

Formulation of the variational principle

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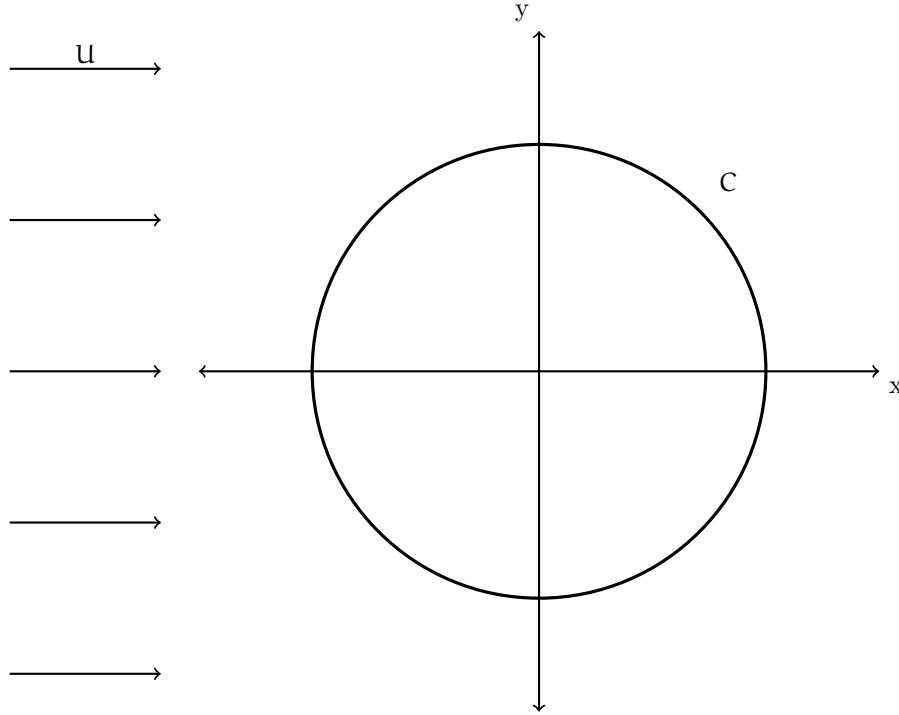


Figure 1: Sketch of the flow field

1 Introduction

The classical problem of steady inviscid subsonic flow past an aerofoil can be formulated in two ways. The usual formulation is as a boundary value problem consisting of a set non linear partial differential equations and a set of boundary conditions. However, it can also be formulated in terms of complementary variational principles. This report describes a variational method for obtaining numerical solutions. The method consists of replacing the infinitely dimensional variational problem by a finitely dimensional problem by means of finite differences, and an approximate maximizing function is then found by standard methods.

2 Formulation

The boundary value problem for plane subsonic flow past an aerofoil can be formulated as follows.

Let $\vec{u} = (u_1, u_2)$ be the velocity vector in the Cartesian coordinate system. Far from the aerofoil, $C \vec{u} = (U, 0)$ where U is a constant.

For an irrotational flow, a velocity potential can be defined as

$$\vec{u} = \nabla \phi$$

The pressure and density are defined by p and ρ respectively. The speed of

sound is defined by

$$c^2 = \frac{dp}{d\rho}$$

$$p = p_0 \left(1 - \frac{q^2}{2\beta c_0^2} \right)^\alpha$$

$$\rho = \rho_0 \left(1 - \frac{q^2}{2\beta c_0^2} \right)^\beta$$

where

$$q^2 = \vec{u} \cdot \vec{u}, \quad \alpha = \frac{\gamma}{\gamma - 1}, \quad \beta = \frac{1}{\gamma - 1}$$

the suffix 0 indicates stagnation values.

The boundary value problem for ϕ is equivalent to a variational problem of maximizing the integral

$$J[\phi] = \int_{R_1} p dV + \int_B \phi h dA \quad (1)$$

(dV = area element, dA = arc length element of $B = C$). Then $J[\phi]$ is a maximum if $\nabla \cdot (\rho \vec{u}) = 0$ and $\rho \vec{u} \cdot \hat{n} = h$ on B . Here the normal mass flu h is prescribes on B such that

$$\text{outflow} = \int_B h dA = 0$$

If the flow region R_1 becomes infinite, the variational integral (1) becomes unbounded. To remove this difficulty, the integral can be formulated as follows

$$J[\phi] = \int \int_{\infty} [p - p_{\infty} + p_{\infty} \cdot \nabla(\phi - \phi_{\infty})] dx dy \quad (2)$$

where

- p_{∞} = pressure at infinity,
- ρ_{∞} = density at infinity,
- ϕ_{∞} = potential for a uniform stream
- ϕ_0 = potential for incompressible flow past C .

To extend the method to a general class of aerofoils, a conformal mapping from the aerofoil C onto a unit circle can be used. Let $z = x + iy$ and $\sigma = r(\cos \theta + i \sin \theta)$, then,, the transform modulus is given by

$$T = \left| \frac{dz}{d\sigma} \right| = (x_r^2 + y_r^2)^{\frac{1}{2}}$$

The Jacobian of the transformation is

$$J = \frac{\partial(x, y)}{\partial(r, \theta)} = x_r y_{\theta} - x_{\theta} y_r$$

Since the transformation is conformal

$$y_\theta = rx_r \quad \text{and} \quad y_r = -\frac{1}{r}x_\theta$$

so

$$T^2 = x_r^2 + \frac{1}{r^2}x_\theta^2$$

and

$$J = r \left(x_r^2 + \frac{1}{r^2}x_\theta^2 \right)$$

Hence

$$J = rT^2 \quad (3)$$

The coordinates r, θ are orthogonal so the element of length $ds = |dz|$ is given by

$$ds^2 = h_1^2 dr^2 + h_2^2 d\theta^2$$

Also

$$\begin{aligned} ds^2 &= |dz|^2 = \left| \frac{dz}{d\sigma} \right|^2 |d\sigma|^2 \\ &= T^2 (dr^2 + r^2 d\theta^2) \end{aligned}$$

Therefore

$$h_1 = T \quad \text{and} \quad h_2 = rT$$

so

$$\nabla \phi = \frac{1}{T} \left(\hat{r} \phi_r + \frac{\hat{\theta}}{r} \phi_\theta \right)$$

Since

$$q^2 = (\nabla \phi)^2$$

and

$$\phi = U(r \cos \theta + \chi)$$

we have

$$q^2 = \frac{U^2}{T^2} \left[1 + 2 \cos \theta \cdot \chi_r - \frac{2}{r} \sin \theta \cdot \chi_\theta + \chi_r^2 + \frac{1}{r^2} \chi_\theta^2 \right] \quad (4)$$

Also

$$p = p_0 \left(1 - \frac{q^2}{2\beta c_0^2} \right)^\alpha$$

and by using the relation

$$p_0 = \left(\frac{\rho_0}{\rho_\infty} \right)^\gamma p_\infty$$

we get after some manipulation

$$p = p_\infty \left[1 + \frac{(\gamma - 1)M_\infty^2}{2T^2} \left(T^2 - 1 - 2 \cos \theta \cdot \chi_r + \frac{2}{r} \sin \theta \cdot \chi_\theta - \chi_r^2 - \frac{1}{r^2} \chi_\theta^2 \right) \right]^\alpha \quad (5)$$

where the free stream Mach number M_∞ is defined by

$$M_\infty^2 = \frac{2\beta U^2}{2\beta c_0^2 - U^2}$$

Now

$$\frac{\gamma p_\infty}{\rho_\infty} = c_0^2 - \frac{U^2}{2\beta}$$

Thus since

$$\phi_o = U \left(r + \frac{1}{r} \right) \cos \theta,$$

$$\rho_\infty \nabla \phi_o \cdot \nabla (\phi - \phi_\infty) = p_\infty \frac{\gamma M_\infty^2}{T^2} \left[\frac{r^2 - 1}{r^2} \cos \theta \chi_r - \frac{r^2 - 1}{r^3} \sin \theta \chi_\theta \right] \quad (6)$$

When the expressions (3), (5) and (6) are used in (2), the variational integral becomes

$$J[\chi] = p_\infty \int_0^{2\pi} \int_1^\infty \left\{ \left[1 + \frac{(\gamma - 1)M_\infty^2}{2T^2} \left(T^2 - 1 - 2 \cos \theta \chi_r + \frac{2}{r} \sin \theta \chi_\theta - \chi_r^2 - \frac{1}{r^2} \chi_\theta^2 \right) \right]^\alpha - 1 + \frac{\gamma M_\infty^2}{T^2} \left(\frac{r^2 - 1}{r^2} \cos \theta \chi_r - \frac{r^2 + 1}{r^3} \sin \theta \chi_\theta \right) \right\} r T^2 dr \quad (7)$$

The boundary conditions on χ are

$$\begin{aligned} \frac{\partial \chi}{\partial r} &= -\cos \theta & \text{at} & \quad r = 1, \\ \chi &= 0 \left(\frac{1}{r} \right) & \text{as} & \quad r \rightarrow \infty \end{aligned} \quad (8)$$

The local Mach number M and the local non dimensional pressure p_L are given by

$$M = M_\infty \frac{q}{U} \left[1 + \frac{1}{2}(\gamma - 1)M_\infty^2 \left(1 - \left(\frac{q}{U} \right)^2 \right) \right]^{\frac{1}{2}} \quad (9)$$

$$p_L = \frac{p}{p_\infty} = \left[1 + \frac{1}{2}(\gamma - 1)M_\infty^2 \left(1 - \left(\frac{q}{U} \right)^2 \right) \right]^{\frac{\gamma}{\gamma - 1}} \quad (10)$$

where q is given by equation (4)

The transform modulus for an ellipse is given by

$$T^2 = \frac{1}{r^4} \left[(r^2 + \lambda^2)^2 - 4\lambda^2 r^2 \cos^2 \theta \right] \quad (11)$$

where λ^2 is the following function of τ , the thickness ratio of the ellipse,

$$\lambda^2 = \frac{1 - \tau}{1 + \tau} \quad (12)$$

3 Numerical Method

The objective of the calculation is to find for given M_∞ and aerofoil shape a function χ which maximizes $J[\chi]$ as given by (7) subject to boundary conditions (8).

3.1 Domain

For non lifting symmetric bodies (about $y = 0$), it is only necessary to treat the interval $0 \leq \theta \leq \pi$.

Since the derivatives in both directions are approximated by finite differences, it is necessary to have a finite computation region. This is obtained by replacing the infinite integration limit on r by a finite limit R and insisting that the reduced potential χ satisfies an appropriate condition at $r = R$. The simplest condition to impose is that χ equals the reduced potential for incompressible flow at $r = R$.

Thus the variational integral reduces to

$$J[\chi] = p_\infty \int_0^\pi \int_1^R F(r, \theta, \chi_r, \chi_\theta) dr \quad (13)$$

where

$$F = \left\{ \left[1 + \frac{(\gamma - 1)M_\infty^2}{2T^2} \left(T^2 - 1 - 2 \cos \theta \chi_r + \frac{2}{r} \sin \theta \chi_\theta - \chi_r^2 - \frac{1}{r^2} \chi_\theta^2 \right) \right]^\alpha - 1 + \frac{\gamma M_\infty^2}{T^2} \left(\frac{r^2 - 1}{r^2} \cos \theta \chi_r - \frac{r^2 + 1}{r^3} \sin \theta \chi_\theta \right) \right\} r T^2$$

The boundary conditions are now

$$\begin{aligned} \frac{\partial \chi}{\partial \theta} &= 0 & \text{at} & \theta = 0, \pi \\ \frac{\partial \chi}{\partial r} &= -\cos \theta & \text{at} & r = 1 \\ \chi &= \frac{1}{R} \cos \theta & \text{at} & r = R \end{aligned} \quad (14)$$

3.2 Discretisation

$$\begin{aligned} r &= 1, \dots, R; & k_j &= r_{j+1} - r_j \\ \theta &= 0, \dots, \pi; & h_i &= \theta_{i+1} - \theta_i \end{aligned}$$

θ varies linearly from 0 to π . r is mapped as $r = 1/\sigma$, where σ varies linearly in the range $[1/R, 1]$.

Thus, the infinitely dimensional variational problem is now replaced by a finite-dimensional problem. Consider four neighbouring points as shown in figure (3). The derivatives of χ in the rectangle (i, j) can be approximated as

$$\frac{\partial \chi}{\partial \theta} = \frac{\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i,j+1}}{2h_i}$$

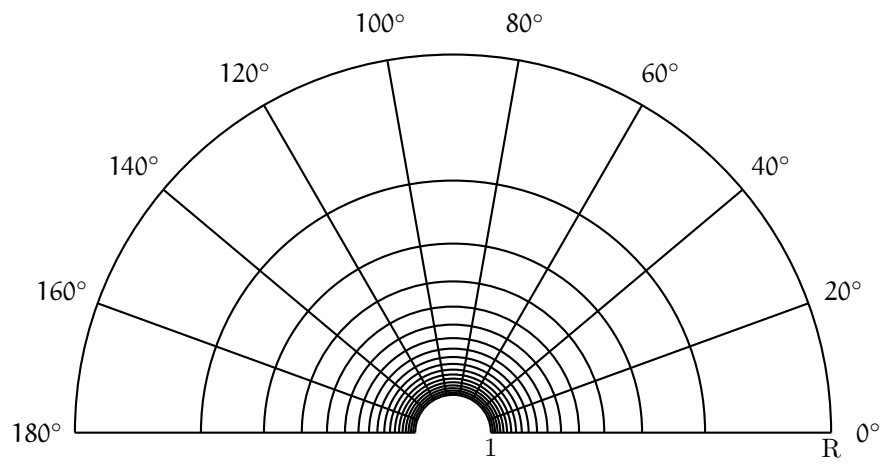


Figure 2: Mesh around the body

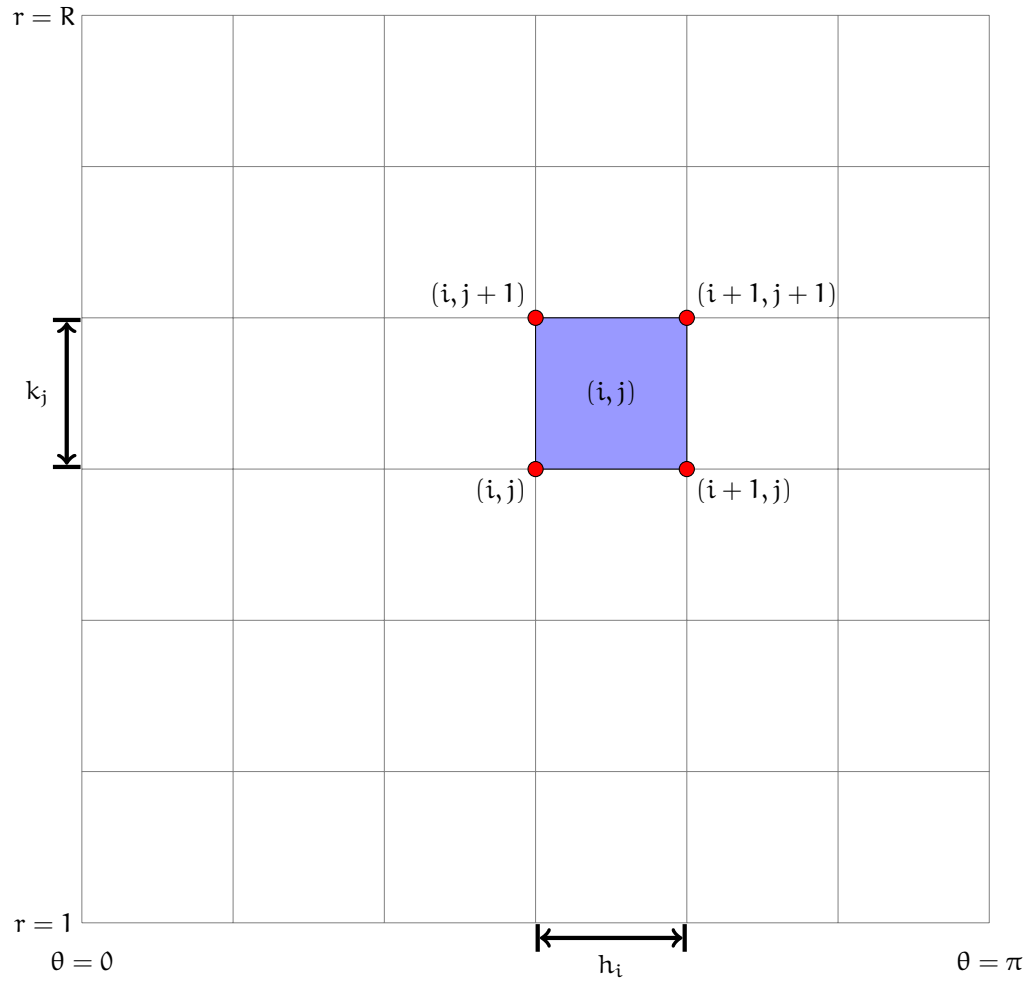


Figure 3: Computational domain

$$\frac{\partial \chi}{\partial \theta} = \frac{\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i+1,j}}{2k_j}$$

For a rectangular region i, j , $J[\chi]$ can be approximated as

$$\begin{aligned} J_{i,j} = p_\infty \left\{ \left[1 + \frac{(\gamma - 1)M_\infty^2}{2T_{i,j}^2} \left(T^2 - 1 - 2 \cos \theta'_i \frac{\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i+1,j}}{2k_j} \right. \right. \right. \\ \left. \left. + \frac{2}{r'_j} \sin \theta'_i \frac{\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i,j+1}}{2h_i} \right. \right. \\ \left. \left. - \left(\frac{\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i+1,j}}{2k_j} \right)^2 \right. \right. \\ \left. \left. - \frac{1}{r^2} \left(\frac{\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i,j+1}}{2h_i} \right)^2 \right) \right]^\alpha \\ - 1 + \frac{\gamma M_\infty^2}{T_{i,j}^2} \left(\left(\frac{r_j'^2 - 1}{r_j'^2} \right) \cos \theta'_i \frac{\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i+1,j}}{2k_j} \right. \\ \left. - \left(\frac{r_j'^2 - 1}{r_j'^3} \right) \sin \theta'_i \frac{\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j} - \chi_{i,j+1}}{2h_i} \right) \right\} \quad (15) \\ \theta'_i = \theta_i + 0.5h_i; \quad r'_j = r_j + 0.5k_j \end{aligned}$$

$$T_{i,j}^2 = \frac{1}{r_j'^4} [(r_j'^2 + \lambda^2)^2 - 4\lambda^2 r_j'^2 \cos^2 \theta'_i]$$

$$\lambda^2 = \frac{1 - \tau}{1 + \tau}$$

Boundary conditions for χ -

1.

$$\frac{\partial \chi}{\partial r} = -\cos \theta; \text{ at } r = 1$$

$$\chi_{i,1} = \frac{1}{k_2 + 2k_1} \left[\frac{1}{k_2} ((k_1 + k_2)^2 \chi_{1,2} - k_1^2 \chi_{i,3}) + k_1 (k_1 + k_2) \cos \theta \right] \quad (16)$$

2.

$$\frac{\partial \chi}{\partial \theta} = 0 \text{ at } \theta = 0, \pi$$

$$\chi_{m+1,j} = \chi_{m-1,j} \quad \chi_{0,j} = \chi_{2,j} \quad (17)$$

3.

$$\chi = \frac{1}{R} \cos \theta \text{ as } r \rightarrow \infty$$

$$\chi_{i,n} = \frac{1}{R} \cos \theta_i \quad (18)$$

Summing the contributions for each rectangle, $J[\chi]$ can be approximated as

$$J[\chi] = \bar{J} = \sum_{i=1}^{n-1} \sum_{j=1}^{m-1} J_{i,j} \quad (19)$$

Therefore, the values of $\chi_{i,j}$ which maximizes this expression is given by the solutions of the equations

$$\begin{aligned} \frac{\partial \bar{J}}{\partial \chi_{i,j}} &= 0 \\ i &= 1, \dots, n \\ j &= 2, \dots, m-1 \end{aligned} \quad (20)$$

for a given i, j , the equation to maximize can be written in the form (by summing the contributions from the four surrounding rectangles)

$$g(\chi_{i,j}) = \frac{\partial \bar{J}}{\partial \chi_{i,j}} \sum_{s=(i,j),(i-1,j),(i,j-1),(i-1,j-1)} \left[\alpha (A_s \chi_{i,j}^2 + B_s \chi_{i,j} + C_s)^{\alpha-1} (2A_s \chi_{i,j} + B_s) + D_s \right] H_s = 0 \quad (21)$$

Using Newton Raphson method, an improved estimate is given by

$$\chi_{i,j}^{(q+1)} = \chi_{i,j}^{(q)} - \frac{g(\chi_{i,j}^{(q)})}{g'(\chi_{i,j}^{(q)})} \quad (22)$$

$$g' = \frac{\partial g_{i,j}}{\partial \chi_{i,j}} \quad (23)$$

3.3 Coefficients A, B, C, D, H

The coefficients A, B, C, D, H are given by -

$$\begin{aligned} A_{i,j} &= -\frac{(\gamma-1)M_\infty^2}{2T_{i,j}^2} \left(\frac{1}{4k_j^2} + \frac{1}{4r_j'^2 h_i} \right) \\ A_{i,j-1} &= -\frac{(\gamma-1)M_\infty^2}{2T_{i,j-1}^2} \left(\frac{1}{4k_{j-1}^2} + \frac{1}{4r_{j-1}'^2 h_i} \right) \\ A_{i-1,j} &= -\frac{(\gamma-1)M_\infty^2}{2T_{i-1,j}^2} \left(\frac{1}{4k_j^2} + \frac{1}{4r_j'^2 h_{i-1}} \right) \\ A_{i-1,j-1} &= -\frac{(\gamma-1)M_\infty^2}{2T_{i-1,j-1}^2} \left(\frac{1}{4k_{j-1}^2} + \frac{1}{4r_{j-1}'^2 h_{i-1}} \right) \end{aligned} \quad (24)$$

$$\begin{aligned}
B_{i,j} &= \frac{(\gamma-1)M_\infty^2}{2T_{i,j}^2} \left(2 \cos \theta'_i \frac{1}{2k_j} - \frac{2}{r'_j} \sin \theta'_i \frac{1}{2h_i} \right. \\
&\quad \left. + \frac{1}{2k_j^2} (\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i+1,j}) \right. \\
&\quad \left. + \frac{1}{2r_j'^2 h_i^2} (\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j+1}) \right) \\
B_{i,j-1} &= \frac{(\gamma-1)M_\infty^2}{2T_{i,j-1}^2} \left(-2 \cos \theta'_i \frac{1}{2k_{j-1}} - \frac{2}{r'_{j-1}} \sin \theta'_i \frac{1}{2h_i} \right. \\
&\quad \left. - \frac{1}{2k_{j-1}^2} (\chi_{i+1,j} - \chi_{i,j-1} - \chi_{i+1,j-1}) \right. \\
&\quad \left. + \frac{1}{2r_{j-1}'^2 h_i^2} (\chi_{i+1,j-1} + \chi_{i+1,j} - \chi_{i,j-1}) \right) \\
B_{i-1,j} &= \frac{(\gamma-1)M_\infty^2}{2T_{i-1,j}^2} \left(2 \cos \theta'_{i-1} \frac{1}{2k_j} + \frac{2}{r'_j} \sin \theta'_{i-1} \frac{1}{2h_{i-1}} \right. \\
&\quad \left. + \frac{1}{2k_j^2} (\chi_{i-1,j+1} + \chi_{i,j+1} - \chi_{i-1,j}) \right. \\
&\quad \left. - \frac{1}{2r_j'^2 h_{i-1}^2} (\chi_{i,j+1} - \chi_{i-1,j+1} - \chi_{i-1,j+1}) \right) \\
B_{i-1,j-1} &= \frac{(\gamma-1)M_\infty^2}{2T_{i-1,j-1}^2} \left(-2 \cos \theta'_{i-1} \frac{1}{2k_{j-1}} + \frac{2}{r'_{j-1}} \sin \theta'_{i-1} \frac{1}{2h_{i-1}} \right. \\
&\quad \left. - \frac{1}{2k_{j-1}^2} (\chi_{i-1,j} - \chi_{i-1,j-1} - \chi_{i,j-1}) \right. \\
&\quad \left. - \frac{1}{2r_{j-1}'^2 h_{i-1}^2} (\chi_{i,j-1} - \chi_{i-1,j-1} - \chi_{i-1,j}) \right)
\end{aligned} \tag{25}$$

$$\begin{aligned}
C_{i,j} &= 1 + \frac{(\gamma-1)M_\infty^2}{2T_{i,j}^2} \left(T_{i,j}^2 - 1 - \frac{\cos \theta'_i}{k_j} (\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i+1,j}) \right. \\
&\quad + \frac{\sin \theta'_i}{r'_j h_i} (\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j+1}) \\
&\quad - \frac{1}{4k_j^2} (\chi_{i,j+1} + \chi_{i+1,j+1} - \chi_{i+1,j})^2 \\
&\quad \left. - \frac{1}{4r_j'^2 h_i^2} (\chi_{i+1,j} + \chi_{i+1,j+1} - \chi_{i,j+1})^2 \right) \\
C_{i,j-1} &= 1 + \frac{(\gamma-1)M_\infty^2}{2T_{i,j-1}^2} \left(T_{i,j-1}^2 - 1 - \frac{\cos \theta'_i}{k_{j-1}} (\chi_{i+1,j} - \chi_{i,j-1} - \chi_{i+1,j-1}) \right. \\
&\quad + \frac{\sin \theta'_i}{r'_{j-1} h_i} (\chi_{i+1,j-1} + \chi_{i+1,j} - \chi_{i,j-1}) \\
&\quad - \frac{1}{4k_{j-1}^2} (\chi_{i+1,j} - \chi_{i,j-1} - \chi_{i+1,j-1})^2 \\
&\quad \left. - \frac{1}{4r_{j-1}'^2 h_i^2} (\chi_{i+1,j-1} + \chi_{i+1,j} - \chi_{i,j-1})^2 \right) \\
C_{i-1,j} &= 1 + \frac{(\gamma-1)M_\infty^2}{2T_{i-1,j}^2} \left(T_{i-1,j}^2 - 1 - \frac{\cos \theta'_{i-1}}{k_j} (\chi_{i-1,j+1} + \chi_{i,j+1} - \chi_{i-1,j}) \right. \\
&\quad + \frac{\sin \theta'_{i-1}}{r'_j h_i} (\chi_{i,j+1} - \chi_{i-1,j+1} - \chi_{i-1,j+1}) \\
&\quad - \frac{1}{4k_j^2} (\chi_{i-1,j+1} + \chi_{i,j+1} - \chi_{i-1,j})^2 \\
&\quad \left. - \frac{1}{4r_j'^2 h_{i-1}^2} (\chi_{i,j+1} - \chi_{i-1,j+1} - \chi_{i-1,j+1})^2 \right) \\
C_{i-1,j-1} &= 1 + \frac{(\gamma-1)M_\infty^2}{2T_{i-1,j-1}^2} \left(T_{i-1,j-1}^2 - 1 - \frac{\cos \theta'_{i-1}}{k_{j-1}} (\chi_{i-1,j} - \chi_{i-1,j-1} - \chi_{i,j-1}) \right. \\
&\quad + \frac{\sin \theta'_{i-1}}{r'_{j-1} h_{i-1}} (\chi_{i,j-1} - \chi_{i-1,j-1} - \chi_{i-1,j}) \\
&\quad - \frac{1}{4k_{j-1}^2} (\chi_{i-1,j} - \chi_{i-1,j-1} - \chi_{i,j-1})^2 \\
&\quad \left. - \frac{1}{4r_{j-1}'^2 h_{i-1}^2} (\chi_{i,j-1} - \chi_{i-1,j-1} - \chi_{i-1,j})^2 \right)
\end{aligned} \tag{26}$$

$$\begin{aligned}
D_{i,j} &= -1 + \frac{\gamma M_\infty^2}{T_{i,j}^2} \left[- \left(\frac{r_j'^2 - 1}{r_j'^2} \right) \frac{\cos \theta_i'}{2k_j} + \left(\frac{r_j'^2 - 1}{r_j'^3} \right) \frac{\sin \theta_i'}{2h_i} \right] \\
D_{i,j-1} &= -1 + \frac{\gamma M_\infty^2}{T_{i,j-1}^2} \left[\left(\frac{r_{j-1}'^2 - 1}{r_{j-1}'^2} \right) \frac{\cos \theta_i'}{2k_{j-1}} + \left(\frac{r_{j-1}'^2 - 1}{r_{j-1}'^3} \right) \frac{\sin \theta_i'}{2h_i} \right] \\
D_{i-1,j} &= -1 + \frac{\gamma M_\infty^2}{T_{i-1,j}^2} \left[- \left(\frac{r_j'^2 - 1}{r_j'^2} \right) \frac{\cos \theta_{i-1}'}{2k_j} - \left(\frac{r_j'^2 - 1}{r_j'^3} \right) \frac{\sin \theta_{i-1}'}{2h_{i-1}} \right] \\
D_{i-1,j-1} &= -1 + \frac{\gamma M_\infty^2}{T_{i-1,j-1}^2} \left[\left(\frac{r_{j-1}'^2 - 1}{r_{j-1}'^2} \right) \frac{\cos \theta_{i-1}'}{2k_{j-1}} - \left(\frac{r_{j-1}'^2 - 1}{r_{j-1}'^3} \right) \frac{\sin \theta_{i-1}'}{2h_{i-1}} \right]
\end{aligned} \tag{27}$$

$$\begin{aligned}
H_{i,j} &= r_j' T_{i,j}^2 h_i k_j \\
H_{i,j-1} &= r_{j-1}' T_{i,j-1}^2 h_i k_{j-1} \\
H_{i-1,j} &= r_j' T_{i-1,j}^2 h_{i-1} k_j \\
H_{i-1,j-1} &= r_{j-1}' T_{i-1,j-1}^2 h_{i-1} k_{j-1}
\end{aligned} \tag{28}$$