## **AE333**

3 Apr, 2019

### **Mechanics of Materials**

Lecture 24 - Strain Transformation
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### schedule

- 3 Apr Strain Transformation, HW7 Due
- 5 Apr Deflection of Beams
- 8 Apr Deflection of Beams, HW8 Due
- 10 Apr Deflection of Beams

#### outline

- mohr's circle
- absolute maximum shear
- plane strain
- principal strains and mohr's circle
- strain rosettes
- material property relationships

- We can visualize plane stress transformation using a technique known as Mohr's circle
- If we re-write the stress transformation equations we find

$$egin{aligned} \sigma_{x'} - \left(rac{\sigma_x + \sigma_y}{2}
ight) &= \left(rac{\sigma_x - \sigma_y}{2}
ight)\cos 2 heta + au_{xy}\sin 2 heta \ au_{x'y'} &= -\left(rac{\sigma_x - \sigma_y}{2}
ight)\sin 2 heta + au_{xy}\cos 2 heta \end{aligned}$$

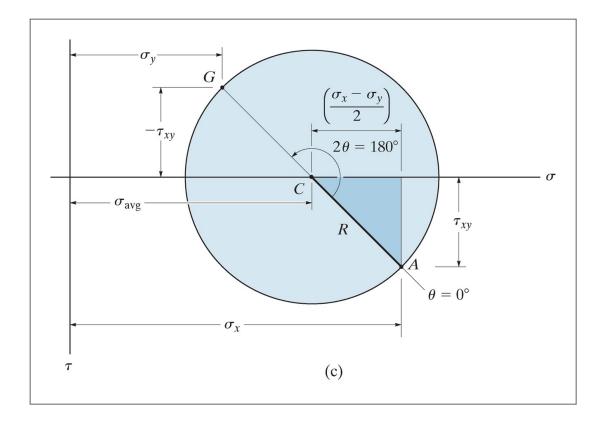
• If we square each equation and add them together, we find

$$\left[\sigma_{x'}-\left(rac{\sigma_x+\sigma_y}{2}
ight)
ight]^2+ au_{x'y'}^2=\left(rac{\sigma_x-\sigma_y}{2}
ight)^2+ au_{xy}^2.$$

• Since  $\sigma_x$ ,  $\sigma_y$  and  $\tau_{xy}$  are known constants, we can write a more compact form by letting

$$(\sigma_{x'}-\sigma_{avg})^2+ au_{x'y'}^2=R^2 \ \sigma_{avg}=rac{\sigma_x+\sigma_y}{2} \ R=\sqrt{\left(rac{\sigma_x-\sigma_y}{2}
ight)^2+ au_{xy}^2}$$

- Re-written in this way, we can see that the previous equation is the equation of a circle on the  $\sigma$ ,  $\tau$  axis
- The center of the circle is at  $\tau = 0$  and  $\sigma = \sigma_{avq}$
- The radius of the circle is  $\sqrt{\left(\frac{\sigma_x-\sigma_y}{2}\right)^2+ au_{xy}^2}$
- Each point on the circle represents  $\sigma_{\chi'}$ ,  $\tau_{\chi' y'}$



#### visual construction of Mohr's circle

- By convention, positive  $\tau$  points down, use this convention to plot the center of the circle and a reference point at  $(\sigma_x, \tau_{xy})$  where the x' axis is coincident with the x axis
- Use these two points to sketch the circle

## principal stress

- The principal stresses,  $\sigma_1$  and  $\sigma_2$  are the coordinates where Mohr's circle intersects the  $\sigma$  axis
- The angles  $\theta_{p1}$  and  $\theta_{p2}$  can be found by calculating the angle between the reference line and the  $\sigma$  axis (note that this angle is equal to  $2\theta_p$ )
- Note that the direction from the the reference point to the  $\sigma$  axis will be the same as the direction from the x axis to the principal axis

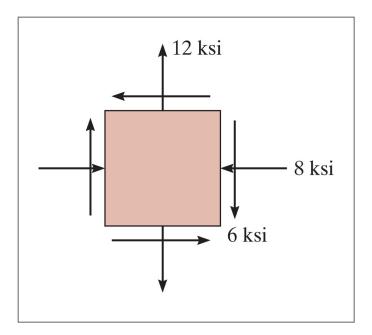
#### maximum shear stress

- The top and bottom of the circle represent the maximum shear stress
- The angles  $\theta_{s1}$  and  $\theta_{s2}$  can be found in a similar method to that described for the principal stress

## stress on arbitrary plane

- To find the stress at some arbitrary plane some known angle  $\theta$  away from our reference plane, we find the angle  $2\theta$  away from the reference line on Mohr's circle
- We can use trigonometry to find the value of the coordinates at that point
- We must draw our angle in the same direction as the desired rotation

# example 9.9

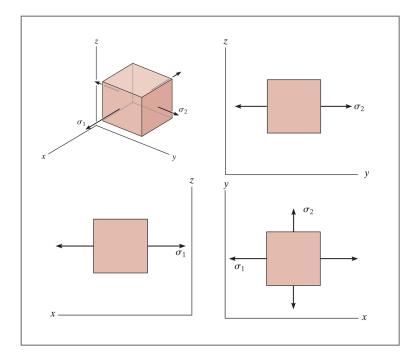


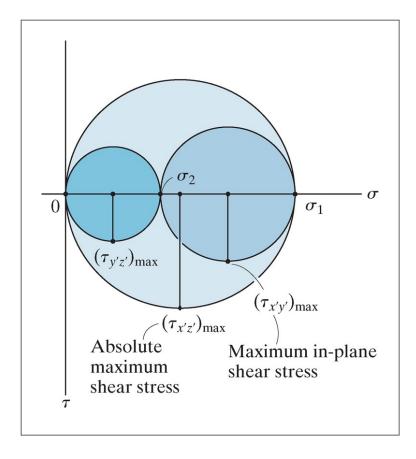
Represent the state of stress shown on an element rotated  $30^{\circ}$  counterclockwise from the position shown.

# absolute maximum shear

#### absolute maximum shear

- We already know how to find the maximum in-plane shear, but sometimes the maximum shear stress can occur in another plane
- We can do this (without treating it as a fully 3D problem) by treating each plane as its own plane stress problem





#### absolute max shear

• The maximum absolute shear will depend on whether  $\sigma_1$  and  $\sigma_2$  are in the same or opposite directions

$$au_{abs,max} = rac{\sigma_1}{2} \hspace{1cm} ext{same direction} \ au_{abs,max} = rac{\sigma_1 - \sigma_2}{2} \hspace{1cm} ext{opposite directions}$$

• Which of the three mohr's circles the maximum occurs in will determine which plane the shear acts in

# plane strain

## plane strain

- Under plane stress we assume no out-of-plane stresses are present
- This is typically a good assumption for very thin materials
- Under plane strain we assume no out-of-plane strains are present
- Typically a good assumption for very thick materials

## sign convention

- Normal strains,  $\epsilon_x$  and  $\epsilon_y$ , are considered positive when they cause elongation, and negative when they cause contraction
- Shear strains,  $\gamma_{xy}$  are positive if the interior angle becomes smaller than 90°, and negative if the angles becomes larger than 90°

## general equations

- We derive the general strain transformation equations by comparing infinitesimal elements before and after deformation
- To find  $\gamma_{x'y'}$  we compare the angle between dx and dy before and after deformation

## general equations

- The equations are nearly exactly the same as the stress transformation equations
- Pay attention to the difference, strain transformation equations are NOT on the equation sheet

$$\epsilon_{x'} = rac{\epsilon_x + \overline{\epsilon_y}}{2} + rac{\epsilon_x - \epsilon_y}{2} \cos 2 heta + rac{\gamma_{xy}}{2} \sin 2 heta \ rac{\gamma_{x'y'}}{2} = -\left(rac{\epsilon_x - \epsilon_y}{2}
ight) \sin 2 heta + rac{\gamma_{xy}}{2} \cos 2 heta$$

• As with  $\sigma_y$ , we find  $\epsilon_y$  by letting  $\theta_y = \theta_x + 90^\circ$ 

## engineering strain

- Side note: there is another definition of shear strain known as tensorial shear strain, where  $\gamma_{xy} = 2\epsilon_{xy}$
- Under tensorial strain, the transformation equations are exactly the same (as in this case both stress and strain are tensors)
- $\gamma_{xy}$  is known as engineering strain, you will need to pay attention to which strain convention is used when extracting data from finite elements or other sources

# principal strains and mohr's circle

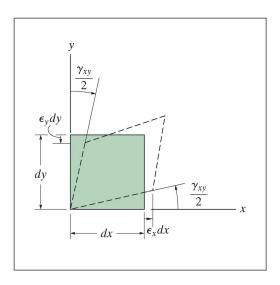
## principal strains

• As you might imagine, since the transformation equations are nearly identical so are the principal strain equations

$$an 2 heta_p = rac{\gamma_{xy}}{\epsilon_x - \epsilon_y} \ \epsilon_{1,2} = rac{\epsilon_x + \epsilon_y}{2} \pm \sqrt{\left(rac{\epsilon_x - \epsilon_y}{2}
ight)^2 + \left(rac{\gamma_{xy}}{2}
ight)^2}$$

- Mohr's circle can also be used in exactly the same way for strain as it is for stress
- The only difference is that the vertical axis is tensor strain, or  $\gamma_{xy}/2$

## example 10.4



The state of plane strain at a point has components of  $\epsilon_x = 250\mu\epsilon$ ,  $\epsilon_y = -150\mu\epsilon$ , and  $\gamma_{xy} = 120\mu\epsilon$ . Determine the principal strains and the direction they act.

## strain rosettes

#### rosettes

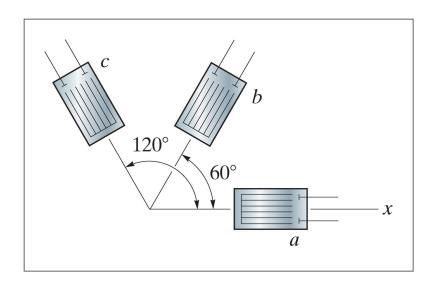
- Normal strain is fairly easy to measure using a strain gage
- Shear strain is more difficult to measure directly, so instead a "rossette" of normal strain gages is used
- We can use the strain transformation equations to determine  $\tau_{\chi y}$

#### rosettes

• Usually, we have  $\theta_a$  = 0,  $\theta_b$  = 90 and  $\theta_c$  = 45 OR  $\theta_a$  = 0,  $\theta_b$  = 60 and  $\theta_c$  = 120

$$egin{aligned} \epsilon_a &= rac{\epsilon_x + \epsilon_y}{2} + rac{\epsilon_x - \epsilon_y}{2} \cos 2 heta_a + rac{\gamma_{xy}}{2} \sin 2 heta_a \ \epsilon_b &= rac{\epsilon_x + \epsilon_y}{2} + rac{\epsilon_x - \epsilon_y}{2} \cos 2 heta_b + rac{\gamma_{xy}}{2} \sin 2 heta_b \ \epsilon_c &= rac{\epsilon_x + \epsilon_y}{2} + rac{\epsilon_x - \epsilon_y}{2} \cos 2 heta_c + rac{\gamma_{xy}}{2} \sin 2 heta_c \end{aligned}$$

## example 10.8



The readings from the rosette shown are  $\epsilon_a=60\mu\epsilon$ ,  $\epsilon_b=135\mu\epsilon$  and  $\epsilon_c=264\mu\epsilon$ . Find the in-plane principal strains and their directions.

# material property relationships

## generalized hooke's law

• We have previously used Hooke's Law in 2D, in 3D we have

$$egin{aligned} \epsilon_x &= rac{1}{E} [\sigma_x - 
u(\sigma_y + \sigma_z)] \ \epsilon_y &= rac{1}{E} [\sigma_y - 
u(\sigma_x + \sigma_z)] \ \epsilon_z &= rac{1}{E} [\sigma_z - 
u(\sigma_x + \sigma_y)] \end{aligned}$$

# generalized hooke's law

• And in shear

$$egin{aligned} \gamma_{xy} &= rac{1}{G} au_{xy} \ \gamma_{yz} &= rac{1}{G} au_{yz} \ \gamma_{xz} &= rac{1}{G} au_{xz} \end{aligned}$$

#### dilatation

- When a material deforms it often changes volume
- The change in volume per unit volume is called "volumetric strain" or dilatation

$$e = rac{\partial V}{\partial V} = \epsilon_x + \epsilon_y + \epsilon_z = rac{1-2
u}{E}(\sigma_x + \sigma_y + \sigma_z)$$

## hydrostatic pressure

- One way of characterizing volumetric response is to apply hydrostatic pressure (equal compression on all sides with no shear)
- Under this case, we have

$$rac{p}{e} = -rac{E}{3(1-2
u)}$$

• We call the term on the right (with no negative sign) the bulk modulus, *k*