



## Master-Thesis

# Path Planning for Dynamic Maneuvers with Micro Aerial Vehicles

Autumn Term 2014

# Declaration of Originality

I hereby declare that the written we	ork I have submitted entitled					
Path Planning for Dynamic Maneuvers with Micro Aerial Vehicles						
is original work which I alone have a	uthored and which is written in my own words. $^{1}$					
A (1 ( )						
Author(s)						
First name	Last name					
Supervising lecturer						
First name	Last name					
citation rules and that I have read etiquette' (http://www.ethz.ch/s	have been informed regarding normal academic and understood the information on 'Citation tudents/exams/plagiarism_s_en.pdf). The scipline in question here have been respected. ted electronically for plagiarism.					
Place and date	Signature					

 $<sup>\</sup>overline{\phantom{a}^{1}\text{Co-authored work: The signatures}}$  of all authors are required. Each signature attests to the originality of the entire piece of written work in its final form.

# Contents

Al	bstra	act	iii
$\mathbf{S}\mathbf{y}$	mbo	ls	iv
1	Intr	roduction	1
	1.1	State of the Art	1
	1.2	Quadratic Programming	1
		1.2.1 Constrained Quadratic Programming	1
		1.2.2 Unconstrained Quadratic Programming	2
2	Pol	ynomial Trajectory Optimization	3
	2.1	Polynomial Trajectory	3
	2.2	Optimization	3
		2.2.1 Cost Function	3
		2.2.2 Polynomial Optimization as a Constrained QP	4
		2.2.3 Polynomial Optimization as a Unconstrained QP	4
		2.2.4 Initial Solution	5
		2.2.5 Time Allocation	7
	2.3	Reduce the Optimization Variables	9
3	RR	Г	11
	3.1	General	11
	3.2	RRT Algorithm	11
		3.2.1 Goal State	11
	3.3	RRT* Algorithm	13
		3.3.1 Rewiring	13
		3.3.2 Bounding Box	14
		3.3.3 Ray Check	15
$\mathbf{A}$	Irge	endwas	17
Bi	bliog	graphy	19

# Abstract

The goal of this master thesis is to develop a numerical robust trajectory planning algorithm for dynaimc multi-copter flights in dense environments. The trajectory generated by this algorithm is represented by polynomials which are jointly optimized. The cost function of the optimization consists of the total trajectory-time as well as the total quadratic snap (second derivation of the acceleration). The inclusion of the snap into the cost function guaranties a trajectory without abrupt or expensive control inputs.

Furthermore, the process of exploring the state space using the Rapidly-Exploring Random Tree (RRT) algorithm is embedded into the numerical robust algorithm. The sampling points oft the RRT (or RRT\*) algorithm are then used as the nodes in the polynomial optimization.

# **Symbols**

## **Symbols**

 $\phi, \theta, \psi$  roll, pitch and yaw angle

### **Indices**

x x axis y y axis

### Terms and Definitions

jerk Derivation of acceleration

snap Derivation of jerk

vertex Fixed sampling point of a polynomial trajectory

### Acronyms and Abbreviations

ETH Eidgenössische Technische Hochschule

UAV Unmanned Aerial Vehicle

RRT Rapidly-Exploring Random Tree

QP Quadratic Programming

## Chapter 1

# Introduction

#### 1.1 State of the Art

A lot of research has been performed in the field of Unmanned Aerial Vehicles (UAV) in the last years leading to a strong improvement in trjectory planning [1] as well as in control ([2], [3]). Another research field is machine learning [4] which is suitable to enhance the performance of aerobatic maneuvers but seams to have a downside regarding motion planning and trajectory generation in dense environments.

Speaking of trajectory planning, there are two different strategies which are pursued. On the one hand, the geometric and the temporal planning are decoupled [5], on the other hand geometric and temporal information are coupled and the trajectory is the result of a minimization problem. For the couplet problem one can make use of the differential flatness of a quadrocopter to derive constraint on the trajectory. A cost function which could be the total trajectory-time [3] or the total snap [6] can be formulated.

Another aspect of planning is exploring the state space in the first place. A strong tool to do so are incremental search techniques as for instance the A\* [7] or the RRT\* algorithm [8]. The sampling points of the solution of the incremental search can then be used as the nodes for the polynomial optimization.

### 1.2 Quadratic Programming

#### 1.2.1 Constrained Quadratic Programming

Quadratic Programming (QP) is a special case of an optimization problem in which a quadratic function f(x) is optimized with respect to its optimization variables (which are represented with the vector x in equation 1.1)

$$f(x) = \frac{1}{2} \cdot x^T Q x + c^T x \tag{1.1}$$

The optimization can be performed under linear constraints on the optimization variables. The linear constraints can be divided in two groups. For one thing there are the inequality constraints

$$Ax \le b \tag{1.2}$$

where the vector b contains the inequality constraints.

For another thing there are the equality constraints

$$Ex = d (1.3)$$

where the vector d contains the equality constraints. In case there are only equality constrains, the solution to the QP is given by the linear system in equation 1.4:

$$\begin{bmatrix} Q & E^T \\ E & 0 \end{bmatrix} \cdot \begin{bmatrix} \mathbf{x} \\ \lambda \end{bmatrix} = \begin{bmatrix} -\mathbf{c} \\ \mathbf{d} \end{bmatrix}$$
 (1.4)

where  $\lambda$  is a set of Lagrange multipliers and c is the linear term of the cost function in equation 1.1.

The constrained QP gets ill-conditioned for a large number of optimization variables which lead to large matrices. The performance of the constraint QP deteriorate even more if the matrices are sparse. This particular case often appears in polynomial optimization for high order polynomials where some polynomial coefficients are close to zero.

To reduce the number of optimization variables, and therefore the size of the matrices, the constrained QP with equality constraints can be converted into a numerical robust unconstrained QP.

#### 1.2.2 Unconstrained Quadratic Programming

For the unconstrained QP the equality constraints  $E\mathbf{x} = \mathbf{d}$  resp.  $\mathbf{x} = E^{-1}\mathbf{d}$  are embedded into the quadratic cost function from equation 1.1 resulting in equation 1.5:

$$f(d) = \frac{1}{2} \cdot d^T E^{-T} Q E^{-1} d + c^T E^{-1} d$$
 (1.5)

Since the vector x is replaced by  $E^{-1}d$  and E is a constant matrix, the new optimization variables are now stored in the vector d.

If the cost function defined in equation 1.1 does not have a linear term, i.e. the vector c is equal to zero, equation 1.5 can be simplified to:

$$f(d) = \frac{1}{2} \cdot d^T E^{-T} Q E^{-1} d \tag{1.6}$$

If we are not interested in the cost itself but only in the optimization variables stored in d, the constant multiplier 1/2 can be dropped:

$$f(d) = d^T E^{-T} Q E^{-1} d (1.7)$$

The theoretical derived equation 1.7 can be compared to to multidimensional cost function in equation 2.5. Equation 2.5 establishes a connection between the numerical advantages of a unconstrained QP and the polynomial coefficients representing a trajectory.

# Chapter 2

# Polynomial Trajectory Optimization

### 2.1 Polynomial Trajectory

Regarding the differentiability of polynomials, they are a profound choice to represent a trajectory. Especially for the use in a differentially flat representation of the UAV dynamics. (Flatness in the proper sense of system theory means that all the states and inputs can be expressed in terms of the flat output and a finite number of its derivative). Furthermore, the differentiability of polynomials enables the possibility to check the derivatives of the trajectory for bounding violations to avoid input saturation. This saturation-check can be performed during trajectory optimization and therefore guarantees the feasibility of the resulting trajectory.

### 2.2 Optimization

The goal of this master thesis is to optimize a trajectory which passes through way-points (also called vertices or nodes) which are defined in advance. This waypoints can be chosen manually or by a path-finding algorithm such as RRT\* which will be discussed in chapter 3. Furthermore, not only the waypoints (i.e. the position) can be fixed in advance but also its derivatives (such as speed, acceleration etc.). The position and its derivatives are then utilized as the equality constrains for a QP (explained in Section 1.2).

#### 2.2.1 Cost Function

Optimization for the purpose of trajectory planning means to minimize a cost function. The cost function in this case is a combination of temporal and geometric cost. The geometric cost penalizes the square of the derivatives of the trajectory. In this master thesis the geometric cost is represented by the squared snap which guarantees a trajectory without abrupt control inputs.

The temporal cost is simply the total trajectory time multiplied by a user chosen factor  $k_T$  which determines the aggressiveness of the resulting trajectory. The impact of  $k_T$  can be seen in equation 2.10 which represents the combined geometric and temporal cost.

To express the geometric cost in a compact way one can utilize the Hessian matrix Q. The Hessian matrix is defined as a squared matrix of second-order partial derivatives which follows from differentiation a function with respect to each of its coefficients,

in this instance the polynomial coefficients. The geometric cost function J(T) for one segment with the duration T can now be written as

$$J(T) = p^T \cdot Q(T) \cdot p \tag{2.1}$$

where Q(T) is the Hessian matrix for a fixed segment time T. p is the vector containing the coefficients of the polynomial trajectory.

If the trajectory consists of more than one segment the Hessian matrix has to be extended to a block-diagonal matrix. The geometric cost function for multiple segments with fixed but individual segment times  $T_i$  can be written as

$$J(T) = \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix}^T \cdot \begin{bmatrix} Q_1(T_1) & & \\ & \ddots & \\ & & Q_n(T_n) \end{bmatrix} \cdot \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix}$$
 (2.2)

#### 2.2.2 Polynomial Optimization as a Constrained QP

The cost function in equation 2.2 has to be minimized under constrains since we want to fix things like start or end position. In a first intuitive approach the constraints on the endpoint derivatives are utilized in a constrained QP. Therefore, a mapping matrix E between endpoint derivatives and polynomial coefficients is needed. The resulting equality constraint for the  $i^{th}$  segment can be written as

$$E_i \cdot p_i = d_i \tag{2.3}$$

where p is the vector containing the polynomial coefficients and d is the vector containing the endpoint derivatives. Regarding the total number of segments of the trajectory, equation 2.3 can be written in matrix form:

$$\begin{bmatrix} E_1 & & & \\ & \ddots & & \\ & & E_n \end{bmatrix} \cdot \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix} = \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix}$$
 (2.4)

The constrained QP is suitable for a small amount of segments but gets ill-conditioned for a large amount of segments and therefore large matrices. Especially if there are matrices which are close to singularity and have coefficients which are close to zero, the constrained QP can get numerical unstable.

#### 2.2.3 Polynomial Optimization as a Unconstrained QP

To avoid the numerical instability of a constrained QP the optimization problem is converted into a unconstrained QP. To achieve this, the polynomial coefficients  $p_i$  from equation 2.2 have to be substituted by the endpoint derivatives  $d_i$  which are now the new optimization variables. The cost function of the unconstrained QP can now be written as

$$J = \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix}^T \cdot \begin{bmatrix} E_1 \\ & \ddots \\ & & E_n \end{bmatrix}^{-T} \cdot \begin{bmatrix} Q_1 \\ & \ddots \\ & & Q_n \end{bmatrix} \cdot \begin{bmatrix} E_1 \\ & \ddots \\ & & E_n \end{bmatrix}^{-1} \cdot \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix}$$
 (2.5)

where  $Q_i$  is the Hessian matrix according to the  $i^{th}$  segment time  $T_i$ .

As mentioned above, the endpoint derivatives are the new optimization variables. Due to the equality constrains some of the endpoint derivatives are already specified consequently reducing the number of optimizations variables. Expediently, the endpoint derivatives are divided in fixed derivatives  $d_f$  and unspecified derivatives  $d_p$  and then reordered using the matrix C which consists of zeros and ones. After reordering the endpoint derivatives equation 2.5 can be rewritten as

$$J = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \underbrace{C^T E^{-T} Q E^{-1} C}_{\mathbf{p}} \begin{bmatrix} d_f \\ d_p \end{bmatrix}$$
 (2.6)

where the product of the reordering matrix C, the mapping matrix E and the Hessian matrix Q can be expressed as a single Matrix R. The matrix R for his part can be divided into four submatrices according to the fixed and unspecified endpoint derivatives which modifies equation 2.6 as follows:

$$J = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \begin{bmatrix} R_{ff} & R_{fp} \\ R_{pf} & R_{pp} \end{bmatrix} \begin{bmatrix} d_f \\ d_p \end{bmatrix}$$
 (2.7)

Partially differenting equation 2.7 with respect to the unspecified derivatives  $d_p$  and equate it to zero yields the optimized/minimized unspecified derivatives  $d_p^*$ 

$$d_p^* = -R_{pp}^{-1} \cdot R_{fp}^T \cdot d_f \tag{2.8}$$

as a function of the fixed derivatives  $d_f$  and two of the submatrixes  $(R_{pp}, R_{fp})$  of R.

#### 2.2.4 Initial Solution

Equation 2.8 can now be used to compute the initial solution. As can be seen in equation 2.5, the Hessian matrix for the  $i^{th}$  segment  $Q_i$  depends on the segment time  $T_i$ . Thus, all the segment times have to be defined in advance. For the initial solution the segment times are calculated based on the Euclidean distance  $d_{norm}$  and on the user specified maximal speed  $(v_{max})$  and maximal acceleration  $(a_{max})$ .

Basically the segment time is determined by the term  $d_{norm}/v_{max} \cdot 2$  which is twice the time the UAV would need for a segment by flying the whole distance at maximal speed. Although this is a good estimation for long segments, for shorter ones the time needed to accelerate gets significant. In order to incorporate acceleration time, a multiplier which is zero for long segments and unequal to zero for short ones is added. The segment time  $T_i$  for the  $i^{th}$  segment can be computed according to

$$T_i = \frac{d_{norm_i}}{v_{max}} \cdot 2 \cdot \left(1 + 6.5 \cdot \frac{v_{max}}{a_{max}} \cdot \frac{1}{e^{\frac{d_{norm_i}}{v_{max}} \cdot 2}}\right)$$
(2.9)

where  $d_{norm_i}$  is the Euclidean distance of the  $i^{th}$  segment,  $v_{max}$  the user specified maximal velocity and  $a_{max}$  the user specified maximal acceleration. The fraction  $v_{max}/a_{max}$  gives an idea how much time is needed to accelerate to maximum velocity whereas 6.5 is a empirical weighting factor.

The result from equation 2.9 is depicted in figure 2.1 whereat the x-axis represents the Euclidean distance  $d_{norm}$  and the y-axis represents the segment time T. For this plot the user specified limitation on speed has been set to  $v_{max} = 3\frac{m}{s}$  and the limitation on acceleration has been set to  $a_{max} = 5\frac{m}{s^2}$ . The dashed green line represents the term  $d_{norm}/v_{max} \cdot 2$  and serves as a reference. The blue graph is the exact representation of equation 2.9.

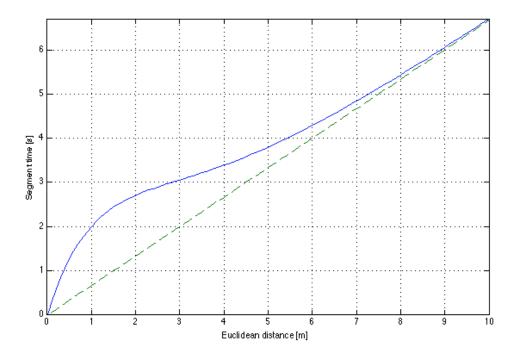


Figure 2.1: The segment time T depends on the Euclidean distance  $d_{norm}$  of a segment as well as on the max. velocity  $v_{max}$  and the max. acceleration  $a_{max}$ . For short segments the acceleration time is incorporated.

Once the segment times are calculated, the initial snap minimized solution can be computed according to equation 2.8. The initial solution for a 3 dimensional problem with 4 segments is depict in figure 2.2. The start and goal of the trajectory is the origin of the Cartesian coordinate system (0/0/0). For both, start and goal state, the velocity, the acceleration, the jerk and the snap are fixed and set to zero. For all the other sampling points (vertices) the derivatives are unspecified. The Cartesian coordinates of the sampling points are chosen manually and are listed in the following table:

Waypoint	x-coordinate	y-coordinate	z-coordinate
Start-Vertex	0	0	0
Vertex 1	5	1	-4
Vertex 2	3	-2	1
Vertex 3	-1	2	3
Goal-Vertex	0	0	0

Figure 2.2 a) shows the position (i.e. the Cartesian coordinates) whereby each of the 3 dimensions is depicted as a single graph. The Cartesian coordinates from the above-noted table are depicted as circles. Plot b) shows the velocity of the individual direction as a solid graph. Additional, the velocity in the three-dimensional space (i.e. the Euclidean norm of the velocity vector) is depicted as a dashed graph. Plot c) is constructed similarly but depicts the acceleration in the individual directions (solid) and the acceleration in the three-dimensional space (dashed). Furthermore the limitation  $(v_{max} = 3\frac{m}{s})$  and  $a_{max} = 2\frac{m}{s^2}$  for this problem) are depicted.

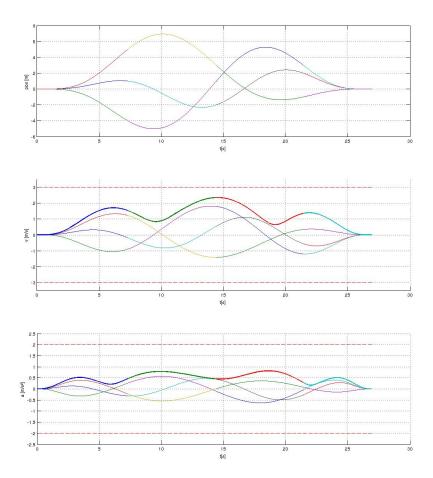


Figure 2.2: Initial solution of a trajectory with 4 segments: Plot a) shows the position (i.e. the Cartesian coordinates). Plot b) shows the velocity and plot c) the acceleration. A dashed graph represents the velocity respectively the acceleration in the three-dimensional space.

The total time of the initial trajectory is the sum of the individual segment times and therefore defined by equation 2.9. Since equation 2.9 does not incorporate the circumstances from one segment to an other it is likely to find a better trajectory (i.e. a trajectory with a smaller total cost) with the same total time but a different ratio between the segment time. Moreover there is no possibility to adjust the aggressiveness of the initial solution since the segment times are calculated up front. Recapitulating, the modification of the individual segment times and therefore the modification of the ratio of the segment times can lead to a better solution. Supplementary, the modification of the individual segment times gives us the opportunity to adjust the aggressiveness of a trajectory.

#### 2.2.5 Time Allocation

So fare, only the geometric cost (i. e. the squared snap) was included in the cost function defined in equation 2.7. Minimization of the geometric cost ensures a

smooth trajectory without abrupt input signal but has no effect on the aggressiveness of a trajectory. Therefore equation 2.7 has to be extended by the temporal cost (i.e. the sum of the segment times) which results in the total cost  $J_{total}$ 

$$J_{total} = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \begin{bmatrix} R_{ff} & R_{fp} \\ R_{pf} & R_{pp} \end{bmatrix} \begin{bmatrix} d_f \\ d_p \end{bmatrix} + k_T \cdot \sum_{i=1}^N T_i$$
 (2.10)

where  $k_T$  is a user specified penalty on time and  $T_i$  is the segment time of the  $i^{th}$  segment.

Since a hight value for  $k_T$  leads to a trajectory with a brief total trajectory time and a small value for  $k_T$  leads to a trajectory with a long total trajectory time, the user specified penalty on time  $k_T$  enables the adjustment of the aggressiveness of a trajectory.

The geometric cost function in equation 2.7 has only the unspecified endpoint derivatives  $d_p$  as optimization variables. This optimization problem can be solved analytically as performed in equation 2.8. The cost function in equation 2.10 has the segment times  $T_i$  as additional optimization variables. Meaning the segment times  $T_i$  are directly represented in the cost function and not only indirectly via the Hessian matrices  $Q_{T_i}$ . Since the segment times are now optimization variables, the Hessian matrices  $Q_{T_i}$  are no longer defined in advance and the problem cannot be solved analytically. Due to that a nonlinear solver is used. In this master thesis NLopt, a open-source library for nonlinear optimization, is applied. The unspecified endpoint derivatives  $d_p$  and the segment times  $T_i$  of the initial solution are the initial values for the nonlinear solver. Hence, the computational cost of the nonlinear optimization is highly depending on the quality of the initial solution.

The termination condition of the optimization can be specified by the optimization variables as well as by the total cost. Generally, the termination conditions are formulated relative to the current value(s). For instance, if the relative termination condition for the total cost  $J_{rel}$  is set to 0.01 the optimization ends if the total cost changes less than a percent during an iteration. The relative termination condition for the optimization variables  $x_{rel}$  is only fulfilled if all of the optimization variables change less then the threshold. Additionally, an absolute termination condition  $x_{abs}$  is applied to the optimization variables but is only called into action if one or several optimization variables are close to zero and the relative criteria therefore don't work properly. Two additional options are the limitation of the number of optimization iterations and the limitation of the duration of the optimization. During the optimization the constraints on velocity and acceleration are checked every iteration.

The result of the nonlinear optimization is depicted in figure 2.3. As can be seen in plot a) the trajectory passes the same waypoints as in Figure 2.2, i.e. the waypoints which are listed in the table in section 2.2.4. The optimized trajectory only needs 18 seconds where as the initial solution required 27 second from start to goal. Figure 2.3 b) shows the velocity of the individual direction as a solid graph. Additional, the velocity in the three-dimensional space (i.e. the Euclidean norm of the velocity vector) is depicted as a dashed graph. The dashed graph tangents the maximal velocity ( $v_{max} = 3m/s$  in this example) partially. The shorter trajectory time and the exploitation of the limits illustrates that the optimized trajectory is more aggressive then the initial trajectory,

The same effect of exploiting the limits can be seen in plot c). Similar to plot b), plot c) shows the acceleration of the individual direction as a solid graph and the acceleration in the three-dimensional space as a dashed graph.

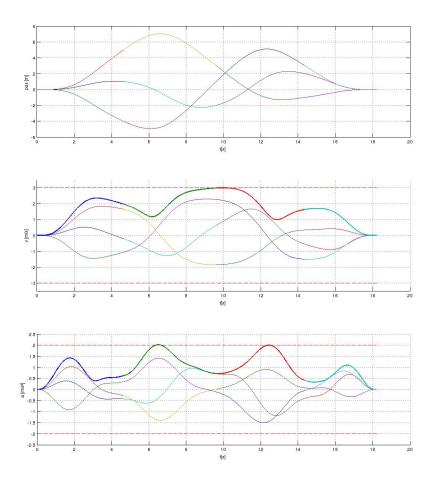


Figure 2.3: Optimized solution of a trajectory with 4 segments: Plot a) shows the position (i.e. the Cartesian coordinates). Plot b) shows the velocity and plot c) the acceleration. A dashed graph represents the velocity respectively the acceleration in the three-dimensional space.

## 2.3 Reduce the Optimization Variables

TODO

## Chapter 3

# RRT

#### 3.1 General

The goal of this thesis was not only to generate a numerically stable, snap optimized polynomial trajectory but also to explore a densely packed (indoor) environment and plan an aggressive trajectory in between the obstacles. Hence, the Rapidly-Exploring Random Tree (RRT) algorithm is used to find a collision-free straight line solution through densely packed environments. The sampling points off the RRT (or RRT\*) algorithm are then used as the vertices in the polynomial optimization.

### 3.2 RRT Algorithm

RRT is a computational efficient algorithm to find a path in a high dimensional space by randomly building a space-filling tree. The sampling points are drawn randomly from the sample space and the tree grows incrementally. For each new sample the algorithm attempts to build a collision-free connection to the nearest state in the tree. If a collision-free connection is possible the sample and the connection are added to the tree.

An iteration of the RRT algorithm can be depicted schematically:

- 1. Generate a random sample
- 2. Find nearest state in tree
- 3. Try to build a collision-free connection to the nearest state
- 4. If feasible, add the sampled state and the connection to the tree

#### 3.2.1 Goal State

As mentioned above, the RRT algorithm is based on random samples. Therefore it is very unlikely that a sampled state perfectly matches the desired goal state.

There are two different strategies to enable the RRT algorithm to get to the goal. One strategy is to define not only a goal state but a goal region. Every random sample which is located within the goal region is considered a goal state. As soon as a collision-free connection to a sample in the goal region is established, this trajectory is stored as the best trajectory. At this point the algorithm can be stopped or further iteration can be performed to find a better trajectory to the goal

region. Once an other state from the goal region is sampled and the cost of the path to the new state is lower then the cost of the best trajectory, the best trajectory is replaced by the new path.

Another strategy is to steer the RRT algorithm directly to the goal state. In addition to the randomly sampled states the exact goal state is added to the algorithm. The schematic description of an iteration of the RRT algorithm listed in section 3.2 can be modified to represent an iteration with the goal state:

- 1. Insert goal state
- 2. Find nearest state in tree
- 3. Try to build a collision-free connection to the nearest state
- 4. If feasible, add the sampled state and the connection to the tree

In all the cases where a direct collision-free connection between start and goal state is not possible the iteration with the goal state will not succeed in a first attempt. Hence the iterations with the randomly sampled states described in section 3.2 are needed to build the space filling tree.

Figure 3.1 depicts the straight line solution of the RRT algorithm with a fixed goal state. The figure is in bird's eye perspective and shows a crossing of different hallways. The blue cells represents the floor and the green cells represents the walls. The map was generated with a stereo camera and was not reworked. Therefore some of the cells of the floor which should be occupied/blue are left free.

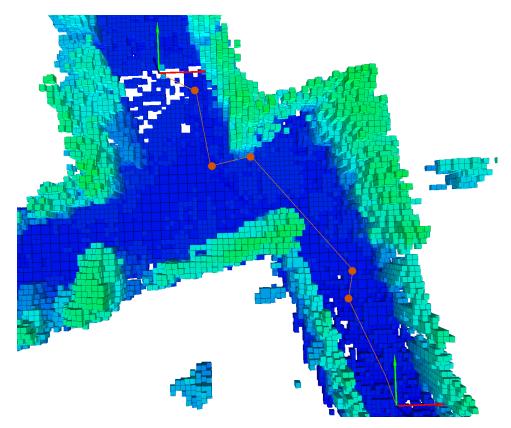


Figure 3.1: Straight line solution of the RRT algorithm with a fixed goal state. Start and goal state are each represented by a red x-vector and a green y-vector. The random sampled states are depicted as orange dots.

### 3.3 RRT\* Algorithm

In contrast to the RRT algorithm the RRT\* (or RRT Star) algorithm not only tries to connect to the nearest state in the tree but to several states near the sampled state. The user can define a threshold (on the distance) which defines which states of the tree belongs to the "near states". If there is no state within the user specified range the algorithm attempts to build a collision-free connection to the nearest state in the tree just as the RRT algorithm.

As a first step, the sampled state is connected to the best state among the near states whereas best means minimum cost/distance. Once the sampled state is added to the tree all the other states among the set of near states are connected to the sampled state. If the connection is collision-free and the cost of the total path is smaller than the cost of the existing path, the old path is replaced.

An iteration of the RRT\* algorithm can be depicted schematically:

- 1. Generate a random sample
- 2. A threshold defines a set of near states
- 3. Try to build a collision-free connection to best state among the near states
- 4. Add the sampled state and the connection to the tree
- 5. Try to connect all the other states from the set with the sampled state
- 6. Replace the old path if the new one has a smaller cost
- 7. If there is no near state within the threshold apply the RRT algorithm

Because the RRT\* algorithm tries to connect to several states each iteration, the procedure to find a path takes longer and is computationally more expensive. However, solutions with lower cost can be found which is more important for most real life applications.

#### 3.3.1 Rewiring

The sequence of step 5 and step 6 of the RRT\* algorithm ("Try to connect all the other states from the set with the sampled state", "Replace the old path if the new one has a smaller cost") are called "rewiring".

The threshold which defines the set of near states depends on a user specified parameter  $\gamma$ . In this case, the threshold is the radius of a sphere, i.e. all the near states are located within this sphere. The radius r can be calculated according to

$$r = \gamma * \left(\frac{\ln(n+1)}{n+1}\right)^{1/d} \tag{3.1}$$

where n is the number of states which are already in the tree. The dimension of the state space d is a fixed parameter and ln represents the natural logarithm.

As mention in section 3.3 the RRT\* algorithm is computationally more expensive but returns trajectory with lower cost. Both aspects are caused by the rewiring and are therefore strongly influenced by the parameter  $\gamma$ . A large  $\gamma$  defines a large sphere, hence it is likely to have more states (which are already in the tree) to be located within the sphere. The rewiring of the near states tends to result in shorter trajectories with fewer segments.

Chapter 3. RRT

Figure 3.2 depicts the straight line solution of the RRT\* algorithm with a fixed goal state. Compared to figure 3.1 a superior trajectory is found since a rewiring of the states in the tree was performed.

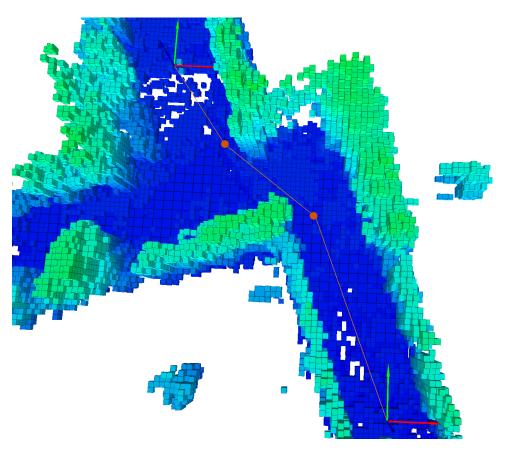


Figure 3.2: Straight line solution of the RRT\* algorithm with a fixed goal state. Start and goal state are each represented by a red x-vector and a green y-vector. The random sampled states are depicted as orange dots. The  $\gamma$  parameter in this example was set to  $\gamma=1.5$ .

#### 3.3.2 Bounding Box

The straight line solution in figure 3.2 is collision-free but passes by very close to the walls. In real life application not only a point mass but a object (in this master thesis a UAV) should follow the trajectory. Therefore a bounding box is installed around the trajectory.

The bounding box is implemented as a cuboid. The 3 dimension of the cuboid can be defined individually. The trajectory is then divided into a discrete trajectory and the bounding box is installed around the discrete points. If there is a obstacle in one of the bounding boxes the hole straight line is considered as in collision.

Figure 3.3 depicts the straight line solution of the RRT\* algorithm with a bounding box. In contrast to figure 3.2 the trajectory is now located more central in the hallway because the bounding box makes it impossible to pass by the wall very close. Because the trajectory proceeds less direct from start to goal state the total distance of the trajectory increases. Although the number of segments increases as well from 3 segments in figure 3.2 to 4 segments in figure 3.3 this is not necessarily

a consequence of the bounding box. Performing additional iterations (including more sampling points and more rewiring) could lead to a solution with 3 segments.

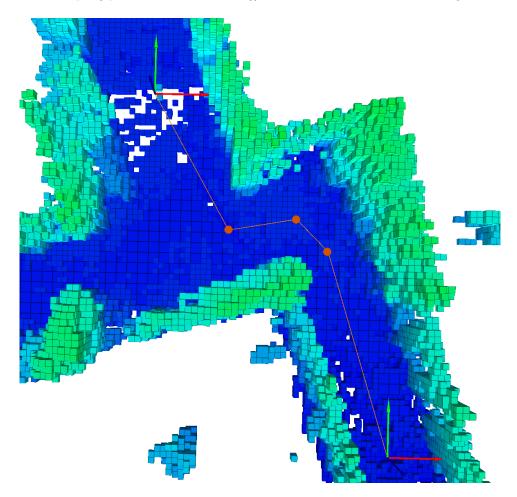


Figure 3.3: Straight line solution of the RRT\* algorithm with bounding box. Due to the bounding box the trajectory is located more central in the hallway because the bounding box makes it impossible to pass by the wall very close. The  $\gamma$  parameter in this example was set to  $\gamma=1.5$ .

### 3.3.3 Ray Check

TODO: verbindung zu Octompa matlab figure

Chapter 3. RRT

# Appendix A

# Irgendwas

Bla bla ...

# **Bibliography**

- [1] R. HE, A. BACHRACH, M. ACHTELIK, A. GERAMIFARD, D. GURDAN, S. PRENTICE, J. STUMPF, AND N. ROY: On the design and use of a micro air vehicle to track and avoid adversaries. The Int. Journal of Robotics Research, vol. 29, pp. 529-546, 2010.
- [2] D. Colling, O. A. Yakimenko, J. F. Whidborne, and A. K. Cooke: *A prototype of an autonomous controller for a quadrotor UAV*. In Proceedings of the European Control Conference, Kos, Greece, 2007, pp. 1-8.
- [3] M. Hehn and R. D'Andrea: Quadrocopter trajectory generation and control. In International Federation of Automatic Control (IFAC), World Congress 2011, 2011.
- [4] S. Lupashin, A. Schollig, M. Sherback, and R. D'Andrea: A simple learning strategy for high-speed quadrocopter multi-flips. In Proc. of the IEEE Int. Conf. on Robotics and Automation, Anchorage, AK, May 2010, pp. 1642-1648.
- [5] Y. BOUKTIR, M. HADDAD, AND T. CHETTIBI: Trajectory Planning for a Quadrotor Helicopter. In Mediterranean Conference on Control and Automation, Jun. 2008, pp. 1258-1263
- [6] D. MELLINGER AND V. KUMAR: Minimum snap trajectory generation and control for quadrotors. In International Conference on Robotics and Automation, 2011, pp. 2520-2525.
- [7] M. LIKHACHEV, G. GORDON AND S. THRUN:  $ARA^*$ : Anytime  $A^*$  with Provable Bounds on Sub-Optimality . Advances in Neural Information Processing Systems, vol. 16, 2003
- [8] C. RICHTER, A. BRY, AND N, ROY: Polynomial Trajectory Planning for Quadrotor Flight. In International Conference on Robotics and Automation, 2013.