

Master-Thesis

# Path Planning for Dynamic Maneuvers with Micro Aerial Vehicles

Autumn Term 2014



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# Abstract

The goal of this Master-Thesis is to develop a numerical robust trajectory-planning algorithm for aggressive multi-copter flights in dense environments. The trajectory generated by this algorithm is represented by polynomials which are jointly optimized. The cost function of the optimization consists of the total trajectory-time as well as the total quadratic snap (second derivation of the acceleration). Including the snap into the cost function guaranties a trajectory without abrupt or expensive control inputs.

Furthermore the process of exploring the state space using the Rapidly-Exploring Random Tree (RRT) algorithm is embedded into the numerical robust algorithm. The sampling points of the RRT (or RRT\*) algorithm are then used as the vertices in the polynomial optimization.

# Symbols

## Symbols

$\phi, \theta, \psi$                       roll, pitch and yaw angle

## Indices

$x$                       x axis  
 $y$                       y axis

## Acronyms and Abbreviations

ETH              Eidgenössische Technische Hochschule  
UAV              Unmanned Aerial Vehicle  
RRT              Rapidly-Exploring Random Tree  
QP                Quadratic Programming

# Chapter 1

## Introduction

### 1.1 State of the Art

A lot of research has been done in the field of Unmanned Aerial Vehicles (UAV) in the last years leading to a strong improvement in planning [1] as well as in control [[2], [3]]. Another research field is machine learning [4] which is suitable to enhance the performance of aerobatic maneuvers but seems to have a downside regarding motion planning and trajectory generation in dense environments.

Speaking of trajectory planning, there are two different strategies which are pursued. On the one hand, the geometric and the temporal planning are decoupled [5] on the other hand, geometric and temporal information are coupled and the trajectory is the result of a minimization problem. For the couplet problem one can make use of the differential flatness of a quadcopter to derive constraint on the trajectory. Then formulate a cost-function which could be the trajectory-time [3] or the total snap [6] (second derivation of acceleration).

Another aspect of planning is exploring the state space in the first place. A strong tool to do so are incremental search techniques as for instance the A\* [7] or the RRT\* algorithm [8]. The sampling points of the solution of the incremental search can then be used as the vertices for the polynomial optimization.

### 1.2 Quadratic Programming

#### 1.2.1 Constrained Quadratic Programming

Quadratic Programming (QP) is a special case of optimization problem in which a quadratic function is optimized with respect to its optimizations variables (which are represented with the vector  $x$  in Equation 1.1)

$$f(x) = \frac{1}{2} \cdot x^T Q x + c^T x \quad (1.1)$$

The optimization is performed under linear constraints on the optimizations variables. Whereas a distinction between equality ( $E\mathbf{x} = \mathbf{d}$ ) and inequality constraints ( $A\mathbf{x} \leq \mathbf{b}$ ) has to be made.

In case there are only equality constraints, the solution to the QP is given by the linear system in Equation 1.2 :

$$\begin{bmatrix} Q & E^T \\ E & 0 \end{bmatrix} \cdot \begin{bmatrix} \mathbf{x} \\ \lambda \end{bmatrix} = \begin{bmatrix} -\mathbf{c} \\ \mathbf{d} \end{bmatrix} \quad (1.2)$$

### 1.2.2 Unconstrained Quadratic Programming

The constrained QP gets ill-conditioned for a large amount of segments or for high order polynomials (which both lead to large and sometimes sparse matrices). To reduce the number of optimization variables, and therefore the size of the matrices, the constrained QP can be converted into a unconstrained QP.

In other words, the polynomial coefficients are no longer the optimization variables but the free endpoint derivatives are optimized. The exact formulas of the unconstrained QP can be seen in Section 2.2.3. Whereby the linear term  $x^T z$  from the quadratic cost-function in Equation 1.1 is equal to zero for the polynomial optimization and therefore absent in Equation 2.5.



## Chapter 2

# Polynomial Trajectory Optimization

### 2.1 Polynomial Trajectory

Regarding the differentiability of polynomials, they are a profound choice to represent a trajectory. Especially for the use in a differentially flat representation of the UAV dynamics. (Flatness in the proper sense of system theory means that all the states and inputs can be expressed in terms of the flat output and a finite number of its derivative).

Furthermore, the differentiability of polynomials enables the possibility to check the derivatives of the trajectory for bounding violations to avoid input saturation. This saturation-check can be performed during trajectory optimization and therefore guarantees the feasibility of the resulting trajectory.

### 2.2 Optimization

The goal is to optimize a trajectory which passes through way-points (also called vertices) which are defined in advance. These way-points can be chosen manually or by a path-finding algorithm such as RRT\* which will be discussed in Chapter 3. Furthermore, not only the way-points (therefore the position) can be fixed in advance but also its derivatives (such as speed, acceleration etc.). The position and its derivatives are then utilized as the equality constraints for a QP (explained in Section 1.2).

#### 2.2.1 Cost Function

Optimization in terms of trajectory planning means minimization of a cost function. The cost function in this case is a combination of temporal and geometric cost. The geometric cost penalizes the (square) of the derivatives of the trajectory. In this Master Thesis the geometric cost is represented by the squared snap which guarantees a trajectory without abrupt control inputs.

The temporal cost is simply the total trajectory-time multiplied by a user chosen factor  $k_T$  which determines the aggressiveness of the resulting trajectory.

To express the geometric cost in a compact way one can make use of the Hessian matrix  $Q$ . The Hessian matrix is defined as a squared matrix of second-order partial derivatives which follows from differentiation a function with respect to each of

its coefficients (in this instance the polynomial coefficients). The geometric cost function  $J(T)$  for a fixed time for one segment can now be written as

$$J(T) = p^T \cdot Q(T) \cdot p \quad (2.1)$$

where  $Q(T)$  is the Hessian matrix for a fixed segment-time  $T$  and  $p$  is the vector containing the coefficients of the polynomial.

If the trajectory consists of more than one segment the Hessian matrix has to be extended to a block-diagonal matrix and the geometric cost function for multiple segments with fixed but individual segment-times can be written as

$$J = \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix}^T \cdot \begin{bmatrix} Q_1(T_1) & & \\ & \ddots & \\ & & Q_n(T_n) \end{bmatrix} \cdot \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix} \quad (2.2)$$

### 2.2.2 Polynomial Optimization as a Constrained QP

In a first, intuitive approach the equality constraints on the endpoint derivatives (mentioned in Section 2.2) are utilized in a constrained QP. Therefore a mapping matrix  $E$  between endpoint derivatives and polynomial coefficients is needed. The resulting formula for the  $i^{th}$  segment can be written as

$$E_i \cdot p_i = d_i \quad (2.3)$$

where  $p$  is the vector containing the polynomial coefficients and  $d$  is the vector containing the endpoint derivatives. Regarding the total number of segments of the trajectory, Formula 2.3 can be written in matrix form:

$$\begin{bmatrix} E_1 & & \\ & \ddots & \\ & & E_n \end{bmatrix} \cdot \begin{bmatrix} p_1 \\ \vdots \\ p_n \end{bmatrix} = \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix} \quad (2.4)$$

The constrained QP is suitable for a small amount of segments but gets ill-conditioned for a large amount of segments and therefore large matrices. Especially if there are matrices which are close to singularity and have coefficients which are close to zero, the constrained QP can get numerical unstable.

### 2.2.3 Polynomial Optimization as a Unconstrained QP

To avoid the numerical instability of a constrained QP the optimization problem is converted into a unconstrained QP. Therefore the polynomial coefficients  $p_i$  from Formula 2.2 have to be substituted by the endpoint derivatives  $d_i$  which are now the new optimization variables. The cost function of the unconstrained QP can now be written as

$$J = \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix}^T \cdot \begin{bmatrix} E_1 & & \\ & \ddots & \\ & & E_n \end{bmatrix}^{-T} \cdot \begin{bmatrix} Q_1 & & \\ & \ddots & \\ & & Q_n \end{bmatrix} \cdot \begin{bmatrix} E_1 & & \\ & \ddots & \\ & & E_n \end{bmatrix}^{-1} \cdot \begin{bmatrix} d_1 \\ \vdots \\ d_n \end{bmatrix} \quad (2.5)$$

where  $Q_i$  is the Hessian matrix according to the  $i^{th}$  segment-time.

As mentioned above, the endpoint derivatives are the new optimization variables. Do to the equality constraints some of the endpoint derivatives are already specified consequently reducing the number of optimization variables. Expediently, the endpoint derivatives are divided in fixed derivatives  $d_f$  and unspecified derivatives  $d_p$  and then reordered using the matrix  $C$  which consists of zeros and ones. After reordering the endpoint derivatives Formula 2.5 can be rewritten as

$$J = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \underbrace{C^T E^{-T} Q E^{-1} C}_R \begin{bmatrix} d_f \\ d_p \end{bmatrix} \quad (2.6)$$

where the product of the reordering matrix  $C$ , the mapping matrix  $E$  and the Hessian matrix  $Q$  can be expressed as a single Matrix  $R$ . The matrix  $R$  for his part can be divided into four submatrices according to the fixed and unspecified endpoint derivatives which modifies Formula 2.6 as follows:

$$J = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \begin{bmatrix} R_{ff} & R_{fp} \\ R_{pf} & R_{pp} \end{bmatrix} \begin{bmatrix} d_f \\ d_p \end{bmatrix} \quad (2.7)$$

Partially differentiating Formula 2.7 with respect to the unspecified derivatives  $d_p$  and equate it to zero yields the optimized/minimized unspecified derivatives  $d_p^*$

$$d_p^* = -R_{pp}^{-1} \cdot R_{fp}^T \cdot d_f \quad (2.8)$$

as a function of the fixed derivatives  $d_f$  and two of the submatrixes ( $R_{pp}, R_{fp}$ ) of  $R$ .

### 2.2.4 Initial Solution

Equation 2.8 can now be used to compute the initial solution. As can be seen in Equation 2.5 the Hessian matrix for the  $i^{th}$  segment  $Q_i$  depends on the segment-time  $i$ . Thus, all the segment-times has to be defined. For the initial solution the segment-times are estimated based on the 2-norm distance  $d_{norm}$  and on the user specified maximal speed ( $v_{max}$ ) and maximal acceleration ( $a_{max}$ ).

Basically the segment-time is determined by the term  $d_{norm}/v_{max} \cdot 2$  which is twice the time the UAV would need for a segment by flying at maximal speed the whole distance. Although this is a good estimation for long segments, for shorter ones the time needed to accelerate gets significant. Therefore a multiplier, which is zero for long segments and unequal to zero for short ones, is added. The segment-time  $t_i$  for the  $i^{th}$  segment can be computed according to

$$t_i = \frac{d_{norm_i}}{v_{max}} \cdot 2 \cdot \left( 1 + 6.5 \cdot \frac{v_{max}}{a_{max}} \cdot \frac{1}{e^{\frac{d_{norm_i}}{v_{max}} \cdot 2}} \right) \quad (2.9)$$

where  $d_{norm_i}$  is the 2-norm distance of the  $i^{th}$  segment,  $v_{max}$  the user specified maximal speed and  $a_{max}$  the user specified maximal acceleration. The fraction  $v_{max}/a_{max}$  gives an idea how much time is needed to accelerate to max. velocity whereas 6.5 is a empirical correction factor.

The result from Equation 2.9 is depicted in Figure 2.1 whereat the  $x$ -axis represents the 2-norm distance  $d_{norm}$  and the  $y$ -axis represents the segment time  $t$ . For this plot the user specified limitation on speed and acceleration has been set to  $v_{max} = 3 \frac{m}{s}$  and  $a_{max} = 5 \frac{m}{s^2}$ . The green line represent the term  $d_{norm}/v_{max} \cdot 2$ , the blue graph takes the time needed for acceleration into account and is therefore the exact representation of Equation 2.9.

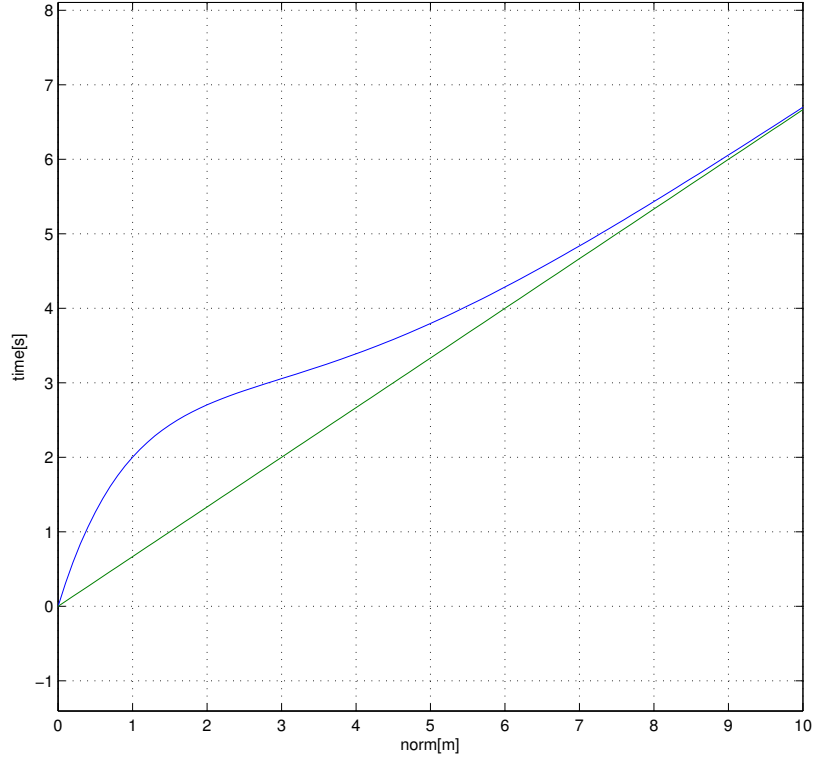


Figure 2.1: The segment-time  $t$  depends on the 2-norm distance  $d_{norm}$  of a segment and on the maximal speed  $v_{max}$  (green line). The blue graph takes the time needed for acceleration into account and is therefore a more elaborate approach to estimate the segment-time.

Once the segment-times are estimated the initial, snap minimized solution can be computed according to Equation 2.8. The initial solution for a 3 dimensional problem with 4 segments is depict in Figure 2.2. The first of this 3 subplots shows the position, the second the velocity and the third the acceleration. The  $x$ -axis for all the 3 subplots is the time. The 3 graphs in the first subplot represent the 3 dimension where the colors are different for each segment. In the second and third subplot, there are also 3 graphs representing a dimension but also a fourth, thicker graph which represents the 2-norm of the velocity respectively the acceleration. Furthermore the limitation ( $v_{max} = 3 \frac{m}{s}$  and  $a_{max} = 2 \frac{m}{s^2}$  for this problem) are depicted.

### 2.2.5 Penalty on Time

So fare, only the geometric cost (i. e. the squared snap) was discussed. Minimization of the geometric cost ensures a smooth trajectory without abrupt input signal but has no effect on the aggressiveness of a trajectory. Therefore Formula 2.7 has to be extended by the temporal cost which results in the total cost  $J_{total}$ :

$$J_{total} = \begin{bmatrix} d_f \\ d_p \end{bmatrix}^T \begin{bmatrix} R_{ff} & R_{fp} \\ R_{pf} & R_{pp} \end{bmatrix} \begin{bmatrix} d_f \\ d_p \end{bmatrix} + k_T \cdot \sum_{i=1}^N T_i \quad (2.10)$$

where  $k_T$  is a user specified penalty on time and  $T_i$  is the segment-time of the  $i^{th}$  segment.

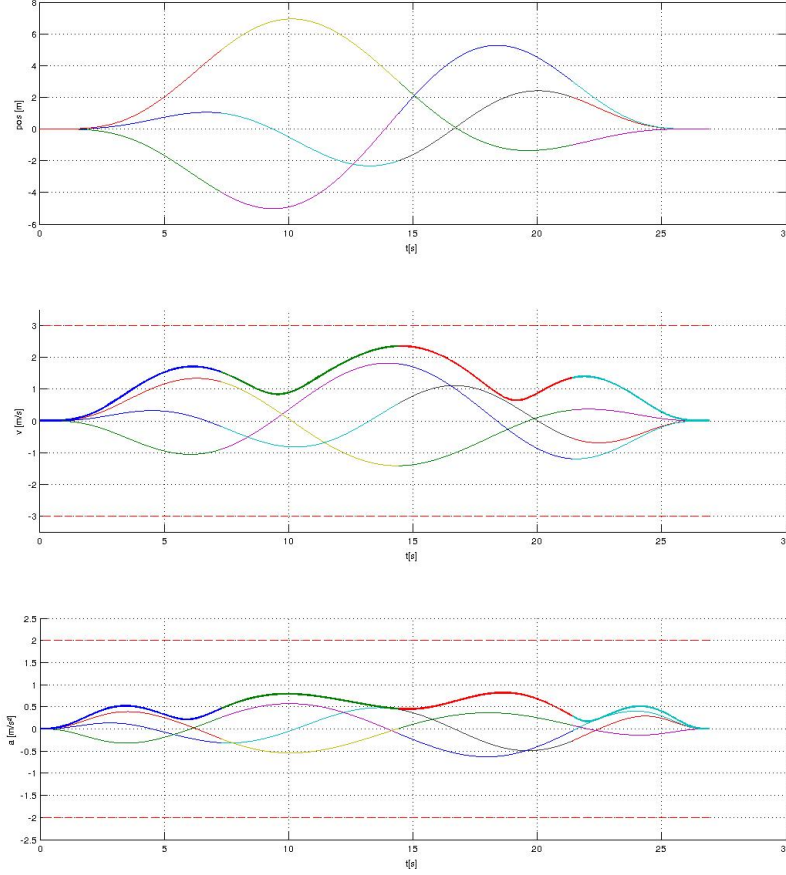


Figure 2.2: Initial solution: The first subplot shows the position, the second the velocity and the third the acceleration of the initial solution. A thicker graph represents the 2-norm of the velocity respectively the acceleration.

In contrast to Equation 2.7, Equation 2.10 cannot be solved analytically. Due to that a nonlinear solver is used. In this thesis NLOpt, a open-source library for nonlinear optimization, is applied. The optimizations variables of the nonlinear optimization are the estimated segment-times and the unspecified endpoint derivatives  $dp$ . The computational cost of the nonlinear optimization is highly depending on the quality of the initial solution.

The termination condition of the optimization can be specified on the optimization variables as well as on the total cost. Generally, the termination conditions are formulated relative to the current value(s). For instance, if the relative termination condition for the total cost  $J_{rel}$  is set to 0.01 the optimization ends if the total cost changes less than a percent during a iteration. The relative termination condition for the optimization variables  $x_{rel}$  is only fulfilled if all of the optimization variables change less then the threshold. Additionally, a absolute termination condition  $x_{abs}$  is applied to the optimization variables but is only called into action if one or several

optimization variables are close to zero and the relative criteria therefore don't work properly. During the optimization the constraints on velocity and acceleration are checked every iteration.

The result of the nonlinear optimization is depicted in Figure 2.3. As can be seen, the trajectory passing the same way-points as in Figure 2.2 only needs around 18 seconds where as the initial solution required around 27 second.

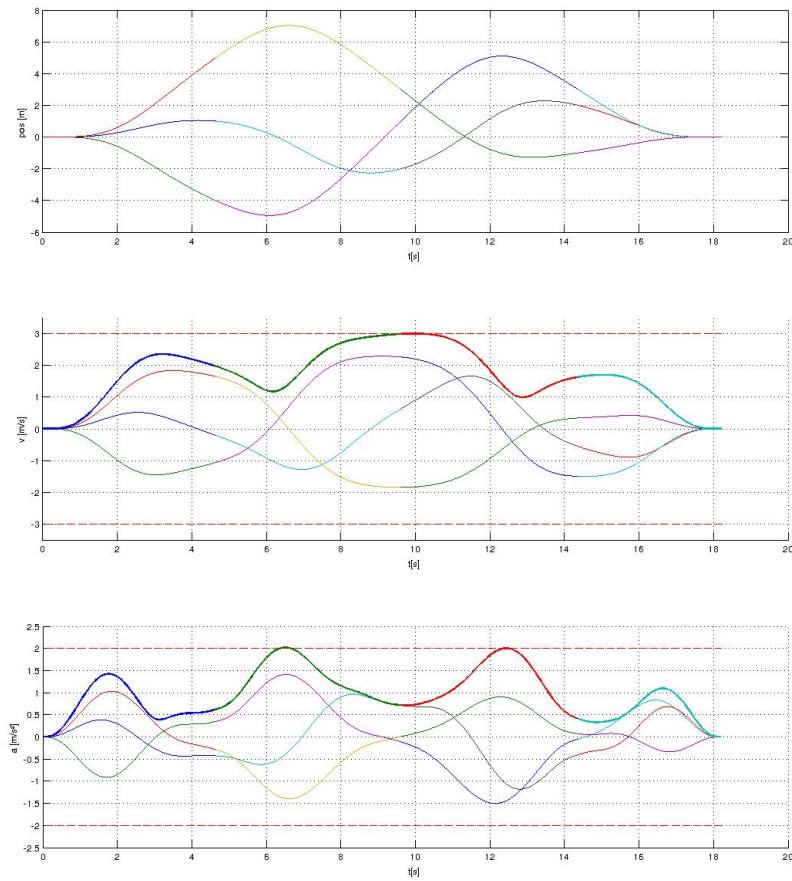


Figure 2.3: Optimized solution: The first subplot shows the position, the second the velocity and the third the acceleration of the optimized solution. A thicker graph represents the 2-norm of the velocity respectively the acceleration.

## Chapter 3

# RRT

### 3.1 General

The goal of this Thesis was not only to generate a numerical stable, snap optimized polynomial trajectory but also to explore a dense (indoor) environment and plan an aggressive trajectory in between the obstacles. Hence, the Rapidly-Exploring Random Tree (RRT) algorithm is used to find a collision free straight-line solution through the dense environment. The sampling points of the RRT (or RRT\*) algorithm are then used as the vertices in the polynomial optimization.

### 3.2 Algorithm

#### 3.2.1 RRT

#### 3.2.2 RRT\*

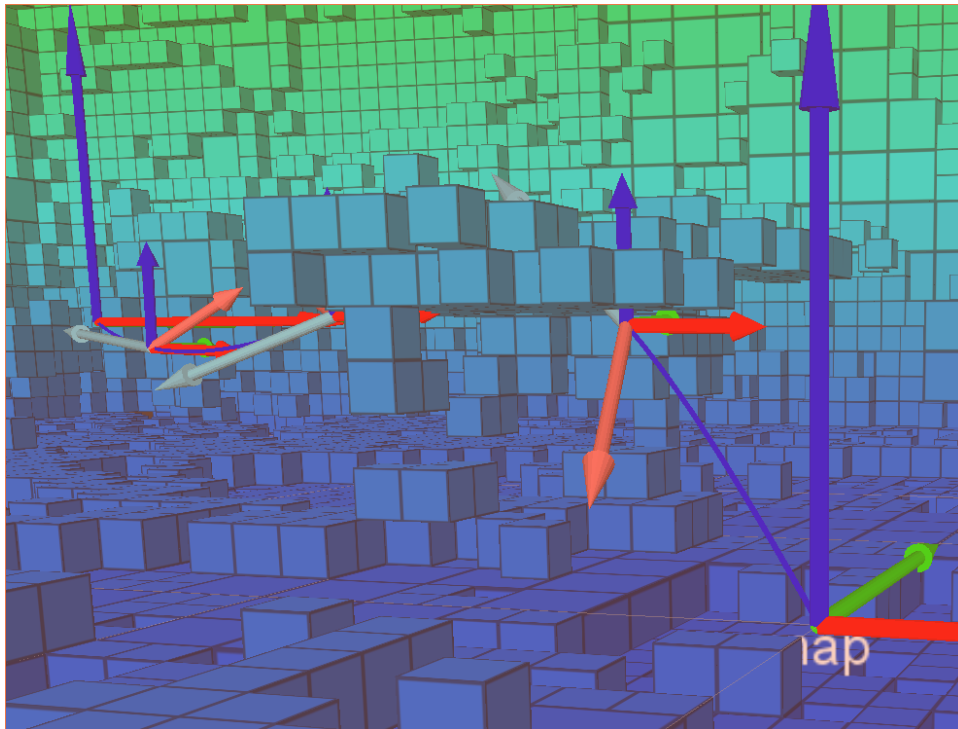


Figure 3.1: Ein Bild.

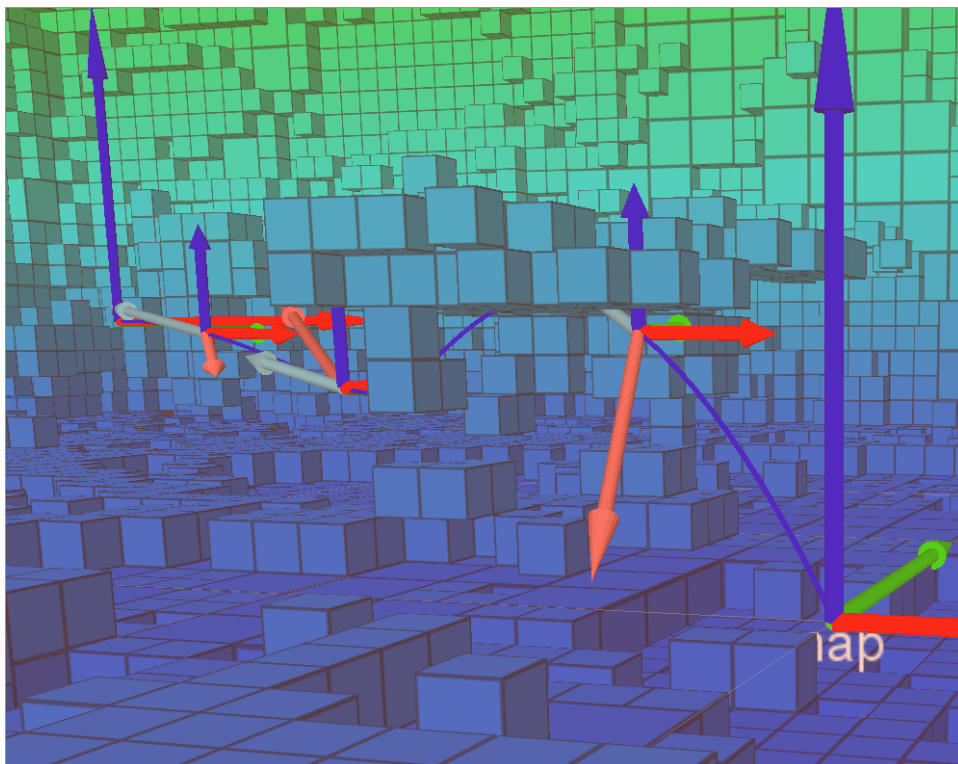


Figure 3.2: Ein Bild.



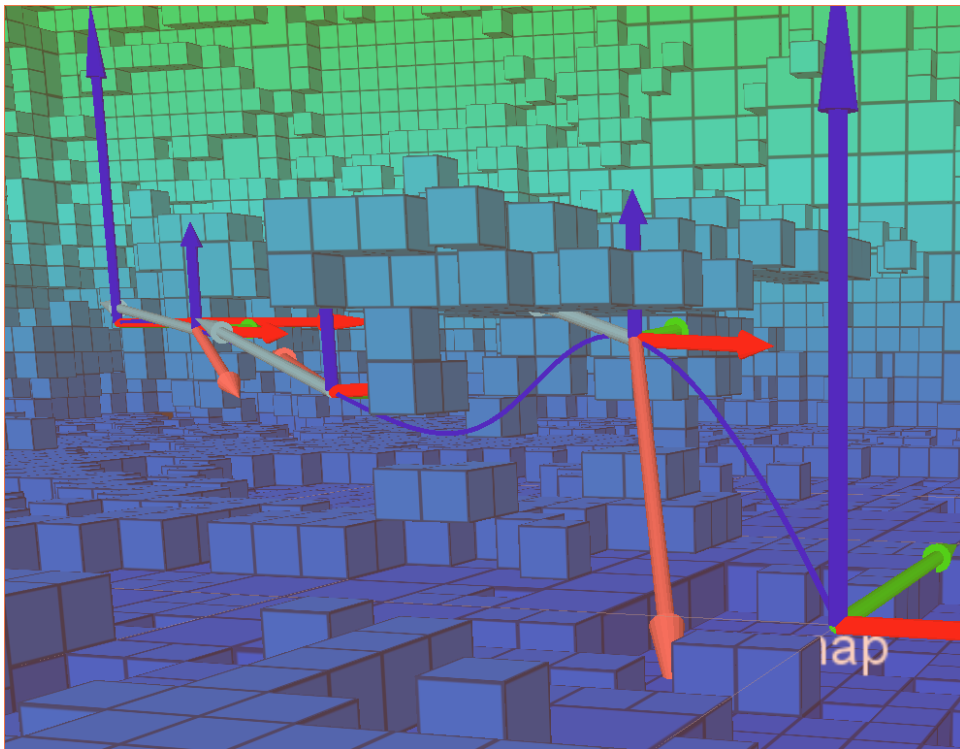


Figure 3.3: Ein Bild.

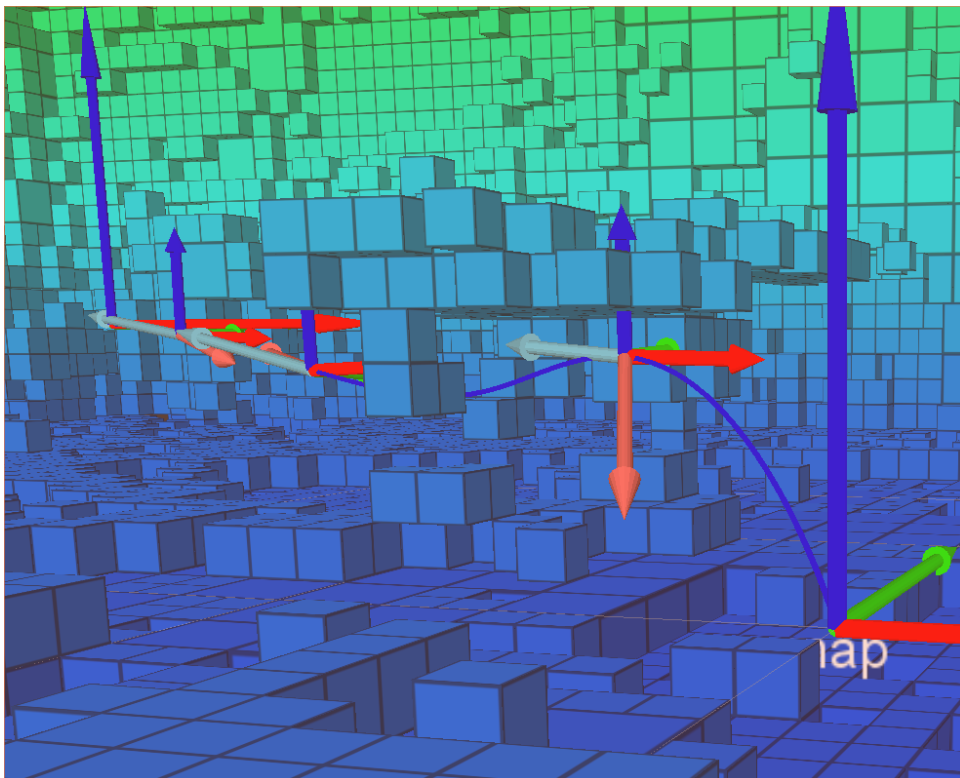


Figure 3.4: Ein Bild.

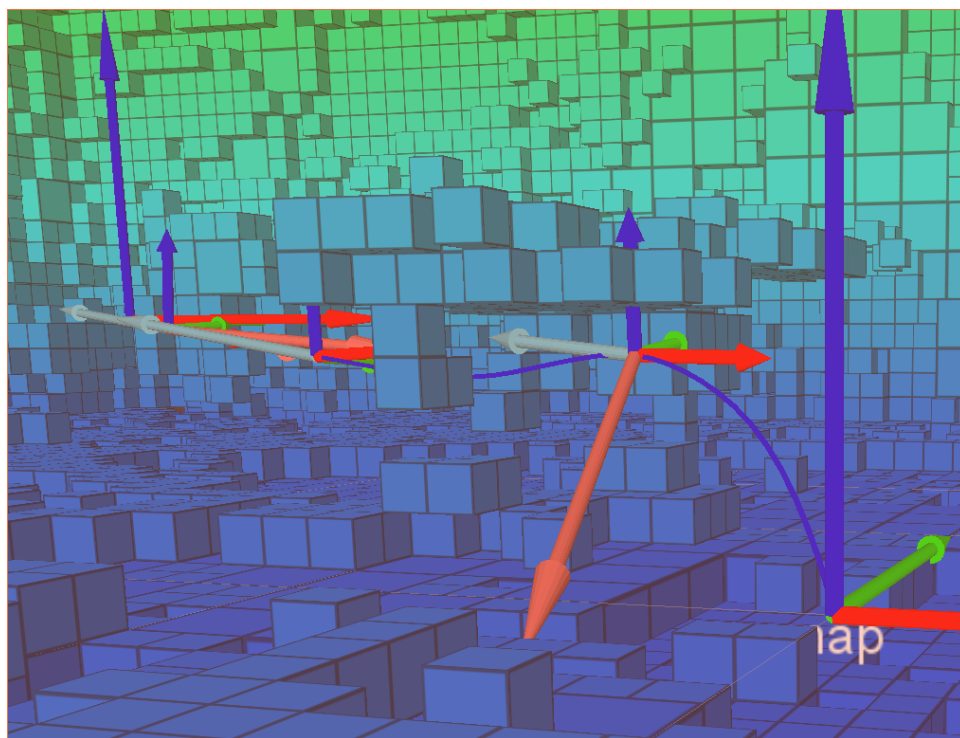


Figure 3.5: Ein Bild.

# Appendix A

## Irgendwas

Bla bla ...



## Appendix B

# Nochmals irgendwas

Bla bla ...



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