

# DESIGN DESCRIPTION OF THE MODEL

REV 0.1.0

January 5, 2018

## Contents

<b>1</b>	<b>Vessel model</b>	<b>2</b>
<b>2</b>	<b>Thruster model</b>	<b>3</b>
<b>3</b>	<b>Wind model</b>	<b>3</b>

# 1 Vessel model

$$\begin{aligned} \dot{\boldsymbol{\eta}} &= \mathbf{R}(\psi)\boldsymbol{\nu} \\ (\mathbf{M}_{RB} + \mathbf{M}_A)\dot{\boldsymbol{\nu}}_r + \mathbf{D}\boldsymbol{\nu}_r|\boldsymbol{\nu}_r| &= \boldsymbol{\tau}_{thr} + \boldsymbol{\tau}_{wind} + \boldsymbol{\tau}_{ext} \end{aligned} \quad (1)$$

The matrix  $\mathbf{M}_{RB}$  can be defined as

$$\mathbf{M}_{RB} = \begin{bmatrix} m & 0 & 0 \\ 0 & m & 0 \\ 0 & 0 & I_z \end{bmatrix}, \quad (2)$$

where  $m$  is the displacement, i.e. the mass of the displaced fluid, or the mass of the vessel, and  $I_z$  is the moment of inertia about the  $z_b$ -axis. The elements that are not on the diagonal of the matrix are ignored.

The added-mass matrix,  $\mathbf{M}_A$ , is calculated in the origin of the coordinate system, CO. This matrix can be written as

$$\mathbf{M}_A = \begin{bmatrix} -X_{\dot{u}} & 0 & 0 \\ 0 & -Y_{\dot{v}} & 0 \\ 0 & 0 & -N_{\dot{r}} \end{bmatrix}, \quad (3)$$

in SNAME notation.  $X_{\dot{u}}$  is added mass in surge,  $Y_{\dot{v}}$  is added mass in sway og  $N_{\dot{r}}$  is added mass ini yaw. The elements that are not on the diagonal of the matrix are ignored.

The matrix  $\mathbf{D}$  can, in SNAME notation, be written as

$$\mathbf{D} = \begin{bmatrix} -X_u & 0 & 0 \\ 0 & -Y_v & 0 \\ 0 & 0 & -N_r \end{bmatrix}, \quad (4)$$

where  $X_u$  is drag in surge,  $Y_v$  is drag in sway and  $N_r$  is drag in yaw. The elements that are not on the diagonal of the matrix are ignored.

$\boldsymbol{\tau}_{thr} = [\tau_{thr,X}, \tau_{thr,Y}, \tau_{thr,N}]^\top$  are forces from thrusters in surge, sway and yaw.  $\boldsymbol{\tau}_{wind} = [\tau_{wind,X}, \tau_{wind,Y}, \tau_{wind,N}]^\top$  and  $\boldsymbol{\tau}_{ext} = [\tau_{ext,X}, \tau_{ext,Y}, \tau_{ext,N}]^\top$  are wind forces and external forces (pipe, winch, etc.) that affects the vessel.

Rotation from vessel coordinates (BODY) to Earth coordinates (NED) can be done with a rotation matrix,  $\mathbf{R}(\psi)$ . For three degrees of freedom, this can be written as

$$\mathbf{R}(\psi) = \begin{bmatrix} \cos(\psi) & -\sin(\psi) & 0 \\ \sin(\psi) & \cos(\psi) & 0 \\ 0 & 0 & 1 \end{bmatrix}, \quad (5)$$

where  $\psi$  is the heading of the vessel.  $\boldsymbol{\eta} = [N, E, \psi]^\top$  is the position in North, East and heading.  $\boldsymbol{\nu} = [u, v, r]^\top$  is velocity in surge, sway and yaw.  $\boldsymbol{\nu}_c = [u_c, v_c, r_c]^\top$  is current velocity in surge, sway and yaw.  $\boldsymbol{\nu}_r = \boldsymbol{\nu} - \boldsymbol{\nu}_c$  is then relative vessel velocity.

## 2 Thruster model

The force that a single thruster can give can be written as

$$T = K_T \rho D^4 n^2, \quad (6)$$

where  $T$  is the thruster force,  $\rho$  is the density of water,  $D$  is the diameter of the propeller and  $n$  is the rpm of the propeller.  $K_T$  is an empirical value that is dependent on water speed into the propeller and the pitch angle of the propeller. A simplified version can be written as

$$K_T = K \cdot \theta^\alpha, \quad (7)$$

where  $K$  is a constant,  $\theta$  is pitch angle normalized to  $0 \rightarrow 1$  ( $0\% \rightarrow 100\%$ ) and  $\alpha$  is a constant. Because  $K$ ,  $\rho$  og  $D$  are constants they can be merged together into one constant.  $n$  can also be normalized such that it's between 0 and 1:  $n_n = K_n \cdot n$ . The final expression for  $T$  is then

$$\begin{aligned} T &= K_T \rho D^4 n^2 \\ &= K \theta^\alpha D^4 (K_n n_n)^2 \\ &= K \theta^\alpha D^4 K_n^2 n_n^2 \\ &= T_K \cdot \theta^\alpha n_n^2, \end{aligned} \quad (8)$$

where  $T_K = K D^4 K_n^2$ .

## 3 Wind model

$\tau_{wind}$  are wind forces that influence the vessel. The forces can be written as:

$$\tau_{wind} = \begin{bmatrix} q \cdot C_X \cdot A_f \\ q \cdot C_Y \cdot A_l \\ q \cdot C_N \cdot A_f \cdot Loa \end{bmatrix}, \quad (9)$$

where  $C_X$ ,  $C_Y$  og  $C_N$  are wind coefficients (drag coefficients) in surge, sway og yaw,  $A_f$  is projected frontal area and  $A_l$  is projected lateral area.  $q = \frac{1}{2} \cdot \rho_{air} \cdot V_{w,r}^2$ , where  $\rho_{air}$  is the density of air and  $V_r$  is relative wind velocity.  $C_X$ ,  $C_Y$  og  $C_N$  are typically found with empirical methods such as Blendermann. These are functions of relative wind velocity.