

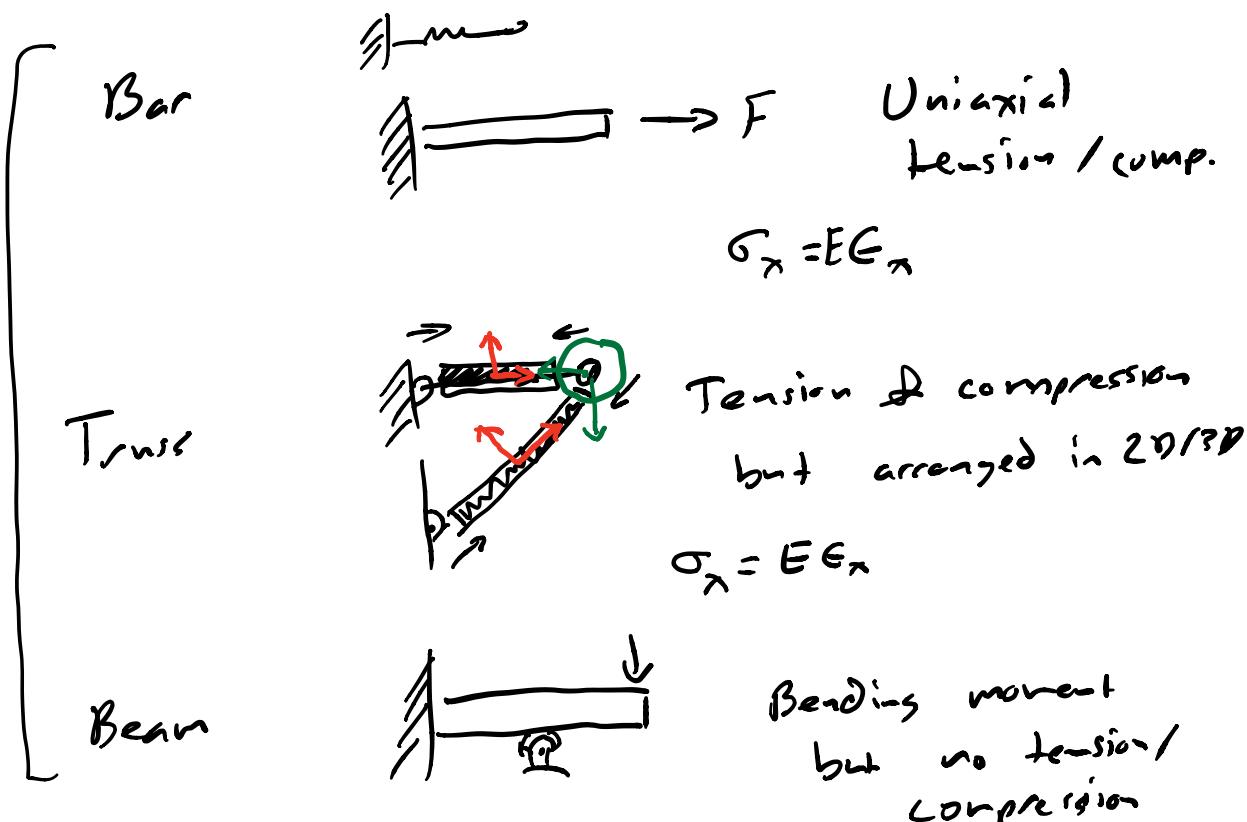
- 1) Introduce derivation of  $\{F\} = [K] \{D\}$   $F=Kx$
- 2) Study bar elements for 1-D FEA
- 3) Study truss elements for 2-D truss structures
- 4) (Thursday) Study beam elements for bending FEA

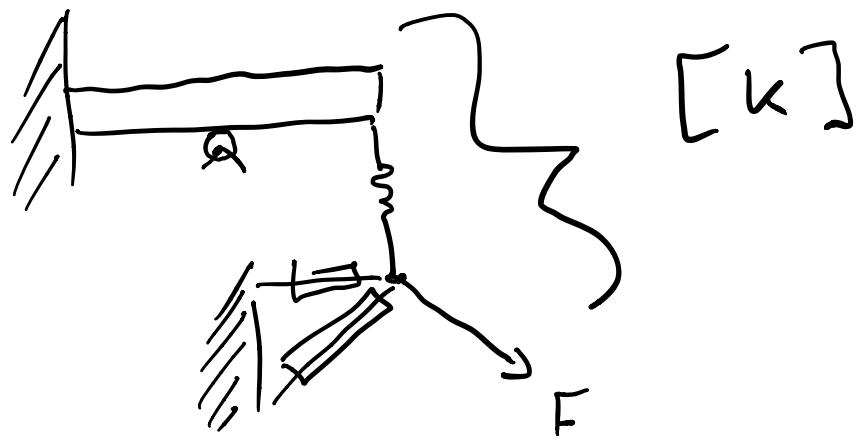
The fundamental FEA equation

$$\rightarrow \{F\} = [K] \{D\}$$

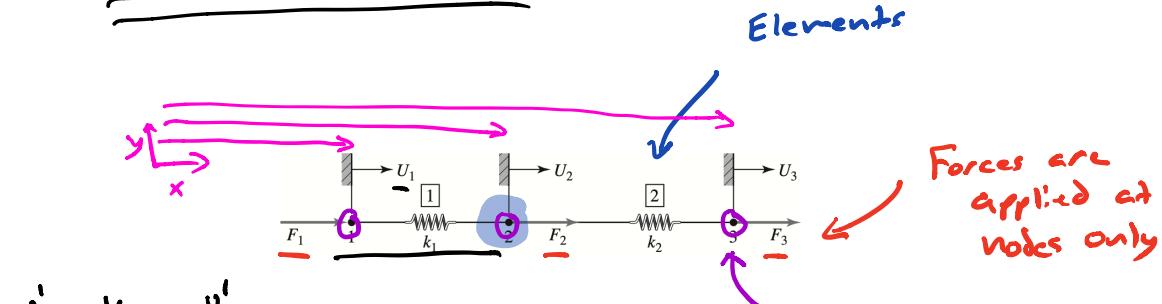
↑                      ↑                      ←  
Forces              Global Stiffness matrix      Displacements  
 $\{ \} \rightarrow$  column vector

We want Displacements of our system, given some boundary conditions and given applied forces





$$\{F\} = [K] \{U\}$$



$$u_1' \quad u_2'$$

$$f_1' \quad f_2'$$

$$f_1' = k u_1 - k u_2$$

$$f_2' = k u_2 - k u_1$$

$$[k' \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}] \begin{Bmatrix} u_1' \\ u_2' \end{Bmatrix} = \begin{Bmatrix} f_1' \\ f_2' \end{Bmatrix}$$

Element stiffness matrix

Element 1

$$\begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1 \end{bmatrix} \begin{Bmatrix} u_1^{(1)} \\ u_2^{(1)} \end{Bmatrix} = \begin{Bmatrix} f_1^{(1)} \\ f_2^{(1)} \end{Bmatrix}$$

Element 2

$$\begin{bmatrix} k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} u_1^{(2)} \\ u_2^{(2)} \end{Bmatrix} = \begin{Bmatrix} f_1^{(2)} \\ f_2^{(2)} \end{Bmatrix}$$

Relative  
elemental  
position

$$u_1^{(1)} = U_1$$

$$u_2^{(1)} = U_2$$

$$u_1^{(2)} = U_2$$

$$u_2^{(2)} = U_3$$

Absolute position

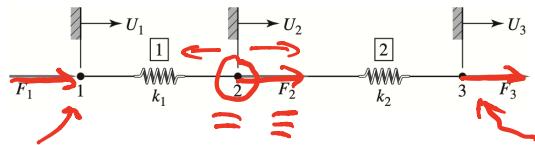
# 1

$$\begin{bmatrix} k_1 & -k_1 \\ -k_1 & k_1 \end{bmatrix} \begin{Bmatrix} U_1 \\ U_2 \end{Bmatrix} = \begin{Bmatrix} f_1^{(1)} \\ f_2^{(1)} \end{Bmatrix}$$

$$\{U\} = \begin{Bmatrix} U_1 \\ U_2 \\ U_3 \end{Bmatrix}$$

# 2

$$\begin{bmatrix} k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} U_2 \\ U_3 \end{Bmatrix} = \begin{Bmatrix} f_1^{(2)} \\ f_2^{(2)} \end{Bmatrix}$$



# 1

$$\begin{array}{c} \downarrow \\ \left[ \begin{array}{ccc} k_1 & -k_1 & 0 \\ -k_1 & k_1 & 0 \\ 0 & 0 & 0 \end{array} \right] \left\{ \begin{array}{c} U_1 \\ U_2 \\ 0 \end{array} \right\} = \left\{ \begin{array}{c} f_1^{(1)} \\ f_2^{(1)} \\ 0 \end{array} \right\} \end{array}$$

# 2

$$\rightarrow \left[ \begin{array}{ccc} 0 & 0 & 0 \\ 0 & k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{array} \right] \left\{ \begin{array}{c} 0 \\ U_2 \\ U_3 \end{array} \right\} = \left\{ \begin{array}{c} 0 \\ f_2^{(2)} \\ f_3^{(2)} \end{array} \right\}$$

$$\begin{array}{l} f_1^{(1)} = F_1 \\ f_2^{(1)} + f_2^{(2)} = F_2 \\ f_3^{(2)} = F_3 \end{array}$$

$$\left[ \begin{array}{ccc} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{array} \right] \left\{ \begin{array}{c} U_1 \\ U_2 \\ U_3 \end{array} \right\} = \left\{ \begin{array}{c} F_1 \\ F_2 \\ F_3 \end{array} \right\}$$

$$[K] \underbrace{\{U\}}_{=} = \{F\}$$

$$\underline{\underline{[K]\{U\} = \{F\}}}$$

$$[K] = \left[ \begin{array}{ccc} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 & -k_2 \\ 0 & -k_2 & k_2 \end{array} \right]$$

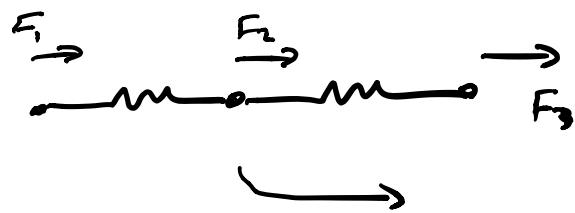
Our goal is to solve for displacements with applied forces  $\{F\}$  at equilibrium

$$\underline{\{U\}} = [K]^{-1} \underline{\{F\}} \quad ??$$



x known variables

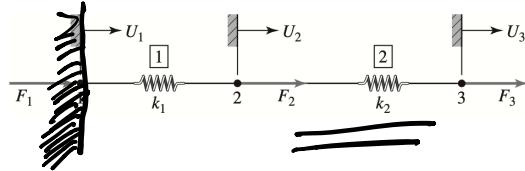
$[K]^{-1}$  as written above is singular, no inverse



Singular matrix  $\Rightarrow$  unsolvable,  
and in this case signifies the  
rigid body motion of the springs

To solve  $\{F\} = [k]\{U\}$   
we need boundary conditions

Node 1 is attached to a fixed support, yielding the displacement constraint  $U_1 = 0$ .  
 $k_1 = 50 \text{ lb./in.}$ ,  $k_2 = 75 \text{ lb./in.}$ ,  $F_2 = F_3 = 75 \text{ lb.}$



$\{U\}$

$$\begin{bmatrix} 50 & -50 & 0 \\ -50 & 125 & -75 \\ 0 & -75 & 75 \end{bmatrix} \begin{Bmatrix} U_1 \\ U_2 \\ U_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ 75 \\ 75 \end{Bmatrix}$$

$$[k] \{U\} = \{F\}$$

$U_2 = 3 \text{ in.}$  and  $U_3 = 4 \text{ in.}$

$$\begin{Bmatrix} U_2 \\ U_3 \end{Bmatrix} = [k]^{-1} \begin{Bmatrix} 75 \\ 75 \end{Bmatrix}$$

$$\begin{bmatrix} 50 & -50 \\ -50 & 50 \end{bmatrix} \begin{Bmatrix} 0 \\ 3 \end{Bmatrix} = \begin{Bmatrix} f_1^{(1)} \\ f_2^{(1)} \end{Bmatrix} = \begin{Bmatrix} -150 \\ 150 \end{Bmatrix} \text{ lb. for element 1}$$

$$\begin{bmatrix} 75 & -75 \\ -75 & 75 \end{bmatrix} \begin{Bmatrix} 3 \\ 4 \end{Bmatrix} = \begin{Bmatrix} f_1^{(2)} \\ f_2^{(2)} \end{Bmatrix} = \begin{Bmatrix} -75 \\ 75 \end{Bmatrix} \text{ lb. for element 2}$$

- 1) Construct element stiffness equations
- 2) Combined into a global stiffness matrix using compatibility conditions at adjacent nodes
- 3) Apply boundary conditions
- 4) Solve reduced system for  $U$
- 5) Solve for unknown forces

### 3 spring elements in series

spring:  $k \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$

#1

$$[k^{(1)}] = \begin{bmatrix} 3k & -3k \\ -3k & 3k \end{bmatrix}$$

$$\rightarrow u_1^{(1)} = U_1 \quad \underline{u_2^{(1)} = u_1^{(2)} = U_2}$$

#2

$$[k^{(2)}] = \begin{bmatrix} 2k & -2k \\ -2k & 2k \end{bmatrix}$$

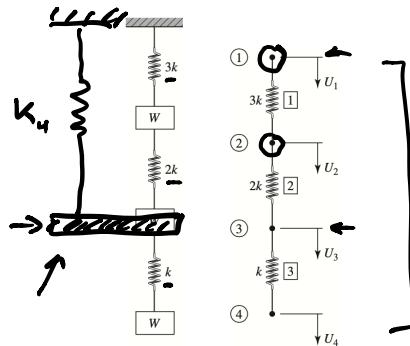
$$\underline{u_2^{(2)} = u_1^{(3)} = U_3} \quad \underline{u_3^{(2)} = U_4}$$

#3

$$[k^{(3)}] = \begin{bmatrix} k & -k \\ -k & k \end{bmatrix}$$

$$\#1 \quad 3k \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \{ \underline{U_1} \} = \{ F_1 \}$$

$$\#2 \quad 2k \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \{ \underline{U_3} \} = \{ F_2 \}$$



$$\rightarrow k_4 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \{ \underline{U_3} \} = \{ F_3 \}$$

$$\#4 \quad \boxed{\begin{bmatrix} k_4 & 0 & -k_4 & 0 \\ 0 & 0 & 0 & 0 \\ -k_4 & 0 & k_4 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}} \{ \underline{U_1} \}$$

$$\left\{ \begin{array}{l} \boxed{\begin{bmatrix} 3k & -3k & 0 & 0 \\ -3k & 3k & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}} \{ \underline{U_1} \} = \{ f_1^{(1)} \} \\ \boxed{\begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 2k & -2k & 0 \\ 0 & -2k & 2k & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}} \{ \underline{U_2} \} = \{ f_1^{(2)} \} \\ \boxed{\begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & k & -k \\ 0 & 0 & -k & k \end{bmatrix}} \{ \underline{U_3} \} = \{ f_1^{(3)} \} \end{array} \right. \quad \#1$$

#2

#3

$\downarrow$   $U_1$

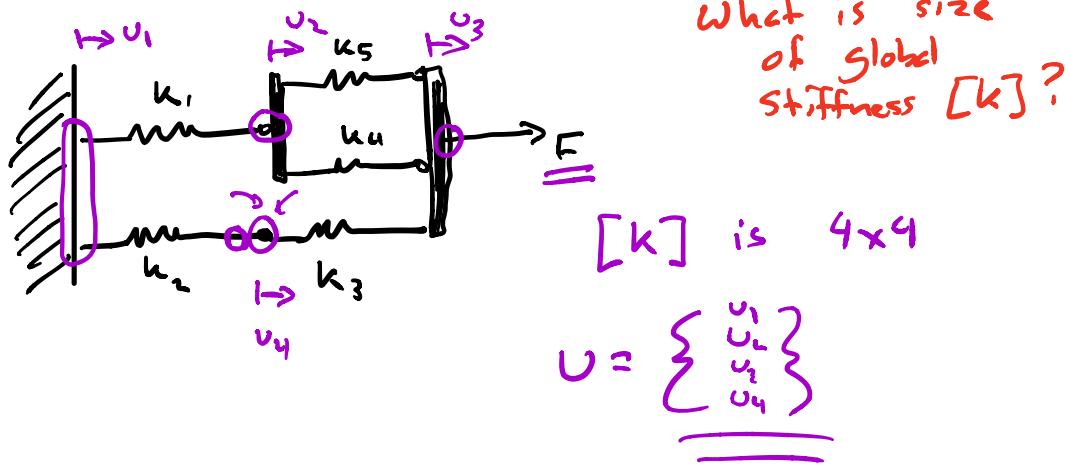
$$U_1 = 0 \quad k \begin{bmatrix} 2 & -3 & 0 & 0 \\ -3 & 5 & -2 & 0 \\ 0 & -2 & 3 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix} \{ \underline{U_2} \} = \{ F_1 \}$$

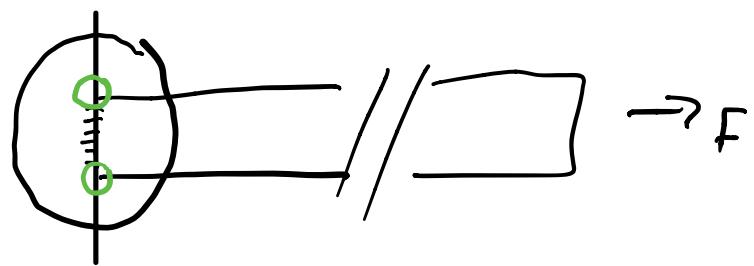
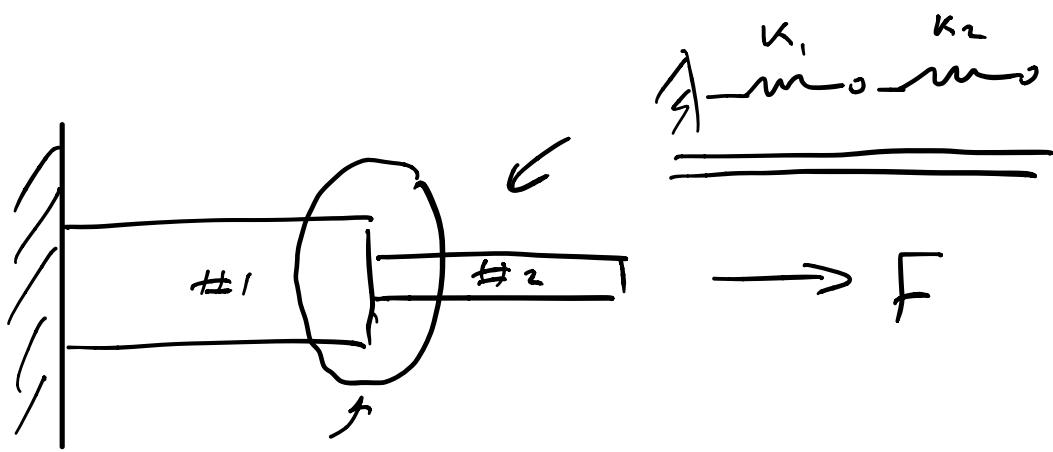
$$k \begin{bmatrix} 5 & -2 & 0 \\ -2 & 3 & -1 \\ 0 & -1 & 1 \end{bmatrix} \{ \underline{U_3} \} = \{ W \}$$

imposed forces  
 $\{ U_3 \} = [k]^{-1} \{ F \}$

$$U_2 = \frac{W}{k} \quad U_3 = \frac{2W}{k} \quad U_4 = \frac{3W}{k}$$

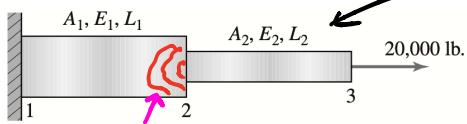
Spring elements have two nodes,  
each described by 1-variable  
of displacement





# Bar element

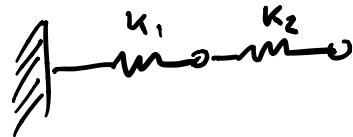
Area, Young's modulus, Length



$$A_1 = 4 \text{ in.}^2 \quad A_2 = 2.25 \text{ in.}^2$$

$$E_1 = 15 \times 10^6 \text{ lb./in.}^2 \quad E_2 = 10 \times 10^6 \text{ lb./in.}^2$$

$$L_1 = 20 \text{ in.} \quad L_2 = 20 \text{ in.}$$



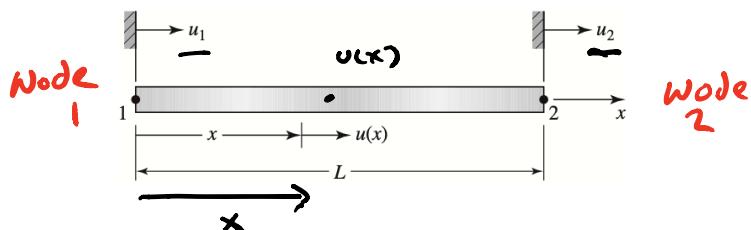
$$k_1 = \frac{A_1 E_1}{L_1} \quad k_2 = \frac{A_2 E_2}{L_2}$$

#1  $k_1 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$

#2  $k_2 \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$

Goal in FEA is to have a continuous representation of stress and strain.

$$u(x) = \underline{N_1(x)u_1 + N_2(x)u_2}$$



B.C.

$$\underline{u(x=0) = u_1} \quad \underline{u(x=L) = u_2}$$

$$\underline{N_1(0) = 1} \quad \underline{N_2(0) = 0}$$

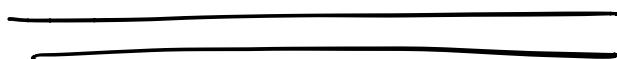
$$\underline{N_1(L) = 0} \quad \underline{N_2(L) = 1}$$

$$N_1(x) = a_0 + a_1x$$

$$N_2(x) = b_0 + b_1x$$

$$N_1(x) = 1 - x/L$$

$$N_2(x) = x/L$$



$$u(x) = \underline{\underline{(1 - x/L)u_1 + (x/L)u_2}}$$

Continuous  
displacement  
field of  
1-D Bar

$$\underline{\underline{u(x)}} = \underline{\underline{[N_1(x) \quad N_2(x)]}} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \underline{\underline{[N]}} \{u\}$$

$$\{u\} = \{u_1 \quad u_2\}$$

$$u(x) = [N] \{u\}$$



shape functions

Interpolation functions

Typically, polynomials

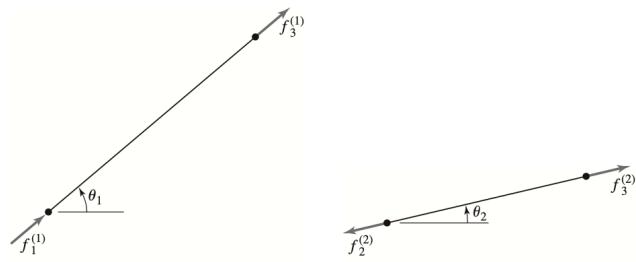
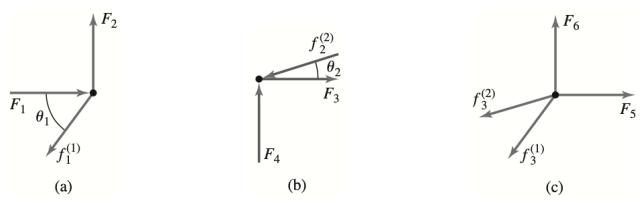
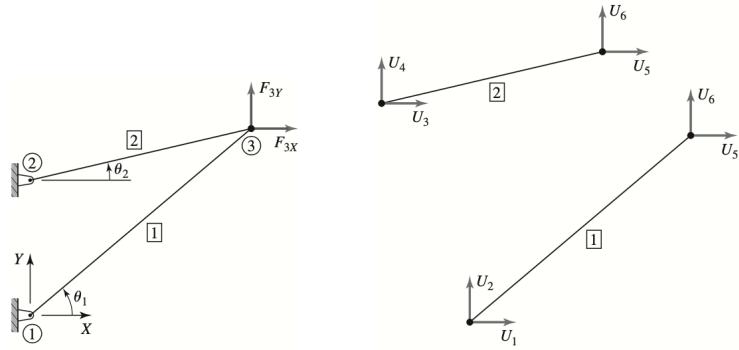








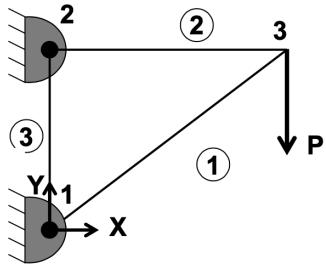




$$\begin{array}{lll}
 F_1 - f_1^{(1)} \cos \theta_1 = 0 & F_3 - f_2^{(2)} \cos \theta_2 = 0 & F_5 - f_3^{(1)} \cos \theta_1 - f_3^{(2)} \cos \theta_2 = 0 \\
 F_2 - f_1^{(1)} \sin \theta_1 = 0 & F_4 - f_2^{(2)} \sin \theta_2 = 0 & F_6 - f_3^{(1)} \sin \theta_1 - f_3^{(2)} \sin \theta_2 = 0
 \end{array}$$

$$\begin{bmatrix} k^{(1)}c^2\theta_1 & k^{(1)}s\theta_1c\theta_1 & 0 & 0 & -k^{(1)}c^2\theta_1 & -k^{(1)}s\theta_1c\theta_1 \\ k^{(1)}s\theta_1c\theta_1 & k^{(1)}s^2\theta_1 & 0 & 0 & -k^{(1)}s\theta_1c\theta_1 & -k^{(1)}s^2\theta_1 \\ 0 & 0 & k^{(2)}c^2\theta_2 & k^{(2)}s\theta_2c\theta_2 & -k^{(2)}c^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 \\ 0 & 0 & k^{(2)}s\theta_2c\theta_2 & k^{(2)}s^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 & -k^{(2)}s^2\theta_2 \\ -k^{(1)}c^2\theta_{12} & -k_1s\theta_1c\theta_1 & -k^{(2)}c^2\theta_2 & -k^{(2)}s\theta_2c\theta_2 & k^{(1)}c^2\theta_1 + \\ & & & & k^{(2)}c^2\theta_2 & k^{(1)}s\theta_1c\theta_1 + \\ -k_1s\theta_1c\theta_1 & -k^{(1)}s^2\theta_1 & -k^{(2)}s\theta_2c\theta_2 & -k^{(2)}s^2\theta_2 & k^{(1)}s\theta_1c\theta_1 + & k^{(1)}s^2\theta_1 + \\ & & & & k^{(2)}s\theta_2c\theta_2 & k^{(2)}s^2\theta_2 \end{bmatrix} = \begin{Bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \\ U_5 \\ U_6 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \\ F_5 \\ F_6 \end{Bmatrix}$$

$$[K^e] = \frac{AE}{L} \begin{bmatrix} c^2 & cs & -c^2 & -cs \\ cs & s^2 & -cs & -s^2 \\ -c^2 & -cs & c^2 & cs \\ -cs & -s^2 & cs & s^2 \end{bmatrix}$$



$$[K^e] = \frac{AE}{L} \begin{bmatrix} c^2 & cs & -c^2 & -cs \\ cs & s^2 & -cs & -s^2 \\ -c^2 & -cs & c^2 & cs \\ -cs & -s^2 & cs & s^2 \end{bmatrix}$$

$$K_1 = \begin{bmatrix} \frac{k_1}{2} & \frac{k_1}{2} & -\frac{k_1}{2} & -\frac{k_1}{2} \\ \frac{k_1}{2} & \frac{k_1}{2} & -\frac{k_1}{2} & -\frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & \frac{k_1}{2} & \frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & \frac{k_1}{2} & \frac{k_1}{2} \end{bmatrix},$$

$$K_2 = \begin{bmatrix} k_2 & 0 & -k_2 & 0 \\ 0 & 0 & 0 & 0 \\ -k_2 & 0 & k_2 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix},$$

$$K_3 = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & k_3 & 0 & -k_3 \\ 0 & 0 & 0 & 0 \\ 0 & -k_3 & 0 & k_3 \end{bmatrix},$$

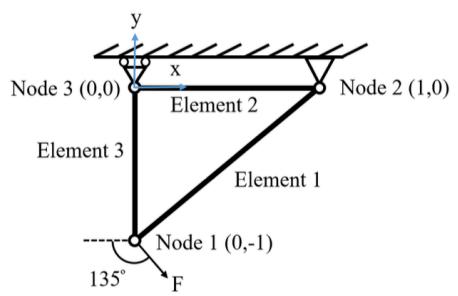
$$\begin{bmatrix} \frac{k_1}{2} & \frac{k_1}{2} & 0 & 0 & -\frac{k_1}{2} & -\frac{k_1}{2} \\ \frac{k_1}{2} & \frac{k_1}{2} & 0 & 0 & -\frac{k_1}{2} & -\frac{k_1}{2} \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ -\frac{k_1}{2} & -\frac{k_1}{2} & 0 & 0 & \frac{k_1}{2} & \frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & 0 & 0 & \frac{k_1}{2} & \frac{k_1}{2} \end{bmatrix} \begin{Bmatrix} u_1 \\ v_1 \\ 0 \\ 0 \\ 0 \\ u_3 \\ v_3 \end{Bmatrix} = \begin{Bmatrix} F_{x1} \\ F_{y1} \\ 0 \\ 0 \\ 0 \\ 0 \\ -P \end{Bmatrix},$$

$$\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & k_2 & 0 & -k_2 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & -k_2 & 0 & k_2 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} 0 \\ 0 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ F_{x2} \\ F_{y2} \\ 0 \\ 0 \end{Bmatrix},$$

$$\begin{bmatrix} \frac{k_1}{2} & \frac{k_1}{2} & 0 & 0 & -\frac{k_1}{2} & -\frac{k_1}{2} \\ \frac{k_1}{2} & \frac{k_1}{2} + k_3 & 0 & -k_3 & -\frac{k_1}{2} & -\frac{k_1}{2} \\ 0 & 0 & k_2 & 0 & -k_2 & 0 \\ 0 & -k_3 & 0 & k_3 & 0 & 0 \\ -\frac{k_1}{2} & -\frac{k_1}{2} & -k_2 & 0 & k_2 + \frac{k_1}{2} & \frac{k_1}{2} \\ -\frac{k_1}{2} & -\frac{k_1}{2} & 0 & 0 & \frac{k_1}{2} & \frac{k_1}{2} \end{bmatrix} \begin{Bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ u_3 \\ v_3 \end{Bmatrix} = \begin{Bmatrix} F_{x1} \\ F_{y1} \\ F_{x2} \\ F_{y2} \\ 0 \\ -P \end{Bmatrix}.$$

$$\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & k_3 & 0 & -k_3 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & -k_3 & 0 & k_3 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix} \begin{Bmatrix} u_1 \\ v_1 \\ u_2 \\ v_2 \\ 0 \\ 0 \end{Bmatrix} = \begin{Bmatrix} F_{x1} \\ F_{y1} \\ F_{x2} \\ F_{y2} \\ 0 \\ 0 \end{Bmatrix},$$

$$\begin{bmatrix} k_2 + \frac{k_1}{2} & \frac{k_1}{2} \\ \frac{k_1}{2} & \frac{k_1}{2} \end{bmatrix} \begin{Bmatrix} u_3 \\ v_3 \end{Bmatrix} = \begin{Bmatrix} 0 \\ -P \end{Bmatrix}$$



## 10.15 | Two dimensional FEM analysis

## 10.15.1 | Triangular elements

Figure 10.12 shows a plane constant-strain triangle. The triangle has six degrees of freedom. The shape functions (interpolation functions)

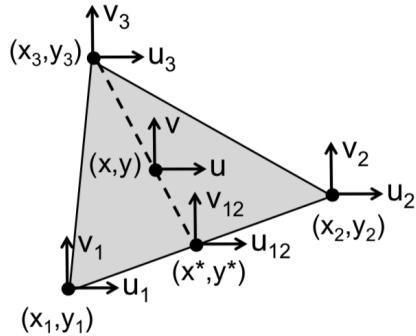


Figure 10.12: Plane, constant strain triangle.

defined at the nodes allows calculating the displacement at each point within the element:

$$u_{12} = \frac{1}{2} \left( \frac{x^* - x_2}{x_1 - x_2} + \frac{y^* - y_2}{y_1 - y_2} \right) u_1 + \frac{1}{2} \left( \frac{x^* - x_1}{x_2 - x_1} + \frac{y^* - y_1}{y_2 - y_1} \right) u_2. \quad (10.164)$$

Along the line  $(x^*, y^*)$  to  $(x_3, y_3)$  one can define

$$u = \frac{1}{2} \left( \frac{x - x_3}{x^* - x_3} + \frac{y - y_3}{y^* - y_3} \right) u_{12} + \frac{1}{2} \left( \frac{x - x^*}{x_3 - x^*} + \frac{y - y^*}{y_3 - y^*} \right) u_3. \quad (10.165)$$

Slope condition:

$$\frac{y^* - y_1}{x^* - x_1} = \frac{y_2 - y_1}{x_2 - x_1} \quad (10.166)$$

and

$$\frac{y - y_3}{x - x_3} = \frac{y^* - y_3}{x^* - x_3}. \quad (10.167)$$

Solving eqns. (10.165), (10.166), and (10.167), one obtains

$$u = \frac{p_1(x, y) u_1 + p_2(x, y) u_2 + p_3(x, y) u_3}{\Delta}, \quad (10.168)$$

where

$$p_1(x, y) = x_2 y_3 - x_3 y_2 + x(y_2 - y_3) + y(x_3 - x_2), \quad (10.169)$$

$$p_2(x, y) = x_3 y_1 - x_1 y_3 + x(y_3 - y_1) + y(x_1 - x_3), \quad (10.170)$$

$$p_3(x, y) = x_1y_2 - x_2y_1 + x(y_1 - y_2) + y(x_2 - x_1), \quad (10.171)$$

and

$$\Delta = x_2y_3 - x_3y_2 + x_1y_2 + x_3y_1 - x_2y_1. \quad (10.172)$$

A similar expression can be derived for  $v$  in terms of  $v_1, v_2$  and  $v_3$ . Thus the displacement of an arbitrary point within an element can be written in terms of the displacement at the nodes as

$$\mathbf{d} = \begin{Bmatrix} u \\ v \end{Bmatrix} = \frac{1}{\Delta} \begin{bmatrix} p_1(x, y) & p_2(x, y) & p_3(x, y) & 0 & 0 & 0 \\ 0 & 0 & 0 & p_1(x, y) & p_2(x, y) & p_3(x, y) \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ v_1 \\ v_2 \\ v_3 \end{Bmatrix}, \quad (10.173)$$

or

$$\mathbf{d} = [N] \mathbf{d}^e. \quad (10.174)$$

Expressing the strains  $\varepsilon$  as a function of the nodal displacements  $\mathbf{d}^e$ , one finds

$$\varepsilon = \begin{Bmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_{xy} \end{Bmatrix} = [\mathbf{B}] \mathbf{d}^e. \quad (10.175)$$

Since  $\varepsilon_x = \frac{\partial u}{\partial x}$ ,  $\varepsilon_y = \frac{\partial v}{\partial y}$  and  $\varepsilon_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$ ,

$$\varepsilon = \begin{bmatrix} \frac{\partial}{\partial x} & 0 \\ 0 & \frac{\partial}{\partial y} \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} \end{bmatrix} \begin{Bmatrix} u \\ v \end{Bmatrix} = \begin{bmatrix} \frac{\partial}{\partial x} & 0 \\ 0 & \frac{\partial}{\partial y} \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} \end{bmatrix} [N] \mathbf{d}^e. \quad (10.176)$$

Thus,

$$[\mathbf{B}] = \begin{bmatrix} \frac{\partial}{\partial x} & 0 \\ 0 & \frac{\partial}{\partial y} \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} \end{bmatrix} [N] \quad (10.177)$$

or

$$[\mathbf{B}] = \frac{1}{\Delta} \begin{bmatrix} y_2 - y_3 & y_3 - y_1 & y_1 - y_2 & 0 & 0 & 0 \\ 0 & 0 & 0 & x_3 - x_2 & x_1 - x_3 & x_2 - x_1 \\ x_3 - x_2 & x_1 - x_3 & x_2 - x_1 & y_2 - y_3 & y_3 - y_1 & y_1 - y_2 \end{bmatrix}. \quad (10.178)$$

Note that  $[\mathbf{B}]$  is not a function of  $x$  or  $y$ . The stiffness matrix

$$[\mathbf{K}] = \int_V [\mathbf{B}]^t [D] [\mathbf{B}] dV \quad (10.179)$$

