

ACTAIR CLASS

1) WHAT IS A SYSTEM?

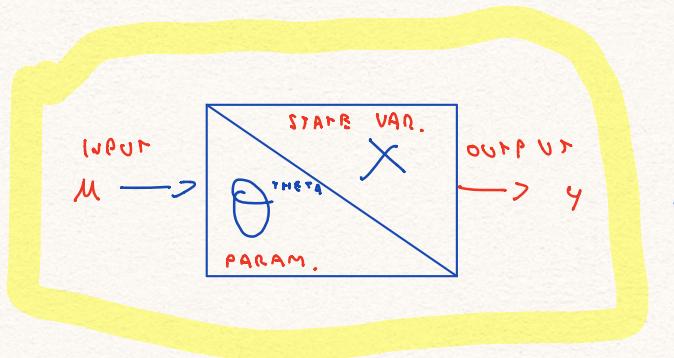
• SPREAD OF A VIRUS

• DEVICE

• EN. CONSUMPTION

RECC--

THEY HAVE IN COMMON TO HAVE A MATH DESCRIPTION



AN EXAMPLE: A PENDULUM
WITH $G(m,L)$ AND $X(\theta)$
CAN WE PREDICT THE MASS
SPEED?
NO! WE NEED TO KNOW
THE WHOLE STATE VECTOR $X(\theta, \dot{\theta})$ → if we know the speed $\dot{\theta}$, we can predict the
SPEED OF THE MASS!

JUST LIKE THIS, WE CAN SAY THE SAME THING FOR A CAR.

~ ~ ~ ~ ~ ~ ~ ~ ~ ~
THIS KIND OF REPRESENTATION OF A SYSTEM COULD BE DIFFERENT
ACCORDING TO WHAT I'M STUDYING, SO I CAN CLASSIFY SYSTEMS
BY THEIR CHARACTERISTIC

TIME INVARIANT: PARAMETERS ARE CONSTANT OVER TIME

PARAMETERS ↗
TIME VARIANT: PARAMETERS ARE NOT CONSTANT OVER TIME
(SHUTTLE CONSUMING FUEL)

LINEAR: MATH. FUNCTION IS A LINEAR COMBINATION OF STATE VARIABLES

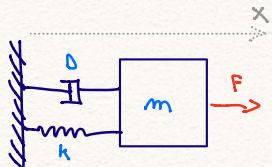
STATE ↗
NON LINEAR: IF THE SYS. IS DESCRIBED BY NONLINEAR EQUATION

INPUT/OUTPUT ↗
• SISO
• SIMO
• MISO
• MIMO

TIME : STATE VARIANCES CHANGES CONTINUITY OVER TIME

DISCRETE TIME: STATE VARIABLES CHANGE BY DISCRETE QUANTITIES ONLY AT DISCRETE SET OF POINT IN TIME.

AN EXAMPLE: SPRING / DUMPER MASS OSCILLATION



$$\text{MATH.} \quad m \ddot{x}(k) + b \dot{x}(k) + k x(k) = F(k)$$

EQ:

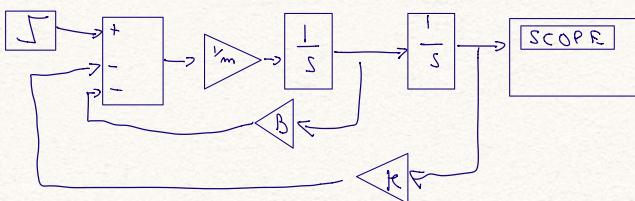
WE CAN MODEL THIS IN DIFF. WAYS

SCRIPT (QD)

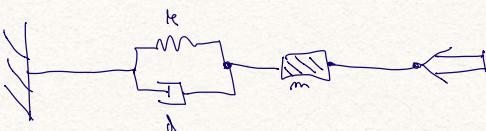
A vertical column of handwritten blue ink marks on white paper. The marks consist of approximately 15 horizontal dashes of varying lengths, arranged in a roughly rectangular grid pattern. The dashes are roughly parallel and evenly spaced vertically.

SCRIPR.m

GRAPHICAL SIGNAC-BASED (10)



GRAPHICAL PHYSICAL BASED



medelic

WE CAN FORGET ABOUT MATH!

ALTAIR COMPOSE

ALTAIR ACTIVATION

usually we solve this kind of problem with ordinary DIFF. EQUATION

The diagram illustrates a state-space model structure. At the top, the word "ACTION" is written in blue. Below it, the state variable $x(t)$ is shown with a derivative chain: $\frac{dx(t)}{dt}$, $\frac{d^2 x(t)}{dt^2}$, ..., $\frac{d^n x(t)}{dt^n}$. To the right of this chain, the word "ORDER" is written vertically above a red box containing n . To the right of the derivatives, the word "INPUT" is written vertically above the term $u(t)$. To the right of $u(t)$, the word "PARAMETER" is written vertically above the term $\theta(t)$. A large orange bracket groups all terms from $x(t)$ down to $\theta(t)$. An arrow labeled "INOP. VARIABLE" points from the bottom left towards this bracket. At the bottom, a box contains two columns: "STATE" and "VARIABLE". Below the box, the text "X = STATE VECTOR" is written.

IMPLICIT FORMULAS

WHAT WE WILL NEED TO USE A SIGNAL-BASED APPROACH?

1) GRANULAR THE IMPLICIT FORMULATION

2) FIND IT'S EXPLICIT FORMULATION.

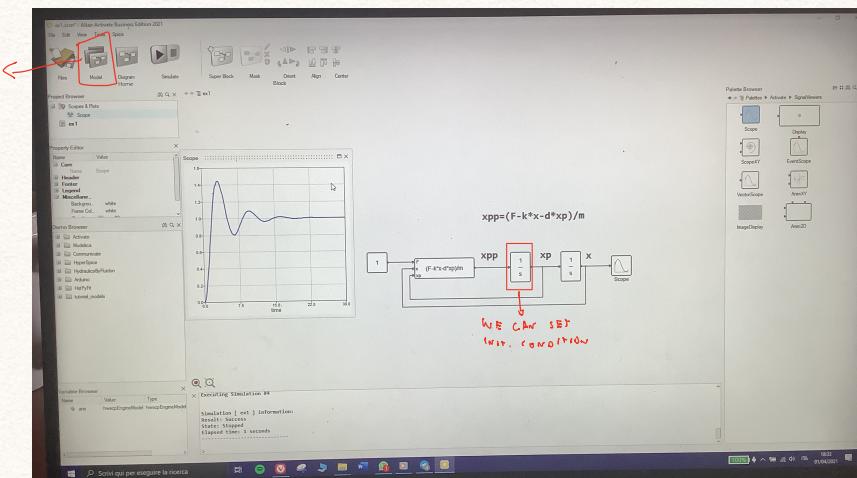
$$\frac{d^m X(t)}{dt^m} = f(t, X(t), \frac{d^2(t)}{t^2}, \dots, \frac{d^{n-1}(t)}{t^{n-1}}, u(t), \theta(t))$$

3) IDENTIFY THE ORDER
WE APPLY ALL
THE STEPS

$$\begin{aligned} 1) \quad m\ddot{x}(t) + d\dot{x}(t) + kx(t) - F &= 0 \\ 2) \quad \ddot{x}(t) &= \frac{1}{m} (F - d\dot{x}(t) - kx(t)) \\ 3) \quad 2^{\text{nd}} \text{ ORDER} \end{aligned}$$

NOW LET'S GO TO ACTIVE
AND DO THE FIRST EXERCISE : EX1.SCM

MODEL
PARAMETERS



NOW WE NEED TO FIND THE PHYSICAL BASED APPROACH AT LEAST REMEMBER THE ORDER

WE DO NOT FOLLOW EULERIAN, AND WE OPEN

MODELICA

it is useful in modeling to add all the initial condition,
THE NUMBER IS THE ORDER N'
FOR EXAMPLE: I SET TRUE TO POSITION AND SPEED OF THE MASS.
AND THAT IS OKAY!

THEN when I simulate, everything works!

AFTER I DID THIS, I SET THE INITIAL POSITION IN THE
SENSOR = 1
SET POSITION = FALSE :) → THIS POSITION IS IGNORED

THEIR ARE 2 CASE

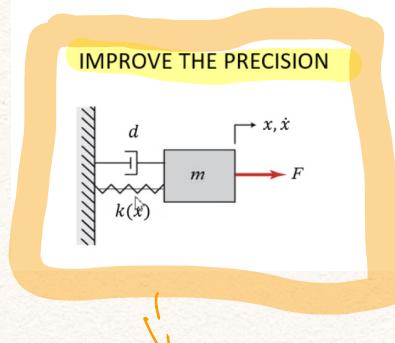
SET POSITION = TRUE :) → ERROR ! WE HAVE 2
INCONSISTENT STARTING POINT!!

PYASICAL APPROACH
HAS A LOT S OF
ADVANTAGES !!

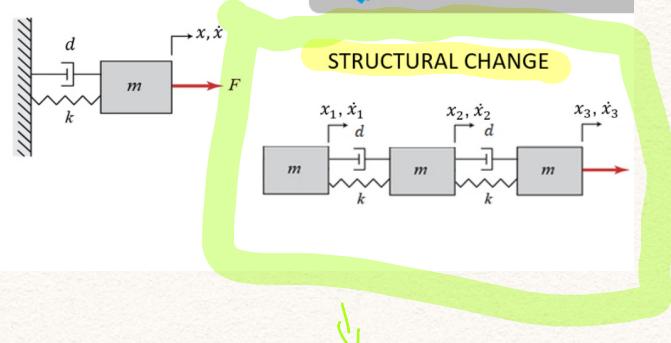
	Signal Approach	Physical Approach
Time required for modeling		X
Approach simplicity		X
Usability/sharing		X
Mistakes during modeling		X
System understanding	X	
Time for model simulation	X	

WHAT WE COULD DO NOW IS SOME
MODIFICATION AT OUR MODEL:

Precision and Structural



SIMPLER TO USE A SIGNAL-BASED



SIMPLER TO USE PHYSICAL-BASED

NOW GO ON

2. SYST. MODELLING / GENERAL-APPROACH. SCM

IMPROVE-PRECISION: SIGNAL BASED WE SIMPLY ADD $k(x) = k + x$
PHYS. BASED \rightarrow WE NEED TO CREATE A
NEW MODELICA MODEL

STRUCTURAL-CHANGE: PHYS. APPROACH WE JUST NEED CREAT-C
CREAT-V

LIN EAR TAEORY

WE HAVE AS ALWAYS A LINEAR SYSTEM



$$a_m \frac{d^m x(t)}{dt^m} + \dots + a_1 \frac{dx(t)}{dt} + a_0 x(t) =$$

↓
STATE VARIABLES

with $m < m$

$$= b_m \frac{d^m u(t)}{dt^m} + \dots + b_1 \frac{du(t)}{dt} + b_0 u(t)$$

if $m=m \rightarrow$ the sys is acting instaneously, and in real life
is not possible

WE KNOW THAT THE SOLUTION OF THIS LINEAR SYST. WILL BE LIKE THIS:

$$X(t) = \sum_{n=1}^m A_n e^{\lambda_n t} + \sum_{k=1}^{M-1} C_k e^{\lambda_k t} + B + X_f(t)$$

NATURAL RESPONSE

FORCED RESPONSE
USUALLY HIGHLIGHTED

NATURAL IS DUE TO THE RESPONSE IF NO INPUT IS ADDED
OTHER FORCED RESPONSE IS THE ONE THAT YOU CAN SEE WHEN WE
GIVE SOMETHING IN INPUT

WHY THOSE ARE IMPORTANT? THINK THAT FAR VAST MAJORITY OF
PROBLEMS ARE NON-LINEAR! :)

↳ 2 MAIN REASOSN

1) A NON-LINEAR SYS COULD BE LINEARIZED → SIMPLER FORM

LYAPUNOV
INDIRECT METHOD

NON-LINEAR SYSTEM IS VERY CLOSE TO A LINEAR ONE
LOCALLY AROUND AT ITS EQUILIBRIUM POINTS

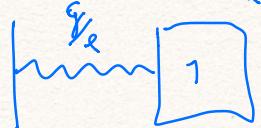
EXAMPLE: PENDULUM

$$\ddot{\alpha} + \frac{g}{l} \sin \alpha = 0$$

$$\ddot{\alpha} < 10^\circ$$

~LINEARIZATION~

$$\ddot{\alpha} + \frac{g}{l} \alpha = 0$$



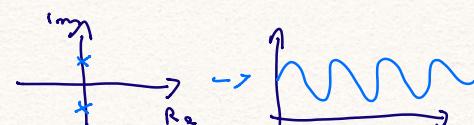
YOU CAN FIND IT IN COMPARISON. SCM

THE 2 SYST. ARE VERY SIMILAR! (if $\theta < 10^\circ$)

2) ALSO BECAUSE LINEAR SYST. ARE MUCH MORE SIMPLE AND

EASY TO UNDERSTAND THE NATURE OF THE SYSTEM! ! → MANY COOL PEOPLE
HAS STUDIED LINEAR MODELS BEFORE US

LINEARIZED; WE CAN FIND
THE ZERO AND POLES:



NOW, IF I HAVE A LTI SYSTEM, I CAN USE LAPLACE TRANSFORM

LTI EQUATIONS

$$\frac{d^m X(s)}{ds^m} + \sum_{k=1}^m a_k s^{m-k} X(s) = \sum_{k=0}^n b_k s^k U(s)$$

$$(a_m s^m + a_{m-1} s^{m-1} + \dots + a_1 s + a_0) X(s) = (b_m s^m + \dots + b_1 s + b_0) U(s)$$

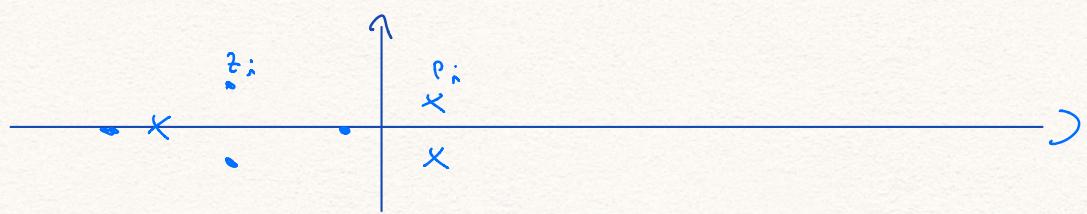
→ LAPLACE DOMAIN (FREQ. DOMAIN) → THOSE ARE ALGEBRAIC FORMULAS,

MUCH MORE EASIER TO DEAL WITH

ONCE WE DO THAT:

TRANSF. FUNCTION: $\frac{\text{OUTPUT}}{\text{INPUT}} = \frac{Y(s)}{U(s)} = \frac{b_m s^m + \dots + b_1 s + b_0}{a_m s^m + \dots + a_1 s + a_0}$ } ZEROS.
} POLES

WE CAN DRAW
ROOTS LOCUS



SOME PROPERTIES

POLES OF THE FCN AFFECTS THE STABILITY OF THE SYSTEM!
BECAUSE THEY ARE IN THE NATURAL RESPONSE OF THE SYS.

$$x(t) = \sum_{i=1}^n A_i e^{s_i t} + x_s(t)$$

POLES

AND FOR POLES AND IN THE DENOMINATOR OF THE T.F. →

Natural Response – Stability : Poles

Transfer function

$s_i = \sigma$

$s_{1,2} = \sigma \pm j\omega$

$Ae^{\sigma t}$ Pure Real

$A(\cos \omega t + \varphi)$ Pure Imaginary

$Ae^{\sigma t} (\cos \omega t + \varphi)$ Complex conjugate

$Ae^{-|sigma|t}$

ASYMPTOTIC STABILITY

$A(\cos \omega t + \varphi)$

MARGINAL STABILITY

$Ae^{+|sigma|t}$

INSTABILITY

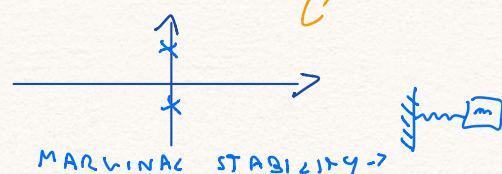
IN OUR CASE, WITH OUR PROBLEM?

$$\ddot{x} + \frac{g}{L} x = 0 \rightarrow \text{LAPLACE} \rightarrow s^2 A(s) + \frac{g}{L} A(s) = 0$$

$$\left(s^2 + \frac{g}{L} \right) A(s) = 0 \quad \Rightarrow \quad s = \pm \sqrt{-\frac{g}{L}}$$

(1) $\sqrt{-\frac{g}{L}}$

(2) $-j\sqrt{\frac{g}{L}}$



PROPERTIES OF FORCED RESPONSE

$$x(t) = \sum_{i=1}^n A_i e^{N_i t} + B + \boxed{x_*(t)} \quad \sim \quad x_*(t) = x_*(\mu_i, (\varepsilon_i), t)$$

FORCED

when we have an input $u(t)$, we can split it in a FOURIER SERIES:

$$u(t) = \sum_{i=1}^m B_i (\cos[\Omega_i t + \varphi_i])$$

THEN WE CAN USE SUPERPOSITION PRINCIPLE AND SIMILARITY METHOD

$$x_*(t) = \sum_{i=1}^m C_i [\cos(\Omega_i t + \varphi_i + \psi_i)] \quad \sim \quad \text{IT IS AGAIN A SUM OF COSIN, WITH THE SAME FREQUENCIES, BUT WITH A PHASE SHIFT } \psi_i$$

THEN THIS TRICK IS TO CONSIDER ONLY THE FORCED RESPONSE, AND SO THE ONLY IMPORTANT PART OF OUR STATE VARIABLE S , IS THE IMR PART: $S = j\Omega$

if we use only $j\Omega$ in place of S , we will obtain THE INFORMATION OF OUR SYSTEM AT REGIME CONDITION!
 L> TF is telling info about the forced response

$$\frac{X(j)}{U(j)} = \frac{X(j\Omega)}{U(j\Omega)} = \frac{b_m (j\Omega)^m + \dots + b_1 j\Omega + b_0}{a_m (j\Omega)^m + \dots + a_1 j\Omega + a_0} = \boxed{\text{CONSTANT COMPLEX NUMBER}}$$

so, for a given $j\Omega^*$, the ratio between $\frac{y}{u}$ is a constant, so it is INDEPENDENT BY THE INPUT AMPLITUDE!!

This means that we can write the BODE PLOT! $\frac{y}{u} = |TF| e^{j\omega}$

- LITTLE NB CAP:
- 1) T.F. gives us the TRANSIENT & STATIONARY behaviour of the system
 - 2) STABILITY DEPENDS ONLY ON POLES p_i
 - 3) POLES p_i + zeros τ_i are linked together and affect the FORCED RESPONSE
 - 4) FOR LTI system we can draw the BODE plot, which represents the frequency response of the system in REAL TIME CONDITION

STATE SPACE FORMULATION OF A SYSTEM

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) \\ y(t) = cx(t) + du(t) \end{cases}$$

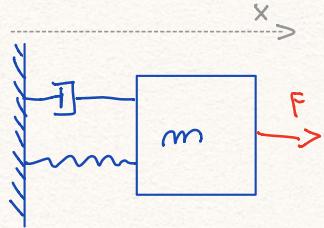
WE GO BACK TO THE MASS + SPRING EXAMPLE

$\cdot A \rightarrow$ STATE MATRIX

$\cdot C \rightarrow$ OUTPUT MATRIX

$\cdot B \rightarrow$ INPUT MATRIX

$\cdot D \rightarrow$ FREE FORWARD MATRIX



INPUT FORMULATION:

$$m\ddot{x}(t) + d\dot{x}(t) + kx(t) - F = 0$$

$$x(s) = \frac{F(s)}{ms^2 + ds + k}$$

EXPLICIT FORMULATION

$$\begin{cases} \dot{x}(t) = -\frac{d}{m}\dot{x}(t) - \frac{k}{m}x(t) + \frac{F}{m} \\ \dot{x}(t) = \ddot{x}(t) \rightarrow \text{WE HAVE AN IDENTITY: NICE TRICK} \end{cases}$$

another trick: we use $\bar{x}(t) = \begin{bmatrix} \dot{x} \\ x \end{bmatrix} \rightarrow$ the new state vector

$$\begin{cases} \dot{\bar{x}}(t) = \begin{bmatrix} [A] \\ [-k/m \quad -b/m] \end{bmatrix} \bar{x} + \begin{bmatrix} [B] \\ [0] \end{bmatrix} F \end{cases}$$

SIMILARITIES BETWEEN

THIS STATE SPACE

FORMULATION \rightarrow AND TRANSF. \rightarrow FUNCTION

$$\frac{y(s)}{u(s)} = \frac{x(s)}{F(s)} = \frac{1}{ms^2 + bs + k}$$

EIGENVALUES OF $[A]$
(AUTOVALORI)

$$\det(A - nI) = \det \begin{pmatrix} [-b/m - n] & [-k/m] \\ 1 & -n \end{pmatrix} =$$

$$= n^2 + n \frac{b}{m} + \frac{k}{m} = \boxed{m \cdot n^2 + d \cdot n + k = 0}$$

$n = s$

\Rightarrow the eigenvalues of A , are the poles of our system

we can do the Laplace Transformation of our state space system:

$$\begin{cases} \dot{x}(t) = Ax(t) + Bu(t) \\ y(t) = cx(t) + du(t) \end{cases} \xrightarrow{\mathcal{L}} \begin{cases} s \cdot x(s) = ax(s) + bu(s) \\ y(s) = cx(s) + du(s) \end{cases} \xrightarrow{sI - A} \begin{cases} (sI - A)x(s) = bu(s) \\ y(s) = cx(s) + du(s) \end{cases}$$

$$= \begin{aligned} x(s) &= [sI - A]^{-1} \cdot b \cdot u(s) \\ y(s) &= (c \cdot [sI - A]^{-1} \cdot b + d) \cdot u(s) \end{aligned} \quad \boxed{\frac{y(s)}{u(s)} = [c \cdot [sI - A]^{-1} + d]} = \begin{bmatrix} TF_{11} & TF_{12} & \dots & TF_{1n} \\ TF_{21} & TF_{22} & \dots & TF_{2n} \\ \vdots & \vdots & \ddots & \vdots \\ TF_{n1} & TF_{n2} & \dots & TF_{nn} \end{bmatrix}$$

THIS MEANS THAT IT IS VERY SIMPLE TO GO FROM

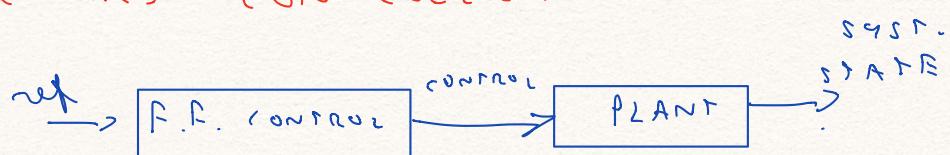
STATE SPACE $\xrightarrow{\text{TO}}$ TRANSFER FUNCTION AND VICE VERSA

IN TF, SS, SCM YOU CAN FIND SOME EXAMPLE OF HOW THE MODEL CAN BE REPRESENTED:

- 1) initial state set
- 2) initial output set
- 3) STABLE STATE \rightarrow WE START IN ORDER THAT EVERYTHING IS IN EQ

A BIT OF CONTROL THEORY

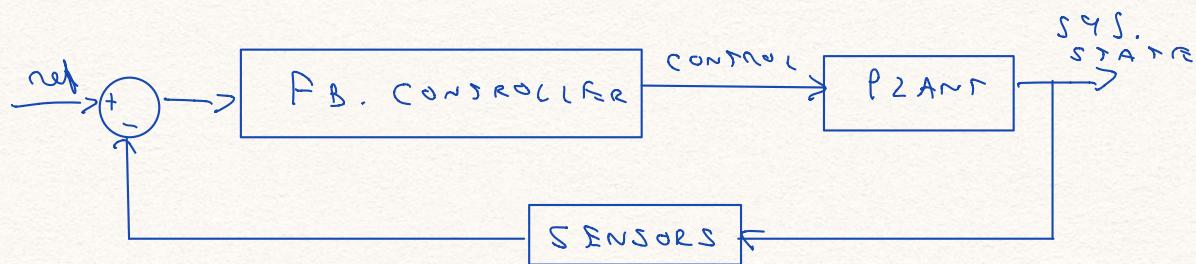
1) FB FORWARD CONTROLLER



- VERY CHRAP ...
- NO FEEDBACK \rightarrow NO REACTION TO DISTURBANCE

(YOU CAN SEE IT IN FFCONTROLLER.SCM)

2) FEED BACK CONTR. OCCURS



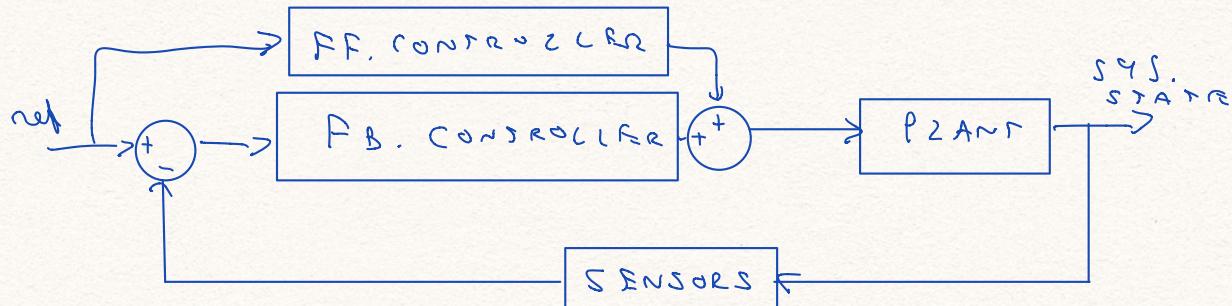
+ ROBUST

- W/B NEEDED SENSORS

(YOU CAN SEE IT IN FB CONTROLLER)

3) FEEDBACK + FEED FORWARD

TAB BEST MEASURES! AND ALSO FASTER



• FF USED TO FOLLOW TAB PER

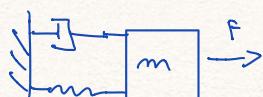
• FB. USED TO COMPENSATE DISTURBANCE

(FB+FF controller . SCM)

PID CONTROLLER

p) PROPORTIONAL ACTION

in our system mass - damper



$$m\ddot{x} + b\dot{x} + kx = F_c = K_p(x_{ref} - x) \rightarrow \text{FORCE PROPORTIONAL TO THE ERROR (FEED FORWARD)}$$

$$m\ddot{x} + b\dot{x} + (k + K_p)x = K_p x_{ref}$$

so now, a proportional controller is changing the stiffness of the system ...

WHAT DOES HAPPEN IN REGIME CONDITIONS?

~~$$m\ddot{x} + b\dot{x} + (k + K_p)x = K_p x_{ref}$$~~

$$x = \frac{K_p}{k + K_p} x_{ref} \rightarrow x \approx x_{ref} \text{ if } K_p \rightarrow \infty$$

P, D) PROPORTIONAL AND DERIVATIVES:

$$m\ddot{x} + b\dot{x} + kx = F_c = K_p(x_{ref} - x) + K_d(\dot{x}_{ref} - \dot{x})$$

$$m\ddot{x} + (b + K_d)\dot{x} + (k + K_p)x = K_p x_{ref} + K_d \dot{x}_{ref}$$

here we also change the equivalent damping!!
in steady state:

$$(k + K_p)x = K_p x_{ref} \rightarrow x = \frac{K_p}{k + K_p} x_{ref}$$

again, we will follow the reference only
if $K_p \rightarrow \infty$

P, D, I) we will also the INTEGRAL:

here we change the variable: $y = \int_0^t x dt$

$$m\ddot{x} + b\dot{x} + kx = F_c = K_p(x_{ref} - x) + K_d(\dot{x}_{ref} - \dot{x}) + K_i \left(\int_0^t (x_{ref} - x) dt \right)$$

$$m\ddot{y} + (b + K_d)\dot{y} + (k + K_p)y = K_d \ddot{y}_{ref} + K_p \dot{y}_{ref} + K_i y_{ref}$$

The effect is that we have increased the order of the system !!

IN STABLE STATE:

$$\cancel{m\ddot{y} + (b+k_d)\dot{y} + (k+k_p)\dot{y} + k_i y = k_d \dot{y}_{ref} + k_p \dot{y}_{ref} + k_i y_{ref}}$$

$$k_i y = k_i y_{ref} \rightarrow \boxed{y = y_{ref}}$$

WE SEE A STABLE POSITION,

BUT WE COULD MAKE OUR SYSTEM UNSTABLE

SO WHY PDO AND USEFUL??

1) INCREASE PERC. OF THE SYSTEM

2) MOVE THE ROOT LOCUS, SO WE CAN MAKE OUR SYSTEM STABLE!

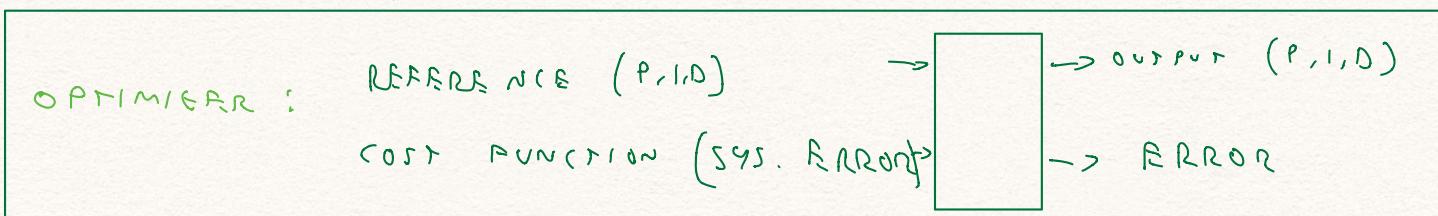
3) CAN USE THE MAXIMUM ERROR

• ALSO CAN SEE THAT FOR A HIGH k_p , THE ACTUATOR COULD GO TO SATURATION, AND THUS WE CAN HAVE CLAMPING

AUTOMATIC GAIN OPTIMIZER!!

OPEN TUTOR-OPT.SCM

In the next hour we will see the optimizer for the suspension



There are many ways to fix the cost function in our case!

if($e > 0$)

$$\text{cost} = e^2$$

else

$$\text{cost} = (100 \cdot e)^2$$

↓ PENALTY, BECAUSE WE WANT TO AVOID THIS CONDITION! When $e < 0$ we are

having an OVERSHOOT

$$\text{cost} = \int_0^k \text{cost} dt$$

↓ THIS IS BECAUSE WE NEED TO EVALUATE ALONG ALL THE TIME!

CONSIDER THE OPTIMIZATOR:

- WE CAN SET UP AND DOWN BOUNDS $\begin{bmatrix} 0 \\ 20, 20, 20 \end{bmatrix}$ ^{LOWER} _{UPPER}
- INITIAL AND FINAL VALUE OF THE TRUST REGION
- MAX ITERATION

NOTE: NOW PARAMETERS P, I, D ARE NOT DEFINED IN THE MODEL

RUNNING TIME THAT OPTIMIZER WORKS, WE WILL

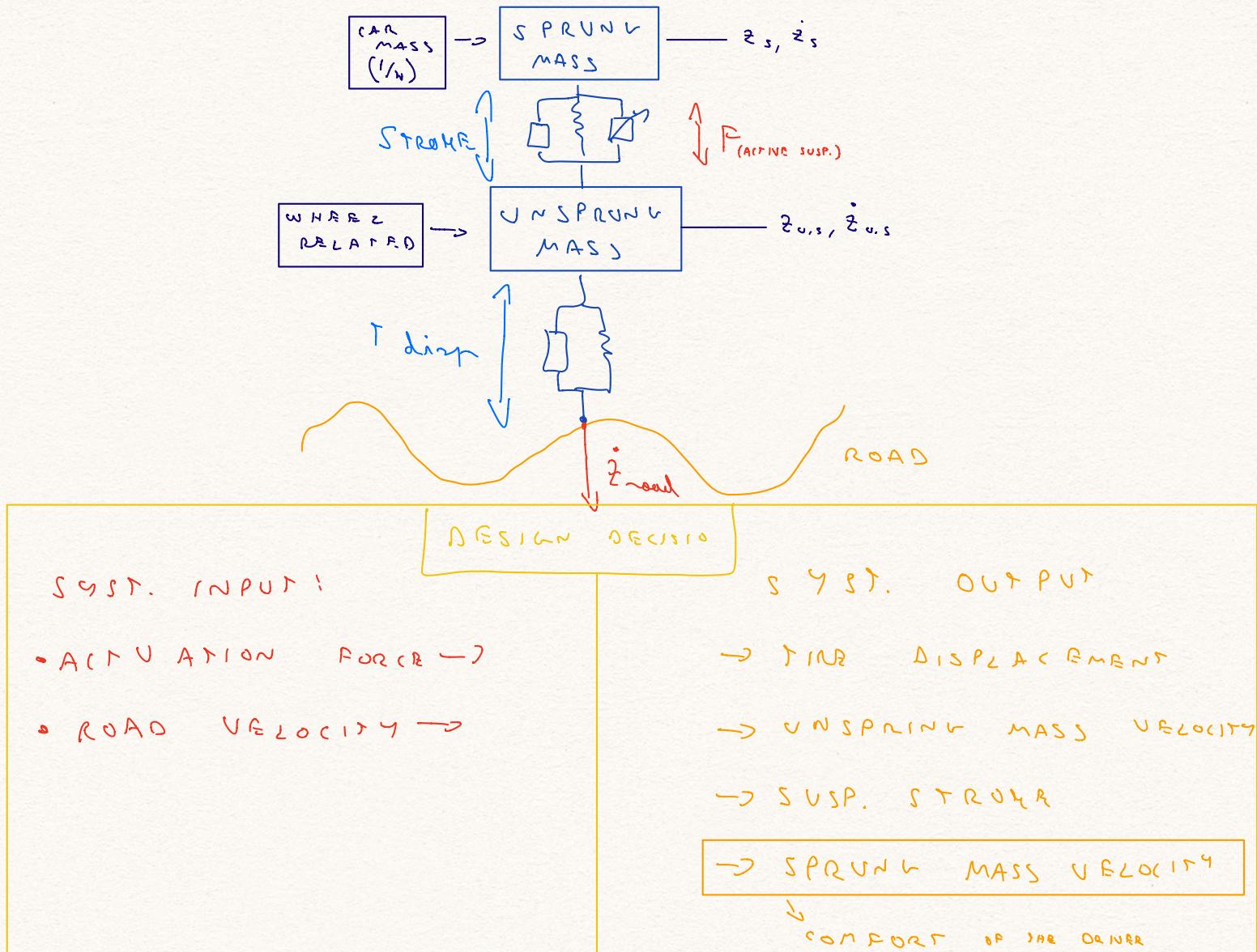
SET NEW PID IN THE BASE, AND THE NEXT RUN

$$\begin{bmatrix} P \\ I \\ D \end{bmatrix} = \text{GET FROM BASE} \left(\begin{bmatrix} P \\ I \\ D \end{bmatrix}, 10 \right)$$

↓ if they are not in base, get 10 (FIRST RUN)

↳ ACTIVE SUSPENSION

- based on the Quater car suspension



WE START NO MODEL YET

SYSTEM IN ALTAIR ACTUATOR

Ex 2

1) we create the model with mass ... etc

TIP:

VARIABLES IN "MODEL" → GLOBAL

VARIABLES IN "DIAGRAM HOME" → LOCAL INSIDE SUPERBLOCK

2) ADD A SIGNAL GEN + LPF

$$LPF = \frac{1}{\omega \cdot \zeta + 1} = \frac{1}{\frac{\omega}{100} + 1} \rightarrow \text{FREQUENCIES AT } \omega = \frac{\omega}{2\pi} = \frac{100}{2\pi} \approx 16 \text{ Hz}$$

if we increase the SPRUNG DAMPING \rightarrow we are ADDING RIGIDITY to THE SYSTEM, so THE ACCELERATION WILL BE HIGHER
 \hookrightarrow SUSPENSION WILL NOT DEFOR MATE TOO MUCH

NOW, IF YOU LOOK PROF'S FILE
 ACTIVS_SUSP._PASSIVE.SCM

This is a 2nd order 2 DOF SYSTEM EACH.

\hookrightarrow SO WE HAVE 2 PEEONES IN THE BODE PLOT

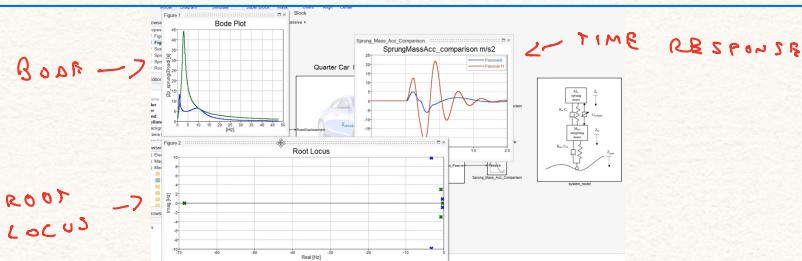
\rightarrow WE CAN ALSO SEE 4 ROOTS ON THE ROOT LOCUS (2 DOF, each 2nd order)

\rightarrow THIS CAN BE DONE BECAUSE WE CAN LINEARIZE AUTOMATICALLY WITH MODELICA!

NOW, WE INCREASE AGAIN SPRUNG DAMPING!

WHAT HAPPEN?

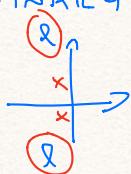
- * IN DOFS Σ WILL HAVE ONLY ONE HIGHER PEAK,
- * THE TIME RESPONSE WILL BE WORSE
- * IN THE ROOT LOCUS, 2 roots have Imag part, and the other 2 have only real \rightarrow this is why we have only ONE PEAK IN BODE



3rd ATTEMPT \rightarrow WE DECREASE THE SPRUNG DUMPING

\hookrightarrow we are making softer DAMPERS! WHAT WILL HAPPEN?

- we are getting less oscillations \rightarrow BETTER
- \hookrightarrow BUT THE SYSTEM WILL HAVE VERY LITTLE OSCILLATIONS
- IN BODY PLOT we will have high spike in lower frequencies (BUT ALSO A SMALLER ONE IN HIGHER FREQ)
- in root locus I will see 2 roots with very high IMAGINARY part near to real (less damped behavior)



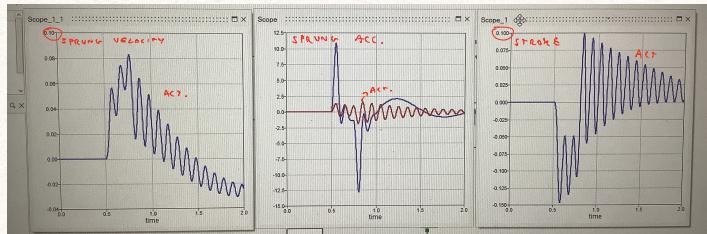
3) NOW WE ADD ACTIVE SUSPENSION!!

WE ADD ACTION-REACTION FORCES AND WE ADD A CONTROLLER!

WE MAKE A SORT OF FEEDBACK LOOP WHERE THE REFERENCE IS 0, AND THE FEEDBACK IS THE SPRUNG MASS: This means that the car vertical speed = 0 \rightarrow ACC = 0

WE ADJUST THIS CONTROLLER BY ADDING A PI REGULATOR $k_p = 10^4$, $k_i = 0$, $k_d = 10^4$

THE RESULT IS PRETTY GOOD!



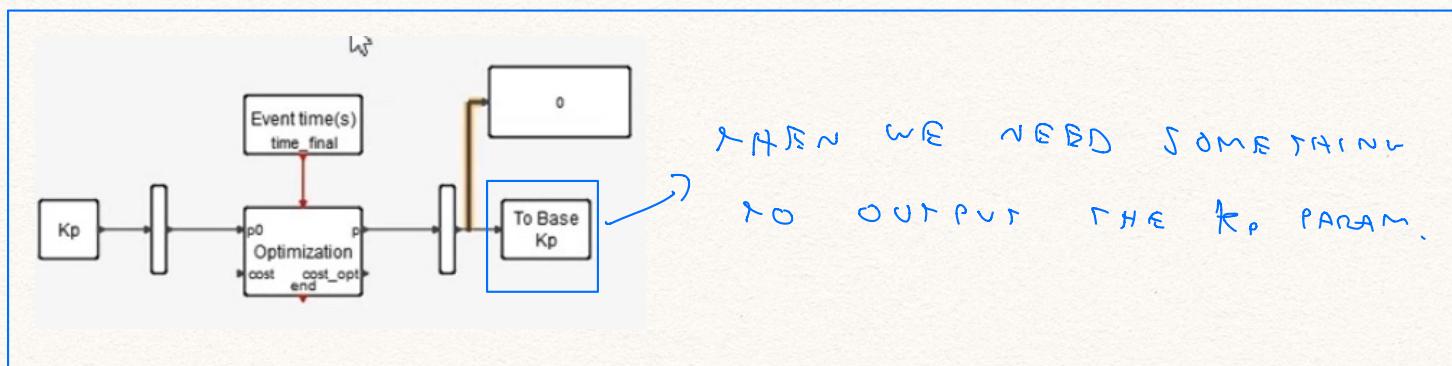
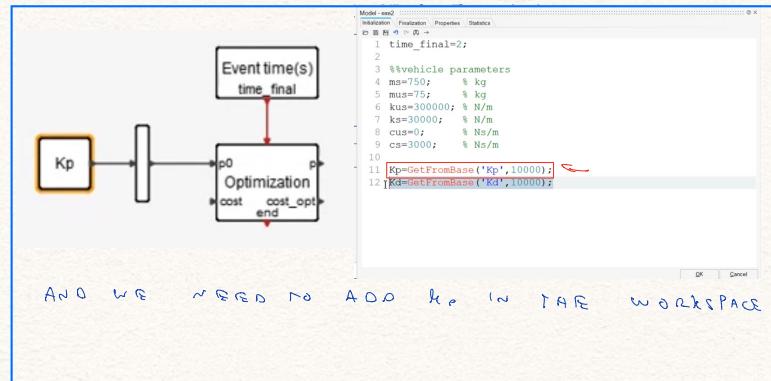
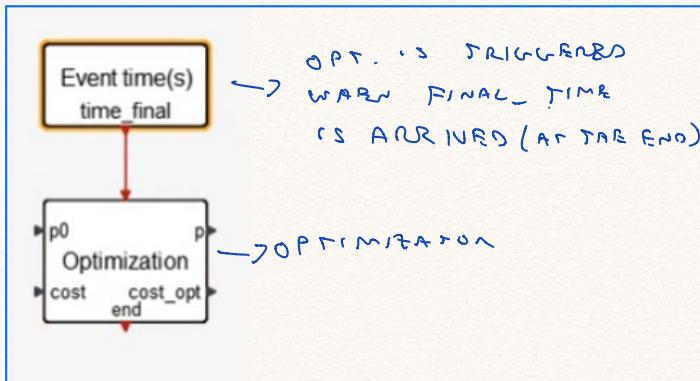
a) WE ADD THE OPTIMIZER

FIRST OF ALL WE ADD A STA. BLOCK AFTER THE PID
BECAUSE PERC. OF ACTUAR. IS LIMITED

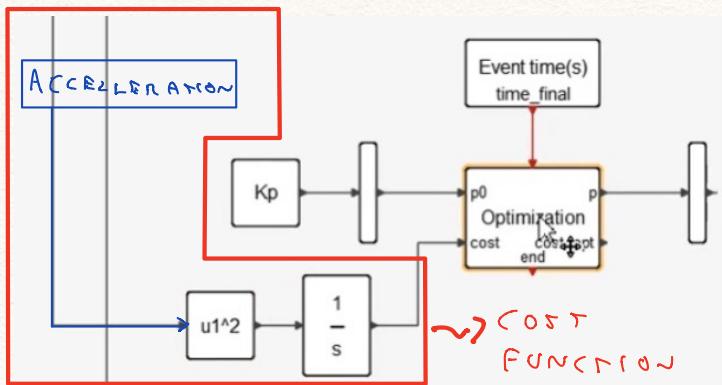
→ NOW WE HAVE WORST BEHAVIOR (MORE SIMILAR TO PASSIVE SUSP.)

AFTER THIS, WE HAVE ALSO ADDED NON LINEARITIES!!

NOW, LET'S START TO OPTIMIZE THE K_p



NOW THE COST FUNCTION:

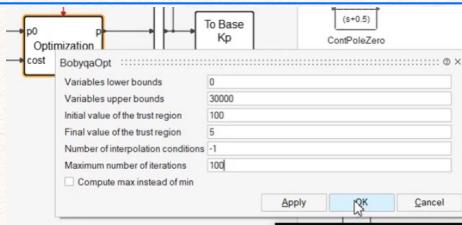


WE HAVE BUILT THE COST FCN IN ORDER TO MINIMIZE THE ACC.

$$\text{COST} = \int_0^t acc^2 dt$$

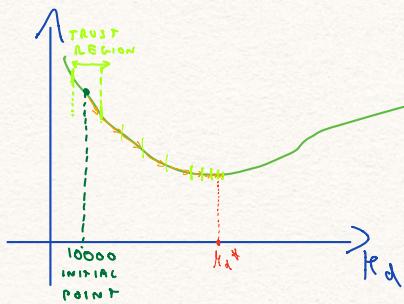
why the squared? because bobyqaopt uses THE GRADIENT FOR THE COST. so if i use the square I'm sure THAT I'M BUILDING A CONTINUOUS DERIVATIVE FOR COST FUNCTION → SIMPLER FOR OPTIMIZER

WE THEN SET ALL THE BOUNDARIES FOR THE OPTIMIZER...



HOW OPTIMIZATION IS WORKING?

WE HAVE ONLY A PARAM. (k_p) AND I WANT
TO MINIMIZE A COST FUNCTION ($\int_{0}^{T_f} \dot{x}^2 dt$)



The alg. starts by APPROXIMATING THE COST VALUE INSIDE THE TRUST REGION, THEN DECIDES WHERE TO MOVE k_p IN ORDER TO MINIMIZE THE COST FCN. WHEN I'M NEAR THE MIN., MY ALG. IS GOING TO SHRINK THE TRUST REGION.

IN k_p^* WE REACHED A VERY LITTLE VALUE OF TRUST REGION AND THE ALG. STOPS -> THIS MEAN THAT WE REACHED THE CONVERGENCE.

The same is if we have 2 parameters, but trust region now is a RADIOUS

IN OUR CASE : START TRUST RADIUS: 100 

FINAL TRUST RADIUS: 5 

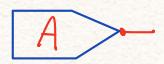
WE RUN THE SIMULATION, AND IT IS GOING TO REPEAT UNTIL IT REACH $k_p = 30'000$

WE CAN ADD ALSO A COOL VIEWER FOR THE COST FCN

NOW FOR HOMEWORK WE CAN ADD k_p, k_i OPTIMIZER!

TIP: SET SIGNALS 

OR REMEMBER TO SET "GLOBAL"

CRT-SIGNAL 

- ACTUATOR CAN BE ALSO IMPLEMENTED BY CHANGING THE DAMPER OPENING! -> REAL CASE -> IF IS CHEAPER, MINIMUM MODIFICATION, BUT WE ONLY CHANGE VELOCITY BECAUSE OUR

• WE CAN ALSO ADJUST THE SPRING CONSTANT
THAT IS NOT ACTIVE

FIND SPRUNG MASS OF A REAL CAR TO USE
MAYBE STRIFRNBGS ARE NOT SIMPLE TO FIND

WHAT IF TARES ARE MORE THAN 1 RECARING MINIMUM?

- 1) Σ CAN START FROM DIFFERENT INITIAL CONDITION
- 2) Σ CAN USE GENERIC OPTIMIZER, BUT A BIT OVERKILL