Name:

## Homework 4

## Due 20 Feb 2020

- 1. A large steel sheet (0.1 in. thick) containing a center-crack with length 2a = 4in. fractures under remote tensile stress of  $\sigma = 30$ ksi. This material has a yield strength of 100 ksi. If you feel you cannot answer any of the following questions with the given data, explain why not.
  - (a) What is the fracture toughness for this sheet of steel?
  - (b) What is  $K_{IC}$  for this material?
  - (c) What remote stress,  $\sigma$  will fracture an identical plate with a 9 in. crack?
  - (d) What is the failure stress for a 4-in. long center crack in a 1-in. thick plate?
  - (e) What is the fracture stress if the original 0.1 in. thick panel contains a 0.1 in. center crack?
- 2. Consider a panel with an edge crack a=2 in.. Predict the failure stress for the following conditions
  - (a) The panel is Aluminum 2024-T351 (L-T direction), 1.5 in. thick, 8 in. wide, and  $\sigma_{YS} = 50$  ksi, at room temperature.
  - (b) Consider an identical panel, but with a crack in the transverse direction (T-L).
  - (c) The panel is Aluminum 7075-T6 (L-T direction), 0.5 in. thick, 8 in. wide, and  $\sigma_{YS} = 72$  ksi at room temperature.
  - (d) Consider the same (7075-T6) panel, at -65°F.
  - (e) The panel is Steel 15-5 PH, 1.0 in. thick, 8 in. wide, with  $\sigma_{YS} = 140$  ksi
- 3. The data file HW4-3.txt contains load-displacement data (P vs. v) for an aluminum alloy of an M(T) specimen. Assuming the specimen had a thickness of 0.1", a yield stress of  $\sigma_{YS} = 48$  ksi, a width of 50", Young's Modulus E = 18Msi, Poisson's ratio of  $\nu = 0.33$  and half-span displacement measurement points at Y = 0.75":
  - (a) use the secant method to calculate the effective crack length throughout the test and plot the  $K_R$  curve
  - (b) find  $K_C$  using the tangent curve method

**Note:** Use the finite width center-cracked panel formula to calculate  $K_I$ .