

What is MIL-STD-1553?

- MIL-STD-1553 is a military standard that specifies the requirements for a digital command/response time division multiplex data bus for integration of aircraft subsystems.
- MIL-STD-1553B defines the term Time Division Multiplexing (TDM) as “the transmission of information from several signal sources through one communications system with different signal samples staggered in time to form a composite pulse train.”
- Review of MIL-STD-1553 Specification

WHY?

What need does it fill

WHAT?

What does the spec say

HOW?

How is it implemented

MIL-STD-1553 Highlights

- The 1553 Standard contains several sections and describes the method of communication, the data bus requirements and the electrical interface requirements for subsystems connected to the data bus.
- MIL-STD-1553B defines the term Time Division Multiplexing (TDM) as “the transmission of information from several signal sources through one communications system with different signal samples staggered in time to form a composite pulse train.”
- MIL-STD-1553 IS A STANDARD, NOT A SPECIFICATION.
- The functional elements of 1553 consist of the data bus and terminals. The Standard defines the data bus to be a single path between the bus controller and all remote terminals.
- This includes the twisted, shielded pair cable, terminators and couplers. Most systems today
- The *bus controller* is the master device and there is only one terminal operating as bus controller at any one point in time.

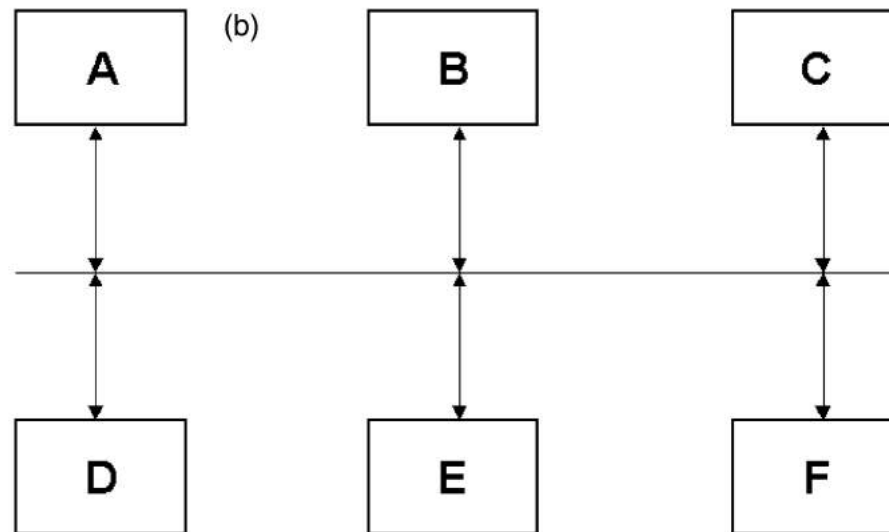
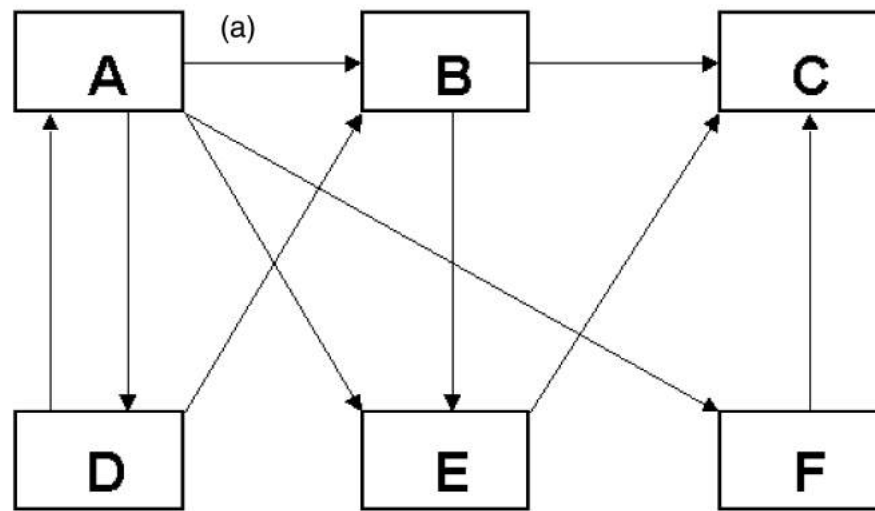


TABLE 1.1 Summary of the 1553 Data Bus Characteristics

Data Rate	1 MHz
Word Length	20 bits
Data Bits per Word	16 bits
Message Length	Maximum of 32 data words
Transmission Technique	Half-Duplex
Operation	Asynchronous
Encoding	Manchester II Bi-phase
Protocol	Command-Response
Bus Control	Single or Multiple
Message Formats	Controller-to-Terminal (BC-RT)
	Terminal-to-Controller (RT-BC)
	Terminal-to-Terminal (RT-RT)
	Broadcast
Number of Remote Terminals	System Control
	Maximum of 31
	Remote Terminal (RT)
	Bus Controller (BC)
Terminal Types	Bus Monitor (BM)
	Twisted Shielded Pair Cable
	Transformer or Direct
Transmission Media	
Coupling	

1553 Communication

- 1553 communication uses three word types: **command**, **status** and **data**. All words are 20 bits long.

Bits Used

- Word Sync-----3 bits
- Information-----16 bits and
- Parity-----1 bit

Data Integrity and System Reliability

- MIL-STD-1553 provides a high degree of data integrity by specifying word and message validation requirements.
- These include **checks for parity**, proper Manchester encoding, **bit count**, **word count** and **proper timing**.

Testing 1553

When working with MIL-STD-1553 there are a number of phases of testing that should be considered:

- development testing,
- validation testing,
- system integration/simulation.

1553 Applications

- The 1553 data bus is the most commonly used military data bus today.
- It is used in systems where data integrity and system reliability are critical.
- It is heavily used in aircraft avionics and stores and in ships, submarines and ground vehicles such as tanks.
- The data bus is also being used in space in numerous satellites and the Space Station and in some commercial applications such as reactors, subway cars and oil drilling.

TERMS

- Avionics Bus
- Remote Terminal
- Bus Controller
- Bus Monitor
- Source
- Sink
- Data Coupler
- Dual Redundant
- Minor Frame
- Major Frame
- Message

Examples of Clients

Altimeter

Active Device

Display

Passive Device

Black Box

Passive Device

Flight Computer

Active and Passive

First Generation Analog Devices

One Device

Altimeter

Speedometer

Compass

One Display

Gauge

Gauge



Second Generation

ARINC-429

One Device

Altimeter

Engine



Multiple Sink

Display

Auto Pilot

Flight Recorder

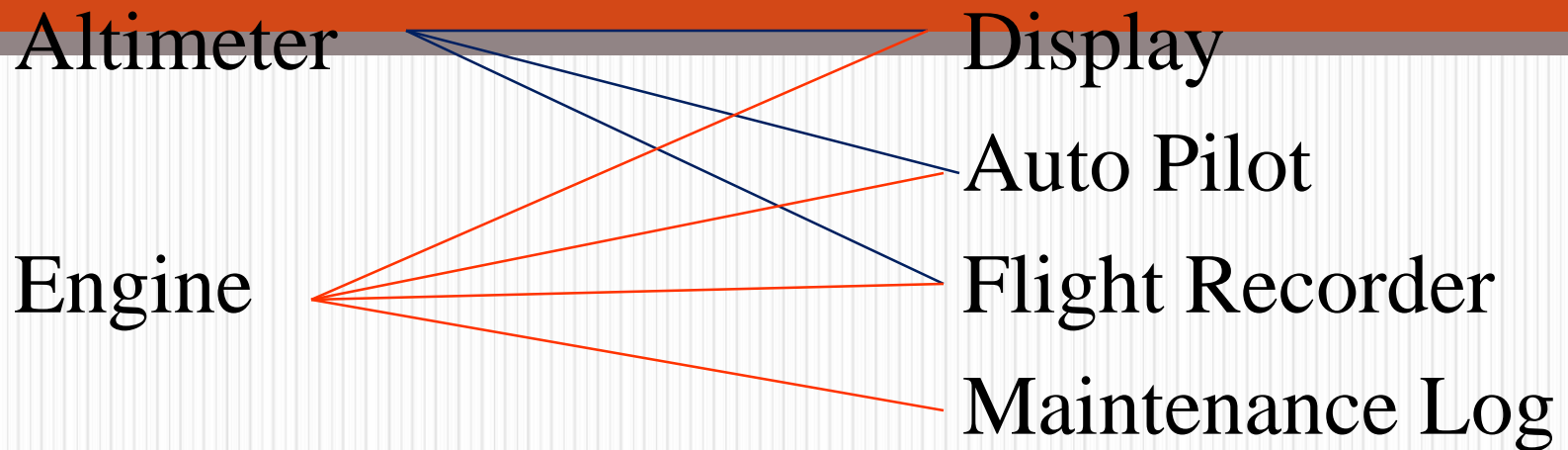
Display

Auto Pilot

Flight Recorder

Maintenance Log

Wiring Diagram



Military Applications

Multiple Device

Missile # 1

Missile # 2

Missile # 3

Missile # 4

Multiple Sink

Display

Trigger

Weapons Test

Flight Recorder

Military Application

ARINC-429 Wiring

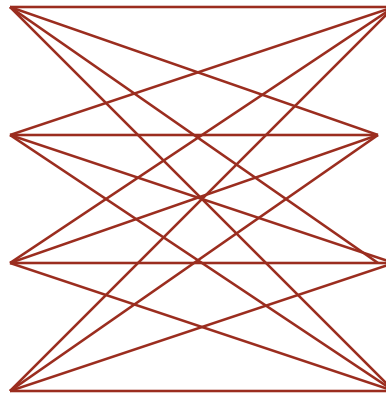
Multiple Device

Missile # 1

Missile # 2

Missile # 3

Missile # 4



Multiple Sink

Display

Trigger

Weapons Test

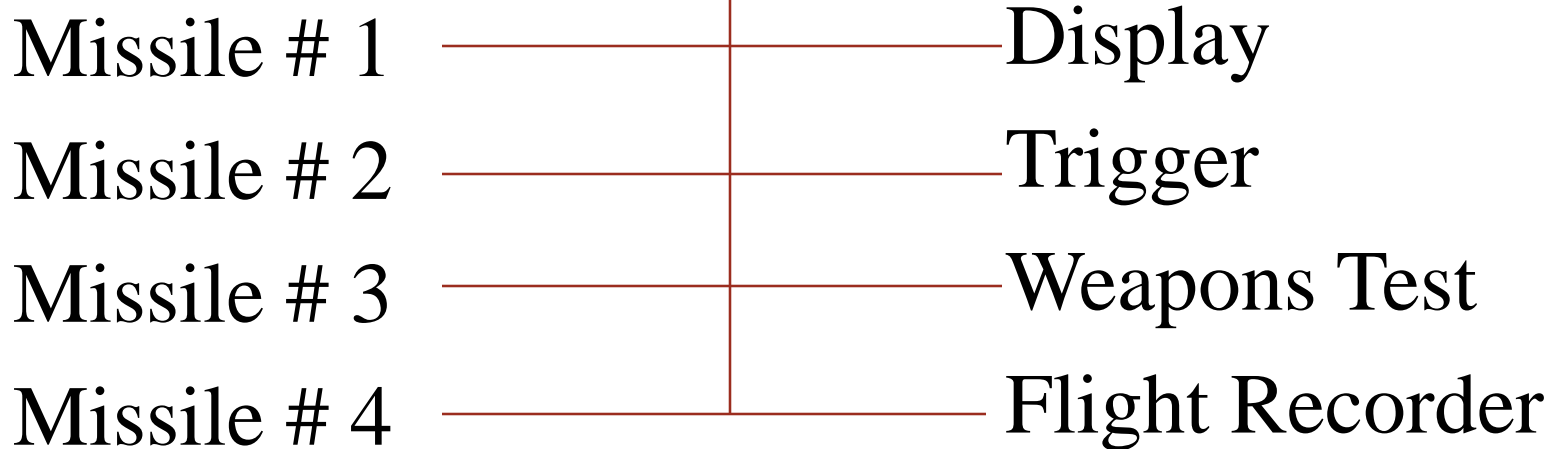
Flight Recorder

Military Applications

MIL-STD-1553 Wiring

Mission Computer

(Bus Controller)



1553 Advantages

- Low Weight
- Easy Bus Installation
- Easy to Add RT's

Bus Controller

- Determines Order of Transmission
- Is Source or Sink for almost all Data
- Can check for bus errors

1553 Problem

Cable Cut = Crash

Inefficiency

Serial Transmission

Xmt to / from BC

BC Overhead

BC Lost = Crash

Solution

Dual Redundant Bus

- 1 Mhz speed vs. 100 Khz for 429
- Minor Frames

Backup BC

Goals of MIL-STD-1553

- Communication between ≤ 32 Boxes
- Low Data Requirement ≤ 32 Words
- High Reliability
 - Ability to detect communication errors
 - Ability to retry on error

Mindset of MIL-STD Design

- Military Approach
 - 1 Commander in Control
 - All others speak when spoken to
 - Commander speaks to one at a time or to all together (Broadcast)

1553 Message

- All Communication is by Message
- All Messages are Initiated by the BC
- All Messages begin with a Command Word

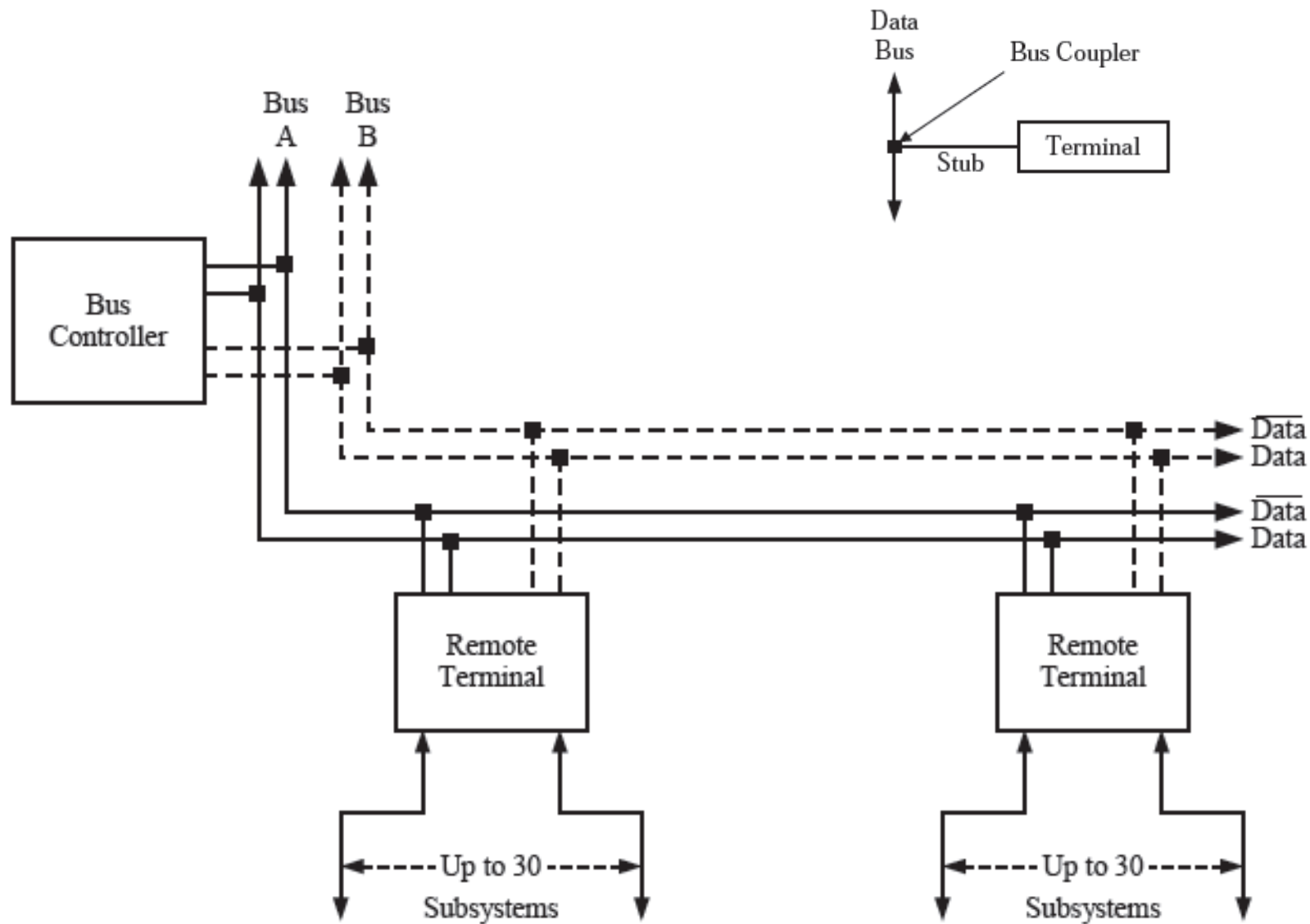
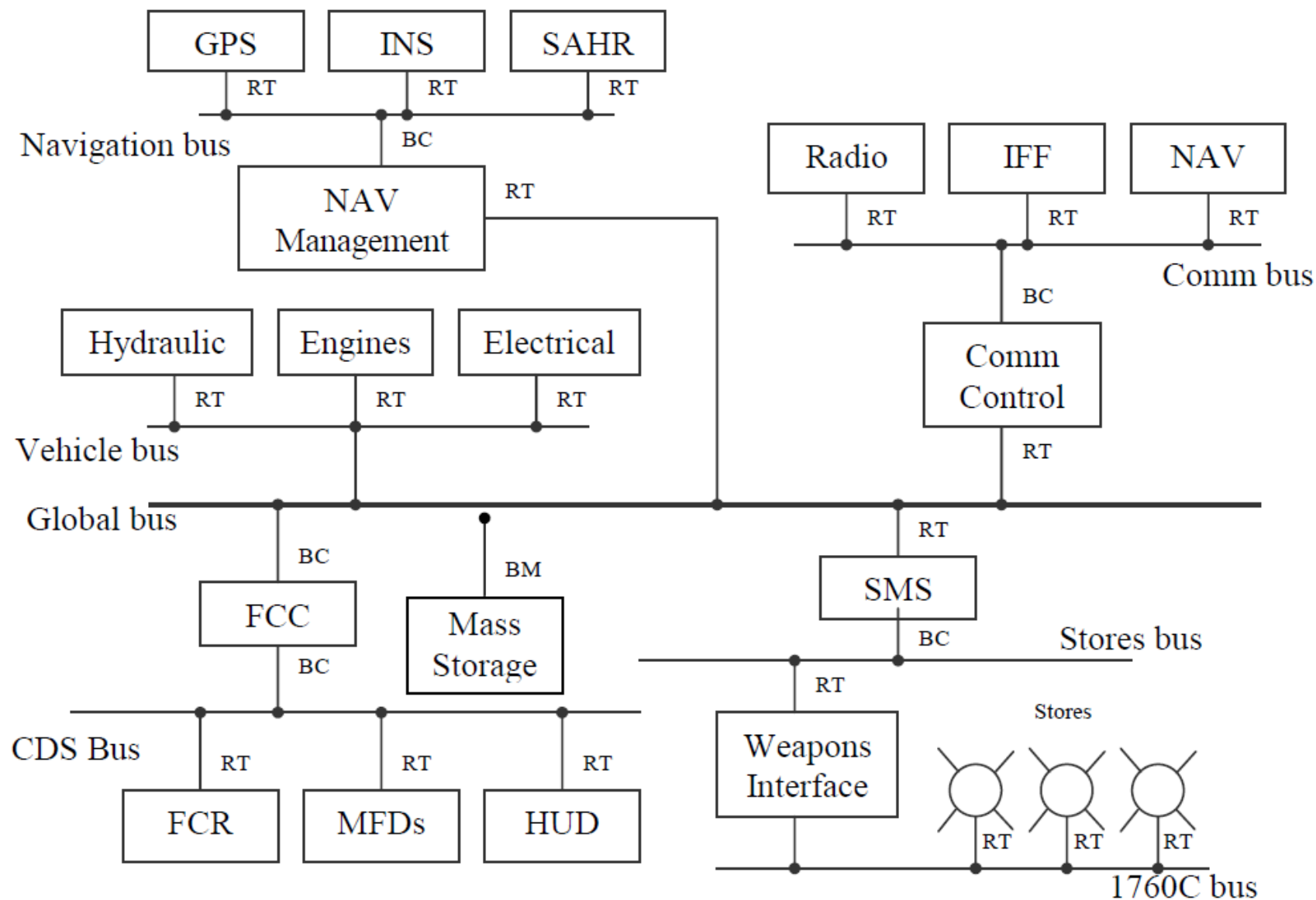


Figure 12.10 MIL-STD-1553B data bus



Bus Controller

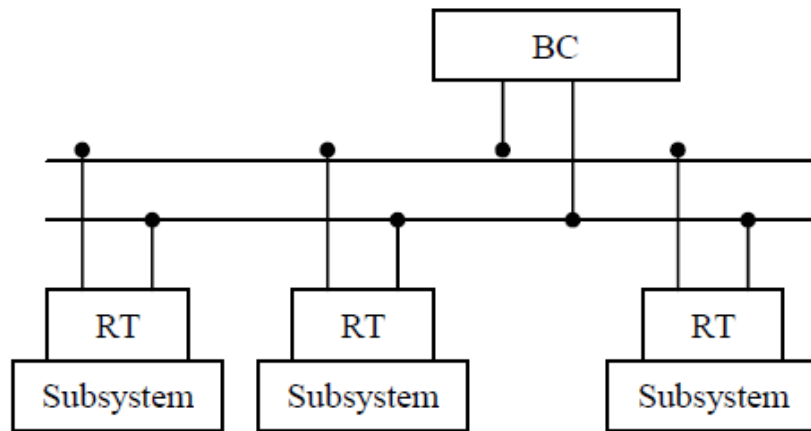


Figure 3-10.1 – Possible Redundancy

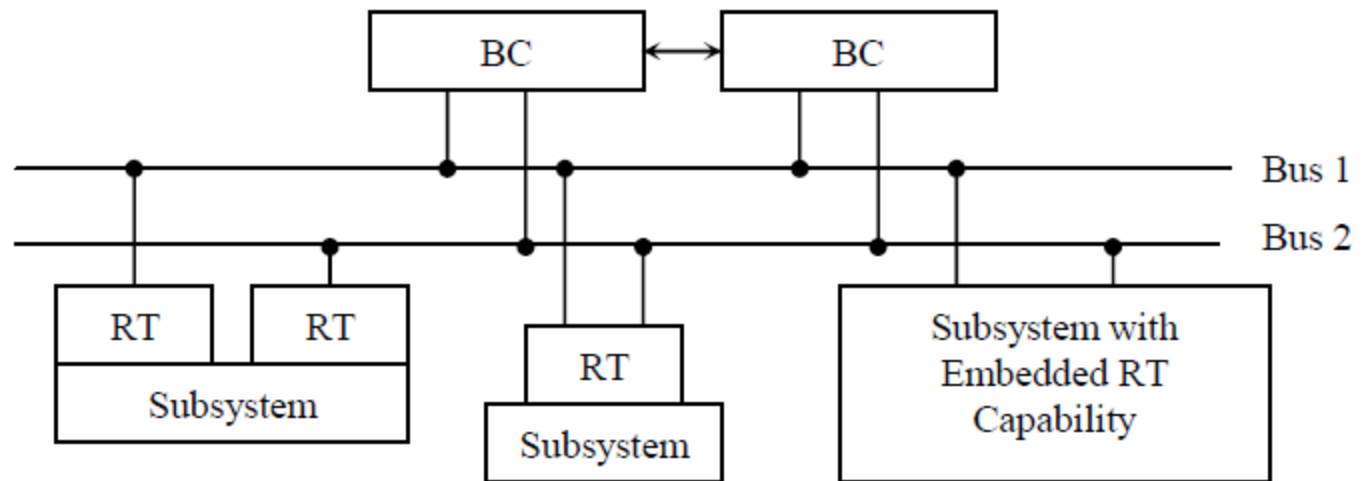


Figure 3-10.2 – Possible Redundancy

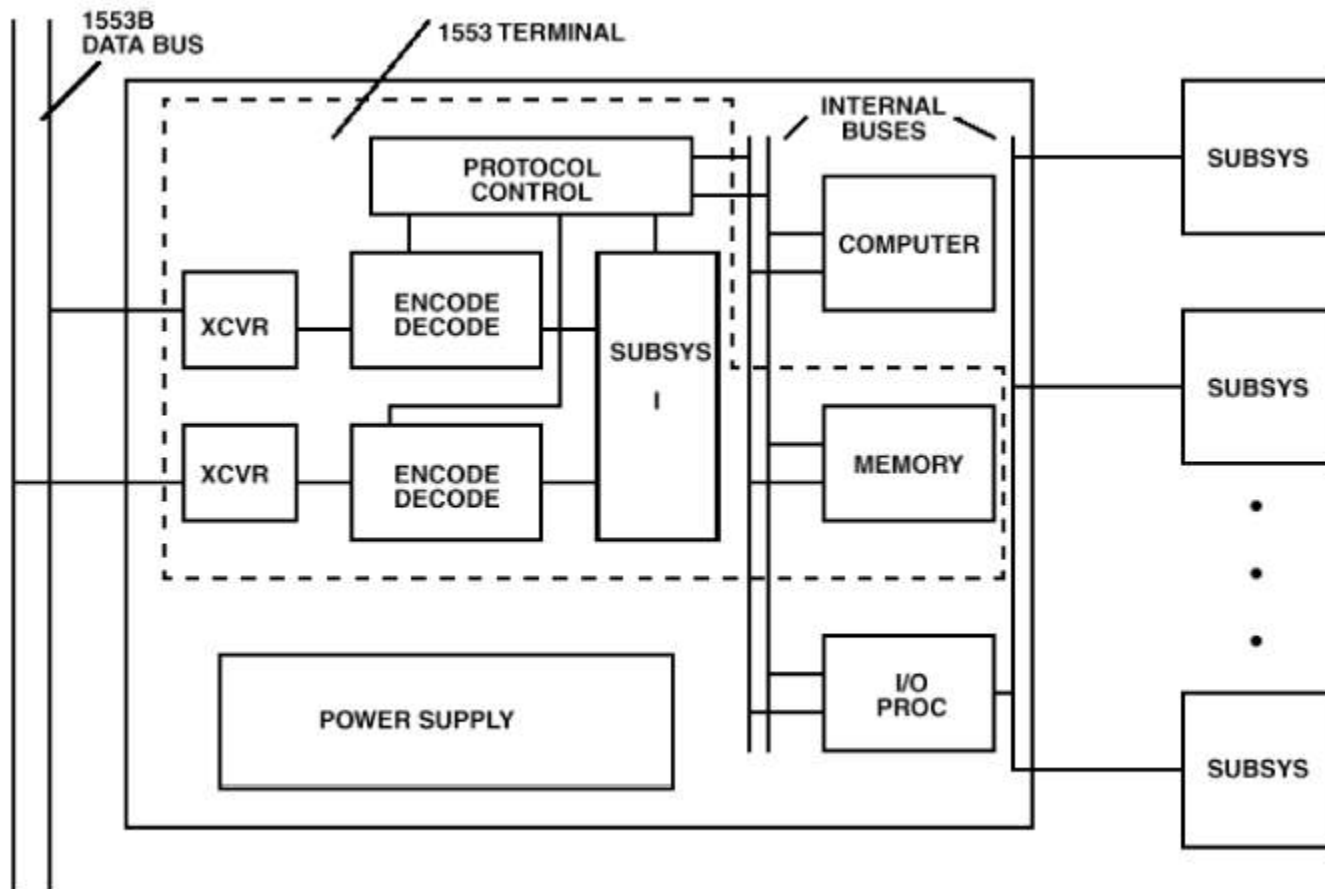


TABLE 1.3 Terminal Electrical Characteristics

Requirement	Transformer Coupled	Direct Coupled	Condition
Input Characteristics			
Input Level	0.86–14.0 V	1.2–20.0 V	p-p, l-l
No Response	0.0–0.2 V	0.0–0.28 V	p-p, l-l
Zero Crossing Stability	±150.0 nsec	±150.0 nsec	
Rise/Fall Times	0 nsec	0 nsec	Sine Wave
Noise Rejection	140.0 mV WGN ^a	200.0 mV WGN	BER 1 ^b per 10 ⁷
Common Mode Rejection	±10.0 V peak	±10.0 V peak	line-gnd, DC–2.0 MHz
Input Impedance	1000 ohms	2000 ohms	75 kHz–1 MHz
Output Characteristics			
Output Level	18.0–27.0 V	6.0–9.0 V	p-p, l-l
Zero Crossing Stability	25.0 nsec	25.0 nsec	
Rise/Fall Times	100–300 nsec	100–300 nsec	10%–90%
Maximum Distortion	±900.0 mV	±300.0 mV	peak, l-l
Maximum Output Noise	14.0 mV	5.0 mV	rms, l-l
Maximum Residual Voltage	±250.0 mV	±90.0 mV	peak, l-l

^a WGN = White Gaussian Noise.

^b BER = Bit Error Rate.

Message Types

- Bus Controller to RT
- Bus Controller to All RT's (Broadcast)
- RT to Bus Controller
- Housekeeping messages (Mode Codes)
- RT to RT Commands

Message Format

- Messages Begin With A Command Word
- Data may flow to/from BC from/to RT
- RT's return a Status Word

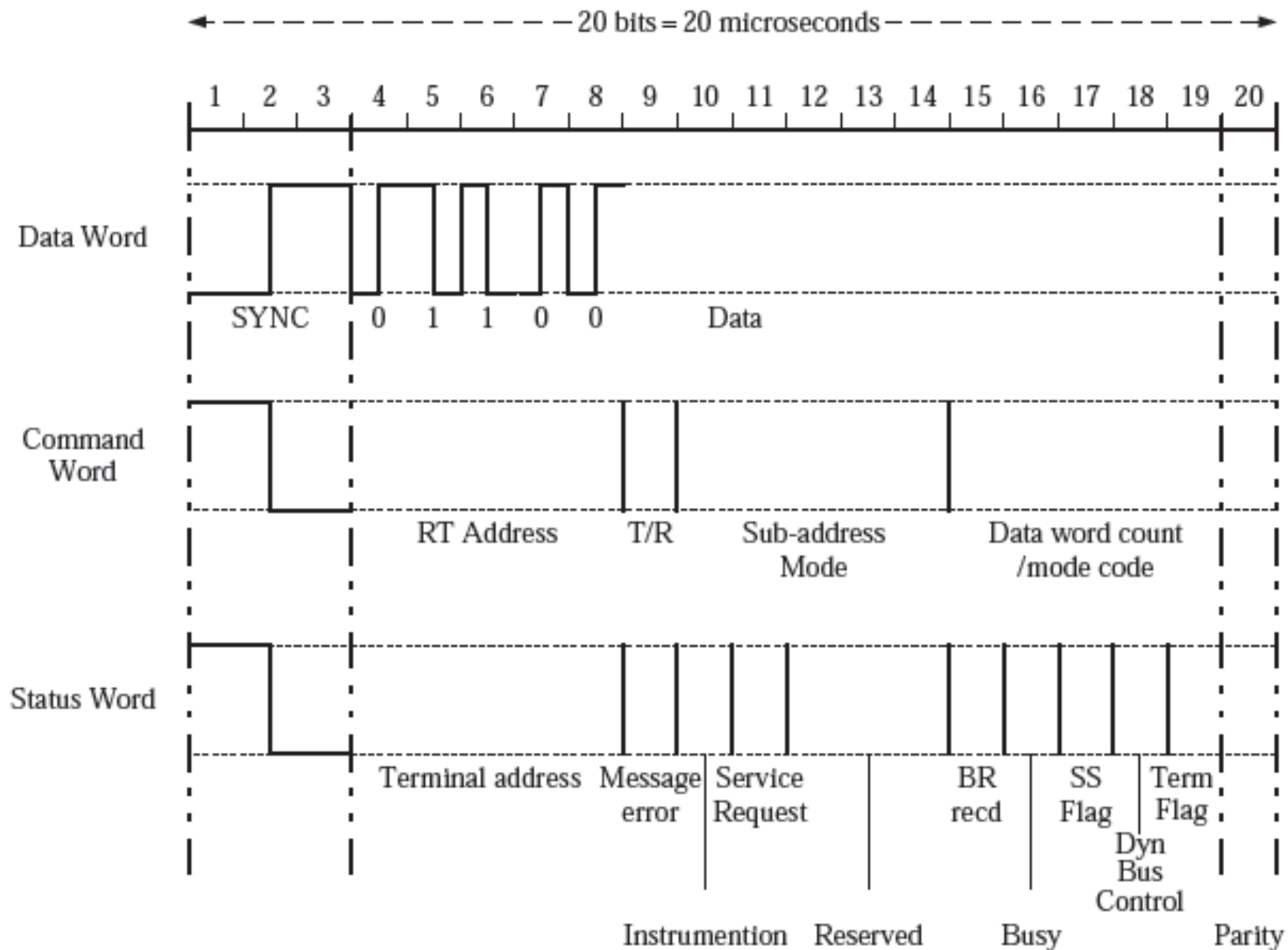


Figure 12.11 MIL-STD-1553B data bus word formats

Command Word

5 Bits

1 Bit

5 Bits

5 Bits

15 11

10

9

5

4

0

RT Address

T/R Bit

Subaddress

Word Count

Command Fields

- RT Address
 - Address range is 0 - 31
 - Some Systems use Address 31 as Broadcast
- T/R Bit
 - If $T/R = 1$, RT Transmits Data
 - If $T/R = 0$, RT Receives Data

Command Fields

- RT Subaddress
 - Additional Routing for Complex RT's
 - May Correspond to Subsystems
 - Subaddress 0 is for Mode Codes
 - Subaddress 31 is MIL-STD-1553B Mode Code

Command Fields

- Word Count
 - Range is 1 to 32 (field value 0 = 32 words)
- For Mode Codes this is Mode Code Type
 - There are 16 Mode Codes with No Data
 - There are 16 Mode Codes with 1 word of Data

Table 3-I – Assigned Mode Codes

T/R Bit	Mode Code	Function	Associated Data Word	Broadcast Command Allowed
1	00000	Dynamic Bus Control	N	N
1	00001	Synchronize	N	Y
1	00010	Transmit Status Word	N	N
1	00011	Initiate Self Test	N	Y
1	00100	Transmitter Shutdown	N	Y
1	00101	Override Transmitter	N	Y
1	00110	Inhibit Terminal Flag Bit	N	Y
1	00111	Override Inhibit Terminal Flag Bit	N	Y
1	01000	Reset RT	N	Y
1	01001	Reserved	N	TBD
<div> <div></div> <div></div> <div></div> <div></div> </div>				
1	01111	Reserved	N	TBD
1	10000	Transmit Vector Word	Y	N
0	10001	Synchronize	Y	Y
1	10010	Transmit Last Command	Y	N
1	10011	Transmit BIT Word	Y	N
0	10100	Selected Transmitter	Y	Y
0	10101	Override Selected Transmitter	Y	Y
1 or 0	10110	Reserved	Y	TBD
<div> <div></div> <div></div> <div></div> <div></div> </div>				
1 or 0	11111	Reserved	Y	TBD

Status Word

<u>Bit #</u>	<u>Description</u>
15-11	RT Address
10	Message Error
8	Service Request
4	Broadcast Received
3	Busy

Status Bit Fields

- RT Address
 - Lets BC Know Correct RT is Responding
 - Usually The Only Field Set
- Message Error
 - Indicates a Communications Error

Status Bit Fields

- Service Request (SRQ)
 - Indicates another Subaddress has info ready
 - Used with Get Vector Mode Command
- Broadcast Received
 - Set in response to the message following a broadcast command
- Busy
 - When RT can't respond - discouraged by spec

Message Sequence

**Receive
Command**

**Data
Word**

**Data
Word**

...

**Data
Word**

*

**Status
Word**

.

**Next
Command**

**Transmit
Command**

*

**Status
Word**

**Data
Word**

**Data
Word**

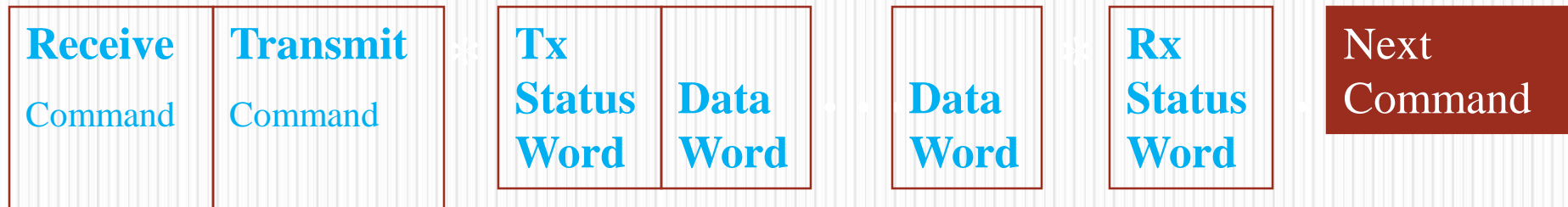
...

**Data
Word**

.

**Next
Command**

RT to RT Command



Mode Commands

**Mode
Command**

*

**Status
Word**

.

**Next
Command**

**Mode
Command**

*

**Status
Word**

**Data
Word**

.

**Next
Command**

**Mode
Command**

**Data
Word**

*

**Status
Word**

.

**Next
Command**

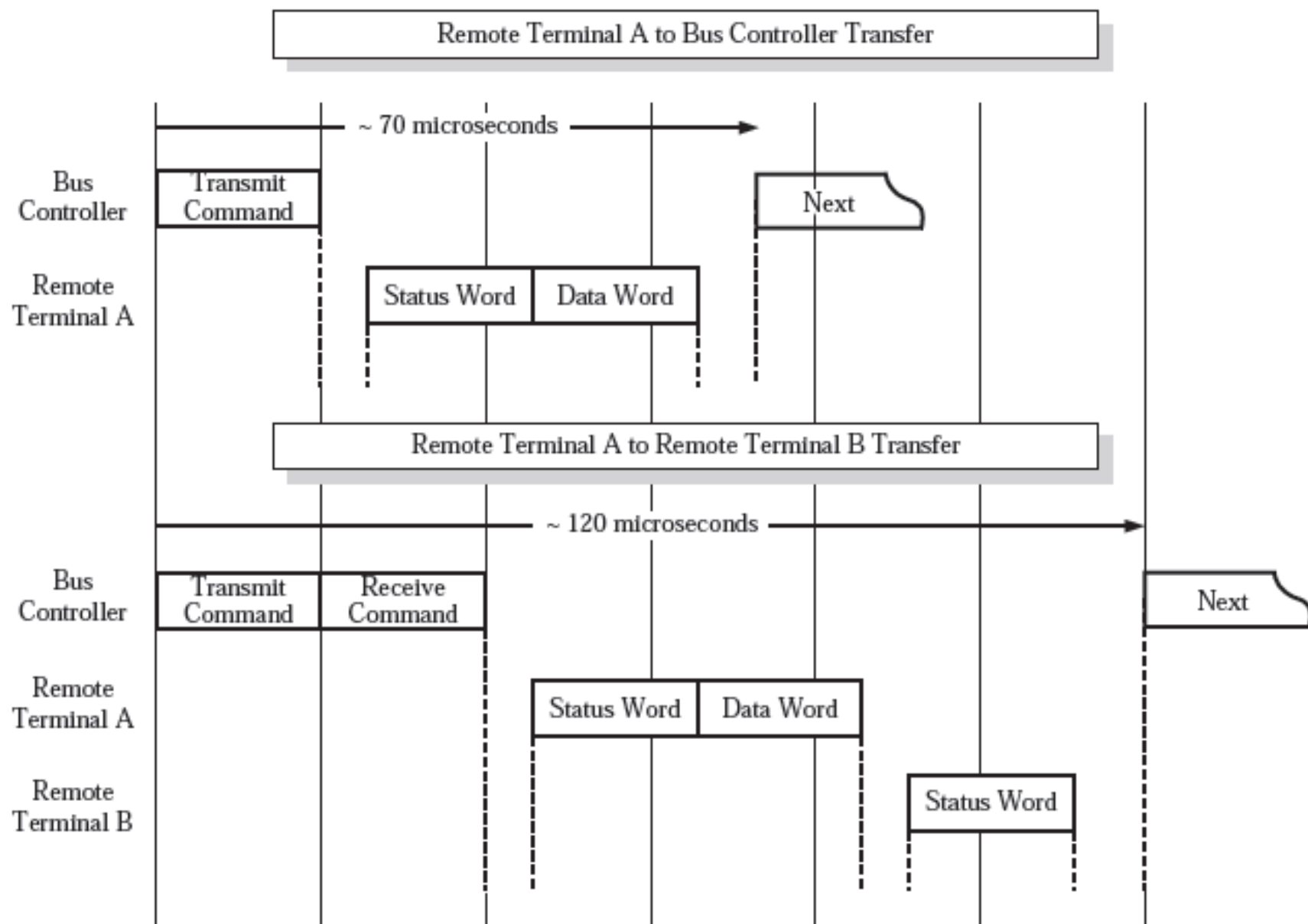


Figure 12.12 MIL-STD-1553B typical data transactions

- Implementation Issues
 - Timing
 - Major / Minor Frames
 - Implementation Examples

Timing Issues

- Intermessage Gap Time
- Response Time
- Major Frame
- Minor Frame



Intermessage Gap Time

- Time Between Messages
 - At Least 4 usec Mid Sync to Mid Parity
 - No Maximum in Specification

Response Time

- Time Until RT Sends A Status Word
 - MIL-STD-1553A Maximum = 7 usec
 - MIL-STD-1553B Maximum = 12 usec

Major Frame

- A Major Frame is the set of all messages in a single cycle
- Typical Cycle is 20 to 80 milliseconds
- Some messages may appear more than once in a single Major Frame

Minor Frames

- Some Messages Are High Priority
- We can alter frequency of specific messages

10 Mill	10 Mill	10 Mill	10 Mill
ABC	AB	A	AB

Example: Missile Test

- Does Pilot Wish To Perform a Test
- Instruct Missile to Execute Self Test
- Get Results Of Self Test
- Display Results On HUD

RT's in Test

- Self Test Button on Console RT2
- Missile RT3
- Heads Up Display (HUD) RT4

Message Frame

RT2 → BC

Button to BC

BC → RT3

BC to Missile

RT3 → BC

Missile to BC

BC → RT4

BC to HUD

Example: Synchronize RT's

- Synchronize Time Tags for All Terminals
- Check If Terminal Received the Command

Set & Check Synch

BC → Broadcast

Synchronize Mode Command

BC → RT1

Last Command Mode

BC → RT2

Last Command Mode

- Hardware Issues

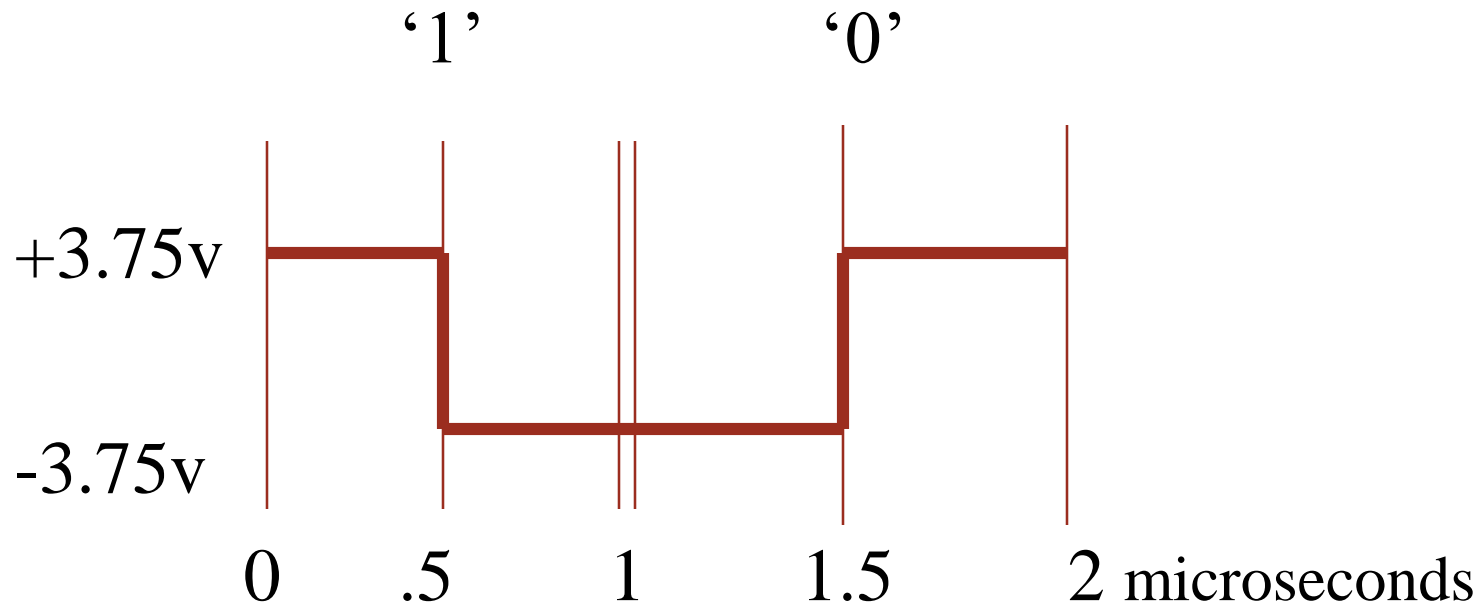
- Manchester 2 coding
- Differential Signals
- Bus Termination



Manchester Properties

- Signal moves between $+3.75\text{v}$ and -3.75v
- Signal always crosses 0v at mid bit
 - Direction of cross determines bit value
- Data Bits are 1 microsecond
 - mid bit after .5 microseconds
- Sync is 3 microseconds
 - mid sync after 1.5 microseconds

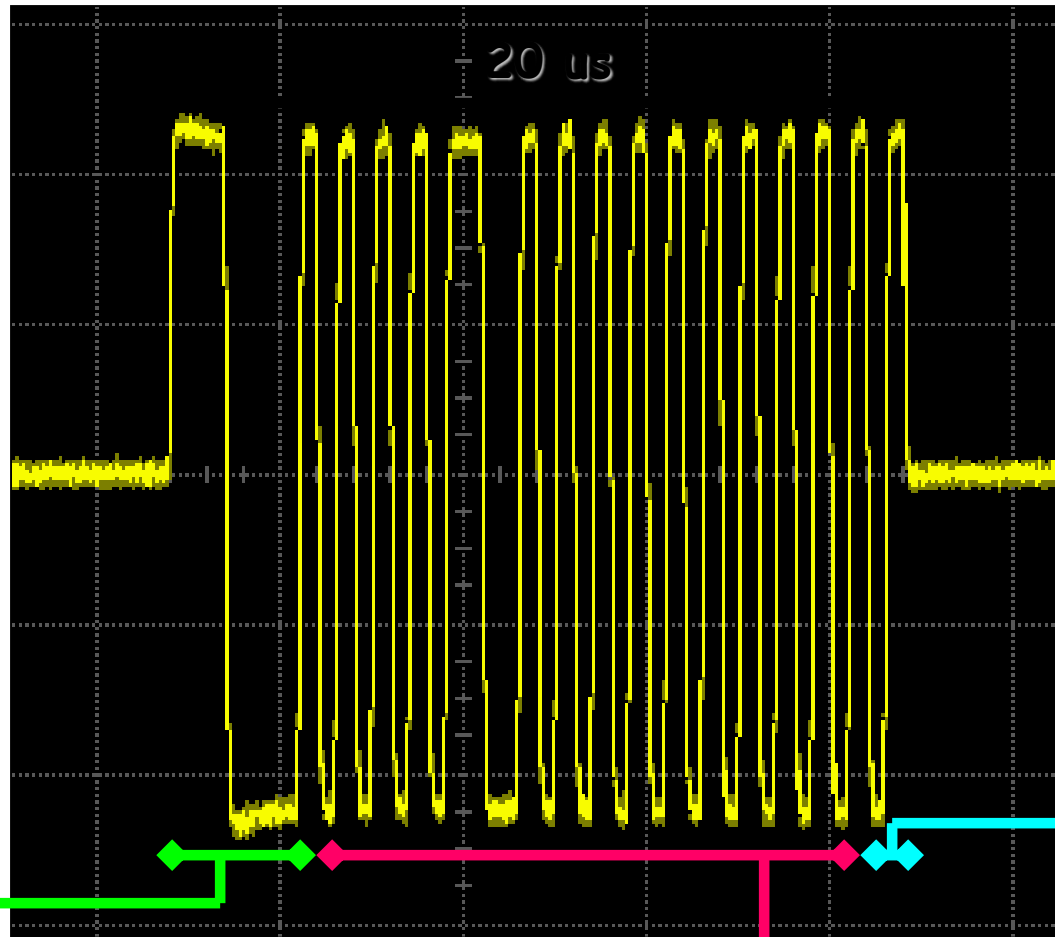
Manchester 2 Coding



Oscilloscope View

- Like preceding chart but less square
- Original spec has Trapezoidal signal
- MacAir introduced Sinusoidal signal
 - Fewer harmonics
 - Cleaner signal (less noise)

Oscilloscope View - 1553 word



Sync 3 μs

Parity 1 μs

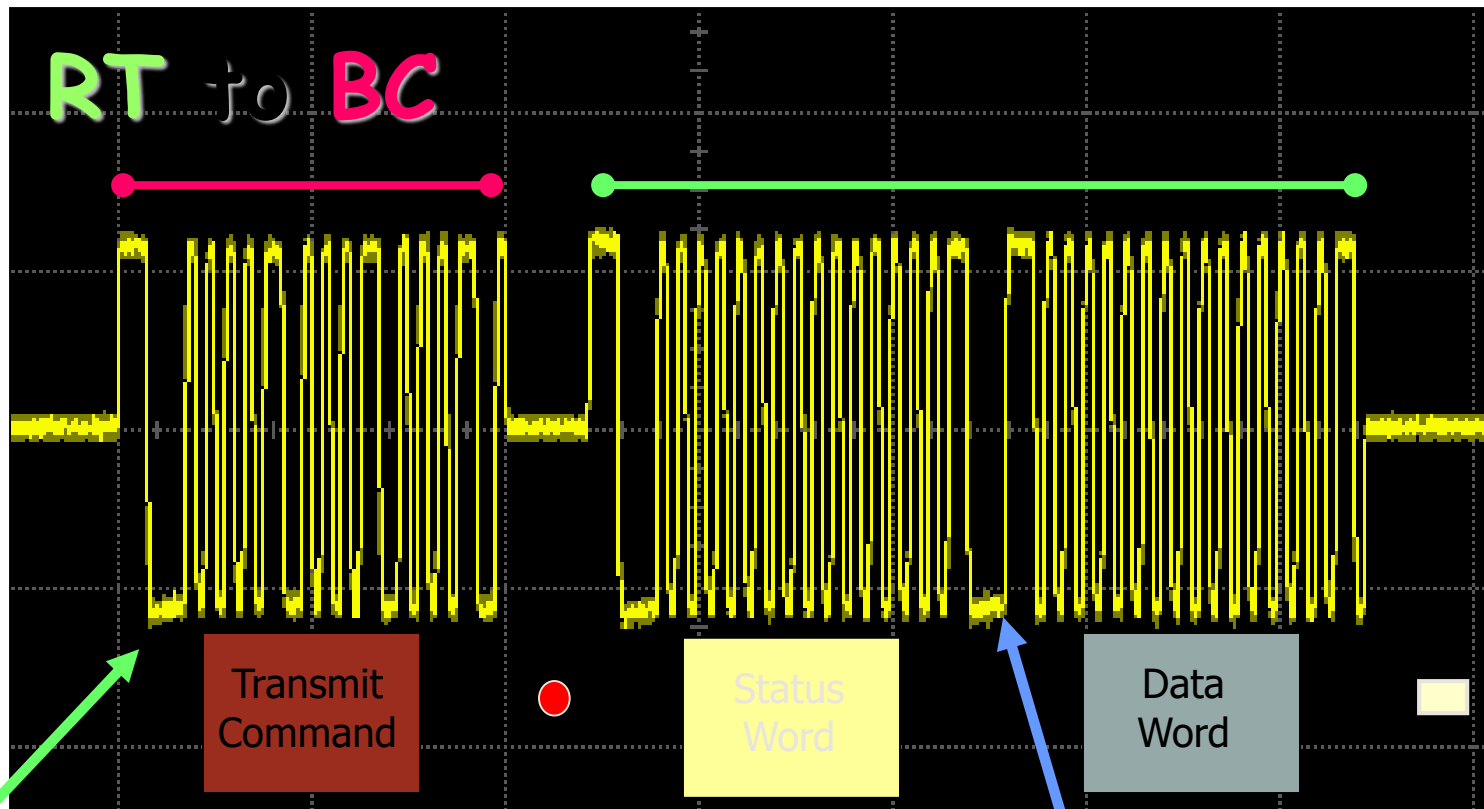
Data 16 μs

Oscilloscope View

• Response Time

1553 message

■ Intermessage Gap time



Command
Sync

Data
Sync

Differential Signals

- 1553 Bus is actually 2 wires
- The first is Manchester described above
- The second is the complement of the first
- During Bus Quiet both lines are 0 volts

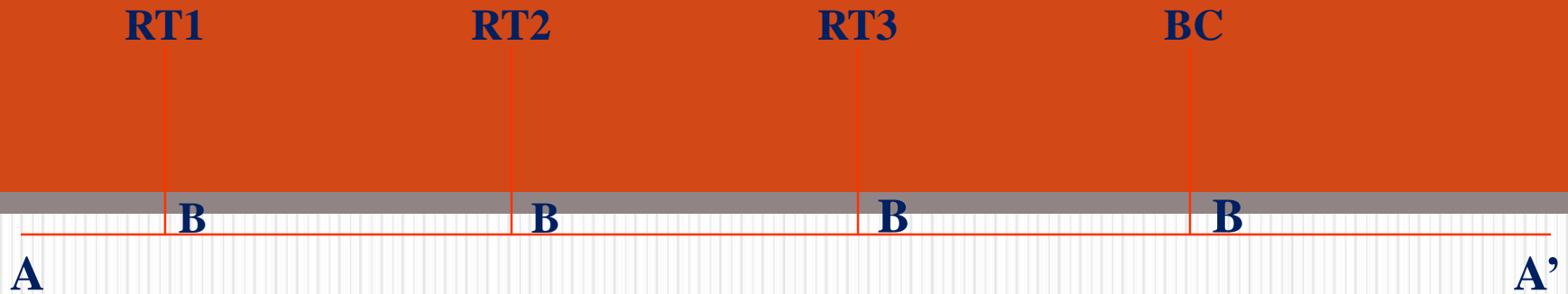
Advantages

- Less Dependent On Ground
- Less Susceptible To Spikes

Bus Termination

- Hi frequency signals are sensitive to reflection
- At the end of the Bus the signal can't continue and tries to "Bounce Back"
- This is caused by Lo Resistance wire meeting Hi Resistance air

1.553 Topology



- Point A, A' and B represent junctions
- We put Terminators on A and A' to make the bus appear infinitely long
- This prevents signal reflection

Coupling

- 'B' Junctions Represent BC's RT's and BM's Connection To The Bus
- Two Methods Are Permitted By The Spec
 - Direct Coupling
 - Transformer Coupling

Direct Coupling

- Simple Point To Point Connection
- Maximum Stub Length Is 1 Foot (30cm)

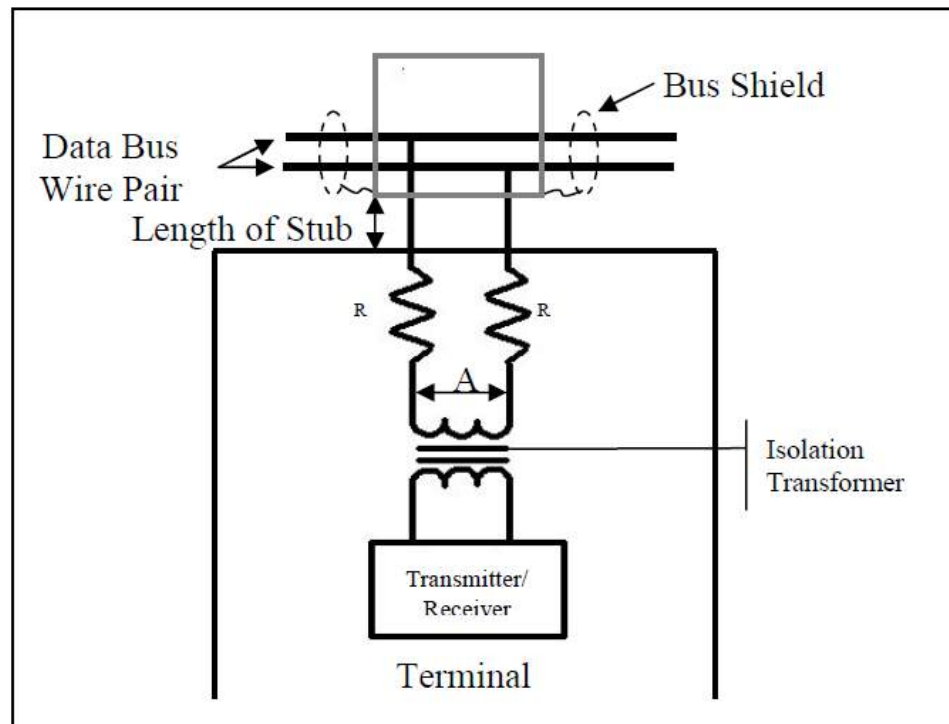


Figure 3-10 – Data Base Interface Using Direct Coupling

Transformer Coupling

- Uses An Isolation Stub Coupler
- Filters out DC and Noise
- Prevents all Reflections
- May Be Used For Up To 20 Foot (6 m) Stubs



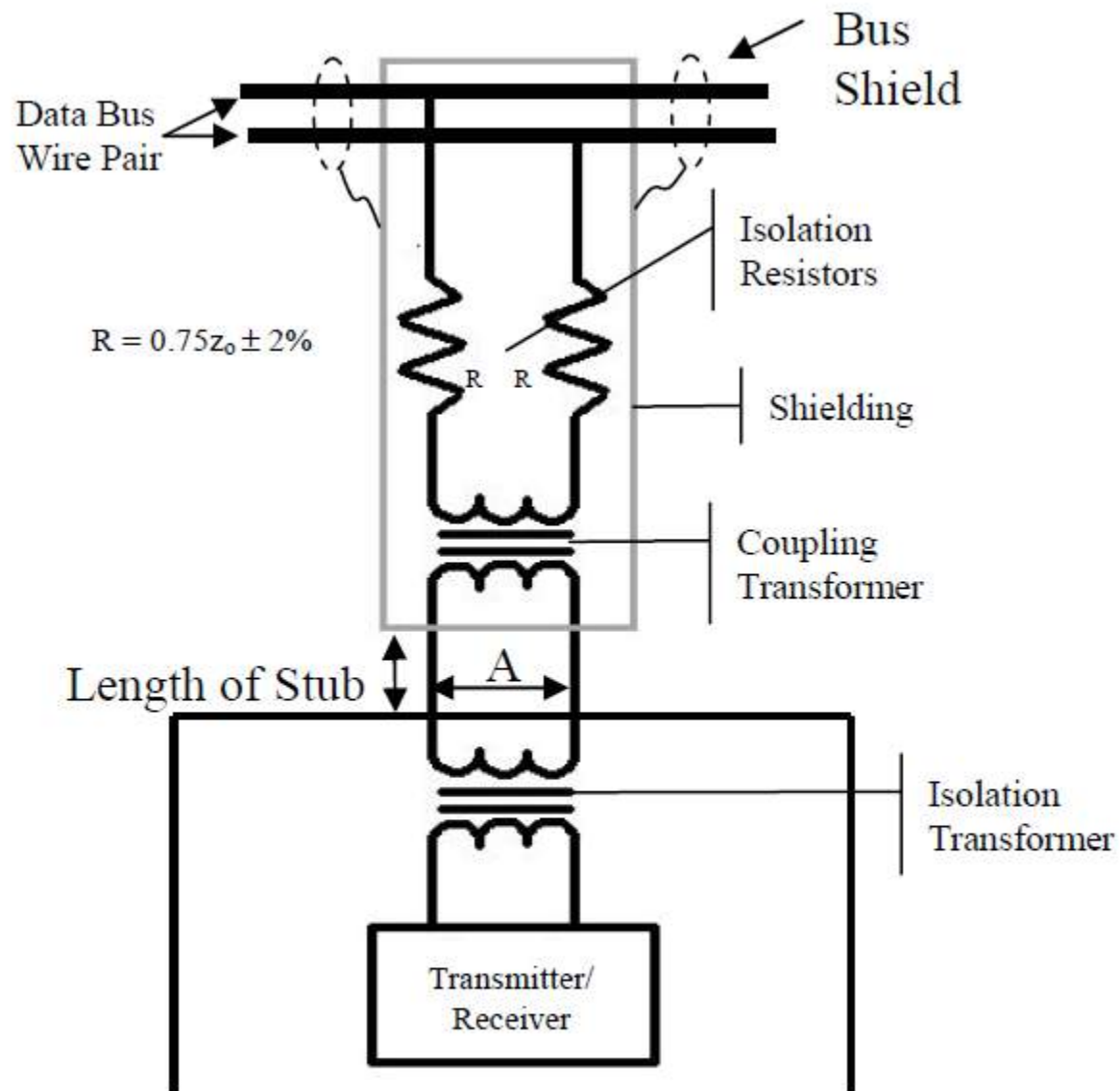
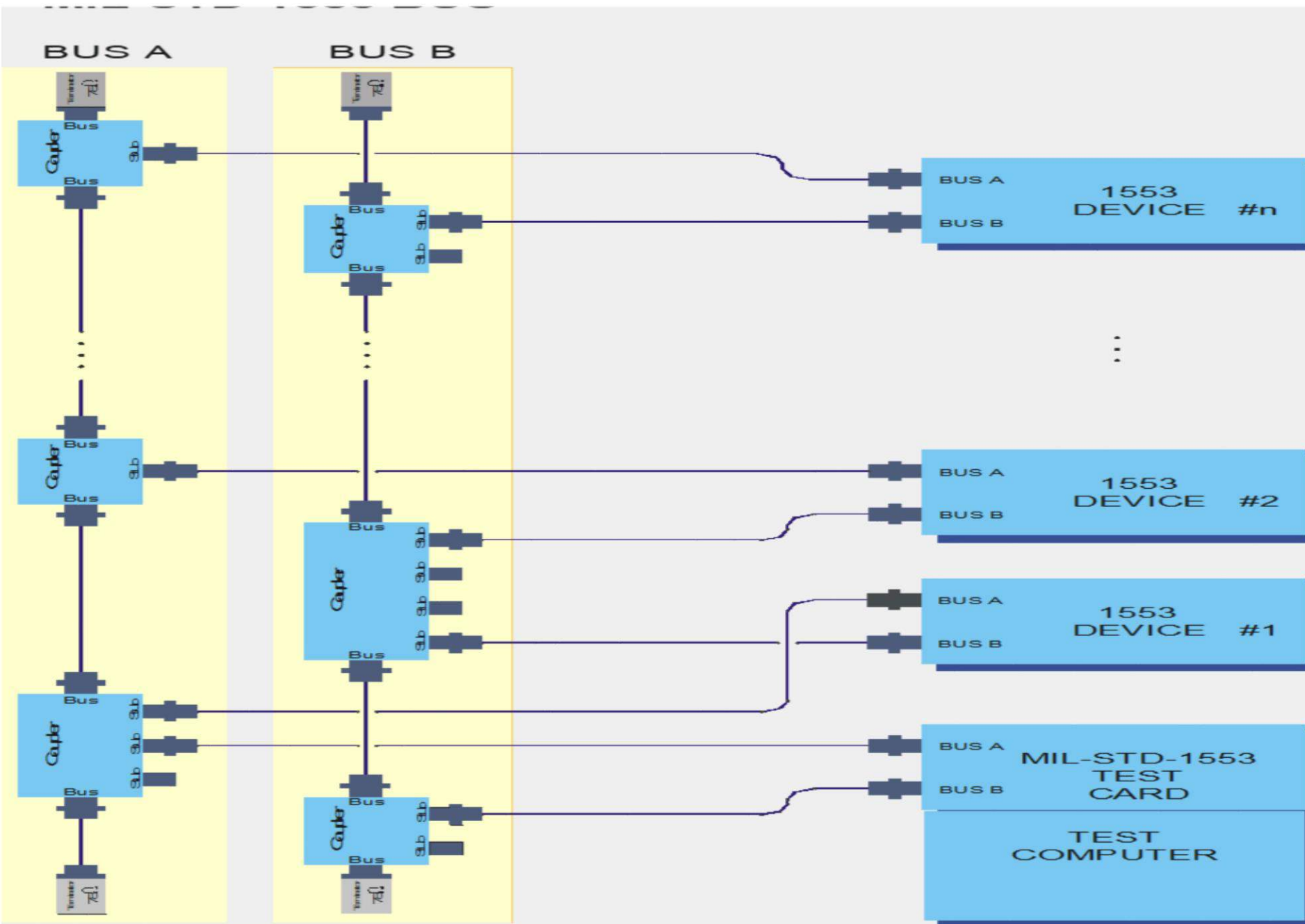


Figure 3-9 – Data Bus Interface Using Transformer Coupling

The US Air Force Prohibits Direct Coupling on Aircraft

- Direct Coupling is convenient when:
 - Used in a lab
 - Connecting Two Boxes Directly



Cable Type	Twisted Shielded Pair
Capacitance	30 pF/ft _{max}
Cable Impedance	70-85 Ω at 1 MHz
Cable Attenuation	1.5 dB per 100 ft at 1 MHz _{max}
Twist Ratio	4 per foot _{min}
Shield Coverage	75% _{min}
Cable Termination	Cable Impedance $\pm 2\%$

1553B Cable Characteristics

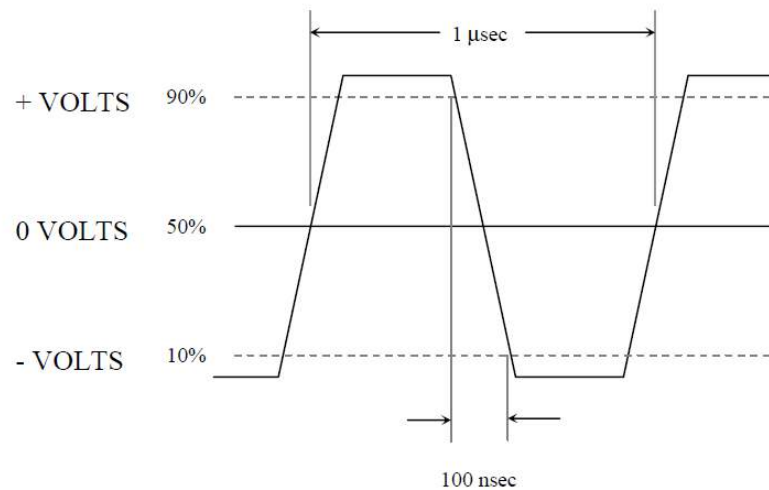
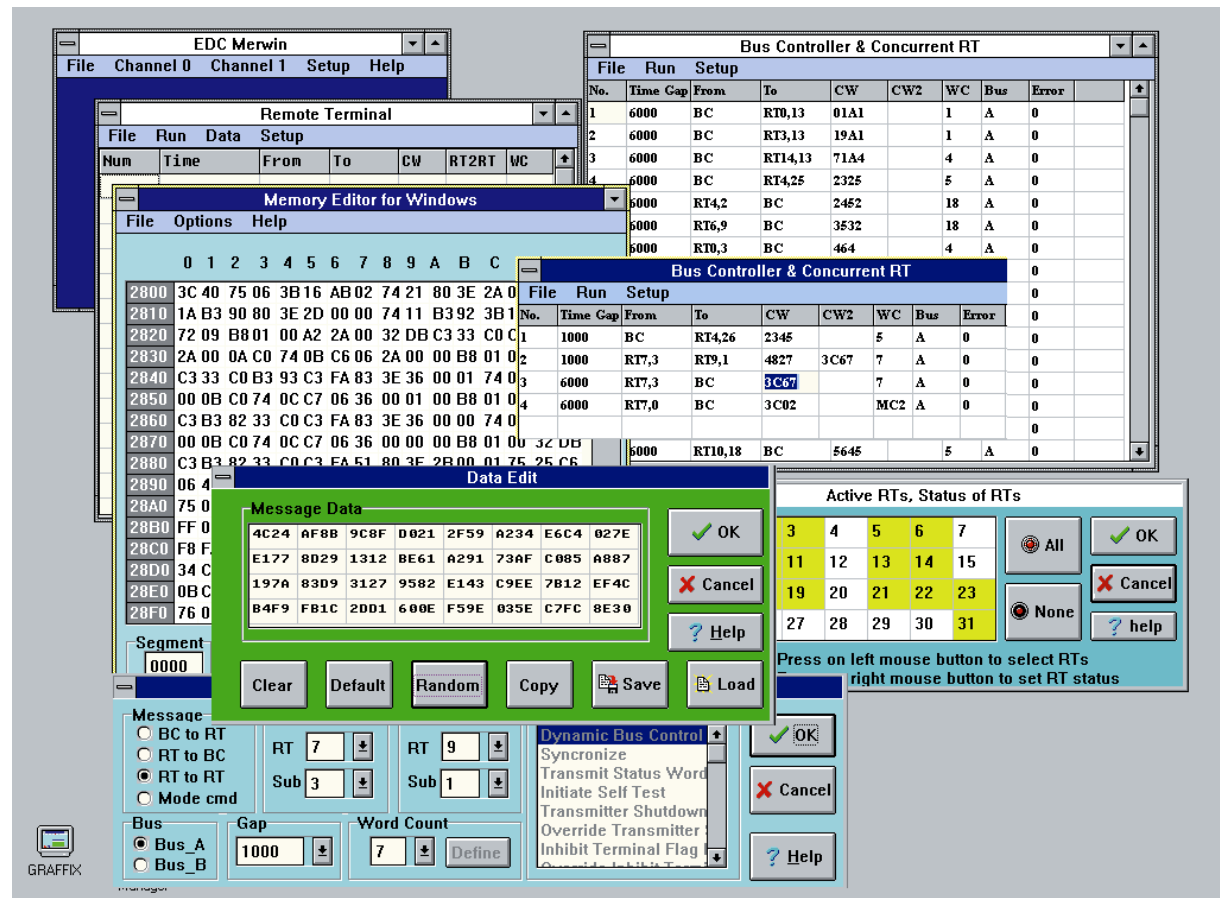


Figure 3-13 – Output Waveform

Software

- Software Applications

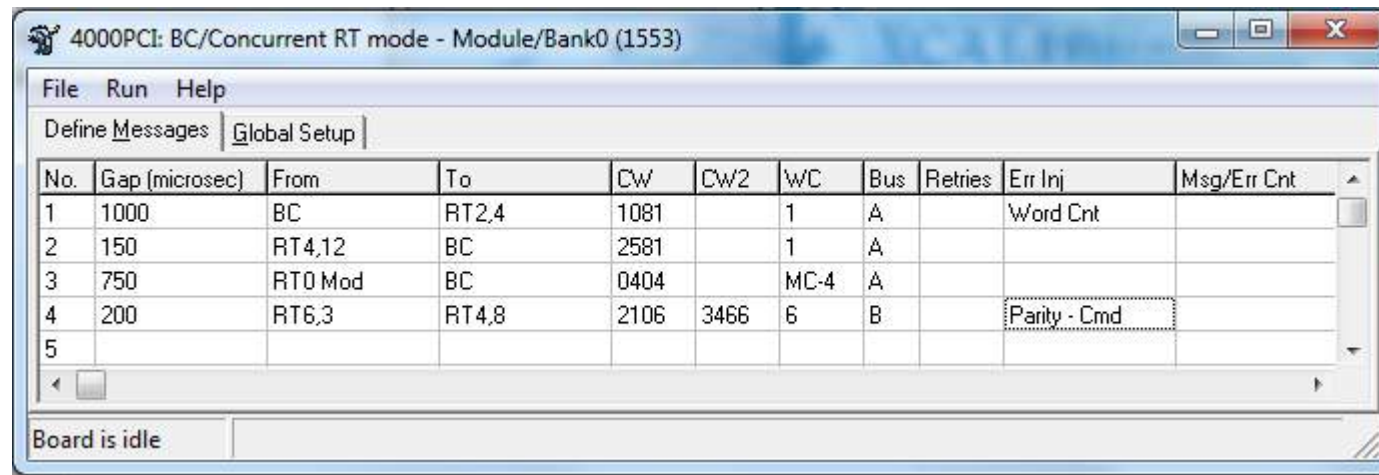


- Systems Integration
- RT Development
- Problem Isolation
- Post Flight Analysis

Systems Integration

- Run multiple RT's in Lab
 - Some RT's may be ready some not
 - Simulate Bus Controller and some RT's
 - Simulate Bus timing and errors
 - Monitor responses for timing, quality and correctness

RT Development



- Simulate BC for single message response
- Simulate other RT for RT to RT commands
- Inject errors to check response
- Alter intermessage timing for stress testing

Problem Isolation

- Reconstruct Bus activity in lab
 - Selectively simulate RT's
 - Match bus timing taken from in flight record
- Perform regression testing



Post Flight Analysis

- Analyze flight data for:
 - Health analysis – error statistics
 - Throughput analysis
 - Engineering data patterns
 - Indirect data analysis i.e., data comprised of other units, e.g.,
acceleration = speed Δ / time
 - Correlations between different data elements, e.g. temperature
relative to altitude

Access Board

SystemGroupsLibraryTransmit

4000PCI - EU names

H009 Flight Data

Flight Data (2)

Air Speed

Altitude

MsgType0

Dev0_Mod1

Multiplex Adapters - EU names

Dev0_Mod0

Unmapped Modules

Default

Default429

Dev0_Mod0

Dev1_Mod1

Dev25_Mod1

Dev25_Mod2

Test Flight 33333

Test Flight 43432

engine20

Fuel Flows

Oil Pressure

Oil Temperature

Revs

temperature

Flight Data

WorkPad

Laboratory Mode > Define > Modify Engineering Units

Name	Description	EU location	Units	Scale /Offset	Display	Discrete
H009 Flight Data Flight Data (2) (PU-8 CF-8 Receive)						
Air Speed		word:0 Bit:0 Len:16 Integer	Units	1.00 / 0	Range:0-10000 Alarm:0-0	
Altitude		word:1 Bit:0 Len:16 Integer	Units	1.00 / 0	Range:0-10000 Alarm:0-0	
H009 Flight Data MsgType0 (PU-2 CF-2 Receive)						
EU182		word:0 Bit:0 Len:16 Integer	Units	1.00 / 0	Range:0-10000 Alarm:0-0	
Test Flight 43432 engine20 (RT-5 SA-6 Receive)						
Fuel Flows	Flight Data	word:5 Bit:15 Len:16 Unsigned	RPM	1.00 / 0	Range:100-700 Alarm:55...	
Oil Pressure	engine 1	word:0 Bit:15 Len:16 Integer	PSI	1.00 / 0	Range:-1000-12000 Alar...	
Oil Temperature		word:1 Bit:15 Len:16 Integer	centigrade	333.00 / 0	Range:1-2 Alarm:3-4	
Revs	Flight Data	word:9 Bit:3 Len:4 Integer	degrees	1.00 / 0	Range:-10-10 Alarm:-5-5	Discrete
temperature	engine 1	word:2 Bit:15 Len:16 Integer	centigrade	1.00 / 0	Range:0-12000 Alarm:20-...	
Test Flight 43432 Flight Data (RT-7 SA-9 Receive)						
Air Speed11		word:0 Bit:0 Len:16 Integer	knots	1.00 / 0	Range:10000-32000 Alar...	
Air Speed11-1		word:0 Bit:0 Len:16 Integer	knots	1.00 / 2	Range:10000-32000 Alar...	
Altitude		word:8 Bit:0 Len:16 Integer	feet	1.00 / 0	Range:10000-32000 Alar...	Discrete
Greenwich Mean ...		word:0 Bit:0 Len:16 Integer	hours	1.00 / 0	Range:0-10000 Alarm:0-0	
Ground Speed	Flight Data	word:5 Bit:0 Len:16 Integer	miles/ho...	1.00 / 0	Range:0-500 Alarm:0-90	Discrete
PresentPosition....	Flight Data	word:2 Bit:0 Len:16 Integer	degrees	1.00 / 0	Range:0-360 Alarm:0-10	Discrete
PresentPosition....	Flight Data	word:9 Bit:3 Len:4 Integer	degrees	1.00 / 0	Range:-10-10 Alarm:-5-5	Discrete
Roll	Flight Data	word:15 Bit:0 Len:16 Integer	degrees	1.00 / 0	Range:0-360 Alarm:0-300	
Selected Course	Flight Data	word:15 Bit:0 Len:16 Integer	degrees	1.00 / 0	Range:0-360 Alarm:0-300	
Selected Heading	Flight Data	word:12 Bit:0 Len:16 Integer	degrees	1.00 / 0	Range:0-700 Alarm:150-5...	
Selected Mach		word:1 Bit:0 Len:16 Integer	kilomete	1.00 / 0	Range:0-0 Alarm:0-0	