

Lecture 15: PDE-constrained optimization

AME60714: Advanced Numerical Methods

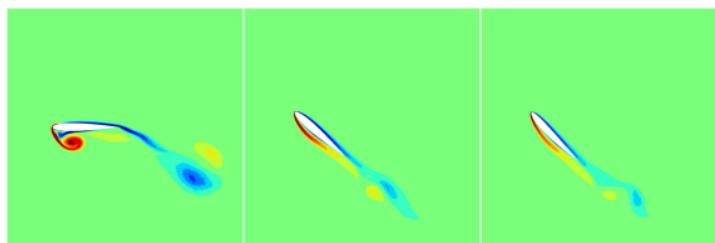
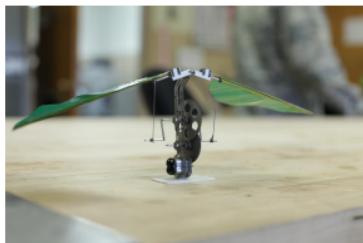
Matthew J. Zahr
Aerospace and Mechanical Engineering
University of Notre Dame

PDE optimization is ubiquitous in science and engineering

Design: Find system that optimizes performance metric, satisfies constraints



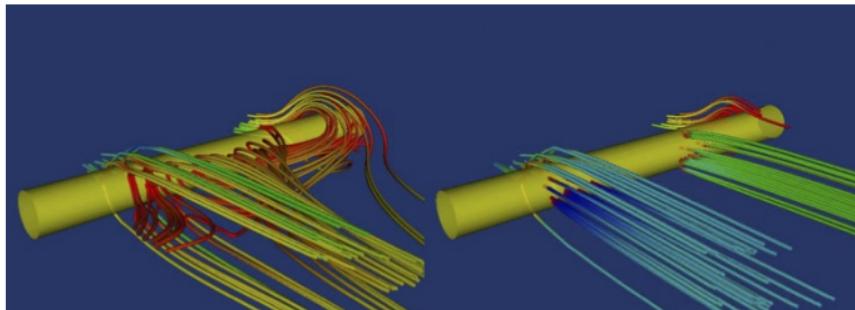
Aerodynamic shape design of automobile



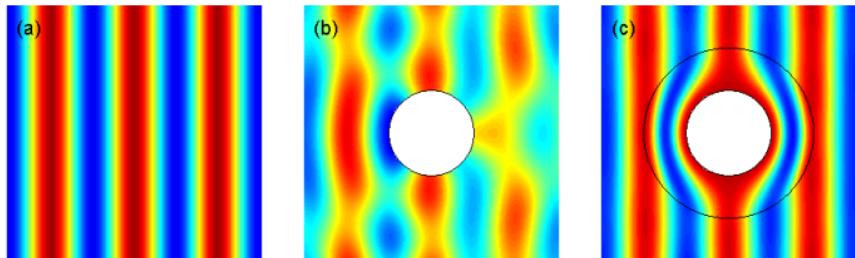
Optimal flapping motion of micro aerial vehicle

PDE optimization is ubiquitous in science and engineering

Control: Drive system to a desired state



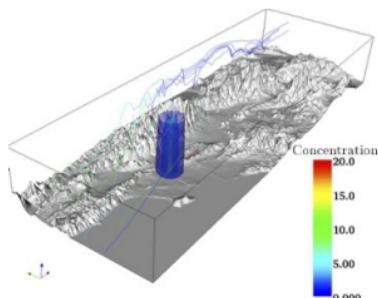
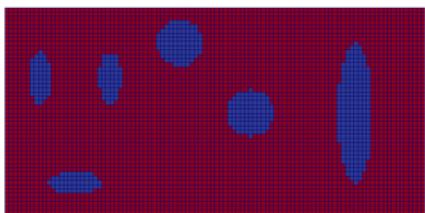
Boundary flow control



Metamaterial cloaking – electromagnetic invisibility

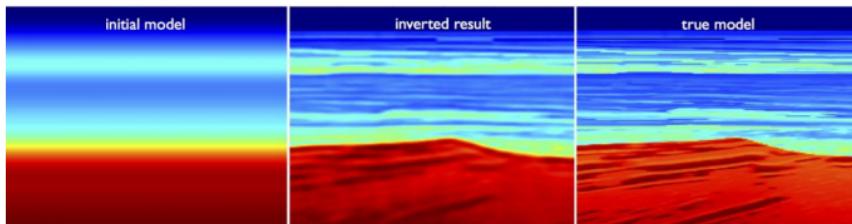
PDE optimization is ubiquitous in science and engineering

Inverse problems: Infer the problem setup given solution observations



Material inversion: find inclusions from acoustic, structural measurements

Source inversion: find source of contaminant from downstream measurements



Full waveform inversion: estimate subsurface of crust from acoustic measurements

Unsteady PDE-constrained optimization formulation

Goal: Find the solution of the *unsteady PDE-constrained optimization* problem

$$\underset{\boldsymbol{U}, \boldsymbol{\mu}}{\text{minimize}} \quad \mathcal{J}(\boldsymbol{U}, \boldsymbol{\mu})$$

$$\text{subject to} \quad \boldsymbol{C}(\boldsymbol{U}, \boldsymbol{\mu}) \leq 0$$

$$\frac{\partial \boldsymbol{U}}{\partial t} + \nabla \cdot \boldsymbol{F}(\boldsymbol{U}, \nabla \boldsymbol{U}) = 0 \quad \text{in } v(\boldsymbol{\mu}, t)$$

$\boldsymbol{U}(\boldsymbol{x}, t)$

PDE solution

$\boldsymbol{\mu}$

design/control parameters

$$\mathcal{J}(\boldsymbol{U}, \boldsymbol{\mu}) = \int_{T_0}^{T_f} \int_{\Gamma} j(\boldsymbol{U}, \boldsymbol{\mu}, t) dS dt$$

objective function

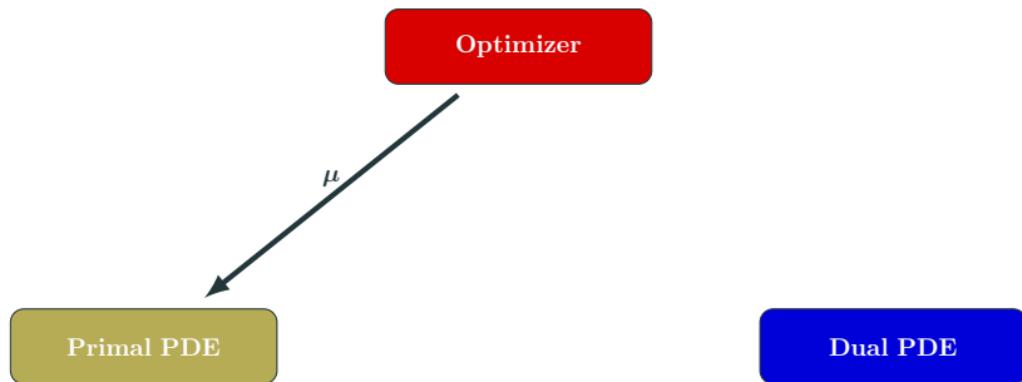
$$\boldsymbol{C}(\boldsymbol{U}, \boldsymbol{\mu}) = \int_{T_0}^{T_f} \int_{\Gamma} \mathbf{c}(\boldsymbol{U}, \boldsymbol{\mu}, t) dS dt$$

constraints

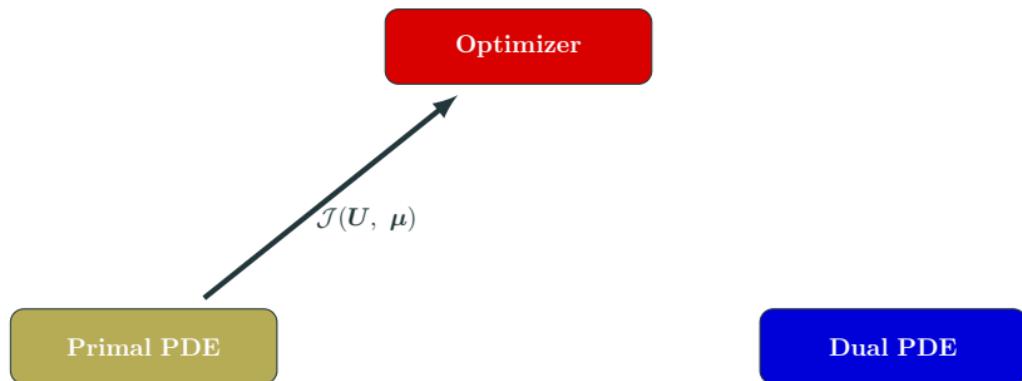
Nested approach to PDE-constrained optimization



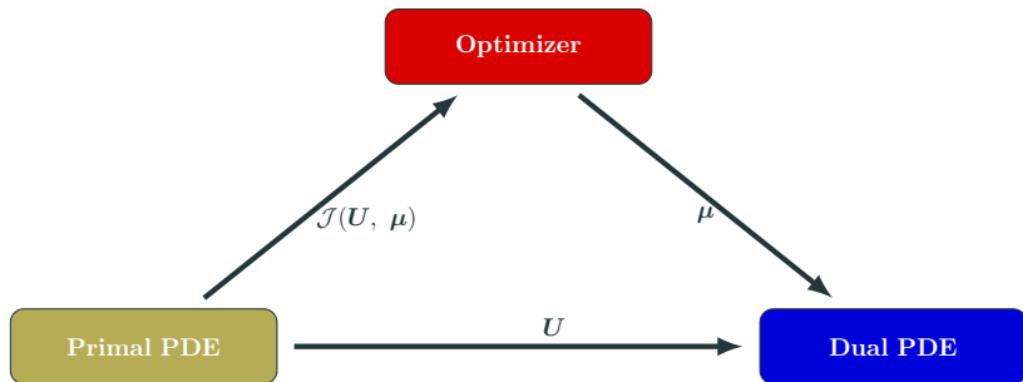
Nested approach to PDE-constrained optimization



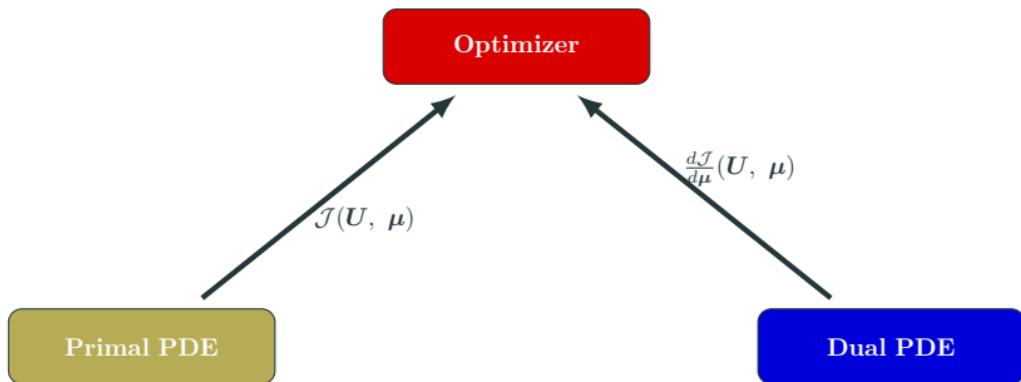
Nested approach to PDE-constrained optimization



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Nested approach to PDE-constrained optimization



High-order discretization of PDE-constrained optimization

- *Continuous* PDE-constrained optimization problem

$$\underset{\boldsymbol{U}, \boldsymbol{\mu}}{\text{minimize}} \quad \mathcal{J}(\boldsymbol{U}, \boldsymbol{\mu})$$

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$$\frac{\partial \boldsymbol{U}}{\partial t} + \nabla \cdot \boldsymbol{F}(\boldsymbol{U}, \nabla \boldsymbol{U}) = 0 \quad \text{in } v(\boldsymbol{\mu}, t)$$

- *Fully discrete* PDE-constrained optimization problem

$$\underset{\begin{array}{l} \boldsymbol{u}_0, \dots, \boldsymbol{u}_{N_t} \in \mathbb{R}^{N_u}, \\ \boldsymbol{k}_{1,1}, \dots, \boldsymbol{k}_{N_t,s} \in \mathbb{R}^{N_u}, \\ \boldsymbol{\mu} \in \mathbb{R}^{n_\mu} \end{array}}{\text{minimize}} \quad J(\boldsymbol{u}_0, \dots, \boldsymbol{u}_{N_t}, \boldsymbol{k}_{1,1}, \dots, \boldsymbol{k}_{N_t,s}, \boldsymbol{\mu})$$

$$\text{subject to} \quad \boldsymbol{C}(\boldsymbol{u}_0, \dots, \boldsymbol{u}_{N_t}, \boldsymbol{k}_{1,1}, \dots, \boldsymbol{k}_{N_t,s}, \boldsymbol{\mu}) \leq 0$$

$$\boldsymbol{u}_0 - \boldsymbol{g}(\boldsymbol{\mu}) = 0$$

$$\boldsymbol{u}_n - \boldsymbol{u}_{n-1} - \sum_{i=1}^s b_i \boldsymbol{k}_{n,i} = 0$$

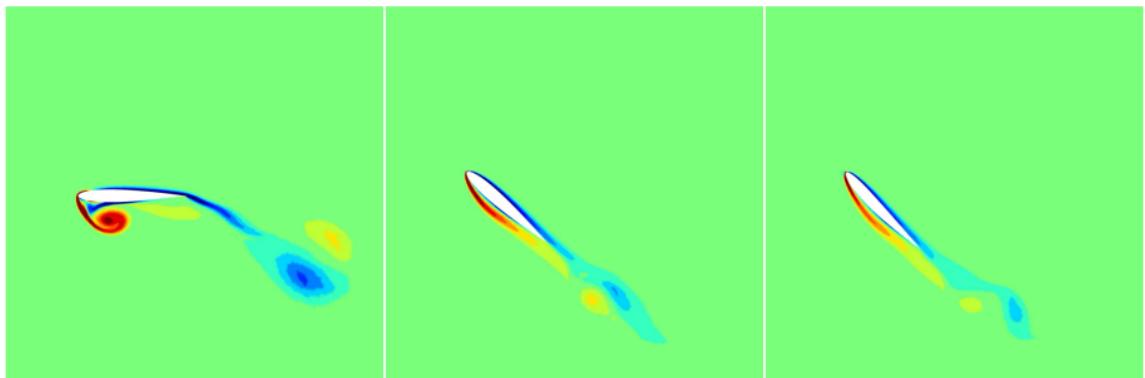
$$M \boldsymbol{k}_{n,i} - \Delta t_n \boldsymbol{r}(\boldsymbol{u}_{n,i}, \boldsymbol{\mu}, t_{n,i}) = 0$$

Optimal rigid body motion (RBM), time-morph geometry (TMG)

Energy = 9.4096
Thrust = 0.1766

Energy = 4.9476
Thrust = 2.500

Energy = 4.6182
Thrust = 2.500



Initial Guess

Optimal RBM
 $T_x = 2.5$

Optimal RBM/TMG
 $T_x = 2.5$

Energetically optimal flapping in three dimensions

Energy = 1.4459e-01
Thrust = -1.1192e-01

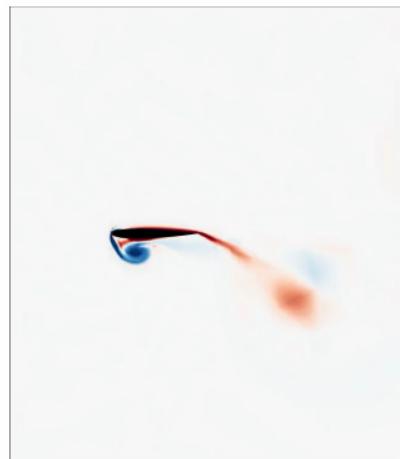


Energy = 3.1378e-01
Thrust = 0.0000e+00



Energetically optimal flapping vs. required thrust

Energy = 1.8445
Thrust = 0.06729



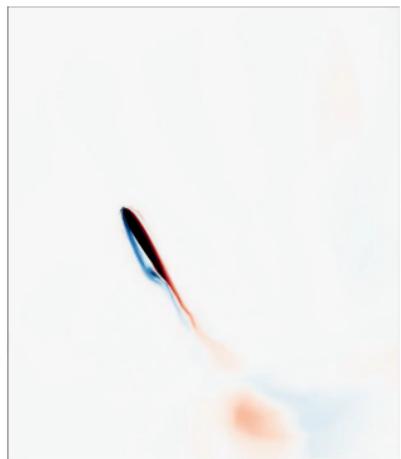
Initial Guess

Energy = 0.21934
Thrust = 0.0000



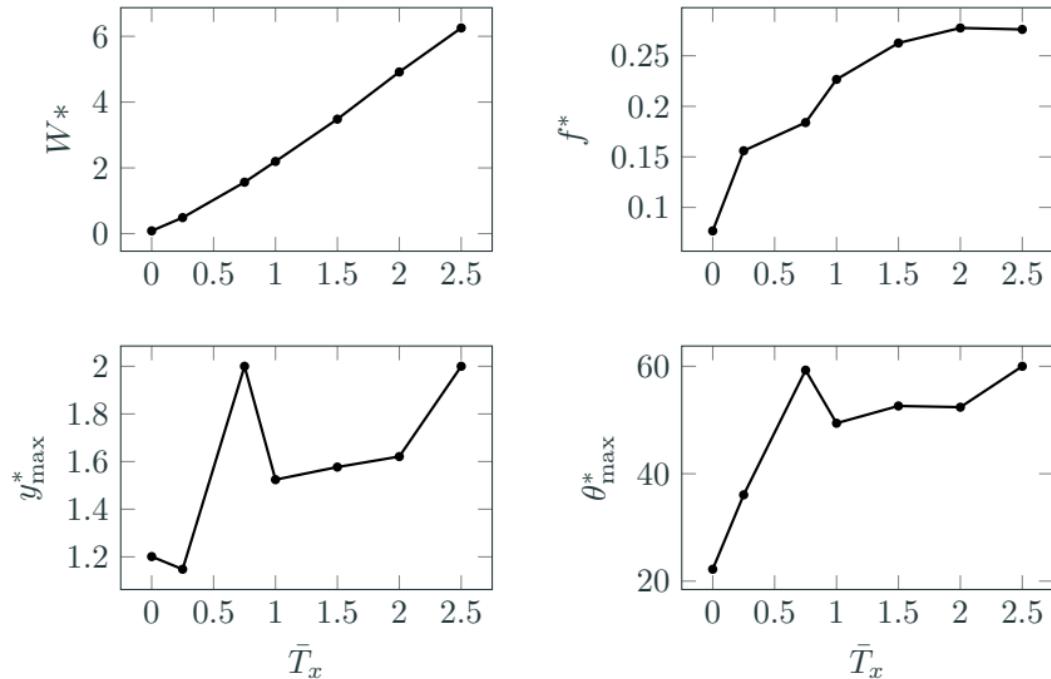
Optimal
 $T_x = 0$

Energy = 6.2869
Thrust = 2.5000



Optimal
 $T_x = 2.5$

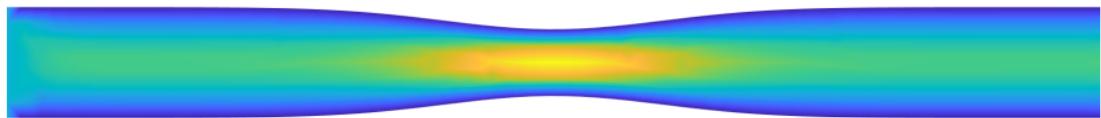
Energetically optimal flapping vs. required thrust: QoI



The optimal flapping energy (W^*), frequency (f^*), maximum heaving amplitude (y_{\max}^*), and maximum pitching amplitude (θ_{\max}^*) as a function of the thrust constraint \bar{T}_x .

Current MRI 4D flow reconstruction: insufficient for infants

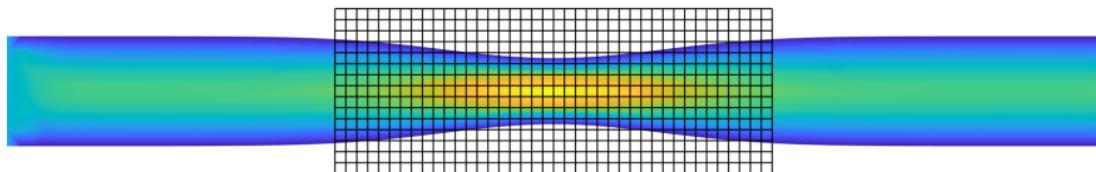
- Resolution: 3mm, 25-100ms in 10-20 minute scan
- Greater resolution = more noise, longer scan
- Biomarkers (WSS) must be computed from noisy velocity measurements
- Still very far from resolution needed for congenital heart disease



True *in vivo* flow

Current MRI 4D flow reconstruction: insufficient for infants

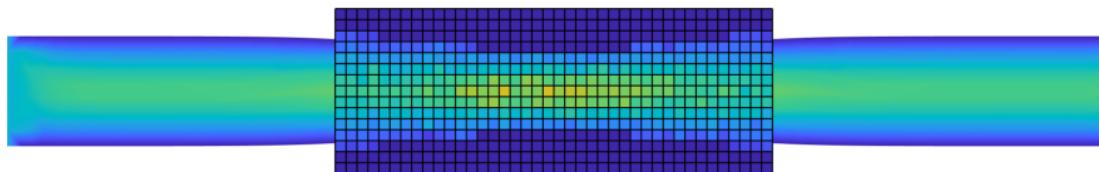
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MRI protocol

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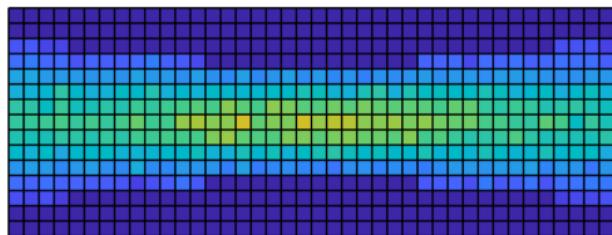
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MRI measurements: noisy, space-time averages over voxels

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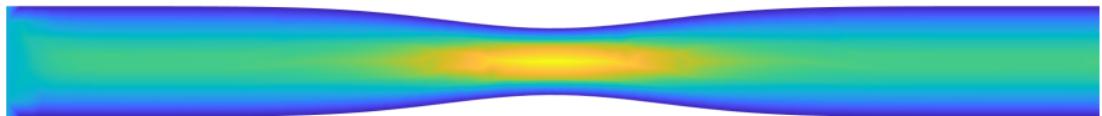
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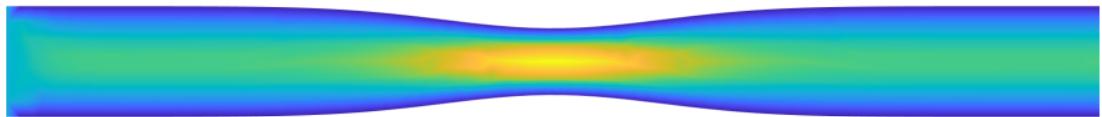
- Resolution: 3mm, 25-100ms in 10-20 minute scan
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Improve with CFD:

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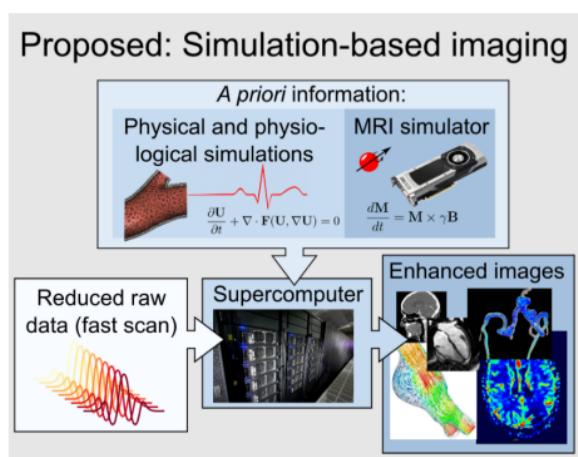
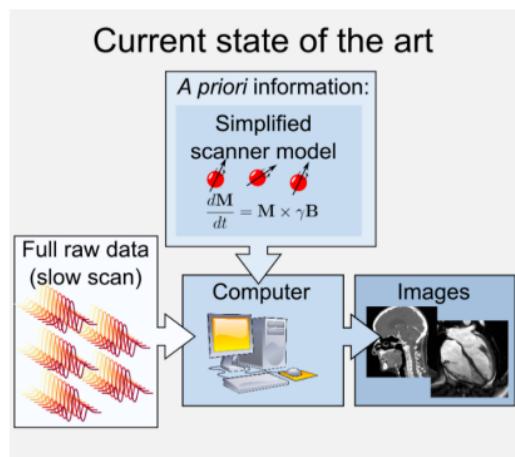


Improve with CFD: BCs? ICs? material properties?

In vivo image reconstruction using *all* prior knowledge

To break resolution-noise barrier, we incorporate all available information into reconstruction procedure

- geometry of patient-specific flow domain
- conservation of mass, momentum, energy (Navier-Stokes)
- low-resolution *in vivo* flow measurements (MRI data)



Simulation-based imaging

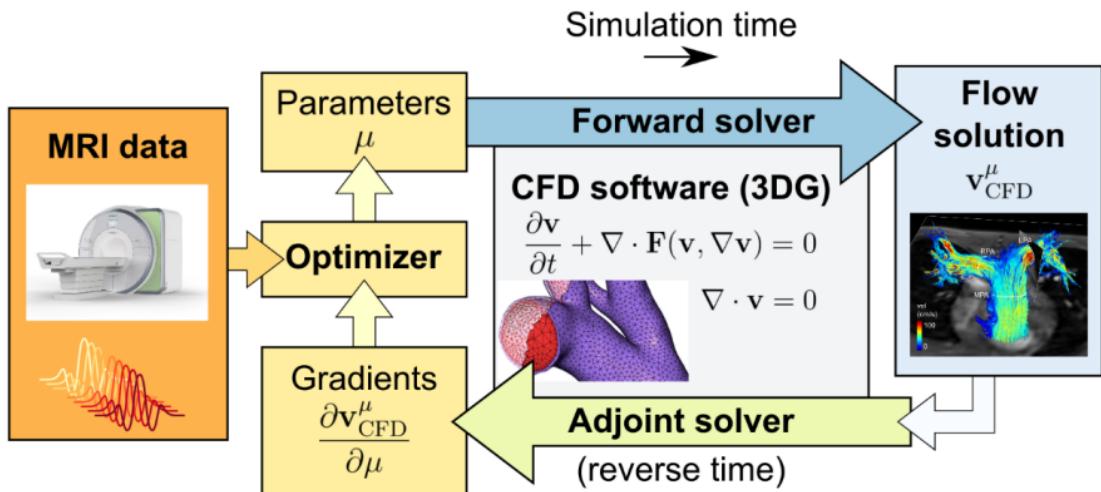
- Phase I: Collect MRI data to extract flow domain and 4D flow measurements
- Phase II: Image segmentation and mesh generation
- Phase III: Find Navier-Stokes solution that best explains flow data
(PDE-constrained optimization to minimize MRI data misfit)

$$\begin{aligned} & \underset{\boldsymbol{U}, \boldsymbol{\mu}}{\text{minimize}} \quad \mathcal{J}(\boldsymbol{U}) \\ & \text{subject to} \quad \frac{\partial \boldsymbol{U}}{\partial t} + \nabla \cdot \boldsymbol{F}(\boldsymbol{U}, \nabla \boldsymbol{U}, \boldsymbol{\mu}) = 0 \end{aligned}$$

\boldsymbol{U}	:	PDE solution
$\boldsymbol{\mu}$:	BCs, IC, material properties
$\mathcal{J}(\boldsymbol{U})$:	CFD/MRI misfit function
$\boldsymbol{F}(\boldsymbol{U}, \nabla \boldsymbol{U}, \boldsymbol{\mu})$:	Navier-Stokes flux function

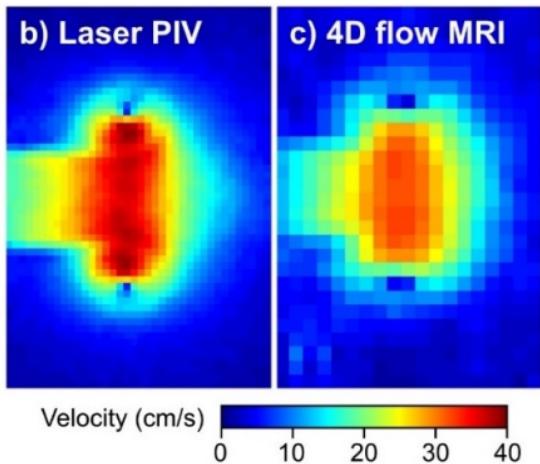
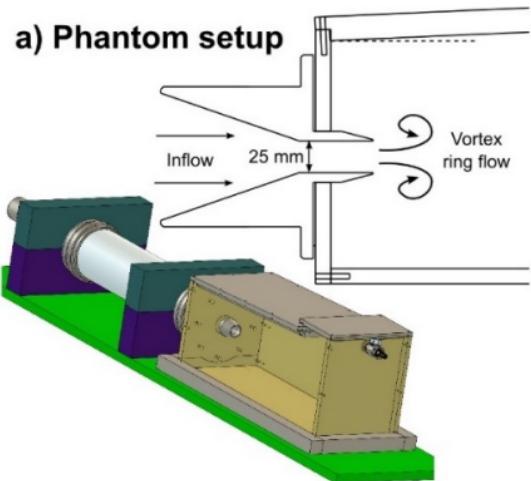
- Phase IV: Visualize solution and compute biomarkers

Simulation-based imaging workflow



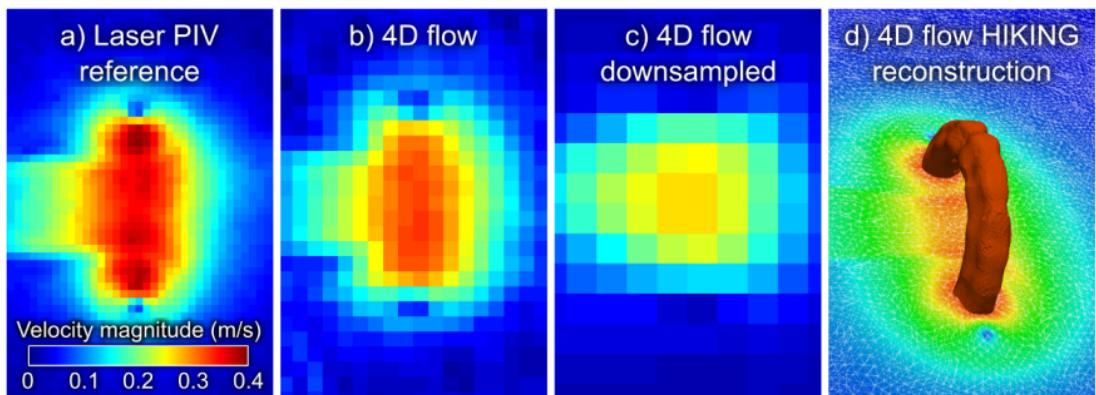
Phantom validation: laser PIV vs. 4D flow MRI vs. SBI

- Experimental setup: water tank with pulsatile inflow
- Precise laser PIV measurements: “true” flow
- Compare standard 4D flow MRI (high-res) and SBI with laser PIV “truth”
- SBI uses low-resolution MRI data

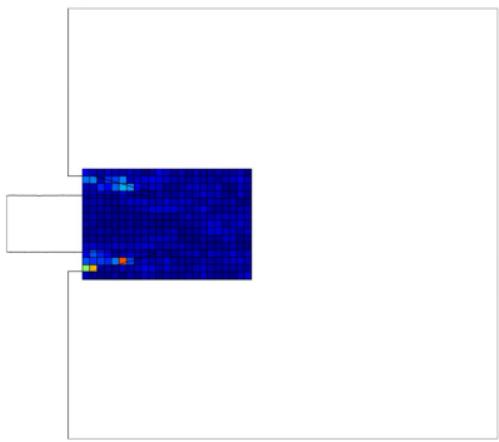


SBI captures features that cannot be resolved with 4D flow MRI

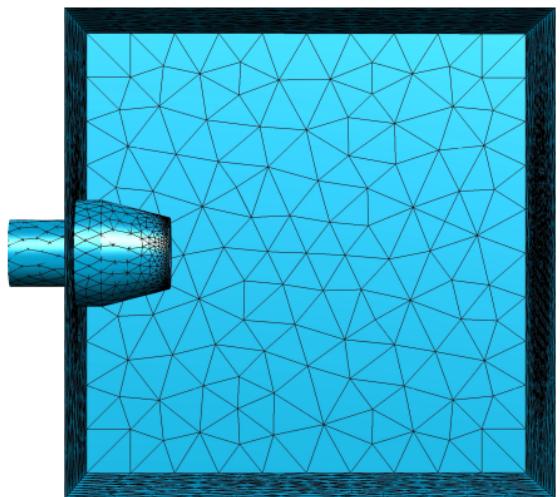
- Resolution of high-resolution 4D flow MRI: $3 \times 3 \times 3\text{mm}^3$, 50ms
- Resolution of low-resolution 4D flow MRI used for SBI: $6 \times 6 \times 6\text{mm}^3$, 100ms



High-quality reconstruction with experimental data: pulsatile flow

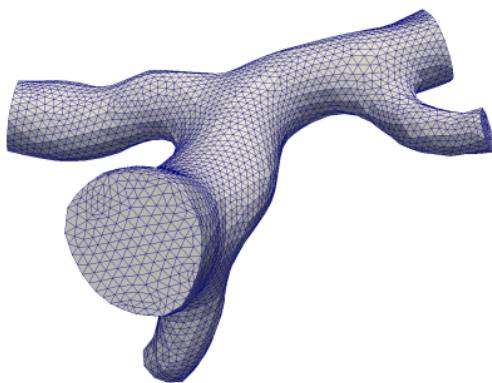


MRI data

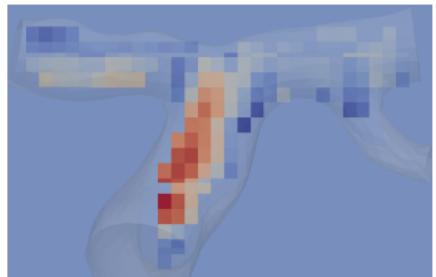


Reconstructed flow

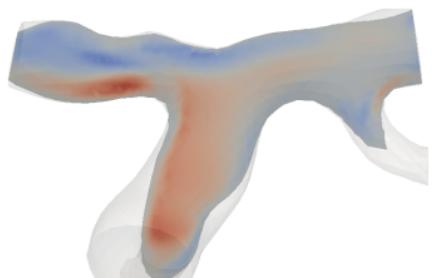
In vivo test of SBI flow reconstruction: Circle of Willis



Patient-specific mesh of brain vessel network
(Circle of Willis)

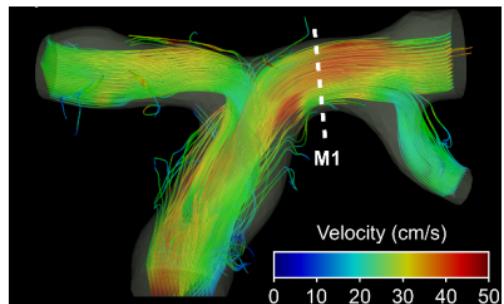


MRI voxel velocity data on
2D spatial slice at time
instance

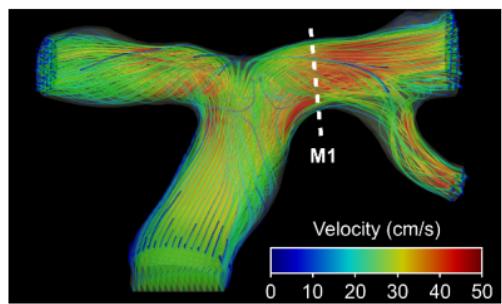


SBI reconstruction

Streamlines of flow: MRI 4D flow vs SBI reconstruction

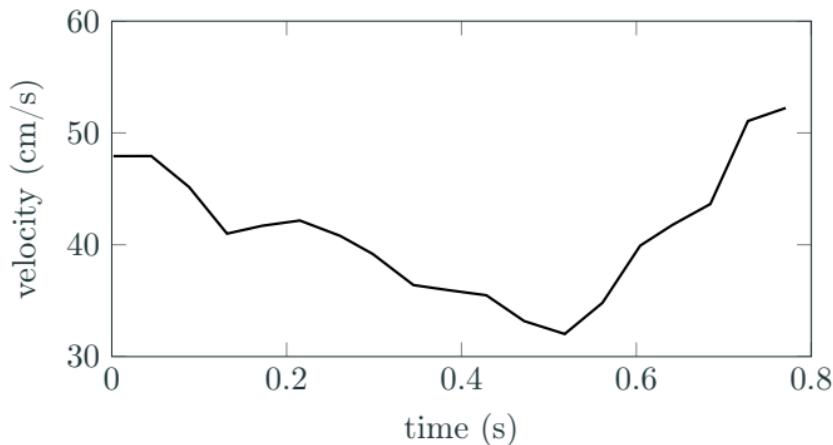


4D flow MRI reconstruction



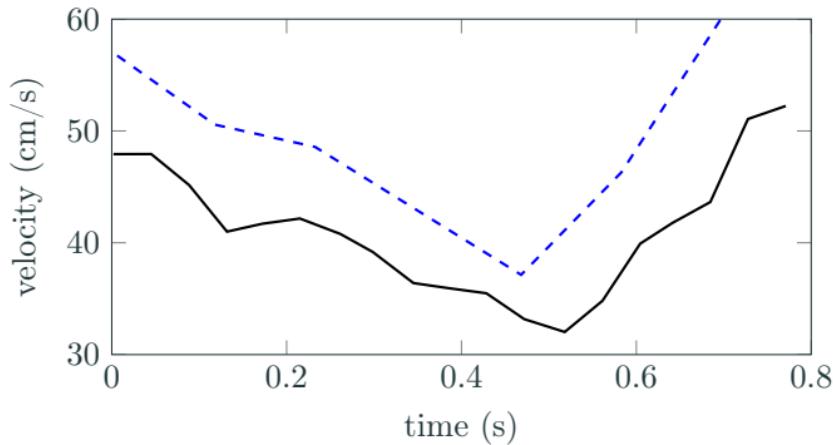
SBI reconstruction

SBI matches reference velocity measurements better than 4D flow MRI even for *in vivo* application



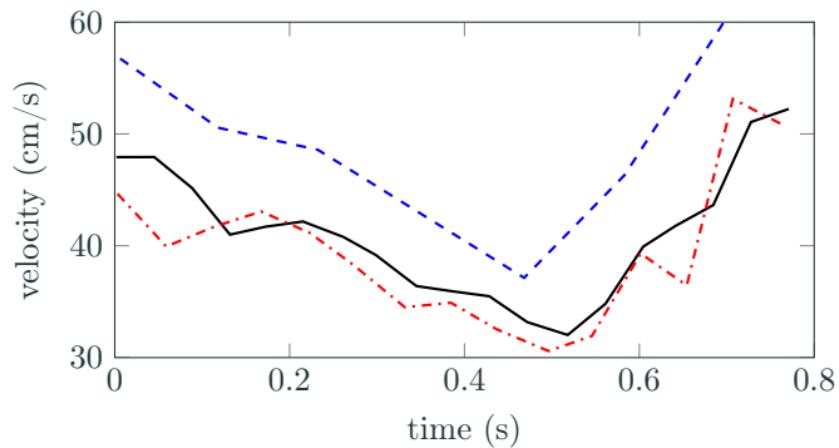
The reconstructed flow field (----) provides better agreement to accurate velocity measurements (—) on a 2D section than the 4D flow MRI measurements (---)

SBI matches reference velocity measurements better than 4D flow MRI even for *in vivo* application



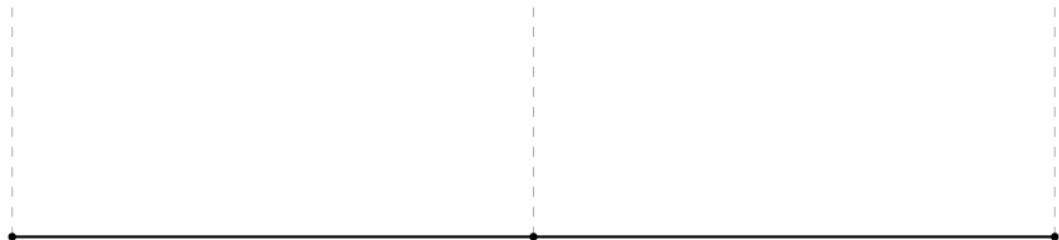
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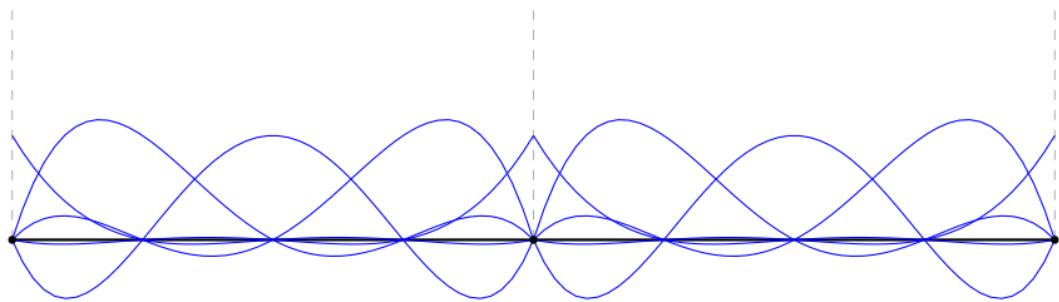
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State-of-the-art numerical methods for resolving shocks



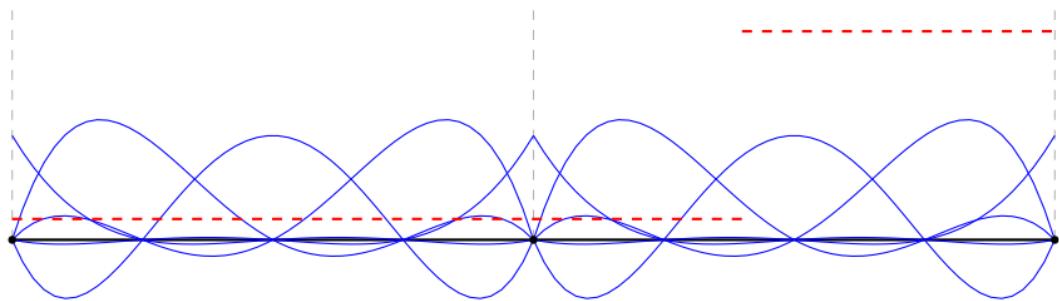
Fundamental issue: approximate discontinuity with polynomial basis

State-of-the-art numerical methods for resolving shocks



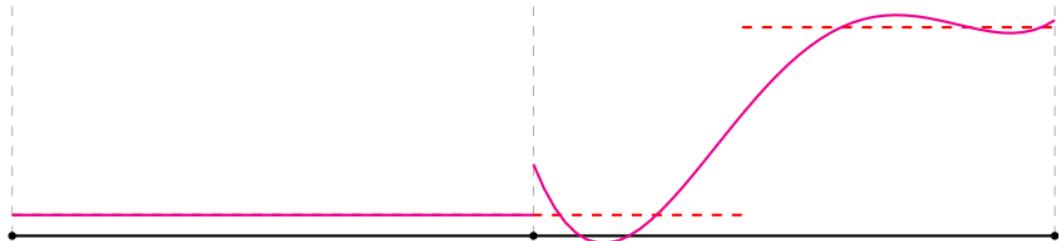
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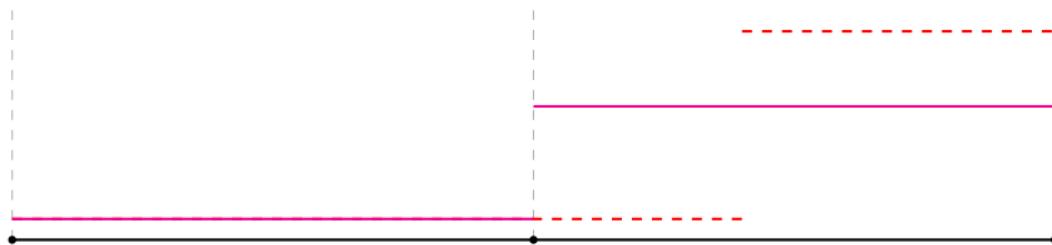
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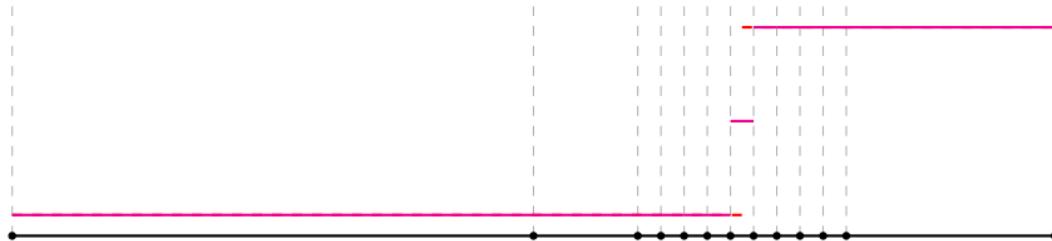


Fundamental issue: approximate discontinuity with polynomial basis

Existing solutions: **limiting**, artificial viscosity

Drawbacks: order reduction, local refinement

State-of-the-art numerical methods for resolving shocks

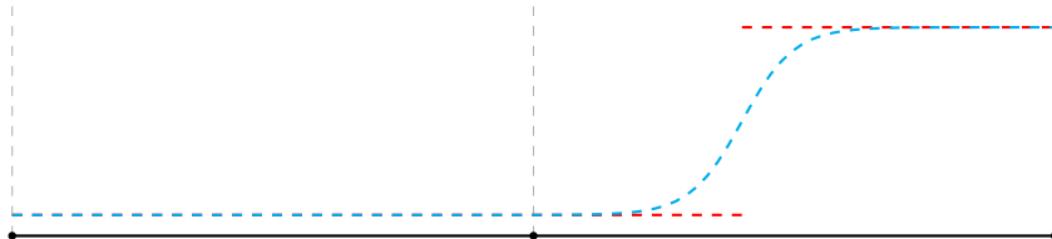


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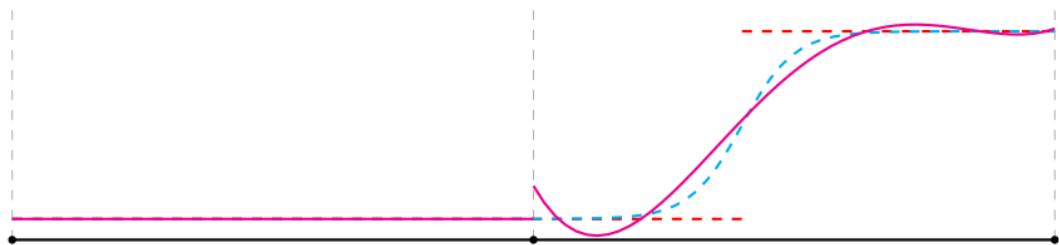


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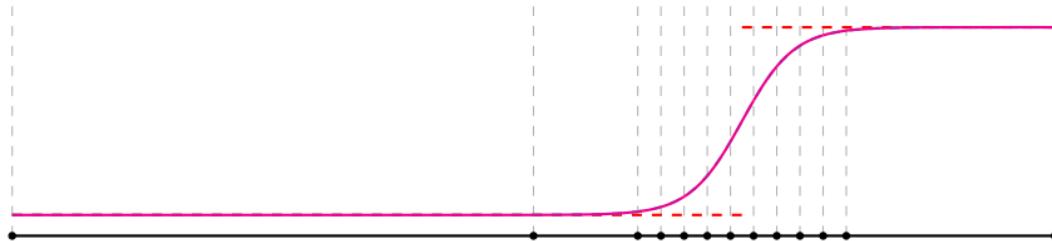


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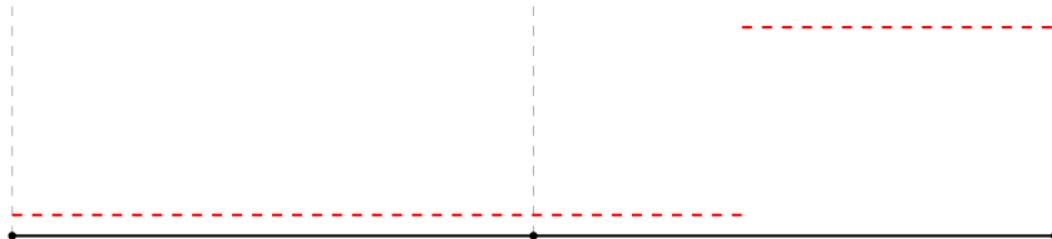


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Proposed solution: align features of solution basis with features in the solution using optimization formulation and solver

State-of-the-art numerical methods for resolving shocks



Fundamental issue: approximate discontinuity with polynomial basis

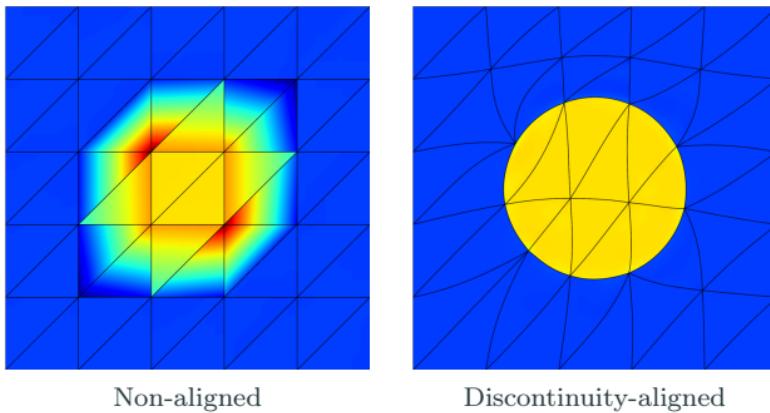
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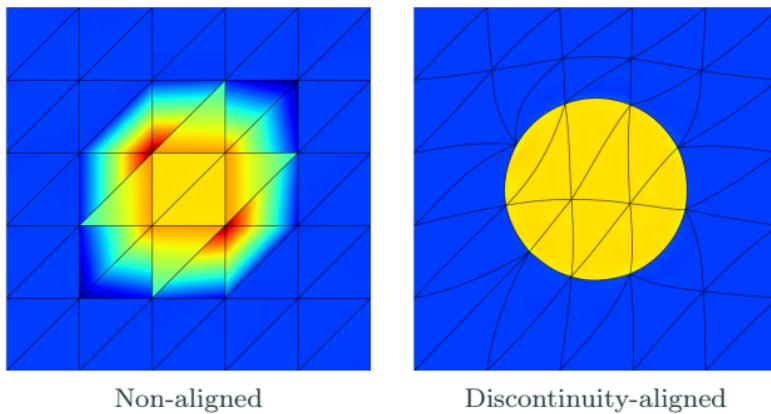
Tracking method for stable, high-order resolution of discontinuities

Goal: Align element faces with (unknown) discontinuities to perfectly capture them and approximate smooth regions to high-order



Tracking method for stable, high-order resolution of discontinuities

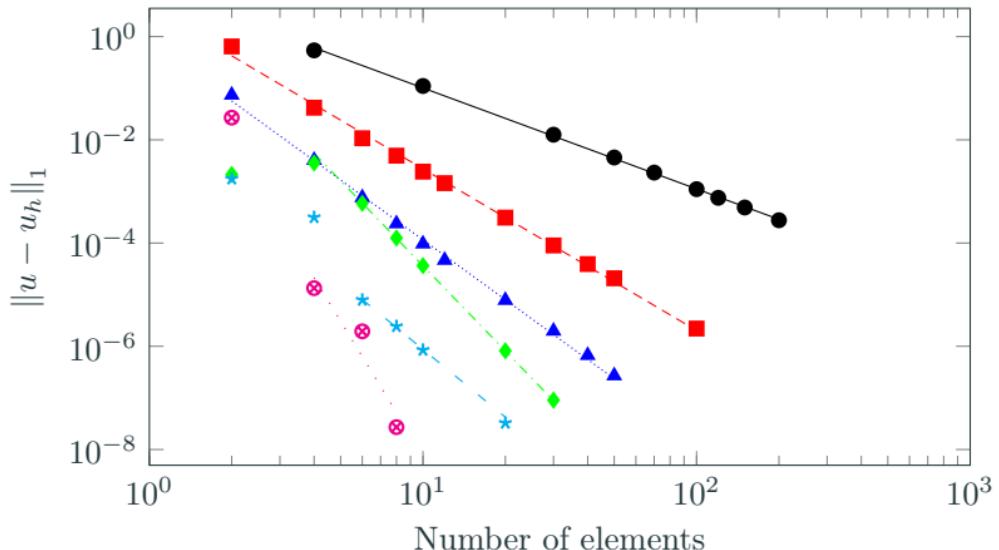
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Ingredients

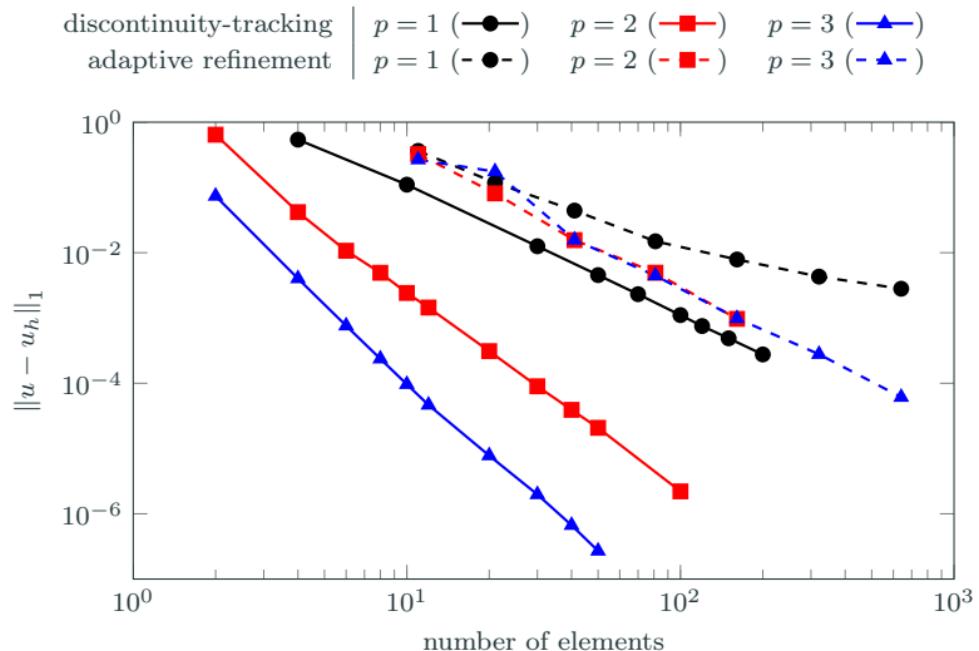
- Discontinuous Galerkin discretization: inter-element jumps, high-order
- Discontinuity-aligned mesh is the solution of an optimization problem constrained by the discrete PDE \implies **implicit shock tracking**
- Full space solver that converges the solution and mesh simultaneously to ensure solution of PDE never required on non-aligned mesh

Why tracking: Recover optimal $\mathcal{O}(h^{p+1})$ convergence rates



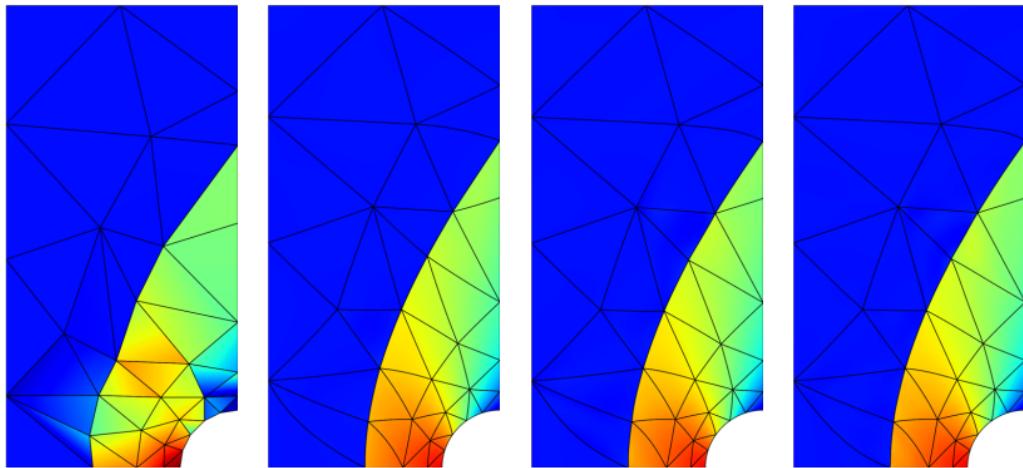
Convergence of DG discretization with implicit shock tracking for the modified inviscid Burgers' equation for polynomial orders $p = 1$ (\bullet), $p = 2$ (\blacksquare), $p = 3$ (\blacktriangle), $p = 4$ (\blacklozenge), $p = 5$ (\star), $p = 6$ (\otimes). The slopes of the best-fit lines to the data points in the asymptotic regime are: $\angle = -1.95$ (—), $\angle = -3.13$ (---), $\angle = -3.85$ (....), $\angle = -5.47$ (-·-), $\angle = -4.36$ (- - -), $\angle = -8.67$ (· · ·).

Why high-order tracking: Benefits more dramatic than low-order



Key observation: Accuracy improvement of tracking approach relative to (specialized) adaptive mesh refinement is more exaggerated for high-order approximations: $\mathcal{O}(10^1)$ for $p = 1$ and $\mathcal{O}(10^6)$ for $p = 3$.

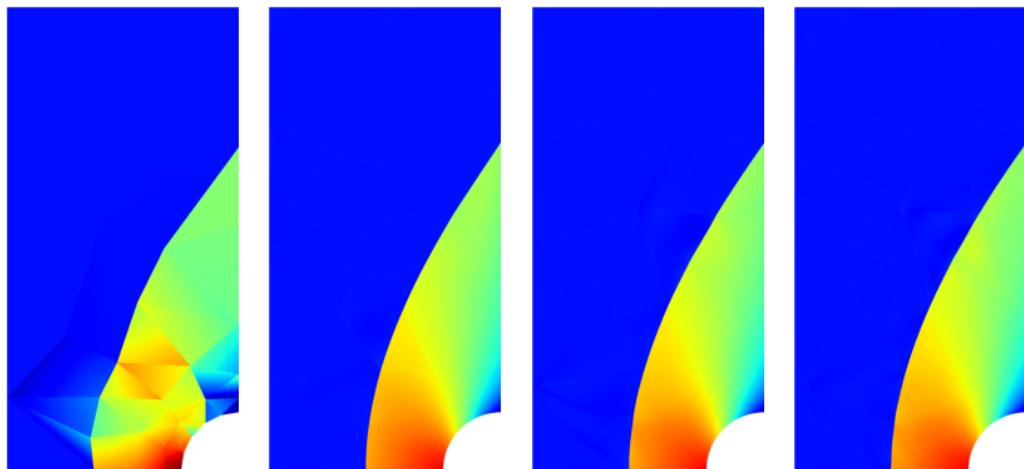
Why high-order tracking: Accurate solutions on coarse meshes



Density of supersonic flow ($M = 2$) past a cylinder using implicit shock tracking with $p = 1$ to $p = 4$ (left to right) DG discretization.

Key observation: High-order tracking enables accurate resolution of 2D supersonic flow with 48 elements; the error in the stagnation enthalpy is $\mathcal{O}(10^{-4})$ for $p = 2$ (1152 DoF).

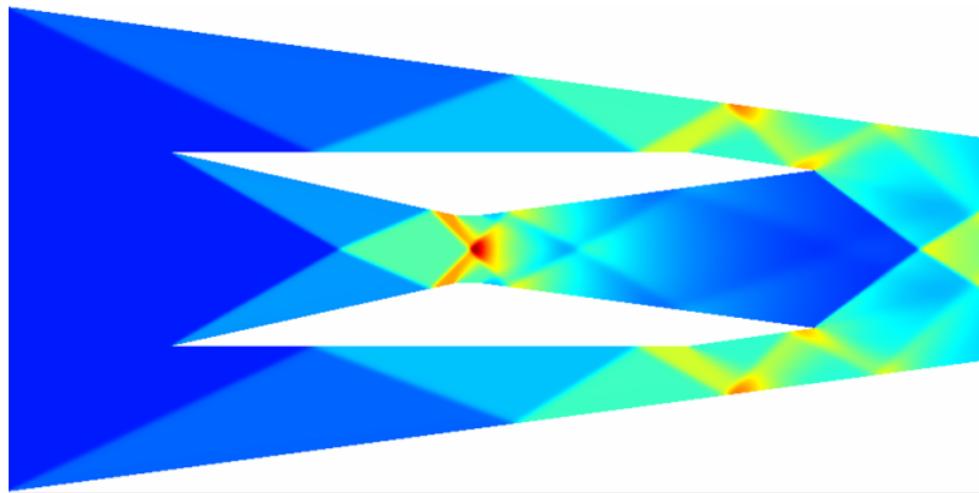
Why high-order tracking: Accurate solutions on coarse meshes



Density of supersonic flow ($M = 2$) past a cylinder using implicit shock tracking with $p = 1$ to $p = 4$ (left to right) DG discretization.

Key observation: High-order tracking enables accurate resolution of 2D supersonic flow with 48 elements; the error in the stagnation enthalpy is $\mathcal{O}(10^{-4})$ for $p = 2$ (1152 DoF).

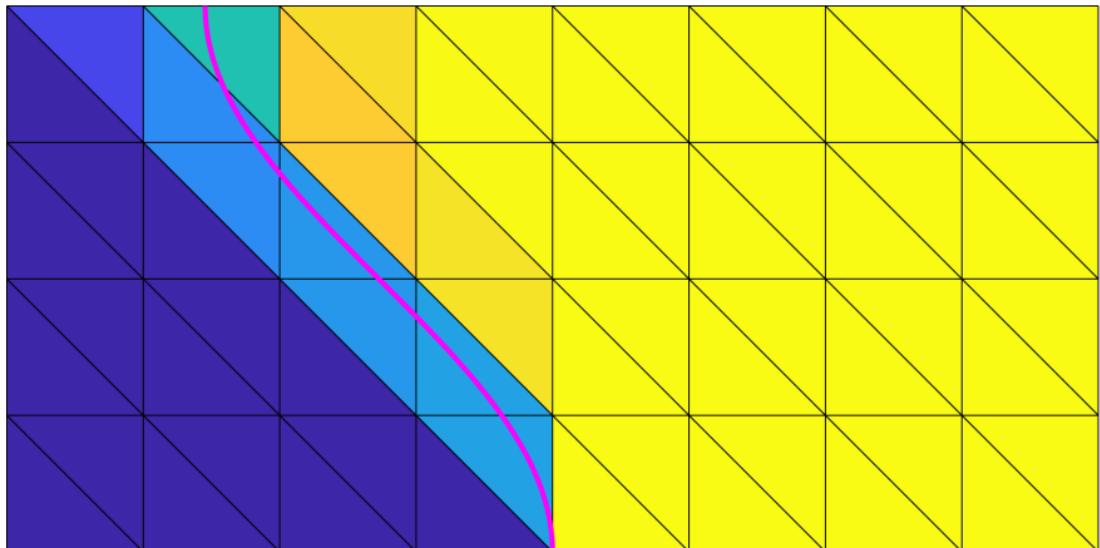
Why not tracking: Difficult for complex discontinuity surfaces



Implicit shock tracking

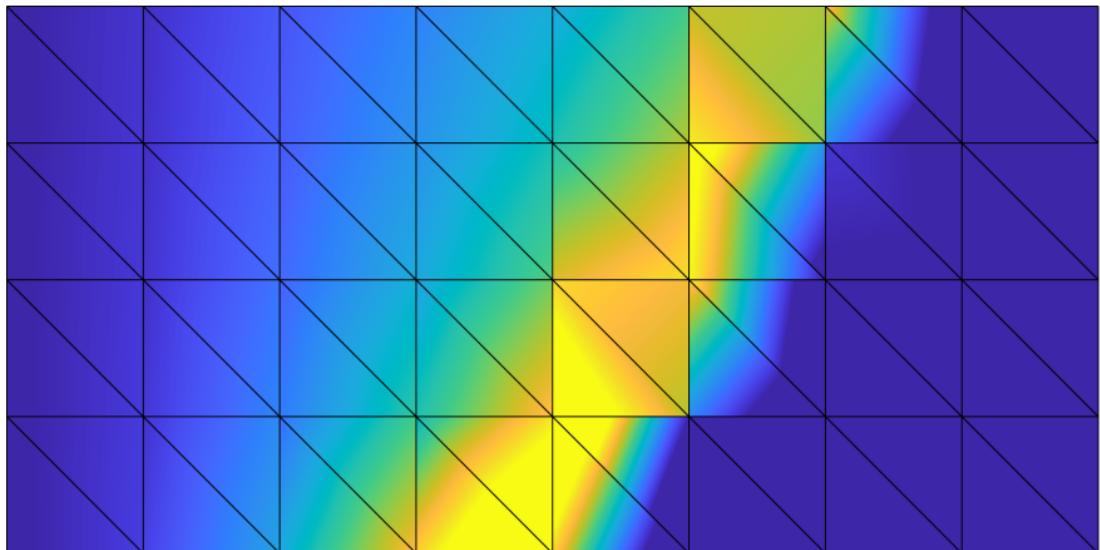
Aims to overcome the difficulty of explicitly meshing the unknown shock surface

Linear advection, trigonometric shock



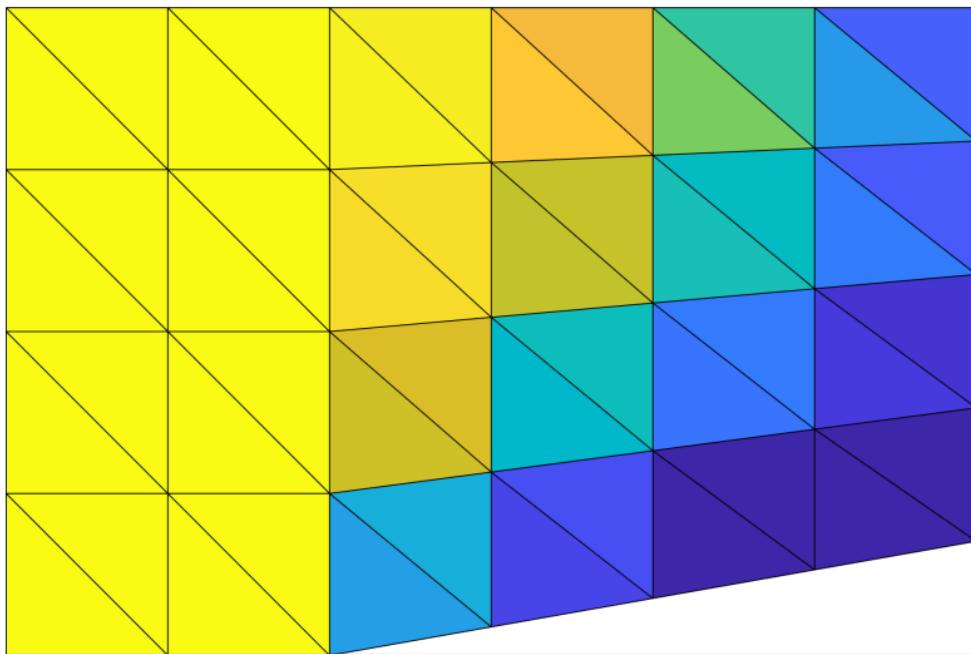
$p = 0$ space for solution, $q = 2$ space for mesh
 L_1 error = 1.15×10^{-3}

Inviscid Burgers' equation: space-time formulation



$p = 0$ space for solution, $q = 3$ space for mesh

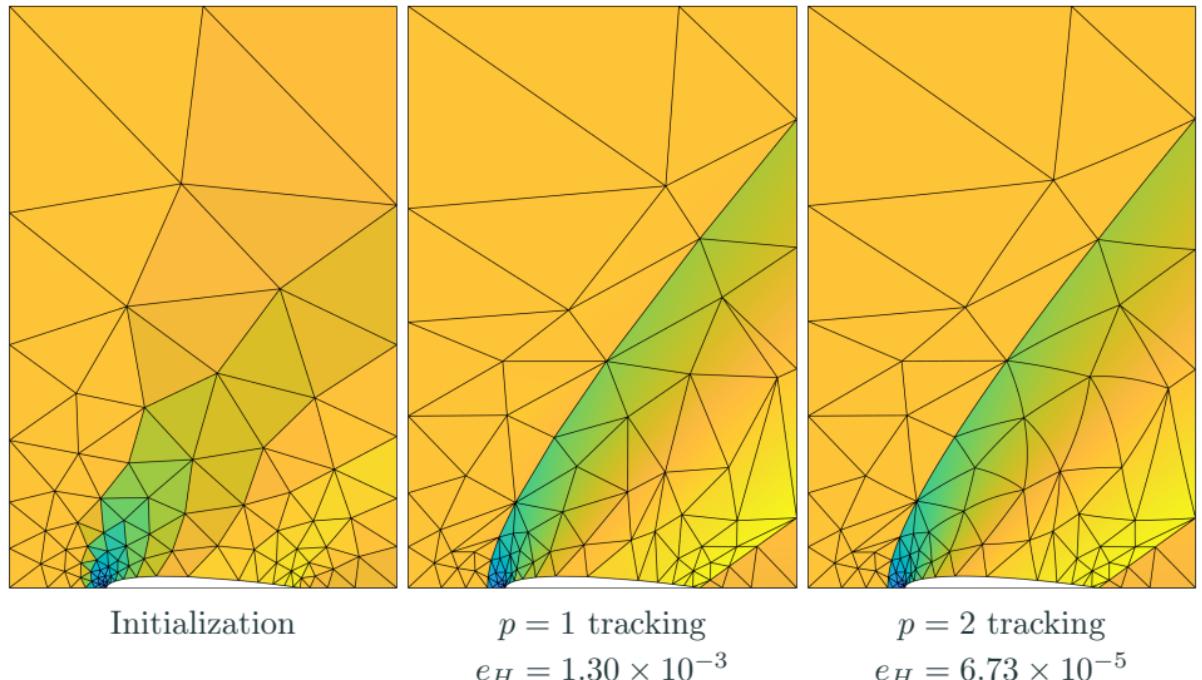
Supersonic flow past wedge ($M = 2$)



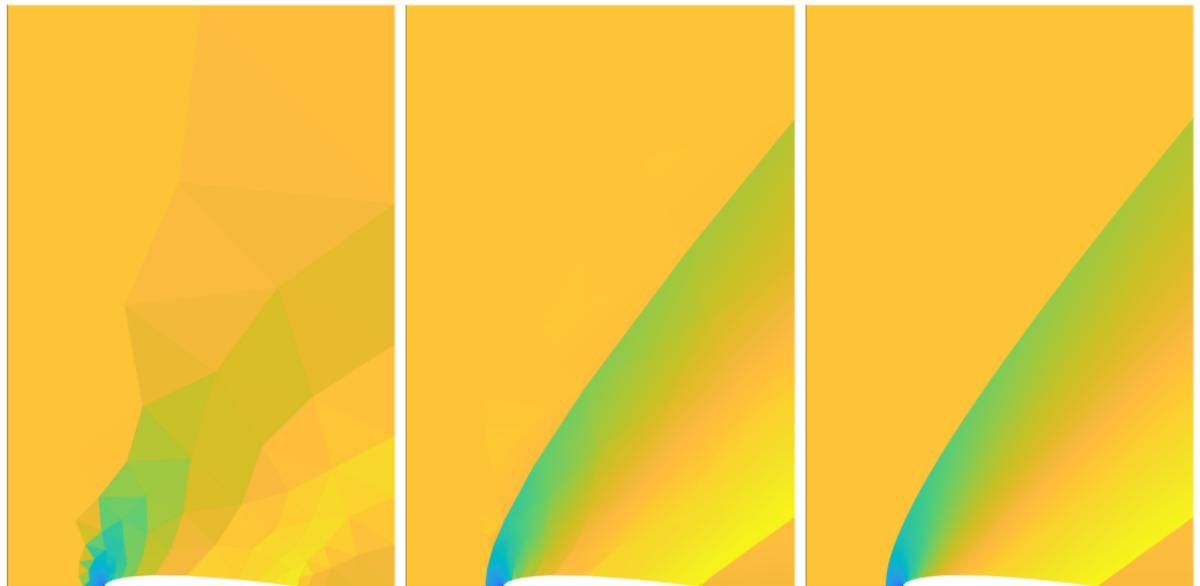
$p = 0$ space for solution, $q = 1$ space for mesh

L^2 stagnation enthalpy error: 7.94×10^{-10}

Supersonic flow past NACA0012 airfoil ($M = 1.5$)



Supersonic flow past NACA0012 airfoil ($M = 1.5$)

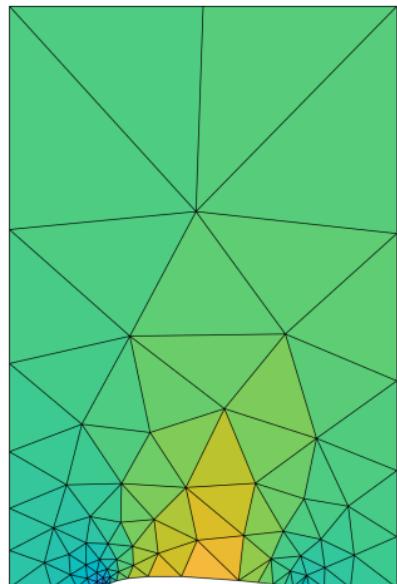


Initialization

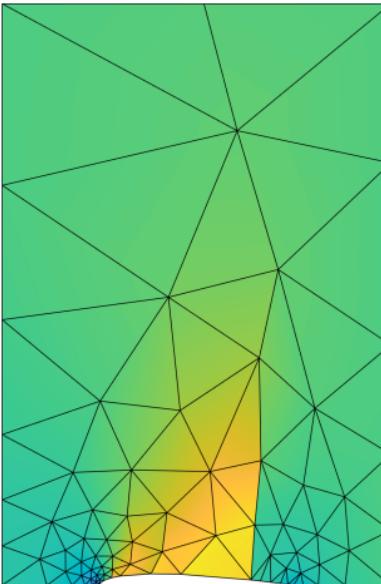
$p = 1$ tracking
 $e_H = 1.30 \times 10^{-3}$

$p = 2$ tracking
 $e_H = 6.73 \times 10^{-5}$

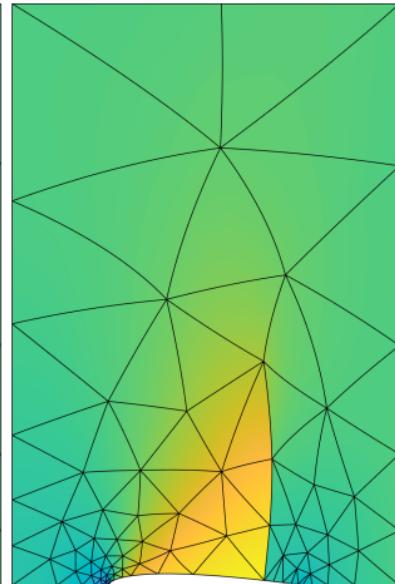
Transonic flow past NACA0012 airfoil ($M = 0.85$)



Initialization

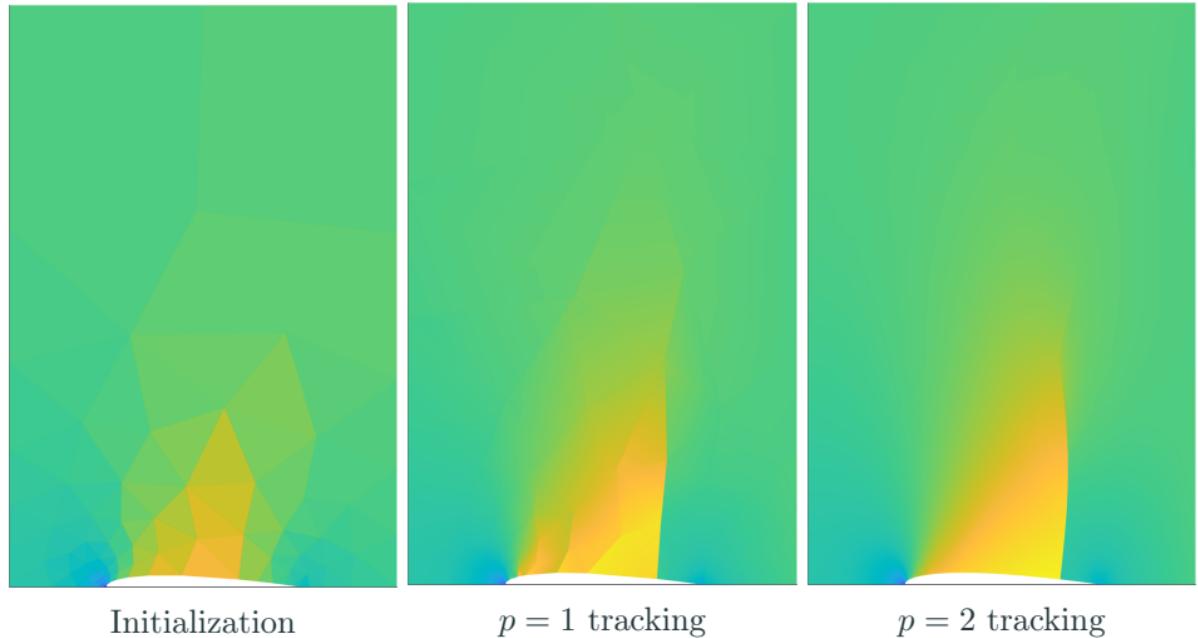


$p = 1$ tracking



$p = 2$ tracking

Transonic flow past NACA0012 airfoil ($M = 0.85$)



Initialization

$p = 1$ tracking

$p = 2$ tracking