

OPENFOAM[®]'A GİRİŞ

ÇALIŞTAYI

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- We will look at the following test cases in this section:
1. **Cavity (transient – 2D)**
 2. **Elbow (transient – 2D)**
 3. **Fully developed pipe flow (steady-state – 3D)**
 - i. **Newtonian fluid**
 - ii. **NonNewtonian fluid (shear-thinning Bird-Carreau model for blood)**

Test case: Cavity

➤ Physics

- Let us assume a two-dimensional flow in a cavity.
- The fluid flow as following parameters:

Top wall velocity: $U_x = 1$ m/s

Cavity edge length: $d = 0.1$ m

Newtonian

Laminar flow, $Re = 10$

Incompressible

Isothermal flow

Transient (unsteady) flow

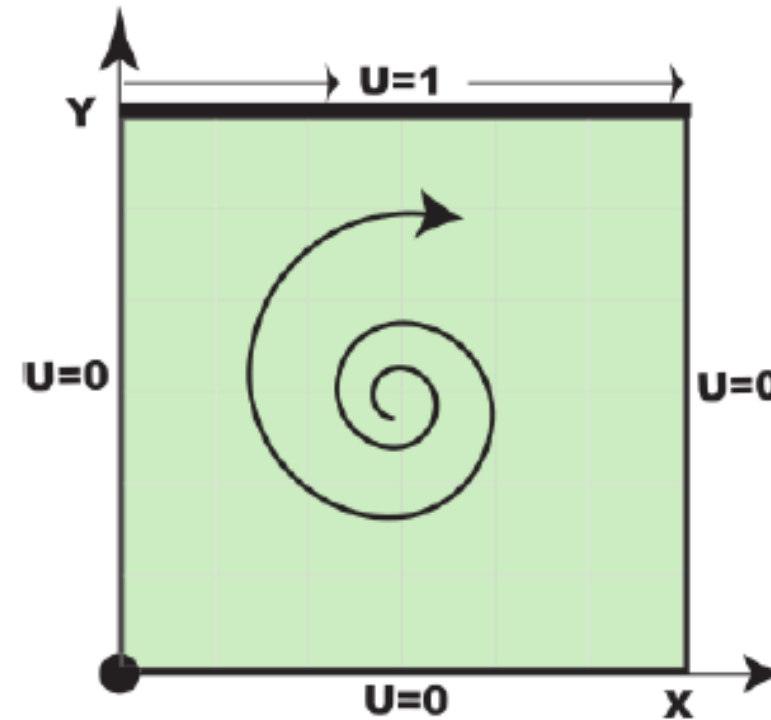


Figure 1: Cavity case geometry

Test case: Cavity

➤ Working with the case

- We (might) need to edit the files.
- Test case directory contains the mesh and all configuration files with boundary conditions, initial conditions, model constants, solver parameters, etc.
- Typical minimal configuration of a test case directory:

caseName	test case directory
└ system	solver parameters
└ controlDict	computation options
└ fvSchemes	numerical schemes options
└ fvSolution	linear solver options
└ constant	constant physics data
└ polyMesh	mesh directory
└ transportProperties	fluid properties
└ turbulenceProperties	turbulence properties
└ 0	initial conditions

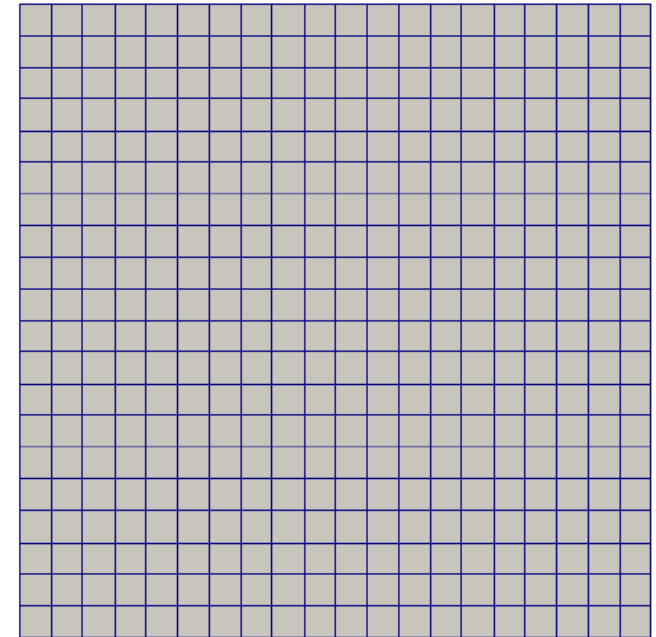


Figure 2: Cavity case computational domain

Test case: Cavity

➤ Generating/importing and checking the mesh

- As mentioned in the first part of the workshop, we can generate our mesh with a mesh utility like **blockMesh** or **snappyHexMesh**; we can also import our “.msh” file with **fluentMeshToFoam**.

- In this example, we use the **blockMesh** utility with its own dictionary file in *caseName/system* folder.

- We run **blockMesh** in the test case directory and display the information about the generated mesh with the **checkMesh** application.

- The mesh is located then in *caseName/constant/polyMesh* folder.

```
Create time
Create mesh for time = 0
Time = 0
Mesh stats
  points:      882
  internal points: 0
  faces:      1640
  internal faces: 760
  cells:      400
  faces per cell: 6
  boundary patches: 3
  point zones: 0
  face zones: 0
  cell zones: 0
Overall number of cells of each type:
  hexahedra: 400
  prisms: 0
  wedges: 0
  pyramids: 0
  tet wedges: 0
  tetrahedra: 0
  polyhedra: 0
Checking topology...
  Boundary definition OK.
  Cell to face addressing OK.
  Point usage OK.
  Upper triangular ordering OK.
  Face vertices OK.
  Number of regions: 1 (OK).
```

checkMesh

```
Checking patch topology for multiply connected surfaces...
  Patch      Faces    Points    Surface topology
  movingWall  20      42      ok (non-closed singly connected)
  fixedWalls  60      122     ok (non-closed singly connected)
  frontAndBack 800     882     ok (non-closed singly connected)
Checking faceZone topology for multiply connected surfaces...
  No faceZones found.
Checking basic cellZone addressing...
  No cellZones found.
Checking geometry...
  Overall domain bounding box (0 0 0) (0.1 0.1 0.01)
  Mesh has 2 geometric (non-empty/wedge) directions (1 1 0)
  Mesh has 2 solution (non-empty) directions (1 1 0)
  All edges aligned with or perpendicular to non-empty directions.
  Boundary openness (8.47033e-18 -8.47033e-18 -4.51751e-17) OK.
  Max cell openness = 1.35525e-16 OK.
  Max aspect ratio = 1 OK.
  Minimum face area = 2.5e-05. Maximum face area = 5e-05. Face area magnitudes OK.
  Min volume = 2.5e-07. Max volume = 2.5e-07. Total volume = 0.0001. Cell volumes OK.
  Mesh non-orthogonality Max: 0 average: 0
  Non-orthogonality check OK.
  Face pyramids OK.
  Max skewness = 1.66533e-14 OK.
  Coupled point location match (average 0) OK.
Mesh OK.
End
```

```
17 scale 0.1; //Scale coordinates
18
19 vertices
20 (
21   (0 0 0) //vertex 0
22   (1 0 0) //vertex 1
23   (1 1 0) //vertex 2
24   (0 1 0) //vertex 3
25   (0 0 0.1) //vertex 4
26   (1 0 0.1) //vertex 5
27   (1 1 0.1) //vertex 6
28   (0 1 0.1) //vertex 7
29 );
30
31 blocks
32 ( //Definition of the hexahedral block. Pay attention to the numbering.
33   hex (0 1 2 3 4 5 6 7) (20 20 1) simpleGrading (1 1 1) //Uniform gradient
34 );
35 //20 cells in x-direction, 20 cells in y-direction,
36 //and only 1 cell in z-direction bcs the simulation will be 2D.
37
38 edges //This is simple geometry, edges can be empty.
39 (
40 );
41 boundary //Definition of boundary of the domain to apply the BC.
42 (
43   movingWall
44   {
45     type wall;
46     faces
47     (
48       (3 7 6 2)
49     );
50   }
51   fixedWalls
52   {
53     type wall;
54     faces
55     (
56       (0 4 7 3)
57       (2 6 5 1)
58       (1 5 4 0)
59     );
60   }
61   frontAndBack
62   {
63     type empty; // Faces normal to Oz are "empty" to specify that the simulation is 2D.
64     faces
65     (
66       (0 3 2 1)
67       (4 5 6 7)
68     );
69   }
70 );
```

blockMeshDict

Test case: Cavity

➤ Boundary and initial conditions

- Initial and boundary conditions are located in directories named with numbers.
- The number in directory means the physical time layer of the solution.
- Usually, the initial and boundary conditions are stored in directory **caseName/0**.
- For the incompressible laminar Newtonian transient fluid flow, we will use **icoFoam** which requires initial pressure (**caseName/0/p**) and velocity (**caseName/0/U**) conditions.

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        volScalarField;
    object       p;
}
// * * * * *

dimensions      [0 2 -2 0 0 0 0];
internalField   uniform 0;
boundaryField
{
    movingWall
    {
        type      zeroGradient;
    }
    fixedWalls
    {
        type      zeroGradient;
    }
    frontAndBack
    {
        type      empty;
    }
}
// *****
```

Pressure conditions

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        volVectorField;
    object       U;
}
// * * * * *

dimensions      [0 1 -1 0 0 0 0];
internalField   uniform (0 0 0);
boundaryField
{
    movingWall
    {
        type      fixedValue;
        value      uniform (1 0 0);
    }
    fixedWalls
    {
        type      noSlip;
    }
    frontAndBack
    {
        type      empty;
    }
}
// *****
```

Velocity conditions

Test case: Cavity

➤ Physical properties

- The physical properties of the fluid flow are located in directory **caseName/constant**.
- Inside this directory we find the mesh and the configuration file named **transportProperties** (may also include **turbulenceProperties**, depends on the problem and the solver)
- In this problem, the fluid is Newtonian, so we need only one parameter which is the **kinematic viscosity ν (nu)** in **m²/s**.
- We also need to check the Reynolds number, since we defined the case as laminar.

$$Re = \frac{U \cdot d}{\nu} = \frac{1.0 \cdot 0.1}{0.01} = 10$$

```
FoamFile
{
  version      2.0;
  format       ascii;
  class        dictionary;
  object       transportProperties;
}
// * * * * *
nu             0.01;
```

transportProperties

➤ Computation setup

- Parameters of computational process are set in files inside **caseName/system** directory.
- The file **controlDict** controls the computation in terms of time interval, time step, writing data etc.

```
FoamFile
{
  version      2.0;
  format       ascii;
  class        dictionary;
  object       controlDict;
}
// * * * * *
application    icoFoam;
startFrom      startTime;
startTime      0;
stopAt         endTime;
endTime        0.5;
deltaT         0.005;
writeControl   timeStep;
writeInterval  2;
purgeWrite     0;
writeFormat    ascii;
writePrecision 6;
writeCompression off;
timeFormat     general;
timePrecision  6;
runTimeModifiable true;
```

controlDict

We use icoFoam.

The simulation starts from $t = 0$ s and ends at $t = 0.5$ s.

The time step size is $\Delta t = 0.005$ s.

The solution is saved after each two timesteps.

We are allowed to change **controlDict** during computation.

Test case: Cavity

➤ Numerical solution setup

- Parameters of numerical solution are stored in files in files inside **caseName/system** directory.
- In file **fvSolution**, there are set parameters of solution of system of linear equations for each quantity and also parameters of numerical methods.
- In file **fvSchemes**, there are setup the numerical schemes for individual quantities and other members.

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       fvSolution;
}
// *****

solvers
{
    p
    {
        solver          PCG;
        preconditioner   DIC;
        tolerance        1e-06;
        relTol           0.05;
    }

    pFinal
    {
        $p;
        relTol          0;
    }

    U
    {
        solver          smoothSolver;
        smoother         symGaussSeidel;
        tolerance        1e-05;
        relTol           0;
    }
}

PISO
{
    nCorrectors      2;
    nNonOrthogonalCorrectors 0;
    pRefCell          0;
    pRefValue          0;
}
```

fvSolution

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       fvSchemes;
}
// *****

ddtSchemes
{
    default         Euler;
}

gradSchemes
{
    default          Gauss linear;
    grad(p)          Gauss linear;
}

divSchemes
{
    default          none;
    div(phi,U)       Gauss linear;
}

laplacianSchemes
{
    default          Gauss linear orthogonal;
}

interpolationSchemes
{
    default          linear;
}

snGradSchemes
{
    default          orthogonal;
}
```

fvSchemes

Test case: Cavity

➤ Running the simulation sequentially

- We run the simulation sequentially on background and save the log file with the command

```
# icoFoam > log.icofoam &
```

- We can watch the log file with

```
# tail -f log.icofoam
```

```
=====
  \  /  F i e l d      OpenFOAM: The Open Source CFD Toolbox
   /  \  O p e r a t i o n      Version: 2206
    /    A n d               Website: www.openfoam.com
   \    M a n i p u l a t i o n

Build : 76d719d1-20220624 OPENFOAM=2206 version=2206
Arch : "LSB:label=32;scalar=64"
Exec : icoFoam
Date : Nov 30 2023
Time : 15:03:34
Host : jcjnb
PID : 46353
I/O : uncollated
Case : /home/jcj/CALISTAY/1_cavity
nProcs : 1
trapFpe: Floating point exception trapping enabled (FOAM_SIGFPE).
fileModificationChecking: Monitoring run-time modified files using timeStampMaster (fileModificationSkew 5, maxFileModificationPolls 20)
allowSystemOperations: Allowing user-supplied system call operations

// *****
Create time

Create mesh for time = 0

PISO: Operating solver in PISO mode

Reading transportProperties

Reading field p

Reading field U

Reading/calculating face flux field phi
```

Log file, initialization

```
Starting time loop

Time = 0.005

Courant Number mean: 0 max: 0
smoothSolver: Solving for Ux, Initial residual = 1, Final residual = 8.90511e-06, No Iterations 19
smoothSolver: Solving for Uy, Initial residual = 0, Final residual = 0, No Iterations 0
DICPCG: Solving for p, Initial residual = 1, Final residual = 0.0492854, No Iterations 12
time step continuity errors : sum local = 0.000466513, global = -1.79995e-19, cumulative = -1.79995e-19
DICPCG: Solving for p, Initial residual = 0.590864, Final residual = 2.65225e-07, No Iterations 35
time step continuity errors : sum local = 2.74685e-09, global = -2.6445e-19, cumulative = -4.44444e-19
ExecutionTime = 0.01 s ClockTime = 0 s

Time = 0.01

Courant Number mean: 0.0976825 max: 0.585607
smoothSolver: Solving for Ux, Initial residual = 0.160686, Final residual = 6.83031e-06, No Iterations 19
smoothSolver: Solving for Uy, Initial residual = 0.260828, Final residual = 9.65039e-06, No Iterations 18
DICPCG: Solving for p, Initial residual = 0.428925, Final residual = 0.0103739, No Iterations 22
time step continuity errors : sum local = 0.000110788, global = 3.77194e-19, cumulative = -6.72498e-20
DICPCG: Solving for p, Initial residual = 0.30209, Final residual = 5.26569e-07, No Iterations 33
time step continuity errors : sum local = 6.61987e-09, global = -2.74872e-19, cumulative = -3.42122e-19
ExecutionTime = 0.01 s ClockTime = 0 s
```

Log file, first two timesteps

```
Time = 0.495

Courant Number mean: 0.222158 max: 0.852134
smoothSolver: Solving for Ux, Initial residual = 2.43518e-07, Final residual = 2.43518e-07, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 5.33314e-07, Final residual = 5.33314e-07, No Iterations 0
DICPCG: Solving for p, Initial residual = 5.25763e-07, Final residual = 5.25763e-07, No Iterations 0
time step continuity errors : sum local = 6.11228e-09, global = 1.51821e-18, cumulative = 1.09644e-17
DICPCG: Solving for p, Initial residual = 7.01015e-07, Final residual = 7.01015e-07, No Iterations 0
time step continuity errors : sum local = 7.5484e-09, global = -4.1723e-19, cumulative = 1.05471e-17
ExecutionTime = 0.16 s ClockTime = 0 s

Time = 0.5

Courant Number mean: 0.222158 max: 0.852134
smoothSolver: Solving for Ux, Initial residual = 2.3091e-07, Final residual = 2.3091e-07, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 5.0684e-07, Final residual = 5.0684e-07, No Iterations 0
DICPCG: Solving for p, Initial residual = 0.63844e-07, Final residual = 0.63844e-07, No Iterations 0
time step continuity errors : sum local = 8.8828e-09, global = 4.94571e-19, cumulative = 1.10417e-17
DICPCG: Solving for p, Initial residual = 9.59103e-07, Final residual = 9.59103e-07, No Iterations 0
time step continuity errors : sum local = 9.66354e-09, global = 1.13175e-18, cumulative = 1.21735e-17
ExecutionTime = 0.16 s ClockTime = 0 s

End
```

Log file, last two timesteps

➤ Iteration output

```
Time = 0.495
Courant Number mean: 0.222158 max: 0.852134
smoothSolver: Solving for Ux, Initial residual = 2.43518e-07, Final residual = 2.43518e-07, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 5.33314e-07, Final residual = 5.33314e-07, No Iterations 0
DICPCG: Solving for p, Initial residual = 5.25763e-07, Final residual = 5.25763e-07, No Iterations 0
time step continuity errors : sum local = 6.11228e-09, global = 1.51821e-18, cumulative = 1.09644e-17
DICPCG: Solving for p, Initial residual = 7.01015e-07, Final residual = 7.01015e-07, No Iterations 0
time step continuity errors : sum local = 7.5484e-09, global = -4.1723e-19, cumulative = 1.05471e-17
ExecutionTime = 0.16 s ClockTime = 0 s
```

- Take a closer look at time 0.495 s.
- **Courant Number** mean/max are the mean/maximal **CFL number** for **convergence condition**, related to the well-known formula $CFL = \frac{|U| \cdot \Delta t}{\Delta x}$.
- **smoothSolver** and **DICPCG** are the linear system solvers for velocity components and pressure, respectively, which are set up in **fvSolution**.
- **Initial residual** is a measure of stability of solution with respect to previous iteration, while **Final residual** defines the residual of the system at the end of that timestep.
- **No Iterations** is the number of iterations of the linear system solver.
- **Time step continuity errors** is continuity equation error, basically it is a sum of fluxes over all mesh faces and ideally should be zero.
- **ExecutionTime** is the time elapsed since start of the computation.
- **ClockTime** is the time spent on processor(s).

Test case: Elbow

➤ Physics

- Two-dimensional flow in an elbow geometry with two inlets and one outlet.
- The fluid flow as following parameters:

Inlet 1 velocity: $U_x = 1$ m/s, Inlet 2 velocity: $U_y = 3$ m/s

Inlet 1 and outlet diameter: $D_1 = 16$ m

Inlet 2 diameter: $D_1 = 4$ m, length in x-axis: $L = 64$ m

Newtonian

Laminar flow, $Re_{D_1} = 1600$

Incompressible

Isothermal flow

Transient (unsteady) flow

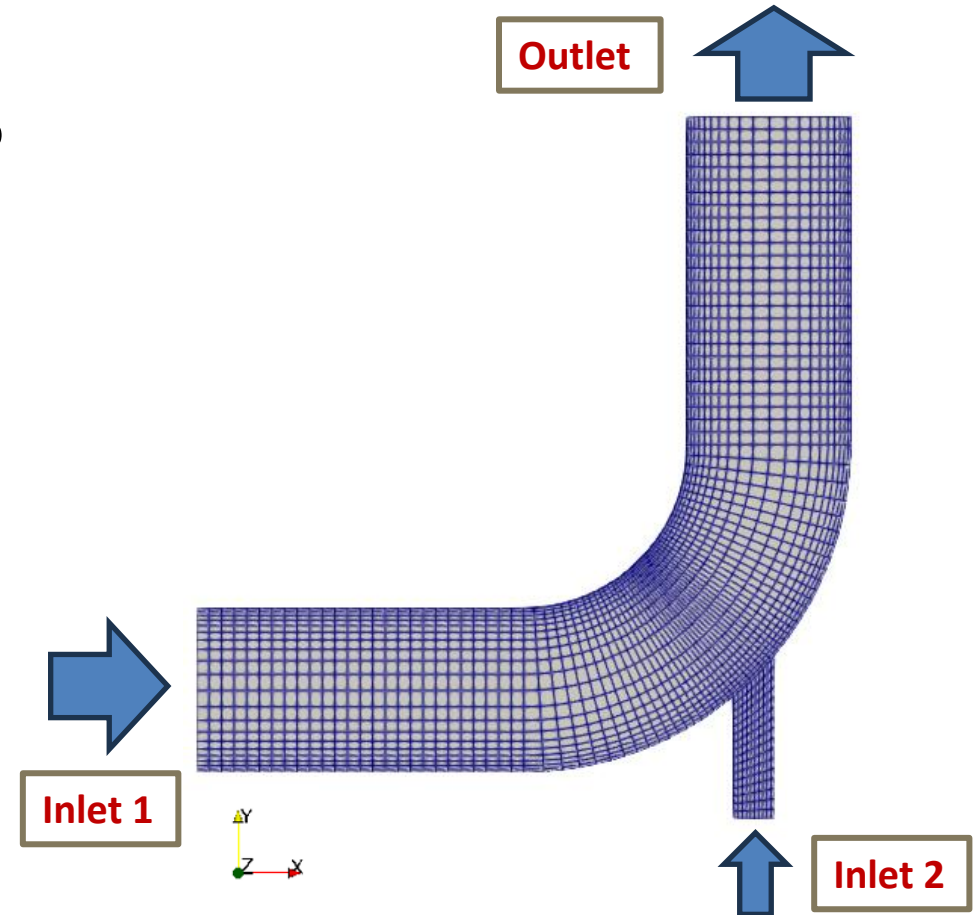


Figure 3: Elbow case geometry and quad mesh

Test case: Elbow

```
Create mesh for time = 0
```

```
Time = 0
```

```
Mesh stats
points:      4678
internal points: 0
faces:      8938
internal faces: 4262
cells:      2200
faces per cell: 6
boundary patches: 5
point zones: 0
face zones: 0
cell zones: 0
```

```
Overall number of cells of each type:
```

```
hexahedra: 2200
prisms: 0
wedges: 0
pyramids: 0
tet wedges: 0
tetrahedra: 0
polyhedra: 0
```

```
Checking topology...
```

```
Boundary definition OK.
Cell to face addressing OK.
Point usage OK.
Upper triangular ordering OK.
Face vertices OK.
Number of regions: 1 (OK).
```

```
Checking patch topology for multiply connected surfaces...
```

Patch	Faces	Points	Surface topology
wall-4	226	458	ok (non-closed singly connected)
pressure-outlet-7	20	42	ok (non-closed singly connected)
velocity-inlet-6	10	22	ok (non-closed singly connected)
velocity-inlet-5	20	42	ok (non-closed singly connected)
frontAndBackPlanes	4400	4678	ok (non-closed singly connected)

```
Checking faceZone topology for multiply connected surfaces...
```

```
No faceZones found.
```

```
Checking basic cellZone addressing...
```

```
No cellZones found.
```

```
Checking geometry...
```

```
Overall domain bounding box (-32 -36.5385 -0.937738) (32 32 0.937738)
Mesh has 2 geometric (non-empty/wedge) directions (1 1 0)
Mesh has 2 solution (non-empty) directions (1 1 0)
All edges aligned with or perpendicular to non-empty directions.
Boundary openness (7.19807e-19 -9.78938e-19 1.15169e-18) OK.
Max cell openness = 2.06735e-16 OK.
Max aspect ratio = 5.34145 OK.
Minimum face area = 0.0623165. Maximum face area = 3.68914. Face area magnitudes OK.
Min volume = 0.116873. Max volume = 4.57578. Total volume = 3155.59. Cell volumes OK.
Mesh non-orthogonality Max: 48.197 average: 7.57964
Non-orthogonality check OK.
Face pyramids OK.
Max skewness = 0.477427 OK.
Coupled point location match (average 0) OK.
```

```
Mesh OK.
```

```
End
```

checkMesh

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        volScalarField;
    object       p;
}
// *****

dimensions      [0 2 -2 0 0 0];

internalField    uniform 0;

boundaryField
{
    wall-4
    {
        type      zeroGradient;
    }

    velocity-inlet-5
    {
        type      zeroGradient;
    }

    velocity-inlet-6
    {
        type      zeroGradient;
    }

    pressure-outlet-7
    {
        type      fixedValue;
        value      uniform 0;
    }

    wall-8
    {
        type      zeroGradient;
    }

    frontAndBackPlanes
    {
        type      empty;
    }
}
```

Pressure conditions

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        volVectorField;
    object       U;
}
// *****

dimensions      [0 1 -1 0 0 0];

internalField    uniform (0 0 0);

boundaryField
{
    wall-4
    {
        type      noSlip;
    }

    velocity-inlet-5
    {
        type      fixedValue;
        value      uniform (1 0 0);
    }

    velocity-inlet-6
    {
        type      fixedValue;
        value      uniform (0 3 0);
    }

    pressure-outlet-7
    {
        type      zeroGradient;
    }

    wall-8
    {
        type      noSlip;
    }

    frontAndBackPlanes
    {
        type      empty;
    }
}
```

Velocity conditions

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       transportProperties;
}
// *****

nu              0.01;
```

transportProperties

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       controlDict;
}
// *****

application     icoFoam;
startFrom        latestTime;
startTime        0;
stopAt           endTime;
endTime          500;
deltaT           0.05;
writeControl     timeStep;
writeInterval    100;
purgeWrite       0;
writeFormat      ascii;
writePrecision   6;
writeCompression off;
timeFormat       general;
timePrecision    6;
runTimeModifiable true;
```

controlDict

Test case: Elbow

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       fvSolution;
}
// *****

solvers
{
    p
    {
        solver      PCG;
        preconditioner DIC;
        tolerance    1e-06;
        relTol       0;
    }

    pFinal
    {
        $p;
        relTol       0;
    }

    U
    {
        solver      smoothSolver;
        smoother     symGaussSeidel;
        tolerance    1e-06;
        relTol       0;
    }
}

PISO
{
    nCorrectors      2;
    nNonOrthogonalCorrectors 2;
}
```

fvSolution

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       fvSchemes;
}
// *****

ddtSchemes
{
    default       Euler;
}

gradSchemes
{
    default       Gauss linear;
}

divSchemes
{
    default       Gauss linear;
}

laplacianSchemes
{
    default       Gauss linear corrected;
}

interpolationSchemes
{
    default       linear;
}

snGradSchemes
{
    default       corrected;
}
```

fvSchemes

➤ Running the simulation in parallel

- To run the simulation in parallel, first, we need decompose the computational domain.

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       decomposeParDict;
}
// *****

numberOfSubdomains 4;

method            scotch;
```

To decompose the domain, we use the command

decomposePar

- We run the simulation in parallel on background and save the log file with the command

```
# mpirun -np 4 icoFoam
-parallel > log.icofoam &
```

Test case: Elbow

```
Build : _76d719d1-20220624 OPENFOAM=2206 version=2206
Arch : "LSB;label=32;scalar=64"
Exec : icoFoam -parallel
Date : Nov 30 2023
Time : 04:56:21
Host : jcnb
PID : 40965
I/O : uncollated
Case : /home/jcj/CALISTAY/3_2_elbow_quad
nProcs : 4
Hosts :
{
  (jcnb 4)
}
Pstream initialized with:
  floatTransfer : 0
  nProcSimpleSum : 0
  commType : nonBlocking
  polling iterations : 0
trapFpe: Floating point exception trapping enabled (FOAM_SIGFPE).
fileModificationChecking: Monitoring run-time modified files using timeStampMaster (fileModificationSkew 5, maxFileModificationPolls 20)
allowSystemOperations: Allowing user-supplied system call operations

// *****
Create time

Create mesh for time = 0

PISO: Operating solver in PISO mode

Reading transportProperties

Reading field p

Reading field U

Reading/calculating face flux field phi
```

Log file, initialization

- To combine the results of a decomposed case run in parallel, we use the command

```
# reconstructPar
```

which will reconstruct all time or solution steps and save the corresponding directories in the case directory.

```
Starting time loop

Time = 0.05

Courant Number mean: 0.000416034 max: 0.0982579
smoothSolver: Solving for Ux, Initial residual = 1, Final residual = 2.42637e-11, No Iterations 2
smoothSolver: Solving for Uy, Initial residual = 1, Final residual = 1.34815e-10, No Iterations 2
DICPCG: Solving for p, Initial residual = 1, Final residual = 9.89997e-07, No Iterations 104
DICPCG: Solving for p, Initial residual = 0.00940477, Final residual = 9.01899e-07, No Iterations 93
DICPCG: Solving for p, Initial residual = 0.00251053, Final residual = 9.36394e-07, No Iterations 88
time step continuity errors : sum local = 1.23326e-07, global = -1.11068e-09, cumulative = -1.11068e-09
DICPCG: Solving for p, Initial residual = 0.00199114, Final residual = 9.62667e-07, No Iterations 87
DICPCG: Solving for p, Initial residual = 0.000954229, Final residual = 7.50071e-07, No Iterations 86
DICPCG: Solving for p, Initial residual = 0.000535386, Final residual = 9.93675e-07, No Iterations 82
time step continuity errors : sum local = 1.31772e-07, global = -3.13951e-09, cumulative = -4.2502e-09
ExecutionTime = 0.03 s ClockTime = 0 s

Time = 0.1

Courant Number mean: 0.0776366 max: 0.692664
smoothSolver: Solving for Ux, Initial residual = 0.410648, Final residual = 6.21884e-07, No Iterations 5
smoothSolver: Solving for Uy, Initial residual = 0.409889, Final residual = 5.64684e-07, No Iterations 5
DICPCG: Solving for p, Initial residual = 0.0120064, Final residual = 9.65585e-07, No Iterations 95
DICPCG: Solving for p, Initial residual = 0.204778, Final residual = 7.11365e-07, No Iterations 102
DICPCG: Solving for p, Initial residual = 0.116282, Final residual = 7.05207e-07, No Iterations 100
time step continuity errors : sum local = 1.94665e-09, global = -1.45069e-11, cumulative = -1.45069e-11
DICPCG: Solving for p, Initial residual = 0.188029, Final residual = 7.83351e-07, No Iterations 100
DICPCG: Solving for p, Initial residual = 0.062367, Final residual = 9.77574e-07, No Iterations 95
DICPCG: Solving for p, Initial residual = 0.0389483, Final residual = 9.07815e-07, No Iterations 96
time step continuity errors : sum local = 1.80072e-09, global = -7.70661e-12, cumulative = -4.27241e-09
ExecutionTime = 0.04 s ClockTime = 0 s
```

Log file, first two timesteps

```
Time = 499.95

Courant Number mean: 0.0818688 max: 0.600555
smoothSolver: Solving for Ux, Initial residual = 0.000272859, Final residual = 8.53207e-07, No Iterations 1
smoothSolver: Solving for Uy, Initial residual = 0.000183287, Final residual = 4.95606e-07, No Iterations 1
DICPCG: Solving for p, Initial residual = 0.00223091, Final residual = 8.36427e-07, No Iterations 80
DICPCG: Solving for p, Initial residual = 6.54528e-06, Final residual = 9.90617e-07, No Iterations 2
DICPCG: Solving for p, Initial residual = 1.31079e-06, Final residual = 7.50271e-07, No Iterations 1
time step continuity errors : sum local = 2.27207e-10, global = 3.99936e-13, cumulative = 3.97177e-09
DICPCG: Solving for p, Initial residual = 5.80851e-05, Final residual = 9.63296e-07, No Iterations 8
DICPCG: Solving for p, Initial residual = 1.51691e-06, Final residual = 7.53125e-07, No Iterations 1
DICPCG: Solving for p, Initial residual = 8.6013e-07, Final residual = 8.6013e-07, No Iterations 0
time step continuity errors : sum local = 2.60485e-10, global = -2.65115e-11, cumulative = 5.64526e-09
ExecutionTime = 45.41 s ClockTime = 45 s

Time = 500

Courant Number mean: 0.0818695 max: 0.600558
smoothSolver: Solving for Ux, Initial residual = 0.000272837, Final residual = 8.54738e-07, No Iterations 1
smoothSolver: Solving for Uy, Initial residual = 0.000183435, Final residual = 4.94544e-07, No Iterations 1
DICPCG: Solving for p, Initial residual = 0.00220913, Final residual = 7.44038e-07, No Iterations 85
DICPCG: Solving for p, Initial residual = 5.98304e-06, Final residual = 5.86548e-07, No Iterations 3
DICPCG: Solving for p, Initial residual = 9.87017e-07, Final residual = 9.87017e-07, No Iterations 0
time step continuity errors : sum local = 2.98907e-10, global = 6.55367e-13, cumulative = 5.64591e-09
DICPCG: Solving for p, Initial residual = 5.08826e-05, Final residual = 9.13068e-07, No Iterations 8
DICPCG: Solving for p, Initial residual = 1.51504e-06, Final residual = 7.53873e-07, No Iterations 1
DICPCG: Solving for p, Initial residual = 8.70743e-07, Final residual = 8.70743e-07, No Iterations 0
time step continuity errors : sum local = 2.63466e-10, global = -2.68716e-11, cumulative = 5.61904e-09
ExecutionTime = 45.42 s ClockTime = 45 s

End

Finalising parallel run
```

Log file, last two timesteps

Test case: Fully-developed pipe flow

➤ Physics

- Three-dimensional flow through a cylindrical pipe geometry with an inlet and outlet.
- The fluid flow as following parameters:

Inlet velocity: $U_z = 0.05$ m/s

Pipe diameter: $D = 0.01$ m, pipe length: $L = 0.1$ m

Blood flow – two cases (**Newtonian & Bird-Carreau shear-thinning model**)

Laminar flow, $Re \approx 150$

Incompressible

Isothermal flow

Steady-state

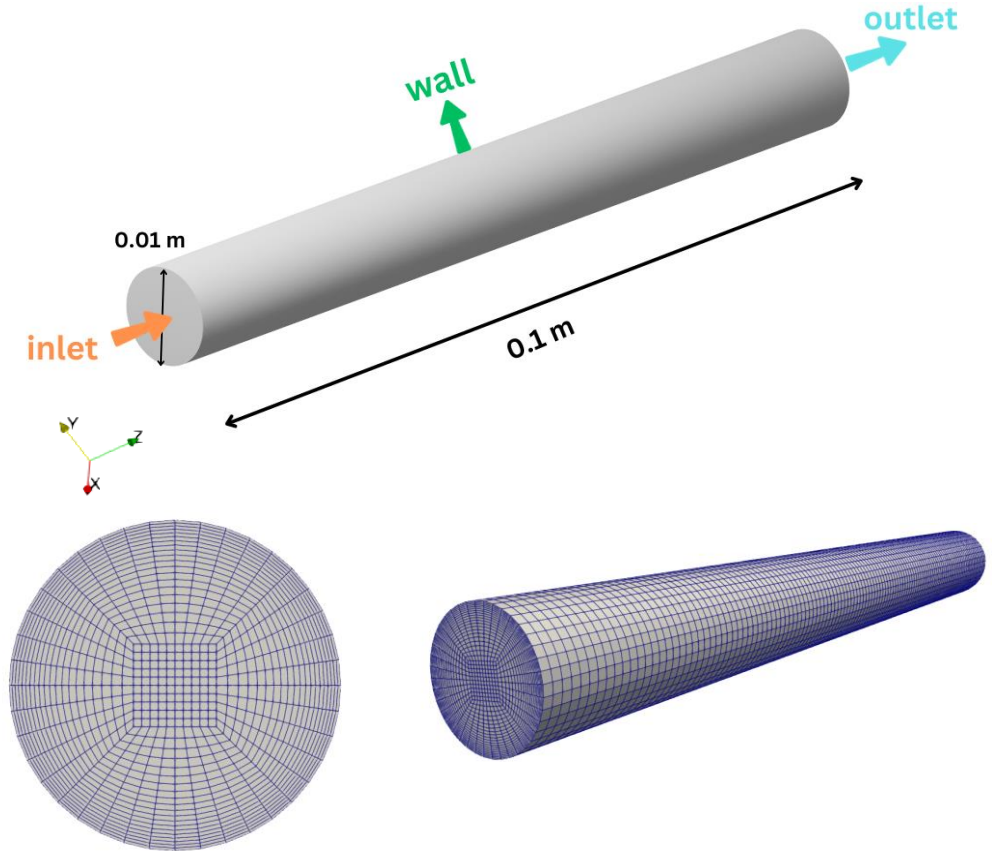


Figure 4: Pipe geometry and hexahedral mesh with o-grid structure.

Test case: Fully-developed pipe flow

```
Create time
Create mesh for time = 0

Time = 0

Mesh stats
  points:      111441
  faces:      327300
  internal faces: 320700
  cells:      108000
  faces per cell: 6
  boundary patches: 3
  point zones: 0
  face zones: 0
  cell zones: 0

Overall number of cells of each type:
  hexahedra: 108000
  prisms: 0
  wedges: 0
  pyramids: 0
  tet wedges: 0
  tetrahedra: 0
  polyhedra: 0

Checking topology...
  Boundary definition OK.
  Cell to face addressing OK.
  Point usage OK.
  Upper triangular ordering OK.
  Face vertices OK.
  Number of regions: 1 (OK).

Checking patch topology for multiply connected surfaces...
  Patch      Faces    Points  Surface topology
  inlet      900      921    ok (non-closed singly connected)
  outlet     900      921    ok (non-closed singly connected)
  wall       4800     4840   ok (non-closed singly connected)

Checking faceZone topology for multiply connected surfaces...
  No faceZones found.

Checking basic cellZone addressing...
  No cellZones found.

Checking geometry...
  Overall domain bounding box (-0.005 -0.005 0) (0.005 0.005 0.1)
  Mesh has 3 geometric (non-empty/wedge) directions (1 1 1)
  Mesh has 3 solution (non-empty) directions (1 1 1)
  Boundary openness (6.08787e-16 6.73362e-16 1.51325e-16) OK.
  Max cell openness = 2.07351e-16 OK.
  Max aspect ratio = 9.83512 OK.
  Minimum face area = 5.88846e-08. Maximum face area = 6.53826e-07. Face area magnitudes OK.
  Min volume = 4.90705e-11. Max volume = 9.48379e-11. Total volume = 7.82172e-06. Cell volumes OK.
  Mesh non-orthogonality Max: 30.3305 average: 5.36658
  Non-orthogonality check OK.
  Face pyramids OK.
  Max skewness = 0.620784 OK.
  Coupled point location match (average 0) OK.

Mesh OK.

End
```

checkMesh

```
FoamFile
{
  version      2.0;
  format       ascii;
  class        dictionary;
  object       controlDict;
}

// *****

application    simpleFoam;

startFrom      startTime;

startTime      0;

stopAt         endTime;

endTime        1000;

deltaT         1;

writeControl    timeStep;

writeInterval   100;

purgeWrite     0;

writeFormat     ascii;

writePrecision  6;

writeCompression off;

timeFormat      general;

timePrecision   6;

runTimeModifiable true;
```

controlDict

```
FoamFile
{
  version      2.0;
  format       ascii;
  class        volScalarField;
  object       p;
}

// *****

dimensions      [0 2 -2 0 0 0];

internalField   uniform 0;

boundaryField
{
  wall
  {
    type        zeroGradient;
  }

  inlet
  {
    type        zeroGradient;
  }

  outlet
  {
    type        fixedValue;
    value       uniform 0;
  }
}
```

Pressure conditions

```
FoamFile
{
  version      2.0;
  format       ascii;
  class        volVectorField;
  object       U;
}

// *****

dimensions      [0 1 -1 0 0 0];

internalField   uniform (0 0 0.05);

boundaryField
{
  wall
  {
    type        fixedValue;
    value       uniform (0 0 0);
  }

  inlet
  {
    type        fixedValue;
    value       uniform (0 0 0.05);
  }

  outlet
  {
    type        zeroGradient;
  }
}
```

Velocity conditions

Test case: Fully-developed pipe flow

➤ Blood properties

- Blood behaves as a non-Newtonian fluid, which shows all signs of non-Newtonian rheology including shear-thinning, viscoelasticity, and thixotropy.
- In a non-Newtonian fluid, the shear stresses and the shear rates have a non-linear relation.
- Two major groups of non-Newtonian fluids are shear thickening and shear thinning fluids.
- Blood shows shear thinning behavior and decreasing shear rate results in a strong increase of viscosity.

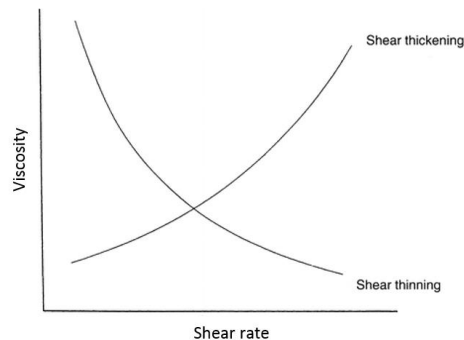


Figure 5: Comparison of shear thickening and shear thinning fluids.

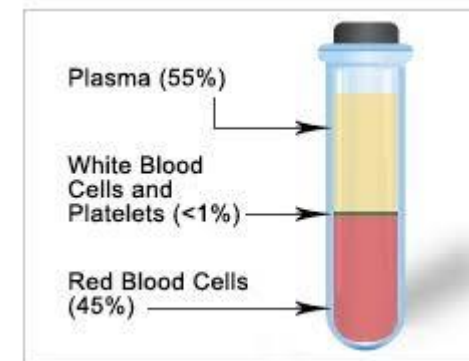


Figure 6: Main components of blood.

Test case: Fully-developed pipe flow

➤ Viscosity models

- Several mathematical models explain the shear-thinning behaviour of blood.
- The most common constitutive equations describing this behaviour are the Power-Law model, Casson model and Bird-Carreau model.
- In this test case, we want to show you the difference between Newtonian and Bird-Carreau models.
- In Bird-Carreau model, the viscosity ν is related to the shear rate $\dot{\gamma}$ by the equation

$$\nu = \nu_{\infty} + (\nu_0 - \nu_{\infty})[1 + (k \cdot \dot{\gamma})^a]^{(n-1)/a}$$

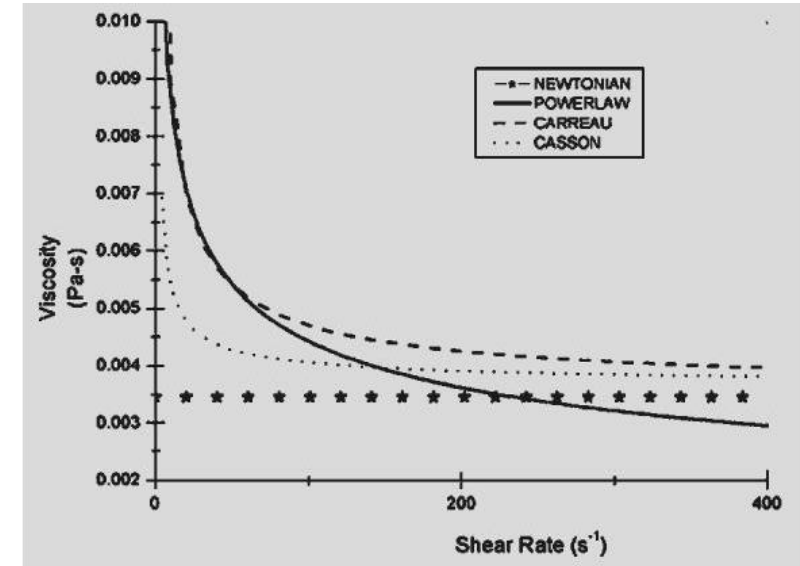


Figure 7: The change of blood viscosity with shear rate for four different viscosity models.

Test case: Fully-developed pipe flow

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       transportProperties;
}
// *****

transportModel  Newtonian;

nu              3.302e-6;

BirdCarreauCoeffs
{
    nu0          5.283e-5; //mu0=0.056
    nuInf        3.302e-6; //muInf=0.0035
    k            3.313;
    n            0.3568;
}
```

transportProperties
Newtonian

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       transportProperties;
}
// *****

transportModel  BirdCarreau;

BirdCarreauCoeffs
{
    nu0          5.283e-5; //mu0=0.056
    nuInf        3.302e-6; //muInf=0.0035
    k            3.313;
    n            0.3568;
}
```

transportProperties
Bird-Carreau

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       fvSolution;
}
// *****

solvers
{
    p
    {
        solver      GAMG;
        smoother    GaussSeidel;
        tolerance    1e-06;
        relTol      0;
    }

    U
    {
        solver      smoothSolver;
        smoother    symGaussSeidel;
        tolerance    1e-06;
        relTol      0;
    }
}

SIMPLE
{
    nNonOrthogonalCorrectors 0;
    residualControl
    {
        p          1e-6;
        U          1e-6;
    }
}

relaxationFactors
{
    fields
    {
        p          0.3;
    }
    equations
    {
        U          0.7;
    }
}
```

fvSolution

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    location     "system";
    object       fvSchemes;
}
// *****

ddtSchemes
{
    default      steadyState;
}

gradSchemes
{
    default      Gauss linear;
}

divSchemes
{
    default      Gauss linear;
}

laplacianSchemes
{
    default      Gauss linear orthogonal;
}

interpolationSchemes
{
    default      linear;
}

snGradSchemes
{
    default      corrected;
}
```

fvSchemes

```
FoamFile
{
    version      2.0;
    format       ascii;
    class        dictionary;
    object       decomposeParDict;
}
// *****

numberOfSubdomains 4;

method            scotch;
```

decomposeParDict

Test case: Fully-developed pipe flow

```
Build : 76d719d1-20220624 OPENFOAM=2206 version=2206
Arch : "LSB;label=32;scalar=64"
Exec : simpleFoam -parallel
Date : Nov 30 2023
Time : 22:33:54
Host : jcjnb
PID : 53276
I/O : uncollated
Case : /home/jcj/CALISTAY_CENK/S_pipe_newtonian_parallel
nProcs : 4
Hosts :
(
  (jcjnb 4)
)
Pstream initialized with:
  floatTransfer : 0
  nProcsSimpleSum : 0
  commType : nonBlocking
  pollingIterations : 0
trapFpe: Floating point exception trapping enabled (FOAM_SIGFPE).
fileModificationChecking: Monitoring run-time modified files using timeStampMaster (fileModificationSkew 5, maxFileModificationPolls 20)
allowSystemOperations: Allowing user-supplied system call operations
// *****
Create time

Create mesh for time = 0

SIMPLE: convergence criteria
  field p tolerance 1e-06
  field U tolerance 1e-06

Reading field p

Reading field U

Reading/calculating face flux field phi

Selecting incompressible transport model Newtonian
Selecting turbulence model type laminar
Selecting laminar stress model Stokes
No MRF models present

No finite volume options present
```

Log file, initialization
Newtonian

```
Starting time loop

Time = 1

smoothSolver: Solving for Ux, Initial residual = 0.125447, Final residual = 2.43747e-07, No Iterations 7
smoothSolver: Solving for Uy, Initial residual = 0.125388, Final residual = 2.43421e-07, No Iterations 7
smoothSolver: Solving for Uz, Initial residual = 1, Final residual = 2.12733e-07, No Iterations 9
GAMG: Solving for p, Initial residual = 1, Final residual = 7.37293e-07, No Iterations 66
time step continuity errors : sum local = 1.10347e-08, global = 1.3752e-09, cumulative = 1.3752e-09
ExecutionTime = 0.68 s ClockTime = 0 s

Time = 2

smoothSolver: Solving for Ux, Initial residual = 0.556019, Final residual = 4.74506e-07, No Iterations 9
smoothSolver: Solving for Uy, Initial residual = 0.556019, Final residual = 4.74271e-07, No Iterations 9
smoothSolver: Solving for Uz, Initial residual = 0.19411, Final residual = 2.80613e-07, No Iterations 8
GAMG: Solving for p, Initial residual = 0.0191035, Final residual = 6.74874e-07, No Iterations 44
time step continuity errors : sum local = 7.81154e-07, global = 5.92055e-08, cumulative = 6.05807e-08
ExecutionTime = 0.95 s ClockTime = 1 s
```

Log file, first two iterations
Newtonian

```
Time = 377

smoothSolver: Solving for Ux, Initial residual = 1.05278e-06, Final residual = 1.07346e-07, No Iterations 1
smoothSolver: Solving for Uy, Initial residual = 1.05952e-06, Final residual = 1.07987e-07, No Iterations 1
smoothSolver: Solving for Uz, Initial residual = 7.15186e-08, Final residual = 7.15186e-08, No Iterations 0
GAMG: Solving for p, Initial residual = 9.45886e-07, Final residual = 9.45886e-07, No Iterations 0
time step continuity errors : sum local = 3.87966e-07, global = -1.02126e-07, cumulative = 2.16188e-06
ExecutionTime = 48.57 s ClockTime = 48 s

Time = 378

smoothSolver: Solving for Ux, Initial residual = 9.88133e-07, Final residual = 9.88133e-07, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 9.94596e-07, Final residual = 9.94596e-07, No Iterations 0
smoothSolver: Solving for Uz, Initial residual = 6.93526e-08, Final residual = 6.93526e-08, No Iterations 0
GAMG: Solving for p, Initial residual = 8.95907e-07, Final residual = 8.95907e-07, No Iterations 0
time step continuity errors : sum local = 3.67467e-07, global = -1.04182e-07, cumulative = 2.0577e-06
ExecutionTime = 48.62 s ClockTime = 48 s

SIMPLE solution converged in 378 iterations

End

Finalising parallel run
```

Log file, last two iterations
Newtonian

Test case: Fully-developed pipe flow

```
Build : 76d719d1-20220624 OPENFOAM=2206 version=2206
Arch : "LSB;label=32;scalar=64"
Exec : simpleFoam -parallel
Date : Nov 30 2023
Time : 22:48:54
Host : jcjnb
PID : 53665
I/O : uncollated
Case : /home/jcj/CALISTAY_CENK/5_pipe_birdcarreau_parallel
nProcs : 4
Hosts :
(
  (jcjnb 4)
)
Pstream initialized with:
  floatTransfer : 0
  nProcsSimpleSum : 0
  commType : nonBlocking
  pollingIterations : 0
trapFpe: Floating point exception trapping enabled (FOAM_SIGFPE).
fileModificationChecking: Monitoring run-time modified files using timeStampMaster (fileModificationSkew 5, maxFileModificationPolls 20)
allowSystemOperations: Allowing user-supplied system call operations

// ***** Create time ***** //
Create time

Create mesh for time = 0

SIMPLE: convergence criteria
  field p tolerance 1e-06
  field U tolerance 1e-06

Reading field p

Reading field U

Reading/calculating face flux field phi

Selecting incompressible transport model BirdCarreau
Selecting turbulence model type laminar
Selecting laminar stress model Stokes
No MRF models present

No finite volume options present
```

Log file, initialization
Bird-Carreau

```
Starting time loop

Time = 1

smoothSolver: Solving for Ux, Initial residual = 0.430471, Final residual = 6.86945e-07, No Iterations 9
smoothSolver: Solving for Uy, Initial residual = 0.429431, Final residual = 6.81165e-07, No Iterations 9
smoothSolver: Solving for Uz, Initial residual = 1, Final residual = 4.44393e-07, No Iterations 12
GAMG: Solving for p, Initial residual = 1, Final residual = 6.70242e-07, No Iterations 64
time step continuity errors : sum local = 3.1172e-09, global = 3.45267e-10, cumulative = 3.45267e-10
ExecutionTime = 0.56 s ClockTime = 0 s

Time = 2

smoothSolver: Solving for Ux, Initial residual = 0.71762, Final residual = 6.48025e-07, No Iterations 11
smoothSolver: Solving for Uy, Initial residual = 0.71762, Final residual = 6.4757e-07, No Iterations 11
smoothSolver: Solving for Uz, Initial residual = 0.413215, Final residual = 5.48578e-07, No Iterations 10
GAMG: Solving for p, Initial residual = 0.0438605, Final residual = 8.16686e-07, No Iterations 52
time step continuity errors : sum local = 3.85537e-07, global = 3.45647e-08, cumulative = 3.491e-08
ExecutionTime = 0.89 s ClockTime = 1 s
```

Log file, first two iterations
Bird-Carreau

```
Time = 859

smoothSolver: Solving for Ux, Initial residual = 1.41232e-06, Final residual = 2.81987e-07, No Iterations 1
smoothSolver: Solving for Uy, Initial residual = 1.06885e-06, Final residual = 2.11971e-07, No Iterations 1
smoothSolver: Solving for Uz, Initial residual = 3.00508e-07, Final residual = 3.00508e-07, No Iterations 0
GAMG: Solving for p, Initial residual = 1.43679e-06, Final residual = 5.25676e-07, No Iterations 1
time step continuity errors : sum local = 1.81752e-07, global = 2.85669e-08, cumulative = 1.03825e-05
ExecutionTime = 105.98 s ClockTime = 106 s

Time = 860

smoothSolver: Solving for Ux, Initial residual = 8.73389e-07, Final residual = 8.73389e-07, No Iterations 0
smoothSolver: Solving for Uy, Initial residual = 9.529e-07, Final residual = 9.529e-07, No Iterations 0
smoothSolver: Solving for Uz, Initial residual = 2.98144e-07, Final residual = 2.98144e-07, No Iterations 0
GAMG: Solving for p, Initial residual = 8.1104e-07, Final residual = 8.1104e-07, No Iterations 0
time step continuity errors : sum local = 2.80417e-07, global = 2.02678e-08, cumulative = 1.04028e-05
ExecutionTime = 106.04 s ClockTime = 106 s

SIMPLE solution converged in 860 iterations

End

Finalising parallel run
```

Log file, last two iterations
Bird-Carreau

Thank you 😊

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