SAS Algorithm Pseudocode:

- 1. Find: sequence of slew maneuvers to avoid sun vector
 - (a) Check the sun vector intrusion
 - i. Find eigenaxis $\hat{e} = \frac{\hat{P}_i \times \hat{P}_f}{|\hat{P}_i \times \hat{P}_f|}$
 - ii. Compute $\alpha = \frac{\pi}{2} \cos^{-1}(\hat{S} \cdot_{\mathcal{N}} \hat{e})$
 - iii. IF $|\alpha| < \epsilon_p$, THEN find $\vec{S}_{||} = \hat{S} \cos \alpha$
 - (b) Compute ϕ_1 : $\phi_1 = \cos^{-1}(\hat{P}_i \cdot_{\mathcal{G}} \hat{S}_{||}) \epsilon_{\mathcal{D}}$
 - (c) Compute ϕ_2 :
 - i. IF $\alpha \neq 0$, THEN $\phi_2 = 2\sin^{-1}\left(\frac{\sin\epsilon_p}{\sin\theta}\right)$, $\theta = \cos^{-1}(\hat{P}_1 \cdot \hat{S})$
 - ii. IF $\alpha = 0$, THEN $\phi_2 = \pi$
 - (d) Compute ϕ_3 : $\phi_3 = \cos^{-1}(_G \hat{P}_f.\hat{P}_2)$
- 2. Find: commanded angular velocity, angular acceleration, and quaternion
 - (a) Compute $\phi_{tt} = \frac{\dot{\phi}_{max}^2}{\ddot{\phi}_{max}}$
 - (b) Compute t_1 , t_2 , and t_f .

IF $\phi > \phi_{tt}$, THEN:

$$\begin{split} t_1 &= t_0 + \frac{\dot{\phi}_{max} - \dot{\phi}_0}{\ddot{\phi}_{max}} \\ t_2 &= t_1 + \frac{1}{\dot{\phi}_{max}} \Big[\phi_f - \dot{\phi}_0(t_1 - t_0) - \frac{1}{2} \ddot{\phi}_{max} (t_f 1 - t_0)^2 - \frac{\dot{\phi}_{max} (\dot{\phi}_{max} - \dot{\phi}_f)}{\ddot{\phi}_{max}} + \frac{(\dot{\phi}_{max} - \dot{\phi}_f)^2}{2\ddot{\phi}_{max}} \Big] \\ t_f &= t_1 + \frac{1}{\dot{\phi}_{max}} \Big[\phi_f - \dot{\phi}_0(t_1 - t_0) - \frac{1}{2} \ddot{\phi}_{max} (t_1 - t_0)^2 + \frac{(\dot{\phi}_{max} - \dot{\phi}_f)^2}{2\ddot{\phi}_{max}} \Big]. \end{split}$$

$$t_f = t_1 + \frac{1}{\dot{\phi}_{max}} \left[\phi_f - \dot{\phi}_0(t_1 - t_0) - \frac{1}{2} \ddot{\phi}_{max} (t_1 - t_0)^2 + \frac{(\dot{\phi}_{max} - \dot{\phi}_f)^2}{2\ddot{\phi}_{max}} \right].$$

$$t_f = \sqrt{\frac{\dot{\phi}_{max}^2}{\ddot{\phi}_{max}}}$$

$$t_2 = t_f/2$$

$$t_1 = t_2$$

- (c) Find ${}^{D}R^{N}$: ${}^{D}R^{N} = [(\cos\phi)I_{3x3} + (1 \cos\phi)\hat{e}\hat{e}^{T} (\sin\alpha)E^{x}]$
- (d) Find $\beta \dot{\omega}^D$: $\beta \dot{\omega}^D = {}^D R^N \ddot{\phi}_{max} \cdot_{\mathcal{N}} \hat{e}$
- (e) Solve for control torque, $u: J \cdot_{\mathcal{B}} \dot{\omega}^D = u -_{\mathcal{B}} \omega^C \times J \cdot_{\mathcal{B}} \omega^C$
- (f) FOR each ϕ between switching times, propagate ω and q between switching times by solving above eqn and $\dot{q} = \frac{1}{2}\Omega q$ where

$$\Omega = \begin{bmatrix} 0 & \omega_3 & -\omega_2 & \omega_1 \\ -\omega_3 & 0 & -\omega_1 & \omega_2 \\ \omega_2 & -\omega_1 & 0 & \omega_3 \\ -\omega_1 & \omega_2 & -\omega_3 & 0 \end{bmatrix}$$

with correct u for each switching time interval.