AE 737: Mechanics of Damage Tolerance

Lecture 11 - Multiple Site Damage, Mixed-Mode Fracture

Dr. Nicholas Smith

Wichita State University, Department of Aerospace Engineering

February 25, 2020

schedule

- 25 Feb Multiple Site Damage, Mixed-Mode Fracture
- 27 Mar Exam Review, Homework 5 Due
- 3 Mar Exam 1
- 5 Mar Fatigue

outline

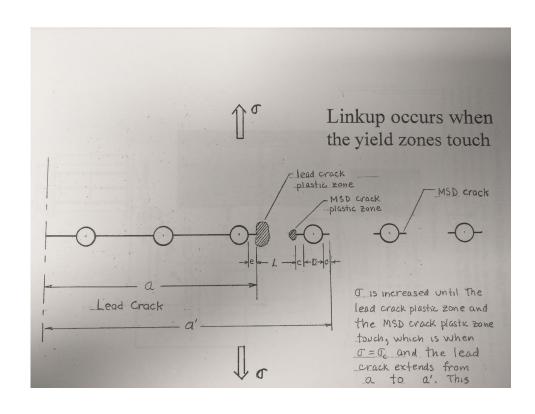
- multiple site damage
- mixed mode fracture

multiple site damage

multiple site damage

- Often damage can accumulate among multiple sources
- This is very common when there are a series of holes, each can develop cracks with a potential to link up
- "link up" occurs when the plastic zones between two adjacent cracks touch

linkup



linkup equation

• We know that

$$R_p = rac{1}{2\pi}igg(rac{K_{Ia}}{\sigma_{YS}}igg)^2$$

$$r_p = rac{1}{2\pi}igg(rac{K_{Il}}{\sigma_{YS}}igg)^2$$

• Where we define the stress intensity factors at a and L as

$$K_{Ia} = \sigma \sqrt{\pi a} \beta_a$$

$$K_{Il} = \sigma \sqrt{\pi l} \beta_l$$

linkup equation

• Since fast cracking occurs when $R_p+r_p=L$, we solve for the condition where $R_p+r_p< L$

$$egin{split} rac{1}{2\pi}igg(rac{K_{Ia}}{\sigma_{YS}}igg)^2 + rac{1}{2\pi}igg(rac{K_{Il}}{\sigma_{YS}}igg)^2 < L \ rac{1}{2\pi\sigma_{YS}^2}ig[K_{Ia}^2 + K_{Il}^2ig] < L \end{split}$$

linkup equation

$$egin{split} rac{1}{2\pi\sigma_{YS}^2}ig[\sigma^2\pi aeta_a^2+\sigma^2\pi leta_l^2ig] < L\ &rac{\sigma^2}{2\sigma_{YS}^2}ig[aeta_a^2+leta_l^2ig] < L\ &\sigma_c=\sigma_{YS}\sqrt{rac{2L}{aeta_a^2+leta_l^2}} \end{split}$$

example

worked link-up example here (http://nbviewer.jupyter.org/github/ndaman /damagetolerance/blob/master/examples/Link-Up.ipynb)

modified linkup equations

- We see that for a brittle material (with a small plastic zone) we predict no effect of "link-up"
- This does not agree with test data
- Even the 2024 predictions don't agree well with test data
- WSU used some empirical parameters to modify the linkup equations and better predict residual strength when multiple site damage is present

modified 2024

- For 2024-T3 we use the following procedure
- First find σ_c from the unmodified equation

$$\sigma_{c,mod} = rac{\sigma_c}{A_1 \ln(L) + A_2}$$

- Where A_1 = 0.3065 and A_2 = 1.3123 for A-basis yield strength and A_1 = 0.3054 and A_2 = 1.3502 for B-basis yield strength
- The same equation can also be used for 2524 with A_1 = 0.1905, A_2 = 0.9683 for A-basis yield and A_1 = 0.2024, A_2 = 1.0719 for B-basis yield

modified 7075

• A similar modification was made for 7075

$$\sigma_{c,mod} = rac{\sigma_c}{B_1 + B_2 L}$$

• Where B_1 = 1.377, B_2 = 1.042 for A-basis yield strength and B_1 = 1.417, B_2 = 1.073 for B-basis yield strength

modified 7075

• However, since general fracture had a closer prediction to real failure than the linkup equation, it may make more sense to modify the brittle fracture equation

$$\sigma_{c,mod} = rac{K_c}{\sqrt{\pi a}(0.856 - 0.496 \ln(L))}$$

mixed mode fracture

mixed-mode fracture

- Most cracks are primarily Mode I, but sometimes Mode II can also have an effect
- We can look at the combined stress field for Mode I and Mode II
- Recall the stress field near the crack tip

stress field

$$\sigma_x = rac{K_I}{\sqrt{2\pi r}} \cosrac{ heta}{2} \left(1 - \sinrac{ heta}{2} \sinrac{3 heta}{2}
ight) \ \sigma_y = rac{K_I}{\sqrt{2\pi r}} \cosrac{ heta}{2} \left(1 + \sinrac{ heta}{2} \sinrac{3 heta}{2}
ight) \ au_{xy} = rac{K_I}{\sqrt{2\pi r}} \sinrac{ heta}{2} \cosrac{ heta}{2} \cosrac{3 heta}{2}$$

mixed-mode fracture

• For Mode II we have

$$\sigma_x = rac{-K_{II}}{\sqrt{2\pi r}} \sinrac{ heta}{2} igg(2 + \cosrac{ heta}{2} \cosrac{3 heta}{2}igg) \ \sigma_y = rac{K_{II}}{\sqrt{2\pi r}} \sinrac{ heta}{2} \cosrac{ heta}{2} \cosrac{3 heta}{2} \ au_{xy} = rac{K_{II}}{\sqrt{2\pi r}} \cosrac{ heta}{2} igg(1 - \sinrac{ heta}{2} \sinrac{3 heta}{2}igg)$$

polar coordinates

- In mixed-mode fracture problems, the crack will generally propagate in a different direction from the initial crack plane
- It is more convenient to handle this scenario in Polar Coordinates
- We can convert stress from Cartesian coordinates to Polar Coordinates using the stress transformation equations

polar coordinates

$$egin{aligned} \sigma_r &= \sigma_x \cos^2 heta + \sigma_y \sin^2 heta + 2 au_{xy} \sin heta \cos heta \ \sigma_ heta &= \sigma_x \sin^2 heta + \sigma_y \cos^2 heta - 2 au_{xy} \sin heta \cos heta \ au_{r heta} &= -\sigma_x \sin heta \cos heta + \sigma_y \sin heta \cos heta + au_{xy} (\cos^ heta - \sin^2 heta) \end{aligned}$$

combined stress field

• When we convert the stress fields from Mode I and Mode II into polar coordinates and combine them, we find

$$\sigma_r = \frac{K_I}{\sqrt{2\pi r}} \left(\frac{5}{4} \cos \frac{\theta}{2} - \frac{1}{4} \cos \frac{3\theta}{2} \right) + \frac{K_{II}}{\sqrt{2\pi r}} \left(-\frac{5}{4} \sin \frac{\theta}{2} + \frac{3}{4} \sin \frac{3\theta}{2} \right)$$

$$\sigma_\theta = \frac{K_I}{\sqrt{2\pi r}} \left(\frac{3}{4} \cos \frac{\theta}{2} + \frac{1}{4} \cos \frac{3\theta}{2} \right) + \frac{K_{II}}{\sqrt{2\pi r}} \left(-\frac{3}{4} \sin \frac{\theta}{2} - \frac{3}{4} \sin \frac{3\theta}{2} \right)$$

$$\tau_{r\theta} = \frac{K_I}{\sqrt{2\pi r}} \left(\frac{1}{4} \sin \frac{\theta}{2} + \frac{1}{4} \sin \frac{3\theta}{2} \right) + \frac{K_{II}}{\sqrt{2\pi r}} \left(\frac{1}{4} \cos \frac{\theta}{2} + \frac{3}{4} \cos \frac{3\theta}{2} \right)$$

- The Maximum Circumferential Stress Criterion assumes that a crack will propagate in the principal direction
- In this direction, the shear stress is o
- The fracture toughness is determined by the Mode I fracture toughness of the material

- **Note:** In this discussion, we will use K_{IC} to differentiate Mode I fracture toughness from Mode II fracture toughness. This does NOT necessarily mean we are referring to plane strain fracture toughness
- Thus fracture begins when

$$\sigma_{ heta}(heta_P) = \sigma_{ heta}(heta=0, K_{II}=0, K_I=K_{Ic}) = rac{K_{IC}}{\sqrt{2\pi r}}$$

- Following the above assumptions, we can solve these equations to find θ_p
- Note: This assumes that we know both K_I and K_{II} , in this class we have not discussed any Mode II stress intensity factors, so they will be given.

• In this case it simplifies to

$$K_I\sin heta_p+K_{II}(3\cos heta_p-1)=0$$

• and

$$4K_{IC}=K_{I}\left(3\cosrac{ heta}{2}+\cosrac{3 heta}{2}
ight)-3K_{II}\left(\sinrac{ heta}{2}+\sinrac{3 heta}{2}
ight)$$

maximum circumferential stress criterion

• The general form for a Mode II stress intensity factor is

$$K_{II} = au \sqrt{\pi a} eta'$$

example

Assuming $\sigma = 4\tau$, $K_{IC} = 60 \text{ ksi} \sqrt{\text{in}}$, and 2a = 1.5 in.

Note: Assume $\beta = \beta' = 1$

principal stress

- In the maximum circumferential stress criterion, we found the principal stress direction near the crack tip in polar coordinates
- We can also find the principal direction (if there were no crack) in Cartesian coordinates
- **Note:** This is not mathematically rigorous, but much easier to calculate and sometimes it's close enough

principal stress

• If we make a free body cut along some angle θ we find, from equilibrium

$$egin{aligned} 0 &= \sigma_{ heta} dA - \sigma_x dA \sin^2 heta - \sigma_y dA \cos^2 heta + 2 au_{xy} dA \cos heta \sin heta \ \sigma_{ heta} &= \sigma_x \sin^2 heta + \sigma_y \cos^2 heta - 2 au_{xy} \sin heta \cos heta \ rac{\partial \sigma_{ heta}}{\partial heta} &= (\sigma_x - \sigma_y) \sin 2 heta_p - 2 au_{xy} \cos 2 heta_P \ an 2 heta_P &= rac{2 au_{xy}}{\sigma_x - \sigma_y} \end{aligned}$$

principal stress

- As before, we consider crack propagation purely due to Mode I
- In the principal stress criterion, we find the maximum Mode I stress as a function of the remote applied stress

$$\sigma_{P1}=C\sigma$$

• We then find the remote failure stress by

$$\sigma_c = rac{K_{IC}}{C\sqrt{\pi a}eta}$$

example

Assuming $\sigma = 4\tau$, $K_{IC} = 60 \text{ ksi} \sqrt{\text{in}}$, and 2a = 1.5 in.

Note: Assume $\beta = \beta' = 1$

example

worked mixed-mode fracture example here (http://nbviewer.jupyter.org/github/ndaman /damagetolerance/blob/master/examples/Mixed%20Mode%20Fracture.ipynb)