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# Heat Transfer

## Heat transfer laws for the forced convection in External Flow

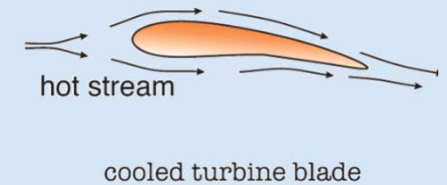
Prof. Dr.-Ing. Reinhold Kneer  
Dr.-Ing. Dr. rer. pol. Wilko Rohlfs  
Prof. dr. ir. Kees Venner



# Learning Goals

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- Forced Convection in External Flow
  - Knowledge and understanding of the Dimensionless numbers
  - Overview of different application cases and associated correlations



# Classifications according to flow regime

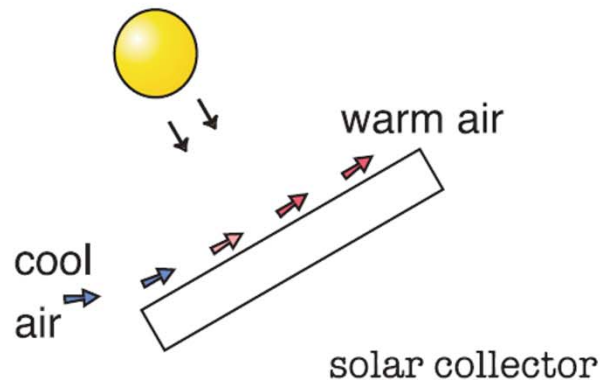
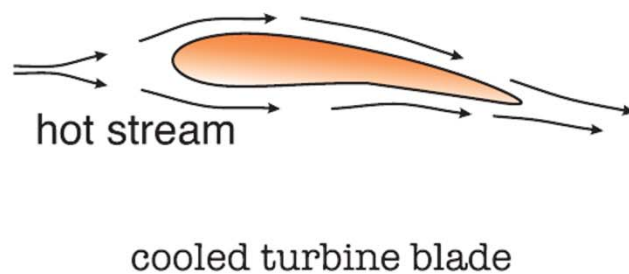
## Forced Convection

- Driven by externally generated movement of the fluid/object

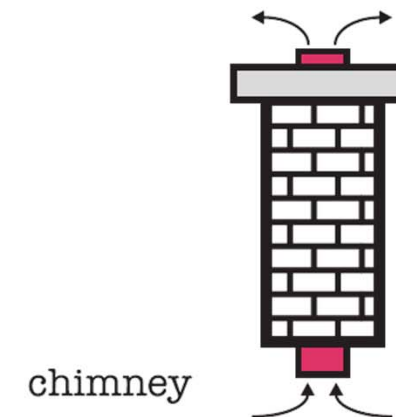
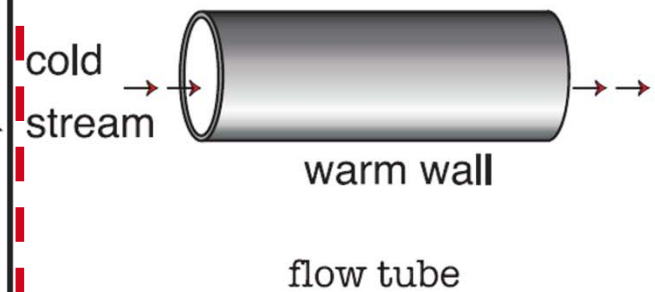
## Free Convection

- Inherently driven due to heat transfer (density differences)

### External



### Internal



# Heat flow and heat transfer coefficient for convection

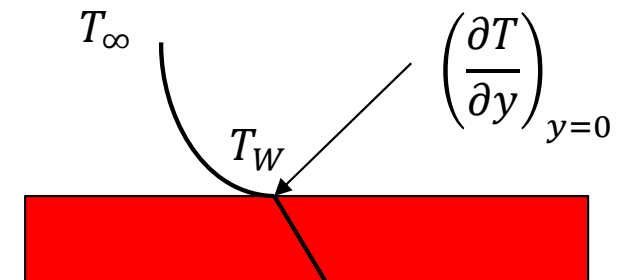
$$\dot{Q} = \alpha A (T_W - T_\infty)$$

$A$  = Convective transport area [m<sup>2</sup>]

$\Delta T$  = Temperature difference between wall temperature  $T_w$  and fluid temperature  $T_\infty$ .

$\alpha$  = Heat transfer coefficient [W/m<sup>2</sup>K]

$$\alpha = \frac{-\lambda_f \left( \frac{\partial T}{\partial y} \right)_{y=0,f}}{(T_W - T_\infty)}$$



Alternative:

$$\alpha = \frac{-\lambda_W \left( \frac{\partial T}{\partial y} \right)_{y=0,W}}{(T_W - T_\infty)}$$



## Important Dimensionless numbers

Nusselt number:  $Nu = \frac{\alpha L}{\lambda}$

The Nusselt number is the dimensionless heat transfer coefficient

Prandtl number:  $Pr = \frac{\eta c_p}{\lambda}$

The Prandtl number compares the momentum transport due to friction with the heat transport due to conduction/diffusion

Reynolds number:  $Re = \frac{\rho u_{\infty} L}{\eta}$

The Reynolds number gives the ratio of inertial forces to viscous forces

Grashof number:  $Gr = \frac{\rho^2 g \beta (T_w - T_{fl}) L^3}{\eta^2}$

The Grashof number describes the relationship between the buoyancy forces of a fluid and the acting viscosity forces



# Nusselt number function in Forced Convection

## General Form

Forced Convection:  $Nu = Nu(Re, Pr)$

## Application criteria

- Geometry
- Flow regime
- Thermal boundary conditions

## Substance Properties

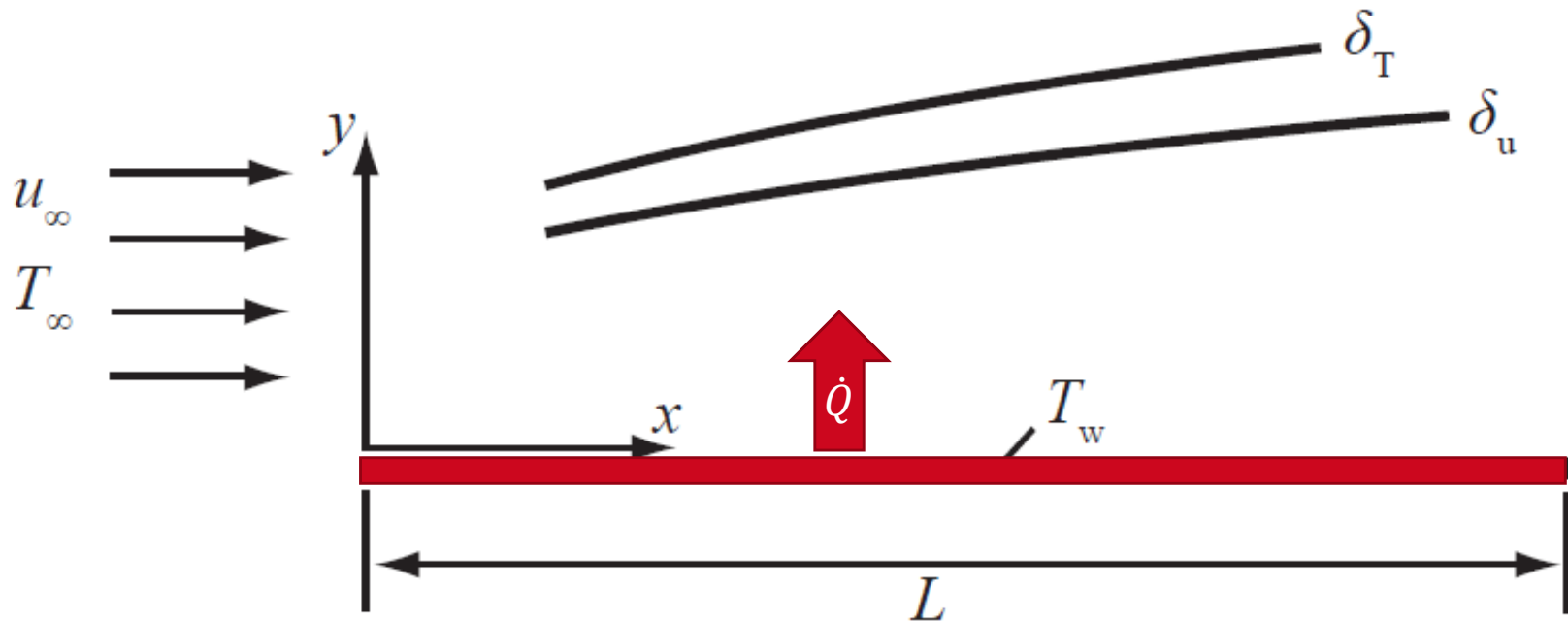
For the calculation of the substance properties, use the average temperature:

$$T_{\text{Prop}} = \frac{T_W + T_{\infty}}{2}$$



## Flat plate – laminar boundary layer flow

### 5.1.1.1 Flat plate - laminar boundary layer flow $Re_{x,crit} \approx 2 \cdot 10^5$



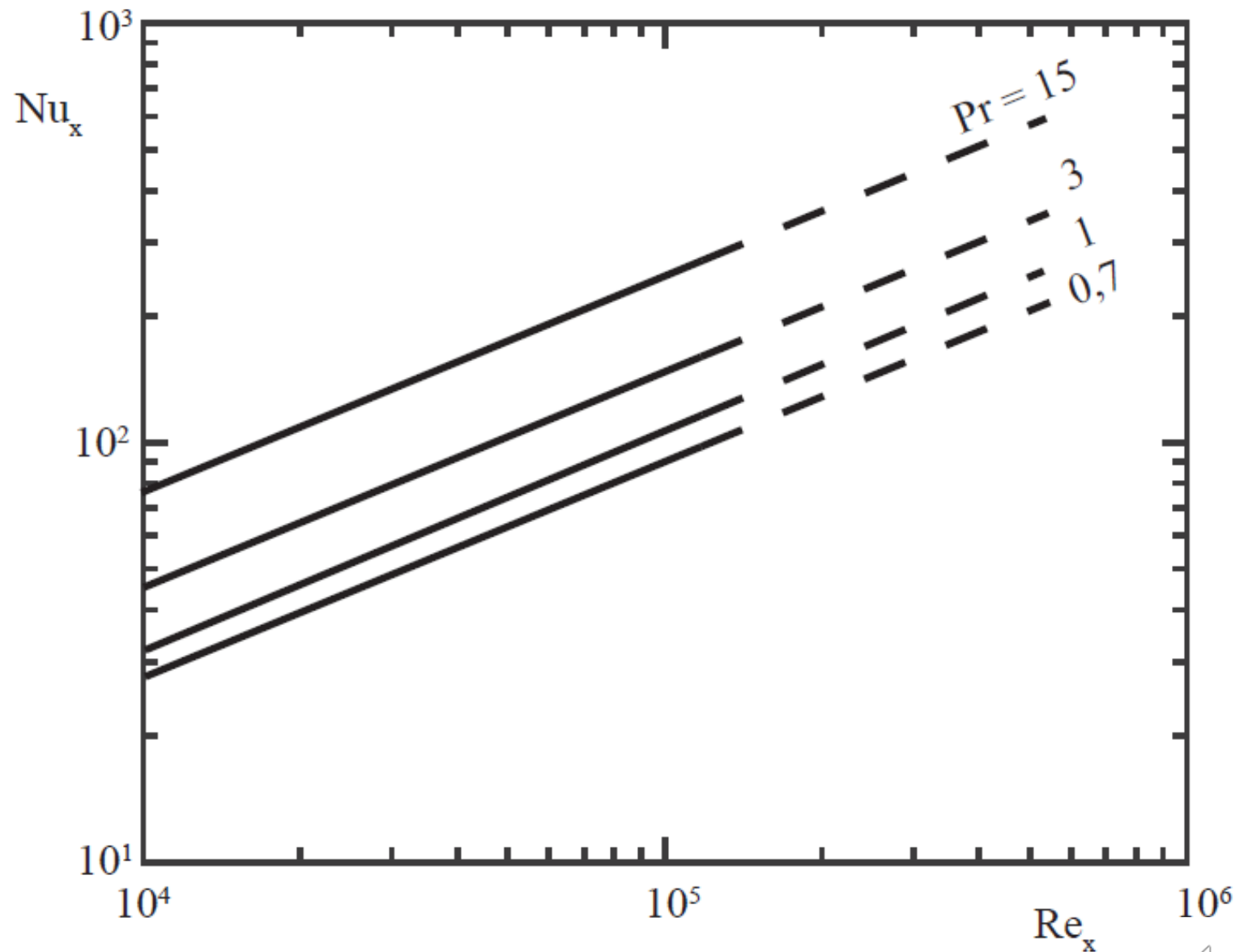
$$Nu_x = \frac{\alpha x}{\lambda}$$

The characteristic length  $x$  is perpendicular to the heat flow!



# Flat plate with laminar boundary layer, isothermal surface, simultaneous hydrodynamic and thermal inflow

$$Nu_x = \frac{\alpha x}{\lambda}$$





# Flat plate with laminar boundary layer, isothermal surface, simultaneous hydrodynamic and thermal inflow

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Nusselt law for the local heat transfer:

$$Nu_x = 0,332 Re_x^{\frac{1}{2}} Pr^{\frac{1}{3}} \quad (\text{HTC.1})$$

$$\begin{aligned} &\text{for } 0,6 < Pr < 10 \\ &\text{and } Re_x < Re_{x,\text{crit}} \approx 2 \cdot 10^5 \end{aligned}$$

$$Nu_x = \frac{\alpha x}{\lambda} = 0.332 \left( \frac{\rho u x}{\eta} \right)^{1/2} \cdot Pr^{1/3}$$

$$\alpha = 0.332 \cdot \lambda \left( \frac{\rho u}{\eta x} \right)^{1/2} \cdot Pr^{1/3}$$

The heat transfer coefficient decreases in the x-direction in the same way as the thermal boundary layer increases



# Flat plate with laminar boundary layer, isothermal surface, simultaneous hydrodynamic and thermal inflow

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Nusselt law for the local heat transfer:

$$\text{Nu}_x = 0,332 \text{Re}_x^{\frac{1}{2}} \text{Pr}^{\frac{1}{3}} \quad (\text{HTC.1})$$

$$\begin{aligned} &\text{for } 0,6 < \text{Pr} < 10 \\ &\text{and } \text{Re}_x < \text{Re}_{x,\text{crit}} \approx 2 \cdot 10^5 \end{aligned}$$

Nusselt law for the mean heat transfer:

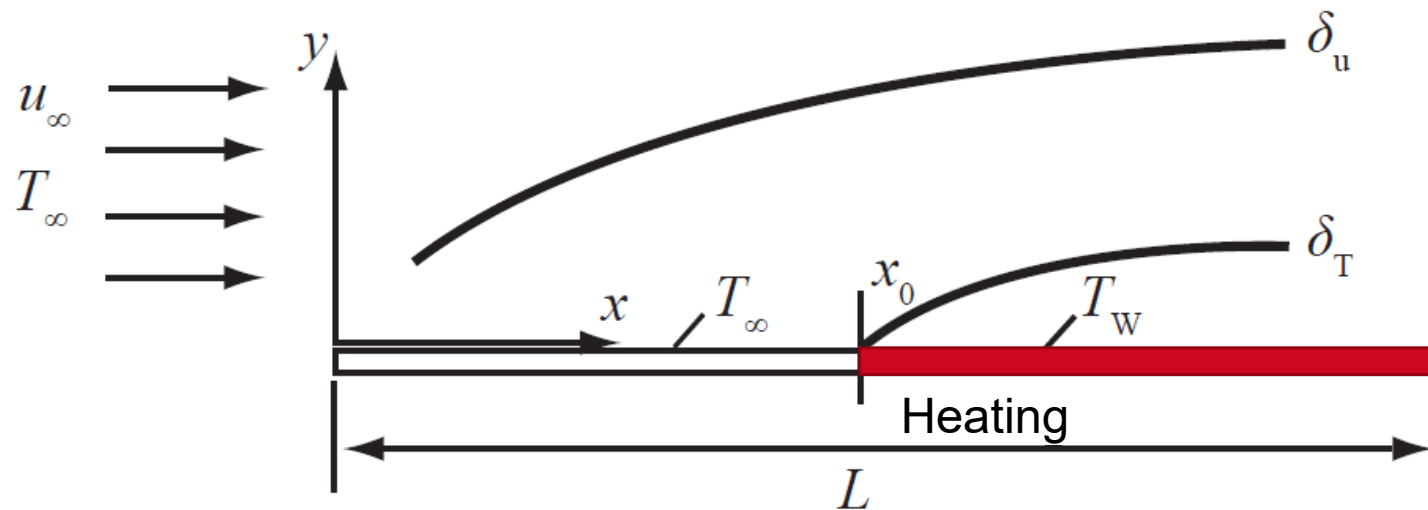
$$\overline{\text{Nu}}_L = \frac{\bar{\alpha}L}{\lambda} = \frac{1}{L} \int_0^L \alpha(x) dx = 2 \text{Nu}_{x=L}$$

$$\overline{\text{Nu}}_L = 0,664 \text{Re}_L^{\frac{1}{2}} \text{Pr}^{\frac{1}{3}} \quad (\text{HTC.2})$$

$$\begin{aligned} &\text{for } 0,6 < \text{Pr} < 10 \\ &\text{and } \text{Re}_L < \text{Re}_{L,\text{crit}} \approx 2 \cdot 10^5 \end{aligned}$$



# Flat plate with laminar boundary layer, isothermal surface, hydrodynamic inlet, heating or cooling starting at $x = x_0$

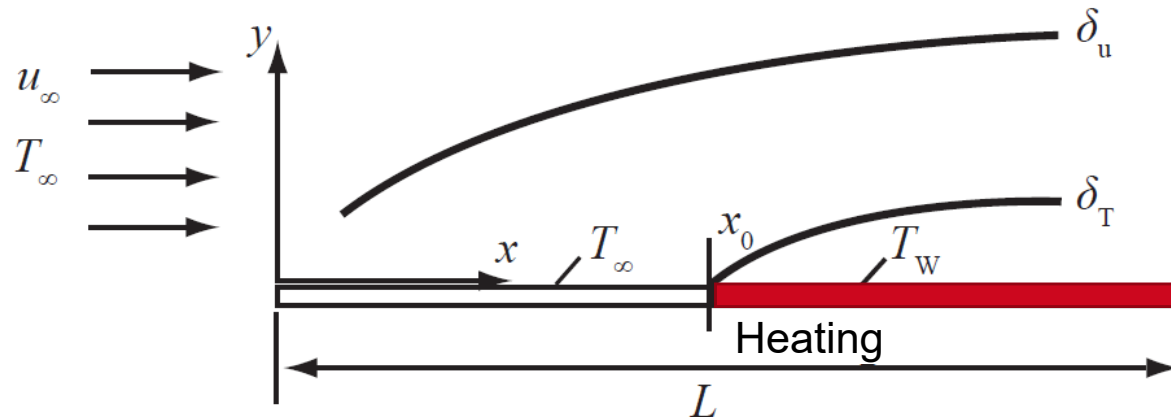


Nusselt law for the local heat transfer:

$$\text{Nu}_x = 0,332 \text{Re}_x^{\frac{1}{2}} \text{Pr}^{\frac{1}{3}} \left[ 1 - \left( \frac{x_0}{x} \right)^{\frac{3}{4}} \right]^{-\frac{1}{3}} \quad (\text{HTC.3})$$

$$\begin{aligned} &\text{for } 0,6 < \text{Pr} < 10 \\ &\text{and } \text{Re}_x < \text{Re}_{x,\text{crit}} \approx 2 \cdot 10^5 \end{aligned}$$

# Flat plate with laminar boundary layer, isothermal surface, hydrodynamic inlet, heating or cooling starting at $x = x_0$



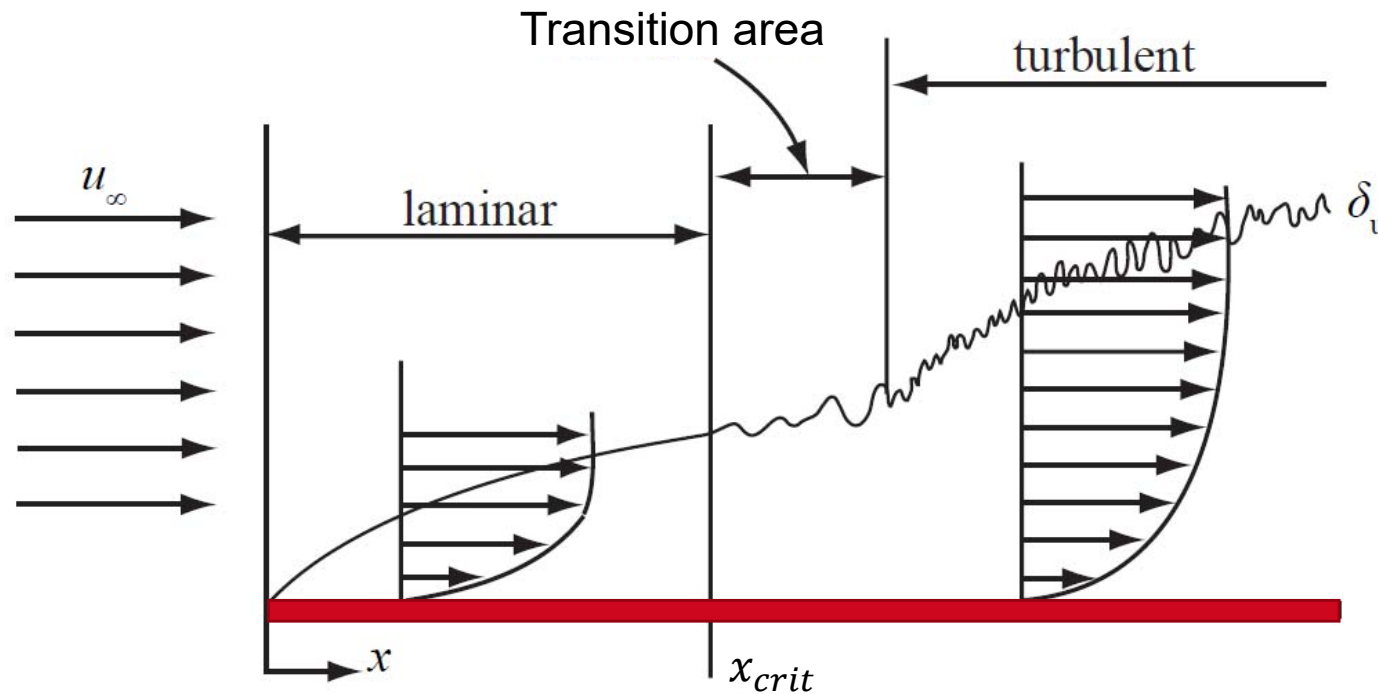
Nusselt law for the mean heat transfer:

$$\text{Nu}_L = \frac{L}{L - x_0} \frac{1}{\lambda} \int_{x_0}^L \alpha(x) dx$$
$$\text{Nu}_L = 0,664 \text{Re}_L^{\frac{1}{2}} \text{Pr}^{\frac{1}{3}} \frac{\left[1 - \left(\frac{x_0}{L}\right)^{\frac{3}{4}}\right]^{\frac{2}{3}}}{\left[1 - \frac{x_0}{L}\right]} \quad (\text{HTC.4})$$

$$\begin{aligned} &\text{for } 0,6 < \text{Pr} < 10 \\ &\text{and } \text{Re}_x < \text{Re}_{x,\text{crit}} \approx 2 \cdot 10^5 \end{aligned}$$



# Flat plate with turbulent boundary layer and isothermal surface



$$Re_{x,crit} \approx 2 \cdot 10^5$$

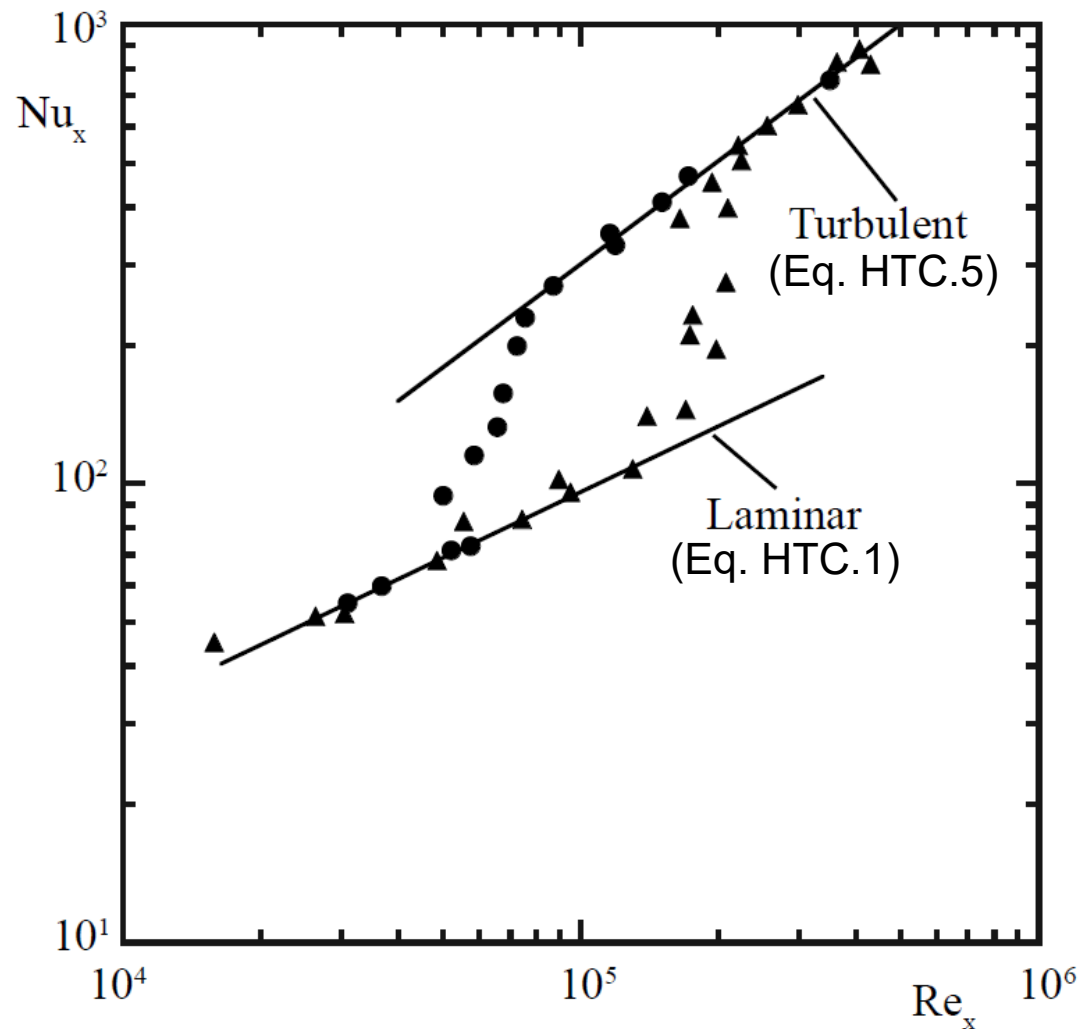
$$Re_{x,crit} = \frac{\rho u x_{crit.}}{\eta} = 2 \cdot 10^5$$

$$x_{crit.} = 2 \cdot 10^5 \frac{\eta}{\rho u}$$

The turbulent envelope shifts backwards with increasing viscosity and forwards with increasing flow velocity



# Flat plate with turbulent boundary layer and isothermal surface



Influence of the average degree of turbulence of the incident flow on the laminar/turbulent transition (triangles and circles)

In both cases, the laminar or turbulent heat transfer is well represented by the two correlations (HTC.1 and HTC.5)

$$\text{Nu}_x = 0,0296 \text{Re}_x^{0,8} \text{Pr}^{0,43} \quad \text{for} \quad 5 \cdot 10^5 < \text{Re}_x < 10^7 \quad (\text{HTC.5})$$

# Average heat transfer for a flat plate with turbulent boundary layer and isothermal surface

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Nusselt law for mean heat transfer:

(Note : The average heat transfer of a plate with length L can be calculated by integrating equations (WÜK.1) and (WÜK.5)

$$\overline{Nu}_L = \frac{\bar{\alpha}L}{\lambda} = \frac{1}{\lambda} \left( \int_0^{x_{crit}} \alpha(x)_{eq. (HTC.1)} dx + \int_{x_{crit}}^L \alpha(x)_{eq. (HTC.5)} dx \right) \quad (5.8)$$

Hence, the result depends on the actual critical Reynolds number).

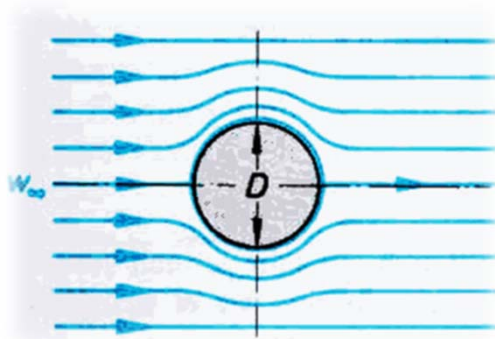
Nusselt law for mean heat transfer:

$$\overline{Nu}_L \approx 0,036 Pr^{0,43} (Re_L^{0,8} - 9400) \quad (HTC.6)$$

$$\text{for } Re_{L,crit} \approx 2 \cdot 10^5$$

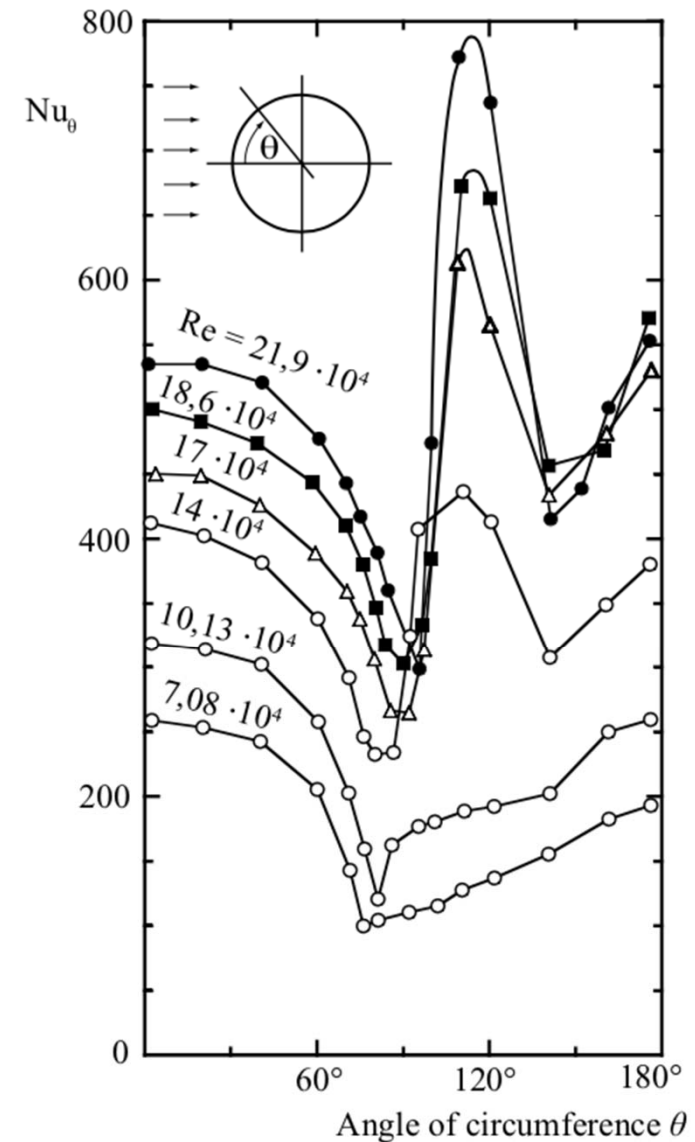


# Transverse flow around cylinder



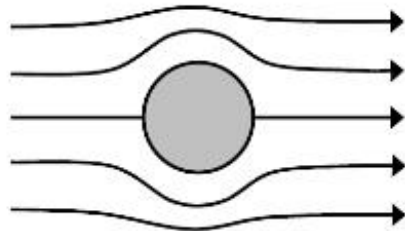
$$\overline{Nu_d} = C Re_d^m Pr^{0,4}$$

$Re_d$	$C$	$m$
0,4 - 4	0,989	0,330
4 - 40	0,911	0,385
40 - 4000	0,683	0,466
4000 - 40000	0,193	0,618
40000 - 400000	0,0266	0,805

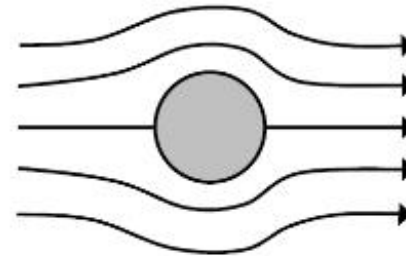




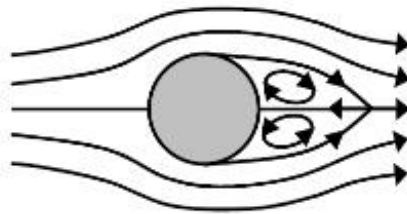
# Transverse flow around cylinder as a function of the Reynolds number



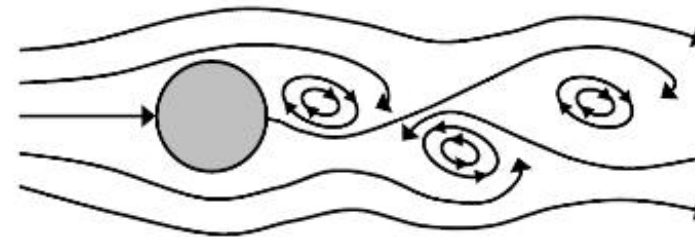
Frictionless flow:  $Re = \infty$



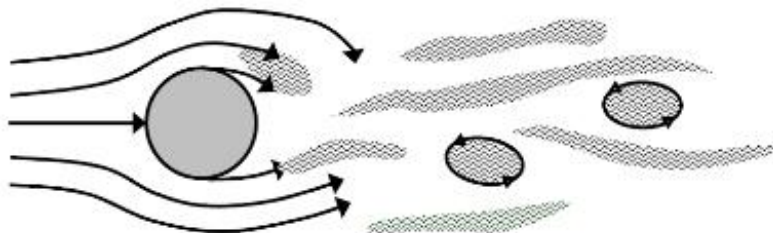
$Re \approx 0.01$



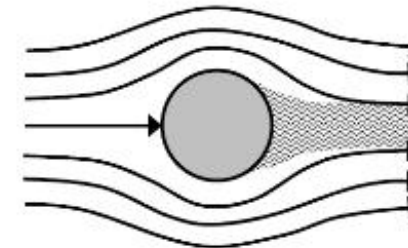
$Re \approx 20$



$Re \approx 100$



$Re \approx 10\,000$



$Re \approx 10\,000\,000$



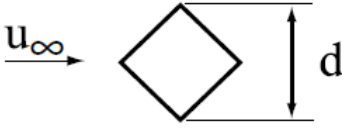
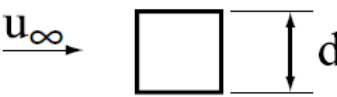
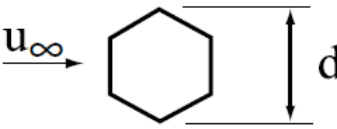
# Transverse flow around non-circular cylinder

Mean heat transfer for non circular cylinders

$$\overline{Nu_d} = C Re_d^m Pr^{0,4}$$

Constants of equation (HTC.9) according to Jakob (1949)

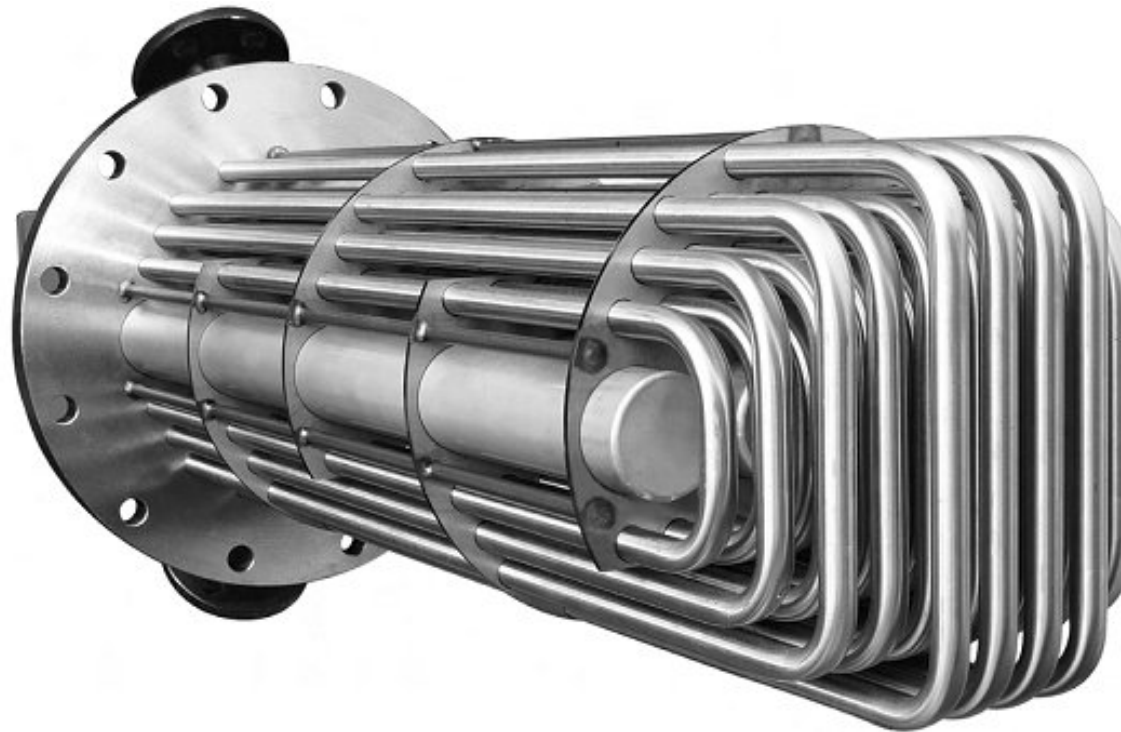
Table 5.2: Constants  $C$  and  $m$  of equation (HTC.9)

Geometry	$Re_d$	$C$	$m$
	$5 \cdot 10^3 - 10^5$	0,246	0,588
	$5 \cdot 10^3 - 10^5$	0,102	0,675
	$5 \cdot 10^3 - 1,94 \cdot 10^4$	0,160	0,638



# Transverse flow around bundle of smooth tubes

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<https://www.heatsystems.de/produkt-details/rohrbuendel-waermetauscher.html>



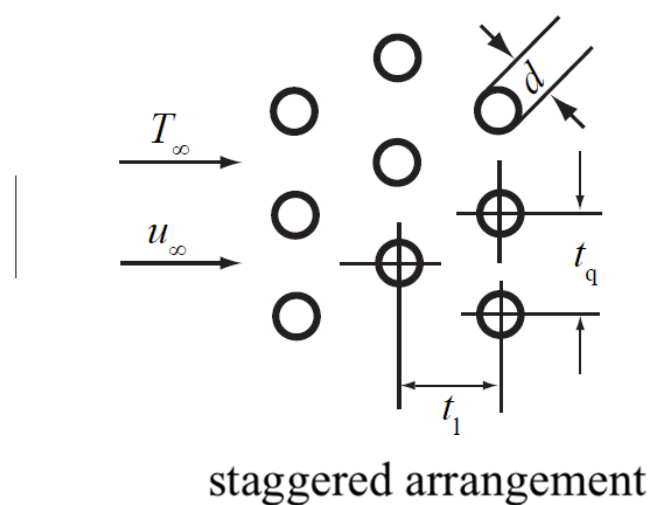
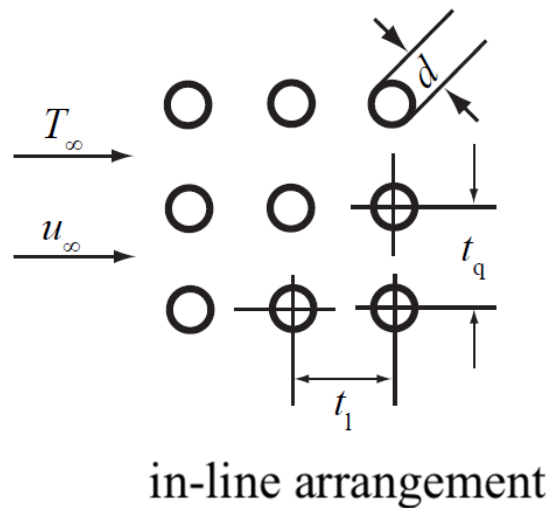
# Transverse flow around bundle of smooth tubes

(Note: Since the temperature of the fluid in the tube bundle changes from row to row, a representative mean temperature difference must be used calculating the heat transfer.)

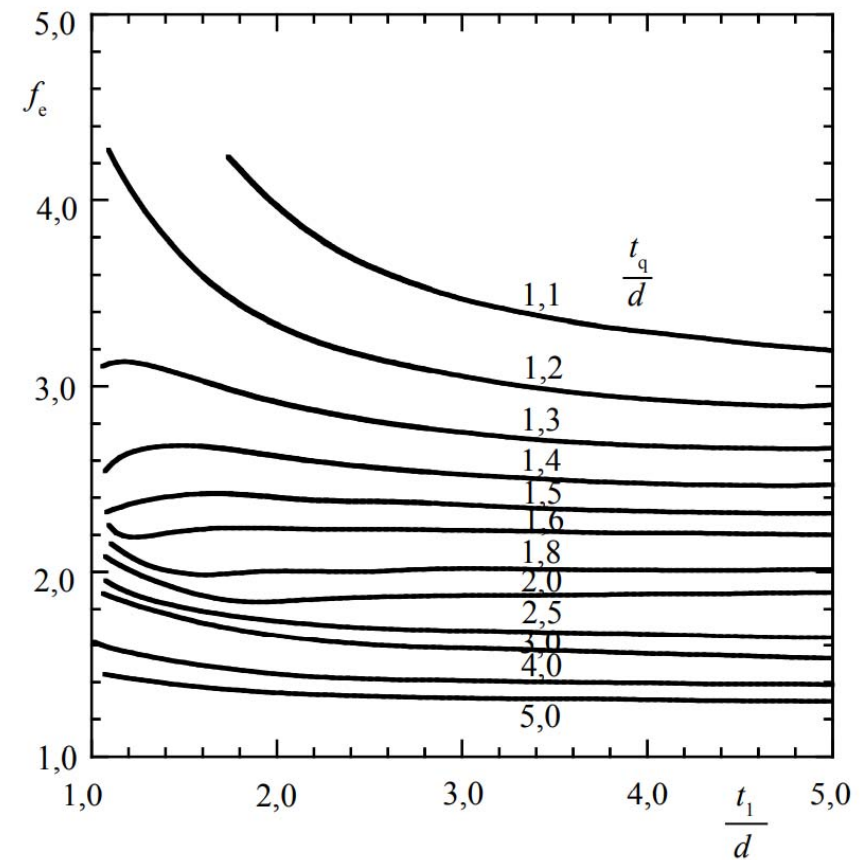
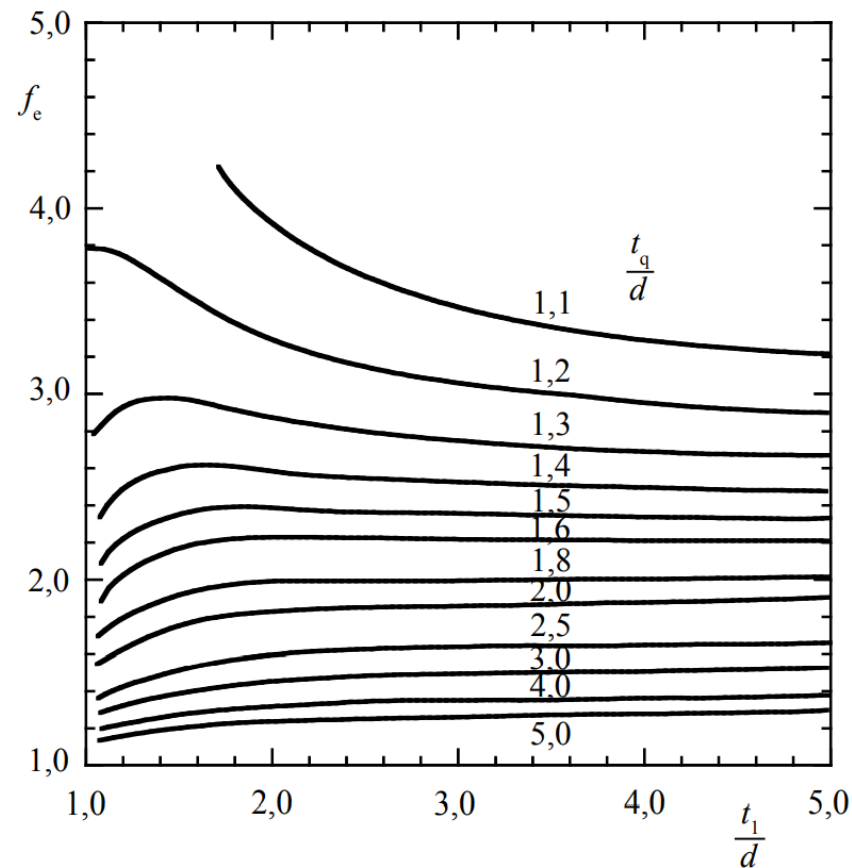
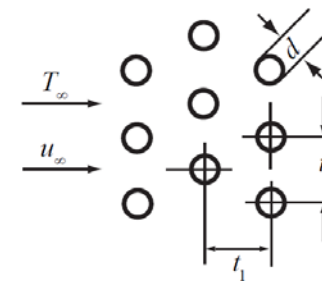
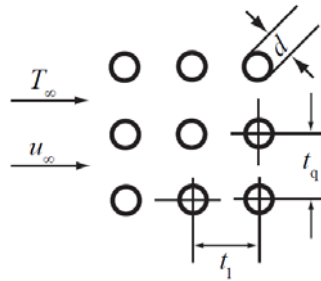
$$\dot{Q}_W = \bar{\alpha} A (T_W - T_{fl})_m = \bar{\alpha} A \Delta T_m \quad (5.9)$$

For many practical purposes, the logarithmic temperature difference is a good approximation

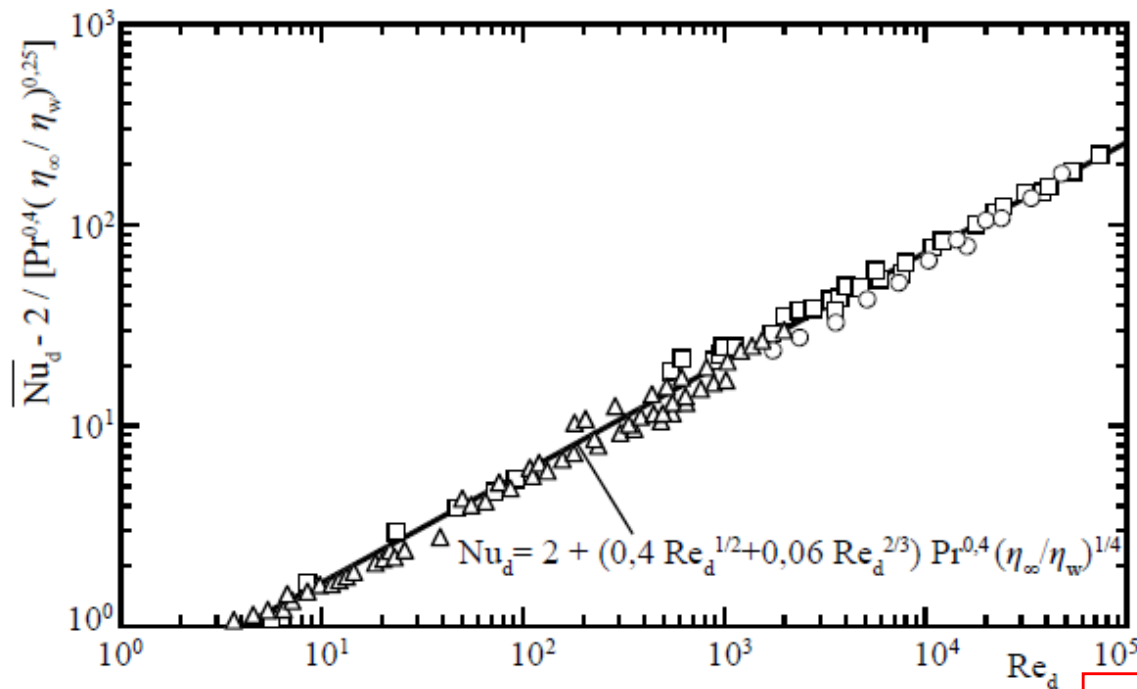
$$\Delta T_m = \Delta T_{ln} \equiv \frac{\Delta T_{inlet} - \Delta T_{outlet}}{\ln \frac{\Delta T_{inlet}}{\Delta T_{outlet}}}$$



# Transverse flow around bundle of smooth tubes



# Flow around a sphere



- Whitaker correlation

$$\overline{Nu}_d = 2 + \left( 0,4 Re_d^{\frac{1}{2}} + 0,06 Re_d^{\frac{2}{3}} \right) Pr^{0,4} \left( \frac{\eta_\infty}{\eta_w} \right)^{\frac{1}{4}} \quad (\text{HTC.11})$$

for  $0,7 < Pr < 380$   
and  $3,5 < Re_d < 8 \cdot 10^4$

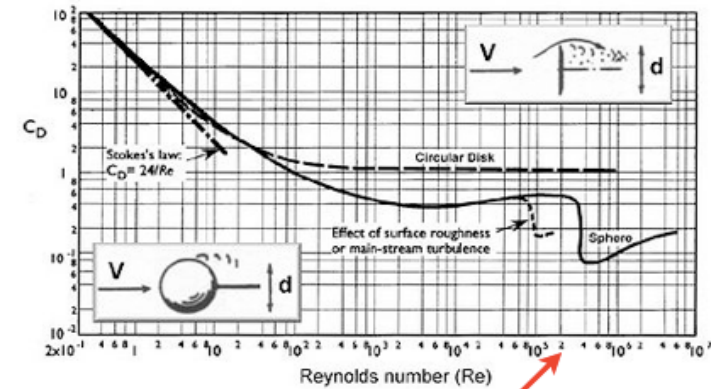
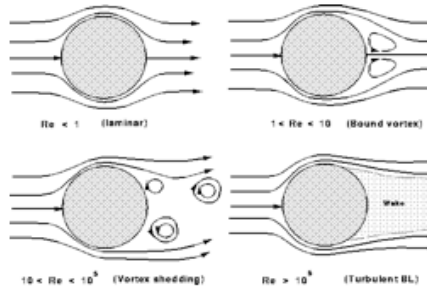
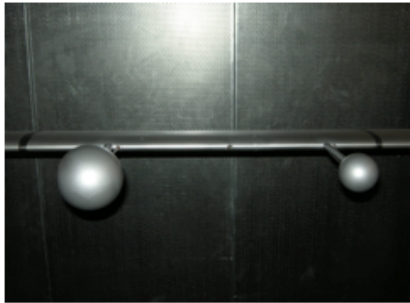
If  $u = 0$  (sphere at rest)  $\rightarrow Nu_d = 2$

3 Whitaker, S. "Forced Convection Heat Transfer Correlations for Flow in Pipes, Past Flat Plates, Single Cylinders, Single Spheres, and for Flow in Packed Beds and Tube Bundles", A. I. Ch. E. Journal, Vol. 18 (1972), pp. 361-371

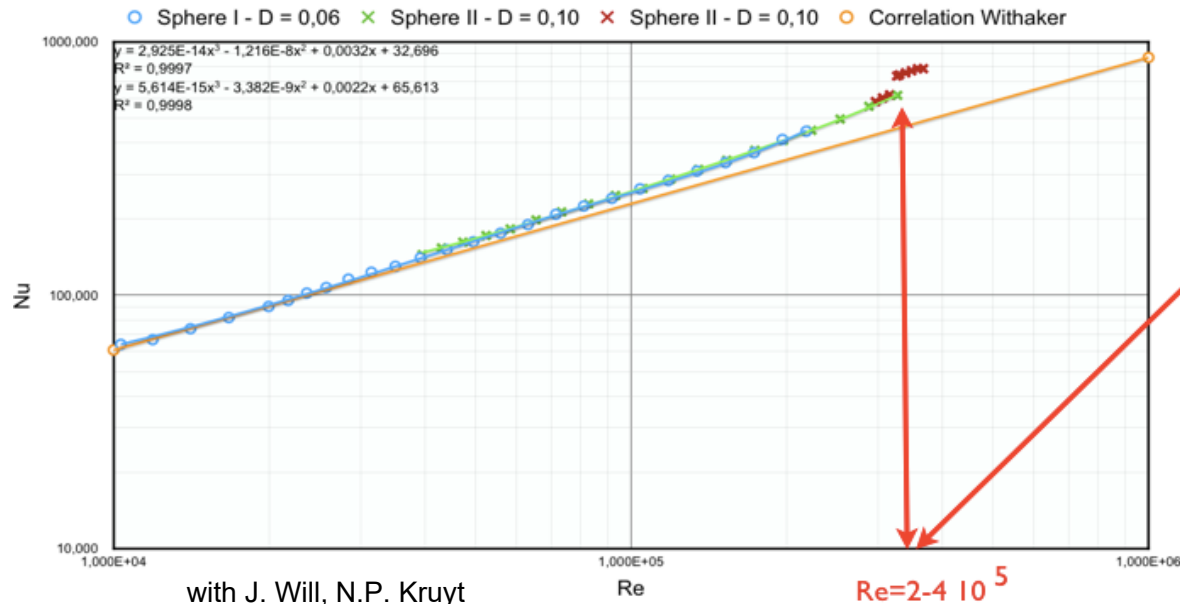


# Flow around a sphere

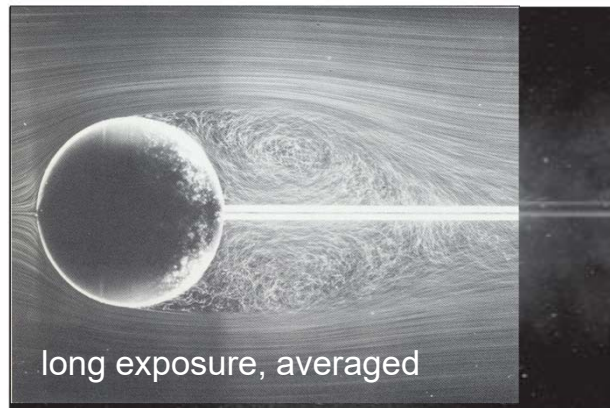
$$\alpha = \dot{Q} / (A (T_W - T_\infty))$$



source: Dr J. Christy

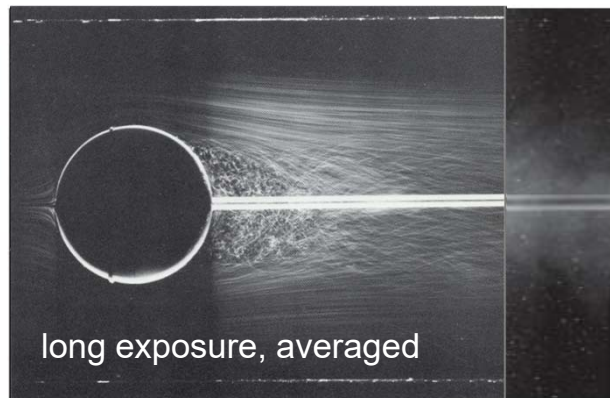


# Sphere in Flow: Visualization



(a)

Instantaneous flow past smooth sphere at  $R=15,000$ . Dye in water shows laminar boundary layer separating before the equator and remaining laminar, and then unstable and turbulent.



(b)

Instantaneous flow past a smooth sphere at  $Re=30,000$  with a trip wire. Classical Experiment of Prandtl and Wieselsberger repeated. Boundary layer turning turbulent delays separation, reduces drag, and enhances heat transfer.

Source: ONERA, H. Werle





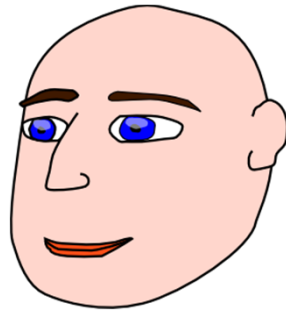
$$\overline{Nu_d} = 2 + \left( 0,4 Re_d^{\frac{1}{2}} + 0,06 Re_d^{\frac{2}{3}} \right) Pr^{0,4} \left( \frac{\eta_{\infty}}{\eta_w} \right)^{\frac{1}{4}}$$

for  $0,7 < Pr < 380$   
and  $3,5 < Re_d < 8 \cdot 10^4$

## Example Application: Heat Transfer from Sphere

$$T_{\infty} = -10 \text{ [}^{\circ}\text{C]}$$

$$U_{\infty} = 5 \text{ [m/s]}$$



$$T_w = 33 \text{ }^{\circ}\text{C}$$

$$D = 0.18 \text{ [m]}$$

$$A = \pi D^2 = 0.102 \text{ [m}^2\text{]}$$

$$Re = 62.230$$

$$Nu = 170.8$$

$$\alpha = Nu\lambda/D = 23.9 \text{ [W/(m}^2\text{K)]}$$

$$\dot{Q} = \alpha A(T_w - T_{\infty}) = 105 \text{ [W]}$$

$$\rho = 1.246 \text{ [kg/m}^3\text{]}$$

$$\lambda = 0.02476 \text{ [W/(mK)]}$$

$$\eta = 1.802 \cdot 10^{-5} \text{ [Pa.s]}$$

$$\eta_s = 1.872 \cdot 10^{-5} \text{ [Pa.s]}$$

$$Pr = 0.7323$$



$$T_w = 10 \text{ }^{\circ}\text{C}$$

$$Nu = 170.8$$

$$\alpha = Nu\lambda/D = 23.9 \text{ [W/(m}^2\text{K)]}$$

$$\dot{Q} = \alpha A(T_w - T_{\infty}) = 49 \text{ [W]}$$

(where I assumed the air parameters to stay the same)



# Comprehension Questions

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**Which dimensionless numbers have to be considered in forced convection?  
How is the applicability of a correlation checked?**

**At what temperature are the material properties occurring in the dimensionless numbers to be determined?**

**What is the difference between local and average heat transfer in a flat plate with heating or cooling?**



# Heat transfer with Natural and Forced Convection

## Example: Cooling a hot surface

