Newton-Euler Method for Rigid Body Dynamics

... about a point that is not the center of mass



Dynamics: Newton-Euler Equations of Motion (non-CM)

© 2021 Mayuresh Patil. Licensed under a Creative Commons Attribution 4.0 license https://creativecommons.org/licenses/by-nc-sa/4.0/ mpatil@gatech.edu

Notations

- There are only two frames, inertial frame *I* and body frame *B*
- All time derivatives are w.r.t. the inertial frame unless indicated
- All angular velocities and angular accelerations are of the body relative to the inertial frame
- Q is a general point on the rigid body (for integration)
- C is the center of mass of the rigid body
 - ullet We already know the equations of motion using the velocities at C and moments about C
- P is a fixed point on the rigid body that is not the center of mass
 - We are trying to write the equations in terms of the velocities at P and moments about P
- O is a point fixed in the inertial frame

Euler's First Law

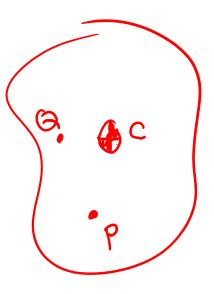
$$\sum \vec{F} = \frac{d\vec{L}}{dt}$$

- The above equation does not reference any point.
- Now consider a rigid body:

$$\vec{L} = \int \vec{v}^Q dm = \sum m; \vec{v}$$

• We can write the velocity at Q relative to a reference point, C or P (using the two-point formula for velocity):

$$ec{v}^Q = ec{v}^C + ec{\omega} imes ec{r}^{CQ}$$
 $ec{v}^Q = ec{v}^P + ec{\omega} imes ec{r}^{PQ}$



• Linear Momentum:

$$\vec{L} = m\vec{v}^C$$

$$\vec{L} = m\vec{v}^P + m\vec{\omega} \times \vec{r}^{PC}$$
$$= m\vec{v}^P - m\vec{r}^{PC} \times \vec{\omega}$$

• Equations of Motion (assuming constant m and constant \vec{r}^{PC} [in B frame]):

$$\sum \vec{F} = \frac{d \left(m\vec{v}^{\,C} \right)}{dt}$$

$$= m\dot{\vec{v}}^{\,C}$$

$$= m\dot{\vec{v}}^{\,P} + m\dot{\vec{\omega}} \times \vec{r}^{\,PC} + m\vec{\omega} \times \dot{\vec{r}}^{\,PC}$$

$$= m\dot{\vec{v}}^{\,P} - m\vec{r}^{\,PC} \times \dot{\vec{\omega}} + m\vec{\omega} \times (\omega \times \vec{r}^{\,PC})$$

 We could have also derived the right equation from the left equation by using the two-point formula for acceleration

$$\vec{a}^C = \vec{a}^P + \vec{\omega} \times (\vec{\omega} \times \vec{r}^{PC}) + \vec{\alpha} \times \vec{r}^{PC}$$
 from family

Euler's Second Law

$$\sum \vec{M}^O = \frac{d\vec{H}^O}{dt}$$

- The above equation is valid for any point fixed in the inertial frame.
- Now, the moment and angular moment about one point can be related to the vectors about another point ...

$$\vec{M}^O = \vec{M}^C + \vec{r}^{OC} \times \vec{F}$$

$$\vec{H}^O = \vec{H}^C + \vec{r}^{OC} \times \vec{L}$$

$$\vec{H}^O$$

• Using the above (and Euler's first law) we get:

$$\sum \vec{M}^C = \dot{\vec{H}}^C$$

$$\vec{M}^O = \vec{M}^P + \vec{r}^{OP} \times \vec{F}$$

$$\vec{H}^O = \vec{H}^P + \vec{r}^{OP} \times \vec{L}$$

$$\sum \vec{M}^P = \dot{\vec{H}}^P + \dot{\vec{r}}^{OP} \times \vec{L}$$

• Now for a rigid body ...

$$\vec{H}^C = \int \vec{r}^{CQ} \times \vec{v}^Q dm$$

$$\vec{H}^P = \int \vec{r}^{PQ} \times \vec{v}^Q dm$$

$$\vec{v}^Q = \vec{v}^Q + \vec{\omega} \times \vec{r}^{CQ}$$

$$\vec{v}^Q = \vec{v}^Q + \vec{\omega} \times \vec{r}^{PQ}$$

Integrate and use the definition of the center of mass ...

$$\vec{H}^{C} = \vec{I}^{C} \cdot \vec{\omega}$$

$$\vec{I}^{C} \cdot \vec{\omega} = \int \left[\vec{r}^{CQ} \times (\vec{\omega} \times \vec{r}^{CQ}) \right] dm$$

$$= \int \left[(\vec{r}^{CQ} \cdot \vec{r}^{CQ}) \vec{\omega} - \vec{r}^{CQ} (\vec{r}^{CQ} \cdot \vec{\omega}) \right] dm$$

$$= \int \left[(\vec{r}^{CQ} \cdot \vec{r}^{CQ}) \vec{U} - \vec{r}^{CQ} \vec{r}^{CQ} \vec{v}^{CQ} \right] dm$$

$$\vec{I}^{P} \cdot \vec{\omega} = \vec{I}^{C} \cdot \vec{\omega} - m\vec{r}^{PC} \times (\vec{r}^{PC} \times \vec{\omega})$$

$$\vec{I}^{C} = \int \left[(\vec{r}^{CQ} \cdot \vec{r}^{CQ}) \vec{U} - \vec{r}^{CQ} \vec{r}^{CQ} \right] dm$$

$$\vec{I}^{C} = \int \left[(\vec{r}^{CQ} \cdot \vec{r}^{CQ}) \vec{U} - \vec{r}^{CQ} \vec{r}^{CQ} \right] dm$$

• Equations of Motion (assuming constant m and constant \vec{r}^{PC} $\vec{\vec{I}}^{C}$ $\vec{\vec{I}}^{P}$ [in B frame]):

$$\sum \vec{M}^C = \vec{ec{I}}^C \cdot \dot{ec{\omega}} + ec{\omega} imes \left(\vec{ec{I}}^C \cdot ec{\omega}
ight)$$
 ______ ~o $\vec{\Box}^C$

$$\sum \vec{M}^P = m\dot{\vec{r}}^{PC} \times \vec{v}^P + m\vec{r}^{PC} \times \dot{\vec{v}}^P + \vec{\vec{I}}^P \cdot \dot{\vec{\omega}} + \vec{\omega} \times \left(\vec{\vec{I}}^P \cdot \vec{\omega}\right) + \dot{\vec{r}}^{OP} \times \vec{L}$$

$$= m\vec{r}^{PC} \times \dot{\vec{v}}^P + \vec{\vec{I}}^P \cdot \dot{\vec{\omega}} + \vec{\omega} \times \left(\vec{\vec{I}}^P \cdot \vec{\omega}\right) - \text{not}$$

• You could also derive the second equation from the first by substituting for the moment about C in terms of the moment about P and the force, and then substituting for the force using the equation of motion in translation

Final Equations of Motion percent find
$$\sum \vec{F} = m\vec{v}^P - m\vec{r}^{PC} \times \dot{\vec{\omega}} + m\vec{\omega} \times (\omega \times \vec{r}^{PC})$$

$$\sum (\vec{M}^{P}) = m\vec{r}^{PC} \times (\vec{v}^{P}) + \vec{\vec{I}}^{P} \cdot (\vec{\omega} + \vec{\omega}) \times (\vec{\vec{I}}^{P} \cdot \vec{\omega})$$

- Note that these are vector equations and all derivatives are in the inertial frame
- The above equations will be solved in terms of components in a frame
- All the vectors except velocity are typically written in the body frame
- The velocity vector may be written in the inertial or body frame and you may need ...

$$\dot{\vec{v}}^P = {}^I\dot{\vec{v}}^P = {}^B\dot{\vec{v}}^P + \vec{\omega} \times \vec{v}^P$$

- You can mix and match, i.e., use velocity at P but write moment equation about C
- For 2D problems, it is possible to easily write all the vectors except $ec{r}^{PC}$ in the inertial frame.

Matrix Form (from wikipedia)

• From https://en.wikipedia.org/wiki/Newton-Euler equations

With respect to a coordinate frame located at point **P** that is fixed in the body and *not* coincident with the center of mass, the equations assume the more complex form:

$$\begin{pmatrix} \mathbf{F} \\ \boldsymbol{\tau}_{\mathrm{p}} \end{pmatrix} = \begin{pmatrix} m\mathbf{I}_{3} & -m[\mathbf{c}]^{\times} \\ m[\mathbf{c}]^{\times} & \mathbf{I}_{\mathrm{cm}} - m[\mathbf{c}]^{\times}[\mathbf{c}]^{\times} \end{pmatrix} \begin{pmatrix} \mathbf{a}_{\mathrm{p}} \\ \boldsymbol{\alpha} \end{pmatrix} + \begin{pmatrix} m[\boldsymbol{\omega}]^{\times}[\boldsymbol{\omega}]^{\times}\mathbf{c} \\ [\boldsymbol{\omega}]^{\times}(\mathbf{I}_{\mathrm{cm}} - m[\mathbf{c}]^{\times}[\mathbf{c}]^{\times})\boldsymbol{\omega} \end{pmatrix},$$

where c is the location of the center of mass expressed in the body-fixed frame, and

$$[\mathbf{c}]^ imes \equiv egin{pmatrix} 0 & -c_z & c_y \ c_z & 0 & -c_x \ -c_y & c_x & 0 \end{pmatrix} \qquad [oldsymbol{\omega}]^ imes \equiv egin{pmatrix} 0 & -\omega_z & \omega_y \ \omega_z & 0 & -\omega_x \ -\omega_y & \omega_x & 0 \end{pmatrix}$$

denote skew-symmetric cross product matrices.

Linear Equations of Motion

- For small rotations, we can derive linear equations of motion.
- We no longer need to differentiate between the inertial and body frame.
- The equations have a "nice" symmetric form.

... where, ξ are the component of $ec{r}^P$

... ξ is the dual or cross-product matrix of ξ

... Δ is a 3 x 3 identity matrix

... also

$$T = \frac{1}{2} \begin{Bmatrix} v^P \\ \omega \end{Bmatrix}^T \begin{bmatrix} m\Delta & m\tilde{\xi}^T \\ m\tilde{\xi} & I^P \end{bmatrix} \begin{Bmatrix} v^P \\ \omega \end{Bmatrix}$$