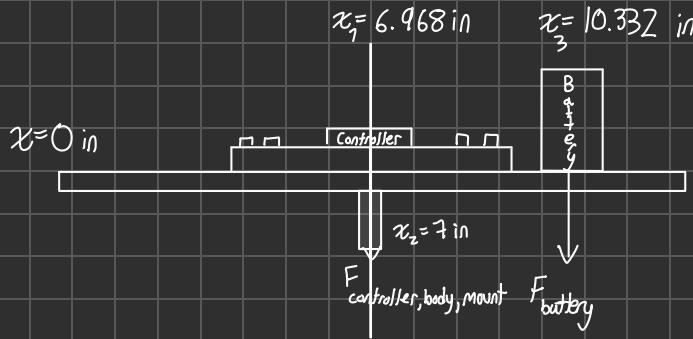
(AT MAX THRUST)

$$F_{net} = 4t - 73.506 \text{ (Net force w/o Thrust)}$$

$$\text{Max payload} = \frac{F_{net}}{9.81} = 0.408t - 1.377 \text{ (Max payload w/o thrust)}$$

Finding the center of mass (body) (NOT RTG/MJ)

* ASSUMING F_{weight} OF CONTROLLER, BATTERY & MOUNT ARE POINT FORCES *



$$x_{cm} = \frac{\sum m_i \vec{r}_i}{M}$$

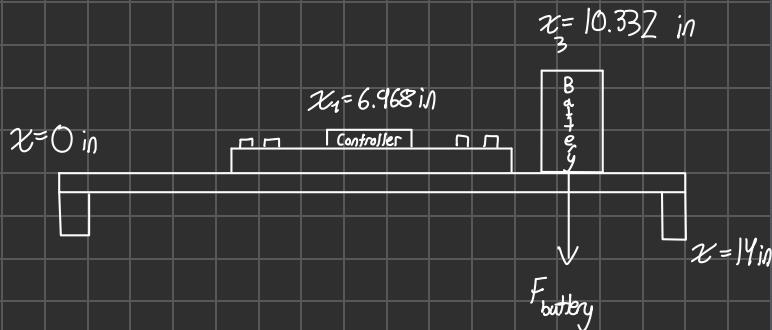
$$7.664 = \frac{(0.612 \cdot 7) + (0.035 \cdot 7) + 0.035x + (0.011 \cdot 6.968) + (0.270 \cdot 10.332) + (0.71 \cdot 6.870)}{1.134}$$

$$8.691 = 8.562 + 0.035x$$

$x = 3.686 \text{ [in]}$ (Location of 2nd tie down to make center of mass to align w/ center of moments)

THIS IS W/O PAYLOAD

Aligning Center of Mass to Center of Moments



Center of Moments: 7.664 in (Needs to = Center of Mass)

$$x_{cm} = \frac{\sum M_i \vec{r}_i}{M}$$

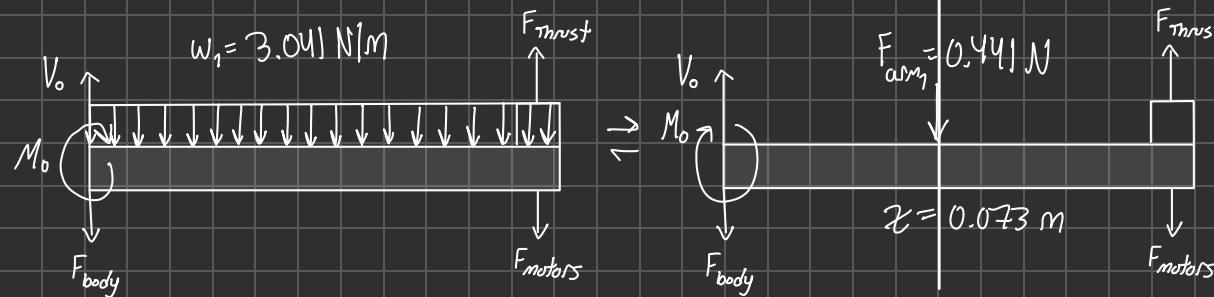
$$\begin{aligned} 7.664 &= (0.617 \cdot 7) + (0.035 \cdot 14) + (0.011 \cdot 6.968) + (0.27 \cdot 10.332) \\ &\quad \frac{(0.171 \cdot 6.82) + (2.454x)}{3.588} \end{aligned}$$

$$27.498 = 8.807 + 2.454x$$

$$x = 7.617 \text{ in} \quad (\text{Location of payload to have center of mass = center of moments})$$

Calculating Moments & Shears of Arms

Arm 1:



$$L = 0.145 \text{ m} \quad F_{arm_1} = 0.441 \text{ N} \quad w_1 = 3.041 \text{ N/m}$$

$$R = 0.441 \text{ N} @ 0.073 \text{ m}$$

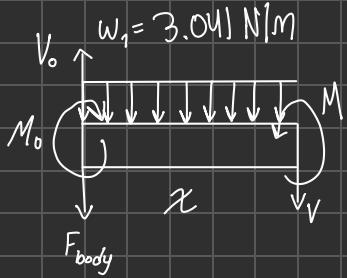
$$\sum F_y = V_o - F_{body} - F_{arm_1} - F_{motors} + F_{thrust} = 0$$

$$V_o - 35.198 - 0.441 - 0.598 + 9.395 = 0$$

$$V_o - 26.842 \therefore V_o = 26.842 \text{ [N]}$$

$$\sum M_{x=0} = -M_o - 0.441(0.073) - 0.598(0.127) + 9.395(0.127) = 0$$

$$-M_o + 1.085 = 0 \therefore M_o = 1.085 \text{ [N.m]}$$



$$\sum F_y = V_0 - w_1 x - F_{body} - V = 0$$

$$26.842 - 3.041x - 35.198 - V = 0$$

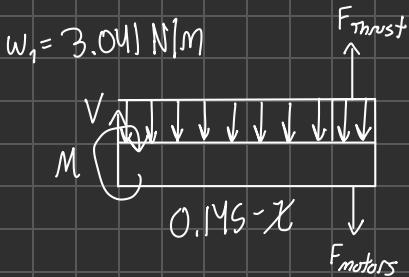
$$V = -3.041x - 8.356 \text{ [N]} \quad (0 < x < 0.073)$$

$$\sum M_{x=0} = -M_0 - 3.041x\left(\frac{x}{2}\right) - Vx + M = 0$$

$$-1.085 - 1.521x^2 + 3.041x^2 + 8.356x + M = 0$$

$$1.521x^2 + 8.356x - 1.085 + M = 0$$

$$M = -1.521x^2 - 8.356x + 1.085 \text{ [N·m]} \quad (0 < x < 0.073)$$



$$\sum F_y = V - 3.041(0.145 - x) - F_{motors} + F_{thrust} = 0$$

$$V - 0.441 + 3.041x - 0.598 + 9.395 = 0$$

$$V + 3.041x + 8.356 = 0 \quad \therefore V = -3.041x - 8.356 \text{ [N]} \quad (0.073 < x < 0.145)$$

$$\sum M_{x=0.073} = -M - ((0.441 - 3.041x)(0.073 - 0.5x)) - 0.598(0.177 - x)$$

$$+ 9.395(0.177 - x) = 0$$

$$-\mathcal{M} - (0.032 - 0.221x - 0.222x^2 + 1.57x^3) - 0.076 + 0.598x \\ + 1.193 - 9.395x = 0$$

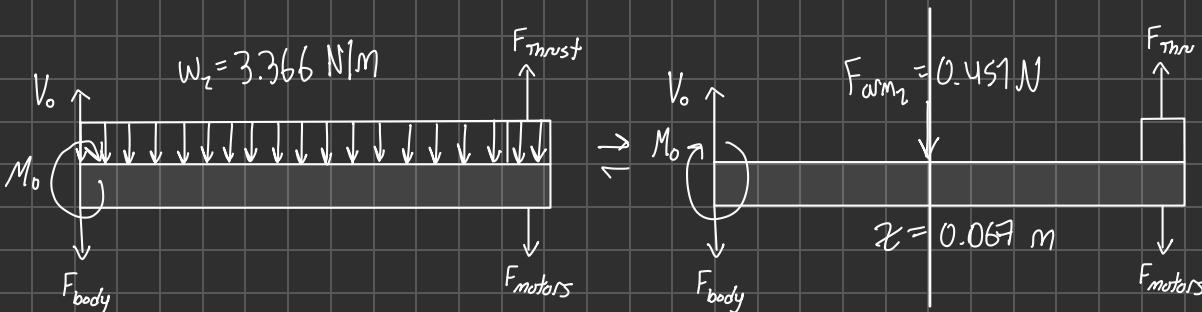
$$-\mathcal{M} - (1.57x^2 - 0.443x + 0.032) + 1.117 - 8.797x = 0$$

$$-\mathcal{M} - 1.57x^2 + 0.443x - 0.032 + 1.117 - 8.797x = 0$$

$$-\mathcal{M} + 1.085 - 8.354x - 1.57x^2 = 0$$

$$\mathcal{M} = -1.57x^2 - 8.354x + 1.085 \text{ (N·m)} \quad (0.073 < x < 0.145)$$

Arm 2:



$$L = 0.134 \text{ m} \quad F_{arm_2} = 0.451 \text{ N} \quad w_2 = 3.366 \text{ N/m}$$

$$R = 0.451 \text{ N} @ 0.067 \text{ m}$$

$$\sum F_y = V_o - F_{body} - F_{arm_2} - F_{motors} + F_{thrust} = 0$$

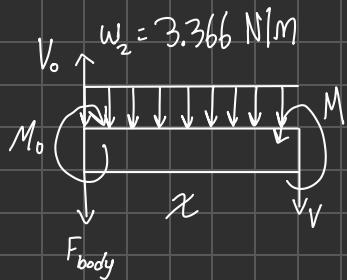
$$V_o - 35.198 - 0.451 - 0.598 + 9.395 = 0$$

$$V_o - 26.852 = 0 \quad \therefore V_o = 26.852 [N]$$

$$\sum_{x=0}^M M = -M_o - 0.451(0.067) - 0.598(0.117) + 0.345(0.117) = 0$$

$$-M_o - 0.03 - 0.07 + 1.099 = 0$$

$$-M_o + 0.999 = 0 \quad \therefore M_o = 0.999 [N \cdot m]$$



$$\sum F_y = 26.852 - 35.148 - 3.366x - V = 0$$

$$-8.346 - 3.366x - V = 0 \quad \therefore V = -3.366x - 8.346 [N]$$

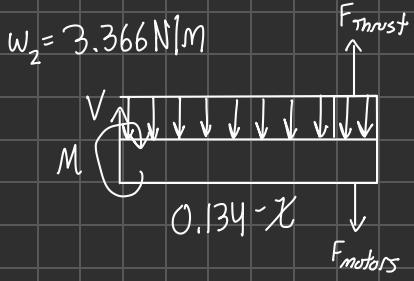
$$(0 < x < 0.067)$$

$$\sum_{x=0}^M M = -0.999 - 3.366x\left(\frac{x}{2}\right) - Vx + M = 0$$

$$-0.999 - 1.683x^2 + 3.366x^2 + 8.346x + M = 0$$

$$1.683x^2 + 8.346x - 0.999 + M = 0$$

$$\therefore M = -1.683x^2 - 8.346x + 0.999 [N \cdot m] \quad (0 < x < 0.067)$$



$$\sum F_y = V - 3.366(0.134 - x) - 0.598 + 9.395 = 0$$

$$V - 0.451 + 3.366x - 0.598 + 9.395 = 0$$

$$V + 3.366x + 8.346 = 0$$

$$V = -3.366x - 8.346 \quad [N] \quad (0.067 < x < 0.134)$$

$$\begin{cases} M \\ x=0.067 \end{cases} = -M - ((0.451 - 3.366x)(0.067 - 0.5x)) - 0.598(0.117 - x) + 9.395(0.117 - x) = 0$$

$$-M - (0.03 - 0.226x - 0.226x^2 + 1.683x^3) - 0.07 + 0.598x + 1.099 - 9.395x = 0$$

$$-M - (1.683x^3 - 0.452x^2 + 0.03) + 1.029 - 8.797x = 0$$

$$-M - 1.683x^3 + 0.452x^2 - 0.03 + 1.029 - 8.797x = 0$$

$$\therefore M = -1.683x^3 - 8.345x^2 + 0.999 \quad [N \cdot m] \quad (0.067 < x < 0.134)$$