

## MECÂNICA DE ESTRUTURAS AEROESPACIAIS

### Problems of practical classes of the chapter TORSION OF SOLID SECTIONS

#### Example 3.3

Determine the

- rate of twist and
- the stress distribution

in a circular section bar of radius  $R$

which is subjected to equal and opposite torques  $T$  at each of its free ends.

If we assume an origin of axes at the centre of the bar the equation of its surface is given by

$$x^2 + y^2 = R^2$$

If we now choose a stress function of the form

$$\phi = C(x^2 + y^2 - R^2)$$

the boundary condition  $\phi = 0$  is satisfied at every point on the boundary of the bar and the constant  $C$  may be chosen to fulfill the remaining requirement of compatibility.

Therefore from Eqs

$$-2G \frac{d\theta}{dz} = \nabla^2 \phi = F \text{ (constant)}$$

and the stress function Eq we have

$$4C = -2G \frac{d\theta}{dz}$$

so that

$$C = -\frac{G}{2} \frac{d\theta}{dz}$$

and

$$\phi = -G \frac{d\theta}{dz} (x^2 + y^2 - R^2)/2$$

Substituting for  $\phi$  in Eq.

$$T = 2 \iint \phi \, dx \, dy$$

we have

$$T = -G \frac{d\theta}{dz} \left( \iint x^2 \, dx \, dy + \iint y^2 \, dx \, dy - R^2 \iint dx \, dy \right)$$

The first and second integrals in this equation both have the value  $\pi R^4/4$  while the third integral is equal to  $\pi R^2$ , the area of cross-section of the bar.

## MECÂNICA DE ESTRUTURAS AEROESPACIAIS

Then

$$T = -G \frac{d\theta}{dz} \left( \frac{\pi R^4}{4} + \frac{\pi R^4}{4} - \pi R^4 \right)$$

which gives

$$T = \frac{\pi R^4}{2} G \frac{d\theta}{dz}$$

i.e.

$$T = GJ \frac{d\theta}{dz}$$

in which

$$J = \pi R^4 / 2 = \pi D^4 / 32$$

(D is the diameter), the *polar second moment of area* of the bar's cross-section.

Substituting for  $G(d\theta/dz)$  in Eq.

$$\phi = -G \frac{d\theta}{dz} (x^2 + y^2 - R^2) / 2$$

from

$$T = GJ \frac{d\theta}{dz}$$

we have

$$\phi = -\frac{T}{2J} (x^2 + y^2 - R^2)$$

and from Prandtl Eqs

$$\tau_{zy} = -\frac{\partial \phi}{\partial x} = \frac{Tx}{J} \quad \tau_{zx} = \frac{\partial \phi}{\partial y} = -\frac{T}{J}y$$

The resultant shear stress at any point on the surface of the bar is then given by

$$\tau = \sqrt{\tau_{zy}^2 + \tau_{zx}^2}$$

i.e.

$$\tau = \frac{T}{J} \sqrt{x^2 + y^2}$$

i.e.

$$\tau = \frac{TR}{J}$$

The above argument may be applied to any annulus of radius  $r$  within the cross-section of the bar so that the stress distribution is given by

$$\tau = \frac{TR}{J}$$

and therefore increases linearly from zero at the centre of the bar to a maximum  $TR/J$  at the surface.

# MECÂNICA DE ESTRUTURAS AEROESPACIAIS

## Problem 3.4.

Determine

- the maximum shear stress and
- the rate of twist in terms of the applied torque  $T$

for the section comprising narrow rectangular strips shown in Fig. P.3.5.

Solution

The torsion constant,  $J$ , for the complete cross-section is found by summing the torsion constants of the narrow rectangular strips which form the section.

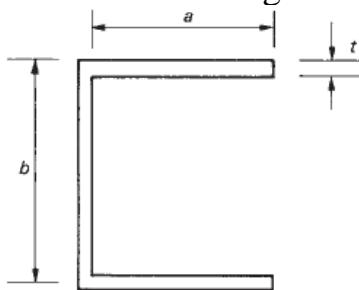


Fig. P.3.5

$$Ans. \quad \tau_{\max} = 3T/(2a+b)t^2, \quad d\theta/dz = 3T/G(2a+b)t^3.$$

Then, from Eq.

$$J = \frac{st^3}{3}$$

we have

$$J = 2 \frac{at^3}{3} + \frac{bt^3}{3} = \frac{(2a+b)t^3}{3}$$

Therefore, from the general torsion equation

$$T = GJ \frac{d\theta}{dz}$$

we have

$$\frac{d\theta}{dz} = \frac{3T}{G(2a+b)t^3}$$

The maximum shear stress follows from the previous Eq and Eq

$$\tau_{zy,\max} = \pm Gt \frac{d\theta}{dz}$$

$$\text{Hence } \tau_{\max} = \pm Gt \frac{d\theta}{dz} = \pm \frac{3T}{(2a+b)t^2}.$$