

# AE333

## Mechanics of Materials

Lecture 21 - Combined loading

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## schedule

- 25 Mar - Combined Loading, HW6  
Due
- 27 Mar - Stress Transformation
- 29 Mar - Stress Transformation
- 1 Apr - Stress Transformation, HW7  
Due

## outline

- combined loading
- group problems
- plane stress  
transformation

# combined loading

## combined loading

- We can use the principle of superposition to treat various loading conditions separately and then add them together to find the total stress

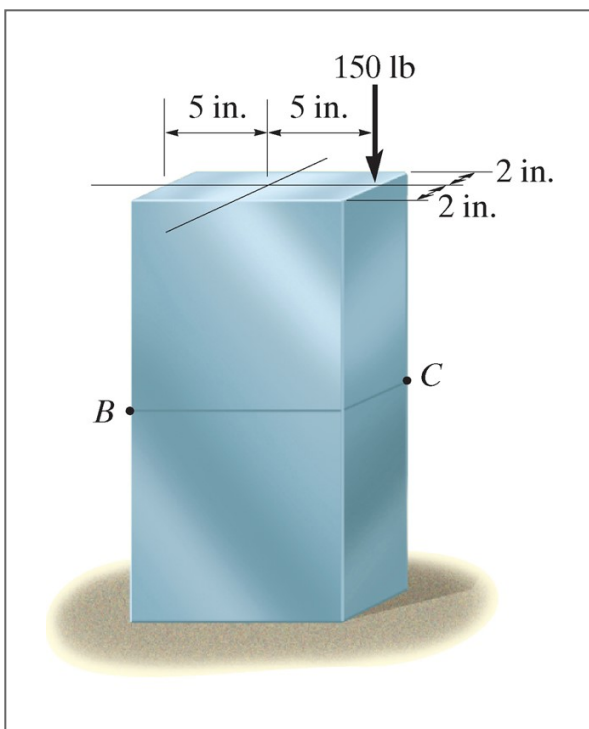
## procedure

- Section the member at the point of interest, internal force components should be drawn acting through the centroid of the section
- Moment components should be calculated about the centroidal axis

## stress components

- Normal stress:  $\sigma = N/A$
- Shear:  $\tau = Q/It$
- Bending:  $\sigma = y/I$
- Torsion:  $\tau = \rho/J$
- Pressure Vessels:  $\sigma_1 = r/t$ ,  $\sigma_2 = r/2t$

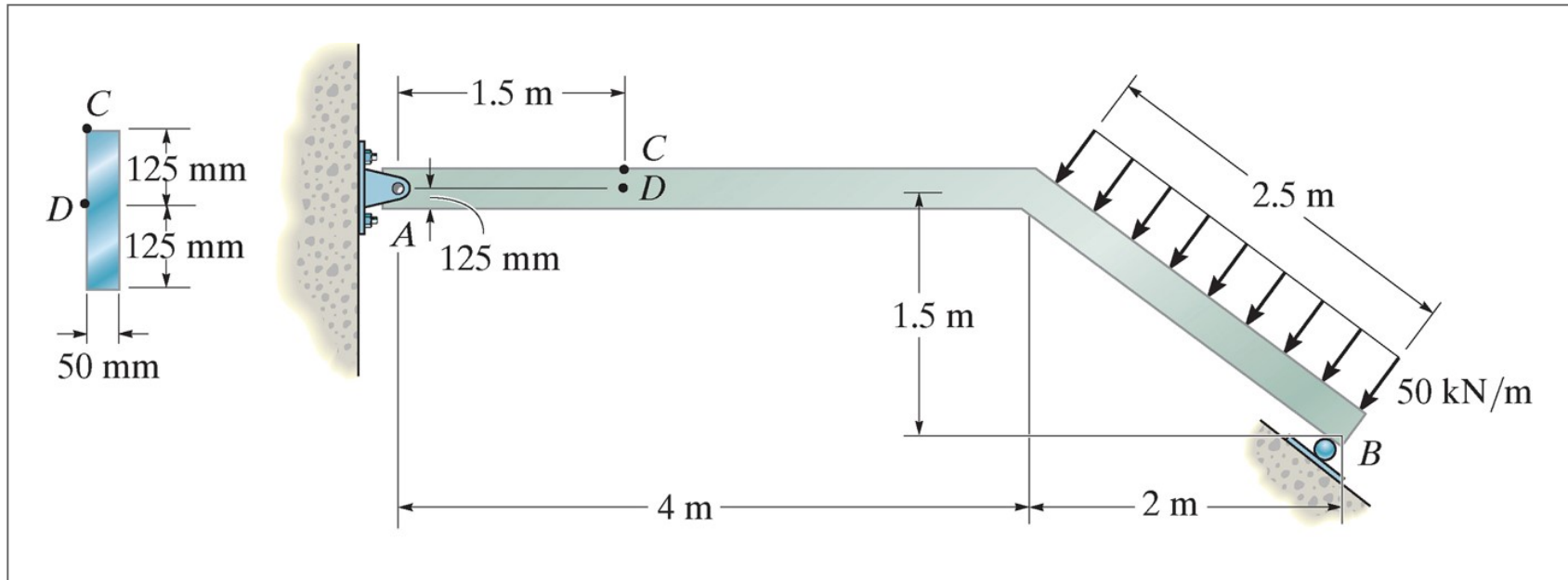
## example 8.2



Neglect the weight of the member and find the stress at B and C.

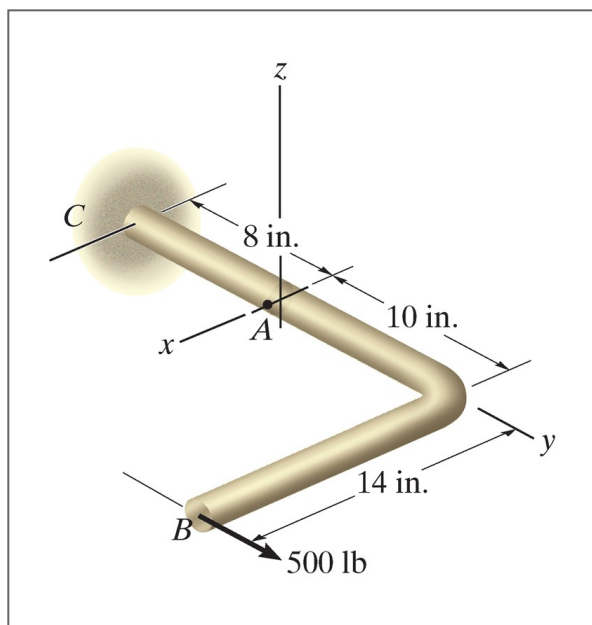


## example 8.4



Determine the stress at C and D.

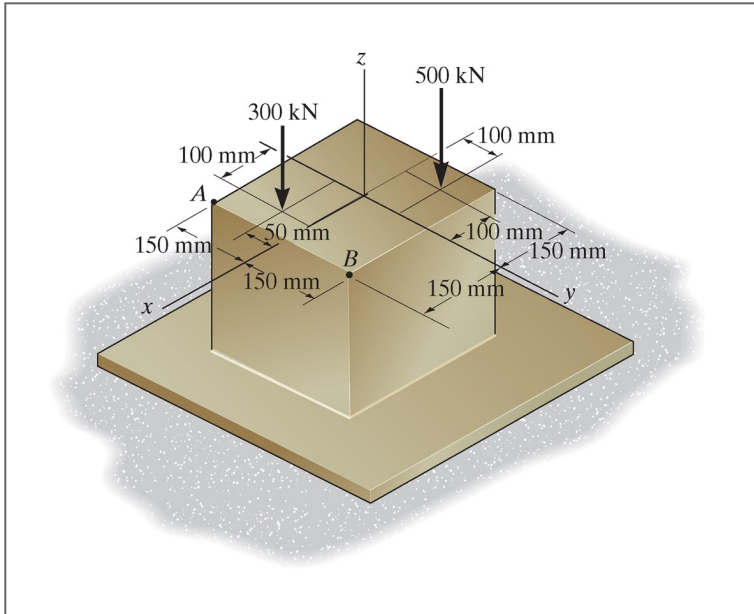
## example 8.5



The rod shown has a radius of 0.75 in. Find the stress at A.

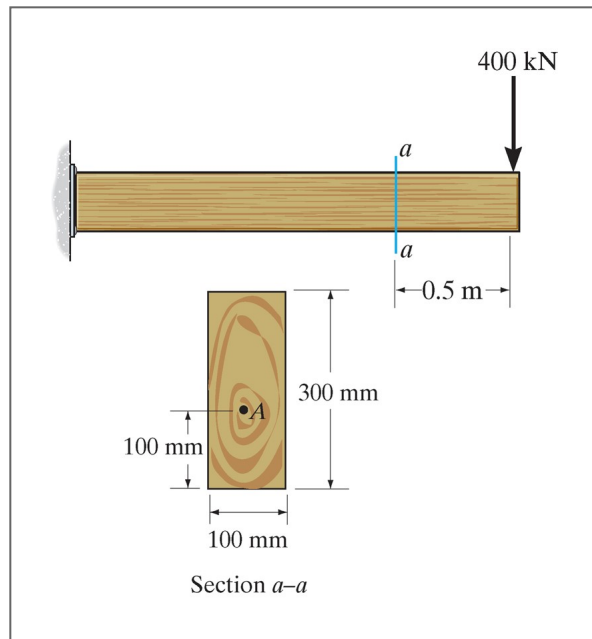
# group problems

## group one



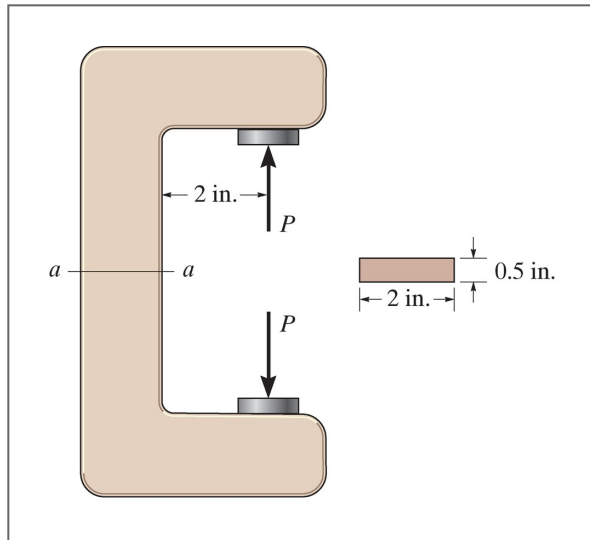
Find the stress at the corners A and B for the column shown.

## group two



Find the stress at point A for the cantilever beam shown.

## group three



Find the load  $P$  that will cause a maximum normal stress of  $\sigma = 30$  ksi along the section  $a-a$ .

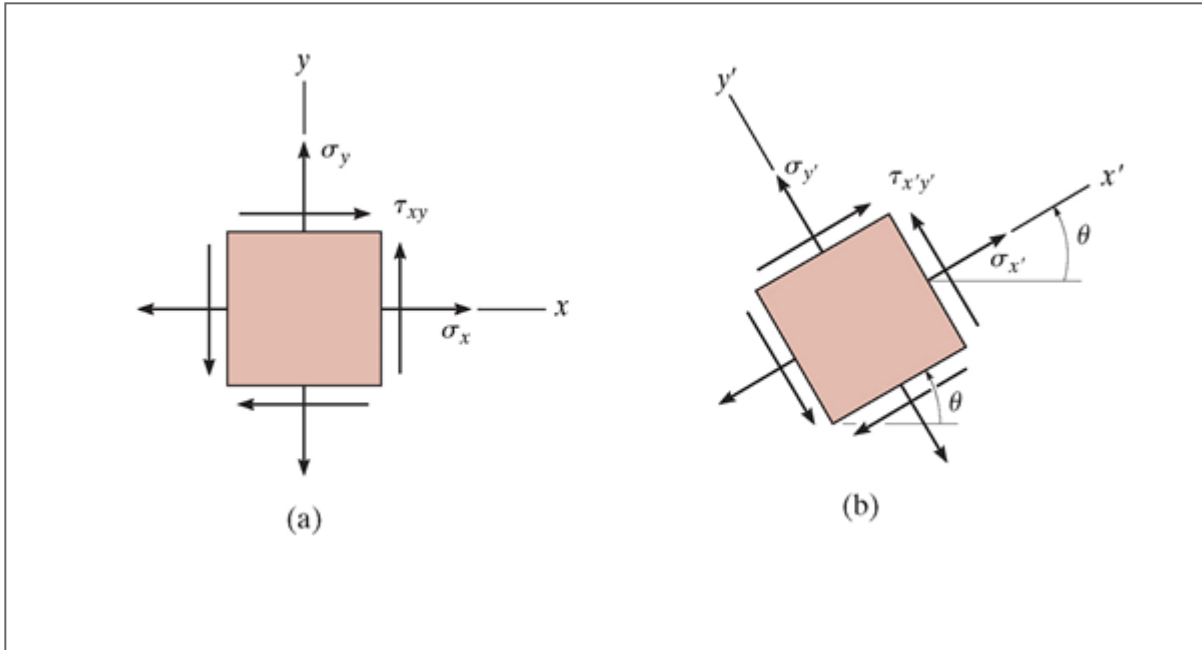
# plane stress transformation

## plane stress

- In general, the state of stress at a point is characterized by six stress components
- In practice, this is rare, as most stresses and forces act in the same plane
- This case is referred to as plane stress



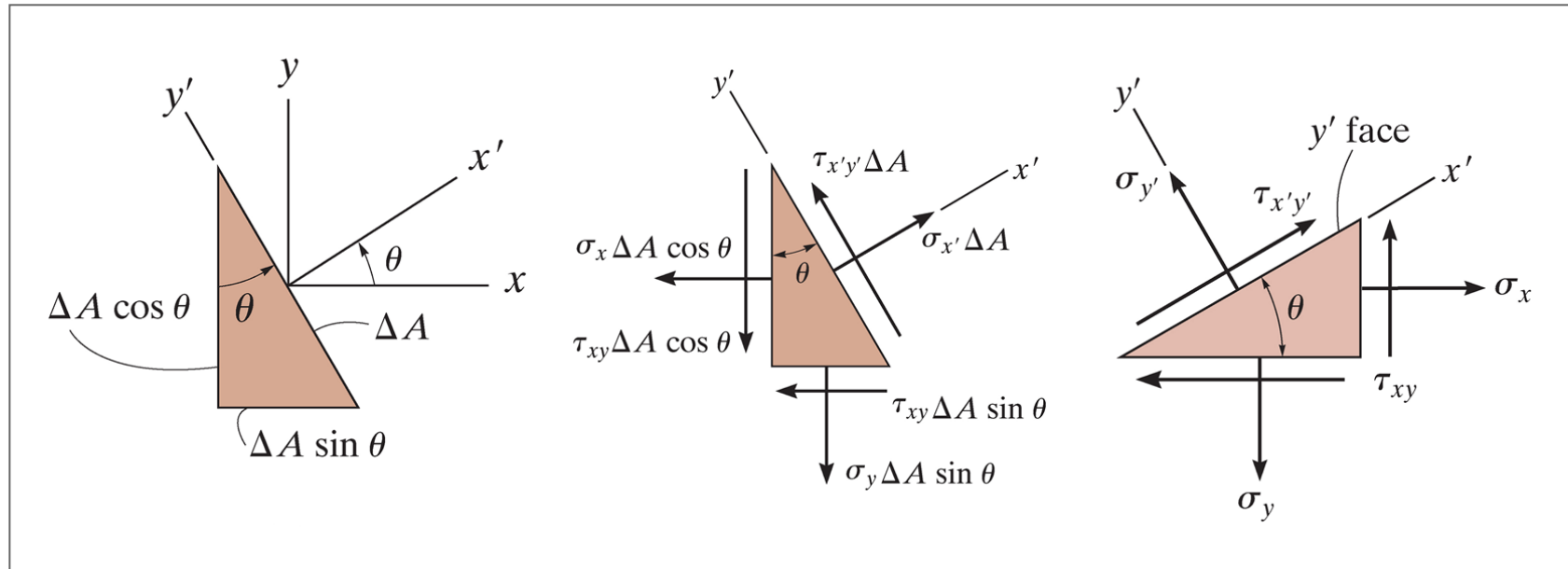
# transformation



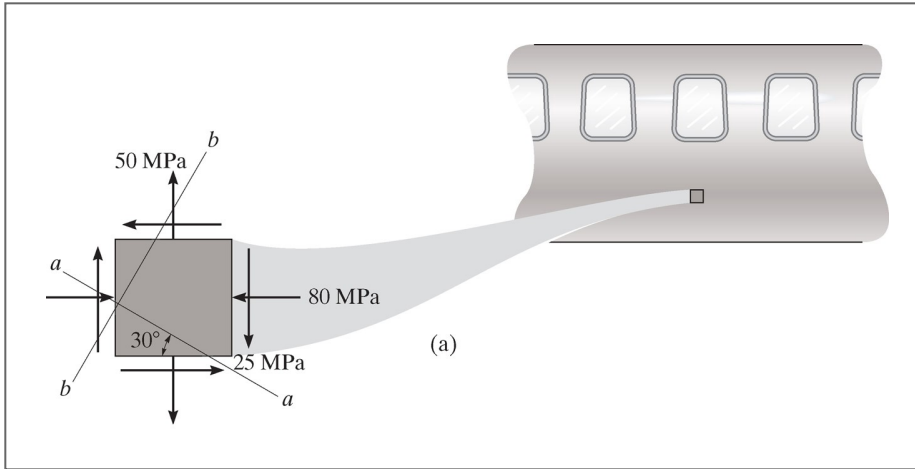
## procedure

- If the state of stress ( $\sigma_x, \sigma_y, \tau_{xy}$ ) is known for a known axis system  $x$  and  $y$ , we can find the stress relative to some rotated coordinate system
- We do this by considering a section of the element perpendicular to the  $x'$  axis
- Sum of forces in  $x$  and  $y$  will give  $\sigma_{x'}$  and  $\tau_{x'y'}$
- A second section is needed to find  $\sigma_{y'}$ , perpendicular to the  $y'$  axis

# procedure



## example 9.1



Represent the state of stress shown on the fuselage section on an element rotated  $30^\circ$  clockwise from the position shown.