5) i)
$$\Omega_1 = 2.3.21a$$
 ignorando efectos propulsivos
$$C_{mx} = \alpha(h - h_{nwb}) - \alpha_x V_{H} \left(1 - J_{E}\right)$$

$$\Omega_1 = 1.5$$

$$\int_{\mathcal{A}} = 2.2.10$$

$$\sqrt{\frac{5}{t}} = \frac{1}{5}$$

$$Q_{1} = 2.3.18$$

$$\alpha = a_{Wb} \left[1 + \frac{3}{4} \cdot \frac{5}{5} \left(1 - \frac{1}{4} \cdot \frac{5}{4} \right) \right]$$

$$\widehat{\pi}$$

Usando Cm<0, substituendo (II) e (III) em (I) $C_{mx} = a \left(h - h_{nwb}\right) - a_t V_{H} \left(1 - \frac{1}{2}E\right) < 0$ $h < h_{nwb} + a_t \left(\frac{1}{2}S_t\right) \left(1 - \frac{1}{2}E\right)$ 4 < 0,5607

ii)
$$D_{i}$$
 2.3. $22a$ ignorando efutos propulsivos

 $C_{m_{o}} = C_{mac_{wb}} + a_{t} V_{H} (\mathcal{E}_{o} + i_{t}) \left[1 - a_{t} \frac{S_{t}}{a} \left(1 - \frac{1}{2} \mathcal{E}_{o} \right) \right] V_{o}$
 $V_{o} = C_{m_{ac_{wb}}} + a_{t} V_{H} (\mathcal{E}_{o} + i_{t}) \left[1 - a_{t} \frac{S_{t}}{a} \left(1 - \frac{1}{2} \mathcal{E}_{o} \right) \right] > 0$
 $C_{m_{o}} = C_{m_{ac_{wb}}} + a_{t} V_{H} (\mathcal{E}_{o} + i_{t}) \left[1 - a_{t} \frac{S_{t}}{a} \left(1 - \frac{1}{2} \mathcal{E}_{o} \right) \right] > 0$

$$i_{t} = \frac{C_{macwb}}{a_{t} \int_{z}^{t} \int_{z}^{$$

(iii) Para voo trumade
$$de = 0$$
, $de 2.3.20a$

$$C_{m} = C_{mo} + C_{ma} \propto = 0$$

$$En voo não acclerado e com $C_{l} = a \propto 1$

$$L = W = a \propto 1$$

$$\sqrt{S}$$$$

sendo $W = mq^2 22680 \cdot 9.81 = 222491 N$ $8^2 (28)^2 0.139 = 86.875 m^2$

Encontra. se a de III logo = = 3,141° Contrinando I, III, VIII em VIII temos it = f(h)

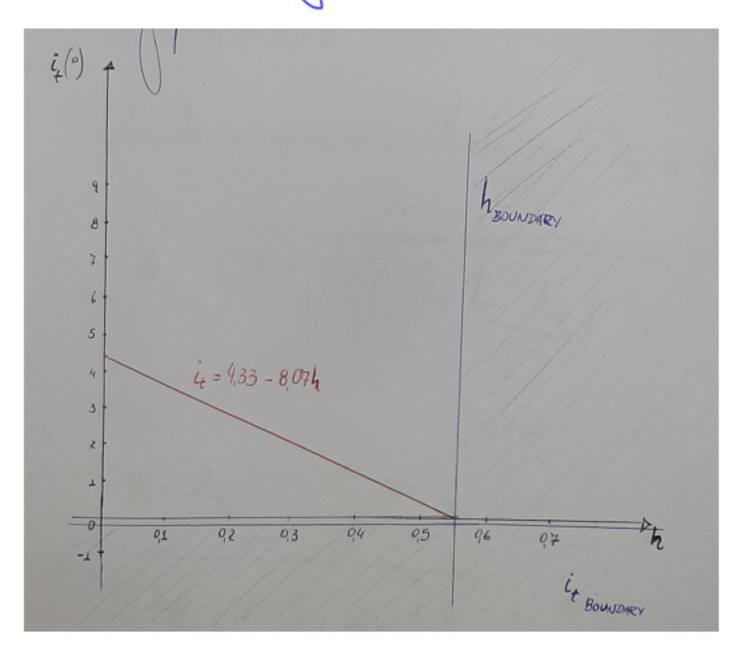
Cm. + Cm. ~ = 0

$$C_{Moc_{Wb}} + a_t \bar{V}_H (\mathcal{E}_0 + i_t) \left[1 + \underline{a}_t \underbrace{S_t} \left(1 - \underline{J} \underbrace{\mathcal{E}} \right) \right] + \left[a(h - h_{n_{Wb}}) - a_t \bar{V}_H \left(1 - \underline{J} \underbrace{\mathcal{E}} \right) \right] \underbrace{\frac{2W}{V^2 S_a}} = 0 \qquad \text{IX}$$

$$i_t = -C_{Moc_{Wb}} - \left[a(h - h_{n_{Wb}}) - a_t \bar{V}_H \left(1 - \underline{J} \underbrace{\mathcal{E}} \right) \right] \underbrace{\frac{2W}{V^2 S_a}} - \mathcal{E}_0$$

$$a_t \bar{V}_H \underbrace{1 + \underline{a}_t \underbrace{S_t}}_{a_s} \left(1 - \underbrace{J} \underbrace{\mathcal{E}} \right)$$

it = (4,33 - 8,074) deg



6) i)
$$Q_{a} C_{a} Z_{b}$$
, T_{envol}

$$A = b^{2} = 7.16$$

Da tabela C.1
$$= \frac{2c_n}{3} \frac{1+\lambda}{1+\lambda}$$

ic) Da eg-include na fig. B. 1-2 $\frac{C_{L_{\alpha}}}{A} = \frac{2\pi}{2 + \frac{A^2 B^2}{B^2}} \left[\frac{1 + \tan^2 A_{\alpha}}{B^2} \right] + 4$ time: tan 1/4 = tan 22° = 0,404 1 + tem? las + 4 = 7,977

$$\mathcal{E}_{m} = \mathcal{C}_{La} = \frac{2\pi A}{2 + A^{2}B^{2} + 4}$$

$$2 + A^{2}B^{2} + 4$$

$$R^{2} + 4$$

$$a_t = 0.068 \text{ dig}^2 = 3.90 \text{ rad}^2$$

$$D_{e} = 2.3.18$$

$$\alpha = a_{Wb} \left[\frac{1}{2} + \frac{a_{t}}{6} + \frac{S_{t}}{6} \left(\frac{1}{2} + \frac{16}{24} \right) \right]$$

$$\alpha = 5.05 \text{ rad}$$

Iv)
$$l_t = l_t$$
 entao $V_H = V_H$
 $Q_t = V_H - S_t$
 $V_H = V_H - S_t$
 $V_H = V_H$
 $V_H = V_$

Compa² a
$$(h - h_{\eta})$$

Substi. (XII) e (XII) em (XIV)

$$C_{m_{\alpha}} = -\alpha_{t} V_{\mu} \left(\frac{1}{2} - \frac{1}{2} \epsilon \right)$$



$$C_{mar} = -a_t \frac{l_t S_t}{l_t S_t} \left(1 - \frac{J \varepsilon}{J \omega} \right) = -l_t 90 \text{ rad}^{-J}$$
 (xv)

v) De grafice C_m versus C_l observa-se que C_{m je} e' independente de C_l e C_m . A de apenas muda as lunhas por uma constante $C_{mfe} = \frac{dC_{m}}{dJ_{a}} = \frac{\Delta C_{m}}{\Delta J_{e}} = \frac{C_{mc} - C_{mi}}{20} = \frac{-0.66}{20}$ $C_{mfe} = -0.020 \text{ deg}^{-1}$

Para
$$h = 0.35$$

Lettin = Lettin = Lettin - Lettin

Lettin - Claim





sende det = Cia Compe - Cije Como

(XX)

De XIX

h-hn = - dit de form Or dC prin Para as duas posições do CG termos;

$$\begin{cases}
0,35 - h_n = 11,4 \text{ det} \\
0,25 - h_n = 15,03 \text{ det} \\
Assin encontraviens
\end{cases}$$

D. Z. 3. 25 c

(XXII)

(a)

(b)

XXIII)