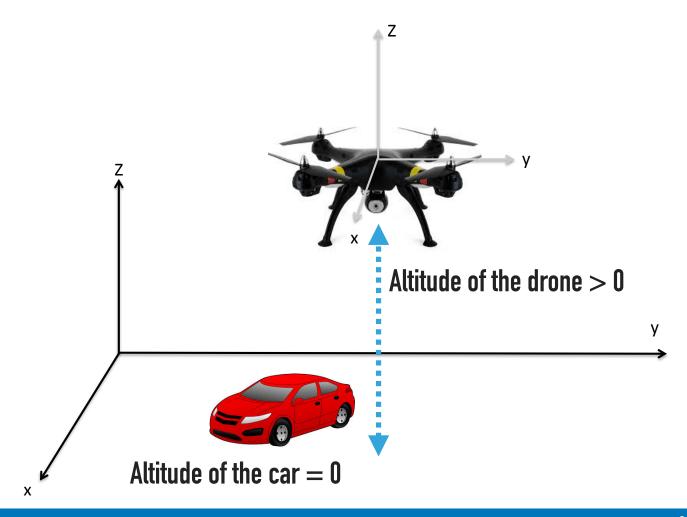




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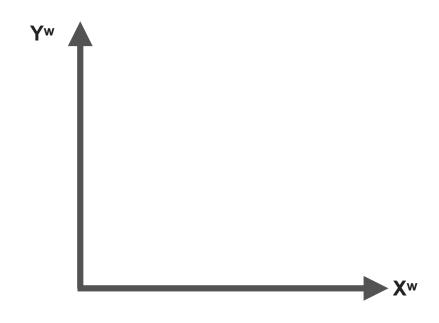
Section Frames and 3D Transformations

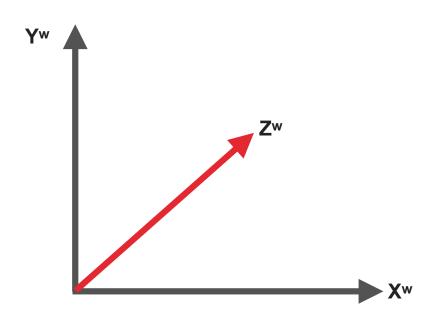
3D COORDINATE SYSTEM

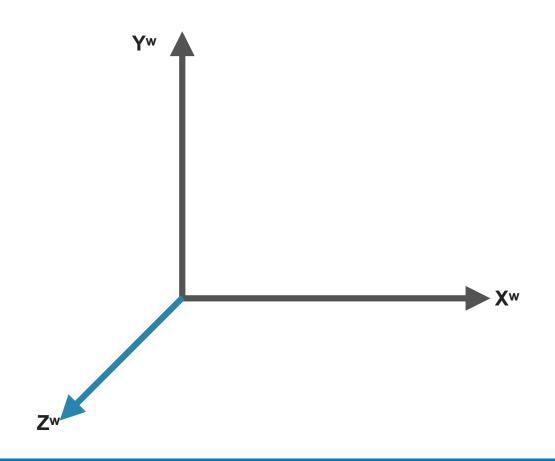


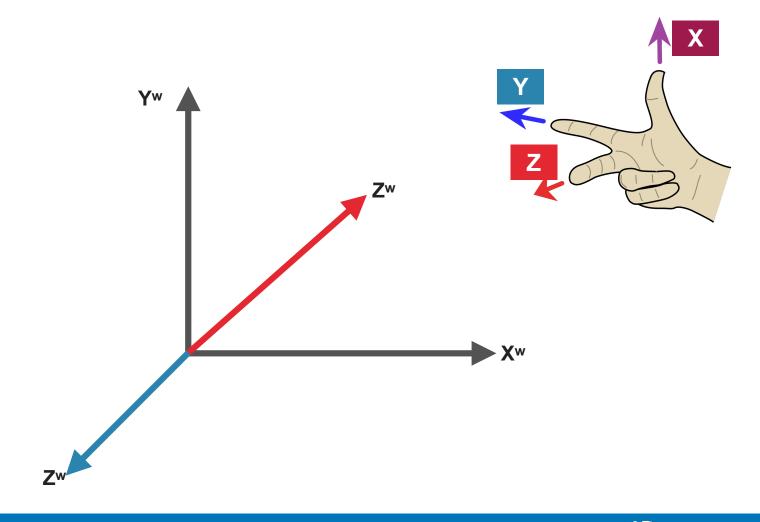
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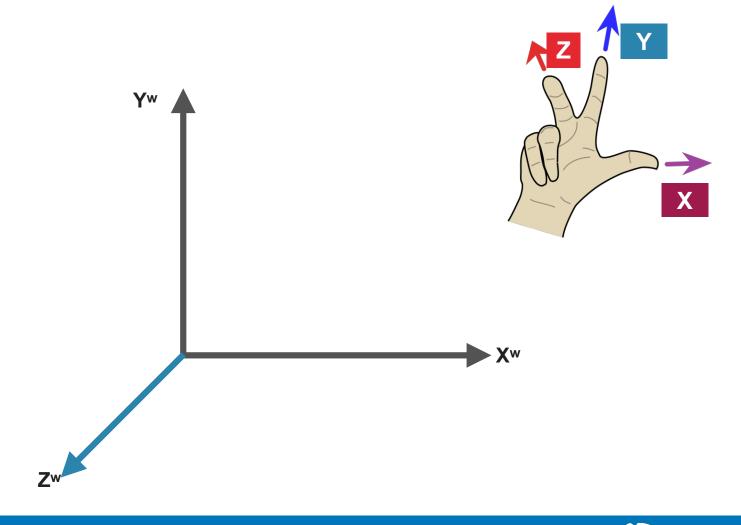
Right-Hand Rule

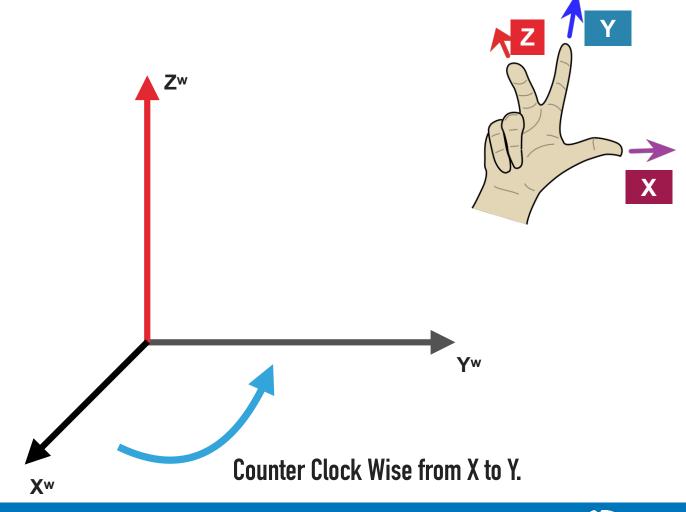












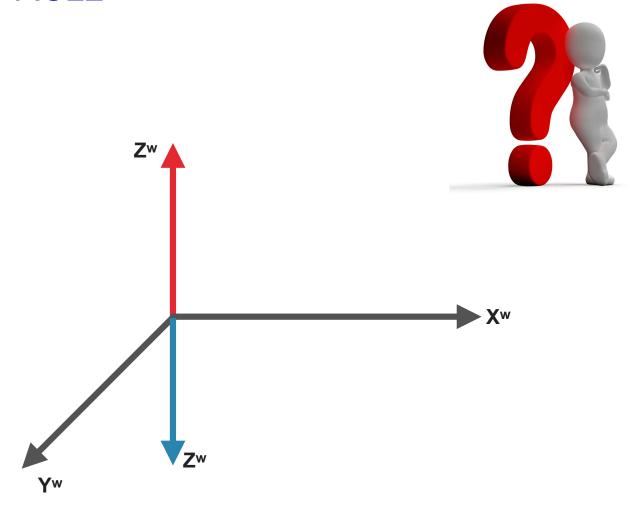




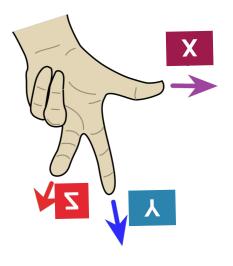
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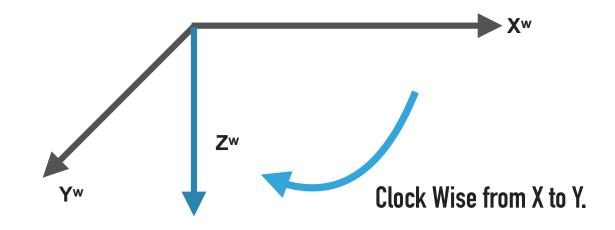
Right-Hand Rule: Exercise

RIGHT-HAND RULE



RIGHT-HAND RULE



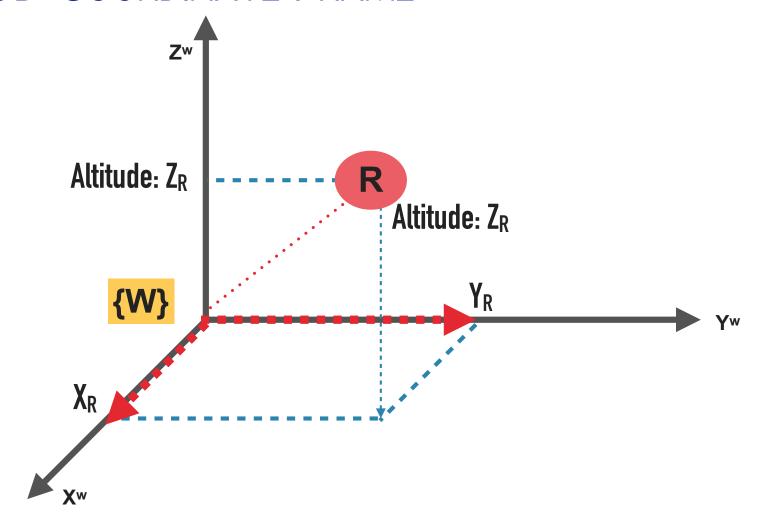


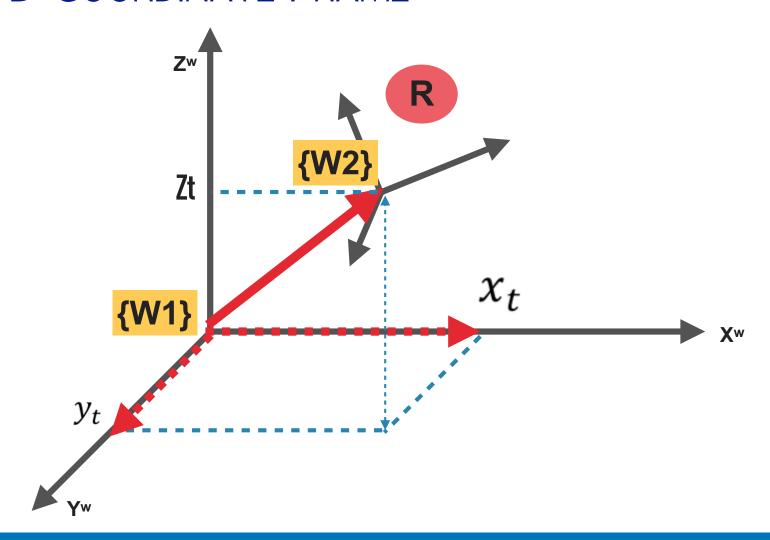




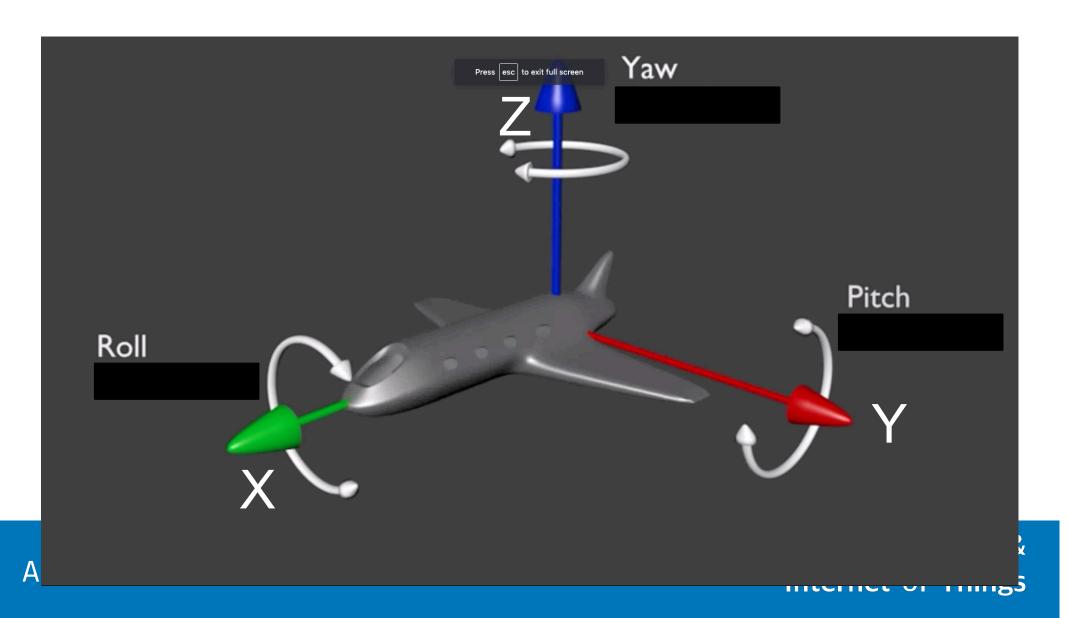
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3D Transformation



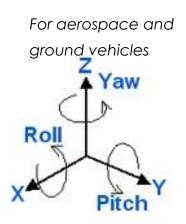


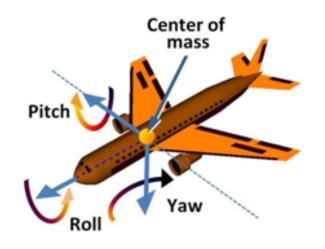
ROLL-PITCH-YAW (X,Y,Z)

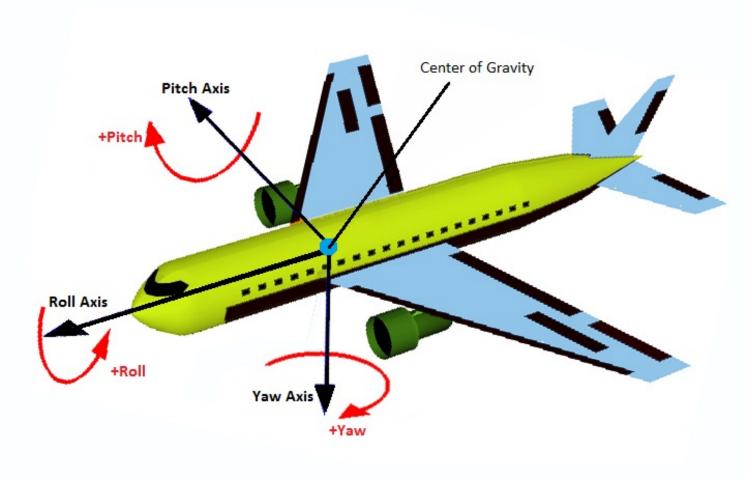


ROLL-PITCH-YAW (X,Y,Z)

- **Roll:** bank
 - Rotation on x-axis
- Pitch: attitude
 - Rotation on y-axis
- Yaw: heading
 - Rotation on z-axis

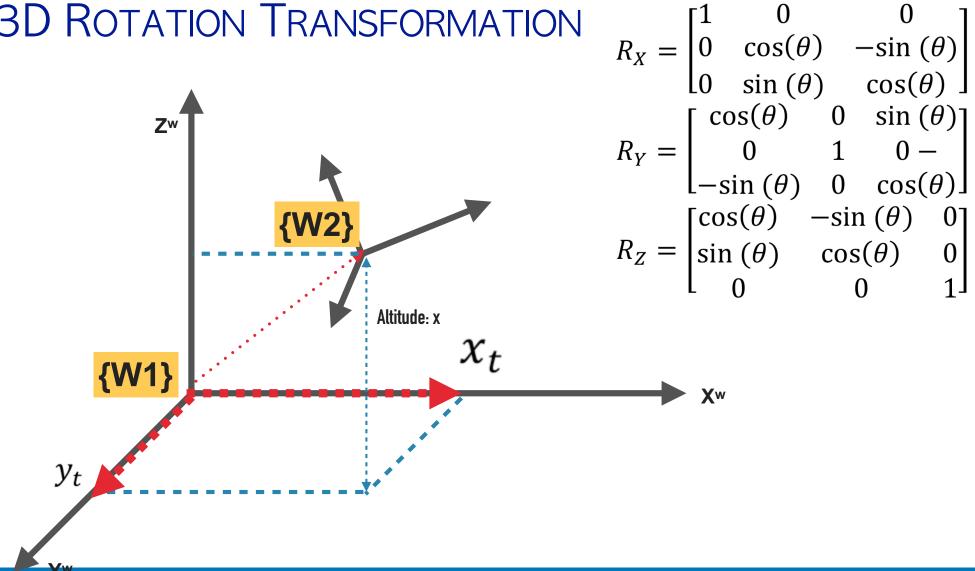






http://elektromot.com/how-to-control-an-aircraft/

3D ROTATION TRANSFORMATION



GENERAL ROTATION MATRIX

- The general rotation matrix can be obtained by a matrix multiplication
- For example, if we perform a rotation around the Z-axis with angle APLHA, then around the Y-axis with angle BETA and then around the X-axis with the angle GAMMA, the resulting rotation matrix will be.

$$R = R_z(\alpha)R_y(\beta)R_x(\gamma)$$

$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} r11 & r12 & r13 & x_t \\ r21 & r22 & r23 & y \\ r31 & r32 & r33 & z_t \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$

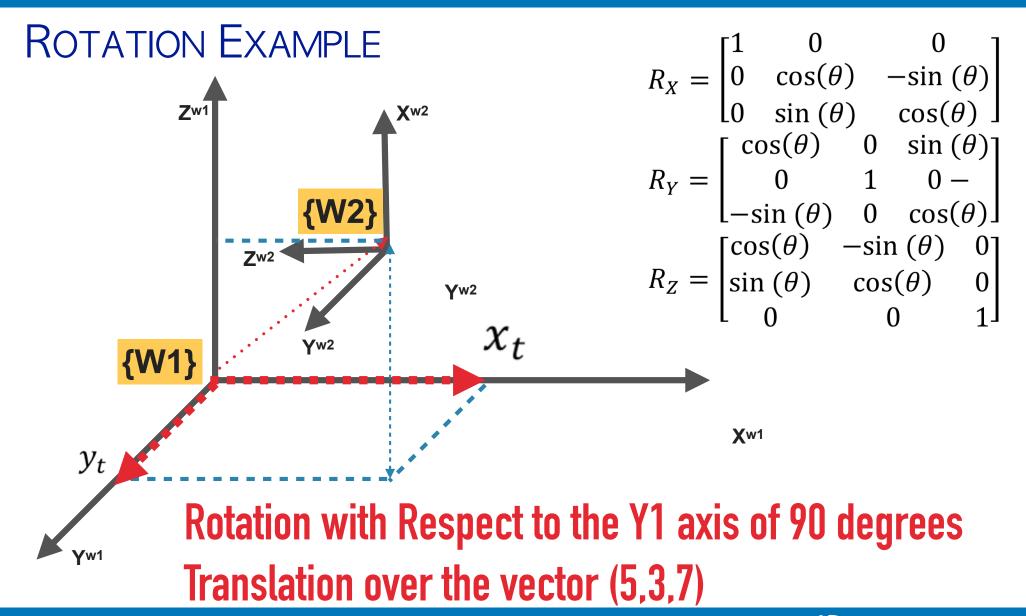
$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} w_1 R_{w2} & t \\ 0_{1x3} & 1 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$

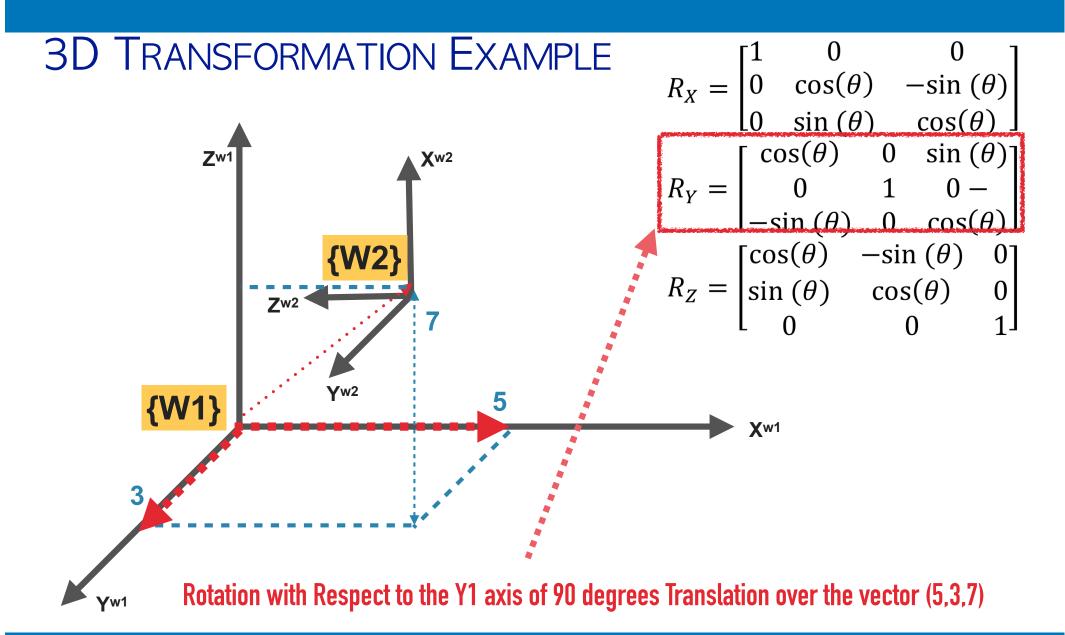




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3D Transformation: Application





$$R = R_{y}(\theta)$$

$$\begin{bmatrix} w^{1}x \\ w^{1}y \\ w^{1}z \\ 1 \end{bmatrix} = \begin{bmatrix} \cos(90) & 0 & \sin(90) & 5 \\ 0 & 0 & 0 & 3 \\ -\sin(90) & 0 & \cos(90) & 7 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w^{2}x \\ w^{2}y \\ w^{2}z \\ 1 \end{bmatrix}$$

$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 5 \\ 0 & 0 & 0 & 3 \\ -1 & 0 & 0 & 7 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$

$$R = R_{y}(\theta)$$

$$\begin{bmatrix} w^{1}x \\ w^{1}y \\ w^{1}z \\ 1 \end{bmatrix} = \begin{bmatrix} \cos(90) & 0 & \sin(90) & 5 \\ 0 & 0 & 0 & 3 \\ -\sin(90) & 0 & \cos(90) & 7 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w^{2}x \\ w^{2}y \\ w^{2}z \\ 1 \end{bmatrix}$$

$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 5 \\ 0 & 0 & 0 & 3 \\ -1 & 0 & 0 & 7 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$

Rotation Matrix Translation Vector

$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} \cos(90) & 0 & \sin(90) & 5 \\ 0 & 0 & 0 & 3 \\ -\sin(90) & 0 & \cos(90) & 7 \\ 0 & 0 & 0 & 1 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$

$$\begin{bmatrix} w_1 x \\ w_1 y \\ w_1 z \\ 1 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 5 \\ 0 & 0 & 0 & 3 \\ -1 & 0 & 0 & 7 \end{bmatrix} * \begin{bmatrix} w_2 x \\ w_2 y \\ w_2 z \\ 1 \end{bmatrix}$$





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3D Orientation Models

ROTATION REPRESENTATION METHODS

- ▶ Three-Angle Representation
 - Euler Rotation vs. Cardan Rotation
- Rotation about Arbitrary Vector
- Quaternions





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Three-Angle Representation

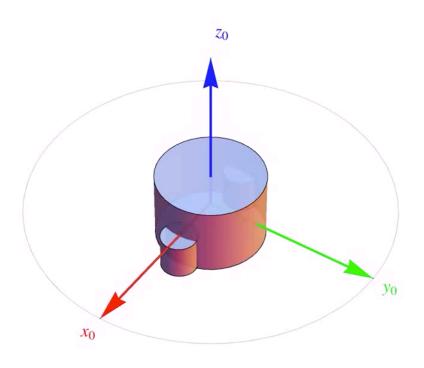
THREE-ANGLE REPRESENTATION: EULER ROTATION

- Euler Rotation Sequence: involves repetition, but not successive, of rotations about one particular axis
- Cardan Rotation Sequence:
 characterized by rotations
 about all three axes

THREE-ANGLE REPRESENTATION: EULER ROTATION

Any <u>two</u> independent <u>orthonormal coordinate frames</u> can be related by a sequence of rotations (not more than three) about coordinate axes, where <u>no two successive</u> rotations may be about the same axis.

EXAMPLE OF EULER ROTATION: Z, X, Z



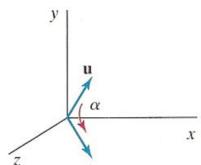




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Rotation about Arbitrary Vector

EULER ROTATION THEOREM



- In 3D space, orientation can be expressed as a rotation about some axis that runs through a fixed point on the rigid body.
- This means, for any rotation, there exists some axis at certain angle around which the rotation occurs.
- Rodrigues Formula

$$R = I\cos\theta + \sin\theta[\mathbf{u}]_{\times} + (1-\cos\theta)\mathbf{u}\otimes\mathbf{u},$$

EULER ROTATION THEOREM

 Given a unit vector u=(ux, uy, uz), the matrix of rotation by an angle THETA about the axis of direction u is

$$R = \begin{bmatrix} \cos\theta + u_x^2 \left(1 - \cos\theta\right) & u_x u_y \left(1 - \cos\theta\right) - u_z \sin\theta & u_x u_z \left(1 - \cos\theta\right) + u_y \sin\theta \\ u_y u_x \left(1 - \cos\theta\right) + u_z \sin\theta & \cos\theta + u_y^2 \left(1 - \cos\theta\right) & u_y u_z \left(1 - \cos\theta\right) - u_x \sin\theta \\ u_z u_x \left(1 - \cos\theta\right) - u_y \sin\theta & u_z u_y \left(1 - \cos\theta\right) + u_x \sin\theta & \cos\theta + u_z^2 \left(1 - \cos\theta\right) \end{bmatrix}.$$



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Quaternion

QUATERNION

- Quaternion are another way of representing rotation
- It is written as a scalar and a vector

$$q = s\langle v_1 | v_2 | v_2 \rangle$$

$$q = s + v$$

$$q = s + v_1 i + v_2 j + v_3 k$$

$$i^2 = j^2 = k^2 = ijk = 1$$

$$\begin{cases} q_0 = q_w = s \\ q_1 = q_x = v_1 \\ q_2 = q_y = v_2 \\ q_3 = q_z = v_3 \end{cases}$$

In ROS: the notation used is (X, Y, Z, W)

EQUIVALENT ROTATION MATRIX

The rotation matrix corresponding to a clockwise/lefthanded rotation by the unit quaternion axis

$$R = egin{bmatrix} q_0^2 + q_1^2 - q_2^2 - q_3^2 & 2(q_1q_2 - q_0q_3) & 2(q_0q_2 + q_1q_3) \ 2(q_1q_2 + q_0q_3) & q_0^2 - q_1^2 + q_2^2 - q_3^2 & 2(q_2q_3 - q_0q_1) \ 2(q_1q_3 - q_0q_2) & 2(q_0q_1 + q_2q_3) & q_0^2 - q_1^2 - q_2^2 + q_3^2 \end{bmatrix}$$

EULER ANGLES TO QUATERNION CONVERSION

We can also obtain the quaternion from Euler angles using the following conversion:

$$\begin{split} \mathbf{q}_{\mathrm{IB}} &= \begin{bmatrix} \cos(\psi/2) \\ 0 \\ \sin(\psi/2) \end{bmatrix} \begin{bmatrix} \cos(\theta/2) \\ \sin(\theta/2) \\ 0 \end{bmatrix} \begin{bmatrix} \cos(\phi/2) \\ \sin(\phi/2) \\ 0 \end{bmatrix} \\ &= \begin{bmatrix} \cos(\phi/2)\cos(\theta/2)\cos(\psi/2) - \sin(\phi/2)\sin(\theta/2)\sin(\psi/2) \\ \sin(\phi/2)\cos(\theta/2)\cos(\psi/2) + \cos(\phi/2)\sin(\theta/2)\sin(\psi/2) \\ \cos(\phi/2)\sin(\theta/2)\cos(\psi/2) - \sin(\phi/2)\cos(\theta/2)\sin(\psi/2) \\ \cos(\phi/2)\sin(\theta/2)\cos(\psi/2) + \sin(\phi/2)\sin(\theta/2)\cos(\psi/2) \end{bmatrix} \begin{matrix} \mathrm{S=QO} \\ \mathrm{V1} \\ \mathrm{V2} \\ \mathrm{V3} \end{matrix} \end{split}$$

QUATERNION TO EULER ANGLES CONVERSION

The Euler angles can be obtained from the quaternions via the relations

$$egin{bmatrix} \phi \ heta \$$

QUATERNION TO EULER ANGLES CONVERSION

Example of Python Code

```
import math

def quaternion_to_euler_angle(w, x, y, z):

    t0 = +2.0 * (w * x + y * z)
    t1 = +1.0 - 2.0 * (x * x + y * y)
    x = math.degrees(math.atan2(t0, t1))

    t2 = +2.0 * (w * y - z * x)
    t2 = +1.0 if t2 > +1.0 else t2
    t2 = -1.0 if t2 < -1.0 else t2
    y = math.degrees(math.asin(t2))

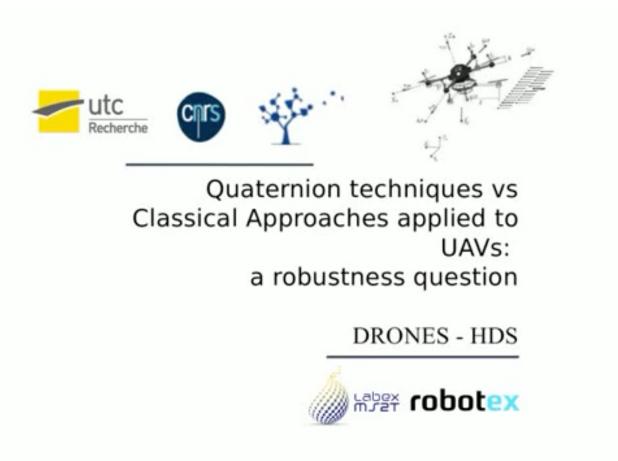
    t3 = +2.0 * (w * z + x * y)
    t4 = +1.0 - 2.0 * (y * y + z * z)
    z = math.degrees(math.atan2(t3, t4))

return X, Y, Z</pre>
```

https://en.wikipedia.org/wiki/Conversion between quaternions and Euler angles

WHY QUATERNION?

WHY QUATERNION?



https://www.youtube.com/watch?v=0VAc_G79POE

QUATERNIONS BENEFITS

- Compared to Euler angles they are simpler to compose and avoid the problem of gimbal lock.
- Compared to rotation matrices they are more compact, more numerically stable, and more efficient.
- Quaternions have applications in computer graphics computer vision, robotics, navigation, molecular dynamics, flight dynamics, orbital mechanics of satellites and crystallographic texture analysis.