Homework 2

February 5, 2019

1 Homework 2

1.1 Problem 1

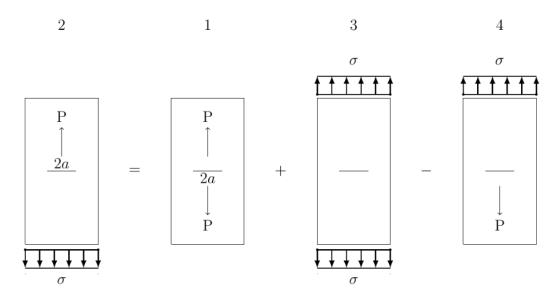
For Panel 1, we use equations (2.6) and (2.7) from the notes, or p. 52 from the text

$$K_I = \frac{P}{t\sqrt{\pi a}}\beta\tag{2.6}$$

$$\beta = \frac{1 - 0.5 \left(\frac{a}{W}\right) + 0.975 \left(\frac{a}{W}\right)^2 - 0.16 \left(\frac{a}{W}\right)^3}{\sqrt{1 - \left(\frac{a}{W}\right)}}$$
(2.7)

In Panel 2 we construct the following superposition

Out[3]:



Since Panels 2 and 4 will give equivalent stress intensity factors, we can add Panel 4 to both sides to get

$$K_{I2} + K_{I4} = K_{I1} + K_{I3}$$

 $2K_{I2} = K_{I1} + K_{I3}$
 $K_{I2} = \frac{1}{2}(K_{I1} + K_{I3})$

We can now substutite the known equations for K_{I1} and K_{I3} to find

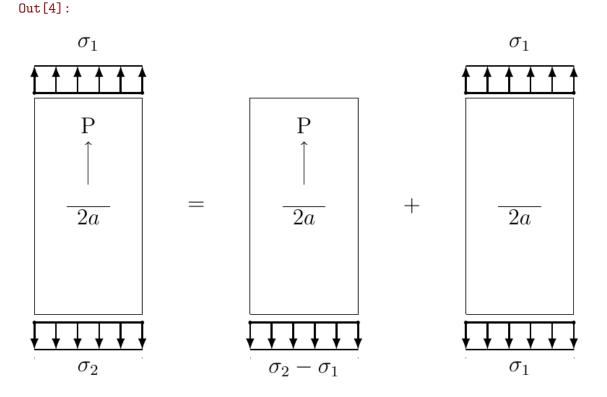
$$K_{I2} = \frac{1}{2} \left(\frac{P}{t\sqrt{\pi a}} \beta_1 + \sigma \sqrt{\pi a} \sqrt{\sec(\pi a/W)} \right)$$
$$\beta_1 = \frac{1 - 0.5 \left(\frac{a}{W} \right) + 0.975 \left(\frac{a}{W} \right)^2 - 0.16 \left(\frac{a}{W} \right)^3}{\sqrt{1 - \left(\frac{a}{W} \right)}}$$

1.2 Problem 2

For a corner crack as indicated in Problem 2, one possible method is to superpose the lug corner-crack and bending corner-crack solutions (p. 63 and p. 64).

1.3 Problem 3

This problem is very similar to Panel 2 from Problem 1. If we superpose the solution from Problem 1 with a center crack under remote stress we will have the same conditions as here.



1.4 Problem 4

In this case we will have two crack tips, one at A and one at B, with different stress intensity factors. We need to consider the effects of both edges, and the hole, but since no height dimension is given, we assume a negligible finite height effect.

From the left edge, we have a=1.25 and b=2.45, looking at the chart gives $\beta_A=1.10$ and $\beta_B=1.06$. From the right edge, we have a=1.25 and b=6-2.45=3.55, notice that we switch the A and B lines from the chart to match our labeling, and we find $\beta_A=1.03$ and $\beta_B=1.05$.

For the circular hole we have R=0.5, c=2.55, and b=2.05. This gives $\beta_A=1.02$ and $\beta_B=1.07$. In summary we have

β_B
1.06
1.05
1.07

We have about 1% difference between methods 1 and 2 for both crack tip A and B, which means we have very little interaction of edge effects and we would expect both values to be fairly accurate.

We use $K_I = \sigma \sqrt{\pi a \beta}$ to find the stress intensity factor

```
#method one
k_a_1 = s*np.sqrt(np.pi*a)*beta_a_1
k_b_1 = s*np.sqrt(np.pi*a)*beta_b_1

#method 2
k_a_2 = s*np.sqrt(np.pi*a)*beta_a_2
k_b_2 = s*np.sqrt(np.pi*a)*beta_b_2

print k_a_1
print k_a_2
print k_b_1
print k_b_2

34.18369794185185
34.351941185635226
35.0754465838132
35.39974583993982
```

1.5 Problem 5

Since we are dealing with cracks along a curved boundary, both "short" and "long," we will use the combined method to interpolate between the two solutions.

For a short crack, we start by finding the stress concentration factor from p. 84. For that chart, $r/d = \frac{0.25}{4 - (0.1 + 0.25)}$, we also have $D/d = \frac{4}{4 - (0.1 + 0.25)}$

```
In [7]: r = 0.25
    d = 4-(.1+.25)
    D = 4
    print r/d
    print D/d
```

0.0684931506849

1.09589041096

We find on the chart that $K_{tn} = 2.5$. Next we relate the given panel to one with a hole in the center and a very short crack on one side of the hole. For this case we find $K_{tg} = 3.05$.

Our next task is to convert the net stress equation into global stress. We do this by making a free-body cut through the crack plane. (See Example 6 on p. 86 of text). Since the force on the cut face acts slightly off-center, we need to also include a bending moment for the body to be in equilibrium.

```
In [8]: s = 6 #ksi, applied stress
    t = 0.375
    P = s*D*t #equivalent force from applied stress
    M = P*(2-d/2) #moment equal to force times eccentricity of load at mid-plane
    sn = P/(d*t) + M*(d/2)/(t*3.65**3/12) #net stress is force/area + My/I
```

 $K_{tg}\sigma = K_{tn}\sigma_n$ (equate net-stress concentration factor to global stress concentration factor)

$$K_{tg} = K_{tn}\sigma_n/\sigma$$

3.52786639144

We can now find what σ_A needs to be by equation $\sigma_{max,B}$ with $\sigma_{max,A}$

$$\sigma_{max,B} = \sigma_{max,A}$$

$$K_{tgB}\sigma = K_{tgA}\sigma_A$$

$$\frac{K_{tgB}}{K_{tgA}}\sigma = \sigma_A$$

6.94006503235

Since $K_{IA} = K_{IB}$ for short cracks, we can now find K_I using a crack on one side of a hole (eq 2.12)

```
In [11]: import numpy as np
    rc = 1 #for short cracks c = 0
    B_3 = .7071+.7548+.3415+.6420+.9196
    Fw = 1/np.cos(np.pi*.25/4)
    Fww = 1 #because c=0
    B_s = B_3*Fw*Fww*K_tg/3.05
    B_s
```

Out[11]: 3.9684728996351324

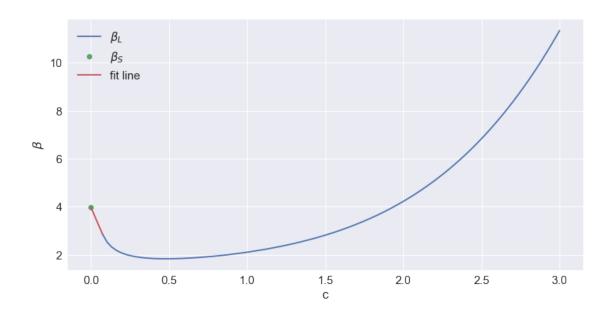
So the β_S for short cracks is 3.97. For long cracks we use the formula for edge cracks in a finite panel, where the extra geometry, e is 0.35

```
In [12]: e = 0.35
    W = 4.0
    c = np.linspace(0,1,100)
    aw = (c+e)/W
    B = 1.122 - 0.231*aw + 10.55*aw**2 - 21.71*aw**3 + 30.82*aw**4
    B_l = np.sqrt((c+e)/c)*B
```

D:\Anaconda\lib\site-packages\ipykernel_launcher.py:6: RuntimeWarning: divide by zero encountere

We now find the tangent curve to generate a cohesive plot

```
In [13]: from scipy import interpolate
         #interpolate our discrete points
         spl = interpolate.splrep(c[1:],B_l[1:])
         x1 = 0.0743 #guess, adjust until they match
         fa = interpolate.splev(x1,spl,der=0)
         fprime = interpolate.splev(x1,spl,der=1)
         print fa-fprime*x1
         print B_s #(to find x1)
3.96788123325723
3.9684728996351324
  And we plot the full result
In [14]: import matplotlib.pyplot as plt
         import seaborn as sb
         sb.set(font_scale=1.5)
         %matplotlib inline
         plt.figure(figsize=(12,6))
         c = np.linspace(x1,3,100)#only plot B_l from end of tangent
         shortc = np.linspace(0,x1)
         aw = (c+e)/W
         B = 1.122 - 0.231*aw + 10.55*aw**2 - 21.71*aw**3 + 30.82*aw**4
        B_1 = np.sqrt((c+e)/c)*B
        plt.figure(figsize=(12,6))
        plt.plot(c,B_1,label=r'$\beta_L$')
        plt.plot(0,B_s,'o',label=r'$\beta_S$')
        plt.plot(shortc,fa+fprime*(shortc-x1),label='fit line')
         plt.xlabel('c')
         plt.ylabel(r'$\beta$')
        plt.legend(loc='best')
Out[14]: <matplotlib.legend.Legend at 0xb48b898>
<Figure size 864x432 with 0 Axes>
```



Visually, we can see that for part a we are using only the long crack formula, calculating K_I :

12.383412965569004

So
$$K_I = 12.38$$
ksi $\sqrt{\text{in}}$.

For part b we need to use our linear interpolation, we find that

3.218144130597965

Substituting this β in gives

7.652735088257687

```
So K_I = 7.65ksi \sqrt{\text{in}}.
```

If we used only the short-crack assumption, we would find $K_I = 3.97(6000)\sqrt{.05\pi}$

```
In [18]: 3.97*6*np.sqrt(np.pi*.05)
Out[18]: 9.440645622897518
```

Which gives a stress intensity factor of $K_I = 9.44$ ksi.

Even though we might assume a 0.05" crack is very short, using the short crack assumptions leads to an error of about 20%.

1.6 Problem 6

We formulate this problem much like Problem 4. For the left crack tip, we have a/b = 0.33, which gives $\beta_A = 1.04$ and $\beta_B = 1.03$. For the right crack tip (with A and B switched from the chart) we have a/b = .05, which gives $\beta_A \approx \beta_B \approx 1$. For the circular hole we have R = 0.3, c = 0.6, and b = 0.3. This gives $\beta_A = 1.19$ and $\beta_B = 1.31$. In summary we have

β_A	β_B
1.04	1.03
1.00	1.00
1.19	1.31

```
In [19]: #method one
    beta_a_1 = 1+ (1.04-1+1.00-1+1.19-1)
    beta_b_1 = 1+ (1.03-1+1.00-1+1.31-1)

#method two
    beta_a_2 = 1.04*1.*1.19
    beta_b_2 = 1.03*1.0*1.31

    print beta_a_1
    print beta_a_2

    print beta_b_1
    print beta_b_2

1.23
1.2376
1.34
1.3493
```

Once again Methods 1 and 2 agree well, indicating little interaction. However we see in this problem that β is dominated by the hole, the edge effects are minimal. The stress intensities are:

```
In [20]: a = 0.1 #in
```

```
#method one
```

```
k_a_1 = np.sqrt(np.pi*a)*beta_a_1
k_b_1 = np.sqrt(np.pi*a)*beta_b_1
```

#method 2

print k_a_1
print k_a_2

print k_b_1
print k_b_2

- 0.6894139196169452
- 0.6936737129414077
- 0.7510688229973225
- 0.7562814648285726

(multiplied by the applied stress)