

# Induced Power Scaling Alone Cannot Explain Paleozoic Griffenfly Gigantism

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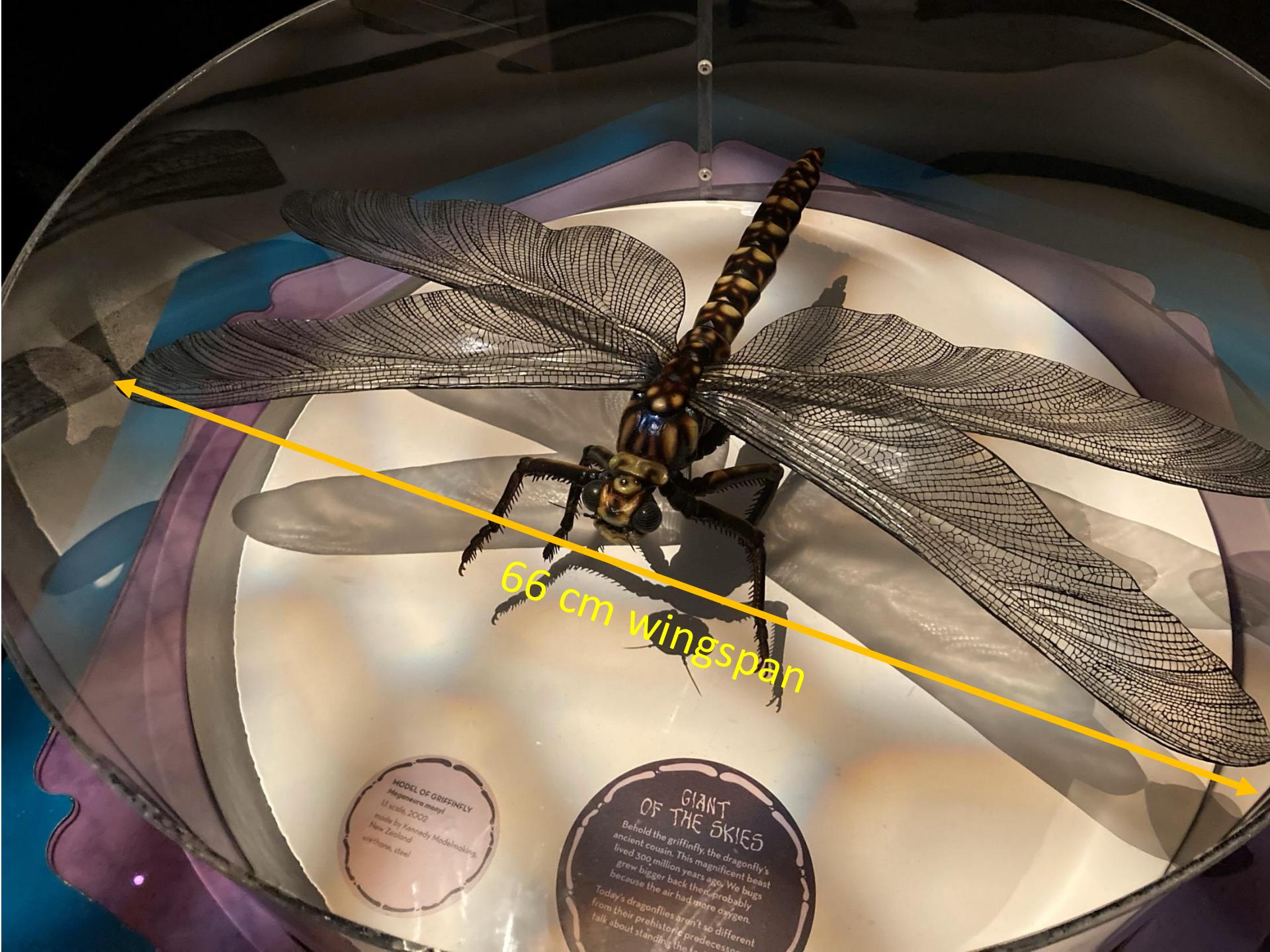
Olaf Ellers<sup>1</sup> and Max Hukill<sup>2</sup>

<sup>1</sup>Bowdoin College

<sup>2</sup>Kaiser Permanente School of Medicine

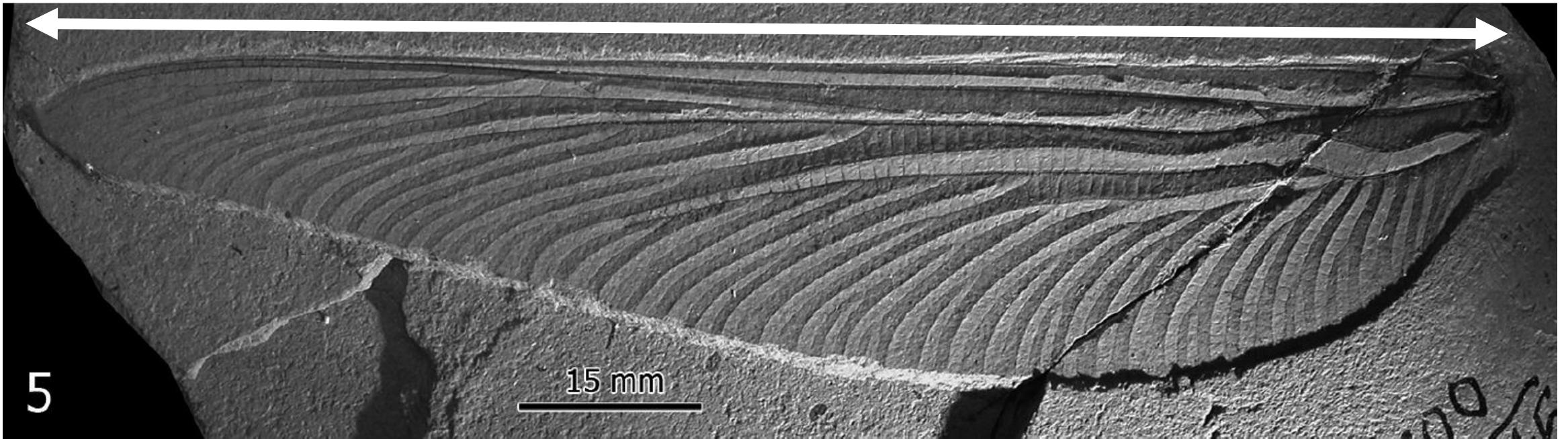


Photo of a model of the griffinfly *Meganeura monyi* at the California Academy of Sciences  
300 mya



May, 1982, 30 cm wing length  
Atkinson, 2005, 66 cm wingspan

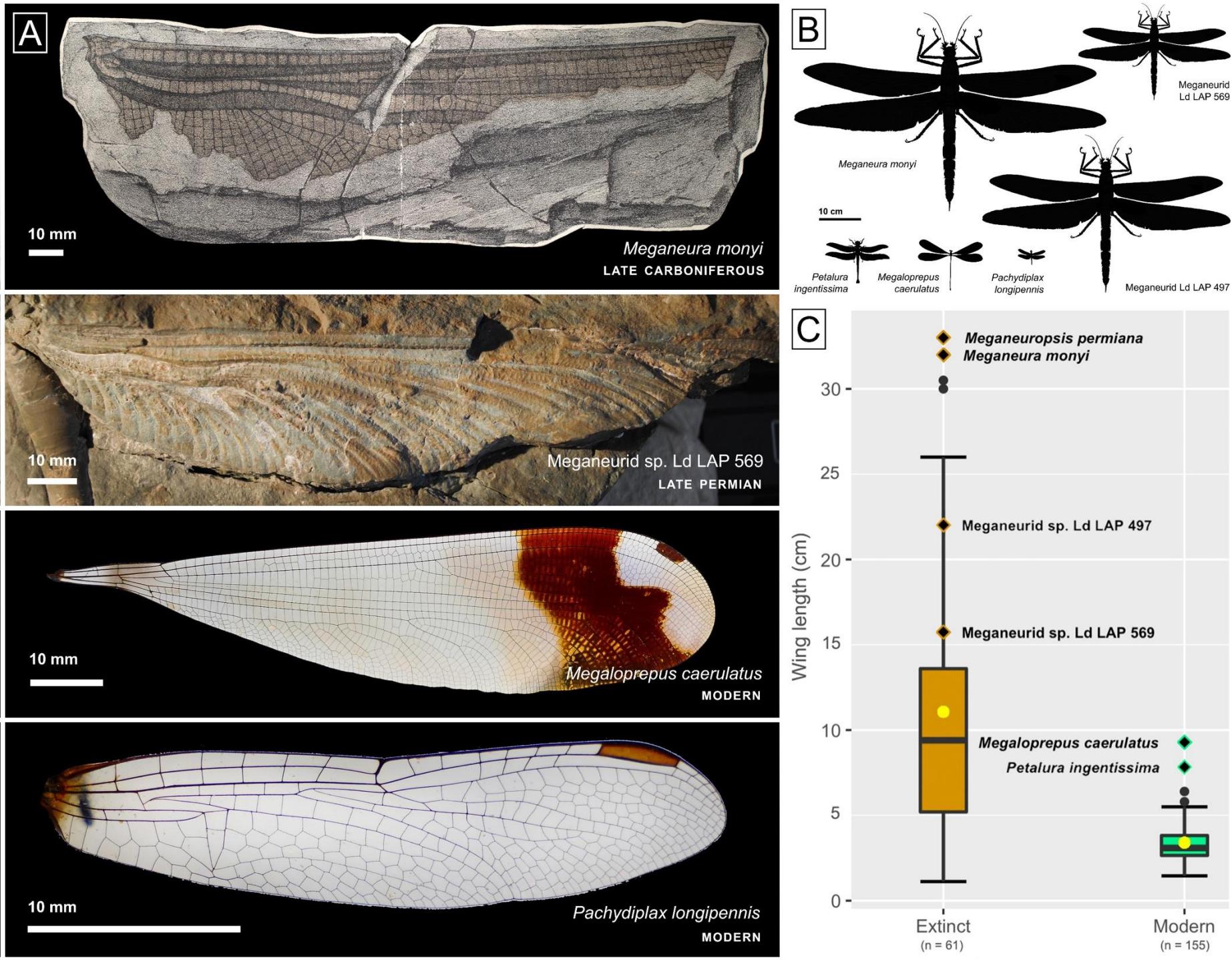
12.5 cm



*Sinomeganeura huangheensis*

Ren et al. 2007



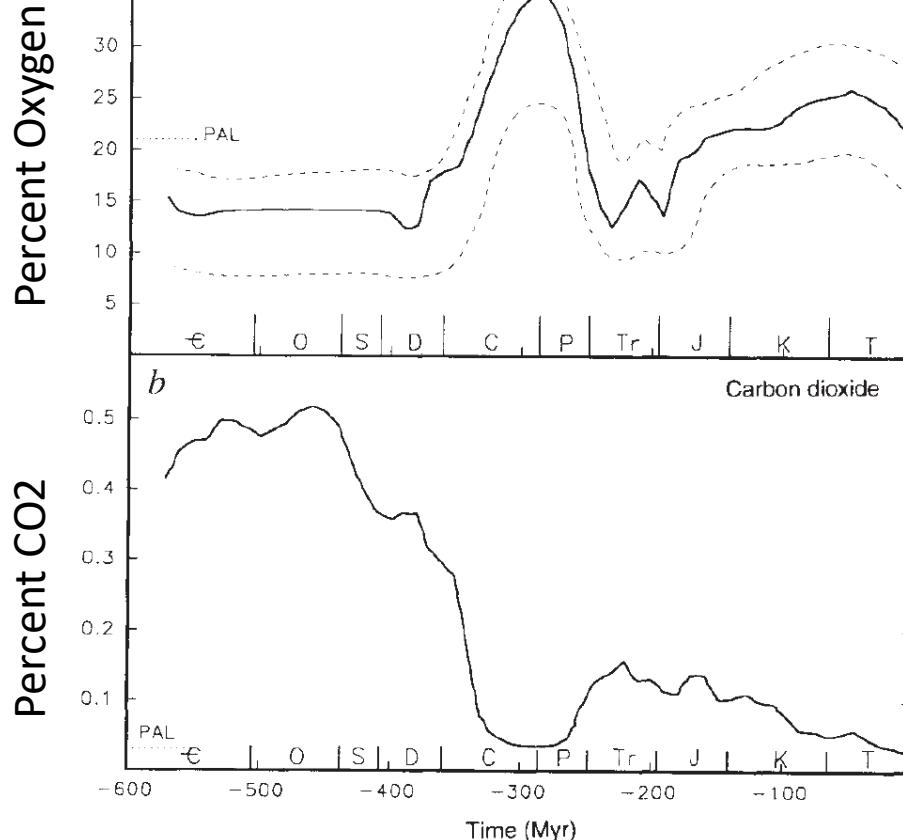


Ellers, et al. 2024

# Implications of the late Palaeozoic oxygen pulse for physiology and evolution

Jeffrey B. Graham, Robert Dudley, Nancy M. Aguilar & Carl Gans

NATURE · VOL 375 · 11 MAY 1995



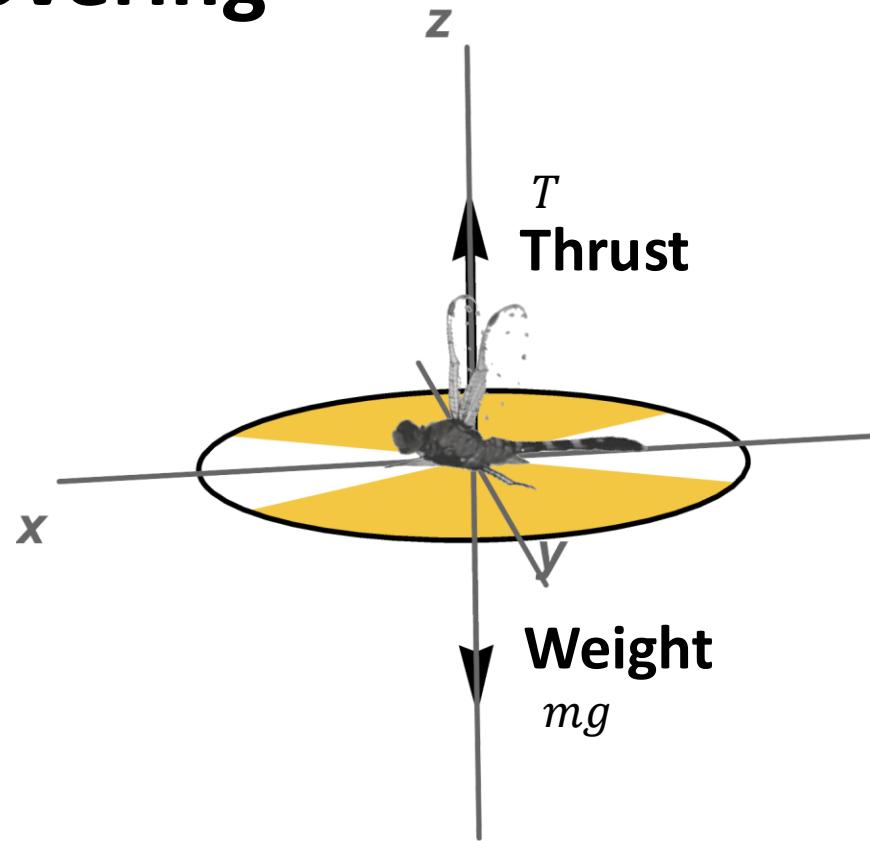
- ↑ O<sub>2</sub> ⇒ growth to larger size
- ↑ O<sub>2</sub> ⇒ more fuel for metabolism and flight
- ↑ O<sub>2</sub> ⇒ ↑ atmospheric density ⇒ greater lift

	% O <sub>2</sub>	Atmospheric Density (kg/m <sup>3</sup> )
Present	21	1.29
Late Carboniferous, 285 MYA	35	1.56

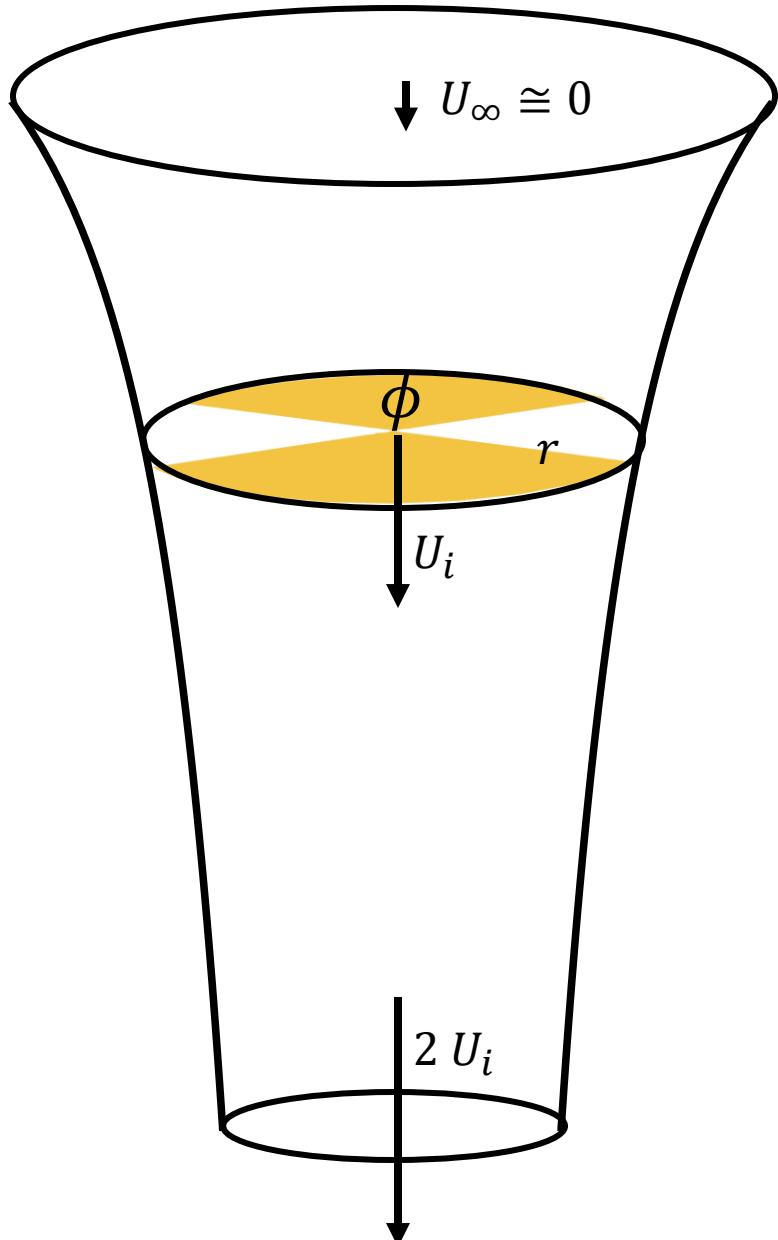
## Question:

In a denser, more oxygen-rich atmosphere,  
how much larger could a dragonfly-like flyer  
theoretically be?

# Hovering



$$T = mg$$



Actuator disk generates thrust

$$U_i = \sqrt{\frac{m g}{2 \rho \phi r^2}}$$

$$P_i = T U_i$$

$$P_i = \sqrt{\frac{(m g)^3}{2 \rho \phi r^2}}$$

Induced Power

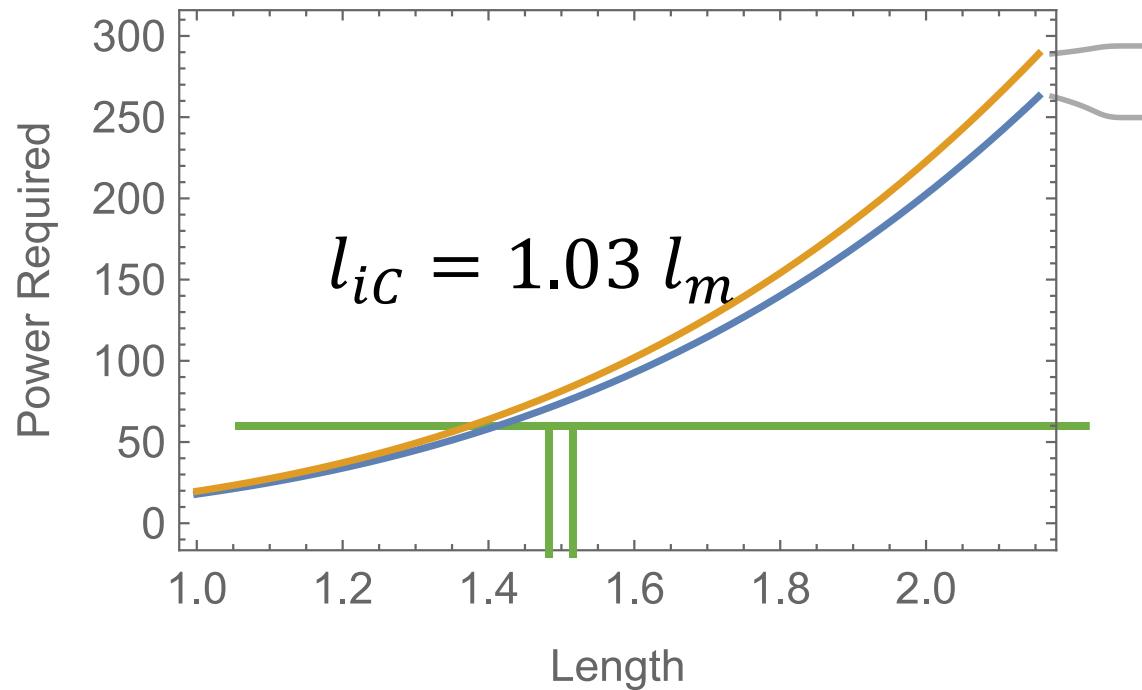
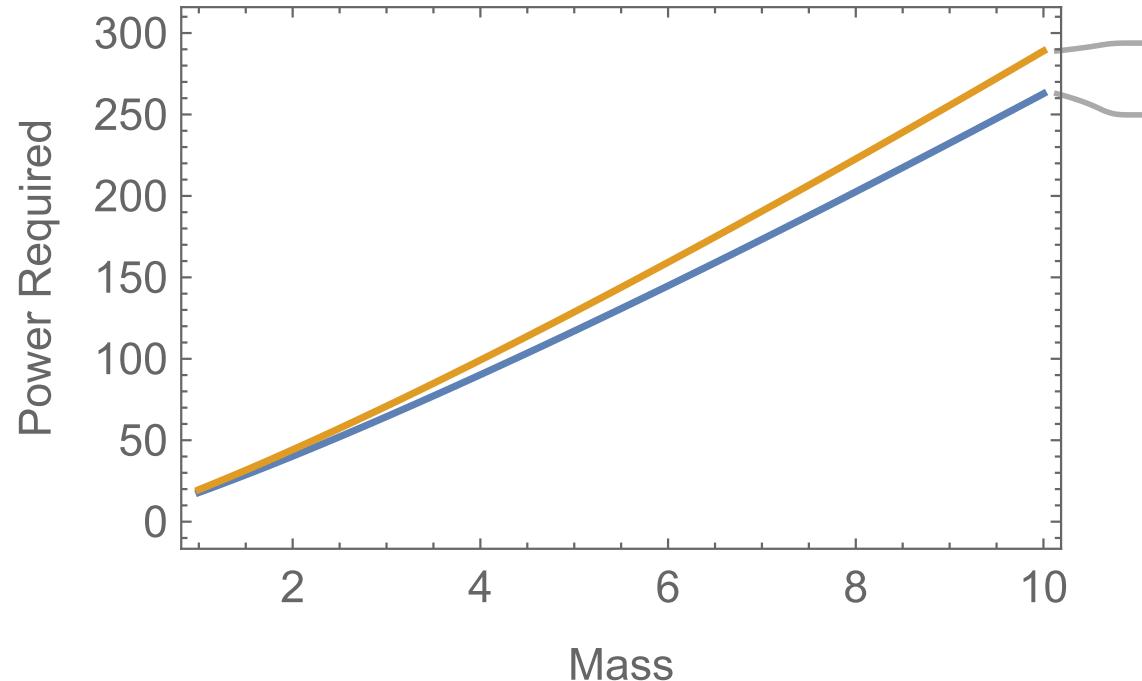
$$P_i = \sqrt{\frac{(mg)^3}{2 \rho \phi r^2}}$$

Induced Power  
required to  
support weight

$$\frac{P_{ic}}{P_{im}} = \frac{\sqrt{\frac{(mg)^3}{2 \rho_c \phi r^2}}}{\sqrt{\frac{(mg)^3}{2 \rho_m \phi r^2}}} = \sqrt{\frac{\rho_m}{\rho_c}} = \sqrt{\frac{1.29}{1.56}} = 0.9$$

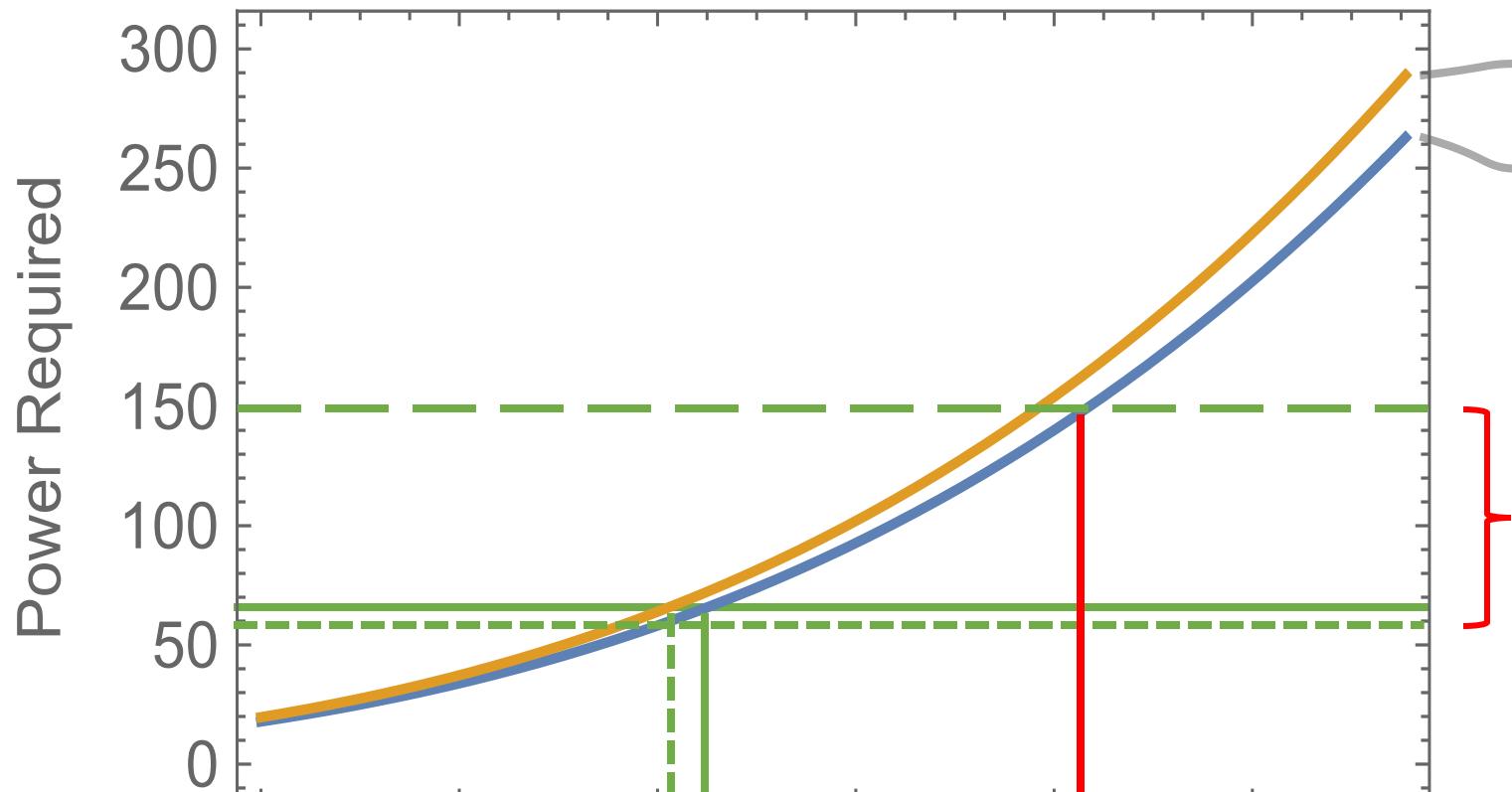
$$P_i = \sqrt{\frac{(k_1 l^3)^3}{2 \rho k_2 l^2}} \propto l^{\frac{7}{2}}$$

$$\frac{P_{ic}}{P_{im}} \propto \frac{l_{ic}^{\frac{7}{2}}}{l_m^{\frac{7}{2}}} \propto 0.9$$



low  $O_2$ , 21%  
high  $O_2$ , 35%

low  $O_2$ , 21%  
high  $O_2$ , 35%



$$l_{ic} = 1.03 l_m$$

Increase in length due to air density  
Aerodynamics

Length

low  $O_2$ , 21%  
high  $O_2$ , 35%

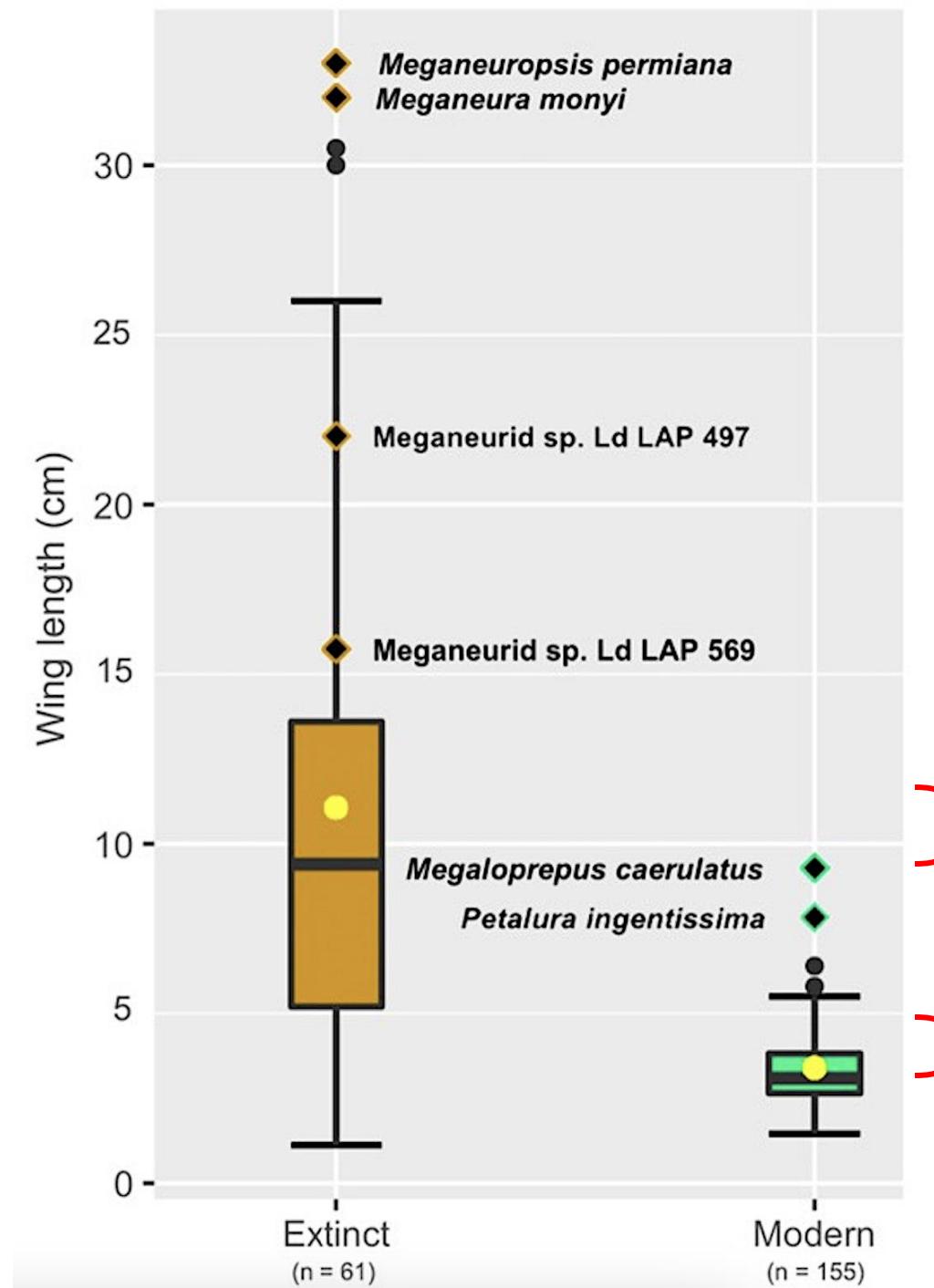
Extra power from  
extra oxygen

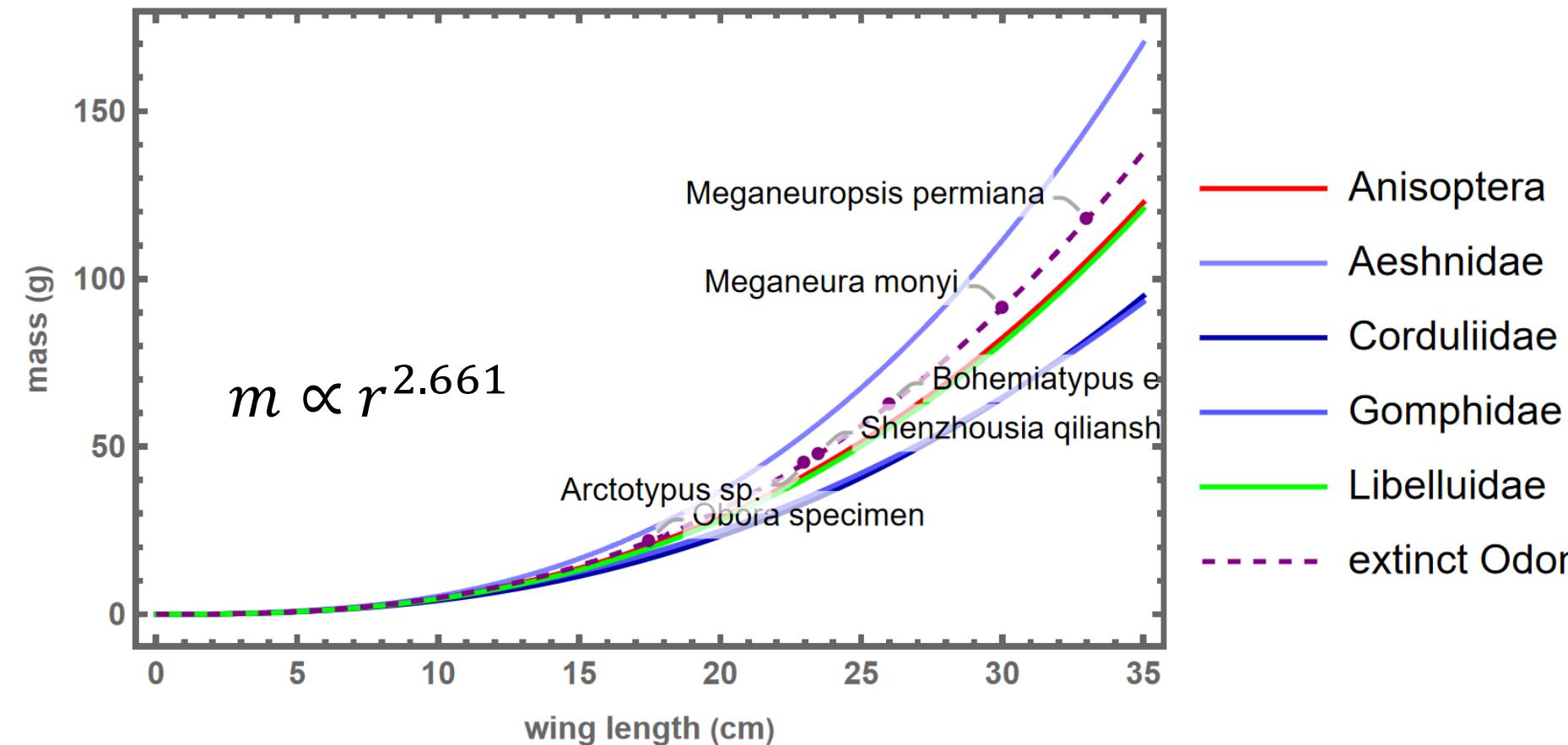
$$\frac{35\%}{21\%} = 1.9$$

$$\frac{P_{ic}}{P_{im}} \propto \frac{l_{ic}^{\frac{7}{2}}}{l_m^{\frac{7}{2}}} \propto 1.9$$

$$l_{ic} = 1.2 l_m$$

Increase in length due to fuel, 35%  $O_2$   
Energy Density





Scaling is anisometric

$$l_c = l_m \left( \frac{P_{i,C}}{P_{i,m}} \right)^{\frac{2}{3*2.661-2}} \left( \frac{\rho_C}{\rho_m} \right)^{\frac{1}{3*2.661-2}}$$

$$= l_m \left( \frac{35}{21} \right)^{\frac{2}{5.983}} \left( \frac{1.56}{1.29} \right)^{\frac{1}{5.983}}$$

$$l_c = 1.22 l_m$$

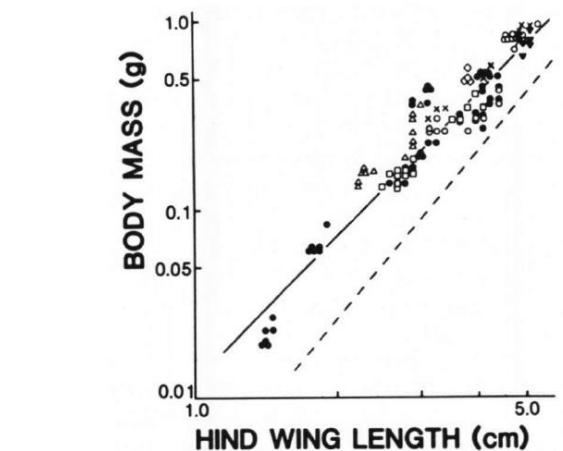
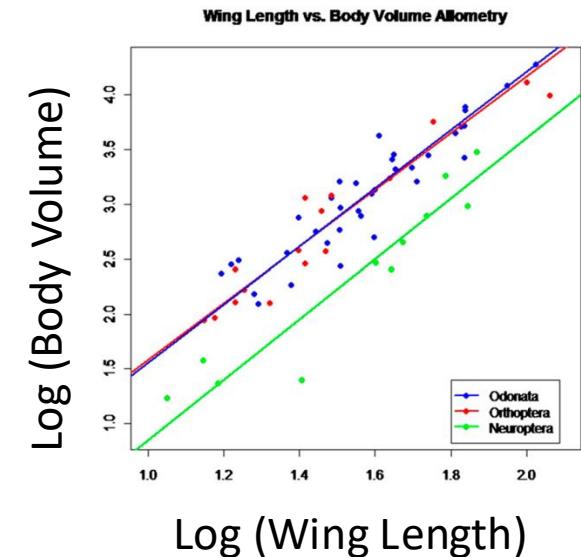
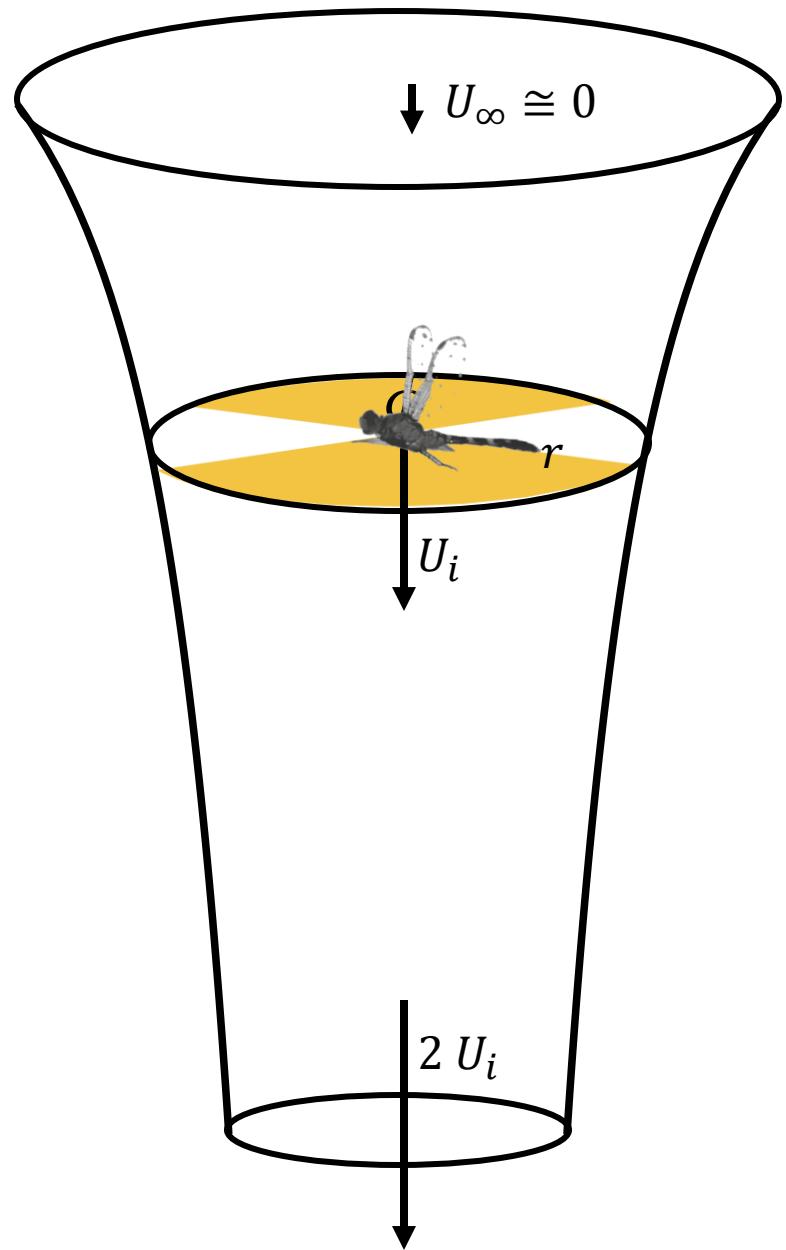


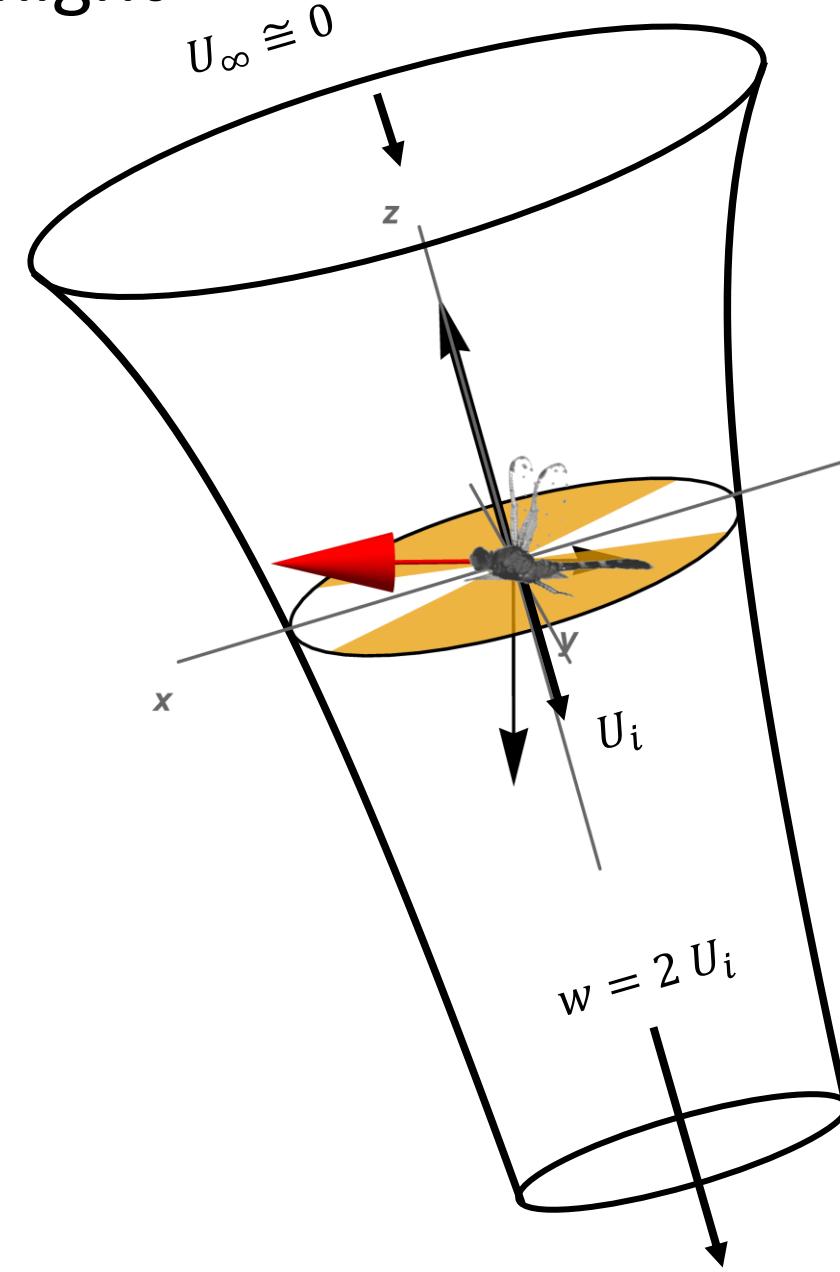
Fig. 2. Total mass ( $M$ ) as a function of hindwing length ( $L_{hw}$ ) in male Anisoptera. Symbols as in Fig. 1.



# Hovering flight



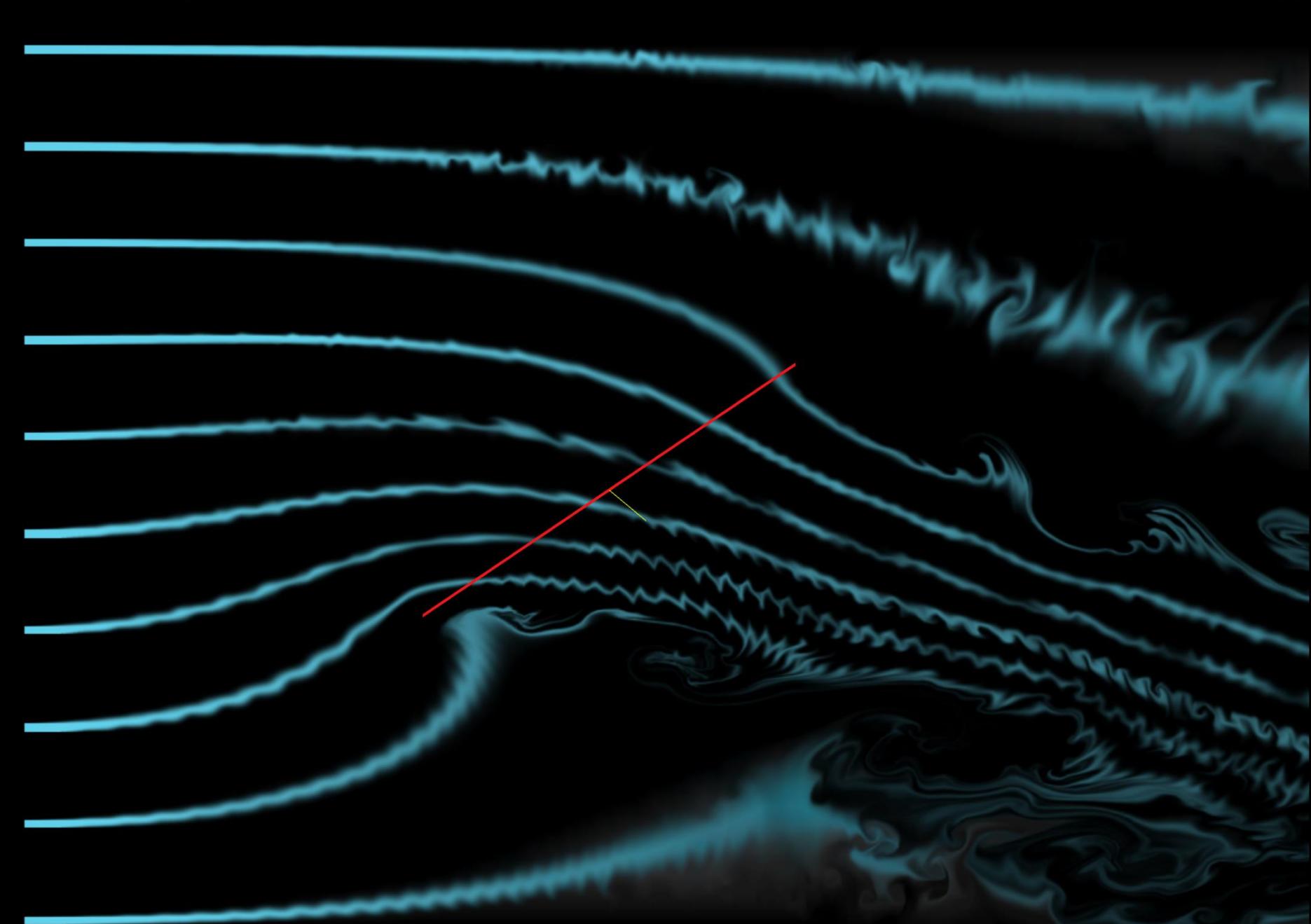
# Level flight





Thomas et al. 2004 JEB



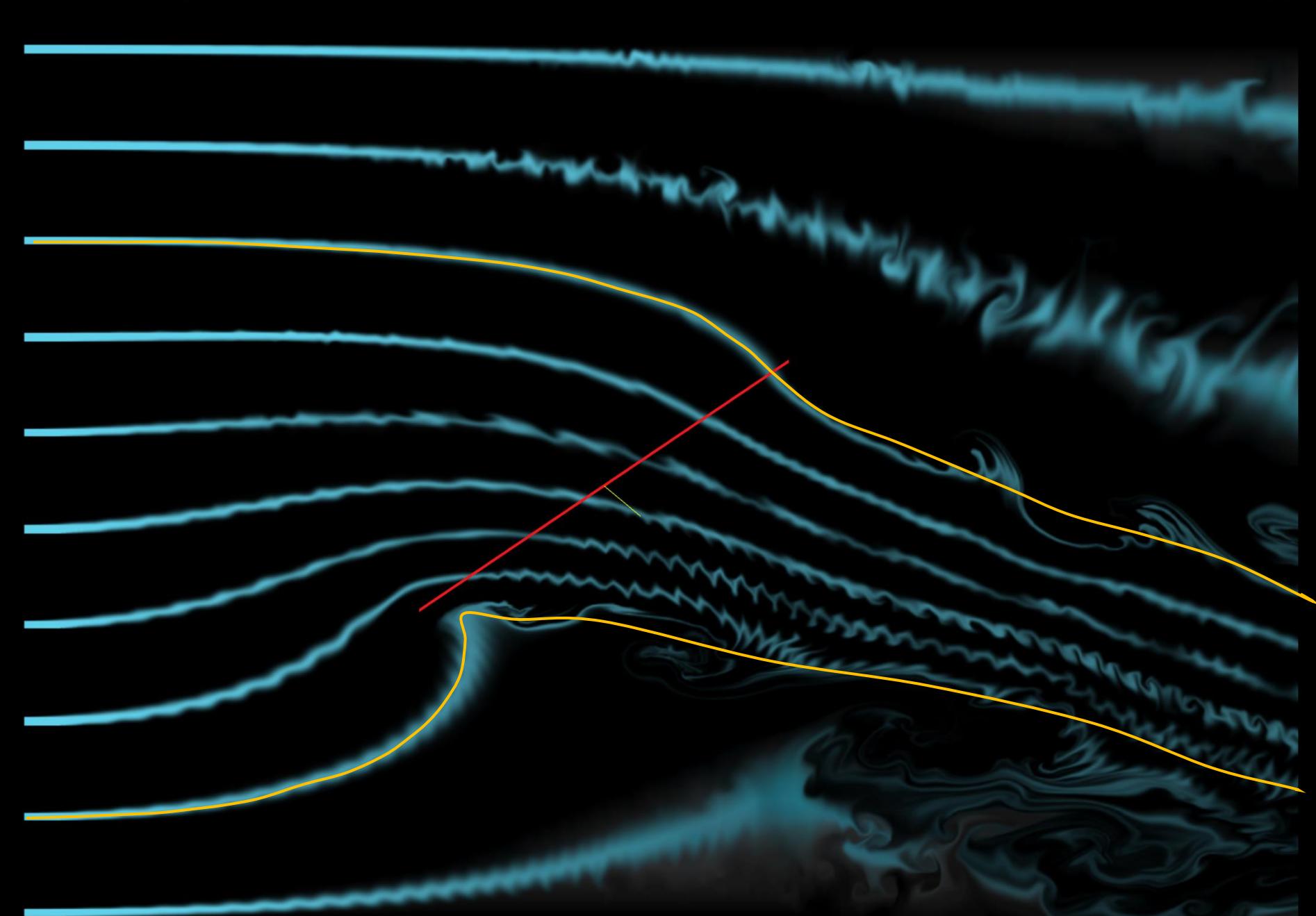


CFD  
Simulation

Streaklines  
Blue

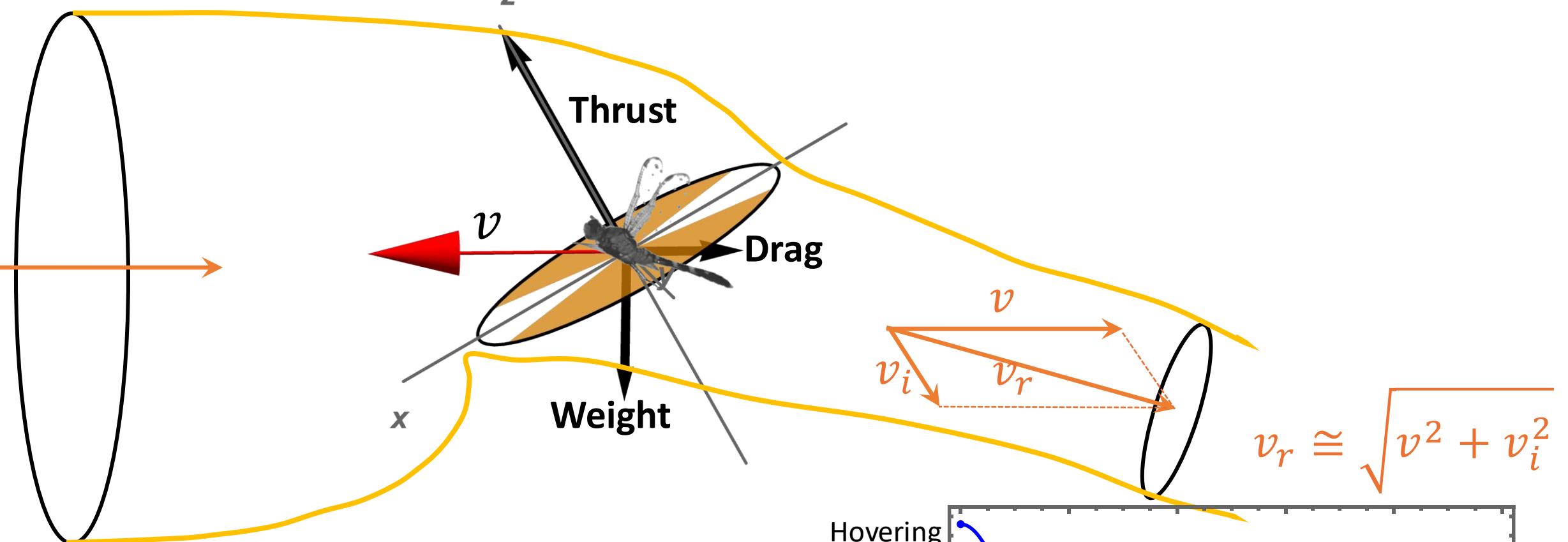
Propeller  
red

Wind tunnel ap



Wind tunnel ap

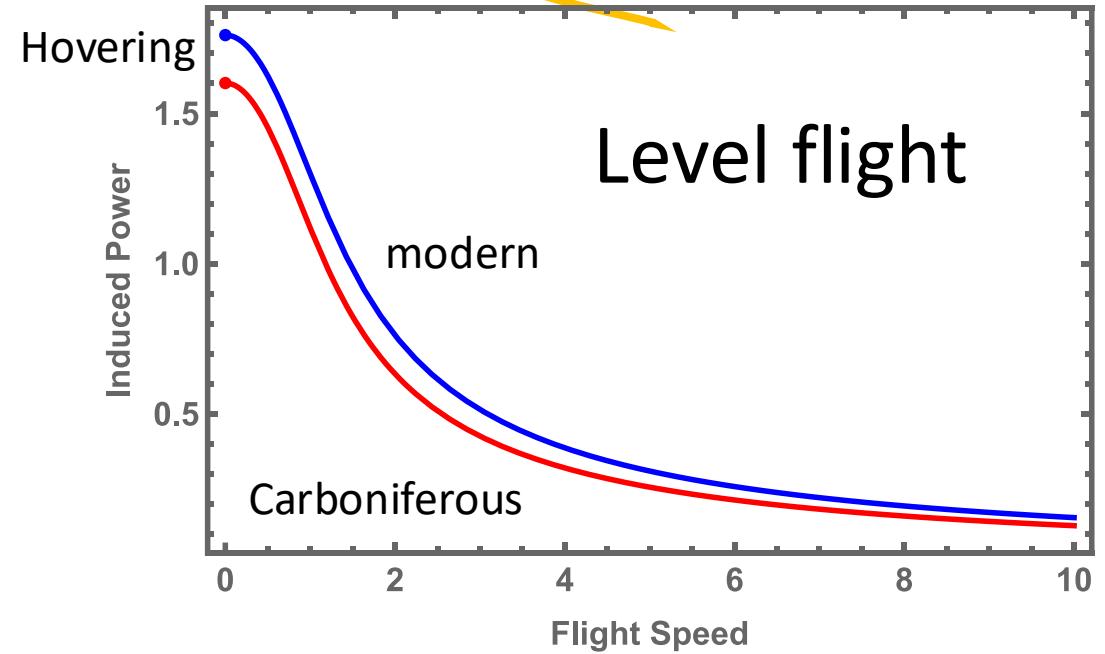
"Tube" of  
flow  
yellow  
lines



Conventional rotor in forward flight:

$$v_i = \frac{T}{2\rho\phi r^2 v_r} = \frac{T}{2\rho\phi r^2 \sqrt{v^2 + v_i^2}}$$

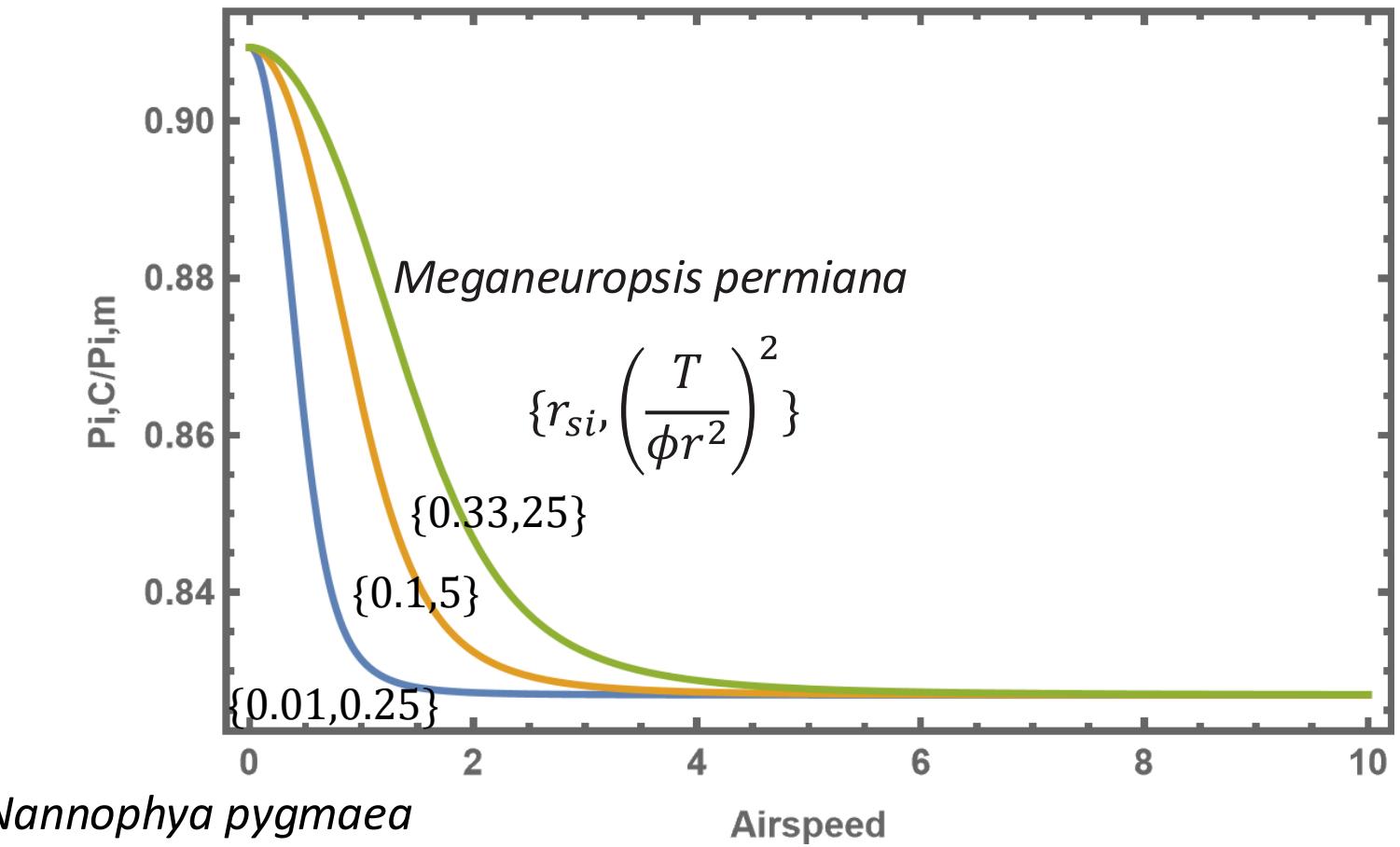
$$P_i = T v_i = \frac{T}{\sqrt{2}} \sqrt{-v^2 + \sqrt{\left(\frac{T}{\rho\phi r^2}\right)^2 + v^4}}$$



$$\frac{P_{i,c}}{P_{i,m}} = \frac{l_c^3 g \sqrt{-v^2 + \sqrt{\left(\frac{l_c^3 g}{\rho_c l_c^2}\right)^2 + v^4}}}{l_m^3 g \sqrt{-v^2 + \sqrt{\left(\frac{l_m^3 g}{\rho_m l_m^2}\right)^2 + v^4}}}.$$

$$\lim_{v \rightarrow 0} \frac{P_{i,c}}{P_{i,m}} = \frac{l_c^{7/2} \rho_m^{1/2}}{l_m^{7/2} \rho_c^{1/2}} = 0.91 \frac{l_c^{7/2}}{l_m^{7/2}},$$

$$\lim_{v \rightarrow \infty} \frac{P_{i,c}}{P_{i,m}} = \frac{l_c^4 \rho_m}{l_m^4 \rho_c} = 0.83 \frac{l_c^4}{l_m^4},$$



Therefore: How much bigger?

$$l_c = l_m \left( \frac{P_{i,c}}{P_{i,m}} \right)^{1/4} \left( \frac{\rho_c}{\rho_m} \right)^{1/4}$$

$$= l_m (1)^{1/4} \left( \frac{1.56}{1.29} \right)^{1/4} = 1.05 l_m$$

**Take Home Haiku:**

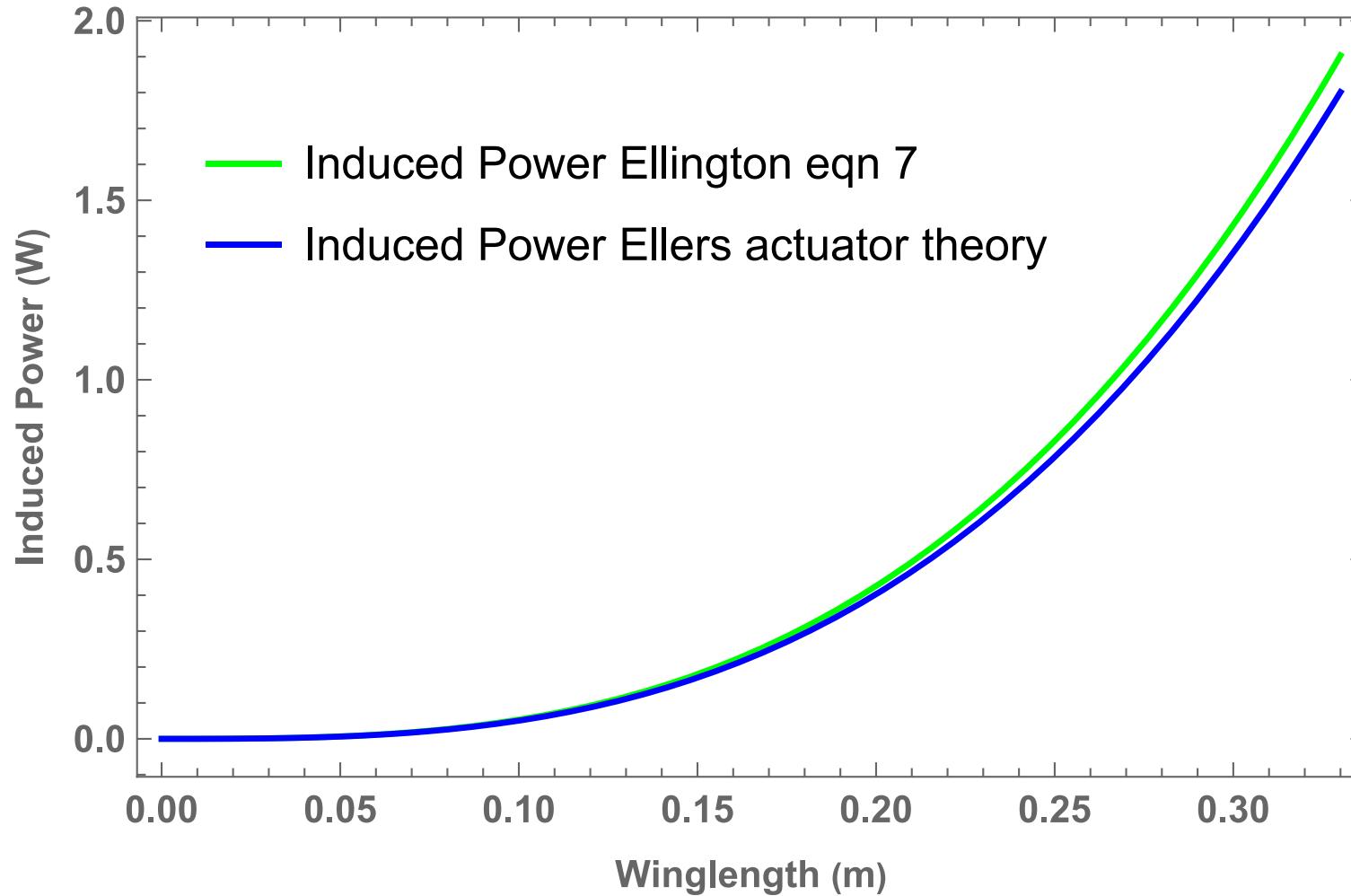
**When in level flight,  
high density atmosphere  
Five percent longer**

# Conclusions

Griffenflies could be:

- 1.03 times longer while hovering due to higher density air
- 1.05 times longer during level flight
- 1.20 times longer due to higher density plus higher oxygen for fuel
- 1.23–1.25 times longer due to the above plus anisometry





Ellington, C . P . 1999. "The Novel Aerodynamics of Insect Flight : Applications to Micro - Air Vehicles ." Journal of Experimental Biology 202 (23) : 3439\,[Dash]48. [https : // doi . org/10.1242/jeb .202 .23 .3439 .](https://doi.org/10.1242/jeb.202.23.3439)

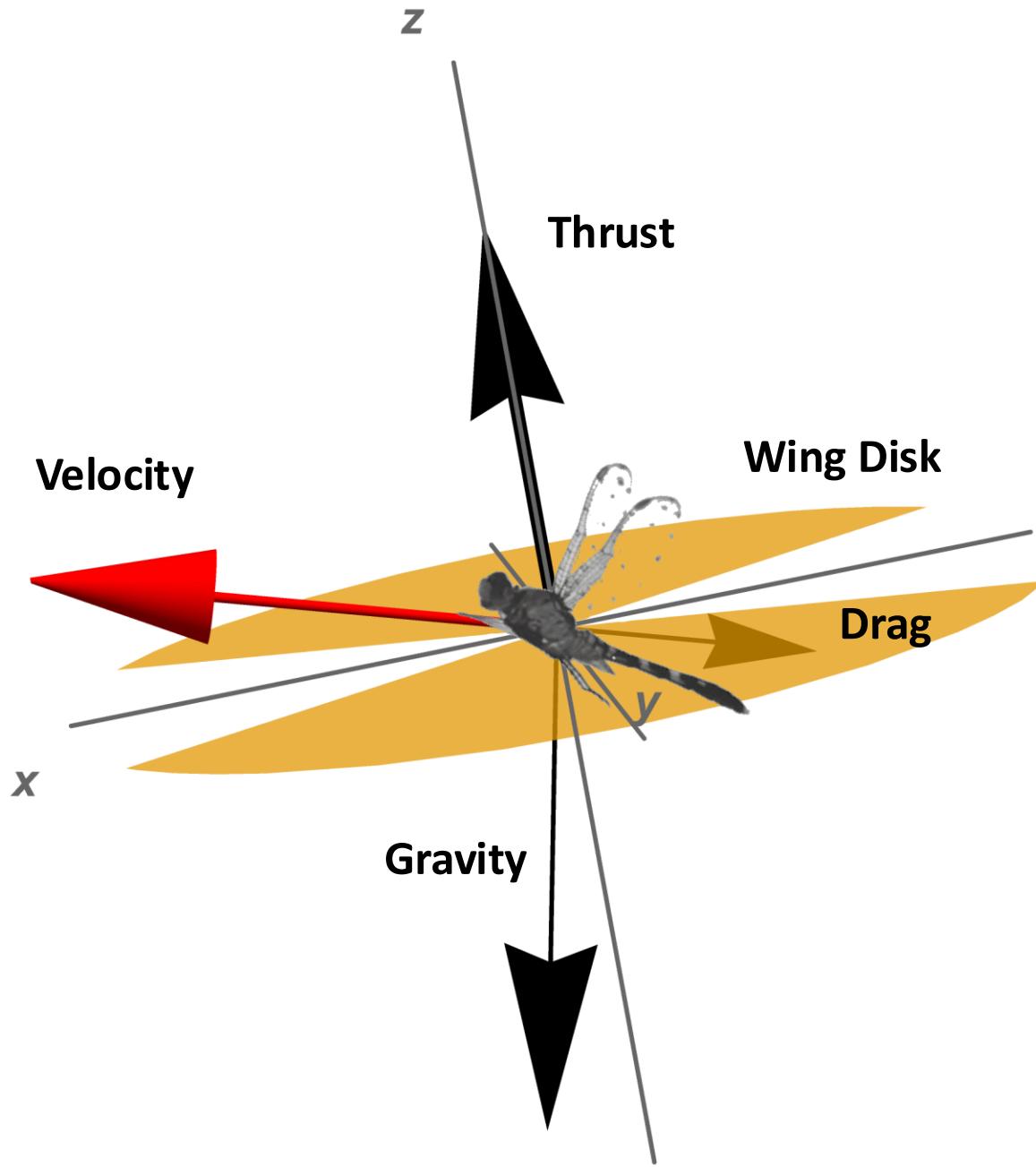
$$\pi = \sqrt{\frac{(mg)^3}{2\rho a}}$$

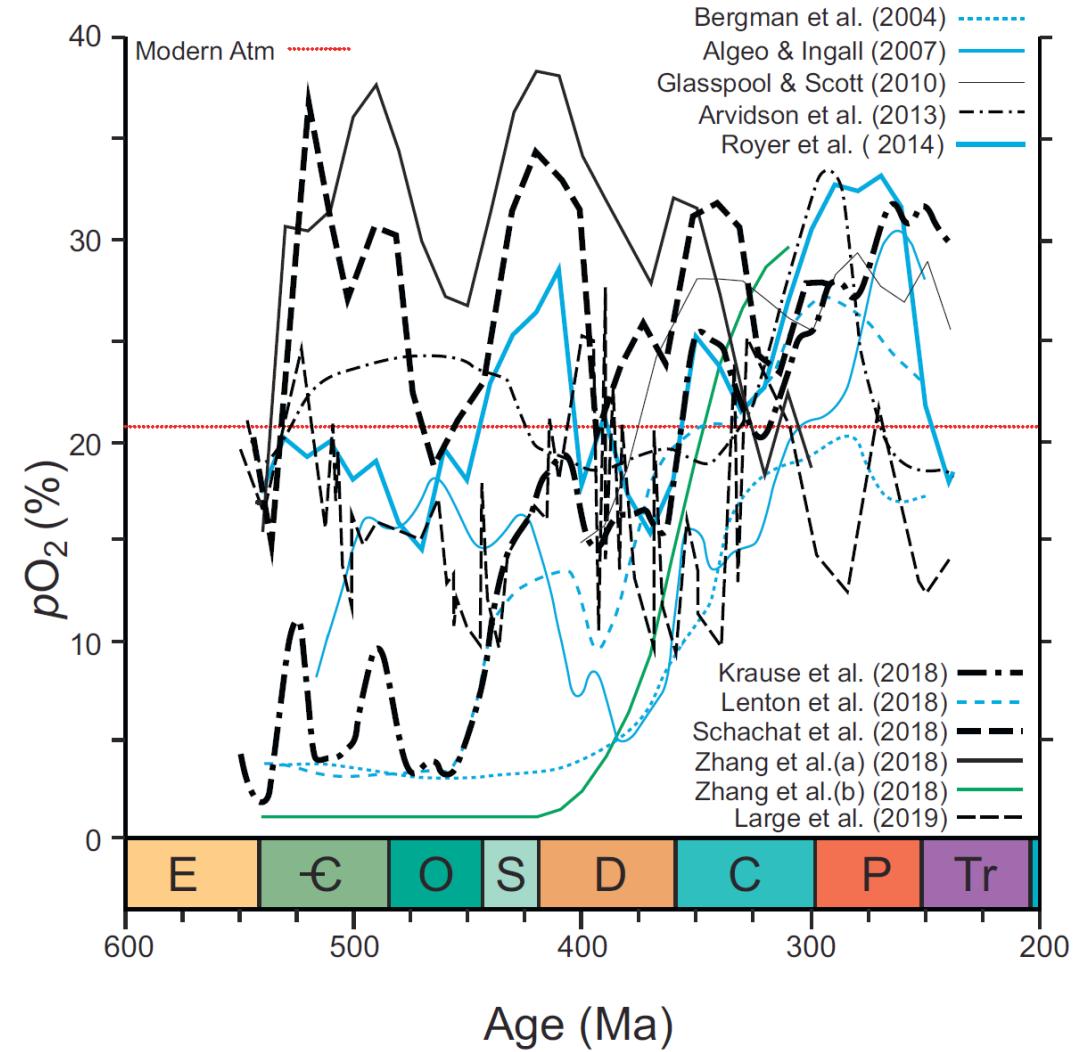
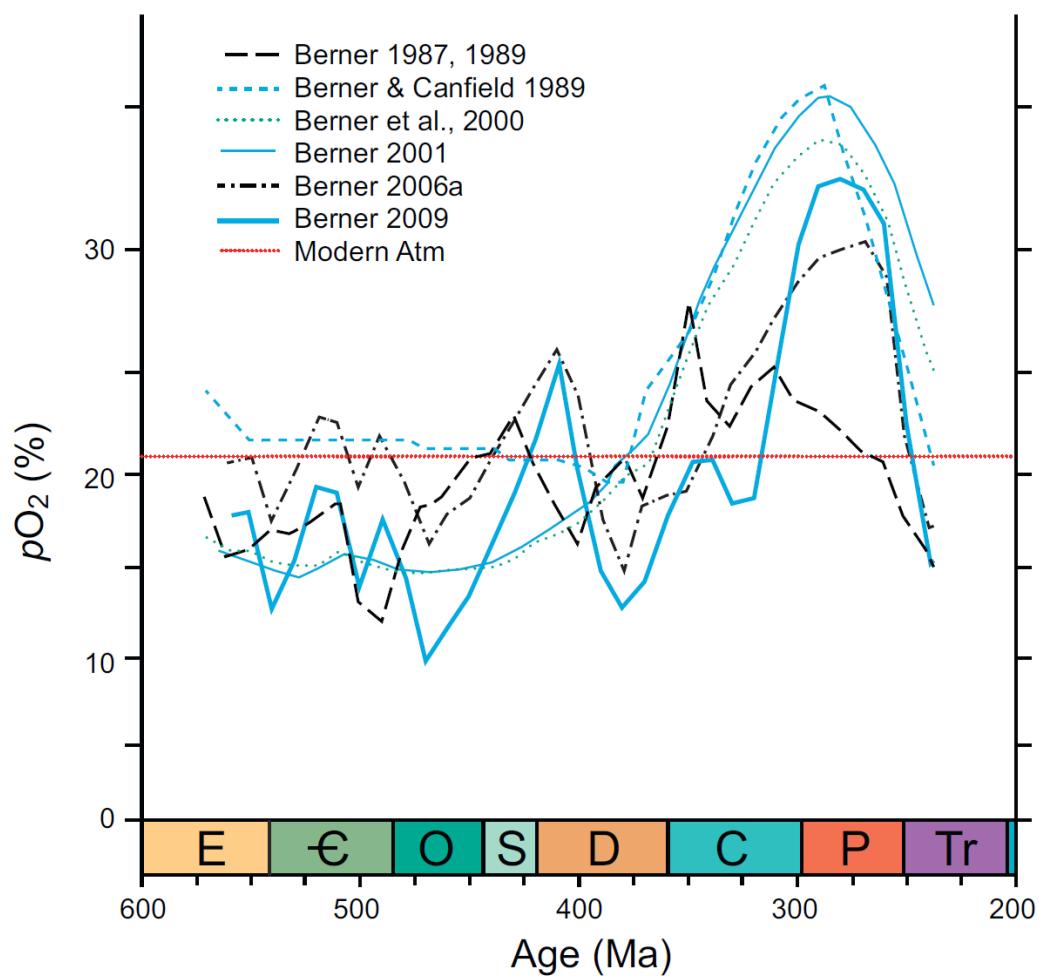
$$a = \Phi r^2$$

$$m_{hov} = 0.387 \frac{\Phi^2 n^2 r^4 cl}{ar}$$

$$P_{ind}^* = 14 n r \sqrt{\frac{\Phi cl}{ar}}$$

- $m_{hov}$  Hovering mass supported
- $m_{hov}$  Power induced per muscle mass
- $ar$  Aspect ratio of wing
- $cl$  Coefficient of lift
- $r$  Wing length
- $n$  Wing beat frequency





Considerable disagreement in estimates of past  $\% \text{O}_2$  and even less information about atmospheric density.