

AE 737 - MECHANICS OF DAMAGE TOLERANCE

LECTURE 18

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SCHEDULE

- 31 Mar - Strain based fatigue, project abstract due
- 5 Apr - Crack Growth, Homework 7 due, Homework 8 assigned
- 7 Apr - Crack Growth, Stress Spectrum
- 12 Apr - Retardation, Boeing Commercial Method
- 14 Apr - Exam Review, Homework 8 Due
- 19 Apr - Exam 2
- 21 Apr - Exam Solutions, Damage Tolerance

1. strain based fatigue

STRAIN BASED FATIGUE

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- It is still valid for high cycle fatigue
- Does not include crack growth analysis or fracture mechanics

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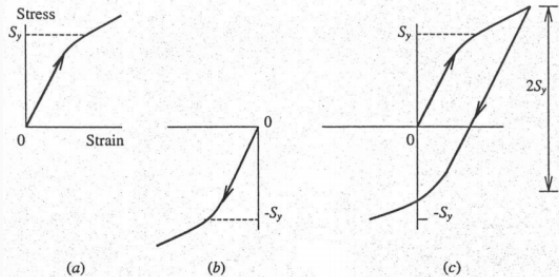
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- Generally plotted on log-log scale

- We can separate the total strain into elastic and plastic components

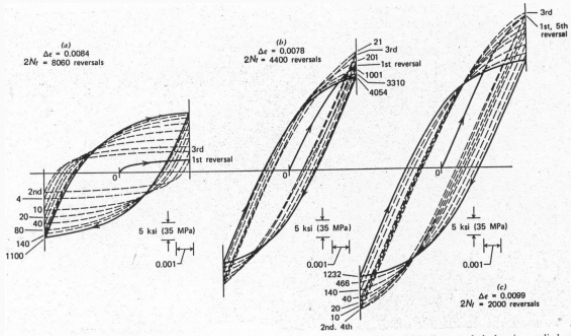
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$$\epsilon_a = \epsilon_{ea} + \epsilon_{pa} \quad (18.1)$$

PLASTIC STRAIN



HYSTERESIS LOOPS



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$$\epsilon_a = \frac{\sigma_a}{E} + \left(\frac{\sigma_a}{H'} \right)^{\frac{1}{n'}} \quad (18.2)$$

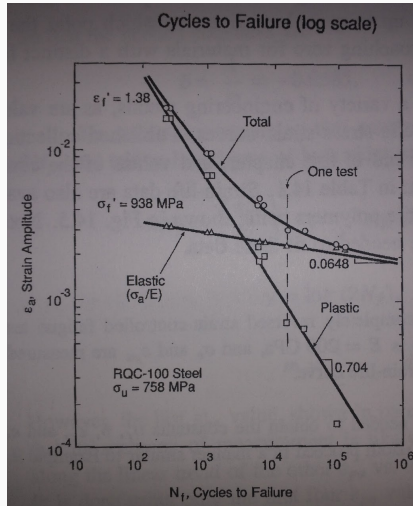
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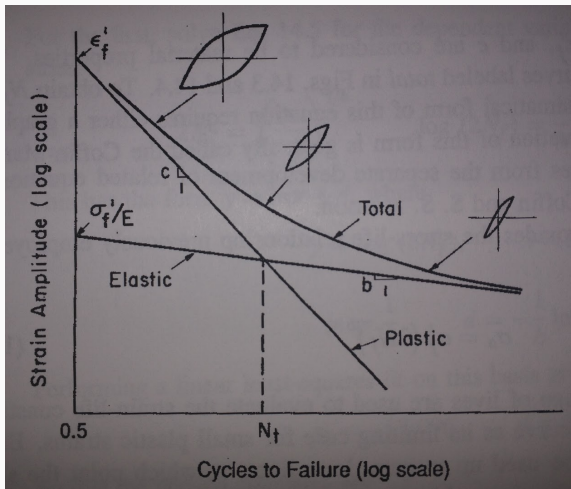
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- Once for total strain, once for plastic strain, and once for elastic strain
- Since plastic strain and elastic strain vary by the number of cycles, a hysteresis loop from half the fatigue life is generally used
- This is considered representative of stable behavior

EXPERIMENTAL DATA





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$$\epsilon_{ea} = \frac{\sigma'_f}{E} (2N_f)^b \quad (18.4)$$

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$$\epsilon_a = \frac{\sigma'_f}{E} (2N_f)^b + \epsilon'_f (2N_f)^c \quad (18.6)$$

Data from p. 270

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$$N_t = \frac{1}{2} \left(\frac{\sigma'_f}{\epsilon'_f} \right)^{\frac{1}{c-b}} \quad (18.7)$$

- If we consider the equation for the cyclic stress train curve

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- We can consider the plastic portion and solve for σ_a

$$\sigma_a = H' \epsilon_{pa}^{n'} \quad (18.9)$$

- We can eliminate $2N_f$ from the plastic strain equation

$$\epsilon_{pa} = \epsilon'_f(2N_f)^c \quad (18.10)$$

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- We then compare with 18.9 and find

$$H' = \frac{\sigma'_f}{(\epsilon'_f)^{b/c}} \quad (18.12a)$$

$$n' = \frac{b}{c} \quad (18.12b)$$

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- However, in practice these constants are fit from different curves
- In some cases there can be large inconsistencies in these values
- One cause for this is data that do not lie on a straight line in the log-log domain
- For ductile materials at short lives, the true stresses and strains may differ significantly from engineering stress and strain