



NORTH CAROLINA A&T STATE UNIVERSITY

FINAL PROJECT FOR ECEN-865 OPTIMAL CONTROL
THEORY

Development of LQ Tracker for Inverted Pendulum

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1 Abstract

In this project a linear quadratic(LQ) tracker is developed for a linear time invariant(LTI) system. The selected LTI system is an inverted pendulum which have four states. The vertical angle of the pendulum is controlled using the LQ tracker. The LQ tracker is developed in MATLAB programming environment and simulated with different reference signal. The controller gain is adjusted in such a way that the overall absolute error between the reference signal and the output signal is less than 0.1. In LQ tracker design the trade off between the input energy and tracking error is shown.

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2 Introduction

The linear quadratic tracker is a very efficient tracker in the context of gain margin and phase margin. The LQ tracker has infinite gain margin and at least 60° phase margin. So, the tracker is quite robust for stability of the system. In this project a LQ tracker is developed for an inverted pendulum. In the design step, first the mathematical model of the system is developed then the state space representation is formulated with the developed model. In next step, a cost function is defined for the tracking problem and then using the calculus of variation the cost function is minimized. After this formulation, we get two differential riccati equation(DRE) which is solved using Runge–Kutta method. At last the system is simulated with two different reference signal(step and sinusoidal). The following sections are organized as section 3: System model of inverted pendulum and State space representation, section 4: LQ tracker formulation, section 5: Simulation, section 6: Conclusion.

3 System model of inverted pendulum and State space representation [1]

The cart with an inverted pendulum, shown in figure 1, is "bumped" with an impulse force, F . Determine the dynamic equations of motion for the system, and linearize about the pendulum's angle, $\theta = \pi$ (in other words, assume that pendulum does not move more than a few degrees away from the vertical, chosen to be at an angle of π).

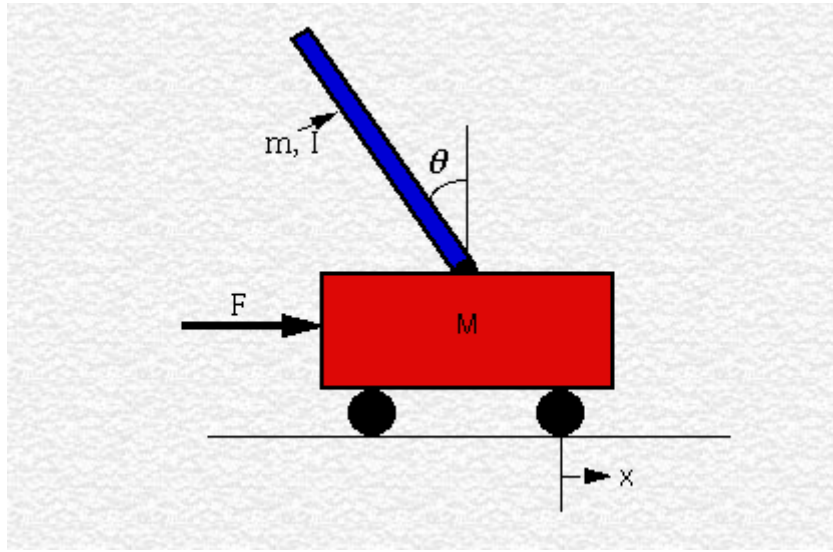


Figure 1: An inverted pendulum

here,

M = mass of the cart

m = mass of the pendulum

b = friction of the cart

I = inertia of the pendulum

L = length of the pendulum's center of mass

F = impulse force applied to cart

3.1 Force analysis and system equations

In figure 2 the free body diagram of the system is shown.

Summing the forces in the Free Body Diagram of the cart in the horizontal direction,

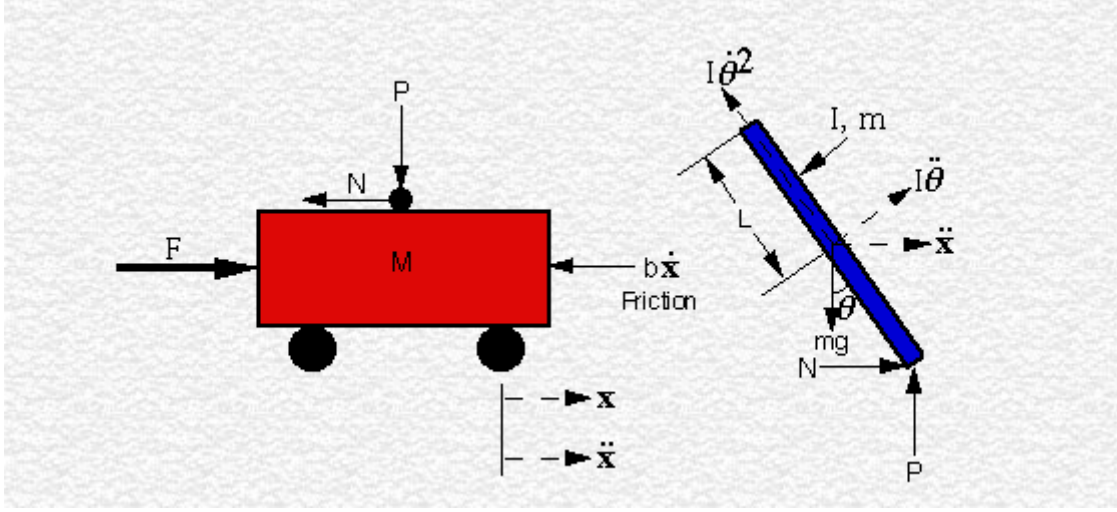


Figure 2: Free body diagrams of the system

we get the following equation of motion:

$$M\ddot{X} + b\dot{X} + N = F \quad (1)$$

Summing the forces in the Free Body Diagram of the pendulum in the horizontal direction, we get an equation for N:

$$N = m\ddot{X} + mL\ddot{\theta}\cos\theta - mL\dot{\theta}^2\sin\theta \quad (2)$$

If we substitute this equation into the first equation, we get the first equation of motion for this system:

$$(M + m)\ddot{X} + b\dot{x} + mL\ddot{\theta}\cos\theta - mL\dot{\theta}^2\sin\theta = F \quad (3)$$

To get the second equation of motion, sum the forces perpendicular to the pendulum. Solving the system along this axis ends up saving you a lot of algebra. You should get the following equation:

$$P\sin\theta + N\cos\theta - mg\sin\theta = mL\ddot{\theta} + m\ddot{X}\cos\theta \quad (4)$$

To get rid of the P and N terms in the equation above, sum the moments around the centroid of the pendulum to get the following equation:

$$-PI\sin\theta - NI\cos\theta = I\ddot{\theta} \quad (5)$$

Combining these last two equations, you get the second dynamic equation:

$$(I + mL^2)\ddot{\theta} + mgL\sin\theta = -mL\ddot{X}\cos\theta \quad (6)$$

This set of equations should be linearized about $\theta = \pi$. Assume that $\theta = \pi + \phi$ (ϕ represents a small angle from the vertical upward direction). Therefore, $\cos(\theta) = -1$, $\sin(\theta) = -\phi$, $\ddot{\theta}^2 = 0$ and $F = u$. After linearization the two equations of motion become:

$$(I + mL^2)\ddot{\phi} - mgL\phi = mL\ddot{X} \quad (7)$$

$$(M + m)\ddot{X} + b\dot{X} - mL\ddot{\phi} = u \quad (8)$$

3.2 State space representation of the system

From equation 7 and 8, we get the state space representation of the system as below:

$$\begin{bmatrix} \dot{X} \\ \ddot{X} \\ \dot{\phi} \\ \ddot{\phi} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & \frac{-(I+ML^2)b}{I(M+m)+MmL^2} & \frac{m^2gL^2}{I(M+m)+MmL^2} & 0 \\ 0 & 0 & 0 & 1 \\ 0 & \frac{-mLb}{I(M+m)+MmL^2} & \frac{mgL(M+m)}{I(M+m)+MmL^2} & 0 \end{bmatrix} \begin{bmatrix} X \\ \dot{X} \\ \phi \\ \dot{\phi} \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{I+mL^2}{I(M+m)+MmL^2} \\ 0 \\ \frac{mL}{I(M+m)+MmL^2} \end{bmatrix} u \quad (9)$$

$$y = \begin{bmatrix} 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} X \\ \dot{X} \\ \phi \\ \dot{\phi} \end{bmatrix} \quad (10)$$

4 LQ tracker formulation

If we have a LTI system like equation 11 and 12 and if we define the cost function as equation 13, then the control input of the system would be like equation 14. But to get this solution we need to solve the two differential riccati equation given in equations 15 and 16 [3]. The block diagram of the system is shown in figure 3. The two DREs are solved simultaneously using Runge–Kutta method [2]. The formula needed to use Runge–Kutta methods is given in equation 17.

$$\dot{X} = AX + Bu \quad (11)$$

$$y = CX \quad (12)$$

Cost function:

$$J = \frac{1}{2}(y(t_f) - r(t_f))^T P(t_f)(y(t_f) - r(t_f)) + \frac{1}{2} \int_{t_0}^{t_f} [(y - r)^T Q(y - r) + u^T R u] dt \quad (13)$$

here, $P \geq 0, Q \geq 0, R > 0$ and all are symmetric.

Solution:

$$u = -kX + R^{-1}B^T \nu \quad (14)$$

here, $k(t) = R^{-1}B^T S(t)$

DREs:

$$-\dot{S} = A^T S + SA - SBR^{-1}B^T S + C^T Q C; S(t_f) = C^T P(t_f) C \quad (15)$$

$$-\dot{\nu} = (A - Bk)^T \nu + C^T Q r; \nu(t_f) = C^T P(t_f) r(t_f) \quad (16)$$

here, $r(t)$ is the reference signal and given over time range $[t_0, t_f]$.

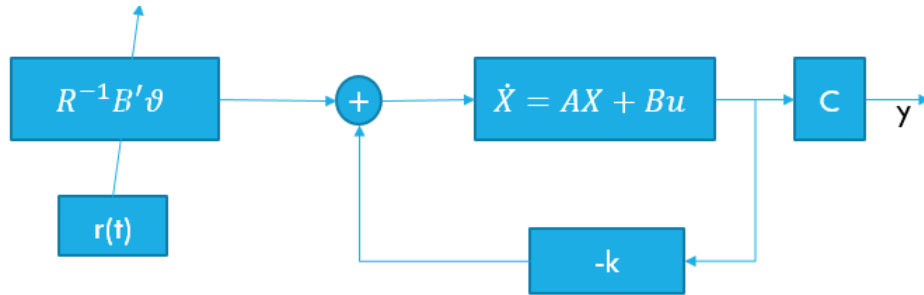


Figure 3: Block diagram of the system with LQ tracker

Equations for Runge–Kutta method:

$$\begin{aligned}
\dot{y} &= f(t, y), \\
y(t_0) &= y_0, \\
y_{n+1} &= y_n + \frac{dt}{6}(k_1 + 2k_2 + 2k_3 + k_4), \\
t_{n+1} &= t_n + dt, \\
k_1 &= f(t_n, y_n), \\
k_2 &= f(t_n + \frac{dt}{2}, y_n + dt\frac{k_1}{2}), \\
k_3 &= f(t_n + \frac{dt}{2}, y_n + dt\frac{k_2}{2}), \\
k_4 &= f(t_n + dt, y_n + dtk_3).
\end{aligned} \tag{17}$$

5 Simulation

Matlab programming language is used for the simulation of the system. The parameters of the simulation is given in table 1. Reference tracking signals are shown in figure 4 and 7. Corresponding output of the system are shown in figure 5 and 8 and the absolute error in tracking are shown in figure 6 and 9 respectively.

Table 1: Simulation parameters

Name	Value
M	0.5 <i>kg</i>
m	0.2 <i>kg</i>
b	0.1
I	0.006 <i>kg.m²</i>
g	9.8 <i>m/s²</i>
L	0.3 <i>m</i>

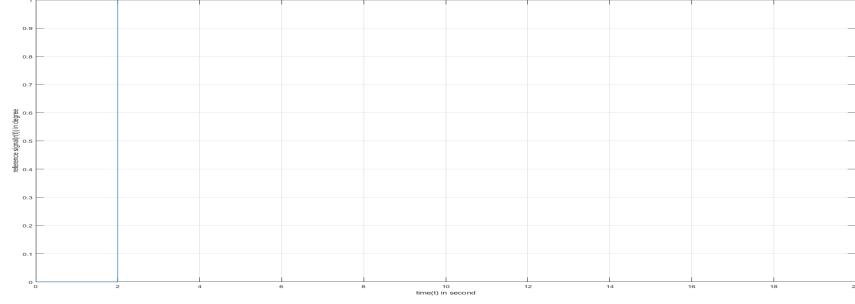


Figure 4: Step reference signal

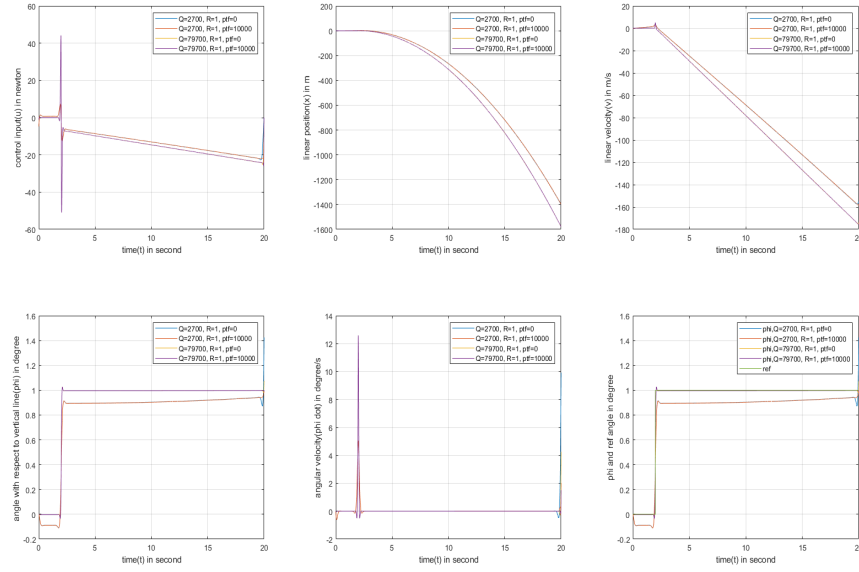


Figure 5: Output of the system for step reference signal

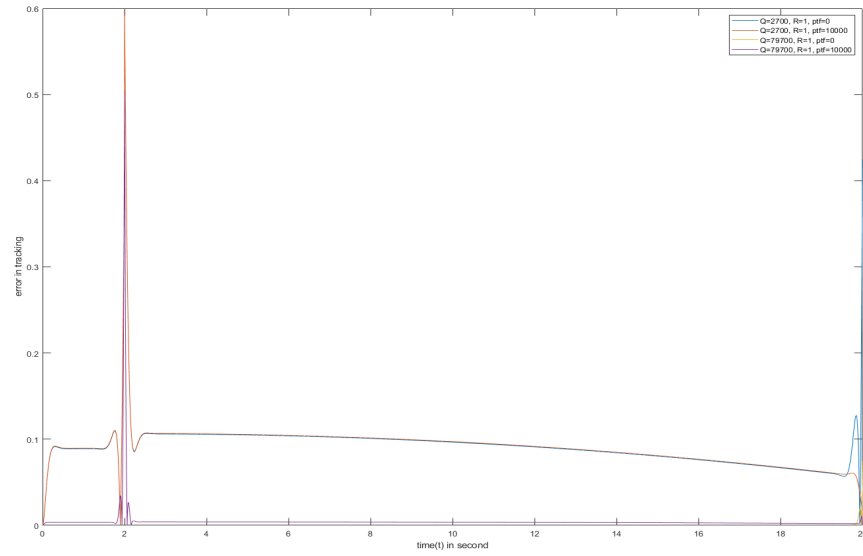


Figure 6: Absolute error in tracking for step reference signal

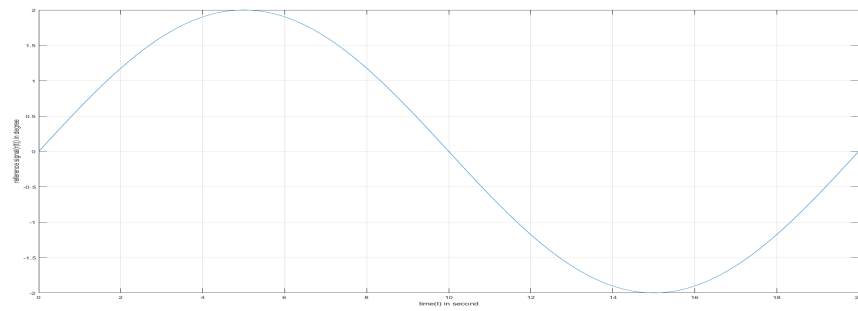


Figure 7: Sine reference signal

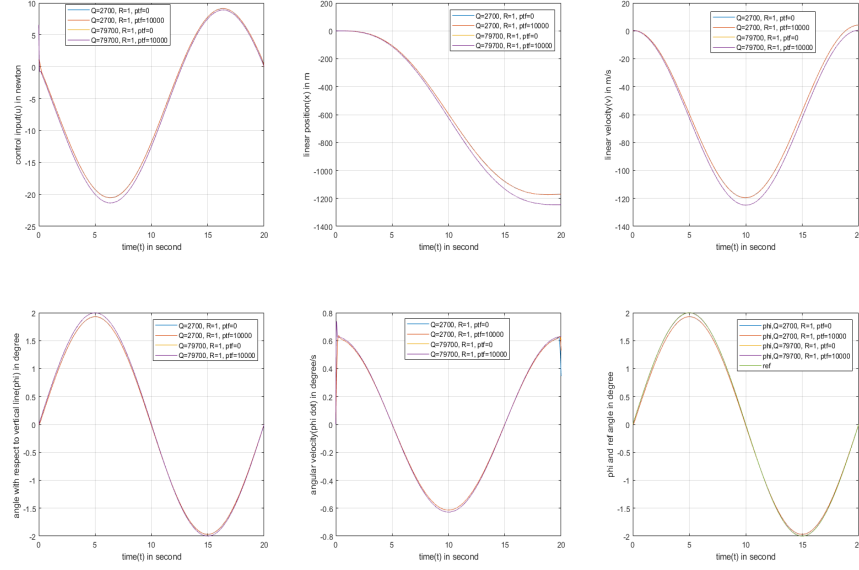


Figure 8: Output of the system for sine reference signal

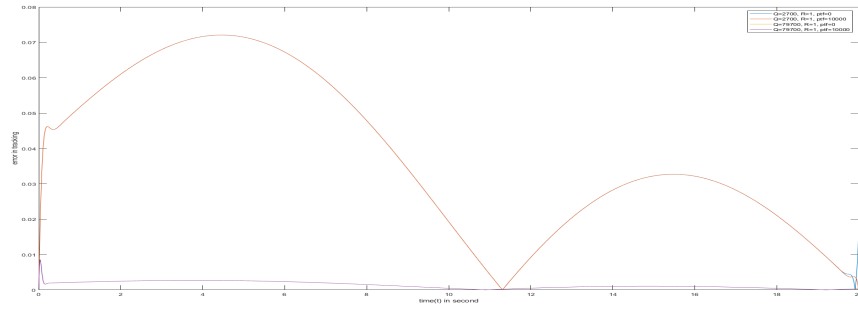


Figure 9: Absolute error in tracking for sine reference signal

6 Conclusion

1. In this project LQ tracker is developed and simulated with two reference signal.
2. The two DRE is solved simultaneously using Runge–Kutta method.
3. The control input is kept in reasonable value.
4. The over all absolute error of the tracker is less than 0.1

A

Matlab Code

```

1 %% author: mrinmoy sarkar
2 % email: msarkar@aggies.ncat.edu
3
4 clear all;
5 close all;
6
7 t0 = 0;
8 tf = 20;
9 dt = 0.001;
10 t = t0:dt:tf;
11
12 no_of_state = 4;
13 no_of_input = 1;
14
15
16
17 M = .5;
18 m = 0.2;
19 b = 0.1;
20 i = 0.006;
21 g = 9.8;
22 l = 0.3;
23
24 p = i*(M+m)+M*m*l^2; %denominator for the A and B matrices
25 A = [0      1      0      0;
26      0 -(i+m*l^2)*b/p (m^2*g*l^2)/p  0;
27      0      0      0      1;

```

```

28      0 -(m*l*b)/p          m*g*l*(M+m)/p  0];
29 B = [      0;
30      (i+m*l^2)/p;
31      0;
32      m*l/p];
33 C = [0 0 1 0];
34 x0 = [0 0 0 0]';
35
36 QQ = [2700,79700];
37 pptf = [0, 10000];
38
39 ref1 = zeros(length(t),1);
40 refdummy = 2*sin(2*pi*t/20);
41 % for sinusoidal input
42 ref1(:,1) = refdummy;
43
44 % for step input
45 %ref1(:,1) = 1;
46 %ref1(1:2000,1)=0;
47
48
49 for qq=1:2
50     for pp=1:2
51
52         Q = QQ(qq);%79700;
53         R = 1;
54         ptf = pptf(pp);% 10000;
55
56
57
58
59         ref1 = flipud(ref1);
60         ref = (ref1(1,:))';
61
62
63
64         S = zeros(no_of_state,no_of_state);
65         nu = zeros(no_of_state,1);
66
67

```

```

68
69     all_nu = zeros(length(t), size(nu,1)*size(nu,2));
70     all_nu(1,:) = C'*ptf*ref;
71
72     all_s = zeros(length(t), size(S,1)*size(S,2));
73     stf = C'*ptf*C;
74     all_s(1,:) = stf(:);
75     all_k = zeros(length(t), no_of_state*no_of_input);
76     K = (R^(-1))*B'*stf;
77     all_k(1,:) = K(:);
78
79     for i=2:length(t)
80         S0 = reshape(all_s(i-1,:), size(S));
81         S = S0;
82         S_dot = A'*S + S*A - S*B*(R^(-1))*B'*S + C'*Q*C;
83         k1 = dt*S_dot;
84
85         S = S0 + k1./2;
86         S_dot = A'*S + S*A - S*B*(R^(-1))*B'*S + C'*Q*C;
87         k2 = dt*S_dot;
88
89         S = S0 + k2./2;
90         S_dot = A'*S + S*A - S*B*(R^(-1))*B'*S + C'*Q*C;
91         k3 = dt*S_dot;
92
93         S = S0 + k3;
94         S_dot = A'*S + S*A - S*B*(R^(-1))*B'*S + C'*Q*C;
95         k4 = dt*S_dot;
96
97         S = S0 + k1./6 + k2./3 + k3./3 + k4./6;
98
99         all_s(i,:) = S(:);
100
101         K = (R^(-1))*B'*S;
102         all_k(i,:) = K(:);
103
104         nu0 = reshape(all_nu(i-1,:), size(nu));
105         nu = nu0;
106         ref = (ref1(i,:))';
107         nu_dot = (A-B*K)'*nu + C'*Q*ref;

```

```

108         k1 = dt*nu_dot;
109
110         nu = nu0 + k1./2;
111         nu_dot = (A-B*K)'*nu + C'*Q*ref;
112         k2 = dt*nu_dot;
113
114         nu = nu0 + k2./2;
115         nu_dot = (A-B*K)'*nu + C'*Q*ref;
116         k3 = dt*nu_dot;
117
118         nu = nu0 + k3;
119         nu_dot = (A-B*K)'*nu + C'*Q*ref;
120         k4 = dt*nu_dot;
121
122         nu = nu0 + k1./6 + k2./3 + k3./3 + k4./6;
123
124         all_nu(i,:) = nu(:);
125     end
126
127     all_k = flipud(all_k);
128     all_nu = flipud(all_nu);
129     all_s = flipud(all_s);
130     ref1 = flipud(ref1);
131
132     u = zeros(length(t), no_of_input);
133     x = zeros(length(t), no_of_state);
134     x(1,:) = x0';
135     F = 1;
136     u(1,:) = -(reshape(all_k(1,:), size(K)))*(x(1,:))' +
        F*(R^(-1))*B'*reshape(all_nu(1,:), size(nu));
137     for i=2:length(t)
138
139         xx0 = (reshape(x(i-1,:), size(x0)));
140         xx = xx0;
141         xx_dot = A*xx + B*(u(i-1,:))';
142         k1 = dt*xx_dot;
143
144         xx = xx0 + k1./2;
145         xx_dot = A*xx + B*(u(i-1,:))';
146         k2 = dt*xx_dot;

```



```

147
148     xx = xx0 + k2./2;
149     xx_dot = A*xx + B*(u(i-1,:))';
150     k3 = dt*xx_dot;
151
152     xx = xx0 + k3;
153     xx_dot = A*xx + B*(u(i-1,:))';
154     k4 = dt*xx_dot;
155
156     xx = xx0 + k1./6 + k2./3 + k3./3 + k4./6;
157
158     x(i,:) = xx(:);
159     %F = pinv(C*pinv(-A+B*(reshape(all_k(i,:),size(K
160     u(i,:) = -(reshape(all_k(i,:),size(K)))*(x(i,:))
161     ' + F*(R^(-1))*B'*reshape(all_nu(i,:),size(nu
162     ));
163
164 end
165 figure(1)
166 plot(t,ref1)
167 xlabel('time(t) in second')
168 ylabel('reference signal(r(t)) in degree')
169 grid on
170 figure(2)
171 %clf
172 subplot(231)
173 plot(t,u)
174 hold on
175 xlabel('time(t) in second')
176 ylabel('control input(u) in newton')
177 grid on
178 subplot(232)
179 plot(t,x(:,1))
180 hold on
181 xlabel('time(t) in second')
182 ylabel('linear position(x) in m')
183 grid on
184 subplot(233)
185 plot(t,x(:,2))
186 hold on

```

```

184         xlabel('time(t) in second')
185         ylabel('linear velocity(v) in m/s')
186         grid on
187         subplot(234)
188         plot(t,x(:,3))
189         hold on
190         xlabel('time(t) in second')
191         ylabel('angle with respect to vertical line(phi) in
            degree ')
192         grid on
193         subplot(235)
194         plot(t,x(:,4))
195         hold on
196         xlabel('time(t) in second')
197         ylabel('angular velocity(phi dot) in degree/s')
198         grid on
199         subplot(236)
200         plot(t,x(:,3))
201         xlabel('time(t) in second')
202         ylabel('phi and ref angle in degree')
203         hold on
204         %plot(t,ref1)
205         %hold on
206         %legend('phi','ref')
207         grid on
208         figure(3)
209         plot(t,abs(x(:,3))-ref1))
210         xlabel('time(t) in second')
211         ylabel('error in tracking')
212         hold on
213     end
214 end
215
216 figure(2)
217 subplot(231)
218 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q
    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')
219
220 subplot(232)
221 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q

```

```

    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')
222 subplot(233)
223 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q
    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')
224 subplot(234)
225 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q
    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')
226 subplot(235)
227 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q
    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')
228 subplot(236)
229 plot(t, ref1)
230 legend('phi,Q=2700, R=1, ptf=0', 'phi,Q=2700, R=1, ptf=10000
    ', 'phi,Q=79700, R=1, ptf=0', 'phi,Q=79700, R=1, ptf
    =10000', 'ref')
231 figure(3)
232 legend('Q=2700, R=1, ptf=0', 'Q=2700, R=1, ptf=10000', 'Q
    =79700, R=1, ptf=0', 'Q=79700, R=1, ptf=10000')

```

References

- [1] https://www.ee.usyd.edu.au/tutorials_online/matlab/examples/pend/invpen.html.
- [2] https://en.wikipedia.org/wiki/Runge%E2%80%93Kutta_methods.
- [3] Ecen-865 lecture notes.