Orientation Angles



Dynamics: Orientation Angles

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Three Angles

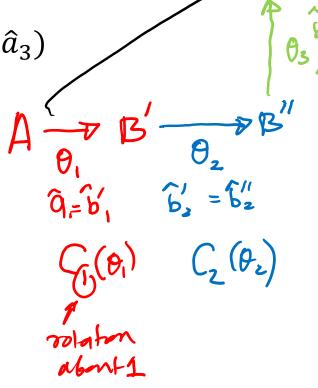
- Orientation of body B in reference frame A can be represented in terms of three angles ...
- Consider unit vectors: $(\hat{a}_1, \hat{a}_2, \hat{a}_3)$ and $(\hat{b}_1, \hat{b}_2, \hat{b}_3)$
- Starting with $(\hat{a}_1, \hat{a}_2, \hat{a}_3)$ and $(\hat{b}_1, \hat{b}_2, \hat{b}_3)$ aligned, one can rotate about three angles about three particular axes to get from $(\hat{a}_1, \hat{a}_2, \hat{a}_3)$ to $(\hat{b}_1, \hat{b}_2, \hat{b}_3)$
- These are three orientation angles
- See Appendix I and II in the book for all the possible angle combinations

Example: Body 1-2-3

• Consider the body $(\hat{b}_1,\hat{b}_2,\hat{b}_3)$ is initially aligned with $(\hat{a}_1,\hat{a}_2,\hat{a}_3)$

• Consider a rotation of θ_1 about \hat{b}_1' = \hat{a}_1

• This takes us to a new reference frame, say $(\hat{b}_1', \hat{b}_2', \hat{b}_3')$



- Next, consider a rotation of θ_2 about $\hat{b}_2' = \hat{b}_2''$
- This takes us to a new reference frame, say $(\hat{b}_1'', \hat{b}_2'', \hat{b}_3'')$

- Finally, consider a rotation of θ_3 about $\hat{b}_3^{\prime\prime}$ = \hat{b}_3
- This takes us to the body reference frame system $(\hat{b}_1, \hat{b}_2, \hat{b}_3)$

$$\begin{pmatrix}
\hat{b}_1'' \\
\hat{b}_2'' \\
\hat{b}_3''
\end{pmatrix} = \begin{bmatrix}
C_3(\theta_3) \\
\hat{b}_2 \\
\hat{b}_3
\end{pmatrix}$$

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• The direction cosine matrix can be written as:
$$\begin{bmatrix} AC^B \end{bmatrix} = \begin{bmatrix} C_1(\theta_1) \end{bmatrix} \begin{bmatrix} C_2(\theta_2) \end{bmatrix} \begin{bmatrix} C_3(\theta_3) \end{bmatrix} = \begin{bmatrix} c_2c_3 & -c_2s_3 & s_2 \\ s_1s_2c_3 + s_3c_1 & -s_1s_2s_3 + c_3c_1 & -s_1c_2 \\ -c_1s_2c_3 + s_3s_1 & c_1s_2s_3 + c_3s_1 & c_1c_2 \end{bmatrix}$$

$$\left\{ \begin{array}{l} \hat{a}_1 \\ \hat{a}_2 \\ \hat{a}_3 \end{array} \right\} = \begin{bmatrix} {}^A C^B \end{bmatrix} \left\{ \begin{array}{l} \hat{b}_1 \\ \hat{b}_2 \\ \hat{b}_3 \end{array} \right\}$$

- One can go from the angles to direction cosine matrix and one can go from the direction cosine matrix back to angles* (see section 10.3)
 - * except at singularities

Angular Velocity

• Use the addition theorem and the simple angular velocities to calculate the angular velocity vectors

$$A\vec{\omega}^B = \dot{\theta}_1 \hat{b}_1' + \dot{\theta}_2 \hat{b}_2'' + \dot{\theta}_3 \hat{b}_3$$
 in mixed basis $= \omega_1 \hat{b}_1 + \omega_2 \hat{b}_2 + \omega_3 \hat{b}_3$ in body frame

• The measure numbers of the angular velocity vector can be represented in any reference frame, specifically we can find the measure numbers in the B reference

frame

... these are the kinematical differential equations

... the above matrix is not always invertible – singularity!

Combal lock.

Other Orientation Angle Representations

- Other axis sequence, e.g., Body 3-2-1 used in aircraft flight mechanics
 - The direction cosine matrix and kinematical differential equations for Body 3-2-1 were presented in Lecture 3

$$\vec{\omega} = p \, \hat{b}_1 + q \, \hat{b}_2 + r \, \hat{b}_3$$

$$\begin{cases} p \\ q \\ r \end{cases} = \begin{vmatrix} 1 & 0 & -s_{\theta} \\ 0 & c_{\phi} & c_{\theta} s_{\phi} \\ 0 & -s_{\phi} & c_{\theta} c_{\phi} \end{vmatrix} \begin{cases} \phi \\ \dot{\theta} \\ \dot{\psi} \end{cases}$$

- Angles measured about the space fixed axes, e.g, Space 1-2-3
- 2-axis (not 2-angle) instead of 3-axis angles, e.g., Space 1-2-1
 See Appendix I and II for the direction cosine matrices and kinematical differential equations for 24 different orientation angle sequences