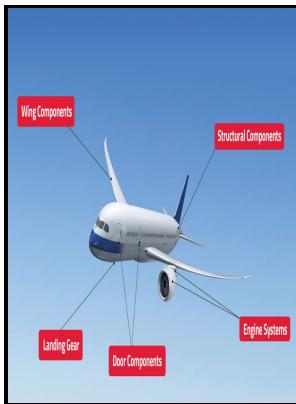


Creep in aircraft structures

Aeronautical Research Laboratories - Design Limits for Buckling in the Creep Range



Description: -

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Notes: Bibliographical references: p. 26-32.

This edition was published in 1959



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Creep Age Forming

Scale-up effects are handled through a building-block approach that relies on testing to verify the anticipated structural performance at each scale level.

Creep (deformation)

The most effective strengthening phases are oxides, carbides, or intermetallic phases, because they are usually much stronger than the host metal and therefore create strong obstacles to dislocation motion. The outstanding creep and stress rupture resistance of nickel-based superalloys is shown in Fig. At this point nucleation begins at the base of the sample and dendrites form in the same way.

Creep in aircraft structures (1959 edition)

The strain rate diminishes to a minimum and becomes near constant as the secondary stage begins.

Creep and Creep Failures

Stress rupture tests at 400, 600, 800, and 1000 deg F, to at least 764 hours in some instances, confirmed the excellent strength retention of hot pressed materials.

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Creep becomes a problem when the stress intensity is approaching the. However, when the range of temperatures is larger than accounted for, the hottest tubes fail sooner than expected. The creep rate of hot pressure-loaded components in a nuclear reactor at power can be a significant design constraint, since the creep rate is enhanced by the flux of energetic particles.

Design Limits for Buckling in the Creep Range

It can contain elements of more than a dozen of atomic components ranging from less than 1% to over 50% with the major components from

transition metals in addition to Ni. The fuel control normally maintains EGT within a safe margin. These cause the material to become brittle and can also lead to discolouration.

Related Books

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