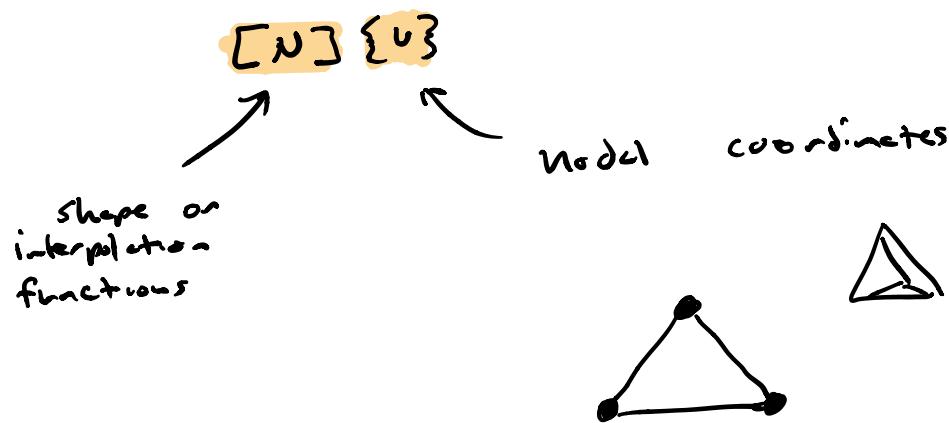


- 1) Generalized definition of elemental stiffness and global stiffness]
- 2) Energy derivation of stiffness eqn]
- 3) Truss structures]
- 4) Beams]
- 5) Fusion 360 Demo

$$u(x) = (1 - x/L)u_1 + (x/L)u_2$$

$$u(x) = [N_1(x) \quad N_2(x)] \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = [N] \{u\}$$

We express the continuous displacement across each element as



$[N] \{u\}$ allows us to estimate the true deflection within all elements

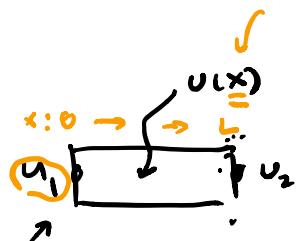
Let's derive the stiffness equation from the shape functions this time:

We want to solve equilibrium equations
thus: want nodal displacements vs forces

Displacement \rightarrow strain

Strain \rightarrow stress

Stress \rightarrow loading



$$\underline{\epsilon_x} = \frac{\partial u(x)}{\partial x}$$

$$\begin{aligned}\underline{\epsilon_x} &= -\frac{u_1}{L} + \frac{u_2}{L} \\ &= \frac{u_2 - u_1}{L}\end{aligned}$$

$$\begin{aligned}u(x) &= [N_1(x) \quad N_2(x)] \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} \\ &= \left(1 - \frac{x}{L}\right) u_1 + \frac{x}{L} u_2 \\ u(0) &= u_1 \\ u(L) &= u_2\end{aligned}$$

$$\underline{\sigma_x} = E \underline{\epsilon_x} = E \left(\frac{u_2 - u_1}{L} \right)$$

$$P = A \epsilon \left(\frac{u_2 - u_1}{L} \right)$$

$$\boxed{P_1} \rightarrow_2 \quad P_1 = -\frac{AE}{L} (u_2 - u_1)$$

$$P_2 = \frac{AE}{L} (u_2 - u_1)$$

$$\xrightarrow{\quad} \frac{AE}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} r_1 \\ r_2 \end{Bmatrix}$$

We derive elemental stiffness matrices by taking appropriate derivatives of $u(x)$ and considering the elastic behavior of material

Minimum potential energy formulation
of equilibrium equations:

$$\boxed{1D} \quad [L]_1 = \frac{\partial}{\partial x} \quad \frac{\partial^2 u(x)}{\partial x^2}$$

$$\underline{\varepsilon} = [L] \frac{u(x)}{x}$$

shape fcn

$$u(x) = [N(x)] \{u\}$$

$$\boxed{2D} \quad [L] = \begin{bmatrix} 2/\partial x & 0 \\ 0 & 2/\partial y \\ 2/\partial y & 2/\partial x \end{bmatrix}$$

2D derivative

$$\underline{\varepsilon} = [L] [N(x)] \{u\}$$

$$\boxed{1D} \quad \{u\} = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

order

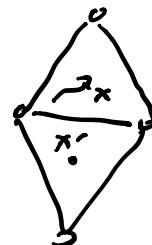
Shape function

Nodal displacements

derivative operator

$$\underline{\sigma} = [E] \underline{\varepsilon} \quad \text{Elasticity matrix}$$

$$\underline{\sigma} = [E] [L] [N(x)] \{u\}$$



[B]

$$\underline{\sigma} = [E] [B] \{u\} \leftarrow$$

$$\Pi_e = \frac{1}{2} \int_{V_e} \underline{\sigma}^T \underline{\varepsilon} dv - \int_{V_e} \underline{\varepsilon} \underline{u}^T p dv - \int_{S_e} \underline{\varepsilon} \underline{u}^T g ds$$

Strain energy

Body forces

Surface forces

$$T_{le} = \frac{1}{2} \int_{V_e} \underline{\underline{\epsilon}}^T [B] [\epsilon] [B] \underline{\underline{\epsilon}} \delta V$$

↗ $- \int_V \underline{\underline{\epsilon}}^T p \delta V - \int_{S_{e-n}} \underline{\underline{\epsilon}}^T q \delta s$

$$\frac{\partial \pi_e}{\partial u_e} = 0$$

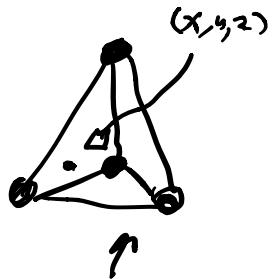
→ $\left(\int_{V_e} [B]^T [\epsilon] [B] \delta V \right) \underline{\underline{\epsilon}} - F_e = 0$

$$[K] \underline{\underline{\epsilon}} = \underline{\underline{F}}$$

$$[B] = [L][N]$$

$$[K] = \int_{V_e} [N] [L] [\epsilon] [L] [N] \delta V$$

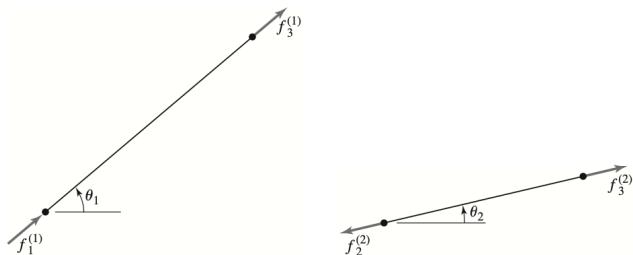
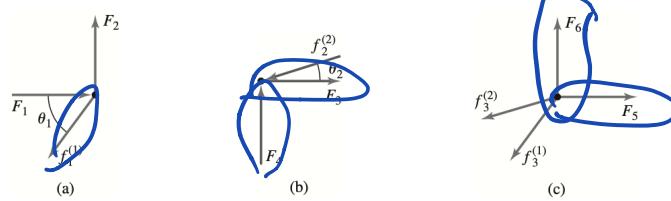
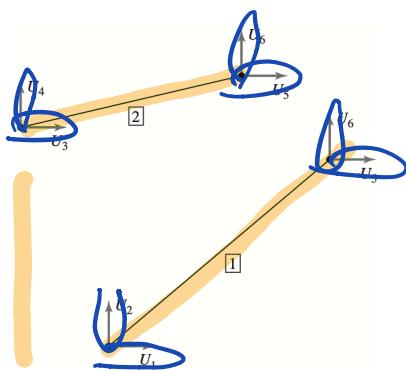
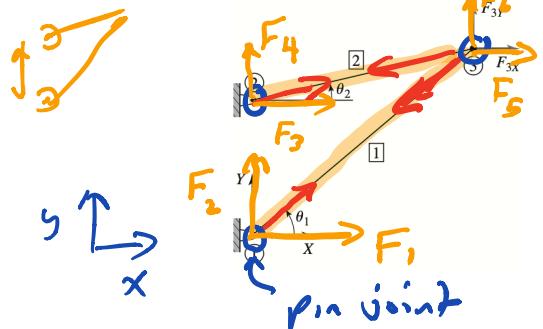
↑ ↗
shape functions



$$[k_e] = \left(\int_{V_e} [B]^T [\epsilon] [B] \delta V \right)$$

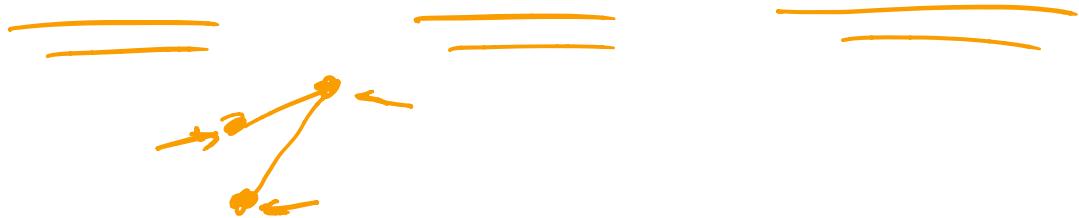
Move on from 1D bars \rightarrow 2D configurations of bars

Truss problems



$$F_1 - f_1^{(1)} \cos \theta_1 = 0 \quad F_3 - f_2^{(2)} \cos \theta_2 = 0 \quad F_5 - f_3^{(1)} \cos \theta_1 - f_3^{(2)} \cos \theta_2 = 0$$

$$F_2 - f_1^{(1)} \sin \theta_1 = 0 \quad F_4 - f_2^{(2)} \sin \theta_2 = 0 \quad F_6 - f_3^{(1)} \sin \theta_1 - f_3^{(2)} \sin \theta_2 = 0$$



$U_1^{(e)}$ = element node 1 displacement in the global X direction
 $U_2^{(e)}$ = element node 1 displacement in the global Y direction
 $U_3^{(e)}$ = element node 2 displacement in the global X direction
 $U_4^{(e)}$ = element node 2 displacement in the global Y direction

$$\begin{aligned}
 u_1^{(e)} &= U_1^{(e)} \cos \theta + U_2^{(e)} \sin \theta \\
 v_1^{(e)} &= -U_1^{(e)} \sin \theta + U_2^{(e)} \cos \theta \\
 u_2^{(e)} &= U_3^{(e)} \cos \theta + U_4^{(e)} \sin \theta \\
 v_2^{(e)} &= -U_3^{(e)} \sin \theta + U_4^{(e)} \cos \theta
 \end{aligned}$$

$-k^{(1)}[(U_5 - U_1)\cos \theta_1 + (U_6 - U_2)\sin \theta_1]\cos \theta_1 = F_1$
 $-k^{(1)}[(U_5 - U_1)\cos \theta_1 + (U_6 - U_2)\sin \theta_1]\sin \theta_1 = F_2$
 $-k^{(2)}[(U_5 - U_3)\cos \theta_2 + (U_6 - U_4)\sin \theta_2]\cos \theta_2 = F_3$
 $-k^{(2)}[(U_5 - U_3)\cos \theta_2 + (U_6 - U_4)\sin \theta_2]\sin \theta_2 = F_4$
 $k^{(2)}[(U_5 - U_3)\cos \theta_2 + (U_6 - U_4)\sin \theta_2]\cos \theta_2$
 $+ k^{(1)}[(U_5 - U_3)\cos \theta_1 + (U_6 - U_4)\sin \theta_1]\cos \theta_1 = F_5$
 $k^{(2)}[(U_5 - U_3)\cos \theta_2 + (U_6 - U_4)\sin \theta_2]\sin \theta_2$
 $+ k^{(1)}[(U_5 - U_3)\cos \theta_1 + (U_6 - U_4)\sin \theta_1]\sin \theta_1 = F_6$

$$c_{\theta_1} \rightarrow \cos \theta_1, \quad s_{\theta_1} \rightarrow \sin \theta_1,$$

$$\underbrace{\begin{bmatrix} k^{(1)}c^2\theta_1 & k^{(1)}s\theta_1c\theta_1 & 0 & 0 & -k^{(1)}c^2\theta_1 & -k^{(1)}s\theta_1c\theta_1 \\ k^{(1)}s\theta_1c\theta_1 & k^{(1)}s^2\theta_1 & 0 & 0 & -k^{(1)}s\theta_1c\theta_1 & -k^{(1)}s^2\theta_1 \\ 0 & 0 & k^{(2)}c^2\theta_2 & k^{(2)}s\theta_2c\theta_2 & -k^{(2)}c^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 \\ 0 & 0 & k^{(2)}s\theta_2c\theta_2 & k^{(2)}s^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 & -k^{(2)}s^2\theta_2 \\ -k^{(1)}c^2\theta_{12} & -k_1s\theta_1c\theta_1 & -k^{(2)}c^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 & k^{(1)}c^2\theta_1 + k^{(2)}c^2\theta_2 & k^{(1)}s\theta_1c\theta_1 + k^{(2)}s\theta_2c\theta_2 \\ -k_1s\theta_1c\theta_1 & -k^{(1)}s^2\theta_1 & -k^{(2)}s\theta_2c\theta_2 & -k^{(2)}s^2\theta_2 & k^{(1)}s\theta_1c\theta_1 + k^{(2)}s\theta_2c\theta_2 & k^{(1)}s^2\theta_1 + k^{(2)}s^2\theta_2 \end{bmatrix}}_{6 \times 1} = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \\ U_5 \\ U_6 \end{bmatrix} = \begin{bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \\ F_5 \\ F_6 \end{bmatrix}$$

$$[K] \{u\} = \{F\}$$

2 m

$k_{ij}^{(e)}$

$$[K^e] = \frac{AE}{L} \begin{bmatrix} c^2 & cs & -c^2 & -cs \\ cs & s^2 & -cs & -s^2 \\ -c^2 & -cs & c^2 & cs \\ -cs & -s^2 & cs & s^2 \end{bmatrix}$$

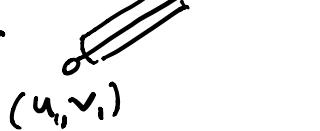
Elemental
Stiffness
matrix
for a
truss

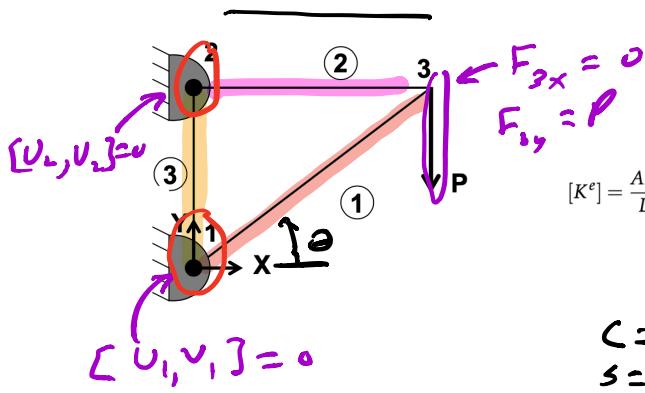
$A = \text{Area}$

$E = \text{Young's mod.}$

$L = \text{Length}$

$$\frac{AE}{L}$$





$$[K^e] = \frac{AE}{L} \begin{bmatrix} c^2 & cs & -c^2 & -cs \\ cs & s^2 & -cs & -s^2 \\ -c^2 & -cs & c^2 & cs \\ -cs & -s^2 & cs & s^2 \end{bmatrix}$$

$$c = \cos \alpha$$

$$s = \sin \alpha$$

$$K_1 = \begin{bmatrix} \frac{k_1}{2} & \frac{k_1}{2} & -\frac{k_1}{2} & -\frac{k_1}{2} \\ \frac{k_1}{2} & \frac{k_1}{2} & -\frac{k_1}{2} & \frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & \frac{k_1}{2} & \frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & \frac{k_1}{2} & -\frac{k_1}{2} \end{bmatrix},$$

$$K_2 = \begin{bmatrix} k_2 & 0 & -k_2 & 0 \\ 0 & 0 & 0 & 0 \\ -k_2 & 0 & k_2 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix},$$

$$K_3 = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & k_3 & 0 & -k_3 \\ 0 & 0 & 0 & 0 \\ 0 & -k_3 & 0 & k_3 \end{bmatrix},$$

$$[K] = \left[\begin{array}{cc|cc} \frac{k_1}{2} & \frac{k_1}{2} & 0 & 0 \\ \frac{k_1}{2} & \frac{k_1}{2} & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ -\frac{k_1}{2} & -\frac{k_1}{2} & 0 & 0 \\ -\frac{k_1}{2} & -\frac{k_1}{2} & 0 & 0 \end{array} \right] \left\{ \begin{array}{l} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{array} \right\} = \left\{ \begin{array}{l} F_{x1} \\ F_{y1} \\ 0 \\ 0 \\ 0 \\ -P \end{array} \right\},$$

$$[K_1] = \left[\begin{array}{cc|cc} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & k_2 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & -k_2 & 0 \\ 0 & 0 & 0 & 0 \end{array} \right] \left\{ \begin{array}{l} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{array} \right\} = \left\{ \begin{array}{l} 0 \\ 0 \\ F_{x2} \\ F_{y2} \\ 0 \\ 0 \end{array} \right\},$$

$$[K_3] = \left[\begin{array}{cc|cc} 0 & 0 & 0 & 0 \\ 0 & k_3 & 0 & -k_3 \\ 0 & 0 & 0 & 0 \\ 0 & -k_3 & 0 & k_3 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{array} \right] \left\{ \begin{array}{l} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{array} \right\} = \left\{ \begin{array}{l} F_{x1} \\ F_{y1} \\ F_{x2} \\ F_{y2} \\ 0 \\ -P \end{array} \right\}.$$

$[K] \{u\} = \{F\}$ Not complete

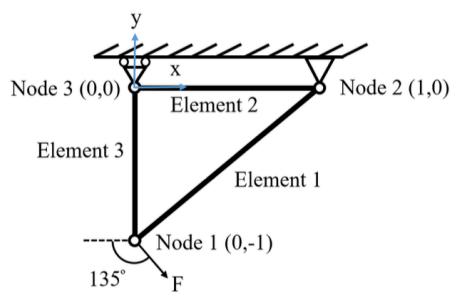
$$\left[\begin{array}{cc|cc} k_2 + \frac{k_1}{2} & \frac{k_1}{2} & u_3 \\ \frac{k_1}{2} & \frac{k_1}{2} & v_3 \end{array} \right] \left\{ \begin{array}{l} u_3 \\ v_3 \end{array} \right\} = \left\{ \begin{array}{l} 0 \\ -P \end{array} \right\}$$

$$(K) \{u\} = \{F\}$$

$$\{u\} = [K]^{-1} \{F\}$$

known disp
 = unknown force
 known force
 = unknown disp





Direction cosines:

We don't actually need the angles, just the nodal coords

$$L = \sqrt{(x_2 - x_1)^2 + (y_2 - y_1)^2}$$

$$\hat{r} = \frac{1}{L} \begin{bmatrix} x_2 - x_1 \\ y_2 - y_1 \end{bmatrix}$$

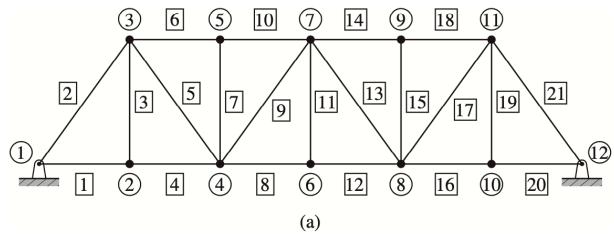
$$\hat{i} = \begin{bmatrix} 1 \\ 0 \end{bmatrix} \quad \hat{j} = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$$

$$\hat{r} \cdot \hat{i} = \cos \theta$$

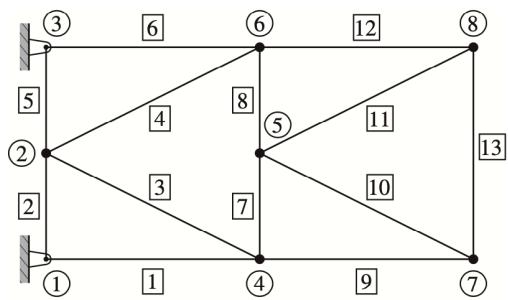
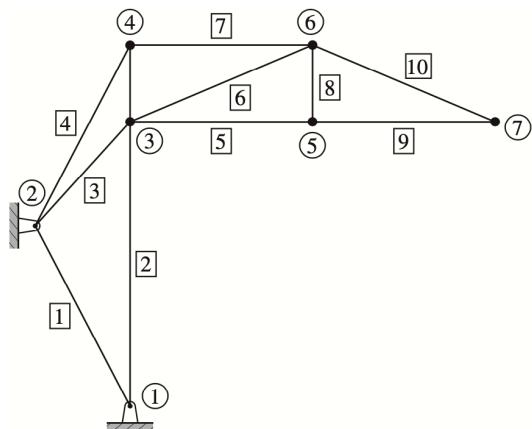
$$\hat{r} \cdot \hat{j} = \sin \theta$$

$$\cos \theta = \frac{x_2 - x_1}{L}, \quad \sin \theta = \frac{y_2 - y_1}{L}$$

These are the direction cosines



(a)



What about

