Analytical solution for Sod shock tube

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This description shows the equations used in sod_shock_exact.py, a script calculating the analytical solution of the Sod shock tube problem after a time t. Equations labelled here bear the same tags as in that script.

The Sod shock tube contains two different regions separated by a thin barrier at $x = x_0$. Each region contains the same gas, but with different pressure and density.

The initial conditions for the left and right domains are

$$p_l = 1$$
, $p_r = 0.1$
 $\rho_l = 1$, $\rho_r = 0.125$
 $u_l = 0$, $u_r = 0$

where p is pressure, ρ is mass density, and u is the gas velocity in the direction along the tube. At t=0, the barrier is removed, and the two regions are allowed to mix.

When the barrier is removed, the tube is split into five domains.

- Left boundary (1): pressure, density, and velocity are constant
- Expansion fan (e): pressure and density decreases
- Contact area (2): region where the initial regions meet
- Shock (1): shock front creates a discontinuity with the right boundary
- Right boundary (r): pressure, density, and velocity are constant The system is described with the one-dimensional Euler system of PDEs

$$\frac{\partial}{\partial t} \begin{bmatrix} \rho \\ \rho u \\ e \end{bmatrix} + \frac{\partial}{\partial x} \begin{bmatrix} \rho u \\ \rho u^2 + p \\ (e+p)u \end{bmatrix} = 0,$$

where e is the internal energy.

We use the Equation Of State (EOS) to close the system of equations. The EOS is given by

$$p = \rho(\gamma - 1)e$$

with $\gamma = 1.4$. We define the constants

$$\Gamma = \frac{\gamma - 1}{\gamma + 1}, \quad \beta = \frac{\gamma - 1}{2\gamma},$$

which will be useful in our calculations.

Another equation needed before diving into the solution, is the equation for sound speed c. Without further ado,

$$c = \sqrt{\gamma \frac{p}{\rho}} \,.$$

First, the *shock* region is described by the Rankine–Hugoniot relations (wikipedia page). These relations can be simplified to

$$u_1 - u_2 = 0.$$

We can express the velocity in terms of pressure,

$$(p' - p_r) \sqrt{\frac{1 - \Gamma}{\rho_r(p' + \Gamma p_r)}} - (p_l^{\beta} - p'^{\beta}) \sqrt{\frac{(1 - \Gamma^2)p_l^{1/\gamma}}{\Gamma^2 \rho_l}} = 0,$$

and the p' that satisfies this relation is in fact the pressure in the shock region. Furthermore,

$$p_{1} = p',$$

$$\rho_{1} = \rho_{r} \frac{p_{1} + \Gamma p_{r}}{p_{r} + \Gamma p_{1}},$$

$$u_{1} = (p_{1} - p_{r}) \sqrt{\frac{1 - \Gamma}{\rho_{r}(p_{1} + \Gamma p_{r})}}.$$
(1)

Then, the physical quantities of the *contact* region are found by the pressure and velocity being continuous, and mass density follows from the adiabatic gas law. This gives

$$u_{2} = u_{1},$$

$$p_{2} = p_{1},$$

$$\rho_{2} = \rho_{l} \left(\frac{p_{2}}{p_{l}}\right)^{1/\gamma}.$$

$$(2)$$

Expressions for *shock* and *contact* regions are inspired by Wikipedia's article on the Sod shock tube.

The expansion fan is solved according to Höfner's lecture notes (chapter 10). We calculate the region accordingly

$$u_e = \frac{2}{\gamma + 1} \left(c_l + \frac{x - x_0}{t} \right) ,$$

$$p_e = p_l \left(\frac{c_e}{c_l} \right)^{2\gamma/(\gamma - 1)} ,$$

$$\rho_e = \gamma \frac{p_e}{c_e^2} .$$
(e)

The sound speed in the expansion fan is not constant, and found by

$$c_e = c_l - (\gamma - 1) \frac{u_e}{2} .$$

Lastly, the boundaries of the different regions are calculated. There are four boundaries that need to be found. The boundaries between *left* boundary and *expansion fan* (le), *expansion fan* and *contact* region (e2), and between *contact* and *shock* region (21) are found using the same expressions as Höfner's lecture notes. The boundary between the *shock* region and the *right* boundary (1r) is inspired by Ibackus' github repository.

$$x_{le} = x_0 - c_l t, (le)$$

$$x_{e2} = x_0 + (u_2 - c_2)t, (e2)$$

$$x_{21} = x_0 + u_2 t \,, \tag{21}$$

$$x_{1r} = x_0 + wt. (1r)$$

(1)

The factor w in equation (1r) is given by

$$w = c_r \sqrt{1 + \frac{\gamma + 1}{2\gamma} \left(\frac{p_2}{p_r} - 1\right)}.$$