# Questionnaire on Gas Dynamics

# **Questionnaire on Gas Dynamics 1**

https://www.youtube.com/watch?v=TZkkdyHtl-A

- 0:00 Why the density is outside of the substantial derivative in the momentum equation
- 19:35 What are the total conditions
- 22:15 Definition of the total conditions for incompressible flow
- 25:26 Definition of the total conditions for compressible flow

### **Questionnaire on Gas Dynamics 2**

https://www.youtube.com/watch?v=li452zObFa8

- 0:00 Why should the flow be isentropic (reversible) to have constant stagnation pressure and density?
- 12:21 Obtaining algebraic form of the energy equation from its integral form
- 21:51 Practical application of the stagnation enthalpy and temperature in the gas dynamics
- 29:11 Stagnation speed of sound
- 33:35 Characteristic speed of sound
- 41:14 Characteristic flow properties
- 42:38 Practical application of the relation between Characteristic and Stagnation flow properties

### **Questionnaire on Gas Dynamics 3**

https://www.youtube.com/watch?v=Jo7AptdVBiw

- 0:00 What is the free-stream Mach number?
- 1:59 When the flow is compressible?
- 4:10 How far from the body the flow properties are considered constant?
- 5:00 What if M is close to 0.3?
- 7:00 Characteristic flow properties (applications)
- 18:43 Limits of the characteristic Mach number
- 21:23 How to use tables to calculate the shockwaves or isentropic flow properties?
- 27:09 Validation of the simulation in one program by the other one

### **Questionnaire on Gas Dynamics 4**

https://www.youtube.com/watch?v=SpO mUuM0-k

- 0:00 Calculation of the flow properties in isentropic and non-isentropic cases (normal shock-wave)
- 28:39 Comment about the Lambda-shock-wave in the nozzle

30:20 Mach number as a function of the pressure

31:33 Limits of constant Cp, Cv, R, gamma and isentropic conditions

# **Questionnaire on Gas Dynamics 5**

https://www.youtube.com/watch?v=1 vYpoBmlNA

0:00 Introduction

0:25 Left-running and right-running shockwaves, the concept of the shockwave interaction model

5:12 Why are the expansion waves isentropic?

7:21 Oblique shockwave and boundary layer interactions

Recommended literature: Shock Wave-Boundary-Layer Interactions (Cambridge Aerospace Series Book 32) 1st Edition, 480 pages, 2011, ISBN-13: 978-0521848527

# **Questionnaire on Gas Dynamics 6**

https://www.youtube.com/watch?v=hqd9jqoH83c

0:00 What is an optimal diffuser inlet?

1:03 Geometry of a 3-shock-wave diffuser

5:34 Losses in a shock-wave

14:48 Total losses in a diffuser

18:38 Optimization parameters in diffuser design and their limits

26:54 References to the textbooks about oblique shock-wave calculations, book: R. Hermann, Supersonic inlet diffusers and introduction to internal aerodynamics, Minneapolis, 1956

28:32 What are the Chapters to study for Work 1

29:31 Planning the TCC in Aero- Gas- Dynamics or Propulsion (experimental or numerical) http://fga.unb.br/olexiy/sugestoes-de-tcc-e-pibic

## **Questionnaire on Gas Dynamics 7**

https://www.youtube.com/watch?v=VMU-29f0Hiw

Chapter 9: Oblique Shock and Expansion Waves
Aspects of the oblique shock-waves and optimal diffuser calculation

### **Questionnaire on Gas Dynamics 8**

https://www.youtube.com/watch?v=0ygzm3LT0VE

Simulation of Supersonic Diffusers and Nozzles

- 0:00 How to prevent the normal shockwave from going out from the diffuser destroying the oblique shockwaves and blocking the flow (case 1)
- 5:14 Moving normal shockwave (case 2)
- 6:15 Flow starts to diverge after some iterations
- 10:19 Other geometry problem in the subsonic section
- 13:11 The exit pressure problem
- 14:45 Why the residuals rise (another explanation)
- 16:39 Importance of studying the Gas Dynamics course
- 17:26 Evaluation problems in the Gas Dynamics course
- 18:52 About the oral test planning
- 20:25 Oral test subjects

# **Questionnaire on Gas Dynamics 9**

https://www.youtube.com/watch?v=nwSEv3kdmT0

The solution of the practical tasks for the oral test - part 1

- 0:00 Introduction
- 0:36 Normal shockwave, example 1
- 3:57 Normal shockwave, example 2
- 4:47 Normal shockwave, example 3
- 7:24 Normal shockwave, example 4
- 9:52 Normal shockwave, example 5
- 15:06 Normal shockwave, example 6
- 23:28 Oblique shockwave, example 2.1a
- 31:19 Oblique shockwave, example 2.1b
- 41:35 Oblique shockwave, example 2.2
- 47:29 Oblique shockwave, example 2.3a
- 50:16 Oblique shockwave, example 2.3b
- 52:36 Oblique shockwave, example 2.3c
- 55:43 Oblique shockwave, example 2.4

# **Questionnaire on Gas Dynamics 10**

https://www.youtube.com/watch?v=HmlBqlMExtl

The solution of the practical tasks for the oral test - part 2

- 0:00 Mach-area relation, example 3.1a
- 13:51 Mach-area relation, example 3.1b
- 16:02 Mach-area relation, example 3.2
- 27:49 Mach-area relation, example 3.3
- 30:31 Mach-area relation, example 3.4
- 39:54 Mach-area relation, example 3.5
- 40:24 Mach-area relation, example 4 with error and further correction

### **Questionnaire on Gas Dynamics 11**

https://www.youtube.com/watch?v=G2nVpmHs2uY

# The solution of the practical tasks for the oral test - part 3 AND Simulation in Ansys Fluent

- 0:00 No convergence of the viscous flow simulation
- 4:42 Oblique shockwave in a non-isentropic nozzle
- 10:19 Convergence of the flow in the nozzle
- 12:12 Simulation of the flow in the nozzle of the low area ratio
- 14:44 Isentropic flow, introduction to examples
- 17:53 Isentropic flow, example 5.1
- 22:44 Isentropic flow, example 5.2
- 28:11 Isentropic flow, example 5.3
- 33:00 Isentropic flow, example 5.4
- 39:44 Expansion waves, introduction to examples
- 45:04 Expansion waves, example 6.1
- 51:29 Expansion waves, example 6.2
- 56:22 Expansion waves, example 6.3a
- 58:53 Expansion waves, example 6.3b
- 1:01:14 Final considerations on the solution of the practical tasks

# **Questionnaire on Gas Dynamics 12**

https://youtu.be/yNePBidaCpQ

# **Chapter 9: Oblique Shock and Expansion Waves**

Aspects of the oblique shock-waves and optimal diffuser calculation continuation of part 7: https://www.youtube.com/watch?v=\_1uWVWNnheM)

### **Questionnaire on Gas Dynamics 13**

https://youtu.be/zwwnn5SS-jk

### **Compressible Flow in a Variable-Area Duct**

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0:00 Introduction

- 4:41 Flow expansion (transition from region 3 to 4)
- 12:56 Heat addition
- 21:17 Flow in the nozzle
- 24:03 Calculation example
- 37:00 Finding the internal and external diffuser size (D and Dint)
- 42:47 Why three shock waves coincide at the same point?
- 47:13 Limitations of the Area-Mach number relation (shaping of the nozzle)
- 50:27 Another comment about the diffuser size D

- 51:34 Conical and bell-shaped nozzle flow results
- 54:37 About a wrong approach to do works in gas dynamics
- 57:00 Can I opt to modify a diffuser or nozzle geometry?
- 59:35 The diffuser and nozzle are planar and not axis-symmetrical.
- 1:03:32 Is there any advantage to use a cylindrical ramjet?
- 1:06:28 Why we don't see ramjets in everyday life?
- 1:09:04 Peaceful applications of ramjets
- 1:09:54 Just look on the SpaceX...

### Additional links:

Mach-area relation: <a href="https://www.youtube.com/watch?v=HmlBqlMExtl">https://www.youtube.com/watch?v=HmlBqlMExtl</a>

The geometry of a 3-shock-wave diffuser: <a href="https://www.youtube.com/watch?v=hqd9jqoH83c">https://www.youtube.com/watch?v=hqd9jqoH83c</a>

How to use tables to calculate the shockwaves or isentropic flow properties?

https://www.youtube.com/watch?v=Jo7AptdVBiw

### Literature:

- 1. Anderson, J. Fundamentals of Aerodynamics, McGraw-Hill, 6<sup>th</sup> edition, 2016.
- 2. BERTSEKAS, Dimitri P; NEDIC, Angelia; OZDAGLAR, Asuman E. Convex analysis and optimization. Belmont: Athena Scientific, 2003. xv, 534 p. ISBN 1886529450
- 3. ANDERSON, John David. Modern compressible flow: with historical perspective. 3rd ed. New York: McGraw-Hill, 2003. xvi, 760 p. ISBN 9780072424430).

## **Questionnaire on Gas Dynamics 14**

### https://youtu.be/arS9ZG2iL0Q

### **Compressible Flow in a Variable-Area Duct**

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- 0:00 Questions asking: thrust optimization, the optimal exit of the diffuser
- 3:14 Answer 1: from the point of view of the thrust optimization problem
- 3:45 Answer 2: the optimal exit of the diffuser and its relation with all the subsonic flow region 19:38 Drag calculation on a diffuser
- 28:51 Drag calculation on a diffuser: an additional question pressure variation in a subsonic section
- 32:16 Application of the theorem of the average value to drag calculation
- 35:46 Matlab implementation of the Area-Mach number relation
- 41:40 Matlab implementation of a diffuser optimization algorithm
- 46:34 Matlab implementation of a heat addition algorithm in
- 49:01 How to evaluate maximum possible heat addition (dT)