AE 737: Mechanics of Damage Tolerance

Lecture 6 - Plastic Zone

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schedule

- 3 Feb Plastic Zone, HW 2 Due, HW 1 Self-grade due
- 8 Feb Fracture Toughness
- 10 Feb Fracture Toughness, HW3 Due, HW 2 Self-grade due
- 15 Feb Residual Strength

outline

- plastic stress intensity ratio
- plastic zone shape
- group problems

plastic stress intensity ratio

plastic stress intensity ratio

- Engineers often use stress intensity to decide which material to use for a certain application
- The ratio of plastic stress intensity to elastic stress intensity, as a function of yield stress over applied stress, can help illustrate the effects of plasticity for different materials

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plastic stress intensity ratio

For an infinitely wide center-cracked panel, we can solve for K_{Ie}/K_{I} symbolically, in plane stress

$$\begin{aligned} & \textit{K}_{\textit{I}} = \sigma \sqrt{\pi \textit{a}} \\ & \textit{K}_{\textit{Ie}} = \sigma \sqrt{\pi (\textit{a} + \textit{r}_{\textit{p}})} \\ & \textit{r}_{\textit{p}} = \frac{1}{2\pi} \left(\frac{\textit{K}_{\textit{Ie}}}{\sigma_{\textit{YS}}} \right)^{2} \\ & \textit{K}_{\textit{Ie}} = \sigma \sqrt{\pi \left(\textit{a} + \frac{1}{2\pi} \left(\frac{\textit{K}_{\textit{Ie}}}{\sigma_{\textit{YS}}} \right)^{2} \right)} \end{aligned}$$

$$\begin{split} \mathcal{K}_{le}^2 &= \sigma^2 \pi \left(a + \frac{1}{2\pi} \left(\frac{\mathcal{K}_{le}}{\sigma_{YS}} \right)^2 \right) \\ \mathcal{K}_{le}^2 &= \sigma^2 \pi a + \frac{\sigma^2}{2} \left(\frac{\mathcal{K}_{le}}{\sigma_{YS}} \right)^2 \\ \mathcal{K}_{le}^2 &- \frac{\sigma^2}{2} \left(\frac{\mathcal{K}_{le}}{\sigma_{YS}} \right)^2 = \sigma^2 \pi a \\ \mathcal{K}_{le}^2 \left(1 - \frac{\sigma^2}{2\sigma_{Ve}^2} \right) &= \sigma^2 \pi a \end{split}$$

Note: square both sides

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plastic stress intensity ratio

$$\begin{split} &K_{le}^2 = \frac{\sigma^2 \pi a}{1 - \frac{\sigma^2}{2\sigma_{YS}^2}} \\ &K_{le} = \frac{\sigma \sqrt{\pi a}}{\sqrt{1 - \frac{\sigma^2}{2\sigma_{YS}^2}}} \\ &K_{le} = \frac{K_I}{\sqrt{1 - \frac{\sigma^2}{2\sigma_{YS}^2}}} \\ &\frac{K_{le}}{K_I} = \frac{1}{\sqrt{1 - \frac{\sigma^2}{2\sigma_{YS}^2}}} \end{split}$$

Note: We divide both sides by $\left(1-\frac{\sigma^2}{2\sigma_{YS}^2}\right)$

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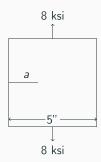
plastic stress intensity ratio

- We can also find the plastic stress intensity ratio numerically for finite width panels
- Panel thickness, yield stress, panel width, crack length could all be variables in this case
- Different heat treatments of metal alloys can give a different yield stress, with most other properties remaining the same
- Typical crack lengths can vary based on inspection cycles

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example

- You are asked to design an inspection cycle for a panel
- Consider the plastic stress intensity ratio, and the effect of varying crack lengths on it



example

online example here1

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plastic zone shape

¹https://colab.research.google.com/drive/1Bb-eznneklW_BILR8po3_fQROB_t56_w?usp=sharing

plastic zone shape

- Although we drew a circle to give a rough idea of the plastic zone in Irwin's method, this solution was only 1D
- We considered $\theta = 0$.
- It is advantageous to model the plastic zone shape, we will do so using two different yield theories
- Von Mises and Tresca

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principal stresses

- Principal stresses are often used in yield theories
- We can determine the principal stresses near the crack tip as

$$\begin{split} &\sigma_1 = \frac{K_I}{\sqrt{2\pi r}}\cos\frac{\theta}{2}\left(1+\sin\frac{\theta}{2}\right) \\ &\sigma_2 = \frac{K_I}{\sqrt{2\pi r}}\cos\frac{\theta}{2}\left(1-\sin\frac{\theta}{2}\right) \\ &\sigma_3 = 0 \\ &\sigma_3 = \frac{2\nu K_I}{\sqrt{2\pi r}}\cos\frac{\theta}{2} \end{aligned} \qquad \text{(plane stress)}$$

Von Mises yield theory

- The Von Mises yield theory is also known as the Distortion Energy Yield Theory
- In this yield theory, we assume that failure or yielding occurs when the strain energy exceeds some threshold
- It has been observed that hydrostatic pressure does not generally cause yielding
- We separate the strain energy into two parts, volumetric and distortional
- Only the distortional strain energy is used to determine the yield strength

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Von Mises yield theory

• The distortional strain energy is given by

$$W_d = \frac{1}{12}G\left[(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 \right]$$

· Which for a uniaxially loaded point becomes

$$W_d = \frac{1}{6} G \sigma_{YS}^2$$

We can equate the two cases and solve

$$\begin{split} &\frac{1}{6}G\sigma_{YS}^{2} = \frac{1}{12}G\left[\left(\sigma_{1} - \sigma_{2}\right)^{2} + \left(\sigma_{2} - \sigma_{3}\right)^{2} + \left(\sigma_{3} - \sigma_{1}\right)^{2}\right] \\ &2\sigma_{YS}^{2} = \left(\sigma_{1} - \sigma_{2}\right)^{2} + \left(\sigma_{2} - \sigma_{3}\right)^{2} + \left(\sigma_{3} - \sigma_{1}\right)^{2} \end{split}$$

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Von Mises yield theory

- We can find the plastic zone size, r_p by substituting the principal stress relations into the distortional strain energy equation
- In plane stress we find

$$2\sigma_{YS}^2 = (\sigma_1 - \sigma_2)^2 + (\sigma_2 - 0)^2 + (0 - \sigma_1)^2$$

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Von Mises yield theory

$$\begin{split} 2\sigma_{YS}^2 &= \left(\frac{K_I}{\sqrt{2\pi r_p}}\cos\frac{\theta}{2}\left(1+\sin\frac{\theta}{2}\right) - \right. \\ &\left. \frac{K_I}{\sqrt{2\pi r_p}}\cos\frac{\theta}{2}\left(1-\sin\frac{\theta}{2}\right)\right)^2 + \\ &\left(\frac{K_I}{\sqrt{2\pi r_p}}\cos\frac{\theta}{2}\left(1-\sin\frac{\theta}{2}\right) - 0\right)^2 + \\ &\left(0-\frac{K_I}{\sqrt{2\pi r_p}}\cos\frac{\theta}{2}\left(1+\sin\frac{\theta}{2}\right)\right)^2 \end{split}$$

Von Mises yield theory

After solving we find

$$r_{p}=\frac{\mathcal{K}_{l}^{2}}{2\pi\sigma_{YS}^{2}}\cos^{2}\frac{\theta}{2}\left(1+3\sin^{2}\frac{\theta}{2}\right)$$

• We can similarly solve for r_p in plane strain to find

$$r_p = \frac{K_I^2}{2\pi\sigma_{YS}^2}\cos^2\frac{\theta}{2}\left(1 - 4\nu + 4\nu^2 + 3\sin^2\frac{\theta}{2}\right)$$

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Tresca yield theory

- Tresca yield theory assumes that yielding begins when the maximum shear stress reaches a critical value
- In uniaxial tension this gives

$$\tau_{0} = \tau_{\textit{max}} = \frac{1}{2} \left(\sigma_{\textit{max}} - \sigma_{\textit{min}} \right) = \frac{1}{2} \left(\sigma_{\textit{YS}} - 0 \right) = \frac{\sigma_{\textit{YS}}}{2}$$

Tresca yield theory

 Using the results for principal stress we found previously, we see that

$$\sigma_{\max} = \frac{K_{l}}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \left(1 + \sin \frac{\theta}{2} \right)$$

$$\sigma_{\min} = 0$$

We can substitute and solve as before to find

$$r_{p} = \frac{K_{l}^{2}}{2\pi\sigma_{VS}^{2}}\cos^{2}\frac{\theta}{2}\left(1+\sin\frac{\theta}{2}\right)^{2}$$

Tresca yield theory

- In plane strain, it is not clear whether σ_2 or σ_3 will be σ_{min}
 - We can solve for when σ_2 will be σ_{min}

$$\begin{split} \sigma_2 &< \sigma_3 \\ \frac{\mathcal{K}_l}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \left(1 - \sin \frac{\theta}{2}\right) &< \frac{2\nu \mathcal{K}_l}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \\ 1 - \sin \frac{\theta}{2} &< 2\nu \\ \theta_t &> 2 \sin^{-1} (1 - 2\nu) \end{split}$$

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Tresca yield theory

- When $2\pi \theta_t < \theta < \theta_t$, σ_2 is the minimum, otherwise σ_3 is the minimum
- Note: Error(s) in text p. 101
- Once we have chosen the appropriate minimum stress (σ₂ or σ₃), we can solve for r_p as before

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Tresca yield theory

$$\begin{split} r_p &= \frac{2K_I^2}{\pi\sigma_{\text{YS}}^2}\cos^2\frac{\theta}{2}\sin^2\frac{\theta}{2} & \theta_t l t; \theta \qquad l t; 2\pi - \theta_t \\ r_p &= \frac{K_I^2}{2\pi\sigma_{\text{YS}}^2}\cos^2\frac{\theta}{2}\left(1 - 2\nu + \sin\frac{\theta}{2}\right)^2 & \theta l t; \theta_t, \theta \quad g t; 2\pi - \theta_t \end{split}$$

3D plastic zone shape

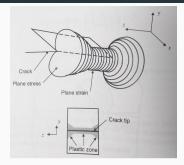


Figure 1: An image showing the 3D plastic zone shape, which looks a little bit like a dumbell. The plastic zone is much larger near the surface, where the material behaves as if in plane stress. In the 23

example

online example $here^2$

²https://colab.research.google.com/drive/1ALdCMw3BzNDn-5clui2nrvfHfWMcN7-5?usp=sharing

group problems

group one

- Calculate the plastic zone size for an infinitely wide, center-cracked panel
- Consider a crack-length of 4 cm, and a yield strength of $\sigma_{YS} = 55$ MPa, with an applied load of $\sigma = 20$ MPa
- Assume the panel is very thin

group two

- Calculate the plastic zone size for an infinitely wide, center-cracked panel
- Consider a crack-length of 4 cm, and a yield strength of $\sigma_{YS} = 55$ MPa, with an applied load of $\sigma = 20$ MPa
- Assume the panel is very thick

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group three

- Calculate the plastic zone size for an infinitely wide, center-cracked panel
- Consider a crack-length of 4 cm, and a yield strength of $\sigma_{YS} = 55$ MPa, with an applied load of $\sigma = 20$ MPa
- The panel thickness is t = 0.65 cm

group four

- Find the plastic stress intensity ratio for an infinitely wide, center-cracked panel
- What factors will increase or decrease the plastic stress intensity ratio?