The CUbesat Solar Polarimeter (CUSP) mission overview

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on behalf of the CUSP Team

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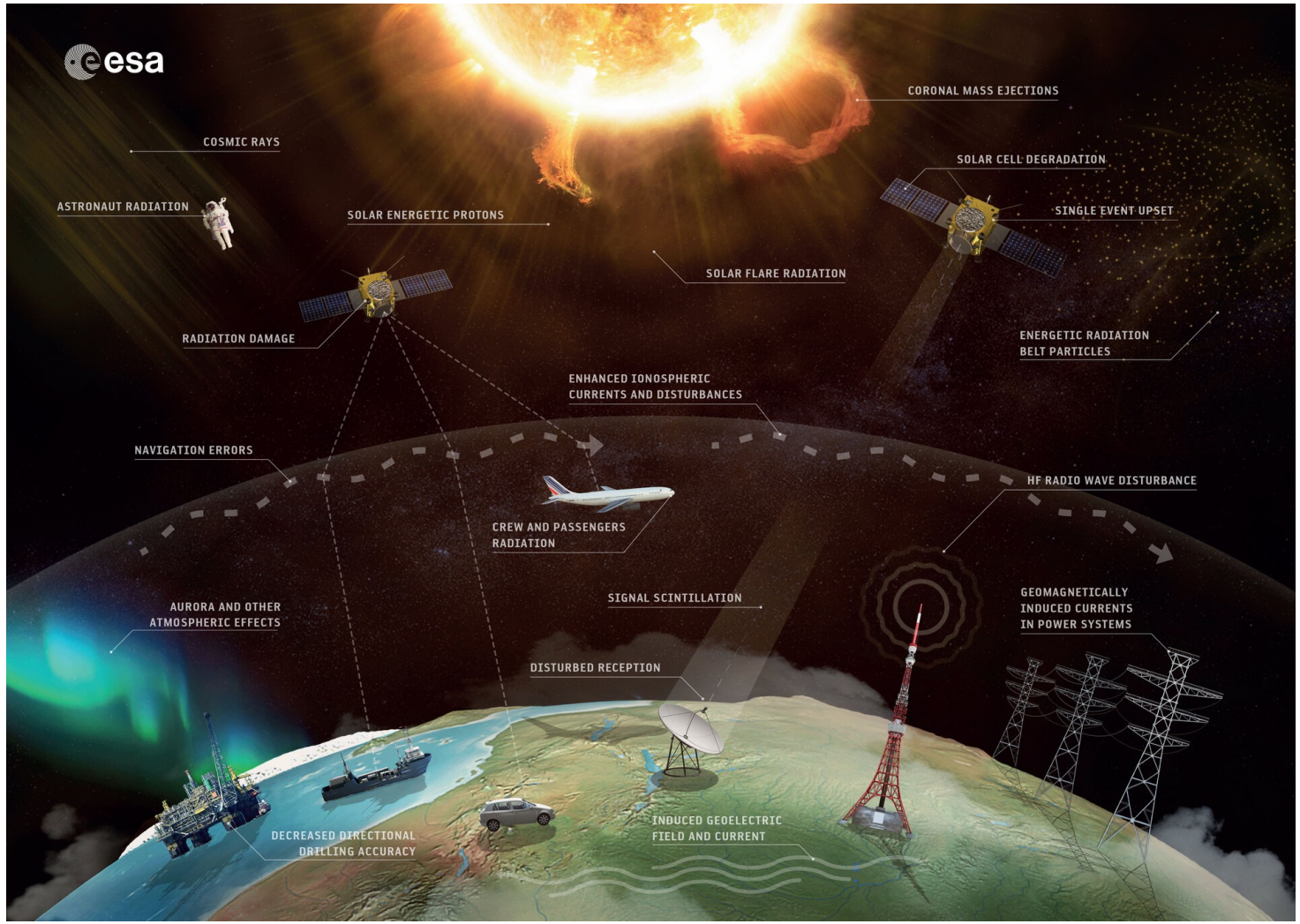
Solar Flares: Heliophysics and Space Weather

- Solar activity, including **Solar flares (SFs)**, can be disruptive for human technological activities in space and on ground
- **The occurrence of SFs** is **very often associated** to Coronal Mass Ejection **(CME)** and Solar Energetic Particle **(SEPs)** events on the ground
- **SF** can also **occur alone** producing a **direct acceleration of particles towards the Earth**

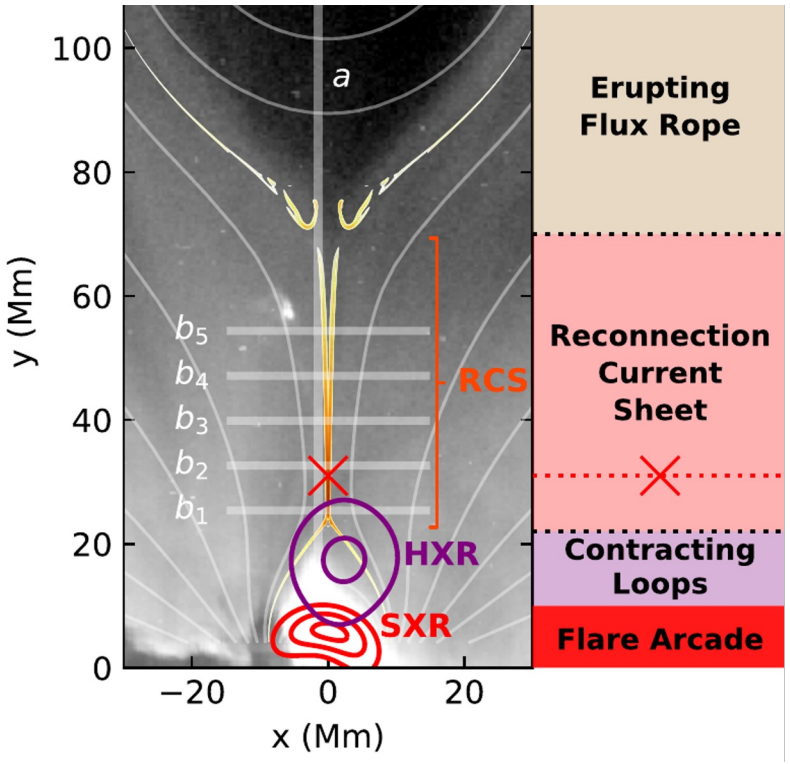


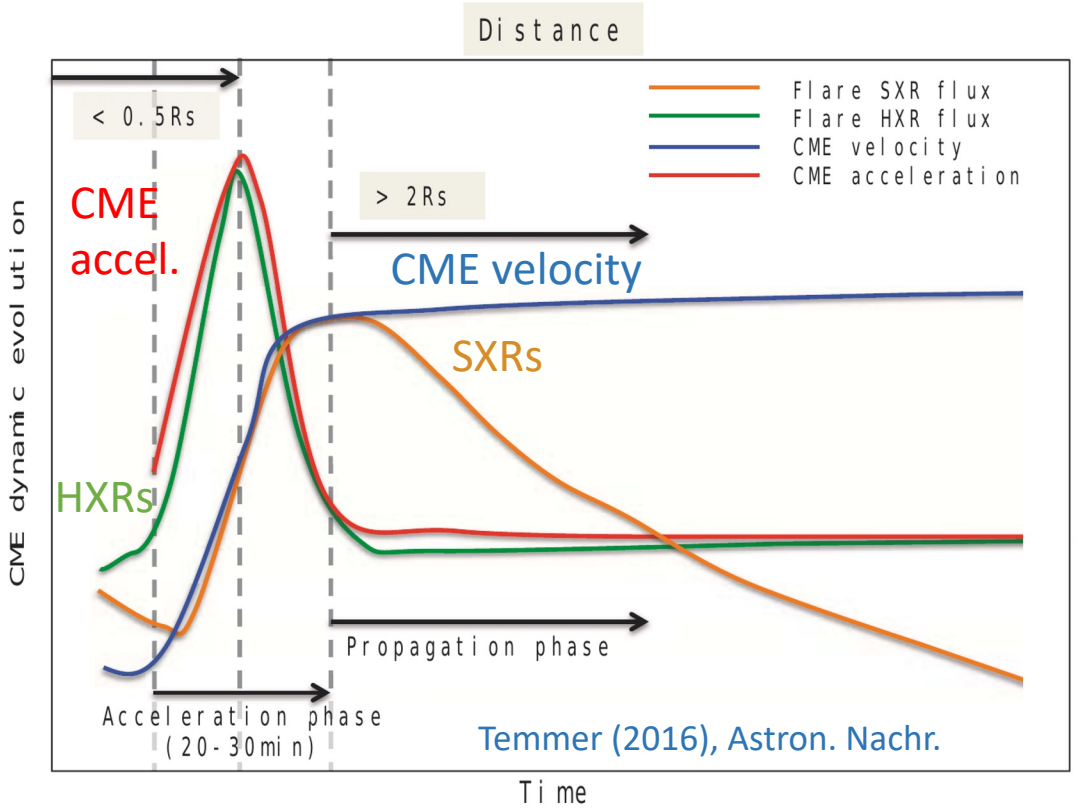
Solar Flares - CME feedback

- Most powerful eruptions associated to powerful flares
- HXRs are related to CME acceleration
- SXRs are related to CME velocity

→ **HXR polarimetry** could improve the knowledge of the **initial conditions** of the eruption of most powerful CME

Chen B. et al. (2020), Nat. Astron.





Temmer (2016), Astron. Nachr.

See also:
Zhang et al. (2001, 2004), ApJ;
Chen & Krall (2003), JGR;
Maričić (2007), Sol. Phys.
Temmer et al. (2008), ApJ Lett.;
Temmer et al. (2010), ApJ.
Berkebile-Stoiser et al., 2012

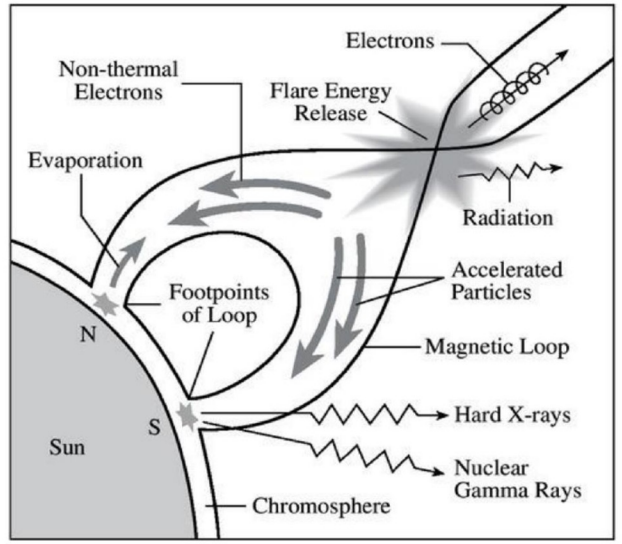
The rapid CME development in the lower corona during the acceleration phase strongly correlates with the associated flare activity.

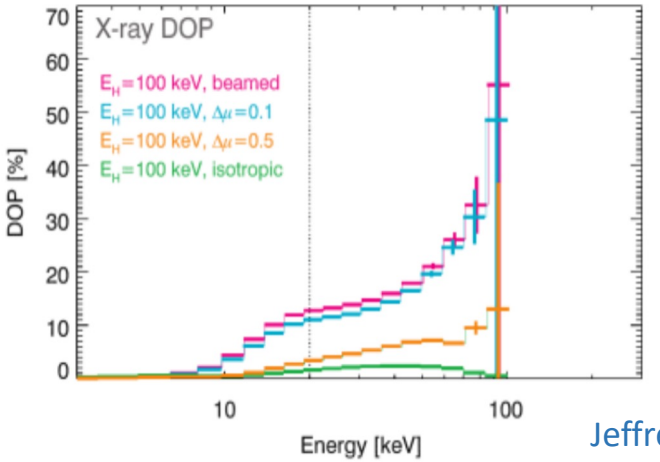
Why Hard X-ray polarimetry of Solar Flares?

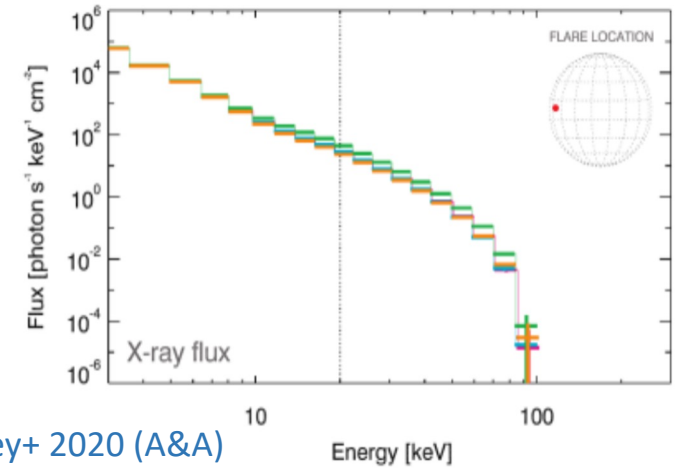
- SFs originate from **magnetic reconnection** in loop structures in solar corona

- SFs energy spectrum in the X-rays is dominated by:
  - **thermal Bremsstrahlung** (due to plasma heating, expected weakly polarized by Emslie & Brown 1980) + emission lines < 10 keV
  - **non-thermal Bremsstrahlung** (at the loop top and footprints, due to particle acceleration along magnetic field lines) expected highly polarized [Zharkova+ (2010)] >10-20 keV

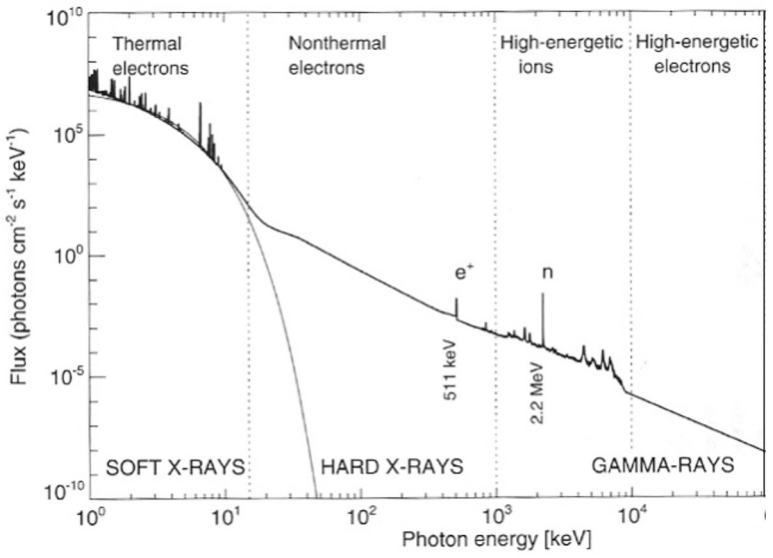
- **X-ray polarimetry** (linear) would allow to **disentangle degeneracies** in **models** of particle beaming and magnetic field structure (also without imaging of the SF)







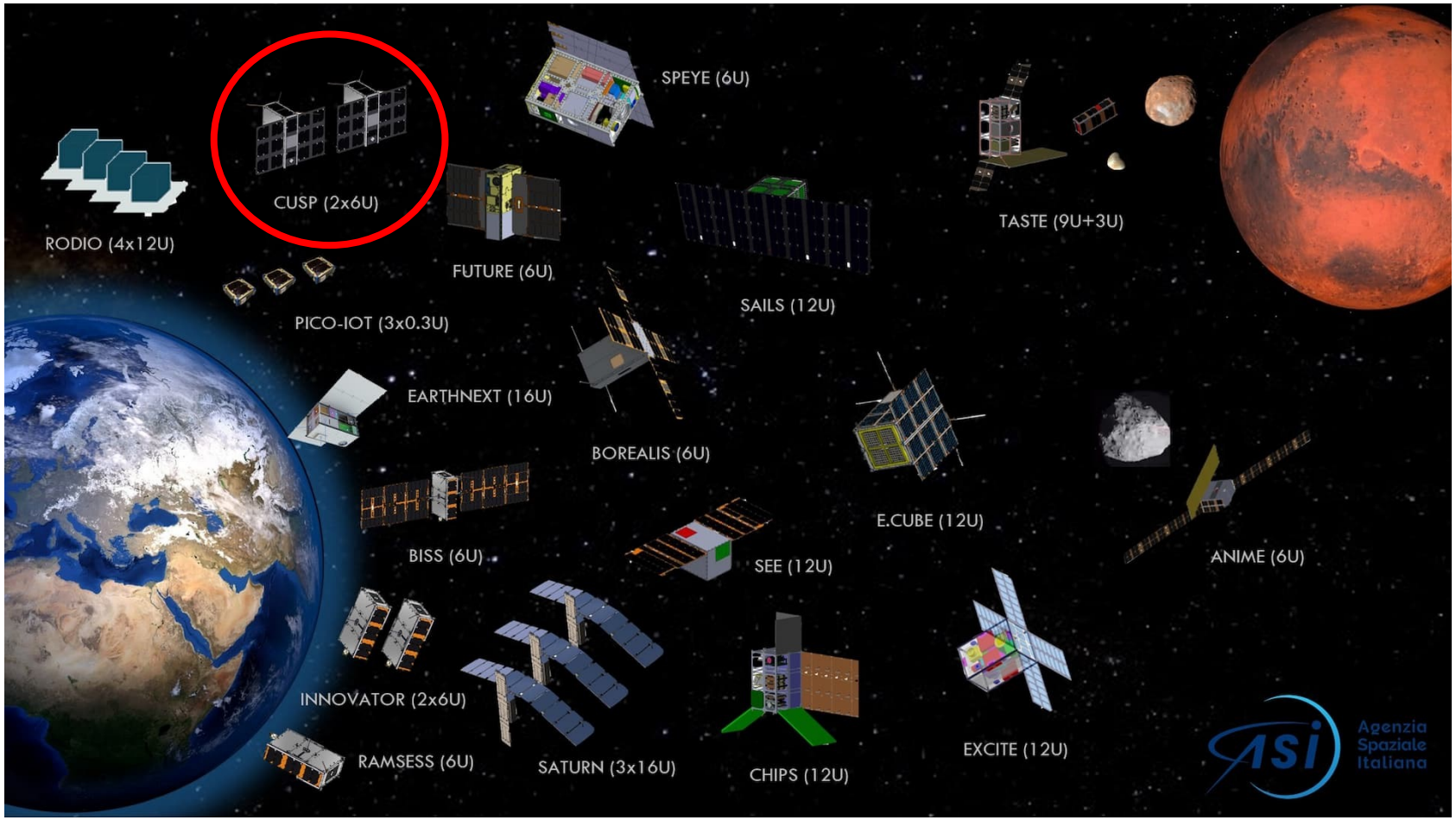
Jeffrey+ 2020 (A&A)



Aschwanden (2005)

The framework

- The **Italian Space Agency** started a new national program named **Alcor** for funding the development of CubeSat technologies and missions
- **CUSP** is one of the 20 selected missions among 49 proposals
  - 22 participants from Research Institutes and Universities and 78 companies, mainly Small and medium-sized enterprises (SMEs)



CUSP Team

- INAF
  - IAPS (Prime and Payload)

    Sergio Fabiani (PI), Enrico Costa, Nicolas De Angelis, Ettore Del Monte, Sergio Di Cosimo, Alessandro Di Marco, Giuseppe Di Persio, Pasqualino Loffredo, Giovanni Lombardi, Gabriele Minervini, Fabio Muleri, Monia Rossi, Alda Rubini, Emanuele Scalise, Paolo Soffitta
  - OAS-Bologna (Detector Simulation)

    Riccardo Campana, Giovanni De Cesare
  - OAR (Lab-SW support)

    Mauro Centrone

- IMT s.r.l. (Platform)

Massimo Perelli, Sergio Bonomo, Giovanni Cucinella , Andrea Negri

- SCAI Connect s.r.l. (Payload Electronics)

Andrea Del Re, Giulia De Iulis, Paolo Leonetti, Alessandro Zambardi

- Università di Bologna – CIRI AERO (Mission Analysis)

Paolo Tortora, Andrea Curatolo, Alfredo Locarini, Dario Modenini

- Università della Tuscia (Ground Segment)

Pierluigi Fanelli, Ilaria Baffo

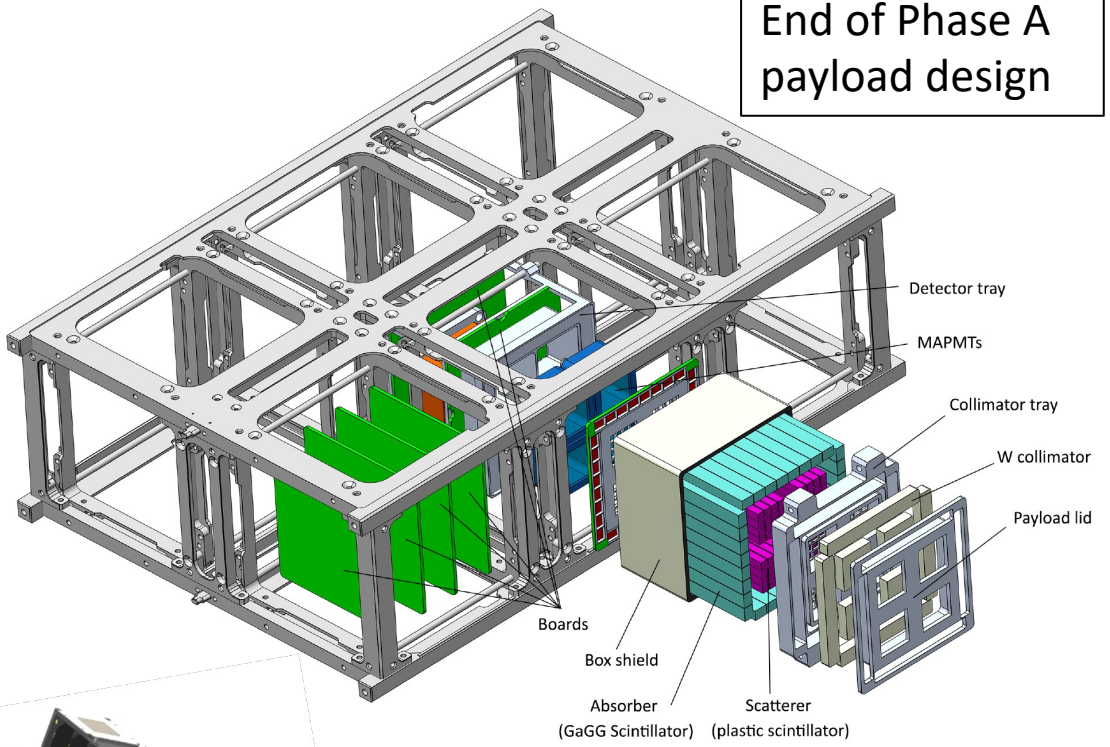
- ASI (Project Control)

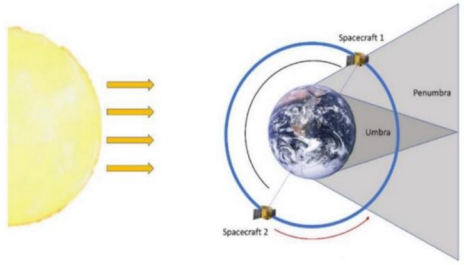
Silvia Natalucci, Daniele Brienza, Immacolata Donnarumma, Andrea Terracciano, Emanuele Zaccagnino

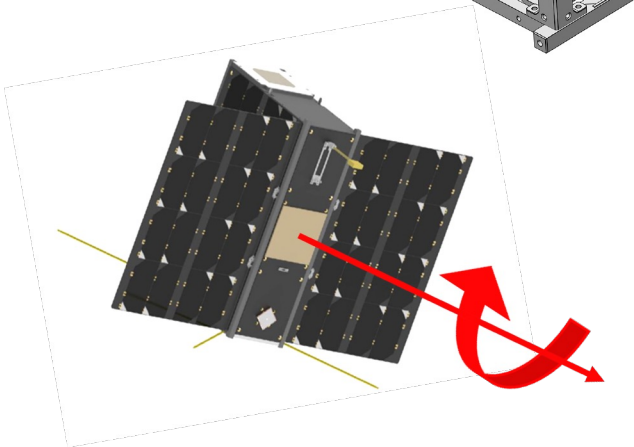
The payload

- **Constellation** of **two 6U CubeSats** orbiting the Earth on SSO orbit (~500-600 km) to observe the Sun
- Monitoring of the Sun with a time fraction >68% during the 3 years nominal life-time
- X-ray polarimetry of Solar Flares in the 25-100 keV energy band
- Each satellite hosts a **dual-phase Compton scattering polarimeter** that exploits **coincidence measurements** between plastic (scatterer) and inorganic (absorber) scintillator rods
- **1 RPM rotation of the spacecraft** around the polarimeter symmetry axis pointing the Sun allows to reduce the systematic effect known as spurious polarization

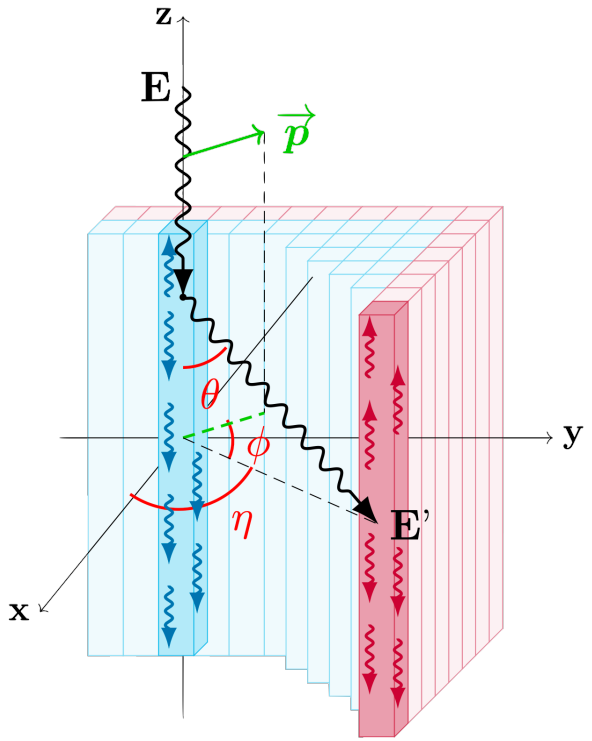
End of Phase A payload design

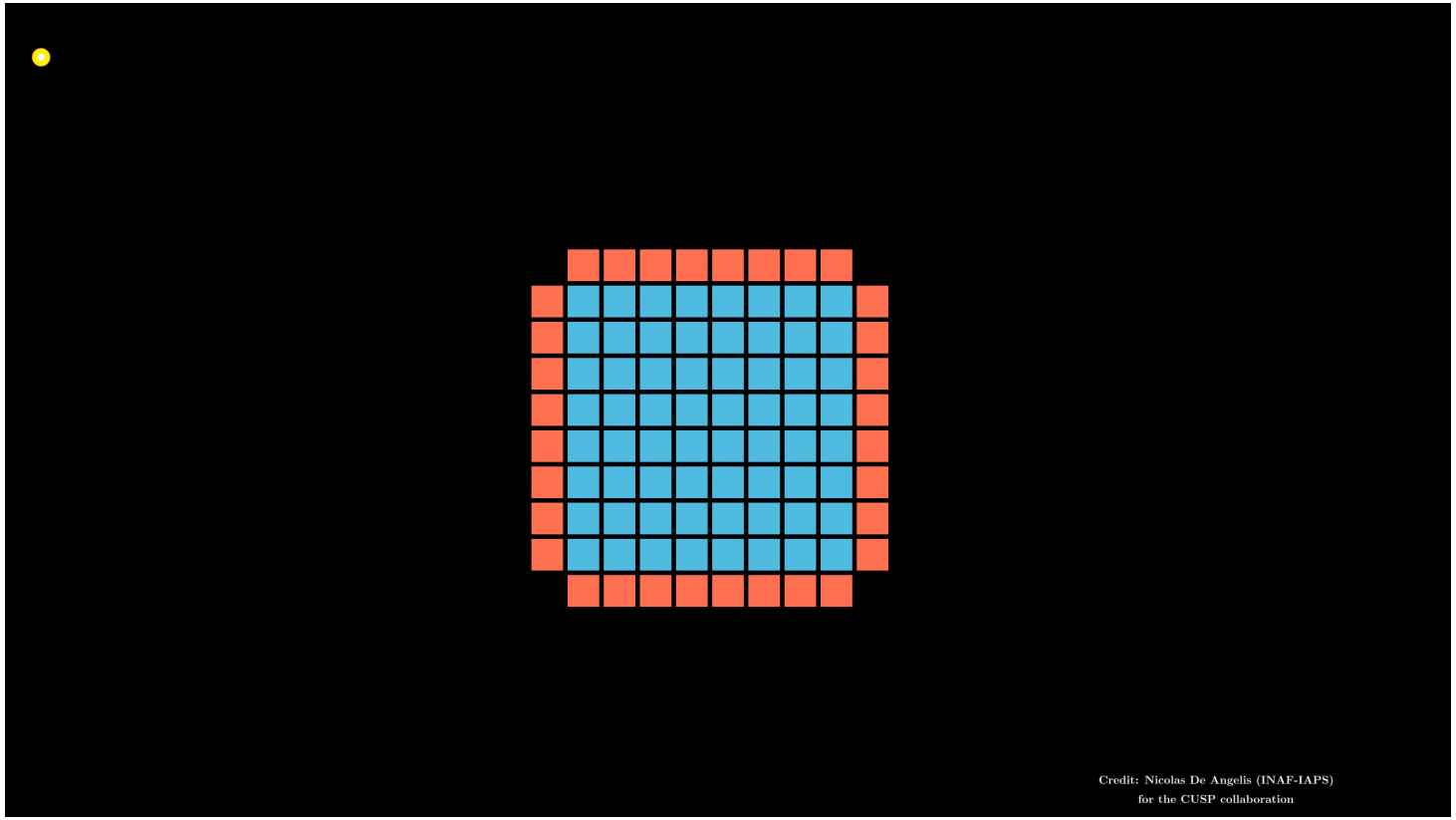


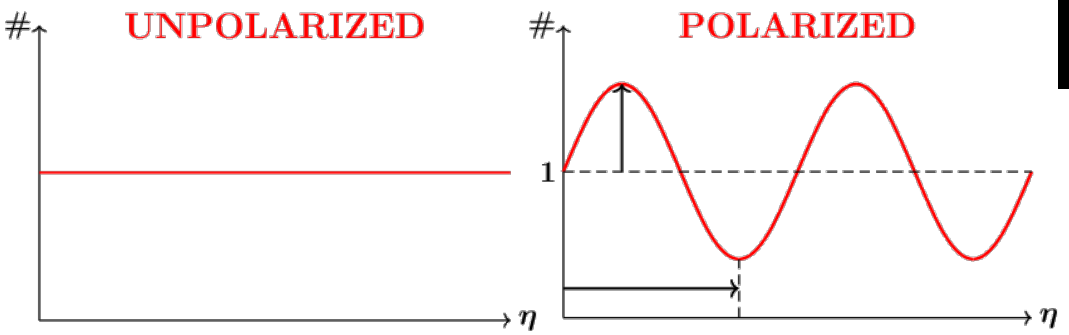




Compton polarimeter working principle





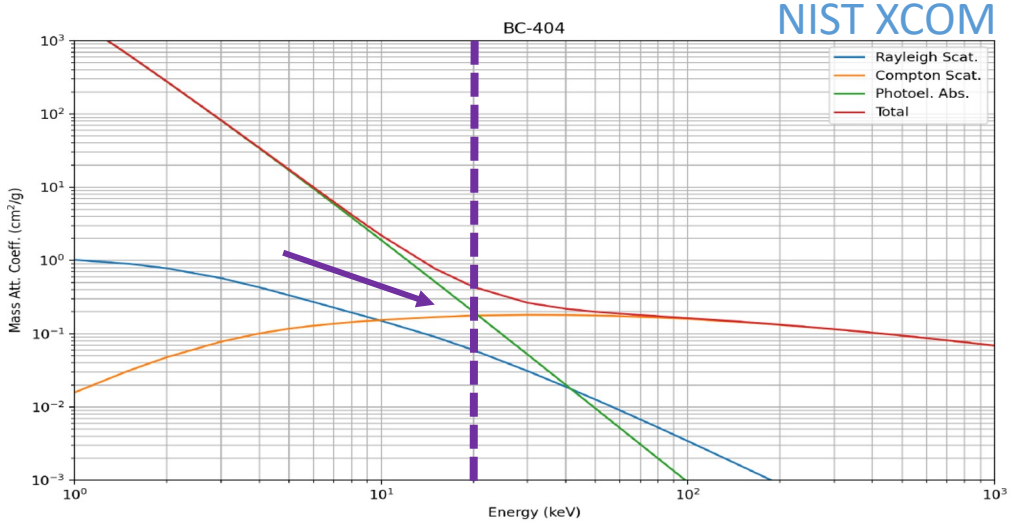


Klein-Nishina: $$\frac{d\sigma}{d\Omega} = \frac{r_e^2}{2}\left(\frac{E'}{E}\right)^2\left[\frac{E'}{E} + \frac{E}{E'} - 2\sin^2(\theta)\cos^2(\phi)\right]$$

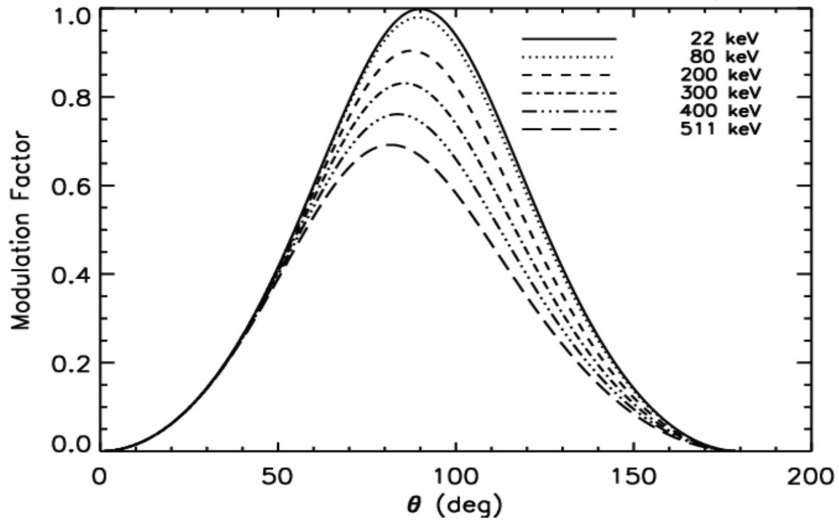
Why dual-phase

- High probability of **scattering** in **plastic** material (4 MAPMTs readout with a MAROC ASIC by WEEROC)
  - 90° scattering produces maximum modulation of the signal
  - @20 keV only 750 eV of Compton energy deposit for scattering at 90°, PMT needed (1-3 optical photons to collect)

NIST XCOM



Fabiani & Muleri, 2014



- High probability of **photoelectric absorption** in the **absorber (GAGG)** material (32 APDs readout with SKIROC ASIC by WEEROC)
- Measurement of **coincidences** Scatterer/Absorber allows effective **background reduction**
- **Fast schedule, no R&D possible,** we need heritage and space proven items as much as possible. We selected APDs from past TSUBAME mission unfortuantely lost in 2015 (and similar MAPMTs) by Hamamatsu (Japan - Yoichi Yatsu 2014, SPIE Proc.)

The payload sensitivity

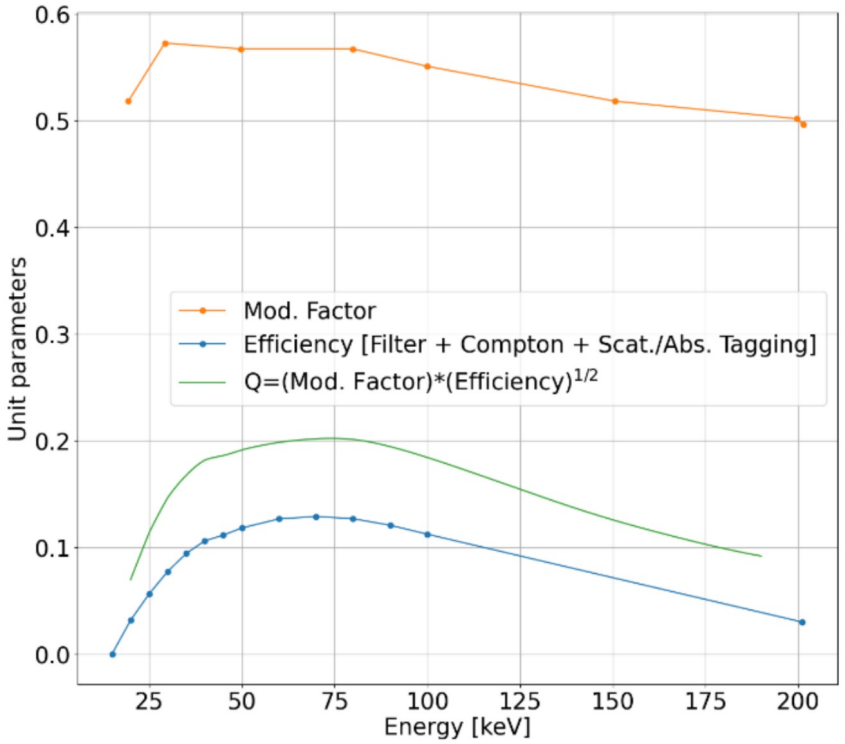
- **Minimum Detectable Polarization** (Weisskopf+ 2010, SPIE, Strohmayer & Kallman 2013, ApJ) in the 25-100 keV energy band (CBE based on benchmark SFs from Saint-Hilaire et al. (2008), Sol. Phys. 250, 53–73)

Flare Class	Integration time (s)	MDP (%) (25-100 keV)
M5.2	284	10.2
X1.2	240	5.0
X10	351	1.1

$$MDP = \frac{4.29}{\mu \cdot R} \cdot \sqrt{\frac{R+B}{T}}$$

$$Q = \mu\sqrt{\varepsilon}$$

R: source rate
B: background rate
T: integration time
μ: modulation factor
ε: quantum efficiency



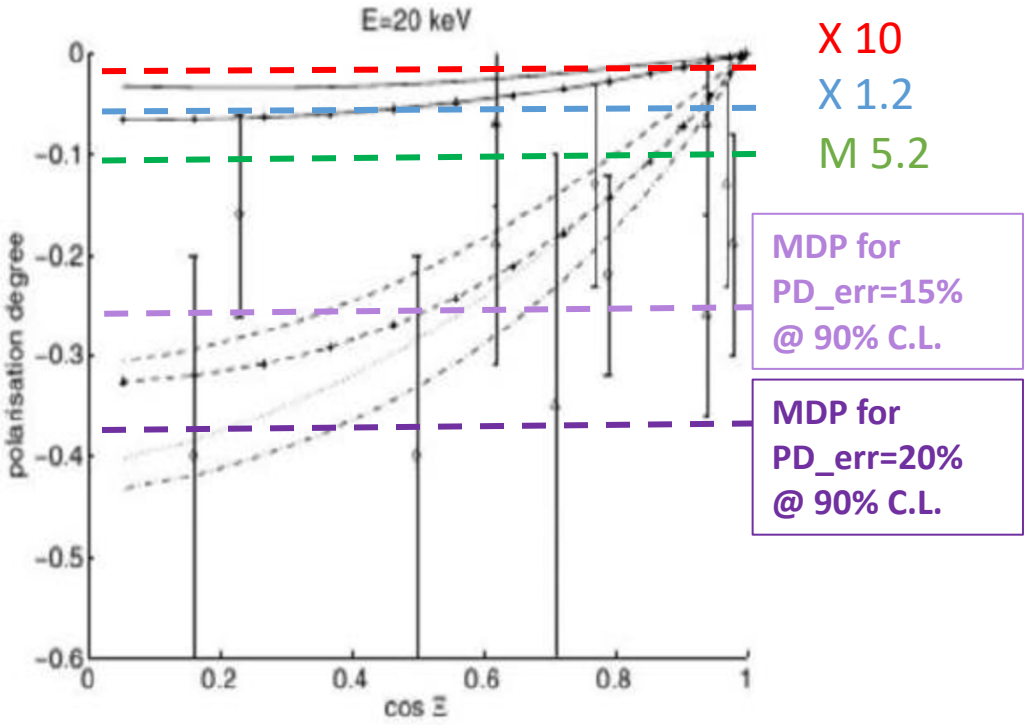
Tagging efficiency by Fabiani+ 2013 Astrop.Phys ., is the probability to detect a scattering event in the plastic if the scattered photon is detected in the absorber.

The payload sensitivity

- MDP~3σ for 1 parameter measurement (Strohmayer & Kallman 2013, ApJ)
- PD errors of about:
  - 15% with a C.L. of 90% (~ 1.645 σ) correspond to an MDP ~27%
  - 20% corresponds to an MDP ~37%

- **CUSP will reduce significantly the MDP wrt past observations.**

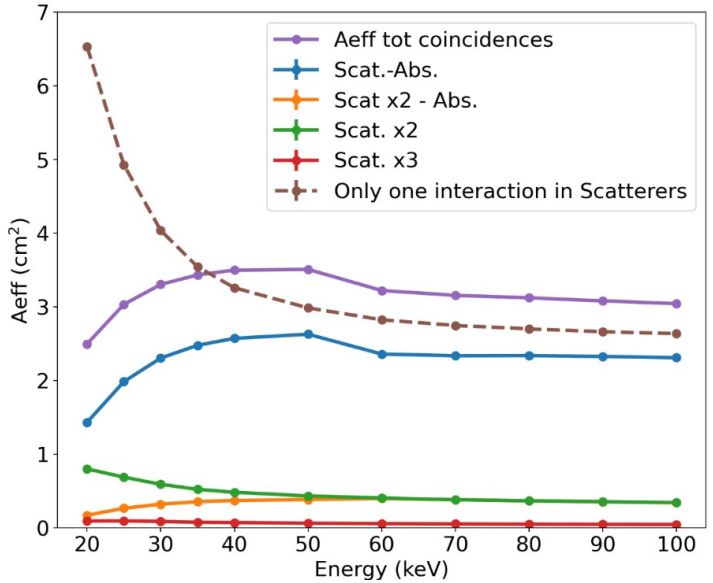
- Flares are expected to be **polarized at tens of %**, few minutes of integration time allow to measure with high significance their polarization

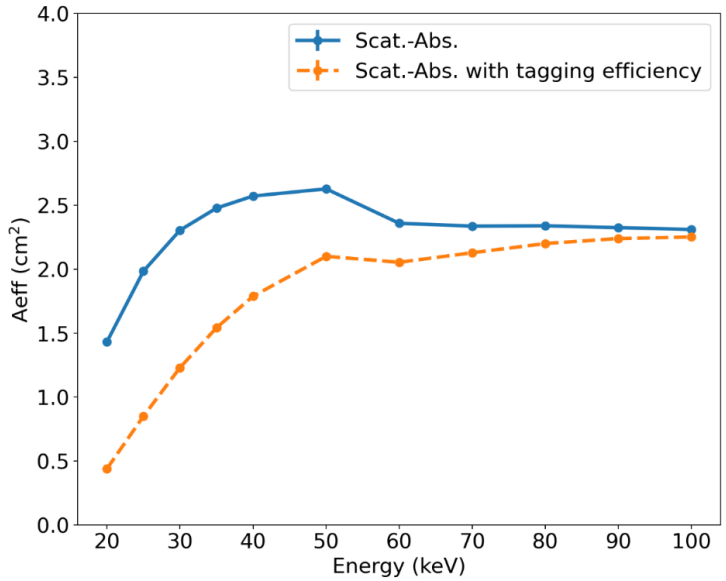


Zharkova+ 2010 (A&A)

Work in progress: design optimization

- Analysis of the **effective area (Aeff)** in progress for design and analysis **optimization**
- Next to come **spurious modulation** and **modulation factor**



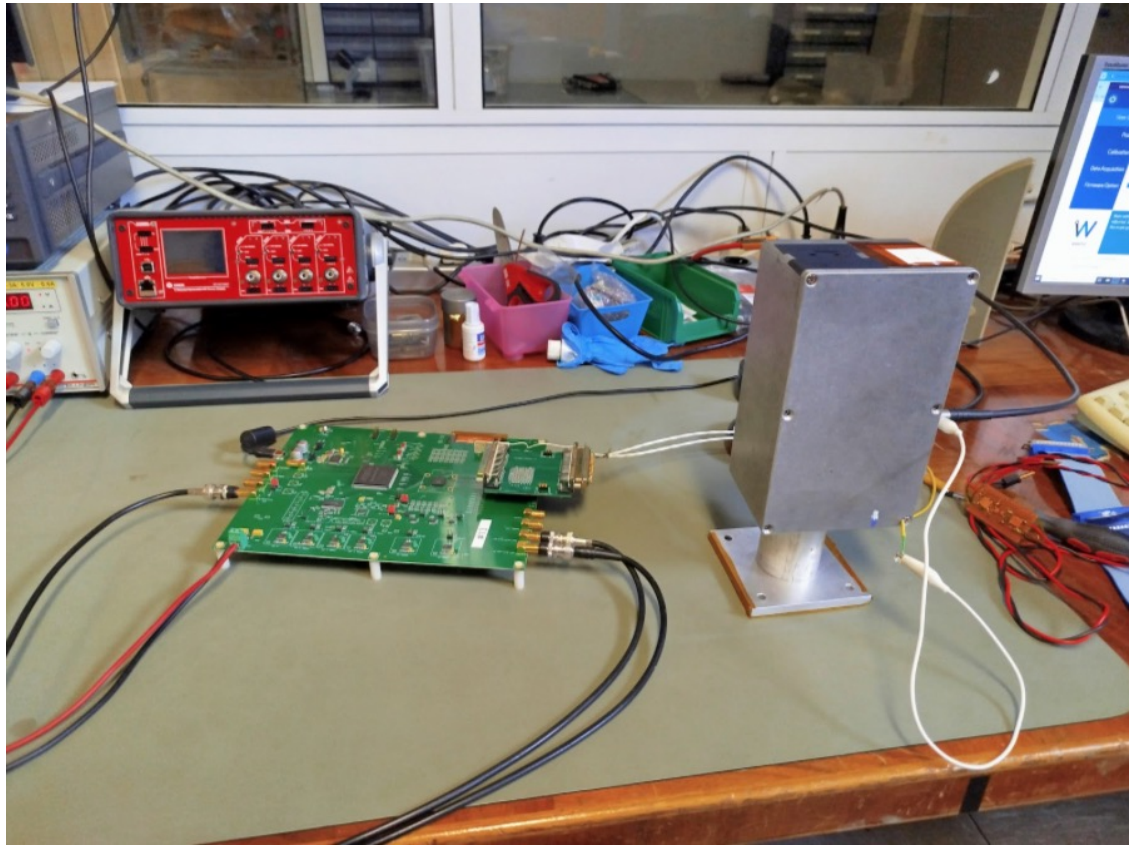


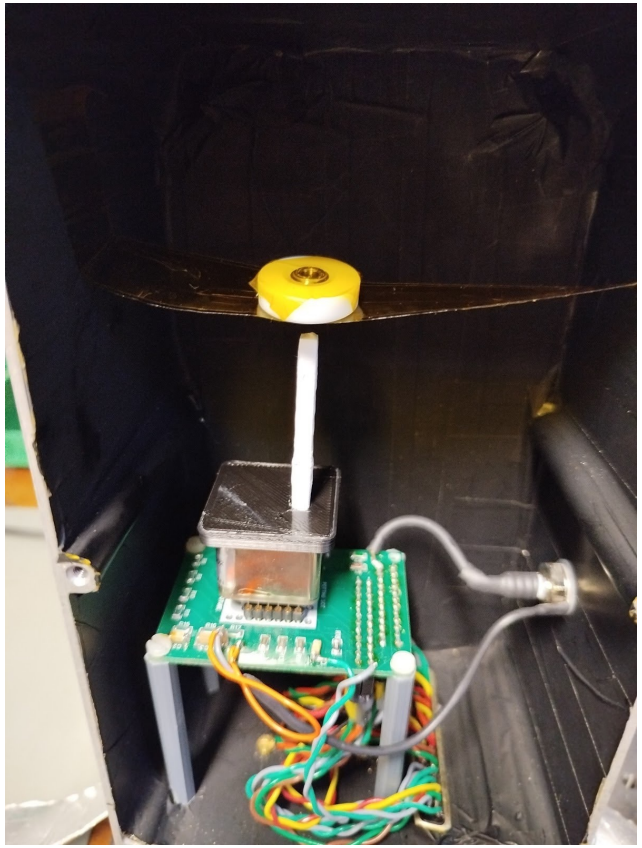
Tagging efficiency
by Fabiani+ 2013
Astrop.Phys .

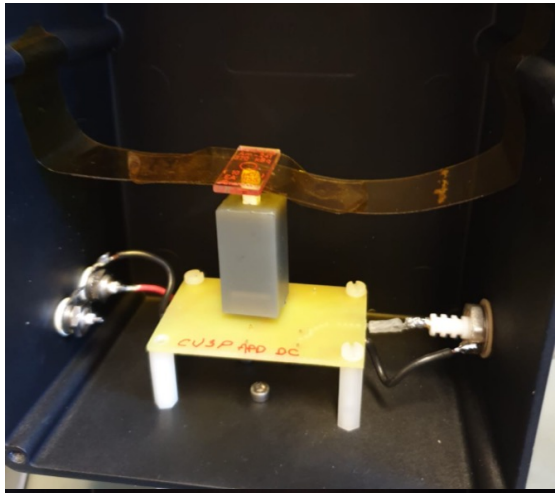
- De Cesare et al. SPIE2024 13093-311, ”The evaluation of the CUSP scientific performance by a GEANT4 Monte Carlo simulation”
- Implementation of **thermo-mechanical simulations** giving some path for design optimization for Phase B
- Lombardi et al. SPIE 2024 13093-307, ”The payload thermo-mechanical design of the CUbesat Solar Polarimeter (CUSP), for Space Weather and Solar flares X-ray polarimetry.”

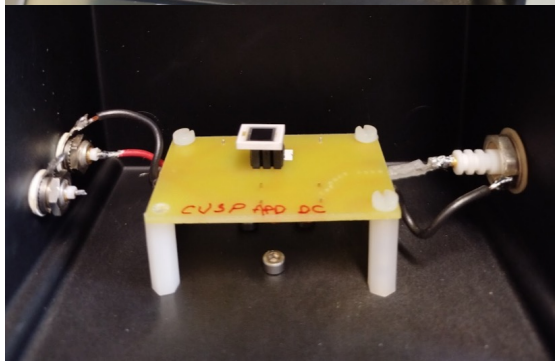
Testing of the two acquisition chains

- CUSP is employing **2 acquisition chains** for the **Absorbers** (GaGG + APD + SKIROC-2A) and **Scatterers** (Plastic + MA-PMTs + MAROC-3A)
- Preliminary tests are being conducted based on ASIC Test Boards (WEEROC) and single scintillator bars coupled to sensors









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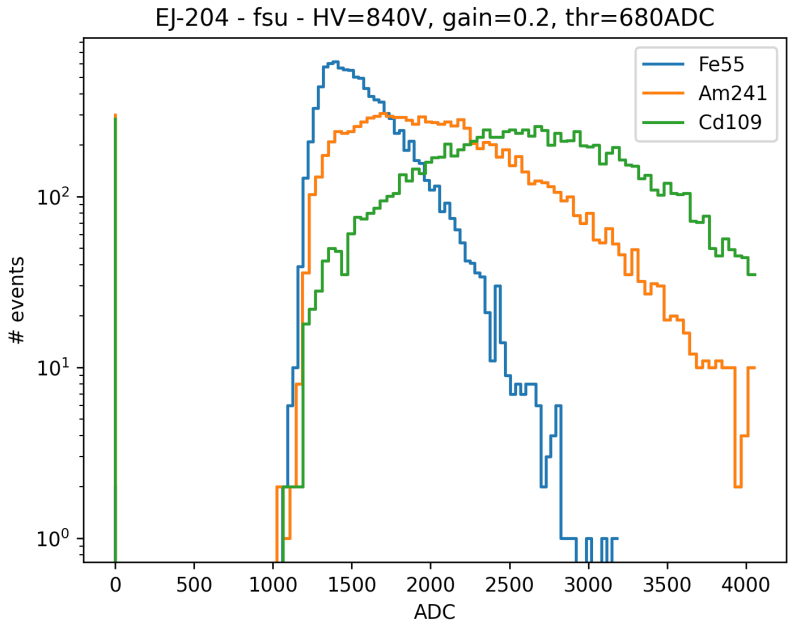
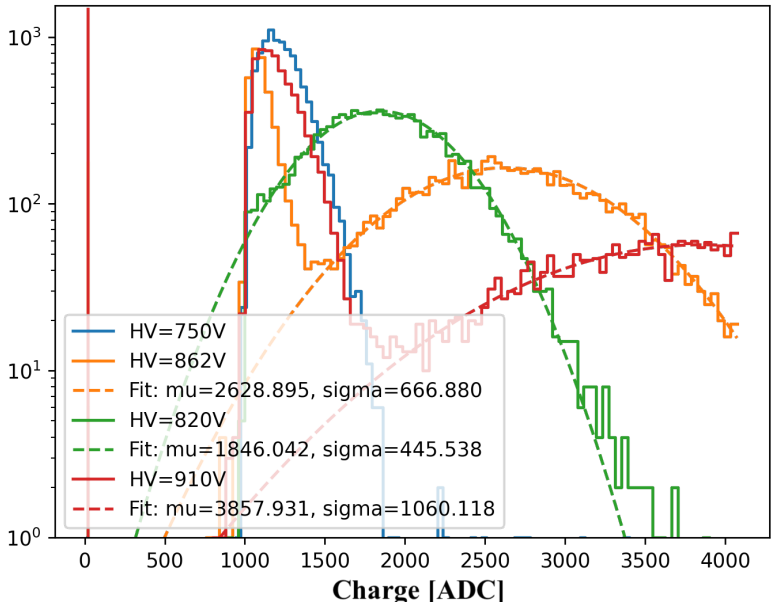
- Preliminary tests are being conducted based on ASIC Test Boards (WEEROC) and single scintillator bars coupled to sensors
- Measuring preliminary performances: energy range, noise optimization and energy threshold, noise rejection with coincidences between the two systems





Impinging Energy (keV)	Scattering angle (deg)	Compton energy deposit (keV)
20	90	0.753
25	90	1.17
30	90	1.66
40	90	2.90
50	90	4.46
60	90	6.31
80	90	10.8
100	90	16.4
	65	10.2

Conclusions & Project Status

- **CUSP** will measure linear X-ray polarization of solar flares for **Heliophysics** and **Space Weather**, with a better MDP w.r.t. past missions
- The 12 months **Phase B** should be starting in **September 2024** to define a preliminary design and deliver a representative prototype of the polarimeter
- Next step would be to propose a 15 months combined Phase C/D
- **Model Philosophy:**
  - 1 detector prototype at the end of Phase B. Representative of the detector front-end (from TRL 3 to TRL 4)
  - 1 payload sub-system Structural Model at the end of Phase B (scintillator bars holding system)
  - 1 payload EQM (design phase B, production and test phase C). Representative of the payload (from TRL 4 to TRL 7)
  - Trade-off assessment to allow ASI to decide if to continue with a 2 CubeSats constallation or with a 1 CubeSat
  - Depending on the trade-off 1 or 2 CubeSats:
    - 1 Proto-flight Model (PFM). To qualify at proto-qualification level
    - 1 additional Flight Model (FM). To qualify at acceptance level
- **Calibration** of the Hard X-ray Polarimeter of each CubeSat will be carried out at INAF-IAPS calibration facility (+ possibility for measurements at Synchrotron facilities like ESRF)

High Solar Activity and the May 2024 Events

ROMATODAY

RT

Redazione

11 maggio 2024 07:41

ATTUALITÀ

Aurora boreale a Roma, lo spettacolo del cielo "rosa" e il tam tam sui social

La tempesta solare che sta investendo la terra sta facendo apparire nei cieli di buona parte dell'Europa delle incredibile aurore boreali

Si parla di

aurora boreale

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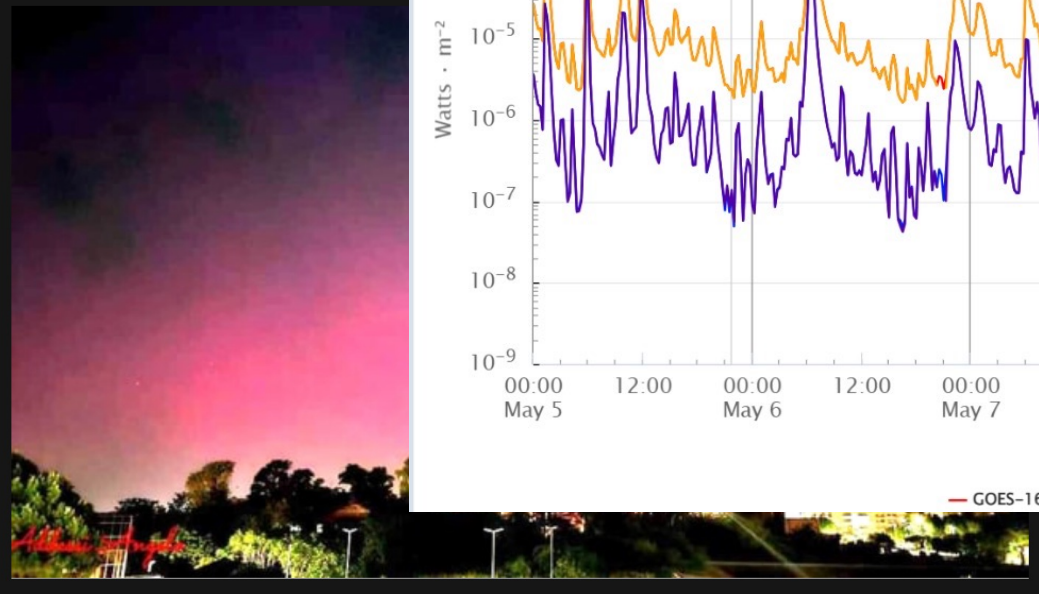

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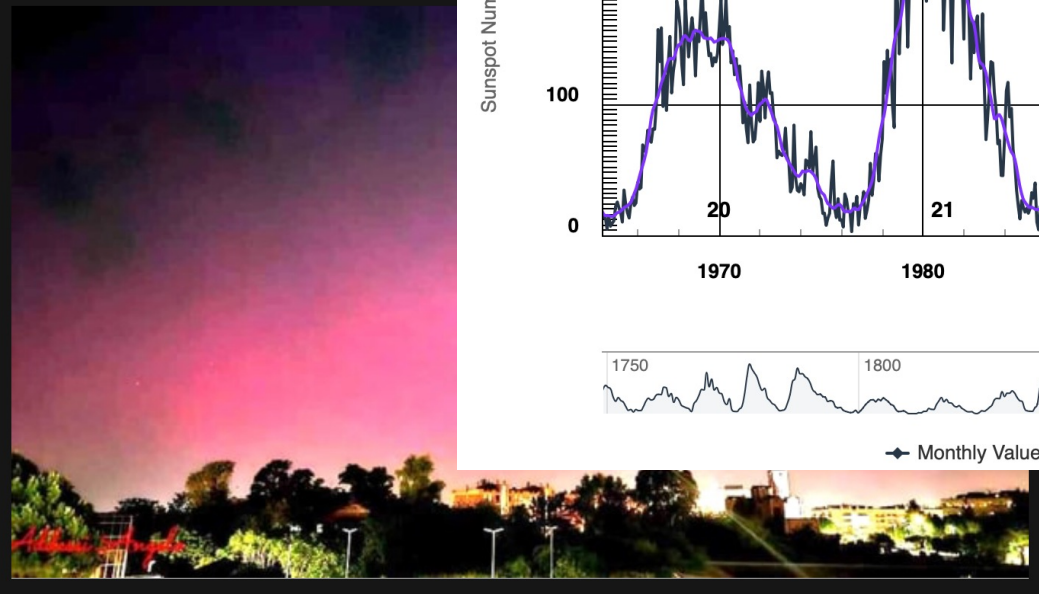

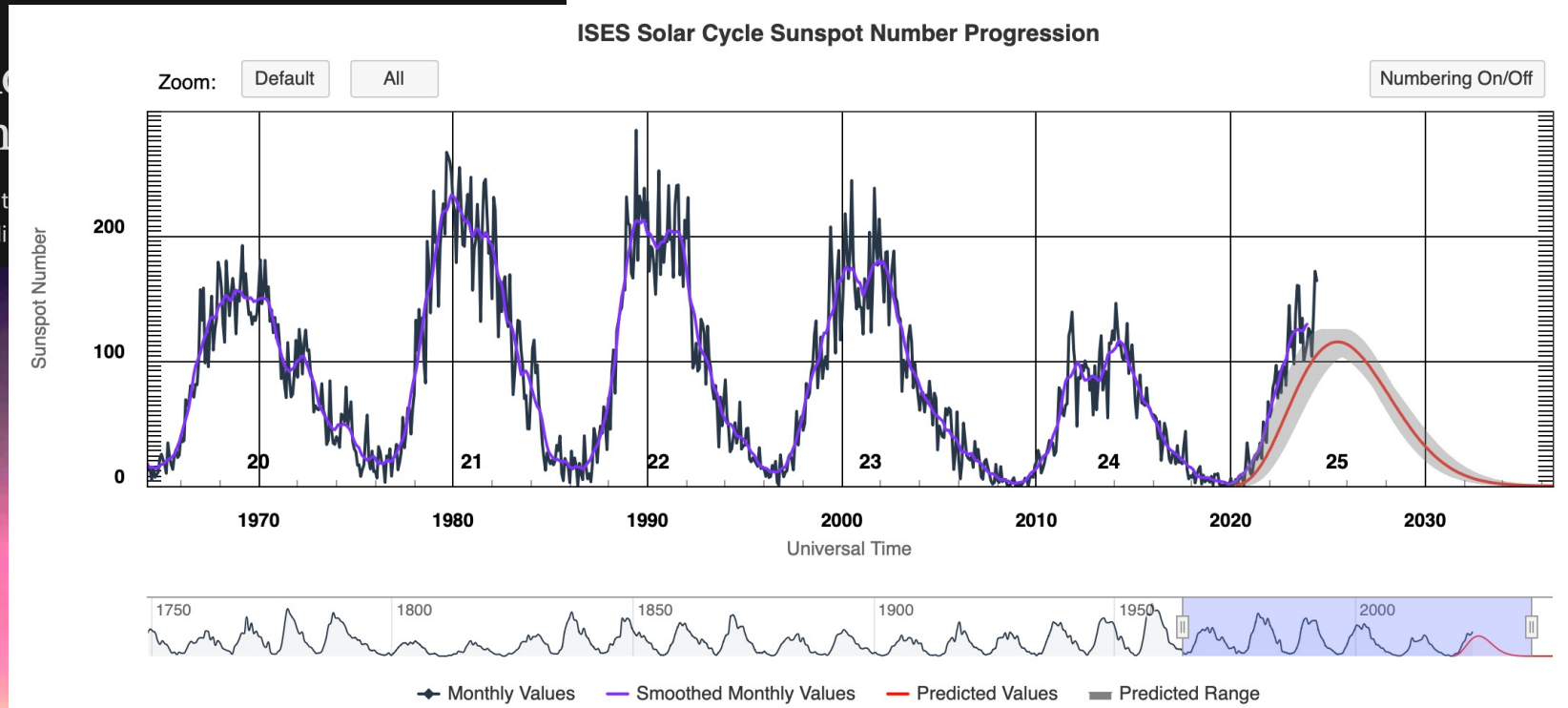

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