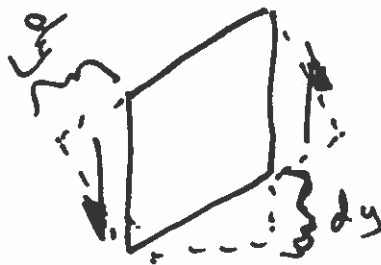
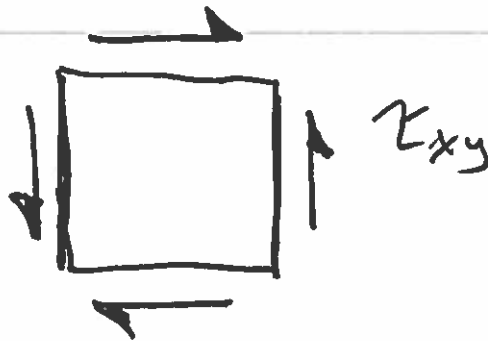
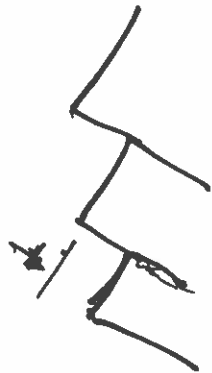
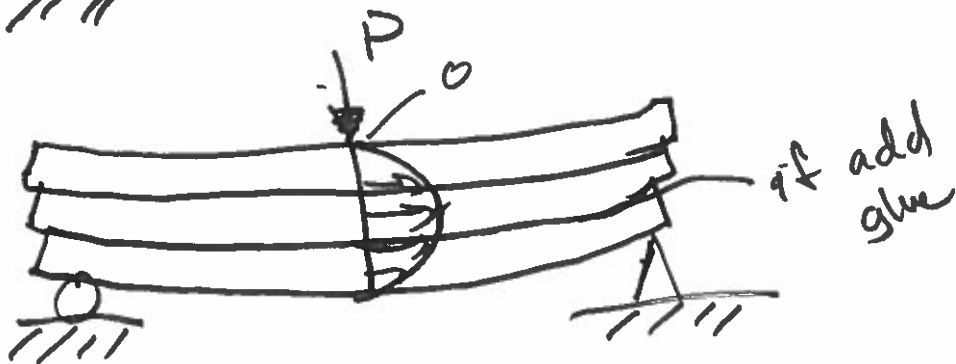
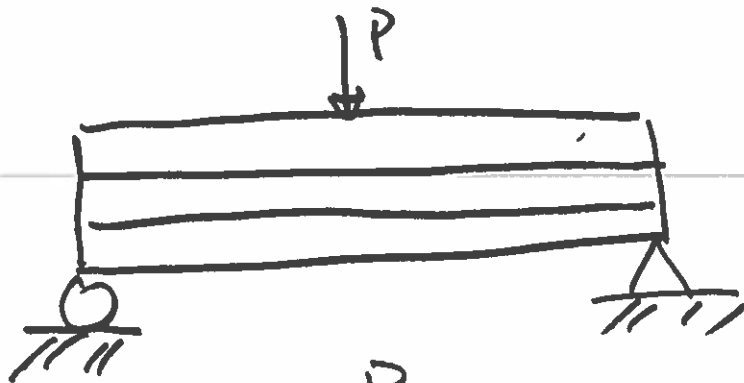


Complementary Shear

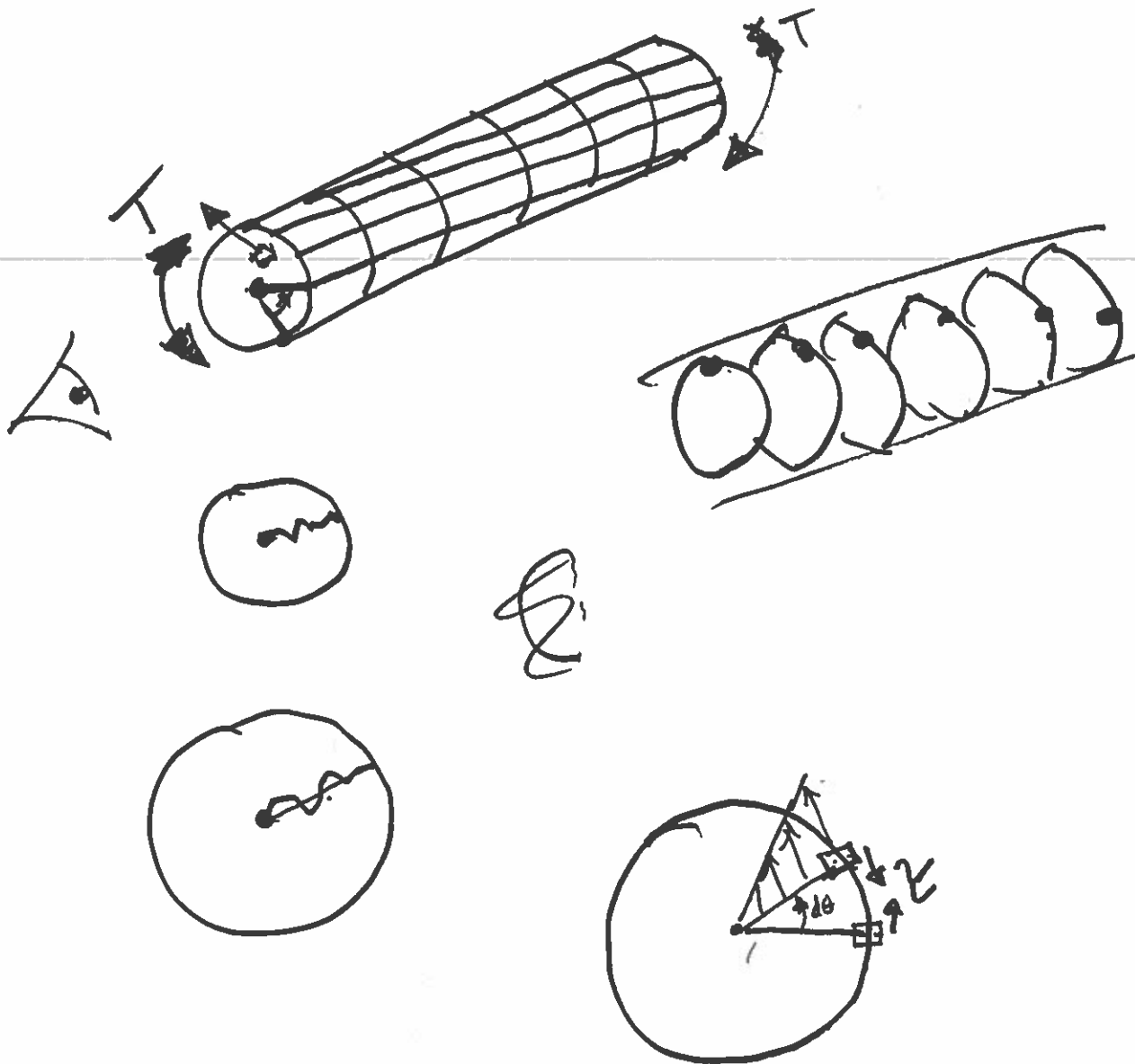
$$\tau_i = \tau_{xy}$$



Transverse Shear in Bending



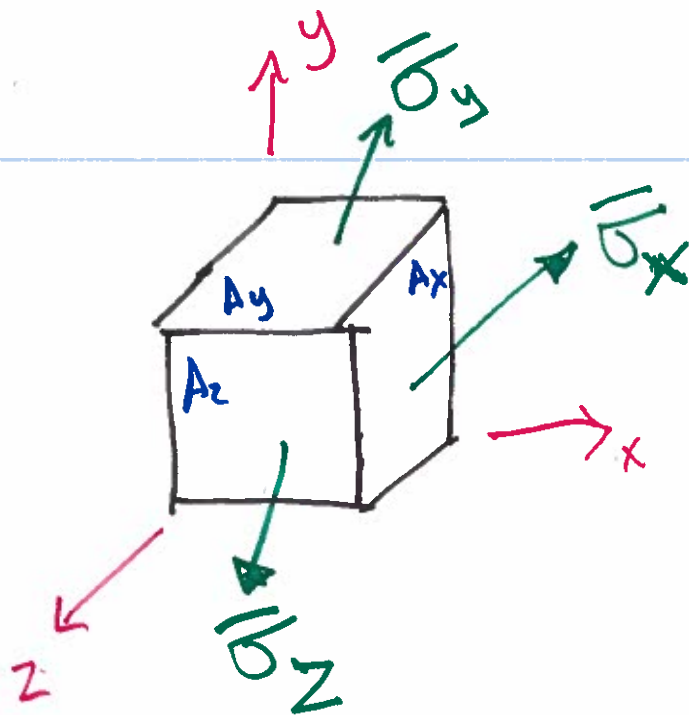
Torsion and Shear



Computing stress on an arbitrary plane

$$\vec{T}_x = \underbrace{\sigma_x}_{\text{normal}} \hat{i} + \underbrace{\sum}_{\text{shear}}$$

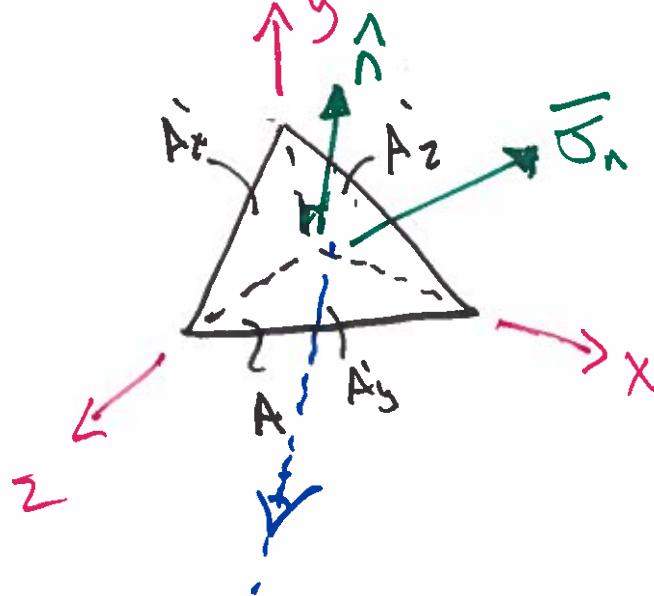
Normal
&
shear
stress
on
the
face



$$\hat{n} = l\hat{i} + m\hat{j} + n\hat{k}$$

$$|\hat{n}| = 1$$

$\left. \begin{matrix} l \\ m \\ n \end{matrix} \right\}$ direction
cosines



$$\sigma_n = \hat{n} \cdot \vec{\sigma}_n$$

$$\vec{\sigma}_x = \sigma_x \hat{i} + \tau_{xy} \hat{j} + \tau_{xz} \hat{k}$$

⋮

σ_z

$$\begin{aligned} \sigma_n = & l^2 \sigma_x + m^2 \sigma_y + n^2 \sigma_z \\ & + 2lm \tau_{xy} + 2ln \tau_{xz} \\ & + 2mn \tau_{yz} \end{aligned}$$

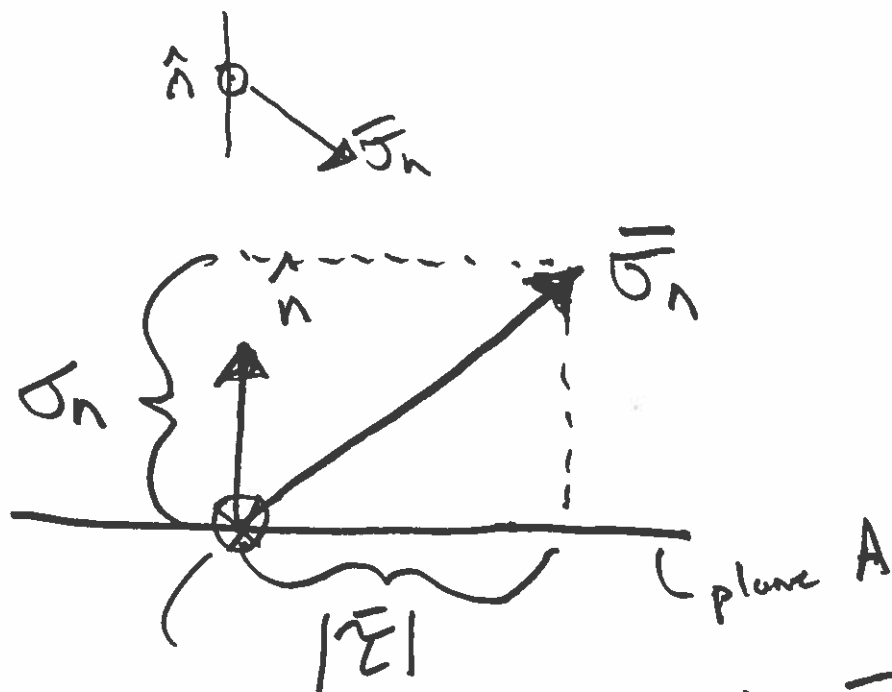
$$\Sigma F=0 = A\bar{\sigma}_n - A'_x \bar{\sigma}_x - A'_y \bar{\sigma}_y - A'_z \bar{\sigma}_z$$

Project A onto xy, xz, yz planes

$$A'_x = lA \quad A'_y = mA \quad A'_z = nA$$

$$\therefore \bar{\sigma}_n = l\bar{\sigma}_x + m\bar{\sigma}_y + n\bar{\sigma}_z$$

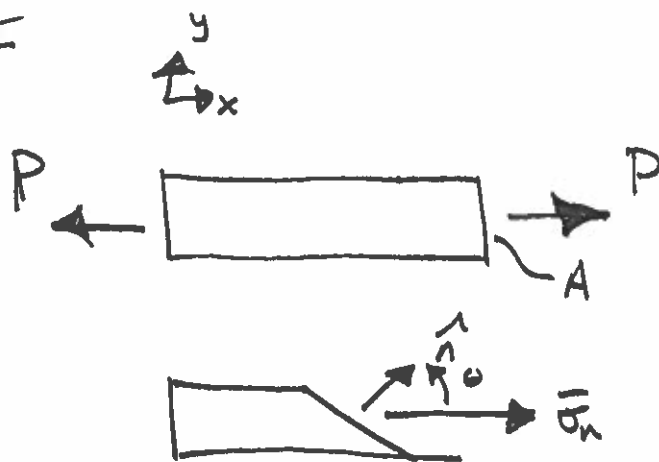
$\bar{\sigma}_n$: stress associated with the face A (defined by \hat{n}), contains the normal and shear stress



$$(\hat{n} \times \bar{\sigma}_n) \times \hat{n} = \bar{\tau}$$

$$|\sigma_n|^2 = \sigma_n^2 + \underbrace{|\tilde{\epsilon}|^2}$$

Example



Find a) $\bar{\sigma}_n$ as function of θ

b) ~~σ_n~~ and $|\bar{\tau}|$

$$\hat{n} = \underbrace{\cos \theta}_l \hat{i} + \underbrace{\sin \theta}_m \hat{j}$$

$$\bar{\sigma}_n = l \bar{\sigma}_x + m \bar{\sigma}_y + n \bar{\sigma}_z \quad n=0$$

$$\bar{\sigma}_n = l \bar{\sigma}_x = \cos \theta \frac{P}{A} \hat{i}$$

$$\bar{\sigma}_n \cdot \hat{n} = \sigma_n = \frac{P}{A} \cos^2 \theta$$

$$\bar{\sigma}_x = \frac{P}{A} \hat{i}$$

$$|\bar{\tau}|^2 + \sigma_n^2 = |\bar{\sigma}_n|^2$$

$$|\bar{\tau}| = \sqrt{|\bar{\sigma}_n|^2 - \sigma_n^2}$$

$$= \sqrt{\cos^2 \theta \frac{P^2}{A^2} - \frac{P^2}{A^2} \cos^4 \theta}$$

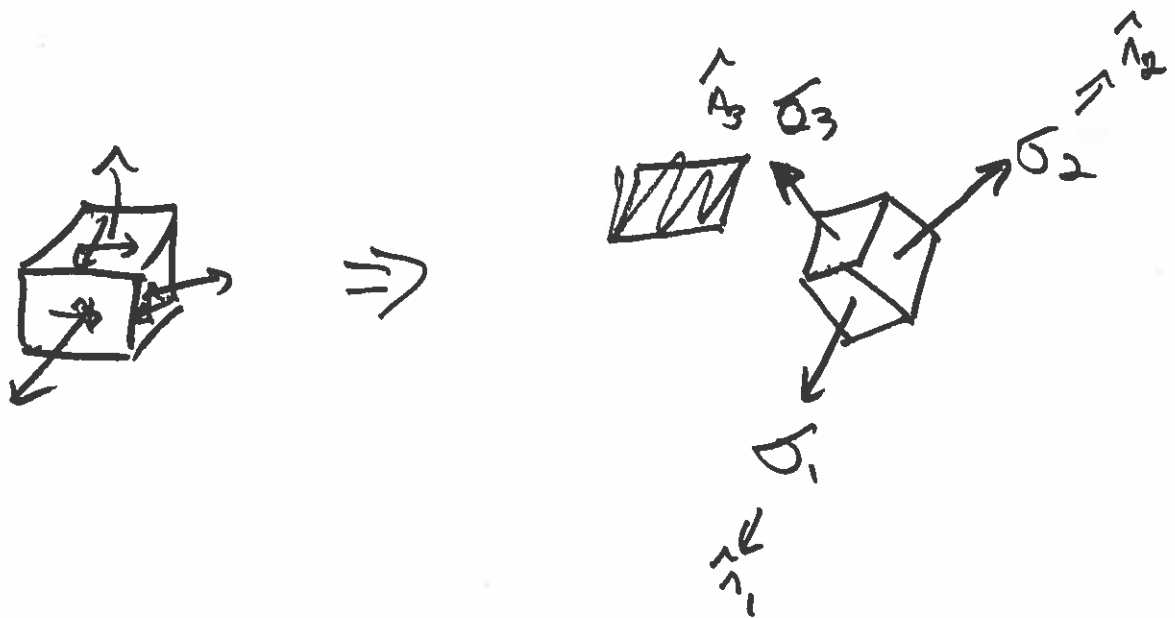
$$= \frac{P}{A} \cos \theta \sqrt{1 - \cos^2 \theta} \sin^2 \theta$$

$$= \frac{P}{A} \cos \theta \sin \theta$$

Principal Stresses

From the theory of Elasticity

At any point with a general state of stress, an element can be oriented such that the shear components vanish.



principal stresses: $\sigma_1, \sigma_2, \sigma_3$

principal directions: $\hat{n}_1, \hat{n}_2, \hat{n}_3$

Eigenvalue and eigenvector problem

$$\begin{array}{c} A \\ \left[\begin{array}{c} 3 \times 3 \end{array} \right] \end{array} \rightarrow \left[\begin{array}{ccc} \sigma_1 & & 0 \\ 0 & \sigma_2 & \\ & & \sigma_3 \end{array} \right]$$

Symmetric

$$A x = 0$$

\uparrow
 \hat{n}_1
 \hat{n}_2
 \hat{n}_3