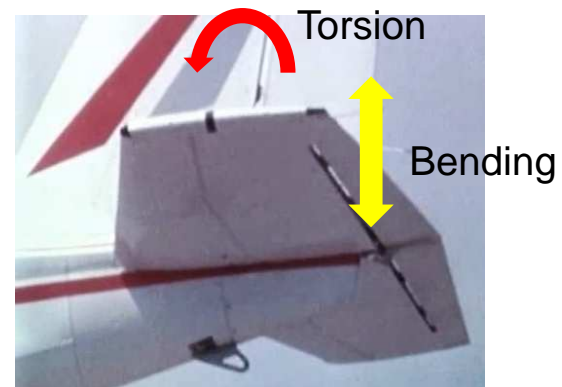
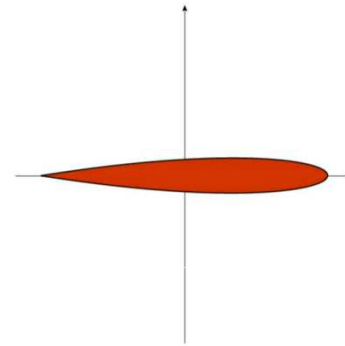


COUPLED-MODE FLUTTER OF A WING

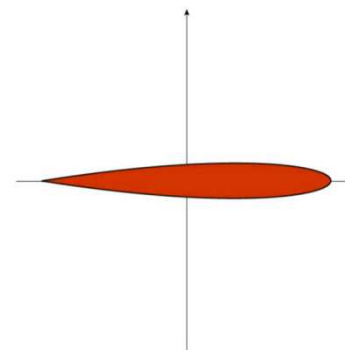


COUPLED-MODE FLUTTER OF A WING

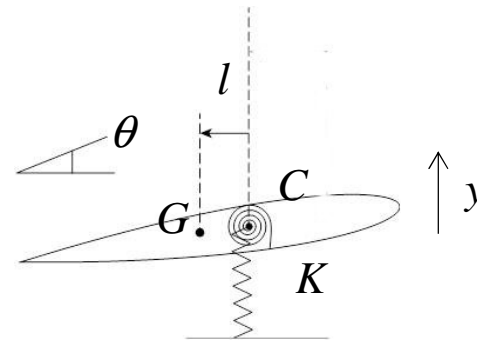
Mode 1 : plunge



Mode 2 : torsion



TWO-MODES APPROXIMATION



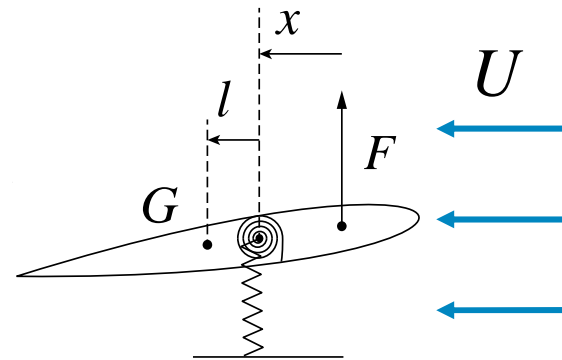
Kinetic energy $K_c = \frac{1}{2}M(\dot{y} - l\dot{\theta})^2 + \frac{1}{2}J\dot{\theta}^2$

Elastic energy $W = \frac{1}{2}Ky^2 + \frac{1}{2}C\theta^2$

Using Lagrange equation

$$\begin{aligned} M\ddot{y} - Ml\ddot{\theta} + Ky &= 0 \\ J\ddot{\theta} + C\theta &= -Kyl \end{aligned}$$

FLUID FORCES



Quasi-static aeroelasticity approximation

$$F = \frac{1}{2} \rho U^2 L C_L(\theta)$$

$$M\ddot{y} - Ml\ddot{\theta} + Ky = \frac{1}{2} \rho U^2 L C_L(\theta)$$

$$J\ddot{\theta} + C\theta = -Kyl + \frac{1}{2} \rho U^2 L(x+l)C_L(\theta)$$

TWO MODES WITH THE FLUID FORCE

Expansion of the force at the first order

$$F = \frac{1}{2} \rho U^2 L C_L(\theta) \approx \frac{1}{2} \rho U^2 L \left(\frac{\partial C_L}{\partial \theta} \right)_0 \theta$$

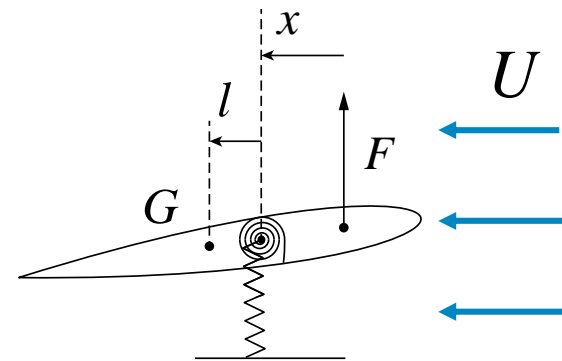
$$\frac{\partial C_L}{\partial \theta} = 2\pi$$

Dimensionless quantities

$$q_1 = \frac{y}{DL}; q_2 = \frac{\theta}{D}$$

$$\Omega = \sqrt{\frac{KJ}{CM}}; \kappa = \frac{KL^2}{C}; \varepsilon = \frac{l}{L}; \chi = \frac{x}{L}$$

$$C_Y = \frac{\rho U^2 L^2}{2C} \quad \bar{t} = \sqrt{\frac{C}{J}} t$$



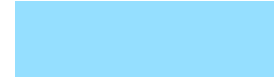
Coupled equations in flow

$$\ddot{q}_1 - \varepsilon \ddot{q}_2 + \Omega^2 q_1 = C_Y 2\pi \frac{\Omega^2}{\kappa} q_2$$

$$\ddot{q}_2 + (1 - C_Y 2\pi(\varepsilon + \chi)) q_2 = -\kappa \varepsilon q_1$$

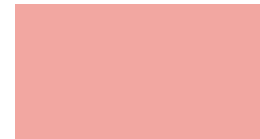
COUPLED EQUATIONS IN FLOW

$$\begin{aligned} \ddot{q}_1 - \varepsilon \ddot{q}_2 + \Omega^2 q_1 &= C_Y 2\pi \frac{\Omega^2}{\kappa} q_2 \\ \ddot{q}_2 + (1 - C_Y 2\pi(\varepsilon + \chi)) q_2 &= -\kappa \varepsilon q_1 \end{aligned}$$



Coincidence

+



Non-symmetric
stiffness coupling

= Dynamic instability !

COUPLED EQUATIONS IN FLOW

$$\begin{aligned} \ddot{q}_1 - \varepsilon \ddot{q}_2 + \Omega^2 q_1 &= C_Y 2\pi \frac{\Omega^2}{\kappa} q_2 \\ \ddot{q}_2 + (1 - C_Y 2\pi(\varepsilon + \chi)) q_2 &= -\kappa \varepsilon q_1 \end{aligned}$$

Modes for a given set of parameters

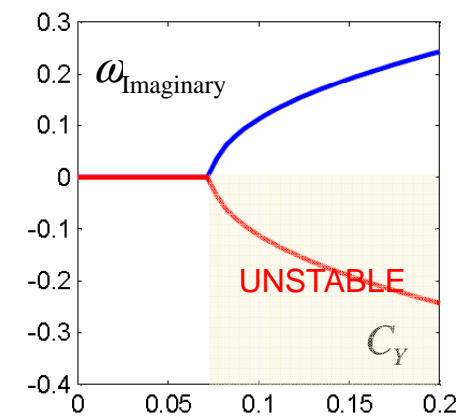
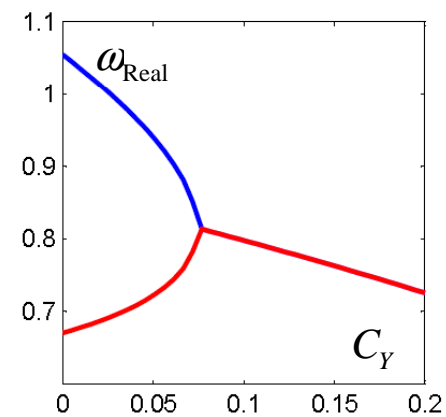
$$\begin{pmatrix} q_1 \\ q_2 \end{pmatrix} = \text{Re} \left[\begin{pmatrix} q_1 \\ q_2 \end{pmatrix}_0 e^{i\omega t} \right] \quad \rightarrow \quad \begin{matrix} \omega_1 & \begin{pmatrix} q_1 \\ q_2 \end{pmatrix}_1 \\ \omega_2 & \begin{pmatrix} q_1 \\ q_2 \end{pmatrix}_2 \end{matrix}$$

$$\omega(C_Y) = \omega_{\text{real}}(C_Y) + i\omega_{\text{imag}}(C_Y)$$

EFFECT OF THE CAUCHY NUMBER ON FREQUENCIES

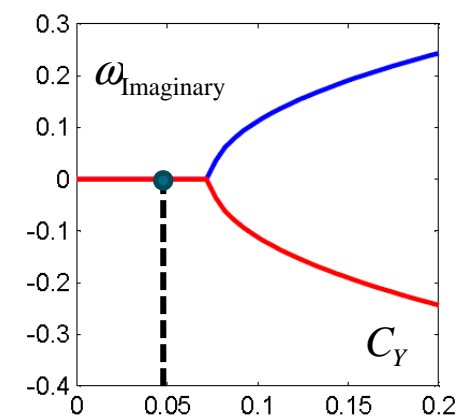
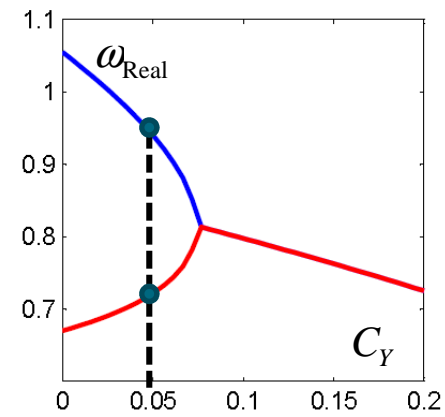


$$\Omega^2 = 1/2; \kappa = 1; \varepsilon = 1/4; \chi = 1/4$$

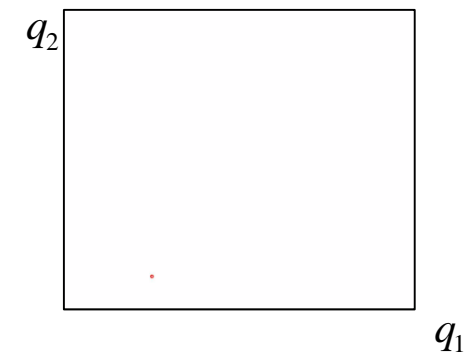
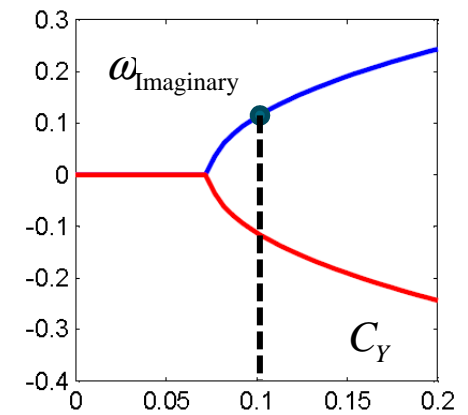
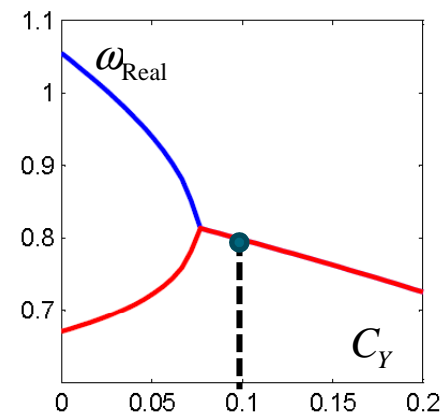


Coincidence

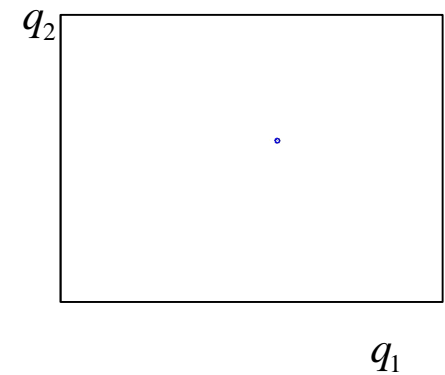
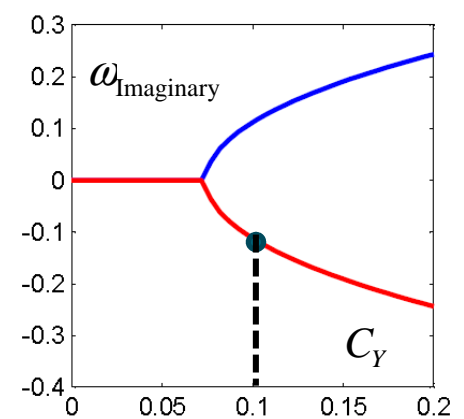
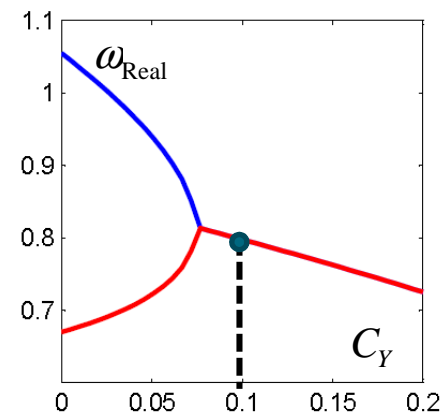
BEFORE COINCIDENCE



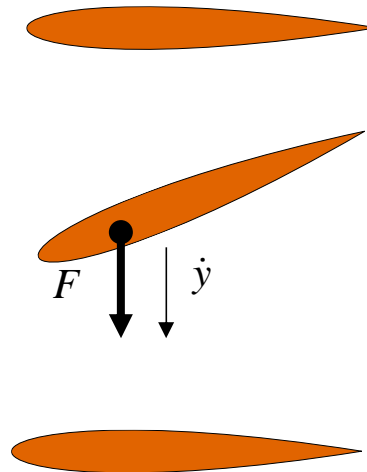
AFTER COINCIDENCE : DAMPED MODE



AFTER COINCIDENCE : UNSTABLE MODE



UNSTABLE MODE : BALANCE OF ENERGY



$$W_{FS} = \int_T F \dot{y} dt > 0$$

COUPLED MODE FLUTTER OF A WING



COUPLED MODE FLUTTER OF A WING

