

Central Force Motion

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Overview:

- Equation of motion for central force motion, and its standard solution.
- Deriving Kepler's Laws.
- Equation of motion for two-bodies system - reducible to one-body motion.

1 The General Solution

An object undergoes a central force motion if it is moving in a potential that depends purely on the radial coordinate of the object. i.e. $V(r, \theta, \phi) = V(r)$. The Newton's 2nd law simply writes:

$$\boxed{m \frac{d^2 \vec{r}}{dt^2} = - \frac{dV(r)}{dr} \hat{r}}$$

Such potential is guaranteed to be conservative, by showing that the curl on the force is 0.

The steps to show curl = 0 (Out of syllabus) :

1. The force only has radial component and also solely depends on the radial coordinate.

$$\begin{aligned} \vec{F}(r, \theta, \phi) &= -\vec{\nabla} V(r) \\ &\stackrel{\text{def}}{=} -\frac{\partial V(r)}{\partial r} \hat{r} + \frac{1}{r} \frac{\partial V(r)}{\partial \theta} \hat{\theta} + \frac{1}{r \sin \theta} \frac{\partial V(r)}{\partial \phi} \hat{\phi} \\ &= -\frac{\partial V(r)}{\partial r} \hat{r} + 0 + 0 \\ &= f(r) \hat{r} \end{aligned}$$

2. Then apply curl in spherical coordinate. Observe that all the terms vanish by knowing that $F_\theta = F_\phi = 0$ and F_r is a function on r only.

$$\begin{aligned} \vec{\nabla} \times \vec{F} &\stackrel{\text{def}}{=} \frac{1}{r \sin \theta} \left(\frac{\partial(F_\phi \sin \theta)}{\partial \theta} - \frac{\partial F_\theta}{\partial \phi} \right) \hat{r} + \frac{1}{r} \left(\frac{1}{\sin \theta} \frac{\partial F_r}{\partial \phi} - \frac{\partial(r F_\phi)}{\partial r} \right) \hat{\theta} + \frac{1}{r} \left(\frac{\partial(r F_\theta)}{\partial r} - \frac{\partial F_r}{\partial \theta} \right) \hat{\phi} \\ &= (\text{all terms} = 0) \end{aligned}$$

1.1 Standard Derivation

- The force from potential is the only force and is conservative \Rightarrow Energy conserves.

$$E = \frac{1}{2}m|\hat{\mathbf{v}}|^2 + V(r) = \frac{1}{2}m\left[\left(\frac{dr}{dt}\right)^2 + \left(r\frac{d\theta}{dt}\right)^2\right] + V(r) = \text{const}$$

- Force is along radial direction \Rightarrow Torque = 0 \Rightarrow Angular momentum conserves.

$$L = m\vec{r} \times \vec{v} = mr^2\frac{d\theta}{dt} = \text{const}$$

The standard rundown (*every textbook does the same*) is to treat E and L as adjustable parameters and derive the function of the object's trajectory $r(\theta)$.

1. Substitute the equation of L into equation of E and rearrange terms.

$$E = \frac{1}{2}m\left[\left(\frac{dr}{dt}\right)^2 + \left(\frac{L}{mr}\right)^2\right] + V(r)$$

$$\Rightarrow \frac{dr}{dt} = \sqrt{\frac{2}{m}\left(E - V(r) - \frac{L^2}{2mr^2}\right)}$$

2. By $L = mr^2\frac{d\theta}{dt} \Rightarrow dt = \frac{mr^2}{L}d\theta$, then move all terms that contain r to one side.

$$\frac{dr}{dt} = \frac{L}{mr^2}\frac{dr}{d\theta} = \sqrt{\frac{2}{m}\left(E - V(r) - \frac{L^2}{2mr^2}\right)}$$

$$d\theta = \frac{1}{r^2} \frac{1}{\underbrace{\sqrt{\frac{2mE}{L^2} - \frac{2mV(r)}{L^2} - \frac{1}{r^2}}}_{\text{This is a function of } r \text{ only.}}} dr$$

This expression is ready to be integrated, but first we need to specify $V(r)$.

1.2 Power Law Potential

A function satisfies **power law** if it is in the form $f(x) \sim x^\alpha$ for any real number α . Most of the time we deal with potential function belongs in such form, i.e. $V(r) = kr^\alpha$. The most common ones are:

- $\alpha = -1$: Columb's Law / Newtonian gravity – $V(r) = \pm \frac{k}{r}$
- $\alpha = 2$: Simple harmonic motion – $V(r) = \frac{1}{2}kr^2$

Substituting the power law potential, mathematicians found that the integral can be resolved into simple functions only when $\alpha = 2, -1$ or -2 .

2 Case $\alpha = -1$: Inverse square force

The R.H.S integral is ready to be integrated. First substitute $u = \frac{1}{r}$ and $du = -\frac{1}{r^2} dr$. Then simplify by trigonometric substitution.

$$\begin{aligned}
 \int d\theta &= \int \frac{1}{r^2} \frac{dr}{\sqrt{\frac{2mE}{L^2} - \frac{2mV(r)}{L^2} - \frac{1}{r^2}}} \\
 &= - \int \frac{du}{\sqrt{\frac{2mE}{L^2} - \frac{2mV(u^{-1})}{L^2} - u^2}} && \text{(subst. } u=\frac{1}{r}, du=-\frac{1}{r^2}dr \text{)} \\
 &= - \int \frac{du}{\sqrt{\frac{2mE}{L^2} \pm \frac{2mk}{L^2}u - u^2}} \\
 &= - \int \frac{du}{\sqrt{\frac{2mE}{L^2} + \frac{m^2k^2}{L^4} - (u \mp \frac{mk}{L^2})^2}} && \text{(completing square)} \\
 &= - \int \frac{d(u \mp \frac{mk}{L^2})}{\sqrt{A^2 - (u \mp \frac{mk}{L^2})^2}} && \text{(take } A=\sqrt{\frac{2mE}{L^2} + \frac{m^2k^2}{L^4}} \text{)} \\
 &= - \int \frac{d(A \cos \phi)}{\sqrt{A^2 - A^2 \cos^2 \phi}} && \text{(take } u \mp \frac{mk}{L^2} = A \cos \phi \text{)} \\
 &= \int d\phi \\
 &= \cos^{-1} \left(\frac{u \mp \frac{mk}{L^2}}{\sqrt{\frac{2mE}{L^2} + \frac{m^2k^2}{L^4}}} \right) + C \\
 \theta &= \cos^{-1} \left(\frac{\frac{L^2}{mk} \mp 1}{\sqrt{\frac{2EL^2}{mk^2} + 1}} \right) + C && \text{(subst. } u=\frac{1}{r} \text{ and rearrange terms)} \\
 \boxed{\frac{1}{r} = \frac{mk}{L^2} \left(\sqrt{1 + \frac{2EL^2}{mk^2}} \cos(\theta - C) \pm 1 \right)}
 \end{aligned}$$

where the \pm sign is for: **+ve = attractive force**, **-ve = repulsive force**. This formula is usually written in a less complicated form by using these symbols:

$$\begin{aligned}
 - \quad \epsilon &\stackrel{\text{def}}{=} \sqrt{1 + \frac{2EL^2}{mk^2}} = \text{eccentricity} \\
 - \quad \theta_0 &\stackrel{\text{def}}{=} C = \text{some initial angle} \\
 - \quad \frac{1}{r_0} &\stackrel{\text{def}}{=} \frac{mk}{L^2} = \text{reciprocal to some initial radial distance}
 \end{aligned}$$

And becomes the **polar equation** of trajectory.

$$\frac{1}{r} = \frac{1}{r_0}(\epsilon \cos(\theta - \theta_0) \pm 1)$$

$$r = \frac{r_0}{\epsilon \cos(\theta - \theta_0) \pm 1} \quad \begin{matrix} (+\text{ve for attractive}) \\ (-\text{ve for repulsive}) \end{matrix}$$

2.1 Kepler's 1st Law - Equation of Trajectory

The original statement from Kepler only concerns elliptical/circular orbits in gravitation, because they were the only types of orbits observed.

The orbit of every planet is an ellipse with the sun at one of the two foci.

With modern mathematics, we can show that **every conic sections are possible trajectories**. Convert to cartesian coordinate by substituting $r = \sqrt{x^2 + y^2}$, $\cos \theta = \frac{x}{r}$ and $\sin \theta = \frac{y}{r}$,

$$\begin{aligned} r &= \frac{r_0}{\epsilon \cos(\theta - \theta_0) \pm 1} \\ \pm r &= r_0 - r\epsilon \cos(\theta - \theta_0) \\ r^2 &= r^2\epsilon^2 \cos^2(\theta - \theta_0) - 2rr_0\epsilon \cos(\theta - \theta_0) + r_0^2 \\ &= r^2\epsilon^2 [\cos \theta \cos \theta_0 + \sin \theta \sin \theta_0]^2 - 2rr_0\epsilon [\cos \theta \cos \theta_0 + \sin \theta \sin \theta_0] + r_0^2 \\ x^2 + y^2 &= \epsilon^2 [x \cos \theta_0 + y \sin \theta_0]^2 - 2r_0\epsilon [x \cos \theta_0 + y \sin \theta_0] + r_0^2 \end{aligned}$$

$$(\epsilon^2 \cos^2 \theta_0 - 1)x^2 + (\epsilon^2 \sin^2 \theta_0 - 1)y^2 + (2\epsilon^2 \sin \theta_0 \cos \theta_0)xy - (2\epsilon r_0 \cos \theta_0)x - (2\epsilon r_0 \sin \theta_0)y + r_0^2 = 0$$

we arrive at the **quadratic form** of conic sections, i.e. $Ax^2 + Bxy + Cy^2 + Dx + Ey + F = 0$, regardless of the force being central attractive or repulsive. Usually textbooks begin their analysis by choosing $\theta_0 = 0$, essentially making the conic section "lies flat".

$$\begin{aligned} (\epsilon^2 - 1)x^2 - y^2 - (2\epsilon r_0)x + r_0^2 &= 0 \\ x^2 - \frac{2\epsilon}{\epsilon^2 - 1}x - \frac{y^2}{\epsilon^2 - 1} &= -\frac{r_0^2}{\epsilon^2 - 1} && \text{(Divide } \epsilon^2 - 1) \\ \left(x + \frac{r_0\epsilon}{1 - \epsilon^2}\right)^2 - \frac{r_0^2\epsilon^2}{(1 - \epsilon^2)^2} + \frac{y^2}{1 - \epsilon^2} &= \frac{r_0^2}{1 - \epsilon^2} && \text{(Completing squares)} \\ \left(x + \frac{r_0\epsilon}{1 - \epsilon^2}\right)^2 + \frac{y^2}{1 - \epsilon^2} &= \left(\frac{r_0}{1 - \epsilon^2}\right)^2 \\ \frac{\left(x + \frac{r_0\epsilon}{1 - \epsilon^2}\right)^2}{\left(\frac{r_0}{1 - \epsilon^2}\right)^2} + \frac{y^2}{\left(\frac{r_0}{\sqrt{1 - \epsilon^2}}\right)^2} &= 1 \\ \Rightarrow \frac{(x + c)^2}{a^2} + \frac{y^2}{b^2} &= 1 \end{aligned}$$

This is the **standard form**, another common way to present conic section.

Conventionally, no matter what kinds of conic section it is, we identify these letters with the following names. Although they are mostly used in circular/elliptical orbits because they are positive real number only if $\epsilon < 1$.

- **Semi-latus rectum** – $r_0 \stackrel{\text{def}}{=} \frac{L^2}{mk}$ (also commonly denoted by l in books)
- **Semi-major axis** – $a \stackrel{\text{def}}{=} \frac{r_0}{1 - \epsilon^2} = -\frac{k}{2E}$
- **Semi-minor axis** – $b \stackrel{\text{def}}{=} \frac{r_0}{\sqrt{1 - \epsilon^2}}$
- **Focal distance** – $c \stackrel{\text{def}}{=} \frac{r_0 \epsilon}{1 - \epsilon^2} = a\epsilon$ (Another name for c = linear encentricity)

Note: The definition of a and r_0 are the most important to remember, because we can compute the object's energy and angular momentum directly from them.

$$\begin{cases} E = -\frac{k}{2a} \\ L = \sqrt{mkr_0} \end{cases} \quad (\text{True for all orbits})$$

(You can also re-derive the definition of encentricity from them).

The value of encentricity determines the ratio of a, b, c to r_0 , so as the type of conic section.

2.1.1 $\epsilon = 0$ – Circle

- Standard form: We can see that $a = b = c = r_0$.

$$(x + r_0)^2 + y^2 = r_0^2$$

which is the equation of a circle at center $(-r_0, 0)$ and radius r_0 .

- Polar equation: $r = \frac{r_0}{\epsilon \cos \theta \pm 1} = \frac{r_0}{0 \pm 1} = \pm r_0$.
 - Attractive force (+ve) : Radial distance = r_0 = constant.
 - Repulsive force (–ve) : $r < 0$ is not physical. i.e. Trajectory cannot be a circle if the central force is repulsive.
- Energy: By substituting $\epsilon = 0$ to a , $E = -\frac{k}{2r_0}$.

(add figure here: circle)

2.1.2 $0 < \epsilon < 1$ – Ellipse

- Standard form: We have $a = \frac{r_0}{1 - \epsilon^2} > \frac{r_0}{\sqrt{1 - \epsilon^2}} = b$.

$$\boxed{\frac{(x + a\epsilon)^2}{a^2} + \frac{y^2}{b^2} = r_0^2}$$

This is the equation of an ellipse with its center shifted to $(-a\epsilon, 0)$, while $(0, 0)$ becomes one of the foci of the ellipse.

- Polar equation:

- Attractive force (+ve) : $r = \frac{r_0}{\epsilon \cos \theta + 1}$, which is in the range between:

- * The **perihelion** point - When $\theta = 0$, r is min - $\boxed{r_p = \frac{r_0}{1 + \epsilon}}$

- * The **aphelion** point - When $\theta = \pi$, r is max - $\boxed{r_a = \frac{r_0}{1 - \epsilon}}$

- Repulsive force (–ve) : $r = \frac{r_0}{\epsilon \cos \theta - 1} < 0$ is not physical. i.e. Trajectory cannot be an ellipse if the central force is repulsive.

- Energy: By substituting ϵ and r_0 into a , $\boxed{-\frac{mk^2}{2L^2} < E < 0}$

(add figure here: ellipse)

2.1.3 $\epsilon = 1$ – Parabola

- Standard form: a, b, c all becomes infinite. So the standard form becomes meaningless. In fact, we have to re-derive it from the first step:

$$(\epsilon^2 - 1)x^2 - y^2 - (2\epsilon r_0)x + r_0^2 = 0$$

$$\boxed{-\frac{1}{2r_0}y^2 + \frac{r_0}{2} = x}$$

This a quadratic equation, i.e. parabola, but lying horizontally. The tip is at $(\frac{r_0}{2}, 0)$.

- Polar form:

- Attractive force (+ve) : $r = \frac{r_0}{\cos \theta + 1}$, which is in the range between:

- * The **perihelion** point - When $\theta = 0$, r is min - $\boxed{r_p = \frac{r_0}{2}}$

- * The **aphelion** point does not exist - When $\theta = \pi$, $r = \infty$.

- Repulsive force (–ve) : $r = \frac{r_0}{\cos \theta - 1} < 0$ is not physical. i.e. Trajectory cannot be a parabola if the central force is repulsive.

- Energy: By substituting $\epsilon = 1$ to the encentricity's definition, $\boxed{E = 0}$

(add figure here: parabola)

2.1.4 $\epsilon > 1$ – Hyperbola

- Standard form: $a, c < 0$ and b becomes imaginary. Take out the magnitude part: $b = i \cdot \frac{r_0}{\sqrt{\epsilon^2 - 1}} = ib'$, the standard form becomes:

$$\frac{(x + a\epsilon)^2}{a^2} - \frac{y^2}{b'^2} = 1$$

This is the equation of an hyperbola with its center shifted to $(-a\epsilon, 0)$, while $(0, 0)$ becomes one of the foci of the ellipse. The trajectories on two sides are corresponding to the cases of attractive and repulsive force.

- Polar form: $r = \frac{r_0}{\epsilon \cos \theta \pm 1} > 0$ only if $\cos \theta > \mp \frac{1}{\epsilon}$.

– Attractive force (+ve) :

* The **perihelion** point - When $\theta = 0$, r is min - $r_p = \frac{r_0}{1 + \epsilon}$

* The **aphelion** point does not exist - When $\cos(\pm\theta) = \frac{1}{\epsilon}$, $r = \infty$.

– Repulsive force (–ve) :

* The **perihelion** point - When $\theta = 0$, r is min - $r_p = \frac{r_0}{\epsilon - 1}$

* The **aphelion** point does not exist - When $\cos(\pm\theta) = -\frac{1}{\epsilon}$, $r = \infty$.

- Energy: By substituting ϵ and r_0 into a , $E > 0$

(add figure here: hyperbola)

2.2 Kepler's 2nd Law - Conservation of Angular Momentum

The original statement for Kelper's 2nd Law says:

A line joining a planet and the Sun sweeps out equal areas during equal intervals of time.

It is written in such way because the concept of angular momentum did not exist during Kelper's life (Newton was birth after Kepler's death). With mathematics, we can show:

- The angle θ of an arc is related to sweeping time by $dt = \frac{mr^2}{L} d\theta$
- The arc's area is $\iint r d\theta dr = \int \frac{1}{2} r^2 d\theta$

So combining,

$$\text{Area}_{(t_2-t_1)} = \int_{t_1}^{t_2} \frac{1}{2} \cdot \frac{L}{m} dt \overset{\substack{\text{True only if} \\ L \text{ is not a function of } t}}{\downarrow} = \frac{L}{2m} (t_2 - t_1)$$

Note that this is true even for parabola and hyperbola orbits.

(add figure here: kepler2 law illustration)

2.3 Kepler's 3rd Law - Period of Elliptical Orbit

The original statement for Kepler's 3rd Law only applies to circular and elliptical orbit, because the concept of "period" only applies to closed orbits.

The ratio of the square of an object's orbital period with the cube of the semi-major axis of its orbit is the same for all objects orbiting the same primary object.

If we go back to the standard proof in section 1.1 and stop at step 1, without substituting $dt = \frac{mr^2}{L} d\theta$, we can integrate and obtain a relation between t and r . Note that we can simplify this integral without removing the \pm sign, so it applies to both attractive and repulsive cases.

$$\begin{aligned}
 dt &= \frac{dr}{\sqrt{\frac{2}{m}(E - V(r) - \frac{L^2}{2mr^2})}} \\
 &= \frac{dr}{\sqrt{\frac{2}{m}(E \pm \frac{k}{r} - \frac{L^2}{2mr^2})}} \\
 \int dt &= \sqrt{-\frac{m}{2E}} \int \frac{r dr}{\sqrt{r^2 \mp \frac{k}{E}r + \frac{L^2}{2mE}}} \quad \left(\begin{array}{l} \text{Denominator} \\ \text{multiply } -\frac{r^2}{E} \end{array} \right) \\
 &= \sqrt{-\frac{m}{2E}} \int \frac{r dr}{\sqrt{-(r \mp \frac{k}{2E})^2 + \frac{k^2}{4E^2} + \frac{L^2}{2mE}}} \quad \left(\begin{array}{l} \text{completing} \\ \text{square} \end{array} \right)
 \end{aligned}$$

We can substitute $a = -\frac{k}{2E}$ and $a^2\epsilon^2 = \frac{k^2}{4E^2} \left(1 + \frac{2EL^2}{mk^2}\right)$ to simplify the notations.

$$\begin{aligned}
 \Rightarrow \int dt &= \sqrt{-\frac{m}{2E}} \int \frac{r dr}{\sqrt{-(r \pm a)^2 + a^2\epsilon^2}} \\
 &= \sqrt{-\frac{m}{2E}} \int \frac{(a\epsilon \cos \varphi \mp a) d(a\epsilon \cos \varphi \mp a)}{\sqrt{a^2\epsilon^2 - (a\epsilon \cos \varphi)^2}} \quad (\text{take } r \pm a = a\epsilon \cos \varphi) \\
 &= \sqrt{-\frac{m}{2E}} \int -(a\epsilon \cos \varphi \mp a) d\varphi \\
 &= \sqrt{-\frac{m}{2E}} (-a\epsilon \sin \varphi \pm a\varphi) + C \\
 &= \sqrt{\frac{a}{k}} (-a\epsilon \sin \varphi \pm a\varphi) + C \quad (\text{by } a = -\frac{k}{2E})
 \end{aligned}$$

The symbol φ is commonly denoted as the **eccentric anomaly**. Its geometrical relation with the rotation angle θ is as depicted:

(add figure here: eccentric anomaly)

φ and θ can be inter-converted through their relations with r . Although this is usually done numerically.

$$\begin{aligned}
 \frac{r_0}{\underbrace{\epsilon \cos \theta \pm 1}_{\text{Polar Equation}}} &= r = \underbrace{a\epsilon \cos \varphi \mp a}_{\text{Definition of } \varphi} \\
 \Rightarrow \frac{r_0}{a} &= 1 - \epsilon^2 = (\text{A constant}) = (1 \mp \epsilon \cos \varphi)(1 \pm \epsilon \cos \theta)
 \end{aligned}$$

We can immediately check that in elliptical orbit, $\theta = 0 \Leftrightarrow \varphi = 0$ and $\theta = \pi \Leftrightarrow \varphi = \pi$. So taking φ from 0 to 2π is the same as revolving one cycle of θ from 0 to 2π . Substitute these into the t v.s. φ relation, we arrive at the familiar Kepler's 3rd law formula.

$$1 \text{ Period} = \int_0^T dt = \sqrt{\frac{a}{k}} (-a\epsilon \sin \varphi \pm a\varphi) \Big|_{\varphi=0}^{\varphi=2\pi}$$

$$\boxed{T = \frac{2\pi}{\sqrt{k}} a^{\frac{3}{2}}}$$

Note: Trajectory are more frequently expressed in terms of φ in space engineering texts because of the t v.s. φ relation – It is a lot easier for computing transit time along any segment of any orbit.

3 Two-body Problem Reduction

In a two body system, where their interaction forces satisfy:

- its magnitude that only depends on the distance between the two objects.
- its direction is along the line that connects the two objects.

Then it is easy to show that the equation of motions can be rewritten into the form of one body subjecting to the same force. Thus we can solve their trajectories by solving only 1 ODE.

(add figure here: two body config)

The pairs of Newton 2nd Law should start in this form:

$$\begin{cases} m_1 \frac{d^2 \vec{r}_1}{dt^2} = \vec{F}(|\vec{r}_2 - \vec{r}_1|) \\ m_2 \frac{d^2 \vec{r}_2}{dt^2} = -\vec{F}(|\vec{r}_2 - \vec{r}_1|) \end{cases}$$

We can immediately see that, after dividing the masses and subtracting one of them by the other, we arrive at

$$\begin{aligned} \frac{d^2 \vec{r}_2}{dt^2} - \frac{d^2 \vec{r}_1}{dt^2} &= -\left(\frac{1}{m_2} + \frac{1}{m_1}\right) \vec{F}(|\vec{r}_2 - \vec{r}_1|) \\ \Rightarrow \frac{d^2 \vec{u}}{dt^2} &= -\left(\frac{1}{m_2} + \frac{1}{m_1}\right) \vec{F}(|\vec{u}|) \end{aligned}$$

by taking $\vec{u} = \vec{r}_2 - \vec{r}_1$. It is common to denote $\mu = \frac{1}{\frac{1}{m_1} + \frac{1}{m_2}}$ as the **reduced mass**, so that it is equivalent to the Newton 2nd Law describing an object of mass μ moving along its trajectory \vec{u} subjecting to a force $-\vec{F}(|\vec{u}|)$. On the other hand, adding the two equation gives the Newton 2nd Law about the center of mass, $\vec{r}_{CM} = \frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2}$:

$$\begin{aligned} m_2 \frac{d^2 \vec{r}_2}{dt^2} + m_1 \frac{d^2 \vec{r}_1}{dt^2} &= 0 \\ \Rightarrow \frac{d^2}{dt^2} (m_1 \vec{r}_1 + m_2 \vec{r}_2) &= \frac{d^2}{dt^2} \left(\frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2} \right) = 0 \end{aligned}$$

(add figure here: equiv: two obj orbit each other == 1 obj, reduced mass, orbit around origin, $\mathbf{r}=\mathbf{r}_2-\mathbf{r}_1$. finally retrieve back $\mathbf{r}_1, \mathbf{r}_2$ from \mathbf{u})

Now the problem is reduced to two independent ODEs — a still annoying one for $\vec{\mathbf{u}}$ and a trivial one for $\vec{\mathbf{r}}_{CM}$ respectively.

$$\begin{cases} \mu \frac{d^2 \vec{\mathbf{u}}}{dt^2} &= -\vec{\mathbf{F}}(|\vec{\mathbf{u}}|) \\ \frac{d^2 \vec{\mathbf{r}}_{CM}}{dt^2} &= 0 \end{cases}$$

which is comparatively easier than solving a system of 2 ODEs. Finally with $\vec{\mathbf{u}}$ and $\vec{\mathbf{r}}_{CM}$, we can retrieve back $\vec{\mathbf{r}}_1$ and $\vec{\mathbf{r}}_2$ by the geometric relation.

$$\begin{cases} \vec{\mathbf{r}}_1 = \vec{\mathbf{r}}_{CM} - \frac{m_2}{m_1+m_2} \vec{\mathbf{u}} \\ \vec{\mathbf{r}}_2 = \vec{\mathbf{r}}_{CM} + \frac{m_1}{m_1+m_2} \vec{\mathbf{u}} \end{cases}$$

— The End —