

TOWARD A GAUSSIAN MIXTURE MODEL FOR CHANCE CONSTRAINED ORBITAL TRANSFER OPTIMAL CONTROL

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MOTIVATION

- Spacecraft often must undergo orbital transfers
- Optimal control is used to determine the solution
- Control solution depends on the state
- State uncertainty is guaranteed
- Can represent the state uncertainty with a Gaussian
- Works for two body problem, run into problems with 3-body

OVERVIEW OF PROJECT



BUILDING OFF CURRENT RESEARCH

- Oguri et al. give a solution for the constrained optimal control problem, solved with indirect approach
 - Derive the necessary conditions with a Gaussian model of the state
- Choose to build off this research with a Gaussian mixture model, to capture higher order moments of the state uncertainty.

DERIVING THE NECESSARY CONDITIONS

- The first half of the semester was devoted to deriving the necessary conditions for the Gaussian Mixture model.
- Stochastic problem to deterministic problem
- New state representation with pdf
- Path constraint is computed with unscented transform
- Leads to optimal control problem
- Augment state with $K \times n$ dimensions, for each component in Gaussian mixture
- Augment Covariance with $K \times n \times K \times n$ dimensions

$$p(t, x) = \sum_{i=1}^K w_i p_{g_i}$$

$$p_{g_i} = \mathcal{N}(x(t) | \mu_i, P_i)$$

$$\Pr[h(x, t) \leq 0] \geq \alpha$$

$$[\bar{h}_i, P_{h,i}] = UT(\mu_i, P_i, h(x, t))$$

$$\bar{h} = \sum_{i=1}^K w_i \bar{h}_i$$

$$P_h = \sum_{i=1}^K w_i P_{h,i}$$

DETERMINISTIC OPTIMAL CONTROL PROBLEM

$$J = \int_{t_0}^{t_f} L dt$$

$$\dot{\mu} = f(\mu, m, u, t), \dot{m} = -\Gamma \dot{m}_{max}$$

$$\dot{P} = AP + PA^T + GG^T$$

$$\psi_s \leq 0$$

$$\psi_0(\mu_i, t)|_{(t_0)} = 0$$

$$\psi_f(\mu_i, t)|_{(t_f)} = 0$$

Where $\mu \in R^{Kn_x}$ and $P \in R^{Kn_x \times Kn_x}$

Time optimal transfer, $L = 1$

IMPLEMENTATION OF UNCONSTRAINED CASE, GAUSSIAN MODEL

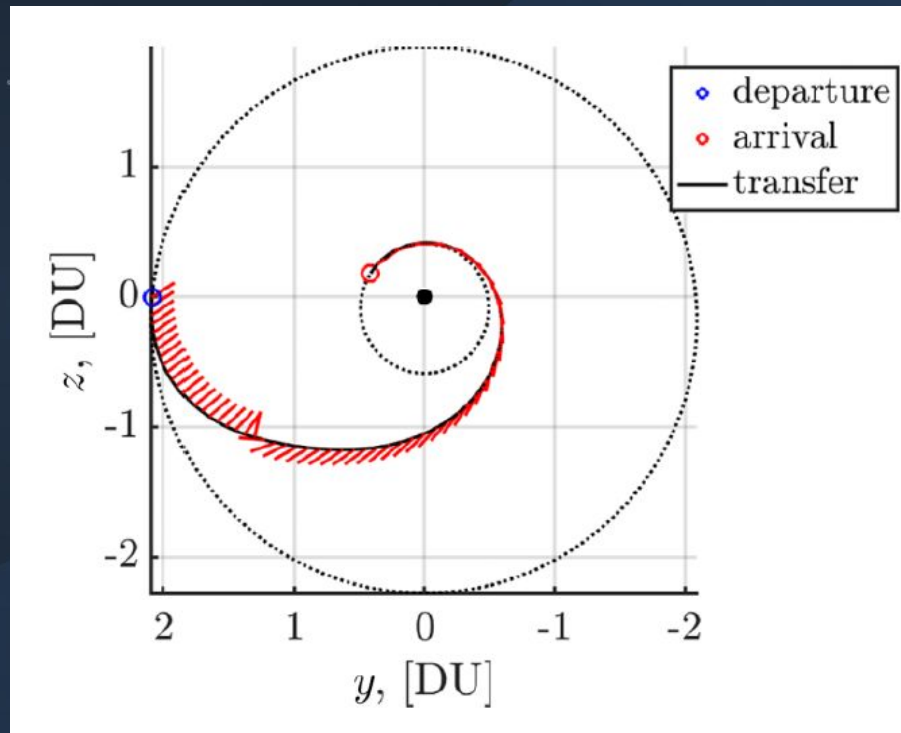
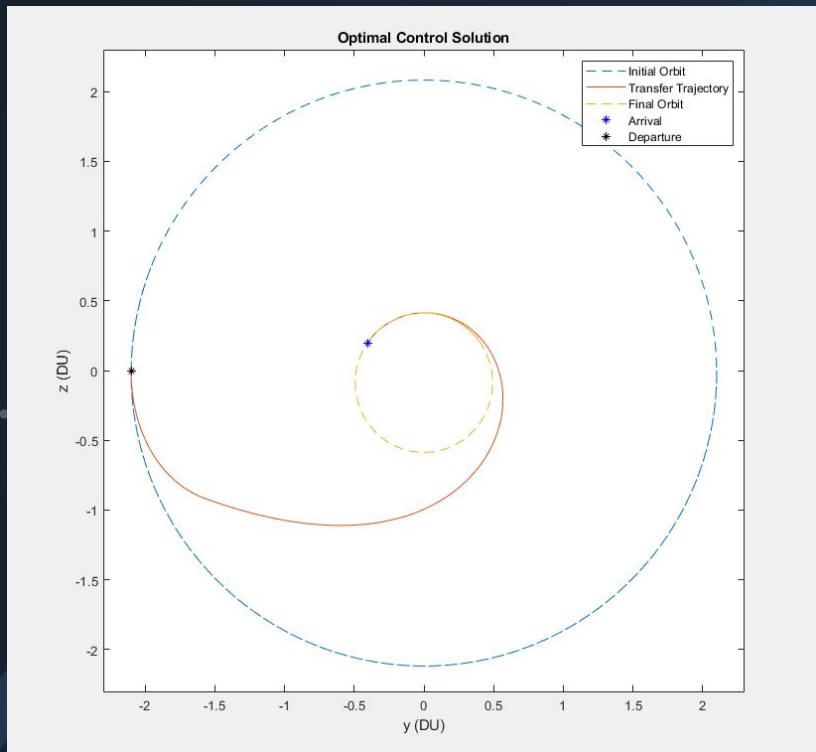
- Solver was implemented using MATLAB
- The 50 coupled ODEs were propagated with ODE45
- The TPBVP was solved with fsolve
- Used same test case as Oguri et. al.:
- Spacecraft Maneuver orbiting an asteroid

semi major axis: 2 km \rightarrow 0.5 km

eccentricity: 0.173 \rightarrow 0.085

true anomaly: free

SOLUTION



FUTURE WORK

- Finalize writeup for development of the Gaussian Mixture necessary conditions
- Now that unconstrained indirect solver is working, add constraint equations to necessary condition propagation which uses the Unscented Transform
- Implement the Gaussian Mixture model on the constrained indirect solver.



THANK YOU FOR YOUR TIME

Any questions?