

..... and Acceleration



Dynamics: Angular Velocity

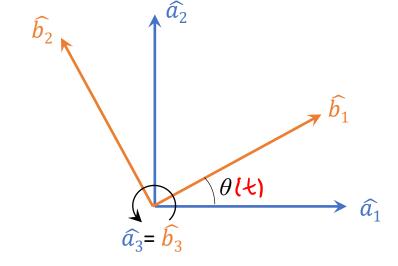
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Angular Velocity

- Angular velocity is a vector!
 - ... even though the general 3D orientation cannot be represented as a vector.
- Angular velocity is derived from small changes in the angle
 ... small (infinitesimal) angles are additive and commutative and can be
 represented as a vector
- For small-angle, linearized, dynamics, we may not need representation of general 3D orientation
 - e.g., linearized attitude dynamic of aircraft including linearized stability analysis

Simple Rotation and Simple Angular Velocity

- Let's consider two reference frames A and B
- If there is a rotation of frame B relative to frame A, about a fixed axis in both frames (simple rotation), say the z-axis, of θ



• This is simple angular velocity of B in A for a rotation about a fixed axis in A and B

B in A for a and B
$$\begin{cases}
\hat{b}_1 = (a \cdot 0 \hat{a}_1 + s n \cdot 0 \hat{a}_2) \\
\hat{b}_2 = (a \cdot 0 \hat{a}_2 + s n \cdot 0 \hat{a}_2)
\end{cases}$$

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$$\begin{cases}$$

Mathematical Definition of Angular Velocity

- The angular velocity is written in terms of the rate of change of unit vectors of B in A
- All the angular velocity expression can be derived from the above equation
- The expression can also be used to derive other very useful relations

Addition Theorem for Angular Velocities

• Lets consider two reference frames A and B and auxiliary or intermediate frames $A_1, A_2, ..., A_{n-1}, A_n$

$${}^{A}\vec{\omega}^{B} = {}^{A}\vec{\omega}^{A_{1}} + {}^{A_{1}}\vec{\omega}^{A_{2}} + \cdots {}^{A_{n-1}}\vec{\omega}^{A_{n}} + {}^{A_{n}}\vec{\omega}^{B}$$

• Thus:

$$^{B}\vec{\omega}^{A}=-^{A}\vec{\omega}^{B}$$

- ullet We will use this in calculation of the aircraft angular velocity in terms of intermediate frames C and D
- For later, angular acceleration does not satisfy any addition theorem!

Angular Velocity in terms of Body Axis Measure Numbers

• It is common to write angular velocity as

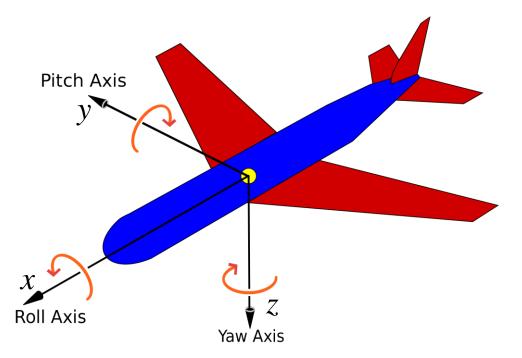
$$\vec{\omega} = p\,\hat{b}_1 + q\,\hat{b}_2 + r\,\hat{b}_3$$

• For small angles:
$$p = \overline{\delta \phi} \qquad \overline{\delta \phi} = \int \rho dt$$

$$q = \overline{\delta \theta}$$

$$r = \overline{\delta \psi}$$

 Above angular velocities cannot be integrated to get corresponding angles for finite rotations



Single rotation + Ang vel Addition Theorem - Life in grand Angular Velocity in terms of Body Axis Measure Numbers — in terms of finite angles

- Consider angles (ϕ, θ, ψ) representing (Body 3-2-1) orientation from Earth frame A to aircraft body frame B:
 - Rotate about Earth z-axis by ψ to get intermediate reference frame C
 - Rotate about ref frame C y-axis by θ to get intermediate ref frame D
 - Rotate about ref frame D x-axis by ϕ to get aircraft body frame B

• There are singularities but not if we restrict the angles

$$\vec{\omega} = \vec{\phi} |\hat{d}_1| + \vec{\theta} |\hat{c}_2| + \vec{\psi} |\hat{a}_3| \qquad p = -\dot{\psi} \sin \theta + \dot{\phi}$$

$$= \vec{\phi} |\hat{b}_1| + \dot{\theta} |\hat{d}_2| + \dot{\psi} |\hat{c}_3| \qquad q = \dot{\psi} \cos \theta \sin \phi + \dot{\theta} \cos \phi$$

$$\vec{\omega} = p \hat{b}_1 + q \hat{b}_2 + r \hat{b}_3 \qquad r = \dot{\psi} \cos \theta \cos \phi - \dot{\theta} \sin \phi$$

$$\vec{\omega} = p \hat{b}_1 + q \hat{b}_2 + r \hat{b}_3 \qquad \vec{\phi} = 0 \quad \vec{\phi} \quad \vec$$

Components of Vectors

 Use transformation matrix to represent a given vector in different reference frames

Differentiation in Two Reference Frames

• For any vector 🕏

$$rac{Adec{v}}{dt} = rac{Bdec{v}}{dt} + {}^Aec{\omega}^B imes ec{v}$$

- This is an important relation used frequently
- Angular velocity is key to the differentiation of vectors in rotating frames ... in addition to giving information about the angular motion of the body
- If vector is fixed in B

$$\frac{^{A}d\vec{v}}{dt} = {^{A}}\vec{\omega}^{B} \times \vec{v}$$

Angular Acceleration

• Lets consider two reference frames A and B, the angular acceleration of B in A is given by:

$$\mathbf{A}_{\vec{\alpha}}\mathbf{B} = \frac{\mathbf{A}_{d}\mathbf{A}_{\vec{\omega}}\mathbf{B}}{dt}$$

ullet We can also differentiate the angular velocity in the B frame to get the angular acceleration

 $AZ^{B} = p \hat{b}_{1} + q \hat{b}_{2} + r \hat{b}_{3}$ $AZ^{B} = p \hat{b}_{1} + q \hat{b}_{2} + r \hat{b}_{3}$

$$\mathbf{A} \stackrel{>}{\sim} \mathbf{B} = \frac{^{A}d^{A}\vec{\omega}^{B}}{dt} = \frac{^{B}d^{A}\vec{\omega}^{B}}{dt} + \frac{^{A}\vec{\omega}^{B} \times ^{A}\vec{\omega}^{B}}{}$$

• For simple rotation and angular velocity, we have

$$\vec{\omega} = \dot{\theta}\hat{k}$$
 $\vec{\alpha} = \ddot{\theta}\hat{k}$

Angular Acceleration

• You can differentiate the angular velocity in A or B but not C

• Angular Acceleration is not Additive! C is an intermediate axis

$$A\overrightarrow{\omega}^{B} = A\overrightarrow{\omega}^{C} + C\overrightarrow{\omega}^{B}$$

$$A\overrightarrow{\delta}^{B} = AAA\overrightarrow{\omega}^{C} + AAC\overrightarrow{\omega}^{B}$$

$$AAAC + AAC\overrightarrow{\omega}^{B}$$

$$AAC + AAC + AAC$$