

MANDATORY APPENDIX 2

RULES FOR BOLTED FLANGE CONNECTIONS WITH RING TYPE GASKETS

(17) 2-1 SCOPE

(a) The rules in [Mandatory Appendix 2](#) apply specifically to the design of bolted flange connections with gaskets that are entirely within the circle enclosed by the bolt holes and with no contact outside this circle, and are to be used in conjunction with the applicable requirements in [Subsections A, B, and C](#) of this Division. The hub thickness of weld neck flanges designed to this Appendix shall also comply with the minimum thickness requirements in [Subsection A](#) of this Division. These rules are not to be used for the determination of the thickness of tubesheets integral with a bolting flange as illustrated in [Figure UW-13.2](#) sketches (h) through (l) or [Figure UW-13.3](#) sketch (c). [Nonmandatory Appendix S](#) provides discussion on Design Considerations for Bolted Flanged Connections.

These rules provide only for hydrostatic end loads and gasket seating. The flange design methods outlined in [2-4](#) through [2-8](#) are applicable to circular flanges under internal pressure. Modifications of these methods are outlined in [2-9](#) and [2-10](#) for the design of split and noncircular flanges. See [2-11](#) for flanges with ring type gaskets subject to external pressure, [2-12](#) for flanges with nut-stops, and [2-13](#) for reverse flanges. Rules for calculating rigidity factors for flanges are provided in [2-14](#). Recommendations for qualification of assembly procedures and assemblers are in [2-15](#). Proper allowance shall be made if connections are subject to external loads other than external pressure.

(b) The design of a flange involves the selection of the gasket (material, type, and dimensions), flange facing, bolting, hub proportions, flange width, and flange thickness. See Note in [2-5\(c\)\(1\)](#). Flange dimensions shall be such that the stresses in the flange, calculated in accordance with [13-1](#), do not exceed the allowable flange stresses specified in [2-8](#). Except as provided for in [2-14\(a\)](#), flanges designed to the rules of this Appendix shall also meet the rigidity requirements of [2-14](#). All calculations shall be made on dimensions in the corroded condition.

(c) It is recommended that bolted flange connections conforming to the standards listed in [UG-44](#) be used for connections to external piping. These standards may be used for other bolted flange connections and dished covers within the limits of size in the standards and the pressure-temperature ratings permitted in [UG-44](#). The ratings in these standards are based on the hub

dimensions given or on the minimum specified thickness of flanged fittings of integral construction. Flanges fabricated from rings may be used in place of the hub flanges in these standards provided that their strength, calculated by the rules in this Appendix, is not less than that calculated for the corresponding size of hub flange.

(d) Except as otherwise provided in (c) above, bolted flange connections for unfired pressure vessels shall satisfy the requirements in this Appendix.

(e) The rules of this Appendix should not be construed to prohibit the use of other types of flanged connections, provided they are designed in accordance with good engineering practice and method of design is acceptable to the Inspector. Some examples of flanged connections which might fall in this category are as follows:

(1) flanged covers as shown in [Figure 1-6](#);

(2) bolted flanges using full-face gaskets;

(3) flanges using means other than bolting to restrain the flange assembly against pressure and other applied loads.

2-2 MATERIALS

(17)

(a) Materials used in the construction of bolted flange connections shall comply with the requirements given in [UG-4](#) through [UG-14](#).

(b) Flanges made from ferritic steel and designed in accordance with this Appendix shall be full-annealed, normalized, normalized and tempered, or quenched and tempered when the thickness of the flange section exceeds 3 in. (75 mm).

(c) Material on which welding is to be performed shall be proved of good weldable quality. Satisfactory qualification of the welding procedure under Section IX is considered as proof. Welding shall not be performed on steel that has a carbon content greater than 0.35%. All welding on flange connections shall comply with the requirements for postweld heat treatment given in this Division.

(d) Fabricated flanges with hubs shall be in accordance with the following:

(1) Flanges with hubs may be machined from a hot-rolled billet, forged billet, or forged bar. The axis of the finished flange shall be parallel to the long axis of the original billet or bar, but these axes need not be concentric.

(2) Flanges with hubs [except as permitted in (1) above] shall not be machined from plate or bar stock material unless the material has been formed into a ring, and further provided that

(-a) in a ring formed from plate, the original plate surfaces are parallel to the axis of the finished flange.

(-b) the joints in the ring are welded butt joints that conform to the requirements of this Division. Thickness to be used to determine postweld heat treatment and radiography requirements shall be the lesser of

$$t \text{ or } \frac{(A-B)}{2}$$

where these symbols are as defined in 2-3.

(-c) the back of the flange and the outer surface of the hub are examined by either the magnetic particle method as per Mandatory Appendix 6 or the liquid penetrant method as per Mandatory Appendix 8.

(e) Bolts, studs, nuts, and washers shall comply with the requirements in this Division. It is recommended that bolts and studs have a nominal diameter of not less than $\frac{1}{2}$ in. (13 mm). If bolts or studs smaller than $\frac{1}{2}$ in. (13 mm) are used, ferrous bolting material shall be of alloy steel. Precautions shall be taken to avoid overstressing small-diameter bolts.

(17) 2-3 NOTATION

The symbols described below are used in the equations for the design of flanges (see also Figure 2-4):

A = outside diameter of flange or, where slotted holes extend to the outside of the flange, the diameter to the bottom of the slots

a = nominal bolt diameter

A_b = cross-sectional area of the bolts using the root diameter of the thread or least diameter of unthreaded position, if less

A_m = total required cross-sectional area of bolts, taken as the greater of A_{m1} and A_{m2}

A_{m1} = total cross-sectional area of bolts at root of thread or section of least diameter under stress, required for the operating conditions

$$= W_{m1} / S_a$$

A_{m2} = total cross-sectional area of bolts at root of thread or section of least diameter under stress, required for gasket seating

$$= W_{m2} / S_a$$

B = inside diameter of flange. When B is less than $20g_1$, it will be optional for the designer to substitute B_1 for B in the formula for longitudinal stress S_H .

b = effective gasket or joint-contact-surface seating width [see Note in 2-5(c)(1)]

B_1 = $B + g_1$ for loose type flanges and for integral type flanges that have calculated values h / h_o and g_1 / g_o which would indicate an f value of less than 1.0, although the minimum value of f permitted is 1.0.

= $B + g_o$ for integral type flanges when f is equal to or greater than one

b_o = basic gasket seating width (from Table 2-5.2)

B_s = bolt spacing. The bolt spacing may be taken as the bolt circle circumference divided by the number of bolts or as the chord length between adjacent bolt locations.

B_{sc} = bolt spacing factor

B_{smax} = maximum bolt spacing

C = bolt-circle diameter

c = basic dimension used for the minimum sizing of welds equal to t_n or t_o whichever is less

C_b = conversion factor

= 0.5 for U.S. Customary calculations; 2.5 for SI calculations

d = factor

$$= \frac{U}{V} h_o g_o^2 \text{ for integral type flanges}$$

$$= \frac{U}{V_L} h_o g_o^2 \text{ for loose type flanges}$$

e = factor

$$= \frac{F}{h_o} \text{ for integral type flanges}$$

$$= \frac{F_L}{h_o} \text{ for loose type flanges}$$

F = factor for integral type flanges (from Figure 2-7.2)

f = hub stress correction factor for integral flanges from Figure 2-7.6 (When greater than one, this is the ratio of the stress in the small end of hub to the stress in the large end.) (For values below limit of figure, use $f = 1$.)

F_L = factor for loose type flanges (from Figure 2-7.4)

G = diameter at location of gasket load reaction. Except as noted in sketch (1) of Figure 2-4, G is defined as follows (see Table 2-5.2):

(a) when $b_o \leq \frac{1}{4}$ in. (6 mm), G = mean diameter of gasket contact face

(b) when $b_o > \frac{1}{4}$ in. (6 mm), G = outside diameter of gasket contact face less $2b$

g_1 = thickness of hub at back of flange

g_o = thickness of hub at small end

(a) for optional type flanges calculated as integral and for integral type flanges per Figure 2-4, illustration (7), $g_o = t_n$

(b) for other integral type flanges, g_o = the smaller of t_n or the thickness of the hub at the small end

H = total hydrostatic end force

$$= 0.785 G^2 P$$

h = hub length

H_D	= hydrostatic end force on area inside of flange = $0.785B^2P$	S_H	= calculated longitudinal stress in hub
h_D	= radial distance from the bolt circle, to the circle on which H_D acts, as prescribed in Table 2-6	S_n	= allowable design stress for material of nozzle neck, vessel or pipe wall, at design temperature (operating condition) or atmospheric temperature (gasket seating), as may apply (see UG-23)
H_G	= gasket load (difference between flange design bolt load and total hydrostatic end force) = $W - H$	S_R	= calculated radial stress in flange
h_G	= radial distance from gasket load reaction to the bolt circle = $(C - G)/2$	S_T	= calculated tangential stress in flange
h_o	= factor = $\sqrt{Bg_0}$	T	= factor involving K (from Figure 2-7.1)
H_p	= total joint-contact surface compression load = $2b \times 3.14 GmP$	t	= flange thickness
H_T	= difference between total hydrostatic end force and the hydrostatic end force on area inside of flange = $H - H_D$	t_n	= nominal thickness of shell or nozzle wall to which flange or lap is attached
h_T	= radial distance from the bolt circle to the circle on which H_T acts as prescribed in Table 2-6	t_x	= two times the thickness g_0 , when the design is calculated as an integral flange or two times the thickness of shell nozzle wall required for internal pressure, when the design is calculated as a loose flange, but not less than $\frac{1}{4}$ in. (6 mm)
K	= ratio of outside diameter of flange to inside diameter of flange = A/B	U	= factor involving K (from Figure 2-7.1)
L	= factor = $\frac{te + 1}{T} + \frac{t^3}{d}$	V	= factor for integral type flanges (from Figure 2-7.3)
m	= gasket factor, obtain from Table 2-5.1 [see Note in 2-5(c)(1)]	V_L	= factor for loose type flanges (from Figure 2-7.5)
M_D	= component of moment due to H_D , = $H_D h_D$	W	= flange design bolt load, for the operating conditions or gasket seating, as may apply [see 2-5(e)]
M_G	= component of moment due to H_G , = $H_G h_G$	w	= width used to determine the basic gasket seating width b_0 , based upon the contact width between the flange facing and the gasket (see Table 2-5.2)
M_o	= total moment acting upon the flange, for the operating conditions or gasket seating as may apply (see 12-4)	W_{m1}	= minimum required bolt load for the operating conditions [see 2-5(c)]. For flange pairs used to contain a tubesheet for a floating head or a U-tube type of heat exchangers, or for any other similar design, W_{m1} shall be the larger of the values as individually calculated for each flange, and that value shall be used for both flanges.
M_T	= component of moment due to H_T = $H_T h_T$	W_{m2}	= minimum required bolt load for gasket seating [see 2-5(c)]. For flange pairs used to contain a tubesheet for a floating head or U-tube type of heat exchanger, or for any other similar design where the flanges or gaskets are not the same, W_{m2} shall be the larger of the values calculated for each flange and that value shall be used for both flanges.
N	= width used to determine the basic gasket seating with b_o , based upon the possible contact width of the gasket (see Table 2-5.2)	Y	= factor involving K (from Figure 2-7.1)
P	= internal design pressure (see UG-21). For flanges subject to external design pressure, see 2-11 .	y	= gasket or joint-contact-surface unit seating load, [see Note 1, 2-5(c)]
R	= radial distance from bolt circle to point of intersection of hub and back of flange. For integral and hub flanges, = $\frac{C-B}{2} - g_1$	Z	= factor involving K (from Figure 2-7.1)
S_a	= allowable bolt stress at atmospheric temperature (see UG-23)		
S_b	= allowable bolt stress at design temperature (see UG-23)		
S_f	= allowable design stress for material of flange at design temperature (operating condition) or atmospheric temperature (gasket seating), as may apply (see UG-23)		

2-4 CIRCULAR FLANGE TYPES

(17)

For purposes of computation, there are three types:

(a) *Loose Type Flanges*. This type covers those designs in which the flange has no direct connection to the nozzle neck, vessel, or pipe wall, and designs where the method of attachment is not considered to give the mechanical strength equivalent of integral attachment. See [Figure 2-4](#) sketches (1), (1a), (2), (2a), (3), (3a), (4), (4a), (4b), and (4c) for typical loose type flanges and the location of the loads and moments. Welds and other details of

construction shall satisfy the dimensional requirements given in [Figure 2-4](#) sketches (1), (1a), (2), (2a), (3), (3a), (4), (4a), (4b), and (4c).

(b) *Integral Type Flanges.* This type covers designs where the flange is cast or forged integrally with the nozzle neck, vessel or pipe wall, butt welded thereto, or attached by other forms of arc or gas welding of such a nature that the flange and nozzle neck, vessel or pipe wall is considered to be the equivalent of an integral structure. In welded construction, the nozzle neck, vessel, or pipe wall is considered to act as a hub. See [Figure 2-4](#) sketches (5), (6), (6a), (6b), and (7) for typical integral type flanges and the location of the loads and moments. Welds and other details of construction shall satisfy the dimensional requirements given in [Figure 2-4](#) sketches (5), (6), (6a), (6b), and (7).

(c) *Optional Type Flanges.* This type covers designs where the attachment of the flange to the nozzle neck, vessel, or pipe wall is such that the assembly is considered to act as a unit, which shall be calculated as an integral flange, except that for simplicity the designer may calculate the construction as a loose type flange, provided none of the following values is exceeded:

$$g_o = \frac{5}{8} \text{ in. (16 mm)}$$

$$B/g_o = 300$$

$$P = 300 \text{ psi (2 MPa)}$$

operating temperature = 700°F (370°C)

See [Figure 2-4](#) sketches (8), (8a), (9), (9a), (10), (10a), and (11) for typical optional type flanges. Welds and other details of construction shall satisfy the dimensional requirements given in [Figure 2-4](#) sketches (8), (8a), (9), (9a), (10), (10a), and (11).

2-5 BOLT LOADS

(a) General Requirements

(1) In the design of a bolted flange connection, calculations shall be made for each of the two design conditions of operating and gasket seating, and the more severe shall control.

(2) In the design of flange pairs used to contain a tubesheet of a heat exchanger or any similar design where the flanges and/or gaskets may not be the same, loads must be determined for the most severe condition of operating and/or gasket seating loads applied to each side at the same time. This most severe condition may be gasket seating on one flange with operating on the other, gasket seating on each flange at the same time, or operating on each flange at the same time. Although no specific rules are given for the design of the flange pairs, after the loads for the most severe conditions are determined, calculations shall be made for each flange following the rules of [Mandatory Appendix 2](#).

(3) Recommended minimum gasket contact widths for sheet and composite gaskets are provided in [Table 2-4](#).

(b) Design Conditions

(1) *Operating Conditions.* The conditions required to resist the hydrostatic end force of the design pressure tending to part the joint, and to maintain on the gasket or joint-contact surface sufficient compression to assure a tight joint, all at the design temperature. The minimum load is a function of the design pressure, the gasket material, and the effective gasket or contact area to be kept tight under pressure, per [eq. \(c\)\(1\)\(1\)](#) below, and determines one of the two requirements for the amount of the bolting A_{m1} . This load is also used for the design of the flange, per [eq. \(d\)\(3\)](#) below.

(2) *Gasket Seating.* The conditions existing when the gasket or joint-contact surface is seated by applying an initial load with the bolts when assembling the joint, at atmospheric temperature and pressure. The minimum initial load considered to be adequate for proper seating is a function of the gasket material, and the effective gasket or contact area to be seated, per [eq. \(c\)\(2\)\(2\)](#) below, and determines the other of the two requirements for the amount of bolting A_{m2} . For the design of the flange, this load is modified per [eq. \(e\)\(4\)](#) below to take account of the operating conditions, when these govern the amount of bolting required A_m , as well as the amount of bolting actually provided A_b .

(c) *Required Bolt Loads.* The flange bolt loads used in calculating the required cross-sectional area of bolts shall be determined as follows.

(1) The required bolt load for the operating conditions W_{m1} shall be sufficient to resist the hydrostatic end force H exerted by the maximum allowable working pressure on the area bounded by the diameter of gasket reaction, and, in addition, to maintain on the gasket or joint-contact surface a compression load H_p , which experience has shown to be sufficient to ensure a tight joint. (This compression load is expressed as a multiple m of the internal pressure. Its value is a function of the gasket material and construction.)

NOTE: [Tables 2-5.1](#) and [2-5.2](#) give a list of many commonly used gasket materials and contact facings, with suggested values of m , b , and y that have proved satisfactory in actual service. These values are suggested only and are not mandatory.

The required bolt load for the operating conditions W_{m1} is determined in accordance with [eq. \(1\)](#).

$$W_{m1} = H + H_p = 0.785G^2P + [2b \times 3.14GmP] \quad (1)$$

(2) Before a tight joint can be obtained, it is necessary to seat the gasket or joint-contact surface properly by applying a minimum initial load (under atmospheric temperature conditions without the presence of internal pressure), which is a function of the gasket material and

Figure 2-4
Types of Flanges

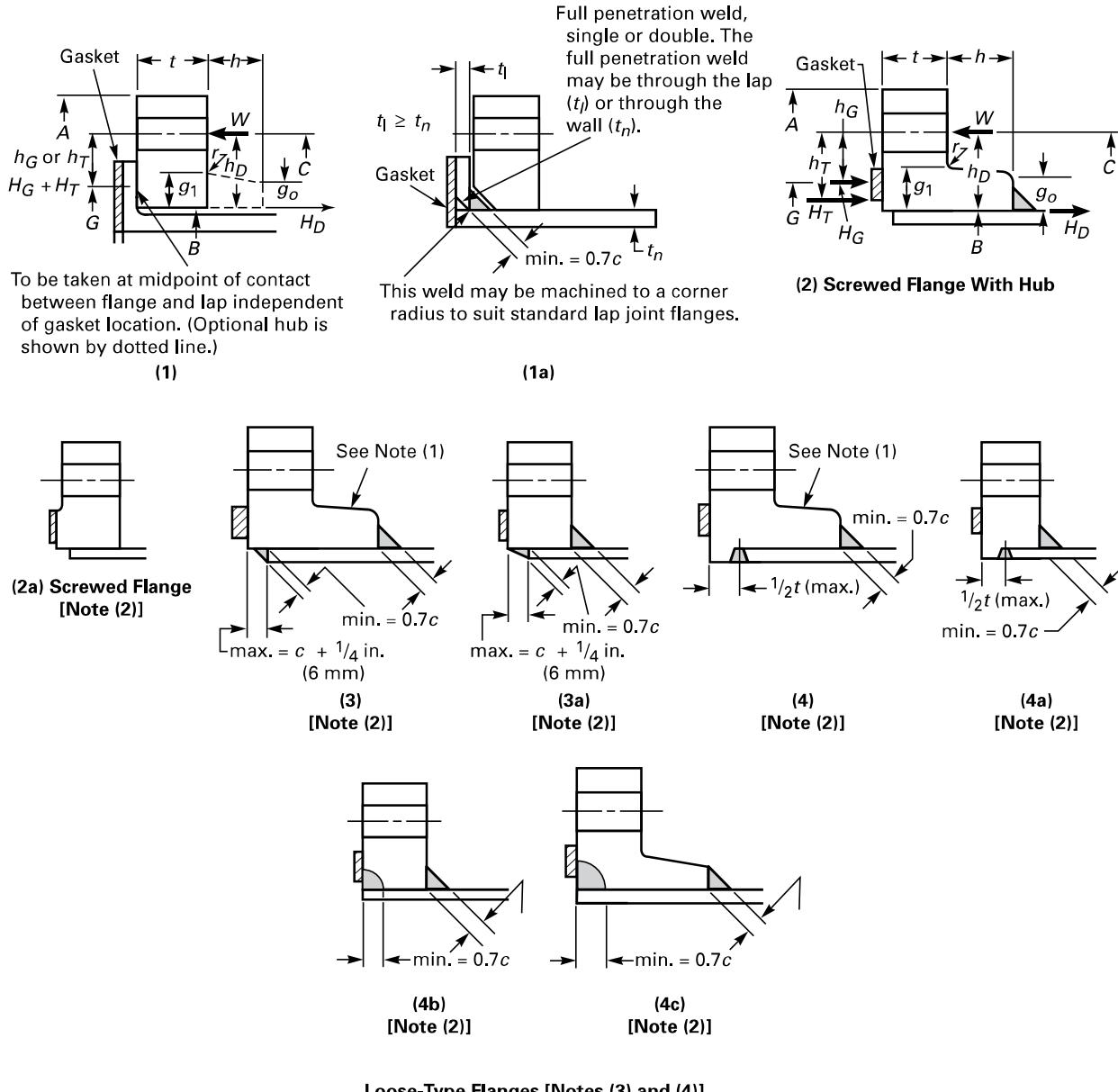
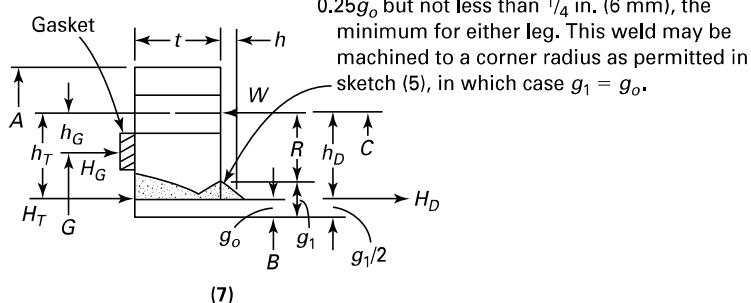
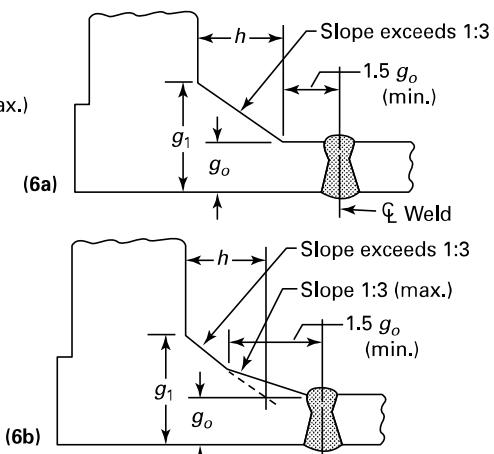
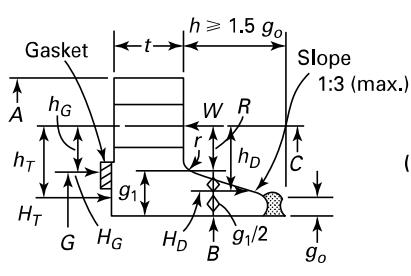
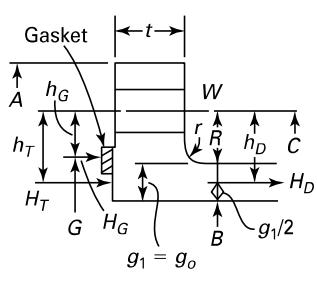
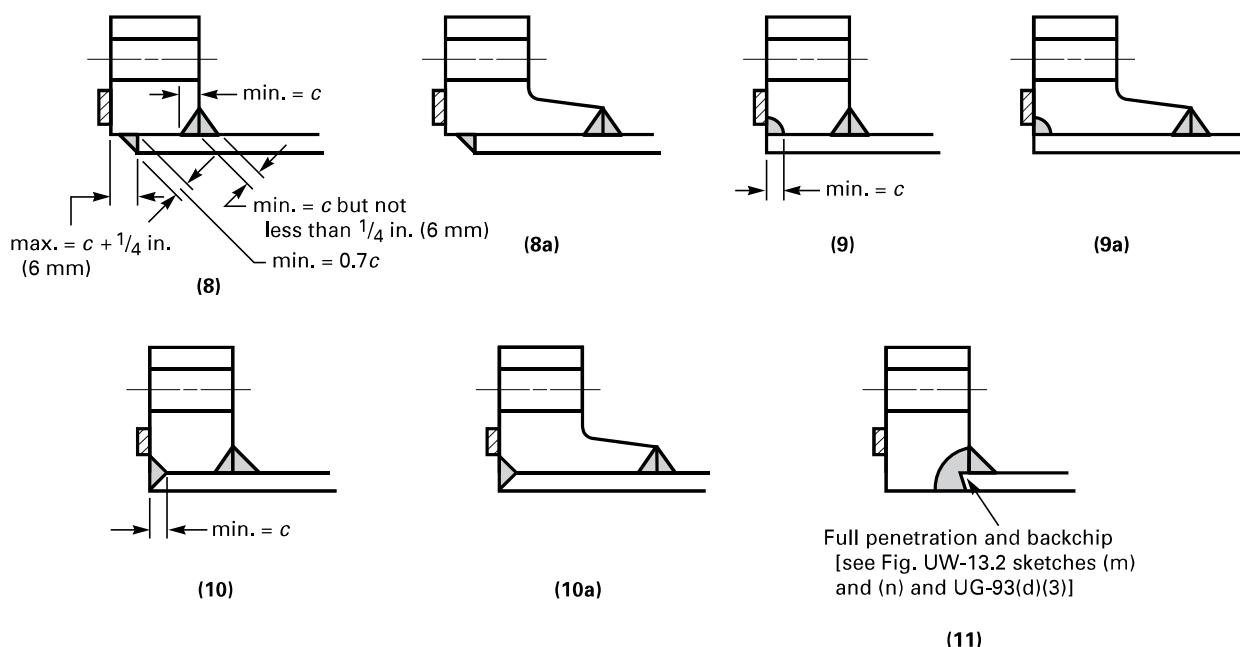


Figure 2-4
Types of Flanges (Cont'd)

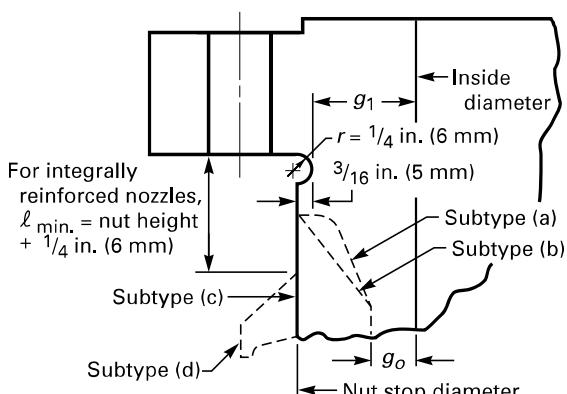


Integral-Type Flanges [Notes (3) and (4)]

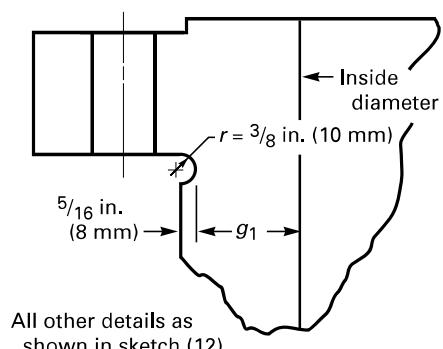
Figure 2-4
Types of Flanges (Cont'd)



Optional-Type Flanges [Notes (5), (6), and (7)]



(12) For Flanged Nozzles 18 in. (460 mm) and Smaller Nominal Size



(12a) For Flanged Nozzles Over 18 in. (460 mm) Nominal Size

Flanges With Nut Stops [Note (8)]

NOTES:

- (1) For hub tapers 6 deg or less, use $g_o = g_1$.
- (2) Loading and dimensions for sketches (2a), (3), (3a), (4), (4a), (4b), and (4c) not shown are the same as for sketch (2).
- (3) Fillet radius r to be at least $0.25 g_1$ but not less than $\frac{3}{16}$ in. (5 mm).

Figure 2-4
Types of Flanges (Cont'd)

NOTES (CONT'D):

- (4) Facing thicknesses or groove depths greater than $\frac{1}{16}$ in. (1.5 mm) shall be in excess of the required minimum flange thickness, t ; those equal to or less than $\frac{1}{16}$ in. (1.5 mm) may be included in the overall flange thickness.
- (5) Optional-type flanges may be calculated as either loose or integral type. See 2-4.
- (6) Loadings and dimensions not shown in sketches (8), (8a), (9), (9a), (10), and (10a) are the same as shown in sketch (2) when the flange is calculated as a loose-type flange, and as shown in sketch (7) when the flange is calculated as an integral-type flange.
- (7) The groove and fillet welds between the flange back face and the shell given in sketch (8) also apply to sketches (8a), (9), (9a), (10), and (10a).
- (8) For subtypes (a) and (b), g_o is the thickness of the hub at the small end. For subtypes (c) and (d), $g_o = g_1$.

the effective gasket area to be seated. The minimum initial bolt load required for this purpose W_{m2} shall be determined in accordance with eq. (2).

$$W_{m2} = 3.14bGy \quad (2)$$

The need for providing sufficient bolt load to seat the gasket or joint-contact surfaces in accordance with eq. (2) will prevail on many low-pressure designs and with facings and materials that require a high seating load, and where the bolt load computed by eq. (1)(1) for the operating conditions is insufficient to seat the joint. Accordingly, it is necessary to furnish bolting and to pretighten the bolts to provide a bolt load sufficient to satisfy both of these requirements, each one being individually investigated. When eq. (2) governs, flange proportions will be a function of the bolting instead of internal pressure.

(3) Bolt loads for flanges using gaskets of the self-energizing type differ from those shown above.

(-a) The required bolt load for the operating conditions W_{m1} shall be sufficient to resist the hydrostatic end force H exerted by the maximum allowable working pressure on the area bounded by the outside diameter of the gasket. H_p is to be considered as 0 for all self-energizing gaskets except certain seal configurations which generate axial loads which must be considered.

(-b) $W_{m2} = 0$.

Self-energizing gaskets may be considered to require an inconsequential amount of bolting force to produce a seal. Bolting, however, must be pretightened to provide a bolt load sufficient to withstand the hydrostatic end force H .

(d) *Total Required and Actual Bolt Areas, A_m and A_b .* The total cross-sectional area of bolts A_m required for both the operating conditions and gasket seating is the greater of the values for A_{m1} and A_{m2} , where $A_{m1} = W_{m1}/S_b$ and $A_{m2} = W_{m2}/S_a$. A selection of bolts to be used shall be made such that the actual total cross-sectional area of bolts A_b will not be less than A_m . For vessels in lethal service or when specified by the user or his designated agent, the maximum bolt spacing shall not exceed the value calculated in accordance with eq. (3).

$$B_{s\max} = 2a + \frac{6t}{m + 0.5} \quad (3)$$

(e) *Flange Design Bolt Load W .* The bolt loads used in the design of the flange shall be the values obtained from eqs. (4) and (5). For operating conditions,

$$W = W_{m1} \quad (4)$$

For gasket seating,

$$W = \frac{(A_m + A_b)S_a}{2} \quad (5)$$

Table 2-4
Recommended Minimum Gasket Contact Widths for Sheet and Composite Gaskets

Flange ID	Gasket Contact Width
24 in. (600 mm) < ID ≤ 36 in. (900 mm)	1 in. (25 mm)
36 in. (900 mm) < ID < 60 in. (1500 mm)	1 $\frac{1}{4}$ in. (32 mm)
ID ≥ 60 in. (1500 mm)	1 $\frac{1}{2}$ (38 mm)

S_a used in eq. (5) shall be not less than that tabulated in the stress tables (see UG-23). In addition to the minimum requirements for safety, eq. (5) provides a margin against abuse of the flange from overbolting. Since the margin against such abuse is needed primarily for the initial, bolting-up operation which is done at atmospheric temperature and before application of internal pressure, the flange design is required to satisfy this loading only under such conditions.

Table 2-5.1
Gasket Materials and Contact Facings
Gasket Factors m for Operating Conditions and Minimum Design Seating Stress y

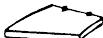
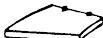
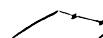
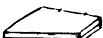
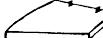
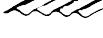
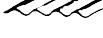
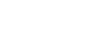
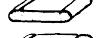
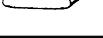
Gasket Material	Gasket Factor m	Min. Design Seating Stress y , psi (MPa)	Sketches	Facing Sketch and Column in Table 2-5.2
Self-energizing types (O-rings, metallic, elastomer, other gasket types considered as self-sealing)	0	0 (0)
Elastomers without fabric or high percent of mineral fiber:				
Below 75A Shore Durometer	0.50	0 (0)		(1a), (1b), (1c), (1d), (4), (5); Column II
75A or higher Shore Durometer	1.00	200 (1.4)		(1a), (1b), (1c), (1d), (4), (5); Column II
Mineral fiber with suitable binder for operating conditions:				
$\frac{1}{8}$ in. (3.2 mm) thick	2.00	1,600 (11)		(1a), (1b), (1c), (1d), (4), (5); Column II
$\frac{1}{16}$ in. (1.6 mm) thick	2.75	3,700 (26)		(1a), (1b), (1c), (1d), (4), (5); Column II
$\frac{1}{32}$ in. (0.8 mm) thick	3.50	6,500 (45)		(1a), (1b), (1c), (1d), (4), (5); Column II
Elastomers with cotton fabric insertion	1.25	400 (2.8)		(1a), (1b), (1c), (1d), (4), (5); Column II
Elastomers with mineral fiber fabric insertion (with or without wire reinforcement):				
3-ply	2.25	2,200 (15)		(1a), (1b), (1c), (1d), (4), (5); Column II
2-ply	2.50	2,900 (20)		(1a), (1b), (1c), (1d), (4), (5); Column II
1-ply	2.75	3,700 (26)		(1a), (1b), (1c), (1d), (4), (5); Column II
Vegetable fiber	1.75	1,100 (7.6)		(1a), (1b), (1c), (1d), (4), (5); Column II
Spiral-wound metal, mineral fiber filled:				
Carbon	2.50	10,000 (69)		(1a), (1b); Column II
Stainless, Monel, and nickel-base alloys	3.00	10,000 (69)		(1a), (1b); Column II
Corrugated metal, mineral fiber inserted, or corrugated metal, jacketed mineral fiber filled:				
Soft aluminum	2.50	2,900 (20)		(1a), (1b); Column II
Soft copper or brass	2.75	3,700 (26)		(1a), (1b); Column II
Iron or soft steel	3.00	4,500 (31)		(1a), (1b); Column II
Monel or 4-6% chrome	3.25	5,500 (38)		(1a), (1b); Column II
Stainless steels and nickel-base alloys	3.50	6,500 (45)		(1a), (1b); Column II
Corrugated metal:				
Soft aluminum	2.75	3,700 (26)		(1a), (1b), (1c), (1d); Column II
Soft copper or brass	3.00	4,500 (31)		(1a), (1b), (1c), (1d); Column II
Iron or soft steel	3.25	5,500 (38)		(1a), (1b), (1c), (1d); Column II
Monel or 4-6% chrome	3.50	6,500 (45)		(1a), (1b), (1c), (1d); Column II
Stainless steels and nickel-base alloys	3.75	7,600 (52)		(1a), (1b), (1c), (1d); Column II
Flat metal, jacketed mineral fiber filled:				
Soft aluminum	3.25	5,500 (38)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II
Soft copper or brass	3.50	6,500 (45)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II
Iron or soft steel	3.75	7,600 (52)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II
Monel	3.50	8,000 (55)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II
4-6% chrome	3.75	9,000 (62)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II
Stainless steels and nickel-base alloys	3.75	9,000 (62)		(1a), (1b), (1c), (1d); [Note (1)], (2); Column II

Table 2-5.1
Gasket Materials and Contact Facings
Gasket Factors m for Operating Conditions and Minimum Design Seating Stress y (Cont'd)

Gasket Material	Gasket Factor m	Min. Design Seating Stress y , psi (MPa)	Sketches	Facing Sketch and Column in Table 2-5.2
Grooved metal:				
Soft aluminum	3.25	5,500 (38)		(1a), (1b), (1c), (1d), (2), (3); Column II
Soft copper or brass	3.50	6,500 (45)		
Iron or soft metal	3.75	7,600 (52)		
Monel or 4-6% chrome	3.75	9,000 (62)		
Stainless steels and nickel-base alloys	4.25	10,100 (70)		
Solid flat metal:				
Soft aluminum	4.00	8,800 (61)		(1a), (1b), (1c), (1d), (2), (3), (4), (5); Column I
Soft copper or brass	4.75	13,000 (90)		
Iron or soft steel	5.50	18,000 (124)		
Monel or 4-6% chrome	6.00	21,800 (150)		
Stainless steels and nickel-base alloys	6.50	26,000 (180)		
Ring joint:				(6); Column I
Iron or soft steel	5.50	18,000 (124)		
Monel or 4-6% chrome	6.00	21,800 (150)		
Stainless steels and nickel-base alloys	6.50	26,000 (180)		

GENERAL NOTE: This Table gives a list of many commonly used gasket materials and contact facings with suggested design values of m and y that have generally proved satisfactory in actual service when using effective gasket seating width b given in Table 2-5.2. The design values and other details given in this Table are suggested only and are not mandatory.

NOTE:

(1) The surface of a gasket having a lap should not be against the nubbin.

NOTE: Where additional safety against abuse is desired, or where it is necessary that the flange be suitable to withstand the full available bolt load $A_b S_a$, the flange may be designed on the basis of this latter quantity.

For gasket seating, the total flange moment M_o is based on the flange design bolt load of eq. 2-5(e)(5), which is opposed only by the gasket load, in which case

$$M_o = W \frac{(C-G)}{2} \quad (6)$$

2-6 FLANGE MOMENTS

In the calculation of flange stress, the moment of a load acting on the flange is the product of the load and its moment arm. The moment arm is determined by the relative position of the bolt circle with respect to that of the load producing the moment (see Figure 2-4). No consideration shall be given to any possible reduction in moment arm due to cupping of the flanges or due to inward shifting of the line of action of the bolts as a result thereof. It is recommended that the value of $h_G [(C-G)/2]$ be kept to a minimum to reduce flange rotation at the sealing surface.

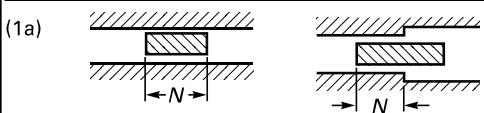
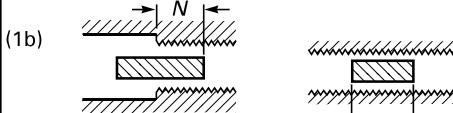
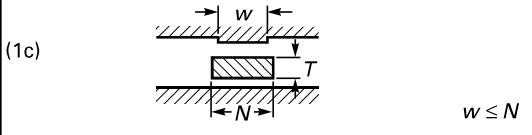
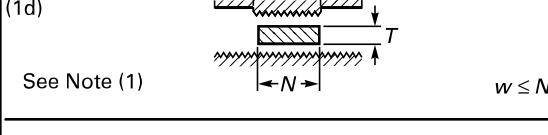
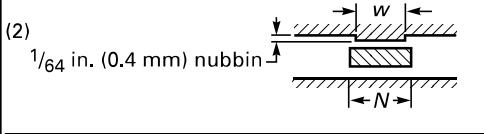
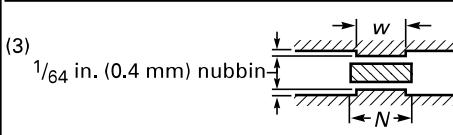
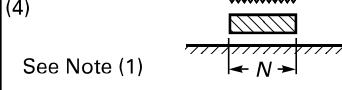
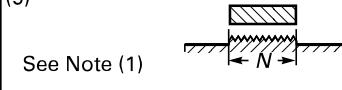
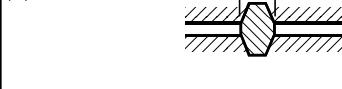
For the operating conditions, the total flange moment M_o is the sum of the three individual moments M_D , M_T , and M_G as defined in 2-3 and based on the flange design load of eq. 2-5(e)(4) with moment arms as given in Table 2-6.

For vessels in lethal service or when specified by the user or his designated agent, the bolt spacing correction shall be applied in calculating the flange stress in 13-1, 2-13(c), and 2-13(d). The flange moment M_o without correction for bolt spacing is used for the calculation of the rigidity index in 2-14.

When the bolt spacing exceeds $2a + t$, multiply M_o by the bolt spacing correction factor B_{SC} for calculating flange stress, where

$$B_{SC} = \sqrt{\frac{B_s}{2a+t}} \quad (7)$$

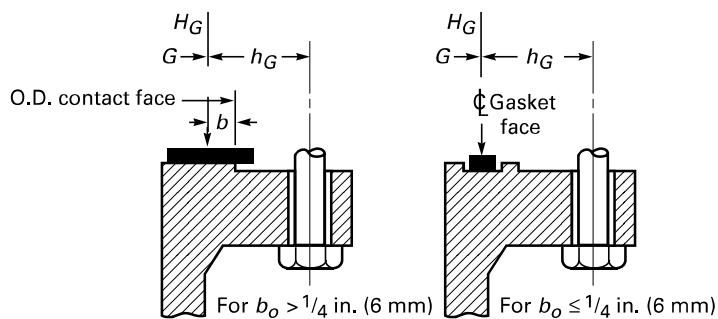
Table 2-5.2
Effective Gasket Width

Facing Sketch (Exaggerated)	Basic Gasket Seating Width, b_o	
	Column I	Column II
(1a) 		
(1b)  See Note (1)	$\frac{N}{2}$	$\frac{N}{2}$
(1c) 	$w + T; \left(\frac{w+N}{4} \max \right)$	$\frac{w+T}{2}; \left(\frac{w+N}{4} \max \right)$
(1d)  See Note (1)	$\frac{w+T}{2}; \left(\frac{w+N}{4} \max \right)$	$\frac{w+T}{2}; \left(\frac{w+N}{4} \max \right)$
(2) 	$\frac{w+N}{4}$	$\frac{w+3N}{8}$
(3) 	$\frac{N}{4}$	$\frac{3N}{8}$
(4)  See Note (1)	$\frac{3N}{8}$	$\frac{7N}{16}$
(5)  See Note (1)	$\frac{N}{4}$	$\frac{3N}{8}$
(6) 	$\frac{w}{8}$...

Effective Gasket Seating Width, b
 $b = b_o$, when $b_o \leq \frac{1}{4}$ in. (6 mm); $b = C_b \sqrt{b_o}$, when $b_o > \frac{1}{4}$ in. (6 mm)

**Table 2-5.2
Effective Gasket Width (Cont'd)**

Location of Gasket Load Reaction



GENERAL NOTE: The gasket factors listed only apply to flanged joints in which the gasket is contained entirely within the inner edges of the bolt holes.

NOTE:

(1) Where serrations do not exceed $\frac{1}{64}$ in. (0.4 mm) depth and $\frac{1}{32}$ in. (0.8 mm) width spacing, sketches (1b) and (1d) shall be used.

2-7 CALCULATION OF FLANGE STRESSES

The stresses in the flange shall be determined for both the operating conditions and gasket seating condition, whichever controls, in accordance with the following equations:

(a) for integral type flanges [Figure 2-4 sketches (5), (6), (6a), (6b), and (7)], for optional type flanges calculated as integral type [Figure 2-4 sketches (8), (8a), (9), (9a), (10), (10a), and (11)], and for loose type flanges with a hub which is considered [Figure 2-4 sketches (1), (1a), (2), (2a), (3), (3a), (4), (4a), (4b), and (4c)];

Longitudinal hub stress

$$S_H = \frac{f M_o}{L g_1^2 B} \quad (8)$$

Radial flange stress

$$S_R = \frac{(1.33te + 1)M_o}{Lt^2 B} \quad (9)$$

**Table 2-6
Moment Arms for Flange Loads Under Operating Conditions**

	h_D	h_T	h_G
Integral-type flanges [see Figure 2-4 sketches (5), (6), (6a), (6b), and (7)] and optional type flanges calculated as integral type [see Figure 2-4 sketches (8), (8a), (9), (9a), (10), (10a), and (11)]	$R + 0.5g_1$	$\frac{R + g_1 + h_G}{2}$	$\frac{C - G}{2}$
Loose type, except lap-joint flanges [see Figure 2-4 sketches (2), (2a), (3), (3a), (4), and (4a)]; and optional type flanges calculated as loose type [see Figure 2-4 sketches (8), (8a), (9), (9a), (10), (10a), and (11)]	$\frac{C - B}{2}$	$\frac{h_D + h_G}{2}$	$\frac{C - G}{2}$
Lap-type flanges [see Figure 2-4 sketches (1) and (1a)]	$\frac{C - B}{2}$	$\frac{C - G}{2}$	$\frac{C - G}{2}$

Tangential flange stress

$$S_T = \frac{Y M_o}{t^2 B} - Z S_R \quad (10)$$

(b) for loose type flanges without hubs and loose type flanges with hubs which the designer chooses to calculate without considering the hub [Figure 2-4 sketches (1), (1a), (2), (2a), (3), (3a), (4), (4a), (4b), and (4c)] and optional type flanges calculated as loose type [Figure 2-4 sketches (8), (8a), (9), (9a), (10), (10a), and (11)]:

$$\begin{aligned} S_T &= \frac{Y M_o}{t^2 B} \\ S_R &= 0 \\ S_H &= 0 \end{aligned} \quad (11)$$

(1) longitudinal hub stress S_H not greater than S_f for cast iron⁹¹ and, except as otherwise limited by (-a) and (-b) below, not greater than $1.5S_f$ for materials other than cast iron:

(-a) longitudinal hub stress S_H not greater than the smaller of $1.5S_f$ or $1.5S_n$ for optional type flanges designed as integral [Figure 2-4 sketches (8), (8a), (9), (9a), (10), (10a), and (11)], also integral type [Figure 2-4 sketch (7)] where the neck material constitutes the hub of the flange;

(-b) longitudinal hub stress S_H not greater than the smaller of $1.5S_f$ or $2.5S_n$ for integral type flanges with hub welded to the neck, pipe or vessel wall [Figure 2-4 sketches (6), (6a), and (6b)].

(2) radial flange stress S_R not greater than S_f ;

(3) tangential flange stress S_T not greater than S_f ;

(4) also $(S_H + S_R)/2$ not greater than S_f and $(S_H + S_T)/2$ not greater than S_f .

(b) For hub flanges attached as shown in Figure 2-4 sketches (2), (2a), (3), (3a), (4), (4a), (4b), and (4c), the nozzle neck, vessel or pipe wall shall not be considered to have any value as a hub.

2-8 ALLOWABLE FLANGE DESIGN STRESSES

(a) The flange stresses calculated by the equations in 13-1 shall not exceed the following values:

Figure 2-7.1
Values of T , U , Y , and Z (Terms Involving K)

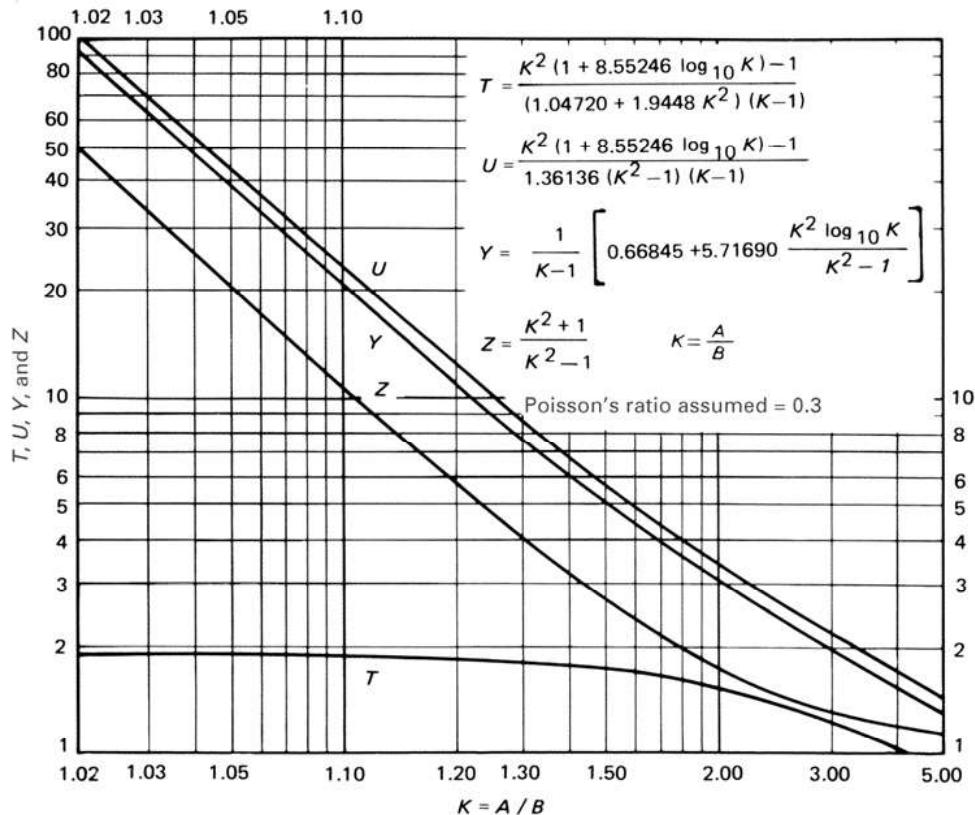
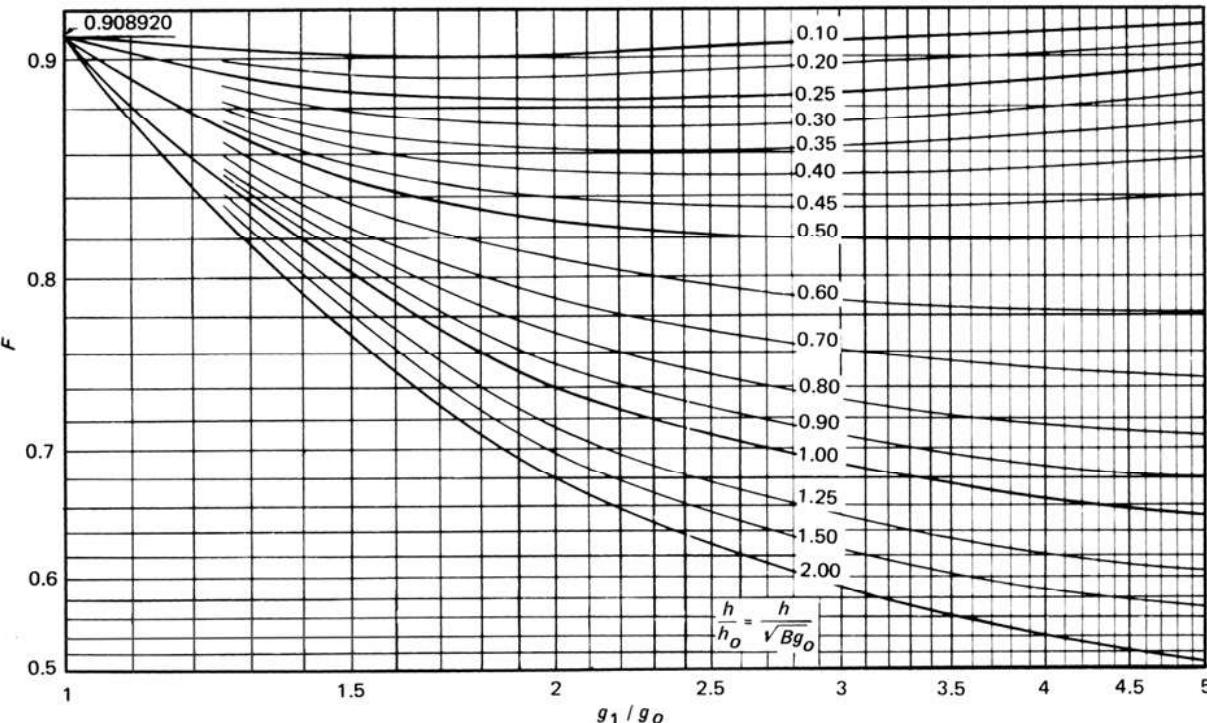


Figure 2-7.2
Values of F (Integral Flange Factors)



GENERAL NOTE: See Table 2-7.1 for equations.

(c) In the case of loose type flanges with laps, as shown in Figure 2-4 sketches (1) and (1a), where the gasket is so located that the lap is subjected to shear, the shearing stress shall not exceed $0.8 S_n$ for the material of the lap, as defined in 2-3. In the case of welded flanges, shown in Figure 2-4 sketches (3), (3a), (4), (4a), (4b), (4c), (7), (8), (8a), (9), (9a), (10), and (10a) where the nozzle neck, vessel, or pipe wall extends near to the flange face and may form the gasket contact face, the shearing stress carried by the welds shall not exceed $0.8 S_n$. The shearing stress shall be calculated on the basis of W_{m1} or W_{m2} as defined in 2-3, whichever is greater. Similar cases where flange parts are subjected to shearing stress shall be governed by the same requirements.

2-9 SPLIT LOOSE FLANGES⁹²

Loose flanges split across a diameter and designed under the rules given in this Appendix may be used under the following provisions.

(a) When the flange consists of a single split flange or flange ring, it shall be designed as if it were a solid flange (without splits), using 200% of the total moment M_o as defined in 12-4.

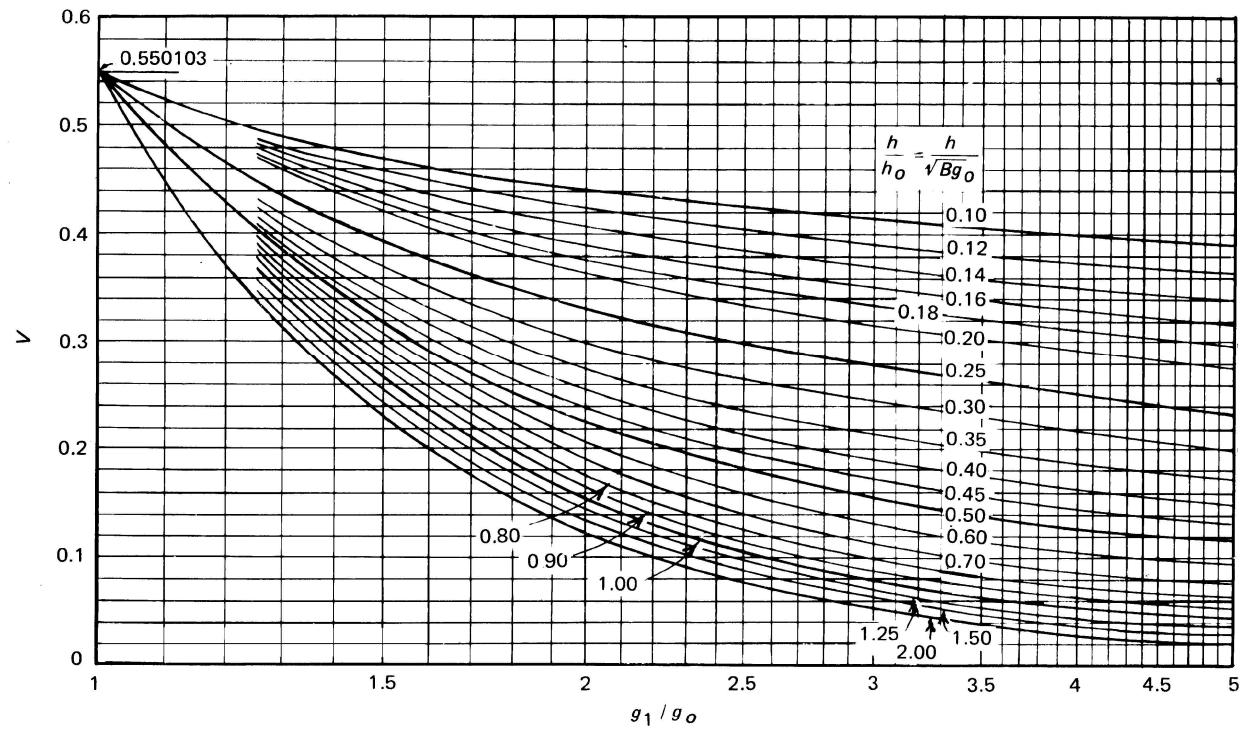
(b) When the flange consists of two split rings each ring shall be designed as if it were a solid flange (without splits), using 75% of the total moment M_o as defined in 12-4. The pair of rings shall be assembled so that the splits in one ring shall be 90 deg from the splits in the other ring.

(c) The splits should preferably be midway between bolt holes.

2-10 NONCIRCULAR SHAPED FLANGES WITH CIRCULAR BORE

The outside diameter A for a noncircular flange with a circular bore shall be taken as the diameter of the largest circle, concentric with the bore, inscribed entirely within the outside edges of the flange. Bolt loads and moments, as well as stresses, are then calculated as for circular flanges, using a bolt circle drawn through the centers of the outermost bolt holes.

Figure 2-7.3
Values of V (Integral Flange Factors)



GENERAL NOTE: See Table 2-7.1 for equations.

2-11 FLANGES SUBJECT TO EXTERNAL PRESSURES

(a) The design of flanges for external pressure only [see UG-99(f)⁹³] shall be based on the equations given in 13-1 for internal pressure except that for operating conditions:

$$M_o = H_D(h_D - h_G) + H_T(h_T - h_G) \quad (10)$$

For gasket seating,

$$M_o = Wh_G \quad (11)$$

where

$$W = \frac{A_m 2 + A_b}{2} S_a \quad (11a)$$

$$H_D = 0.785 B^2 P_e \quad (11b)$$

$$H_T = H - H_D \quad (11c)$$

$$H = 0.785 G^2 P_e \quad (11d)$$

P_e = external design pressure

See 2-3 for definitions of other symbols. S_a used in eq. (11a) shall be not less than that tabulated in the stress tables (see UG-23).

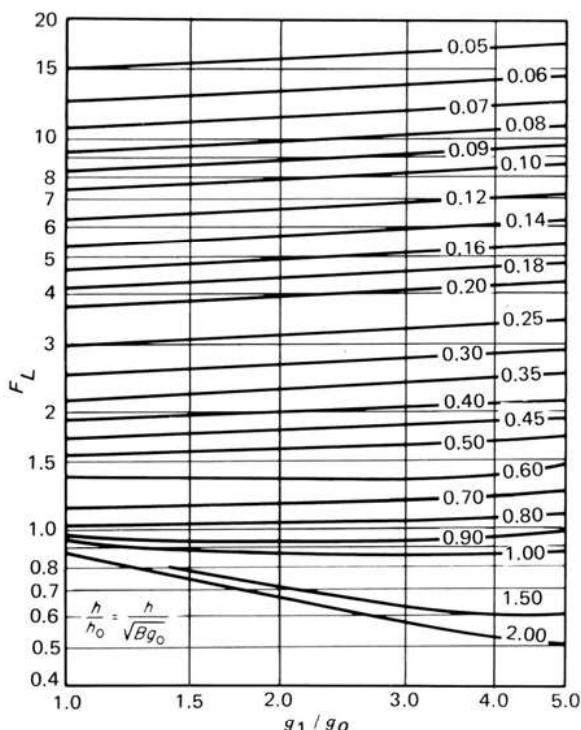
(b) When flanges are subject at different times during operation to external or internal pressure, the design shall satisfy the external pressure design requirements given in (a) above and the internal pressure design requirements given elsewhere in this Appendix.

NOTE: The combined force of external pressure and bolt loading may plastically deform certain gaskets to result in loss of gasket contact pressure when the connection is depressurized. To maintain a tight joint when the unit is repressurized, consideration should be given to gasket and facing details so that excessive deformation of the gasket will not occur. Joints subject to pressure reversals, such as in heat exchanger floating heads, are in this type of service.

2-12 FLANGES WITH NUT-STOPS

(a) When flanges are designed per this Appendix, or are fabricated to the dimensions of ASME B16.5 or other acceptable standards [see UG-44(b)], except that the dimension R is decreased to provide a nut-stop, the fillet radius relief shall be as shown in Figure 2-4 sketches (12) and (12a) except that:

Figure 2-7.4
Values of F_L (Loose Hub Flange Factors)

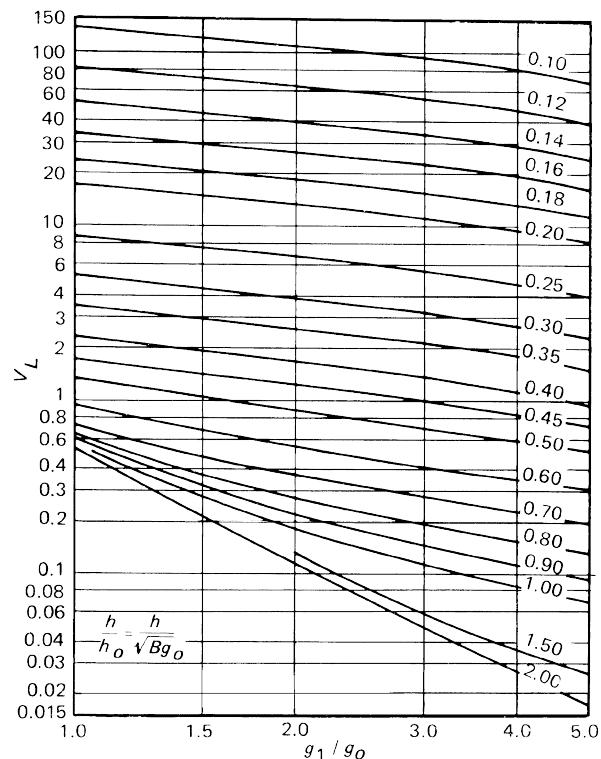


GENERAL NOTE: See Table 2-7.1 for equations.

(1) for flanges designed to this Appendix, the minimum dimension g_1 must be the lesser of $2t$ (t from UG-27) or $4r$, but in no case less than $\frac{1}{2}$ in. (13 mm), where

r = the radius of the undercut

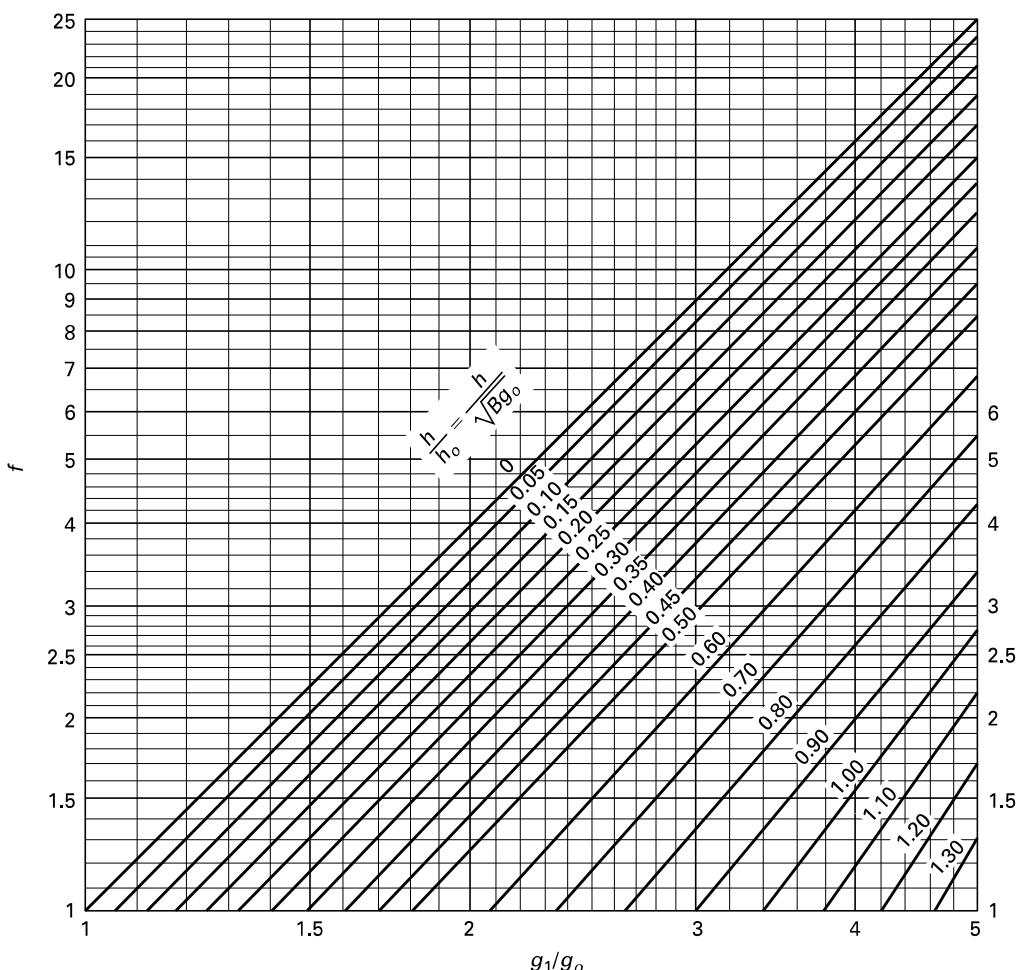
Figure 2-7.5
Values of V_L (Loose Hub Flange Factors)



GENERAL NOTE: See Table 2-7.1 for equations.

(2) for ASME B16.5 or other standard flanges, the dimension of the hub g_o shall be increased as necessary to provide a nut-stop.

Figure 2-7.6
Values of f (Hub Stress Correction Factor)



- $f = 1$ (minimum)
- $= 1$ for hubs of uniform thickness ($g_1 / g_o = 1$)
- $= 1$ for loose hubbed flanges

GENERAL NOTE: See [Table 2-7.1](#) for equations.

Table 2-7.1
Flange Factors in Formula Form

Integral Flange	Loose Hub Flange
Factor F per Figure 2-7.2 is then solved by	Factor F_L per Figure 2-7.4 is solved by
$F = -\frac{E_6}{\left(\frac{C}{2.73}\right)^{\frac{1}{4}} \frac{(1+A)^3}{C}}$	$F_L = -\frac{C_{18}\left(\frac{1}{2} + \frac{A}{6}\right) + C_{21}\left(\frac{1}{4} + \frac{11A}{84}\right) + C_{24}\left(\frac{1}{70} + \frac{A}{105}\right) - \left(\frac{1}{40} + \frac{A}{72}\right)}{\left(\frac{C}{2.73}\right)^{\frac{1}{4}} \frac{(1+A)^3}{C}}$
Factor V per Figure 2-7.3 is then solved by	Factor V_L per Figure 2-7.5 is solved by
$V = \frac{E_4}{\left(\frac{2.73}{C}\right)^{\frac{1}{4}} (1+A)^3}$	$V_L = \frac{\frac{1}{4} - \frac{C_{24}}{5} - \frac{3C_{21}}{2} - C_{18}}{\left(\frac{2.73}{C}\right)^{\frac{1}{4}} (1+A)^3}$
Factor f per Figure 2-7.6 is then solved by	Factor f per Figure 2-7.6 is set equal to 1.
$f = C_{36} / (1+A)$	$f = 1$
The values used in the above equations are solved using eqs. (1) through (45) below based on the values g_1, g_o, h , and h_o as defined by 2-3 . When $g_1 = g_o$, $F = 0.908920$, $V = 0.550103$, and $f = 1$; thus eqs. (1) through (45) need not be solved.	The values used in the above equations are solved using eqs. (1) through (5), (7), (9), (10), (12), (14), (16), (18), (20), (23), and (26) below based on the values of g_1, g_o, h , and h_o as defined by 2-3 .
Equations	
(1) $A = (g_1/g_o) - 1$	
(2) $C = 43.68(h/h_o)^4$	
(3) $C_1 = 1/3 + A/12$	
(4) $C_2 = 5/42 + 17A/336$	
(5) $C_3 = 1/210 + A/360$	
(6) $C_4 = 11/360 + 59A/5040 + (1+3A)/C$	
(7) $C_5 = 1/90 + 5A/1008 - (1+A)^3/C$	
(8) $C_6 = 1/120 + 17A/5040 + 1/C$	
(9) $C_7 = 215/2772 + 51A/1232 + (60/7 + 225A/14 + 75A^2/7 + 5A^3/2)/C$	
(10) $C_8 = 31/6930 + 128A/45,045 + (6/7 + 15A/7 + 12A^2/7 + 5A^3/11)/C$	
(11) $C_9 = 533/30,240 + 653A/73,920 + (1/2 + 33A/14 + 39A^2/28 + 25A^3/84)/C$	
(12) $C_{10} = 29/3780 + 3A/704 - (1/2 + 33A/14 + 81A^2/28 + 13A^3/12)/C$	
(13) $C_{11} = 31/6048 + 1763A/665,280 + (1/2 + 6A/7 + 15A^2/28 + 5A^3/42)/C$	
(14) $C_{12} = 1/2925 + 71A/300,300 + (8/35 + 18A/35 + 156A^2/385 + 6A^3/55)/C$	
(15) $C_{13} = 761/831,600 + 937A/1,663,200 + (1/35 + 6A/35 + 11A^2/70 + 3A^3/70)/C$	
(16) $C_{14} = 197/415,800 + 103A/332,640 - (1/35 + 6A/35 + 17A^2/70 + A^3/10)/C$	
(17) $C_{15} = 233/831,600 + 97A/554,400 + (1/35 + 3A/35 + A^2/14 + 2A^3/105)/C$	
(18) $C_{16} = C_1 C_7 C_{12} + C_2 C_8 C_3 + C_3 C_8 C_2 - (C_3^2 C_7 + C_8^2 C_1 + C_2^2 C_{12})$	
(19) $C_{17} = [C_4 C_7 C_{12} + C_2 C_8 C_{13} + C_3 C_8 C_9 - (C_{13} C_7 C_3 + C_8^2 C_4 + C_{12} C_2 C_9)]/C_{16}$	
(20) $C_{18} = [C_5 C_7 C_{12} + C_2 C_8 C_{14} + C_3 C_8 C_{10} - (C_{14} C_7 C_3 + C_8^2 C_5 + C_{12} C_2 C_{10})]/C_{16}$	
(21) $C_{19} = [C_6 C_7 C_{12} + C_2 C_8 C_{15} + C_3 C_8 C_{11} - (C_{15} C_7 C_3 + C_8^2 C_6 + C_{12} C_2 C_{11})]/C_{16}$	
(22) $C_{20} = [C_1 C_9 C_{12} + C_4 C_8 C_3 + C_3 C_{13} C_2 - (C_3^2 C_9 + C_{13} C_8 C_1 + C_{12} C_4 C_2)]/C_{16}$	
(23) $C_{21} = [C_1 C_{10} C_{12} + C_5 C_8 C_3 + C_3 C_{14} C_2 - (C_3^2 C_{10} + C_{14} C_8 C_1 + C_{12} C_5 C_2)]/C_{16}$	
(24) $C_{22} = [C_1 C_{11} C_{12} + C_6 C_8 C_3 + C_3 C_{15} C_2 - (C_3^2 C_{11} + C_{15} C_8 C_1 + C_{12} C_6 C_2)]/C_{16}$	
(25) $C_{23} = [C_1 C_7 C_{13} + C_2 C_9 C_3 + C_4 C_8 C_2 - (C_3 C_7 C_4 + C_8 C_9 C_1 + C_2^2 C_{13})]/C_{16}$	
(26) $C_{24} = [C_1 C_7 C_{14} + C_2 C_{10} C_3 + C_5 C_8 C_2 - (C_3 C_7 C_5 + C_8 C_{10} C_1 + C_2^2 C_{14})]/C_{16}$	
(27) $C_{25} = [C_1 C_7 C_{15} + C_2 C_{11} C_3 + C_6 C_8 C_2 - (C_3 C_7 C_6 + C_8 C_{11} C_1 + C_2^2 C_{15})]/C_{16}$	
(28) $C_{26} = -(C/4)^{1/4}$	
(29) $C_{27} = C_{20} - C_{17} - 5/12 + C_{17} C_{26}$	
(30) $C_{28} = C_{22} - C_{19} - 1/12 + C_{19} C_{26}$	
(31) $C_{29} = -(C/4)^{1/2}$	
(32) $C_{30} = -(C/4)^{3/4}$	
(33) $C_{31} = 3A/2 - C_{17} C_{30}$	
(34) $C_{32} = 1/2 - C_{19} C_{30}$	
(35) $C_{33} = 0.5C_{26} C_{32} + C_{28} C_{31} C_{29} - (0.5C_{30} C_{28} + C_{32} C_{27} C_{29})$	
(36) $C_{34} = 1/12 + C_{18} - C_{21} - C_{18} C_{26}$	
(37) $C_{35} = -C_{18}(C/4)^{3/4}$	
(38) $C_{36} = (C_{28} C_{35} C_{29} - C_{32} C_{34} C_{29})/C_{33}$	
(39) $C_{37} = [0.5C_{26} C_{35} + C_{34} C_{31} C_{29} - (0.5C_{30} C_{34} + C_{35} C_{27} C_{29})]/C_{33}$	
(40) $E_1 = C_{17} C_{36} + C_{18} + C_{19} C_{37}$	

Table 2-7.1
Flange Factors in Formula Form (Cont'd)

Equations (Cont'd)

- (41) $E_2 = C_{20}C_{36} + C_{21} + C_{22}C_{37}$
- (42) $E_3 = C_{23}C_{36} + C_{24} + C_{25}C_{37}$
- (43) $E_4 = 1/4 + C_{37}/12 + C_{36}/4 - E_3/5 - 3E_2/2 - E_1$
- (44) $E_5 = E_1(1/2 + A/6) + E_2(1/4 + 11A/84) + E_3(1/70 + A/105)$
- (45) $E_6 = E_5 - C_{36}(7/120 + A/36 + 3A/C) - 1/40 - A/72 - C_{37}(1/60 + A/120 + 1/C)$

2-13 REVERSE FLANGES

(a) Flanges with the configuration as indicated in [Figure 2-13.1](#) shall be designed as integral reverse flanges and those in [Figure 2-13.2](#) shall be designed as loose ring type reverse flanges. These flanges shall be designed in conformance with the rules in [2-3](#) through [2-8](#), but with the modifications as described in the following.

Mandatory use of these rules is limited to $K \leq 2$. When $K > 2$, results become increasingly conservative and [U-2\(g\)](#) may be used.

(1) *Integral Type Reverse Flange*. The shell-to-flange attachment of integral type reverse flanges may be attached as shown in [Figure 2-4](#) sketches (5) through (11), as well as [Figure UW-13.2](#) sketches (a) and (b).

Figure 2-13.1
Reverse Flange

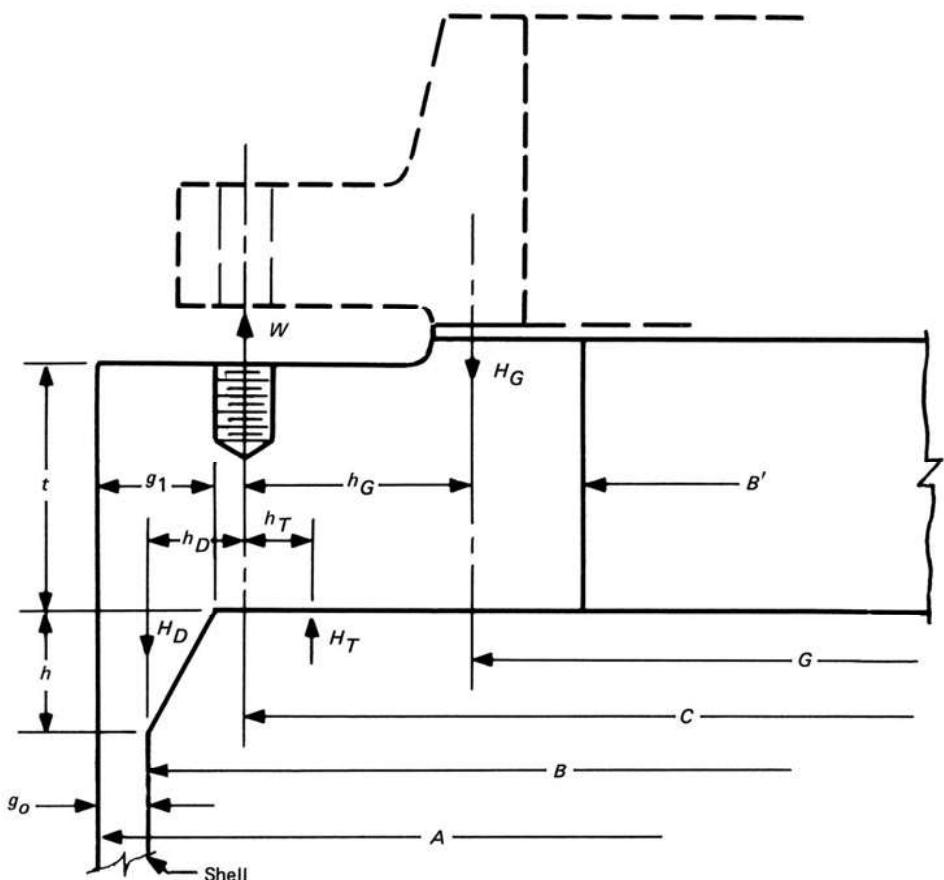
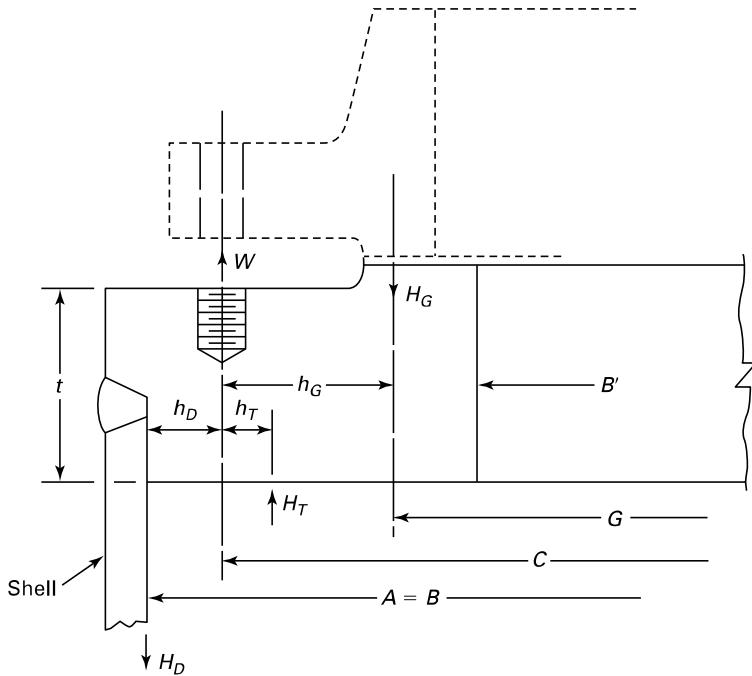


Figure 2-13.2
Loose Ring Type Reverse Flange



The requirements of 2-4(c) apply to Figure 2-4 sketches (8) through (11) as well as Figure UW-13.2 sketches (a) and (b).

(2) *Loose Ring Type Reverse Flange.* The shell-to-flange attachment of loose ring type reverse flanges may be attached as shown in Figure 2-4 sketches (3a), (4a), (8), (9), (10), and (11) as well as Figure UW-13.2 sketches (c) and (d). When Figure UW-13.2 sketches (c) and (d) are used, the maximum wall thickness of the shell shall not exceed $\frac{3}{8}$ in. (10 mm), and the maximum design metal temperature shall not exceed 650°F (340°C).

The symbols and definitions in this paragraph pertain specifically to reverse flanges. Except as noted in (b) below, the symbols used in the equations of this paragraph are defined in 2-3.

The equations for S_H , S_{R_1} , and S_{T_1} correspond, respectively, to eqs. 2-7(a)(8), 2-7(a)(9), and 2-7(a)(10), in direction, but are located at the flange *outside* diameter. The sole stress at the flange inside diameter is a tangential stress and is given by the formula for S_{T_2} .

(b) *Notation*

B = inside diameter of shell

B' = inside diameter of reverse flange

$d_r = U_r h_{or} g_o^2 / V$

$e_r = F / h_{or}$

F = factor (use h_{or} for h_o in Figure 2-7.2)

f = factor (use h_{or} for h_o in Figure 2-7.6)

H = total hydrostatic end force on attached component

$$= 0.785G^2P$$

$$H_D = \text{hydrostatic end force on area inside of flange}$$

$$= 0.785B^2P$$

$$H_T = \text{difference between hydrostatic end force on attached component and hydrostatic end force on area inside of flange}$$

$$= H - H_D$$

$$h_D = \text{radial distance from the bolt circle to the circle on which } H_D \text{ acts}$$

$$= (C + g_1 - 2g_o - B) / 2 \text{ for integral type reverse flanges}$$

$$= (C - B) / 2 \text{ for loose ring type reverse flanges}$$

$$h_{or} = \text{factor}$$

$$= \sqrt{Ag_o}$$

$$h_T = \text{radial distance from the bolt circle, to the circle on which } H_T \text{ acts}$$

$$= \frac{1}{2} \left(C - \frac{B + G}{2} \right)$$

$$K = \text{ratio of outside diameter of flange to inside diameter of flange}$$

$$= A/B'$$

$$L_r = \text{factor}$$

$$= \frac{te_r + 1}{T_r} + \frac{t^3}{d_r}$$

$$M_o = \text{total moment acting on the flange, for the operating conditions or gasket seating as may apply}$$

= algebraic sum of M_D , M_T , and M_G . Values of load H_T and moment arm h_D are negative; value of moment arm h_T may be positive as in Figure 2-13.1, or negative. If M_o is negative, use its absolute value in calculating stresses to obtain positive stresses for comparison with allowable stresses.

$$T_r = \left[\frac{Z + 0.3}{Z - 0.3} \right] \alpha_r T$$

$$U_r = \alpha_r U$$

V = factor (use h_{or} for h_o in Figure 2-7.3)

$$Y_r = \alpha_r Y$$

$$\alpha_r = \left[1 + \frac{0.668(K+1)}{Y} \right] / K^2$$

(c) For Integral Type Reverse Flanges

(1) Stresses at the Outside Diameter

$$S_H = f M_o / L_r g_1^2 B'$$

$$S_R = (1.33te_r + 1)M_o / L_r t^2 B'$$

$$S_{T1} = \left(Y_r M_o / t^2 B' \right) - Z S_R (0.67te_r + 1) / (1.33te_r + 1)$$

(2) Stress at Inside Diameter B'

$$S_{T2} = \left(M_o / t^2 B' \right) \left[Y - \frac{2K^2(1 + \frac{2}{3}te_r)}{(K^2 - 1)L_r} \right]$$

(d) For Loose Ring Type Reverse Flanges

$$S_T = YM_o / t^2 B'$$

$$S_R = 0$$

$$S_H = 0$$

(b) The notation is as follows:

E = modulus of elasticity for the flange material at design temperature (operating condition) or at atmospheric temperature (gasket seating condition), psi

J = rigidity index ≤ 1

K_I = rigidity factor for integral or optional flange types = 0.3

K_L = rigidity factor for loose-type flanges = 0.2

Experience has indicated that K_I and K_L provided above are sufficient for most services; other values may be used with the User's agreement.

Other notation is defined in 2-3 for flanges and 2-13 for reverse flanges.

(c) The rigidity criterion for an integral type flange and for a loose type flange without a hub is applicable to the reverse flanges in Figures 2-13.1 and 2-13.2, respectively. The values of h_{or} shall be substituted for h_o , and the value L_r shall be substituted for the value L in the rigidity equation for integral type flanges. Also substitute h_{or} for h_o in determining the factor V in the equation for integral type flanges.

(d) If the value of J , when calculated by the appropriate formula above, is greater than 1.0, the thickness of the flange, t , shall be increased and J recalculated until $J \leq 1$ for both gasket seating and operating conditions.

2-15 QUALIFICATION OF ASSEMBLY PROCEDURES AND ASSEMBLERS

It is recommended that flange joints designed to this Appendix be assembled by qualified procedures and by qualified assemblers. ASME PCC-1 may be used as a guide.

2-14 FLANGE RIGIDITY

(a) Flanges that have been designed based on allowable stress limits alone may not be sufficiently rigid to control leakage. This paragraph provides a method of checking flange rigidity. The rigidity factors provided in Table 2-14 have been proven through extensive user experience for a wide variety of joint design and service conditions. The use of the rigidity index does not guarantee a leakage rate within established limits. The use of the factors must be considered as only part of the system of joint design and assembly requirements to ensure leak tightness. Successful service experience may be used as an alternative to the flange rigidity rules for fluid services that are non-lethal and nonflammable and designed within the temperature range of -20°F (-29°C) to 366°F (186°C) without exceeding design pressures of 150 psi (1 035 kPa).

Flange Type	Rigidity Criterion
Integral-type flanges and optional type flanges designed as integral-type flanges	$J = \frac{52.14VM_o}{LEg_o^2Kph_o} \leq 1.0$
Loose-type flanges with hubs	$J = \frac{52.14VLM_o}{LEg_o^2K_Lh_o} \leq 1.0$
Loose-type flanges without hubs and optional flanges designed as loose-type flanges	$J = \frac{109.4M_o}{Et^3K_L(\ln K)} \leq 1.0$

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