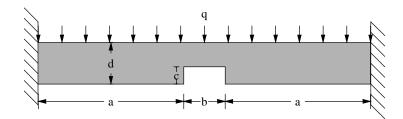
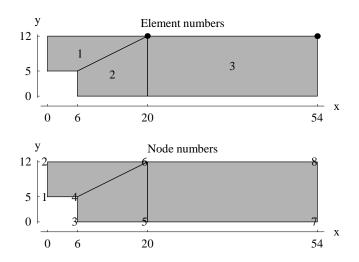
# Example 7.7: Notched beam (p. 510)

Find stresses in a notched beam of rectangular cross-section shown in Figure. The following numerical values are used.

 $a = 48 \text{ in}; \quad b = 12 \text{ in}; \quad c = 5 \text{ in}; \quad d = 12 \text{ in}; \quad \text{Thickness} = 4 \text{ in}; \quad E = 3 \times 10^6 \text{ lb} / \text{in}^2; \quad v = 0.2; \quad q = 50 \text{ lb} / \text{in}^2$ 



Since the thickness of the beam is much smaller than the other dimensions, and there are no out of plane loads, the problem can be treated as a plane stress situation. Using symmetry half of the beam is modeled. To show all calculations, a very coarse model involving only three elements is used.



Global equations at start of the element assembly process

#### Equations for element 1

$$E = 3000000;$$
  $v = 0.2;$   $h = 4$ 

Nodal coordinates

Element node	Global node number	X	y
1	1	0.	5.
2	4	6.	5.
3	6	20.	12.
4	2	0.	12.

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\ N_2,\ N_3,\ N_4\} &= \Big\{\frac{1}{4}\,(s-1)\,(t-1),\ -\frac{1}{4}\,(s+1)\,(t-1),\ \frac{1}{4}\,(s+1)\,(t+1),\ -\frac{1}{4}\,(s-1)\,(t+1)\Big\} \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{\frac{t-1}{4},\ \frac{1-t}{4},\ \frac{t+1}{4},\ \frac{1}{4}\,(-t-1)\Big\} \\ \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} &= \Big\{\frac{s-1}{4},\ \frac{1}{4}\,(-s-1),\ \frac{s+1}{4},\ \frac{1-s}{4}\Big\} \end{split}$$

Mapping to the master element

$$\begin{aligned} x(s,t) &= \textit{N}^{T}\textit{x}_{n} = 3.5 \text{ t } s + 6.5 \text{ s} + 3.5 \text{ t} + 6.5 \\ y(s,t) &= \textit{N}^{T}\textit{y}_{n} = 3.5 \text{ t} + 8.5 \end{aligned}$$

For numerical integration the Gauss quadrature points and weights are

Computation of element matrices at  $\{-0.57735, -0.57735\}$  with weight = 1.

$$\boldsymbol{J} = \begin{pmatrix} 4.47927 & 1.47927 \\ 0 & 3.5 \end{pmatrix} \qquad \text{detJ} = 15.6775$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.622008,\ 0.166667,\ 0.0446582,\ 0.166667\}$ 

 $\{\partial N_1/\partial s, \partial N_2/\partial s, \partial N_3/\partial s, \partial N_4/\partial s\} = \{-0.394338, 0.394338, 0.105662, -0.105662\}$ 

 $\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \{-0.394338,\; -0.105662,\; 0.105662,\; 0.394338\}$ 

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.088036,\ 0.088036,\ 0.0235892,\ -0.0235892\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0754595,\ -0.0673977,\ 0.0202193,\ 0.122638\}$ 

$${\bf B}^{\rm T} = \begin{pmatrix} -0.088036 & 0 & 0.088036 & 0 & 0.0235892 & 0 & -0.0235892 & 0 \\ 0 & -0.0754595 & 0 & -0.0673977 & 0 & 0.0202193 & 0 & 0.0202193 & 0 & 0.0202193 & 0 & 0.0202193 & 0.076547 & 0.0202193 & 0.0202193 & 0.0202193 & 0.076547 & 0.0202193 & 0.02021$$

Computation of element matrices at  $\{-0.57735, 0.57735\}$  with weight = 1.

$$J = \begin{pmatrix} 8.52073 & 1.47927 \\ 0 & 3.5 \end{pmatrix} \qquad \text{det } J = 29.8225$$

```
\{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} = \{-0.105662,\ 0.105662,\ 0.394338,\ -0.394338\}
          \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} = \{-0.394338,\ -0.105662,\ 0.105662,\ 0.394338\}
          \{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.0124006,\ 0.0124006,\ 0.0462798,\ -0.0462798\}
          \{\partial N_1/\partial y, \partial N_2/\partial y, \partial N_3/\partial y, \partial N_4/\partial y\} = \{-0.107427, -0.0354304, 0.0106291, 0.132228\}
                                                                                                                              -0.0462798
                     -0.0124006
                                         0
                                                            0.0124006
                                                                              0
                                                                                              0.0462798 0
           \boldsymbol{B}^{\mathrm{T}} =
                                       -0.107427
                                                            0
                                                                            -0.0354304 0
                                                                                                              0.0106291
                     -0.107427
                                       -0.0124006 \quad -0.0354304
                                                                              0.0124006 \quad 0.0106291 \quad 0.0462798
                                                                                                                                0.132228
          k = 10^6
         1.77816
                          0.297963
                                             0.510224
                                                             -0.165885
                                                                               -0.384204 -0.751169
                                                                                                                  -1.90418
                                                                                                                                     0.619091
         0.297963
                          4.32502
                                           -0.0338069
                                                               1.39594
                                                                               -0.390325 \quad -0.511237
                                                                                                                    0.126169
                                                                                                                                   -5.20972
                                                                                 0.157784 \quad -0.234675
         0.510224
                        -0.0338069
                                                             -0.0982711
                                                                                                                                     0.366753
                                             0.244508
                                                                                                                  -0.912516
        -0.165885
                          1.39594
                                           -0.0982711
                                                               0.490888
                                                                               -0.102597 \quad -0.0548117
                                                                                                                    0.366753 - 1.83202
       -0.384204 \quad -0.390325
                                             0.157784
                                                             -0.102597
                                                                                 0.815278
                                                                                                  0.110026
                                                                                                                  -0.588859
                                                                                                                                     0.382896
                                           -0.234675
       -0.751169 \quad -0.511237
                                                             -0.0548117
                                                                                 0.110026
                                                                                                  0.361489
                                                                                                                    0.875818
                                                                                                                                     0.20456
        -1.90418
                                          -0.912516
                                                                                                                                   -1.36874
                          0.126169
                                                               0.366753
                                                                               -0.588859
                                                                                                  0.875818
                                                                                                                    3.40556
         0.619091 -5.20972
                                             0.366753
                                                             -1.83202
                                                                                 0.382896
                                                                                                  0.20456
                                                                                                                  -1.36874
                                                                                                                                     6.83718
Computation of element matrices at \{0.57735, -0.57735\} with weight = 1.
          J = \begin{pmatrix} 4.47927 & 5.52073 \\ 0 & 2^{\text{F}} \end{pmatrix}
                                                         detJ = 15.6775
          \{N_1, N_2, N_3, N_4\} = \{0.166667, 0.622008, 0.166667, 0.0446582\}
          \{\partial N_1/\partial s, \, \partial N_2/\partial s, \, \partial N_3/\partial s, \, \partial N_4/\partial s\} = \{-0.394338, \, 0.394338, \, 0.105662, \, -0.105662\}
          \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} = \{-0.105662,\ -0.394338,\ 0.394338,\ 0.105662\}
          \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.088036, 0.088036, 0.0235892, -0.0235892\}
          \{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\}=\{0.108674,\ -0.251532,\ 0.0754595,\ 0.0673977\}
          \mathbf{B}^{\mathrm{T}} =
        -0.088036
                          0
                                           0.088036
                                                           0
                                                                          0.0235892 0
                                                                                                          -0.0235892
                          0.108674
                                                         -0.251532 0
                                                                                                                               0.0673977
         0
                                                                                          0.0754595
                                                                                                            0
```

 $0.088036 \ 0.0754595 \ 0.0235892$ 

0.0673977 - 0.0235892

0

0.

-0.

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.166667,\ 0.0446582,\ 0.166667,\ 0.622008\}$ 

0.108674 - 0.088036 - 0.251532

$$\mathbf{k} = 10^6 \begin{vmatrix} 2.44459 & -1.12493 & -3.66154 & 1.61785 & 0.235849 & -0.0594203 & 0.981107 & -0.12493 & 2.92194 & 2.11077 & -5.96433 & -0.420264 & 1.44425 & -0.56558 \\ -3.66154 & 2.11077 & 6.47824 & -2.60369 & -1.08086 & -0.204736 & -1.73584 \\ 1.61785 & -5.96433 & -2.60369 & 13.0061 & 0.288186 & -3.55678 & 0.697658 & -0.235849 & -0.420264 & -1.08086 & 0.288186 & 0.555394 & 0.209297 & 0.289615 & -0.0594203 & 1.44425 & -0.204736 & -3.55678 & 0.209297 & 1.15949 & 0.0548588 \\ -0.981107 & -0.56558 & -1.73584 & 0.697658 & 0.289615 & 0.0548588 & 0.465117 & -0.433502 & 1.59814 & 0.697658 & -3.48497 & -0.0772193 & 0.953035 & -0.186937 \end{vmatrix}$$

Computation of element matrices at  $\{0.57735, 0.57735\}$  with weight = 1.

$$J = \begin{pmatrix} 8.52073 & 5.52073 \\ 0 & 3.5 \end{pmatrix} \qquad \text{detJ} = 29.8225$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.0446582,\ 0.166667,\ 0.622008,\ 0.166667\}$ 

 $\{\partial N_1/\partial s, \, \partial N_2/\partial s, \, \partial N_3/\partial s, \, \partial N_4/\partial s\} = \{-0.105662, \, 0.105662, \, 0.394338, \, -0.394338\}$ 

 $\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \{-0.105662,\; -0.394338,\; 0.394338,\; 0.105662\}$ 

 $\{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0124006, 0.0124006, 0.0462798, -0.0462798\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\}=\{-0.0106291,\ -0.132228,\ 0.0396684,\ 0.103189\}$ 

Summing contributions from all points we get

$$k = 10^6$$

$$\begin{pmatrix} 6.26209 & -0.0163755 & -4.11923 & 1.26638 & -0.951731 & -1.12991 & -1.19113 & -0.120087 \\ -0.0163755 & 9.0354 & 2.51638 & -3.67826 & -1.12991 & 0.228478 & -1.37009 & -5.58562 \\ -4.11923 & 2.51638 & 11.2621 & -3.76638 & -1.19113 & -1.37009 & -5.95173 & 2.62009 \\ 1.26638 & -3.67826 & -3.76638 & 21.5354 & -0.120087 & -5.58562 & 2.62009 & -12.2715 \\ -0.951731 & -1.12991 & -1.19113 & -0.120087 & 2.54484 & 0.786026 & -0.401981 & 0.463974 \\ -1.12991 & 0.228478 & -1.37009 & -5.58562 & 0.786026 & 2.55069 & 1.71397 & 2.80646 \\ -1.19113 & -1.37009 & -5.95173 & 2.62009 & -0.401981 & 1.71397 & 7.54484 & -2.96397 \\ -0.120087 & -5.58562 & 2.62009 & -12.2715 & 0.463974 & 2.80646 & -2.96397 & 15.0507 \\ \end{pmatrix}$$

$$\mathbf{r}^{\mathrm{T}} = (0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0)$$

Computation of element matrices resulting from NBC

NBC on side 3 with  $\{q_n,\,q_t\}=\{-50,\,0\}$ 

$$\left\{ N_{1},\ N_{2},\ N_{3},\ N_{4}\right\} _{c}=\left\{ 0,\ 0,\ \frac{1-a}{2},\ \frac{a+1}{2}\right\}$$

$$x(a) = 10. - 10. a;$$

$$y(a) = 12.$$

$$dx/da = -10.$$
;

$$dy/da = 0$$
.

$$J_{c} = 10.$$

Gauss point = 
$$-0.57735$$
; Weight = 1.;  $J_c = 10$ .

$$\{N_1,\ N_2,\ N_3,\ N_4\}_c=\{0,\ 0,\ 0.788675,\ 0.211325\}$$

$$\boldsymbol{r}_{q}^{T} = (\ 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad -1577.35 \quad 0 \quad -422.65 \ )$$

Gauss point = 0.57735; Weight = 1.; 
$$J_c = 10$$
.

$$\{N_1,\ N_2,\ N_3,\ N_4\}_c=\{0,\ 0,\ 0.211325,\ 0.788675\}$$

$$\mathbf{r}_{0}^{T} = (0 \ 0 \ 0 \ 0 \ 0 \ -422.65 \ 0 \ -1577.35)$$

Summing contributions from all Gauss points

$$\mathbf{r}_{0}^{T} = (0 \ 0 \ 0 \ 0 \ 0 \ -2000. \ 0 \ -2000.)$$

Complete element equations for element 1

$$\begin{pmatrix} 6.26209 & -0.0163755 & -4.11923 & 1.26638 & -0.951731 & -1.12991 & -1.19113 & -0.120087 \\ -0.0163755 & 9.0354 & 2.51638 & -3.67826 & -1.12991 & 0.228478 & -1.37009 & -5.58562 \\ -4.11923 & 2.51638 & 11.2621 & -3.76638 & -1.19113 & -1.37009 & -5.95173 & 2.62009 \\ 1.26638 & -3.67826 & -3.76638 & 21.5354 & -0.120087 & -5.58562 & 2.62009 & -12.2715 \\ -0.951731 & -1.12991 & -1.19113 & -0.120087 & 2.54484 & 0.786026 & -0.401981 & 0.463974 \\ -1.12991 & 0.228478 & -1.37009 & -5.58562 & 0.786026 & 2.55069 & 1.71397 & 2.80646 \\ -1.19113 & -1.37009 & -5.95173 & 2.62009 & -0.401981 & 1.71397 & 7.54484 & -2.96397 \\ -0.120087 & -5.58562 & 2.62009 & -12.2715 & 0.463974 & 2.80646 & -2.96397 & 15.0507 \end{pmatrix}$$

$$\begin{pmatrix} u_1 \\ v_1 \\ u_4 \\ v_4 \\ u_6 \\ v_6 \\ u_2 \\ v_2 \end{pmatrix} = \begin{pmatrix} 0. \\ 0. \\ 0. \\ 0. \\ -2000. \\ 0. \\ -2000. \\ -2000. \end{pmatrix}$$

The element contributes to {1, 2, 7, 8, 11, 12, 3, 4} global degrees of freedom.

Locations for element contributions to a global vector: 2
7
8
11
12
3
4

and to a global matrix: 
$$\begin{bmatrix} [1,\ 1] & [1,\ 2] & [1,\ 7] & [1,\ 8] & [1,\ 11] & [1,\ 12] & [1,\ 3] & [1,\ 4] \\ [2,\ 1] & [2,\ 2] & [2,\ 7] & [2,\ 8] & [2,\ 11] & [2,\ 12] & [2,\ 3] & [2,\ 4] \\ [7,\ 1] & [7,\ 2] & [7,\ 7] & [7,\ 8] & [7,\ 11] & [7,\ 12] & [7,\ 3] & [7,\ 4] \\ [8,\ 1] & [8,\ 2] & [8,\ 7] & [8,\ 8] & [8,\ 11] & [8,\ 12] & [8,\ 3] & [8,\ 4] \\ [11,\ 1] & [11,\ 2] & [11,\ 7] & [11,\ 8] & [11,\ 11] & [11,\ 12] & [11,\ 3] & [11,\ 4] \\ [12,\ 1] & [12,\ 2] & [12,\ 7] & [12,\ 8] & [12,\ 11] & [12,\ 12] & [12,\ 3] & [12,\ 4] \\ [3,\ 1] & [3,\ 2] & [3,\ 7] & [3,\ 8] & [3,\ 11] & [3,\ 12] & [3,\ 3] & [4,\ 4] \\ [4,\ 1] & [4,\ 2] & [4,\ 7] & [4,\ 8] & [4,\ 11] & [4,\ 12] & [4,\ 3] & [4,\ 4] \\ \end{bmatrix}$$

Adding element equations into appropriate locations we have

### Equations for element 2

$$E = 3000000;$$
  $\vee = 0.2;$   $h = 4$ 

Nodal coordinates

Element node	Global node number	X	y
1	3	6.	0.
2	5	20.	0.
3	6	20.	12.
4	4	6.	5.

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\ N_2,\ N_3,\ N_4\} &= \Big\{\frac{1}{4}\,(s-1)\,(t-1),\ -\frac{1}{4}\,(s+1)\,(t-1),\ \frac{1}{4}\,(s+1)\,(t+1),\ -\frac{1}{4}\,(s-1)\,(t+1)\Big\} \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{\frac{t-1}{4}\,,\ \frac{1-t}{4}\,,\ \frac{t+1}{4}\,,\ \frac{1}{4}\,(-t-1)\Big\} \\ \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} &= \Big\{\frac{s-1}{4}\,,\ \frac{1}{4}\,(-s-1),\ \frac{s+1}{4}\,,\ \frac{1-s}{4}\,\Big\} \end{split}$$

Mapping to the master element

$$\mathbf{x}(\mathbf{s}, t) = \mathbf{N}^{T} \mathbf{x}_{n} = 7. \, \mathbf{s} + 13.$$
 
$$\mathbf{y}(\mathbf{s}, t) = \mathbf{N}^{T} \mathbf{y}_{n} = 1.75 \, t \, \mathbf{s} + 1.75 \, \mathbf{s} + 4.25 \, t + 4.25$$

$$\textbf{\textit{J}} = \left( \begin{array}{ccc} 7. & 0 \\ 1.75 \ t + 1.75 & 1.75 \ s + 4.25 \end{array} \right) ; \hspace{1cm} det \textbf{\textit{J}} = 12.25 \ s + 29.75$$

Plane stress 
$$C = \begin{pmatrix} 3.125 \times 10^6 & 625000. & 0 \\ 625000. & 3.125 \times 10^6 & 0 \\ 0 & 0 & 1.25 \times 10^6 \end{pmatrix}$$

For numerical integration the Gauss quadrature points and weights are

Computation of element matrices at  $\{-0.57735, -0.57735\}$  with weight = 1.

$$\textbf{\textit{J}} = \left( \begin{array}{cc} 7. & 0 \\ 0.739637 & 3.23964 \end{array} \right) \hspace{1cm} det \textbf{\textit{J}} = 22.6775$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.622008,\ 0.166667,\ 0.0446582,\ 0.166667\}$ 

 $\{\partial N_1/\partial s, \partial N_2/\partial s, \partial N_3/\partial s, \partial N_4/\partial s\} = \{-0.394338, 0.394338, 0.105662, -0.105662\}$ 

 $\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \{-0.394338,\; -0.105662,\; 0.105662,\; 0.394338\}$ 

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.0434724,\ 0.0597802,\ 0.0116484,\ -0.0279562\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\}=\{-0.121723,\ -0.0326155,\ 0.0326155,\ 0.121723\}$ 

Computation of element matrices at  $\{-0.57735, 0.57735\}$  with weight = 1.

$$J = \begin{pmatrix} 7. & 0 \\ 2.76036 & 3.23964 \end{pmatrix}$$
 det  $J = 22.6775$ 

$$\begin{array}{c} (\partial N_1/\partial s, \ \partial N_2/\partial s, \ \partial N_3/\partial s, \ \partial N_4/\partial s) = \{-0.105662, \ 0.105662, \ 0.394338, \ -0.394338\} \\ (\partial N_1/\partial t, \ \partial N_2/\partial t, \ \partial N_3/\partial t, \ \partial N_4/\partial t) = \{-0.394338, \ -0.105662, \ 0.105662, \ 0.394338\} \\ (\partial N_1/\partial t, \ \partial N_2/\partial t, \ \partial N_3/\partial t, \ \partial N_4/\partial t) = \{-0.329052, \ 0.0279562, \ 0.0434724, \ -0.104334\} \\ (\partial N_1/\partial t, \ \partial N_2/\partial t, \ \partial N_3/\partial t, \ \partial N_4/\partial t) = \{-0.121723, \ -0.0326155, \ 0.0326155, \ 0.121723\} \\ \textbf{B}^T = \\ \begin{pmatrix} 0.0329052 & 0 & 0.0279562 & 0 & 0.0434724 & 0 & -0.104334 & 0 \\ 0 & -0.121723 & 0 & -0.0326155 & 0 & 0.0326155 & 0 & 0.121723 \\ -0.121723 & 0.0329052 & -0.0326155 & 0.0279562 & 0.0326155 & 0.0434724 & 0.121723 & -0.104334 \end{pmatrix} \\ \textbf{k} = 10^6 \\ \begin{pmatrix} 1.98692 & -0.681228 & 0.710917 & -0.446691 & -0.0446606 & -0.539153 & -2.65318 & 1.66707 \\ -0.681228 & 4.32276 & -0.314612 & 1.22969 & -0.178309 & -0.963186 & 1.17415 & -4.58926 \\ 0.710917 & -0.314612 & 0.342162 & -0.155081 & 0.223887 & -0.109076 & -1.27697 & 0.578769 \\ -0.446691 & 1.22969 & -0.155081 & 0.390163 & 0.0230025 & -0.163744 & 0.578769 & -1.45611 \\ -0.0446606 & -0.178309 & 0.223887 & 0.0230025 & 0.656331 & 0.241154 & -0.835557 & -0.0858466 \\ -0.539153 & -0.963186 & -0.109076 & -0.163744 & 0.241154 & 0.515831 & 0.407075 & 0.611099 \\ -2.65318 & 1.17415 & -1.27697 & 0.578769 & -0.835557 & 0.407075 & 4.7657 & -2.15999 \\ 1.66707 & -4.58926 & 0.578769 & -1.45611 & -0.0858466 & 0.611099 & -2.15999 & 5.43427 \end{pmatrix} \\ \text{Computation of element matrices at } \{0.57735, -0.57735\} \text{ with weight } = 1. \end{cases}$$

$$J = \begin{pmatrix} 7 & 0 & 0 & 0.066667, 0.622008, 0.166667, 0.0446582\} \\ \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.05662, -0.394338, 0.394338, 0.105662, -0.105662\} \\ \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0542115, 0.0642548, 0.00717376, -0.017217\} \\ \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0542145, 0.0642548, 0.00717376, 0 & -0.017217\} \\ \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0542148, 0.00717376, 0 & -0.017217\} \\ \{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0542148, 0.00717376, 0 & -0.017217\} \\ \{\partial N_1/\partial x, \partial N_$$

 $0.0642548 \quad 0.0749639$ 

0.00717376

0.0200865 -

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.166667,\ 0.0446582,\ 0.166667,\ 0.622008\}$ 

 $-0.0200865 \quad -0.0542115 \quad -0.0749639$ 

Computation of element matrices at  $\{0.57735, 0.57735\}$  with weight = 1.

$$\mathbf{J} = \begin{pmatrix} 7. & 0 \\ 2.76036 & 5.26036 \end{pmatrix} \qquad \qquad \det \mathbf{J} = 36.8225$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.0446582,\ 0.166667,\ 0.622008,\ 0.166667\}$ 

 $\{\partial N_1/\partial s, \, \partial N_2/\partial s, \, \partial N_3/\partial s, \, \partial N_4/\partial s\} = \{-0.105662, \, 0.105662, \, 0.394338, \, -0.394338\}$ 

 $\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \{-0.105662,\; -0.394338,\; 0.394338,\; 0.105662\}$ 

 $\{\partial N_1/\partial x, \, \partial N_2/\partial x, \, \partial N_3/\partial x, \, \partial N_4/\partial x\} = \{-0.00717376, \, 0.0446557, \, 0.0267728, \, -0.0642548\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0200865,\ -0.0749639,\ 0.0749639,\ 0.0200865\}$ 

$$\mathbf{k} = 10^6 \begin{bmatrix} -0.00717376 & 0 & 0.0446557 & 0 & 0.0267728 & 0 & -0.0642548 \\ 0 & -0.0200865 & 0 & -0.0749639 & 0 & 0.0749639 & 0 \\ -0.0200865 & -0.00717376 & -0.0749639 & 0.0446557 & 0.0749639 & 0.0267728 & 0.0200865 & -0.0979711 & 0.0397948 & 0.12978 & -0.11564 & -0.365633 & -0.148516 & 0.137883 \\ 0.0397948 & 0.195184 & 0.0164383 & 0.634096 & -0.148516 & -0.728437 & 0.0922832 & 0.12978 & 0.0164383 & 1.9525 & -0.924495 & -0.484344 & -0.0613487 & -1.59794 \\ -0.11564 & 0.634096 & -0.924495 & 2.95374 & 0.431573 & -2.36648 & 0.608562 & -0.365633 & -0.148516 & -0.484344 & 0.431573 & 1.36456 & 0.554271 & -0.514585 & -0.148516 & -0.728437 & -0.0613487 & -2.36648 & 0.554271 & 2.71857 & -0.344406 & 0.137883 & 0.0922832 & -1.59794 & 0.608562 & -0.514585 & -0.344406 & 1.97464 & 0.224361 & -0.100843 & 0.969406 & -1.22136 & -0.837328 & 0.37635 & -0.356439 \end{bmatrix}$$

Summing contributions from all points we get

$$\mathbf{k} = 10^6 \begin{bmatrix} 5.7276 & 0.559291 & -0.771918 & -1.17054 & -1.46023 & -1.32946 & -3.49546 & 1.94071 \\ 0.559291 & 9.65901 & 0.0794621 & 2.74625 & -1.32946 & -3.6391 & 0.690709 & -8.76615 \\ -0.771918 & 0.0794621 & 6.3633 & -2.74144 & -1.00616 & 0.241443 & -4.58523 & 2.42054 \\ -1.17054 & 2.74625 & -2.74144 & 7.3974 & 1.49144 & -5.25454 & 2.42054 & -4.8891 \\ -1.46023 & -1.32946 & -1.00616 & 1.49144 & 3.2383 & 1.00856 & -0.771918 & -1.17054 \\ -1.32946 & -3.6391 & 0.241443 & -5.25454 & 1.00856 & 6.1474 & 0.0794621 & 2.74625 \\ -3.49546 & 0.690709 & -4.58523 & 2.42054 & -0.771918 & 0.0794621 & 8.8526 & -3.19071 \\ 1.94071 & -8.76615 & 2.42054 & -4.8891 & -1.17054 & 2.74625 & -3.19071 & 10.909 \end{bmatrix}$$

 $\mathbf{r}^{\mathrm{T}} = (0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0)$ 

Complete element equations for element 2

The element contributes to  $\{5,\,6,\,9,\,10,\,11,\,12,\,7,\,8\}$  global degrees of freedom.

[5, 5] [5, 6] [5, 9] [5, 10] [5, 11][5, 12][6, 10][6, 11][6, 12][6, 7][6, 8][9, 5] [9, 6] [9, 9] [9, 10] [9, 11] [9, 12][9, 7][9, 8]  $[10,\, 5] \ [10,\, 6] \ [10,\, 9] \ [10,\, 10] \ [10,\, 11] \ [10,\, 12] \ [10,\, 7] \ [10,\, 8]$ and to a global matrix: [11, 5] [11, 6] [11, 9] [11, 10] [11, 11] [11, 12] [11, 7] [11, 8]  $[12,\, 5] \ [12,\, 6] \ [12,\, 9] \ [12,\, 10] \ [12,\, 11] \ [12,\, 12] \ [12,\, 7] \ [12,\, 8]$ [7, 9] [7, 10] [7, 11] [7, 12] [8, 12] [8, 5] [8, 6] [8, 9] [8, 10] [8, 11] [8, 7] [8, 8]

Adding element equations into appropriate locations we have

	· ·	-							
	6.26209	-0.0163755	-1.19113	-0.120087	0	0	-4.11923	1.26638	(
	-0.0163755	9.0354	-1.37009	-5.58562	0	0	2.51638	-3.67826	(
	-1.19113	-1.37009	7.54484	-2.96397	0	0	-5.95173	2.62009	(
	-0.120087	-5.58562	-2.96397	15.0507	0	0	2.62009	-12.2715	(
	0	0	0	0	5.7276	0.559291	-3.49546	1.94071	-(
	0	0	0	0	0.559291	9.65901	0.690709	-8.76615	(
	-4.11923	2.51638	-5.95173	2.62009	-3.49546	0.690709	20.1147	-6.95708	-4
$10^6$	1.26638	-3.67826	2.62009	-12.2715	1.94071	-8.76615	-6.95708	32.4444	4
10	0	0	0	0	-0.771918	0.0794621	-4.58523	2.42054	(
	0	0	0	0	-1.17054	2.74625	2.42054	-4.8891	-:
	-0.951731	-1.12991	-0.401981	0.463974	-1.46023	-1.32946	-1.96304	-1.29063	<b>-</b> [
	-1.12991	0.228478	1.71397	2.80646	-1.32946	-3.6391	-1.29063	-2.83938	(
	0	0	0	0	0	0	0	0	(
	0	0	0	0	0	0	0	0	(
	0	0	0	0	0	0	0	0	(
	0	0	0	0	0	0	0	0	(

# Equations for element 3

E = 3000000;  $\vee = 0.2;$  h = 4

Nodal coordinates

Element node	Global node number	X	y
1	5	20.	0.
2	7	54.	0.
3	8	54.	12.
4	6	20.	12.

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\ N_2,\ N_3,\ N_4\} &= \Big\{\frac{1}{4}\,(s-1)\,(t-1),\ -\frac{1}{4}\,(s+1)\,(t-1),\ \frac{1}{4}\,(s+1)\,(t+1),\ -\frac{1}{4}\,(s-1)\,(t+1)\Big\} \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{\frac{t-1}{4},\ \frac{1-t}{4},\ \frac{t+1}{4},\ \frac{1}{4}\,(-t-1)\Big\} \\ \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} &= \Big\{\frac{s-1}{4},\ \frac{1}{4}\,(-s-1),\ \frac{s+1}{4},\ \frac{1-s}{4}\Big\} \end{split}$$

Mapping to the master element

$$\mathbf{x}(\mathbf{s}, \mathbf{t}) = \mathbf{N}^{T} \mathbf{x}_{n} = 17. \, \mathbf{s} + 37.$$

$$\mathbf{y}(\mathbf{s}, \mathbf{t}) = \mathbf{N}^{T} \mathbf{y}_{n} = 6. \, \mathbf{t} + 6.$$

$$\mathbf{J} = \begin{pmatrix} 17. & 0 \\ 0 & 6. \end{pmatrix}; \qquad \det \mathbf{J} = 102.$$
Plane stress  $\mathbf{C} = \begin{pmatrix} 3.125 \times 10^{6} & 625000. & 0 \\ 625000. & 3.125 \times 10^{6} & 0 \\ 0 & 0 & 1.25 \times 10^{6} \end{pmatrix}$ 

For numerical integration the Gauss quadrature points and weights are

Computation of element matrices at  $\{-0.57735, -0.57735\}$  with weight = 1.

$$\boldsymbol{J} = \begin{pmatrix} 17. & 0 \\ 0 & 6. \end{pmatrix} \qquad \text{det } J = 102.$$

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.622008,\ 0.166667,\ 0.0446582,\ 0.166667\}$$

$$\{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} = \{-0.394338,\ 0.394338,\ 0.105662,\ -0.105662\}$$

$$\{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} = \{-0.394338,\ -0.105662,\ 0.105662,\ 0.394338\}$$

$$\{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0231963, 0.0231963, 0.00621544, -0.00621544\}$$

$$\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0657229,\ -0.0176104,\ 0.0176104,\ 0.0657229\}$$

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.0231963 & 0 & 0.0231963 & 0 & 0.00621544 & 0 & -0.00621544 \\ 0 & -0.0657229 & 0 & -0.0176104 & 0 & 0.0176104 & 0 \\ -0.0657229 & -0.0231963 & -0.0176104 & 0.0231963 & 0.0176104 & 0.00621544 & 0.0657229 \end{pmatrix}$$

$$k = 10^6$$

$$\begin{pmatrix} 2.88899 & 1.16627 & -0.095761 & -0.673344 & -0.774101 & -0.3125 & -2.01912 & -0.180422 \\ 1.16627 & 5.78178 & -0.180422 & 1.20128 & -0.3125 & -1.54922 & -0.673344 & -5.43384 \\ -0.095761 & -0.180422 & 0.844203 & -0.3125 & 0.0256591 & 0.0483439 & -0.774101 & 0.444578 \\ -0.673344 & 1.20128 & -0.3125 & 0.669827 & 0.180422 & -0.321882 & 0.805422 & -1.54922 \\ -0.774101 & -0.3125 & 0.0256591 & 0.180422 & 0.20742 & 0.0837341 & 0.541022 & 0.0483439 \\ -0.3125 & -1.54922 & 0.0483439 & -0.321882 & 0.0837341 & 0.415113 & 0.180422 & 1.45599 \\ -2.01912 & -0.673344 & -0.774101 & 0.805422 & 0.541022 & 0.180422 & 2.2522 & -0.3125 \\ -0.180422 & -5.43384 & 0.444578 & -1.54922 & 0.0483439 & 1.45599 & -0.3125 & 5.52707 \end{pmatrix}$$

Computation of element matrices at  $\{-0.57735, 0.57735\}$  with weight = 1.

$$\mathbf{J} = \begin{pmatrix} 17. & 0 \\ 0 & 6. \end{pmatrix} \qquad \text{detJ} = 102.$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.166667,\ 0.0446582,\ 0.166667,\ 0.622008\}$ 

 $\{\partial N_1/\partial s, \partial N_2/\partial s, \partial N_3/\partial s, \partial N_4/\partial s\} = \{-0.105662, 0.105662, 0.394338, -0.394338\}$ 

 $\{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} = \{-0.394338,\ -0.105662,\ 0.105662,\ 0.394338\}$ 

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.00621544,\ 0.00621544,\ 0.0231963,\ -0.0231963\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0657229,\ -0.0176104,\ 0.0176104,\ 0.0657229\}$ 

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.00621544 & 0 & 0.00621544 & 0 & 0.0231963 & 0 & -0.0231963 \\ 0 & -0.0657229 & 0 & -0.0176104 & 0 & 0.0176104 & 0 \\ -0.0657229 & -0.00621544 & -0.0176104 & 0.00621544 & 0.0176104 & 0.0231963 & 0.0657229 \end{pmatrix}$$

$$k = 10^6$$

Computation of element matrices at  $\{0.57735, -0.57735\}$  with weight = 1.

$$\textbf{\textit{J}} = \left( \begin{array}{cc} 17. & 0 \\ 0 & 6. \end{array} \right) \hspace{1cm} det \textbf{\textit{J}} = 102.$$

 $\{N_1, N_2, N_3, N_4\} = \{0.166667, 0.622008, 0.166667, 0.0446582\}$ 

$$\begin{split} & \{\partial N_1/\partial s,\,\partial N_2/\partial s,\,\partial N_3/\partial s,\,\partial N_4/\partial s\} = \{-0.394338,\,0.394338,\,0.105662,\,-0.105662\} \\ & \{\partial N_1/\partial t,\,\partial N_2/\partial t,\,\partial N_3/\partial t,\,\partial N_4/\partial t\} = \{-0.105662,\,-0.394338,\,0.394338,\,0.105662\} \\ & \{\partial N_1/\partial x,\,\partial N_2/\partial x,\,\partial N_3/\partial x,\,\partial N_4/\partial x\} = \{-0.0231963,\,0.0231963,\,0.00621544,\,-0.00621544\} \\ & \{\partial N_1/\partial y,\,\partial N_2/\partial y,\,\partial N_3/\partial y,\,\partial N_4/\partial y\} = \{-0.0176104,\,-0.0657229,\,0.0657229,\,0.0176104\} \\ & (-0.0231963 \quad 0 \quad 0.0231963 \quad 0 \quad 0.00621544 \quad 0 \quad -0.00621544) \end{split}$$

$${\pmb B}^{\rm T} = \begin{pmatrix} -0.0231963 & 0 & 0.0231963 & 0 & 0.00621544 & 0 & -0.00621544 \\ 0 & -0.0176104 & 0 & -0.0657229 & 0 & 0.0657229 & 0 \\ -0.0176104 & -0.0231963 & -0.0657229 & 0.0231963 & 0.0657229 & 0.00621544 & 0.0176104 \\ {\pmb k} = 10^6 \\ \end{cases}$$

Computation of element matrices at  $\{0.57735, 0.57735\}$  with weight = 1.

$$\boldsymbol{J} = \begin{pmatrix} 17. & 0 \\ 0 & 6. \end{pmatrix} \qquad \text{det } J = 102.$$

 $\{N_1,\ N_2,\ N_3,\ N_4\}=\{0.0446582,\ 0.166667,\ 0.622008,\ 0.166667\}$ 

 $\{\partial N_1/\partial s, \partial N_2/\partial s, \partial N_3/\partial s, \partial N_4/\partial s\} = \{-0.105662, 0.105662, 0.394338, -0.394338\}$ 

 $\{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} = \{-0.105662,\ -0.394338,\ 0.394338,\ 0.105662\}$ 

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.00621544,\ 0.00621544,\ 0.0231963,\ -0.0231963\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0176104,\ -0.0657229,\ 0.0657229,\ 0.0176104\}$ 

$$\boldsymbol{B}^T = \begin{pmatrix} -0.00621544 & 0 & 0.00621544 & 0 & 0.0231963 & 0 & -0.0231963 \\ 0 & -0.0176104 & 0 & -0.0657229 & 0 & 0.0657229 & 0 \\ -0.0176104 & -0.00621544 & -0.0657229 & 0.00621544 & 0.0657229 & 0.0231963 & 0.0176104 \end{pmatrix}$$

$$k = 10^6$$

$$\begin{pmatrix} 0.20742 & 0.0837341 & 0.541022 & 0.0483439 & -0.774101 & -0.3125 & 0.0256591 & 0.180422 \\ 0.0837341 & 0.415113 & 0.180422 & 1.45599 & -0.3125 & -1.54922 & 0.0483439 & -0.321882 \\ 0.541022 & 0.180422 & 2.2522 & -0.3125 & -2.01912 & -0.673344 & -0.774101 & 0.805422 \\ 0.0483439 & 1.45599 & -0.3125 & 5.52707 & -0.180422 & -5.43384 & 0.444578 & -1.54922 \\ -0.774101 & -0.3125 & -2.01912 & -0.180422 & 2.88899 & 1.16627 & -0.095761 & -0.673344 \\ -0.3125 & -1.54922 & -0.673344 & -5.43384 & 1.16627 & 5.78178 & -0.180422 & 1.20128 \\ 0.0256591 & 0.0483439 & -0.774101 & 0.444578 & -0.095761 & -0.180422 & 0.844203 & -0.3125 \\ 0.180422 & -0.321882 & 0.805422 & -1.54922 & -0.673344 & 1.20128 & -0.3125 & 0.669827 \\ \end{pmatrix}$$

Summing contributions from all points we get

$$\mathbf{k} = 10^6 \begin{bmatrix} 6.19281 & 1.875 & 0.890523 & -0.625 & -3.09641 & -1.875 & -3.98693 & 0.625 \\ 1.875 & 12.3938 & 0.625 & 5.31454 & -1.875 & -6.1969 & -0.625 & -11.5114 \\ 0.890523 & 0.625 & 6.19281 & -1.875 & -3.98693 & -0.625 & -3.09641 & 1.875 \\ -0.625 & 5.31454 & -1.875 & 12.3938 & 0.625 & -11.5114 & 1.875 & -6.1969 \\ -3.09641 & -1.875 & -3.98693 & 0.625 & 6.19281 & 1.875 & 0.890523 & -0.625 \\ -1.875 & -6.1969 & -0.625 & -11.5114 & 1.875 & 12.3938 & 0.625 & 5.31454 \\ -3.98693 & -0.625 & -3.09641 & 1.875 & 0.890523 & 0.625 & 6.19281 & -1.875 \\ 0.625 & -11.5114 & 1.875 & -6.1969 & -0.625 & 5.31454 & -1.875 & 12.3938 \end{bmatrix}$$

$$\mathbf{r}^{\mathrm{T}} = (0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0)$$

Computation of element matrices resulting from NBC

NBC on side 3 with 
$$\{q_n, q_t\} = \{-50, 0\}$$

$$\left\{ N_{1},\;N_{2},\;N_{3},\;N_{4}\right\} _{c}=\left\{ 0,\;0,\;\frac{1-a}{2},\;\frac{a+1}{2}\right\}$$

$$x(a) = 37. - 17. a;$$

$$y(a) = 12.$$

$$dx/da = -17.;$$

$$dy/da = 0$$
.

$$J_c = 17$$
.

Gauss point = 
$$-0.57735$$
; Weight = 1.;  $J_c = 17$ .

$$\left\{ N_{1},\;N_{2},\;N_{3},\;N_{4}\right\} _{c}=\left\{ 0,\;0,\;0.788675,\;0.211325\right\}$$

$$\mathbf{r}_{q}^{T} = (0 \ 0 \ 0 \ 0 \ 0 \ -2681.5 \ 0 \ -718.505)$$

Gauss point = 
$$0.57735$$
; Weight = 1.;  $J_c = 17$ .

$$\{N_1,\ N_2,\ N_3,\ N_4\}_c=\{0,\ 0,\ 0.211325,\ 0.788675\}$$

$$\mathbf{r}_{q}^{T} = (0 \ 0 \ 0 \ 0 \ 0 \ -718.505 \ 0 \ -2681.5)$$

Summing contributions from all Gauss points

$$\mathbf{r}_{q}^{T} = (0 \ 0 \ 0 \ 0 \ 0 \ -3400. \ 0 \ -3400.)$$

Complete element equations for element 3

$$10^{6} \begin{pmatrix} 6.19281 & 1.875 & 0.890523 & -0.625 & -3.09641 & -1.875 & -3.98693 & 0.625 \\ 1.875 & 12.3938 & 0.625 & 5.31454 & -1.875 & -6.1969 & -0.625 & -11.5114 \\ 0.890523 & 0.625 & 6.19281 & -1.875 & -3.98693 & -0.625 & -3.09641 & 1.875 \\ -0.625 & 5.31454 & -1.875 & 12.3938 & 0.625 & -11.5114 & 1.875 & -6.1969 \\ -3.09641 & -1.875 & -3.98693 & 0.625 & 6.19281 & 1.875 & 0.890523 & -0.625 \\ -1.875 & -6.1969 & -0.625 & -11.5114 & 1.875 & 12.3938 & 0.625 & 5.31454 \\ -3.98693 & -0.625 & -3.09641 & 1.875 & 0.890523 & 0.625 & 6.19281 & -1.875 \\ 0.625 & -11.5114 & 1.875 & -6.1969 & -0.625 & 5.31454 & -1.875 & 12.3938 \end{pmatrix}$$

$$\begin{pmatrix} u_5 \\ v_5 \\ u_7 \\ v_7 \\ u_8 \\ v_8 \\ u_6 \\ v_6 \end{pmatrix} = \begin{pmatrix} 0. \\ 0. \\ 0. \\ 0. \\ -3400. \\ 0. \\ -3400. \\ 0. \\ -3400. \end{pmatrix}$$

The element contributes to {9, 10, 13, 14, 15, 16, 11, 12} global degrees of freedom.

# and to a global matrix:

# Adding element equations into appropriate locations we have

1	6.26209	-0.0163755	-1.19113	-0.120087	0	0	-4.11923	1.26638	(
	-0.0163755	9.0354	-1.37009	-5.58562	0	0	2.51638	-3.67826	(
	-1.19113	-1.37009	7.54484	-2.96397	0	0	-5.95173	2.62009	(
	-0.120087	-5.58562	-2.96397	15.0507	0	0	2.62009	-12.2715	(
	0	0	0	0	5.7276	0.559291	-3.49546	1.94071	-(
	0	0	0	0	0.559291	9.65901	0.690709	-8.76615	(
	-4.11923	2.51638	-5.95173	2.62009	-3.49546	0.690709	20.1147	-6.95708	-4
10 <sup>6</sup>	1.26638	-3.67826	2.62009	-12.2715	1.94071	-8.76615	-6.95708	32.4444	1
10	0	0	0	0	-0.771918	0.0794621	-4.58523	2.42054	1;
	0	0	0	0	-1.17054	2.74625	2.42054	-4.8891	-(
	-0.951731	-1.12991	-0.401981	0.463974	-1.46023	-1.32946	-1.96304	-1.29063	_4
	-1.12991	0.228478	1.71397	2.80646	-1.32946	-3.6391	-1.29063	-2.83938	(
	0	0	0	0	0	0	0	0	(
	0	0	0	0	0	0	0	0	-(
	0	0	0	0	0	0	0	0	-5
1	0	0	0	0	0	0	0	0	<b>-</b> [

# Essential boundary conditions

Node	dof	Valu
1	$\mathbf{u}_1$	0
2	$\mathbf{u}_2$	0
7	$\mathbf{u}_7$	0
,	$\mathbf{v}_7$	0
8	$u_8$	0
U	${ m u_8} \ { m v_8}$	0

Remove {1, 3, 13, 14, 15, 16} rows and columns.

After adjusting for essential boundary conditions we have

```
9.0354
                         0
                                    0
            -5.58562
                                                2.51638
                                                           -3.67826
                                                                       0
                                                                                     0
-5.58562
            15.0507
                         0
                                    0
                                                2.62009
                                                          -12.2715
                                                                       0
                                                                                     0
                                                                                                (
 0
                                    0.559291
                                                             1.94071 - 0.771918
                                                                                    -1.17054
             0
                         5.7276
                                               -3.49546
 0
             0
                         0.559291
                                    9.65901
                                                0.690709
                                                           -8.76615
                                                                       0.0794621
                                                                                     2.74625
 2.51638
             2.62009
                       -3.49546
                                    0.690709
                                               20.1147
                                                           -6.95708 -4.58523
                                                                                     2.42054
-3.67826
           -12.2715
                         1.94071
                                   -8.76615
                                               -6.95708
                                                            32.4444
                                                                       2.42054
                                                                                   -4.8891
 0
             0
                       -0.771918
                                    0.0794621 \ -4.58523
                                                             2.42054 \quad 12.5561
                                                                                   -0.866443 -4
 0
             0
                       -1.17054
                                    2.74625
                                                2.42054
                                                           -4.8891
                                                                      -0.866443
                                                                                    19.7912
                                                                                                (
-1.12991
             0.463974 - 1.46023
                                   -1.32946
                                               -1.96304
                                                           -1.29063 -4.99308
                                                                                     0.866443
                                                                                               11
 0.228478
             2.80646
                       -1.32946
                                  -3.6391
                                               -1.29063
                                                           -2.83938
                                                                       0.866443
                                                                                  -16.766
                                                                                               -(
```

Solving the final system of global equations we get

```
 \begin{aligned} \{v_1 = -0.0183155, \ v_2 = -0.0183204, \ u_3 = 0.00275915, \ v_3 = -0.0166486, \ u_4 = 0.00114552, \\ v_4 = -0.0164634, \ u_5 = 0.00305003, \ v_5 = -0.0113566, \ u_6 = -0.00210128, \ v_6 = -0.0116254\} \end{aligned}
```

# Complete table of nodal values

	u	v
1	0	-0.0183155
2	0	-0.0183204
3	0.00275915	-0.0166486
4	0.00114552	-0.0164634
5	0.00305003	-0.0113566
6	-0.00210128	-0.0116254
7	0	0
8	0	0

## Computation of reactions

Equation numbers of dof with specified values: {1, 3, 13, 14, 15, 16}

Extracting equations {1, 3, 13, 14, 15, 16} from the global system we have

Substituting the nodal values and re-arranging

Carrying out computations, the reactions are as follows.

Label	dof	Reaction
$R_1$	$\mathbf{u}_1$	-7931.98
$R_2$	$\mathbf{u_2}$	10360.8
$R_3$	$\mathbf{u}_7$	-19673.
$R_4$	$\mathbf{v}_7$	5839.98
$R_5$	$u_8$	17244.2
$R_6$	$v_8$	4960.02

### Sum of Reactions

dof: u 0 dof: v 10800.

## Solution for element 1

### Element nodal displacements

Element node	Global node number	u	$\mathbf{v}$
1	1	0	-0.0183155
2	4	0.00114552	-0.0164634
3	6	-0.00210128	-0.0116254
4	2	0	-0.0183204

$$\boldsymbol{d}^{T} = (0 \quad -0.0183155 \quad 0.00114552 \quad -0.0164634 \quad -0.00210128 \quad -0.0116254 \quad 0 \quad -0.0183204)$$

$$E = 3000000;$$

Plane stress 
$$C = \begin{pmatrix} 3.125 \times 10^6 & 625000. & 0 \\ 625000. & 3.125 \times 10^6 & 0 \\ 0 & 0 & 1.25 \times 10^6 \end{pmatrix}$$

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\ N_2,\ N_3,\ N_4\} &= \Big\{\frac{1}{4}\,(s-1)\,(t-1),\ -\frac{1}{4}\,(s+1)\,(t-1),\ \frac{1}{4}\,(s+1)\,(t+1),\ -\frac{1}{4}\,(s-1)\,(t+1)\Big\} \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{\frac{t-1}{4},\ \frac{1-t}{4},\ \frac{t+1}{4},\ \frac{1}{4}\,(-t-1)\Big\} \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{\frac{s-1}{4},\ \frac{1}{4}\,(-s-1),\ \frac{s+1}{4},\ \frac{1-s}{4}\Big\} \end{split}$$

Nodal coordinates

Element node	Global node number	X	y
1	1	0.	5.
2	4	6.	5.
3	6	20.	12.
4	2	0.	12.

Mapping to the master element

$$\begin{split} x(s,t) &= 1.5 \ (s+1) \ (1-t) + 5. \ (s+1) \ (t+1) \\ y(s,t) &= 1.25 \ (1-s) \ (1-t) + 1.25 \ (s+1) \ (1-t) + 3. \ (1-s) \ (t+1) + 3. \ (s+1) \ (t+1) \\ J &= \begin{pmatrix} 1.5 \ (1-t) + 5. \ (t+1) & 3.5 \ (s+1) \\ 0 & 1.75 \ (1-s) + 1.75 \ (s+1) \end{pmatrix}; \\ \text{detJ} &= 12.25 \ t + 22.75 \end{split}$$

Element solution at  $\{s, t\} = \{0, 0\} \Longrightarrow \{x, y\} = \{6.5, 8.5\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\} = \Big\{\frac{1}{4},\ \frac{1}{4},\ \frac{1}{4},\ \frac{1}{4}\Big\}$$

$$\{\partial N_1/\partial s,\; \partial N_2/\partial s,\; \partial N_3/\partial s,\; \partial N_4/\partial s\} = \left\{-\frac{1}{4},\; \frac{1}{4},\; \frac{1}{4},\; -\frac{1}{4}\right\}$$

$$\{\partial N_1/\partial t,\;\partial N_2/\partial t,\;\partial N_3/\partial t,\;\partial N_4/\partial t\} = \left\{-\frac{1}{4},\;-\frac{1}{4},\;\frac{1}{4},\;\frac{1}{4}\right\}$$

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.0384615,\ 0.0384615,\ 0.0384615,\ -0.0384615\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.032967,\ -0.10989,\ 0.032967,\ 0.10989\}$ 

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.00003676 \ 0.0000164861 \ 0.000133582)$ 

In-plane stress components, $\sigma = C\epsilon = (-104.571 \ 28.544 \ 166.978)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = (-0.00003676 \ 0.0000164861 \ 5.06848 \times 10^{-6} \ 0.000133582 \ 0 \ 0)$$

 $\sigma^{\mathrm{T}} = (-104.571 \ 28.544 \ 0 \ 166.978 \ 0 \ 0)$ 

Substituting these stress components into appropriate formulas

Principal stresses =  $(141.741 \ 0. \ -217.768)$ 

Effective stress (von Mises) = 313.656

Element solution at  $\{s, t\} = \{-1, -1\} \Longrightarrow \{x, y\} = \{0, 5, 5\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{1, 0, 0, 0\}$$

$$\left\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\right\} = \left\{-\frac{1}{2},\;\frac{1}{2},\;0,\;0\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{-\frac{1}{2},\; 0,\; 0,\; \frac{1}{2}\right\}$$

$$\{\partial N_1/\partial x,\; \partial N_2/\partial x,\; \partial N_3/\partial x,\; \partial N_4/\partial x\} = \{-0.166667,\; 0.166667,\; 0,\; 0.\}$$

$$\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.142857,\ 0.,\ 0,\ 0.142857\}$$

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.166667 & 0 & 0.166667 & 0 & 0 & 0 & 0 \\ 0 & -0.142857 & 0 & 0 & 0 & 0 & 0.142857 \\ -0.142857 & -0.166667 & 0 & 0.166667 & 0 & 0 & 0.142857 & 0 \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{\mathrm{T}} \mathbf{d} = (0.00019092 - 6.97447 \times 10^{-7} 0.000308689)$ 

In-plane stress components, $\sigma = C\epsilon = (596.188 \ 117.145 \ 385.861)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = (\ 0.00019092 \ \ -6.97447 \times 10^{-7} \ \ \ -0.0000475555 \ \ 0.000308689 \ \ 0 \ \ 0 \ )$$

$$\sigma^{\mathrm{T}} = (596.188 \ 117.145 \ 0 \ 385.861 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(810.824 \ 0. \ -97.4913)$ 

Effective stress (von Mises) = 863.706

$$\{N_1, N_2, N_3, N_4\} = \{0, 0, 0, 1\}$$

$$\{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} = \left\{0,\ 0,\ \frac{1}{2},\ -\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\;\partial N_2/\partial t,\;\partial N_3/\partial t,\;\partial N_4/\partial t\}=\left\{-\,\frac{1}{2}\,,\;0,\;0,\;\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{0., 0, 0.05, -0.05\}$$

$$\{\partial N_1/\partial y, \, \partial N_2/\partial y, \, \partial N_3/\partial y, \, \partial N_4/\partial y\} = \{-0.142857, \, 0, \, 0., \, 0.142857\}$$

In–plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.000105064 -6.97447 \times 10^{-7} 0.000334751)$ 

In-plane stress components,  $\sigma = C\epsilon = (-328.76 -67.8444 \ 418.438)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{T} = (-0.000105064 -6.97447 \times 10^{-7} \ 0.0000264403 \ 0.000334751 \ 0 \ 0)$$

$$\sigma^{T} = (-328.76 -67.8444 \ 0 \ 418.438 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(240.001 \ 0. \ -636.606)$ 

Effective stress (von Mises) = 784.636

Element solution at  $\{s, t\} = \{1, -1\} \Longrightarrow \{x, y\} = \{6, 5, 5\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{0,\ 1,\ 0,\ 0\}$$

$$\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\}=\left\{-\,\frac{1}{2},\;\frac{1}{2},\;0,\;0\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{0,\; -\frac{1}{2},\; \frac{1}{2},\; 0\right\}$$

$$\{\partial N_1/\partial x, \, \partial N_2/\partial x, \, \partial N_3/\partial x, \, \partial N_4/\partial x\} = \{-0.166667, \, 0.166667, \, 0., \, 0\}$$

$$\{\partial N_1/\partial y, \partial N_2/\partial y, \partial N_3/\partial y, \partial N_4/\partial y\} = \{0.333333, -0.47619, 0.142857, 0\}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (0.00019092 \ 0.0000737645 \ -0.000536978)$ 

In-plane stress components,  $\sigma = C\epsilon = (642.726 \ 349.839 \ -671.222)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{\text{T}} = (0.00019092 \ 0.0000737645 \ -0.000066171 \ -0.000536978 \ 0 \ 0)$$

$$\sigma^{\text{T}} = (642.726 \ 349.839 \ 0 \ -671.222 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(1183.29 \ 0. \ -190.729)$ 

Effective stress (von Mises) = 1289.28

Element solution at  $\{s, t\} = \{1, 1\} \Longrightarrow \{x, y\} = \{20, 12.\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{0, 0, 1, 0\}$$

$$\left\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\right\} = \left\{0,\;0,\;\frac{1}{2}\,,\;-\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{0,\; -\frac{1}{2},\; \frac{1}{2},\; 0\right\}$$

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\}=\{0,\ 0.,\ 0.05,\ -0.05\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{0,\ -0.142857,\ 0.0428571,\ 0.1\}$ 

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} 0 & 0 & 0 & 0 & 0.05 & 0 & -0.05 & 0 \\ 0 & 0 & 0 & -0.142857 & 0 & 0.0428571 & 0 & 0.1 \\ 0 & 0 & -0.142857 & 0 & 0.0428571 & 0.05 & 0.1 & -0.05 \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.000105064 \ 0.0000216411 \ 0.0000810505)$ 

In-plane stress components,  $\sigma = C\epsilon = (-314.799 \ 1.96363 \ 101.313)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = ( \ -0.000105064 \quad 0.0000216411 \quad 0.0000208557 \quad 0.0000810505 \quad 0 \quad 0 \ )$$

$$\sigma^{\mathrm{T}} = (-314.799 \ 1.96363 \ 0 \ 101.313 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(31.5956 \ 0. \ -344.431)$ 

Effective stress (von Mises) = 361.266

#### Solution for element 2

Element nodal displacements

Element node	Global node number	u	v
1	3	0.00275915	-0.0166486
2	5	0.00305003	-0.0113566
3	6	-0.00210128	-0.0116254
4	4	0.00114552	-0.0164634

$$E = 3000000;$$
  $\vee = 0.2;$   $h = 4$ 

Plane stress 
$$C = \begin{pmatrix} 3.125 \times 10^6 & 625000. & 0 \\ 625000. & 3.125 \times 10^6 & 0 \\ 0 & 0 & 1.25 \times 10^6 \end{pmatrix}$$

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\,N_2,\,N_3,\,N_4\} &= \Big\{\frac{1}{4}\,(s-1)\,(t-1),\,-\frac{1}{4}\,(s+1)\,(t-1),\,\frac{1}{4}\,(s+1)\,(t+1),\,-\frac{1}{4}\,(s-1)\,(t+1)\Big\} \\ \{\partial N_1/\partial s,\,\partial N_2/\partial s,\,\partial N_3/\partial s,\,\partial N_4/\partial s\} &= \Big\{\frac{t-1}{4},\,\frac{1-t}{4},\,\frac{t+1}{4},\,\frac{1}{4}\,(-t-1)\Big\} \\ \{\partial N_1/\partial s,\,\partial N_2/\partial s,\,\partial N_3/\partial s,\,\partial N_4/\partial s\} &= \Big\{\frac{s-1}{4},\,\frac{1}{4}\,(-s-1),\,\frac{s+1}{4},\,\frac{1-s}{4}\Big\} \end{split}$$

Nodal coordinates

Element node	Global node number	X	y
1	3	6.	0.
2	5	20.	0.
3	6	20.	12.
4	4	6.	5.

Mapping to the master element

$$\begin{split} x(s,t) &= 1.5 \ (1-s) \ (1-t) + 5. \ (s+1) \ (1-t) + 1.5 \ (1-s) \ (t+1) + 5. \ (s+1) \ (t+1) \\ y(s,t) &= 1.25 \ (1-s) \ (t+1) + 3. \ (s+1) \ (t+1) \\ J &= \left( \begin{array}{ccc} 3.5 \ (1-t) + 3.5 \ (t+1) & 0 \\ 1.75 \ (t+1) & 1.25 \ (1-s) + 3. \ (s+1) \end{array} \right); \end{split} \qquad det J = 12.25 \ s + 29.75 \end{split}$$

Element solution at  $\{s, t\} = \{0, 0\} \Longrightarrow \{x, y\} = \{13., 4.25\}$ 

$$\begin{split} \{N_1, \ N_2, \ N_3, \ N_4\} &= \Big\{\frac{1}{4}, \ \frac{1}{4}, \ \frac{1}{4}, \ \frac{1}{4}\Big\} \\ \\ \{\partial N_1/\partial s, \ \partial N_2/\partial s, \ \partial N_3/\partial s, \ \partial N_4/\partial s\} &= \Big\{-\frac{1}{4}, \ \frac{1}{4}, \ \frac{1}{4}, \ -\frac{1}{4}\Big\} \\ \\ \{\partial N_1/\partial t, \ \partial N_2/\partial t, \ \partial N_3/\partial t, \ \partial N_4/\partial t\} &= \Big\{-\frac{1}{4}, -\frac{1}{4}, \ \frac{1}{4}, \ \frac{1}{4}\Big\} \end{split}$$

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.0210084,\ 0.0504202,\ 0.0210084,\ -0.0504202\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0588235,\ -0.0588235,\ 0.0588235,\ 0.0588235\}$ 

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.0210084 & 0 & 0.0504202 & 0 & 0.0210084 & 0 & -0.0504202 & 0 \\ 0 & -0.0588235 & 0 & -0.0588235 & 0 & 0.0588235 & 0 & 0.\\ -0.0588235 & -0.0210084 & -0.0588235 & 0.0504202 & 0.0588235 & 0.0210084 & 0.0588235 & -0. \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-6.08383 \times 10^{-6} -4.91655 \times 10^{-6} -0.0000349244)$ 

In-plane stress components, $\sigma = C\epsilon = (-22.0848 - 19.1666 - 43.6555)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{T} = (\ -6.08383 \times 10^{-6} \quad -4.91655 \times 10^{-6} \quad 2.7501 \times 10^{-6} \quad -0.0000349244 \quad 0 \quad 0 \ )$$

$$\sigma^{\mathrm{T}} = (-22.0848 - 19.1666 \ 0 \ -43.6555 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(23.0542 \ 0. \ -64.3056)$ 

Effective stress (von Mises) = 78.4169

Element solution at  $\{s, t\} = \{-1, -1\} \Longrightarrow \{x, y\} = \{6, 0.\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{1,\ 0,\ 0,\ 0\}$$

$$\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\} = \left\{-\frac{1}{2},\;\frac{1}{2},\;0,\;0\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{-\frac{1}{2},\; 0,\; 0,\; \frac{1}{2}\right\}$$

$$\{\partial N_1/\partial x,\; \partial N_2/\partial x,\; \partial N_3/\partial x,\; \partial N_4/\partial x\} = \{-0.0714286,\; 0.0714286,\; 0,\; 0.\}$$

$$\{\partial N_1/\partial y, \, \partial N_2/\partial y, \, \partial N_3/\partial y, \, \partial N_4/\partial y\} = \{-0.2, \, 0., \, 0, \, 0.2\}$$

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.0714286 & 0 & 0.0714286 & 0 & 0 & 0 & 0 \\ 0 & -0.2 & 0 & 0 & 0 & 0 & 0.2 \\ -0.2 & -0.0714286 & 0 & 0.0714286 & 0 & 0 & 0.2 & 0 \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (0.0000207772 \ 0.0000370391 \ 0.0000552707)$ 

In-plane stress components,  $\sigma = C\epsilon = (88.0783 \ 128.733 \ 69.0884)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{\mathrm{T}} = (0.0000207772 \ 0.0000370391 \ -0.0000144541 \ 0.0000552707 \ 0 \ 0)$$

$$\sigma^{\mathrm{T}} = (88.0783 \ 128.733 \ 0 \ 69.0884 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(180.422 \quad 36.3889 \quad 0.)$ 

Effective stress (von Mises) = 165.26

Element solution at  $\{s, t\} = \{-1, 1\} \Longrightarrow \{x, y\} = \{6, 5, 5\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{0,\ 0,\ 0,\ 1\}$$

$$\{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} = \left\{0,\ 0,\ \frac{1}{2},\ -\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{-\frac{1}{2},\; 0,\; 0,\; \frac{1}{2}\right\}$$

$$\{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{0.1, 0, 0.0714286, -0.171429\}$$

$$\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\}=\{-0.2,\ 0,\ 0.,\ 0.2\}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.0000705504 \ 0.0000370391 \ 4.32456 \times 10^{-6})$ 

In–plane stress components, $\sigma = C\epsilon = (-197.32 71.6533 5.4057)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = (-0.0000705504 \ 0.0000370391 \ 8.37781 \times 10^{-6} \ 4.32456 \times 10^{-6} \ 0 \ 0)$$

$$\boldsymbol{\sigma}^{\mathrm{T}} = (-197.32 \ 71.6533 \ 0 \ 5.4057 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(71.7619 \ 0. \ -197.429)$ 

Effective stress (von Mises) = 241.445

Element solution at  $\{s, t\} = \{1, -1\} \Longrightarrow \{x, y\} = \{20, 0.\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{0,\ 1,\ 0,\ 0\}$$

$$\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\} = \left\{-\,\frac{1}{2}\,,\;\frac{1}{2},\;0,\;0\right\}$$

$$\left\{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\right\} = \left\{0,\ -\frac{1}{2},\ \frac{1}{2},\ 0\right\}$$

 $\{\partial N_1/\partial x, \partial N_2/\partial x, \partial N_3/\partial x, \partial N_4/\partial x\} = \{-0.0714286, 0.0714286, 0., 0\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\}=\{0.,\ -0.0833333,\ 0.0833333,\ 0\}$ 

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (0.0000207772 - 0.0000223981 - 0.0000512781)$ 

In-plane stress components,  $\sigma = C\epsilon = (50.93 -57.0082 -64.0977)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = (\ 0.0000207772 \ -0.0000223981 \ 4.05212 \times 10^{-7} \ -0.0000512781 \ 0 \ 0)$$

$$\boldsymbol{\sigma}^{\mathrm{T}} = (\ 50.93 \ -57.0082 \ 0 \ -64.0977 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(80.7534 \ 0. -86.8316)$ 

Effective stress (von Mises) = 145.165

Element solution at  $\{s, t\} = \{1, 1\} \Longrightarrow \{x, y\} = \{20, 12.\}$ 

$$\{N_1,\ N_2,\ N_3,\ N_4\}=\{0,\ 0,\ 1,\ 0\}$$

$$\left\{\partial N_1/\partial s,\; \partial N_2/\partial s,\; \partial N_3/\partial s,\; \partial N_4/\partial s\right\} = \left\{0,\; 0,\; \frac{1}{2}\,,\; -\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{0,\; -\frac{1}{2},\; \frac{1}{2},\; 0\right\}$$

 $\{\partial N_1/\partial x, \, \partial N_2/\partial x, \, \partial N_3/\partial x, \, \partial N_4/\partial x\} = \{0, \, 0.0416667, \, 0.0297619, \, -0.0714286\}$ 

$$\boldsymbol{B}^{T} = \begin{pmatrix} 0 & 0 & 0.0416667 & 0 & 0.0297619 & 0 & -0.0714286 & 0 \\ 0 & 0 & 0 & -0.0833333 & 0 & 0.0833333 & 0 & 0 \\ 0 & 0 & -0.0833333 & 0.0416667 & 0.0833333 & 0.0297619 & 0 & -0.0714286 \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.0000172759 -0.0000223981 -0.0000725057)$ 

In-plane stress components,  $\sigma = C\epsilon = (-67.9861 -80.7914 -90.6321)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{T} = (-0.0000172759 \ -0.0000223981 \ 9.9185 \times 10^{-6} \ -0.0000725057 \ 0 \ 0)$$

$$\sigma^{T} = (-67.9861 \ -80.7914 \ 0 \ -90.6321 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(16.4692 \ 0. -165.247)$ 

Effective stress (von Mises) = 174.067

#### Solution for element 3

Element nodal displacements

Element node	Global node number	u	$\mathbf{v}$
1	5	0.00305003	-0.0113566
2	7	0	0
3	8	0	0
4	6	-0.00210128	-0.0116254

$$\mathbf{d}^{\mathrm{T}} = (0.00305003 \ -0.0113566 \ 0 \ 0 \ 0 \ -0.00210128 \ -0.0116254)$$

$$E = 3000000;$$
  $v = 0.2;$   $h = 4$ 

Plane stress 
$$C = \begin{pmatrix} 3.125 \times 10^6 & 625000. & 0 \\ 625000. & 3.125 \times 10^6 & 0 \\ 0 & 0 & 1.25 \times 10^6 \end{pmatrix}$$

Interpolation functions and their derivatives

$$\begin{split} \{N_1,\,N_2,\,N_3,\,N_4\} &= \left\{\frac{1}{4}\,(s-1)\,(t-1),\,-\frac{1}{4}\,(s+1)\,(t-1),\,\frac{1}{4}\,(s+1)\,(t+1),\,-\frac{1}{4}\,(s-1)\,(t+1)\right\} \\ \{\partial N_1/\partial s,\,\partial N_2/\partial s,\,\partial N_3/\partial s,\,\partial N_4/\partial s\} &= \left\{\frac{t-1}{4},\,\frac{1-t}{4},\,\frac{t+1}{4},\,\frac{1}{4}\,(-t-1)\right\} \\ \{\partial N_1/\partial s,\,\partial N_2/\partial s,\,\partial N_3/\partial s,\,\partial N_4/\partial s\} &= \left\{\frac{s-1}{4},\,\frac{1}{4}\,(-s-1),\,\frac{s+1}{4},\,\frac{1-s}{4}\right\} \end{split}$$

#### Nodal coordinates

Element node	Global node number	X	$\mathbf{y}$
1	5	20.	0.
2	7	54.	0.
3	8	54.	12.
4	6	20.	12.

Mapping to the master element

$$\begin{split} x(s,t) &= 5.\,(1-s)\,(1-t) + 13.5\,(s+1)\,(1-t) + 5.\,(1-s)\,(t+1) + 13.5\,(s+1)\,(t+1) \\ y(s,t) &= 3.\,(1-s)\,(t+1) + 3.\,(s+1)\,(t+1) \\ \boldsymbol{J} &= \left( \begin{array}{cc} 8.5\,(1-t) + 8.5\,(t+1) & 0 \\ 0 & 3.\,(1-s) + 3.\,(s+1) \end{array} \right); \end{split} \qquad det \boldsymbol{J} = 102. \end{split}$$

Element solution at  $\{s, t\} = \{0, 0\} \Longrightarrow \{x, y\} = \{37., 6.\}$ 

$$\begin{split} \{N_1,\ N_2,\ N_3,\ N_4\} &= \Big\{\frac{1}{4},\ \frac{1}{4},\ \frac{1}{4},\ \frac{1}{4}\Big\} \\ \\ \{\partial N_1/\partial s,\ \partial N_2/\partial s,\ \partial N_3/\partial s,\ \partial N_4/\partial s\} &= \Big\{-\frac{1}{4},\ \frac{1}{4},\ \frac{1}{4},\ -\frac{1}{4}\Big\} \\ \\ \{\partial N_1/\partial t,\ \partial N_2/\partial t,\ \partial N_3/\partial t,\ \partial N_4/\partial t\} &= \Big\{-\frac{1}{4},\ -\frac{1}{4},\ \frac{1}{4},\ \frac{1}{4}\Big\} \end{split}$$

 $\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\} = \{-0.0147059,\ 0.0147059,\ 0.0147059,\ -0.0147059\}$ 

 $\{\partial N_1/\partial y, \partial N_2/\partial y, \partial N_3/\partial y, \partial N_4/\partial y\} = \{-0.0416667, -0.0416667, 0.0416667, 0.0416667\}$ 

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.0147059 & 0 & 0.0147059 & 0 & 0.0147059 & 0 & -0.0147059 & 0 \\ 0 & -0.0416667 & 0 & -0.0416667 & 0 & 0.0416667 & 0 & 0. \\ -0.0416667 & -0.0147059 & -0.0416667 & 0.0147059 & 0.0416667 & 0.0147059 & 0.0416667 & -0. \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{\mathrm{T}} \mathbf{d} = (-0.0000139523 -0.000011199 \ 0.000123333)$ 

In-plane stress components,  $\sigma = C\epsilon = (-50.6003 - 43.7172 \ 154.167)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{T} = (-0.0000139523 - 0.000011199 \ 6.28783 \times 10^{-6} \ 0.000123333 \ 0 \ 0)$$

$$\sigma^{T} = (-50.6003 \ -43.7172 \ 0 \ 154.167 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(107.046 \ 0. -201.364)$ 

Effective stress (von Mises) = 271.222

Element solution at  $\{s, t\} = \{-1, -1\} \Longrightarrow \{x, y\} = \{20, 0.\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{1, 0, 0, 0\}$$

$$\left\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\right\} = \left\{-\frac{1}{2},\;\frac{1}{2},\;0,\;0\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{-\frac{1}{2},\; 0,\; 0,\; \frac{1}{2}\right\}$$

$$\{\partial N_1/\partial x,\; \partial N_2/\partial x,\; \partial N_3/\partial x,\; \partial N_4/\partial x\} = \{-0.0294118,\; 0.0294118,\; 0,\; 0.\}$$

 $\{\partial N_1/\partial y, \, \partial N_2/\partial y, \, \partial N_3/\partial y, \, \partial N_4/\partial y\} = \{-0.0833333, \, 0., \, 0, \, 0.0833333\}$ 

$$\boldsymbol{B}^{\mathrm{T}} = \begin{pmatrix} -0.0294118 & 0 & 0.0294118 & 0 & 0 & 0 & 0 \\ 0 & -0.0833333 & 0 & 0 & 0 & 0 & 0 & 0.0833333 \\ -0.0833333 & -0.0294118 & 0 & 0.0294118 & 0 & 0 & 0.0833333 & 0 \end{pmatrix}$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.0000897069 -0.0000223981 -0.0000952572)$ 

In-plane stress components,  $\sigma = C\epsilon = (-294.333 - 126.061 - 119.072)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{\mathrm{T}} = (-0.0000897069 \ -0.0000223981 \ 0.0000280262 \ -0.0000952572 \ 0 \ 0)$$
 
$$\boldsymbol{\sigma}^{\mathrm{T}} = (-294.333 \ -126.061 \ 0 \ -119.072 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses = (0. -64.3993 -355.994)

Effective stress (von Mises) = 328.563

Element solution at  $\{s, t\} = \{-1, 1\} \Longrightarrow \{x, y\} = \{20, 12.\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{0, 0, 0, 1\}$$

$$\left\{\partial N_1/\partial s,\; \partial N_2/\partial s,\; \partial N_3/\partial s,\; \partial N_4/\partial s\right\} = \left\{0,\; 0,\; \frac{1}{2},\; -\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{-\frac{1}{2},\; 0,\; 0,\; \frac{1}{2}\right\}$$

 $\{\partial N_1/\partial x, \, \partial N_2/\partial x, \, \partial N_3/\partial x, \, \partial N_4/\partial x\} = \{0., \, 0, \, 0.0294118, \, -0.0294118\}$ 

 $\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{-0.0833333,\ 0,\ 0.,\ 0.0833333\}$ 

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (0.0000618023 - 0.0000223981 - 0.000087352)$ 

In-plane stress components,  $\sigma = C\epsilon = (179.133 -31.3676 -109.19)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{T} = (0.0000618023 - 0.0000223981 - 9.85105 \times 10^{-6} - 0.000087352 \ 0 \ 0)$$

$$\sigma^{T} = (179.133 - 31.3676 \ 0 \ -109.19 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(225.541 \ 0. \ -77.775)$ 

Effective stress (von Mises) = 272.872

Element solution at  $\{s, t\} = \{1, -1\} \Longrightarrow \{x, y\} = \{54, 0.\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{0, 1, 0, 0\}$$

$$\left\{\partial N_1/\partial s,\; \partial N_2/\partial s,\; \partial N_3/\partial s,\; \partial N_4/\partial s\right\} = \left\{-\frac{1}{2},\; \frac{1}{2},\; 0,\; 0\right\}$$

$$\left\{\partial N_1/\partial t,\;\partial N_2/\partial t,\;\partial N_3/\partial t,\;\partial N_4/\partial t\right\} = \left\{0,\;-\frac{1}{2},\;\frac{1}{2},\;0\right\}$$

 $\{\partial N_1/\partial x,\; \partial N_2/\partial x,\; \partial N_3/\partial x,\; \partial N_4/\partial x\} = \{-0.0294118,\; 0.0294118,\; 0.,\; 0\}$ 

 $\{\partial N_1/\partial y, \ \partial N_2/\partial y, \ \partial N_3/\partial y, \ \partial N_4/\partial y\} = \{0., \ -0.0833333, \ 0.0833333, \ 0\}$ 

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (-0.0000897069 \ 0. \ 0.000334019)$ 

In-plane stress components,  $\sigma = C\epsilon = (-280.334 - 56.0668 417.523)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\boldsymbol{\epsilon}^{T} = (-0.0000897069 \ 0. \ 0.0000224267 \ 0.000334019 \ 0 \ 0)$$

$$\sigma^{\mathrm{T}} = (-280.334 - 56.0668 \ 0 \ 417.523 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(264.119 \ 0. \ -600.519)$ 

Effective stress (von Mises) = 767.457

Element solution at  $\{s, t\} = \{1, 1\} \Longrightarrow \{x, y\} = \{54., 12.\}$ 

$$\{N_1, N_2, N_3, N_4\} = \{0, 0, 1, 0\}$$

$$\left\{\partial N_1/\partial s,\;\partial N_2/\partial s,\;\partial N_3/\partial s,\;\partial N_4/\partial s\right\} = \left\{0,\;0,\;\frac{1}{2}\,,\;-\frac{1}{2}\right\}$$

$$\{\partial N_1/\partial t,\; \partial N_2/\partial t,\; \partial N_3/\partial t,\; \partial N_4/\partial t\} = \left\{0,\; -\frac{1}{2},\; \frac{1}{2},\; 0\right\}$$

$$\{\partial N_1/\partial x,\ \partial N_2/\partial x,\ \partial N_3/\partial x,\ \partial N_4/\partial x\}=\{0,\ 0.,\ 0.0294118,\ -0.0294118\}$$

$$\{\partial N_1/\partial y,\ \partial N_2/\partial y,\ \partial N_3/\partial y,\ \partial N_4/\partial y\} = \{0,\ -0.0833333,\ 0.08333333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.0833333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.083333333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.08333333,\ 0.083333333,\ 0.083333333,\ 0.083333333,\ 0.083333333,\ 0.083333333,\ 0.0833333333,\ 0.08333333333,\ 0.0833333333,\ 0.0833333333333333$$

In-plane strain components,  $\epsilon = \mathbf{B}^{T} \mathbf{d} = (0.0000618023 \ 0. \ 0.000341924)$ 

In-plane stress components,  $\sigma = C\epsilon = (193.132 \ 38.6264 \ 427.405)$ 

Computing out-of-plane strain and stress components using appropriate formulas, the complete strain and stress vectors are as follows.

$$\epsilon^{\mathrm{T}} = (0.0000618023 \ 0. \ -0.0000154506 \ 0.000341924 \ 0 \ 0)$$

$$\sigma^{\mathrm{T}} = (193.132 \ 38.6264 \ 0 \ 427.405 \ 0 \ 0)$$

Substituting these stress components into appropriate formulas

Principal stresses =  $(550.21 \ 0. \ -318.451)$ 

Effective stress (von Mises) = 761.155

#### Solution summary

Nodal solution

	X	y	u	V
1	0.	5.	0	-0.0183155
2	0.	12.	0	-0.0183204
3	6.	0.	0.00275915	-0.0166486
4	6.	5.	0.00114552	-0.0164634
5	20.	0.	0.00305003	-0.0113566
6	20.	12.	-0.00210128	-0.0116254
7	54.	0.	0	0
8	<b>54</b> .	12.	0	0

# Solution at selected points on elements

	Coord	Disp	Stresses	Principal stresses	Effective Stress
1	6.5 8.5	-0.00023894 -0.0161812	-104.571 28.544 0 166.978 0	141.741 0. -217.768	313.656
2	13. 4.25	0.00121336 -0.0140235	-22.0848 -19.1666 0 -43.6555 0	23.0542 0. -64.3056	78.4169
3	37. 6.	0.000237189 -0.00574551	-50.6003 -43.7172 0 154.167 0	107.046 0. -201.364	271.222

# Support reactions

Node	dof	Reaction
1	1	-7931.98
2	1	10360.8
7	1	-19673.
7	2	5839.98
8	1	17244.2
8	2	4960.02

Sum of applied loads  $\rightarrow$  (0 -10800.)

Sum of support reactions  $\rightarrow$  (0 10800.)