



# PECOS

Predictive Engineering and Computational Sciences

## Tools and Techniques for Software Verification using Manufactured Solutions

Nicholas Malaya, Chris Simmons

Center for Predictive Engineering and Computational Sciences (PECOS)  
Institute for Computational Engineering and Sciences (ICES)  
The University of Texas at Austin

July 9th, 2012

# Outline

This talk is online:

`users.ices.utexas.edu/~nick/masa-lanl.pdf`

## Lecture

- Motivation
- V&V-UQ
- Creating Manufactured Solutions
- The MASA Library

# Scientific Computing In Practice

## Numerical simulations have a broad range of applicability

- Informing a decision-making process
- Helping to further our understanding of the physical world
- Computer Aided Design (e.g. Boeing 777)
- Human treatment and drug discovery
- Societal impact (global warming yes/no?)
- Disaster recovery and prediction (e.g. earthquakes, hurricanes, storm surges)
- *Possibly getting you a degree...*

# Scientific Computing

## Most types of programming:

- Some errors are tolerable, some physics are negotiable (e.g. shaders, rendering)
- Speed is negotiable
- Often has to be pretty or easy to use, or both

## Scientific and Technical Computing:

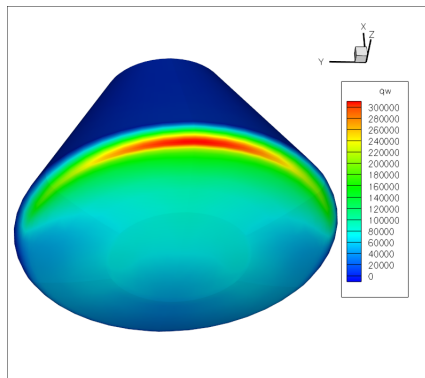
- You have to be correct (or at least quantifiably wrong)
- We often don't know the “correct” answer
- Sometimes we cannot make new experiments (e.g. NNSA stockpile stewardship)
- Needs to be fast (*think hurricane weather prediction*)
- Software does not need to be easy to use

# The PECOS Center

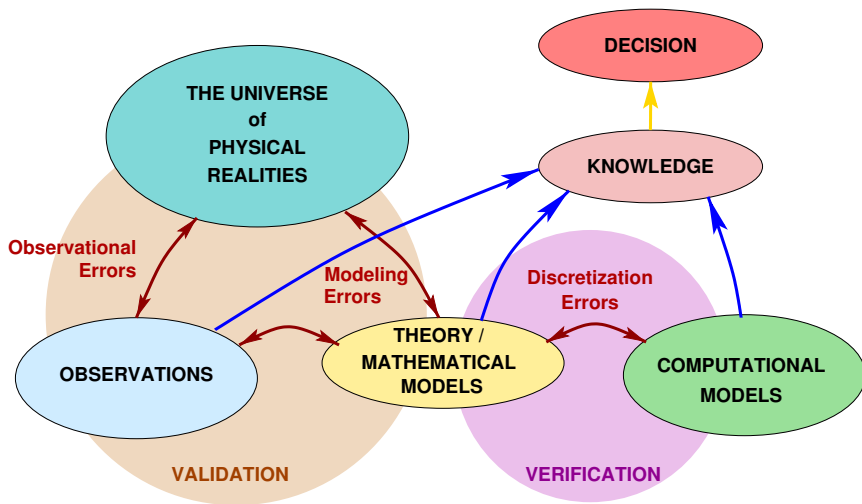
## Predictive Engineering and COmputational Science

### Physics Modeling

- Modeling Reentry vehicle
- Complex multiphysics problems:
  - ▶ turbulence
  - ▶ radiation
  - ▶ chemistry
  - ▶ ablation
  - ▶ etc.
- **Are we confident in our predictions?**



# Verification, Validation and Uncertainty Quantification



# Verification

## Verification of Scientific Software

- Verification ensures that the outputs of a computation accurately reflect the solution of the mathematical models.

## Code Verification

- Ensuring that the code used in the simulation correctly implements the intended numerical discretization of the model.

## Solution Verification

- Are the errors introduced by the numerical discretization sufficiently small?
- Is the convergence rate consistent with the numerical scheme?

# Introduction to the Method of Manufactured Solutions

## Method of Exact Solutions

- Numerically solve the governing equations for which the solution can be determined analytically.

## MMS

- Often, analytical solutions either:
  - ▶ Do not exist
  - ▶ Does not fully exercise equations (e.g. a symmetric solution, nonlinearities)
- Alleviate this using Method of Manufactured Solutions (MMS)
  - ▶ Simply put, we “create” our own solutions



# Manufactured solution to Laplace's Equation

Laplace's Equation:

$$\nabla^2 \phi = 0$$

In two dimensions:

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$$

“Manufacture” a solution, with two constants:

$$\phi(x, y) = (Ly - y)^2(Ly + y)^2 + (Lx - x)^2(Lx + x)^2$$

# Calculating the Source Term

We insert our manufactured solution back into the governing equations:

$$\frac{\partial^2((Lx - x)^2(Lx + x)^2)}{\partial x^2} + \frac{\partial^2((Ly - y)^2(Ly + y)^2)}{\partial y^2} = 0$$

$$\begin{aligned} &= 2(Lx - x)^2 - 8(Lx - x)(Lx + x) + 2(Lx + x)^2 \\ &+ 2(Ly - y)^2 - 8(Ly - y)(Ly + y) + 2(Ly + y)^2 \\ &\neq 0 \end{aligned}$$

This does not satisfy Laplace's Equation!

To balance the equation, add the residual to the RHS as a source term.

## Example Verification Use Case

To solve Laplace's Equation numerically, we need a discretization scheme.

Let's use a 2nd order finite central difference:

$$\phi_i'' \approx \frac{\phi_{i+1} - 2\phi_i + \phi_{i-1}}{h^2} + O(h^2)$$

This requires solving the implicit system of equations:

$$A\vec{\phi} = \textcolor{red}{f}$$

You can use your favorite linear solver (e.g. PETSc) to solve the system.

# Problem: Solve 2D Laplacian using Finite-Differencing

## Outline

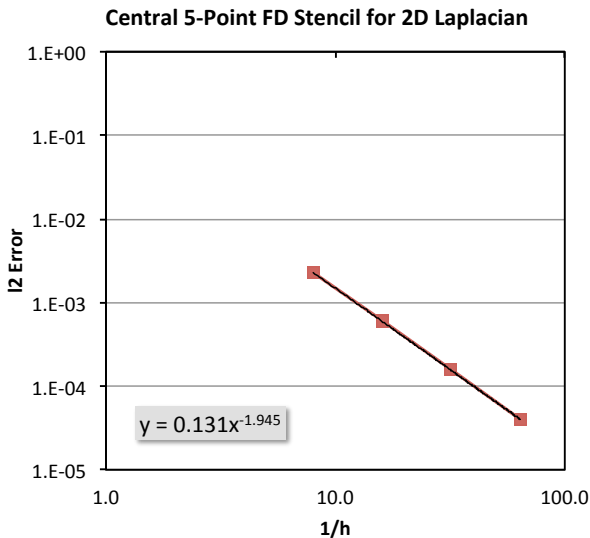
- *Goal*: Write a program in C/C++, F90
- *Inputs*:
  - ▶ # of points in one direction ( $npts$ )
  - ▶ the physical dimension of one side ( $L_x, L_y$ )
- *Output*:  $l_2$  error between your numerical solution and an exact solution derived from a manufactured solution

$$l_2 = \sqrt{\frac{\sum_{i=1}^N (\phi_i - \phi_i^{\text{exact}})^2}{N}}$$

- *Runs*: Run your snazzy code for  $npts = 5, 9, 17$ , and  $33$  and plot  $l_2$  norm as a function of  $1/h$  where  $h = \text{length}/(npts - 1)$

# Example Results: What we're hoping for

## 2nd Order Central Finite-difference Scheme



# Useful for Detecting Bugs

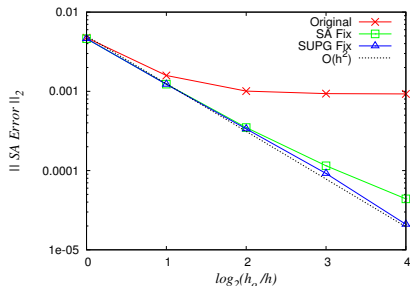
## Verification of FIN-S

- FANS, Spalart-Allmaras
- Derivative:

$$\frac{d(sa)}{dx} = \frac{1}{\rho} * \left( \frac{d(\rho * sa)}{dx} - sa \frac{d\rho}{dx} \right)$$

- In code:

$$\frac{d(sa)}{dx} = \frac{1}{\rho} * \frac{d(\rho * sa)}{dx} - sa \frac{d\rho}{dx}$$



# Why is this not more commonly done?

# A Real Example

## MMS Creation Process

- Start by “manufacturing” a suitable closed-form exact solution
- For example, the 10 parameter trigonometric solution of the form:  
(Roy, 2002)

$$\hat{u}(x, y, z, t) = \hat{u}_0 + \hat{u}_x f_s \left( \frac{a \hat{u}_x \pi x}{L} \right) + \hat{u}_y f_s \left( \frac{a \hat{u}_y \pi y}{L} \right) + \\ + \hat{u}_z f_s \left( \frac{a \hat{u}_z \pi z}{L} \right) + \hat{u}_t f_s \left( \frac{a \hat{u}_t \pi t}{L} \right)$$

- Apply this solution to equations of interest, solve for source terms (residual)

Accomplished using packages such as Maple, Mathematica, SymPy, Macsyma, etc.



# Maple MMS: 3D Navier-Stokes Energy Term

$$\begin{aligned}
 Qe = & -\frac{a_{px}\pi p_x}{L} \frac{\gamma}{\gamma-1} \sin\left(\frac{a_{px}\pi x}{L}\right) \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right] + \\
 & + \frac{a_{py}\pi p_y}{L} \frac{\gamma}{\gamma-1} \cos\left(\frac{a_{py}\pi y}{L}\right) \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right] + \\
 & - \frac{a_{pz}\pi p_z}{L} \frac{\gamma}{\gamma-1} \sin\left(\frac{a_{pz}\pi z}{L}\right) \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right] + \\
 & + \frac{a_{px}\pi \rho_x}{2L} \cos\left(\frac{a_{px}\pi x}{L}\right) \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right] \left( \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right]^2 + \right. \\
 & \quad \left. + \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right]^2 + \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right]^2 \right) + \\
 & - \frac{a_{py}\pi \rho_y}{2L} \sin\left(\frac{a_{py}\pi y}{L}\right) \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right] \left( \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right]^2 + \right. \\
 & \quad \left. + \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right]^2 + \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right]^2 \right) + \\
 & + \frac{a_{pz}\pi \rho_z}{2L} \cos\left(\frac{a_{pz}\pi z}{L}\right) \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right] \left( \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right]^2 + \right. \\
 & \quad \left. + \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right]^2 + \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right]^2 \right) + \\
 & + \frac{a_{ux}\pi u_x}{2L} \cos\left(\frac{a_{ux}\pi x}{L}\right) \left\{ \left( \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right]^2 + \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right]^2 \right) \right. \\
 & \quad \left. + 3 \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right]^2 \right\} \left[ \rho_0 + \rho_x \sin\left(\frac{a_{px}\pi x}{L}\right) + \rho_y \cos\left(\frac{a_{py}\pi y}{L}\right) + \rho_z \sin\left(\frac{a_{pz}\pi z}{L}\right) \right] + \\
 & + \left[ p_0 + p_x \cos\left(\frac{a_{px}\pi x}{L}\right) + p_y \sin\left(\frac{a_{py}\pi y}{L}\right) + p_z \cos\left(\frac{a_{pz}\pi z}{L}\right) \right] \frac{2\gamma}{(\gamma-1)} + \\
 & - \frac{a_{uy}\pi u_y}{L} \sin\left(\frac{a_{uy}\pi y}{L}\right) \left[ v_0 + v_x \cos\left(\frac{a_{vx}\pi x}{L}\right) + v_y \sin\left(\frac{a_{vy}\pi y}{L}\right) + v_z \sin\left(\frac{a_{vz}\pi z}{L}\right) \right] \left[ \rho_0 + \rho_x \sin\left(\frac{a_{px}\pi x}{L}\right) + \rho_y \cos\left(\frac{a_{py}\pi y}{L}\right) + \rho_z \sin\left(\frac{a_{pz}\pi z}{L}\right) \right] \cdot \\
 & \quad \cdot \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right] + \\
 & - \frac{a_{wz}\pi u_z}{L} \sin\left(\frac{a_{wz}\pi z}{L}\right) \left[ w_0 + w_x \sin\left(\frac{a_{wx}\pi x}{L}\right) + w_y \sin\left(\frac{a_{wy}\pi y}{L}\right) + w_z \cos\left(\frac{a_{wz}\pi z}{L}\right) \right] \left[ \rho_0 + \rho_x \sin\left(\frac{a_{px}\pi x}{L}\right) + \rho_y \cos\left(\frac{a_{py}\pi y}{L}\right) + \rho_z \sin\left(\frac{a_{pz}\pi z}{L}\right) \right] \cdot \\
 & \quad \cdot \left[ u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_y \cos\left(\frac{a_{uy}\pi y}{L}\right) + u_z \cos\left(\frac{a_{uz}\pi z}{L}\right) \right] +
 \end{aligned}$$

## But wait, there's more!

[illegible]

## C-code output

[illegible]

# Robust Verification

## What Constitutes a Robust Test?

- How “strong” is a verification test for a particular codebase?
  - ▶ Can you characterize your confidence in a codebase?
  - ▶ What constitutes a strong test?

## Verification Metrics

- In general, you cannot “verify” a codebase
  - ▶ You can, however, detect **verification failures**

## Physically-based MS

- Exercise each term in the PDE in a similar way to that of a real solution

# Example Problem

## Complex Codebase

- Finite Element hypersonic code, fully implicit Navier-Stokes (FIN-S)
- Favre-Averaged Navier Stokes (FANS) + Spalart-Allmaras (SA) turbulence model

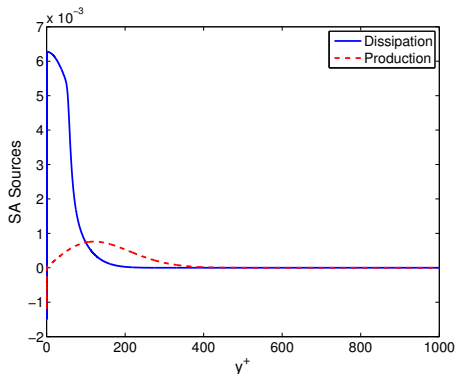
## FANS + SA

$$\begin{aligned}\frac{\partial \bar{\rho}}{\partial t} + \frac{\partial}{\partial x_i} (\bar{\rho} \tilde{u}_i) &= 0 \\ \frac{\partial}{\partial t} (\bar{\rho} \tilde{u}_i) + \frac{\partial}{\partial x_j} (\bar{\rho} \tilde{u}_j \tilde{u}_i) &= - \frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} \left( 2(\mu + \mu_t) \tilde{S}_{ji} \right) \\ \frac{\partial}{\partial t} \left[ \bar{\rho} \left( \tilde{e} + \frac{1}{2} \tilde{u}_i \tilde{u}_i \right) \right] + \frac{\partial}{\partial x_j} \left[ \bar{\rho} \tilde{u}_j \left( \tilde{h} + \frac{1}{2} \tilde{u}_i \tilde{u}_i \right) \right] &= \frac{\partial}{\partial x_j} \left( 2(\mu + \mu_t) \tilde{S}_{ji} \tilde{u}_i \right) + \frac{\partial}{\partial x_j} \left[ \left( \frac{\mu}{\text{Pr}} + \frac{\mu_t}{\text{Pr}_t} \right) \frac{\partial \tilde{h}}{\partial x_j} \right] \\ \frac{\partial}{\partial t} (\bar{\rho} \nu_{\text{sa}}) + \frac{\partial}{\partial x_j} (\bar{\rho} \tilde{u}_j \nu_{\text{sa}}) &= c_{b1} S_{\text{sa}} \bar{\rho} \nu_{\text{sa}} - c_{w1} f_w \bar{\rho} \left( \frac{\nu_{\text{sa}}}{d} \right)^2 + \frac{1}{\sigma} \frac{\partial}{\partial x_k} \left[ (\mu + \bar{\rho} \nu_{\text{sa}}) \frac{\partial \nu_{\text{sa}}}{\partial x_k} \right] + \frac{c_{b2}}{\sigma} \bar{\rho} \frac{\partial \nu_{\text{sa}}}{\partial x_k} \frac{\partial \nu_{\text{sa}}}{\partial x_k}\end{aligned}$$

# Insufficient Verification

## Shortcoming of other SA MS

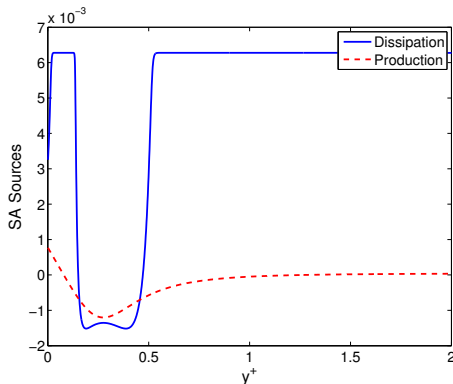
- Bond solution: sinusoidal, only satisfies no-slip.
- Eça solutions were noted to have a suboptimal rate of convergence



# A Closer Look

## Shortcoming of other SA MS

- Eça solutions are shown to have instabilities or near-wall features that disrupt the correct rate of convergence
  - ▶ Offending term:  $\nu_{sa} = \tilde{\nu}_{\max} \eta_{\nu}^2 e^{1-\eta_{\nu}^2}$



# Our SA Manufactured Solution

- Parameters colored in red (values appear in paper appendix)
- Using our understanding of incompressible flow physics to inform our modeling assumptions for this MS

## Streamwise Velocity

- The mean streamwise velocity is given by,

$$\tilde{u} = \frac{u_{\infty}}{A} \sin \left( \frac{A}{u_{\infty}} u_{eq} \right)$$

The van Driest equivalent velocity can be written as,

$$u_{eq} = u_{\tau} u_{eq}^{+},$$

- Must specify both  $u_{\tau}$  and  $u_{eq}^{+}$



# Completing Streamwise Velocity Specification

## Correlations

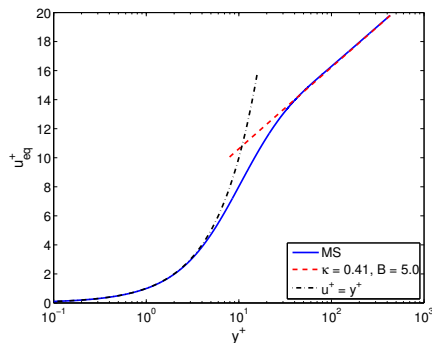
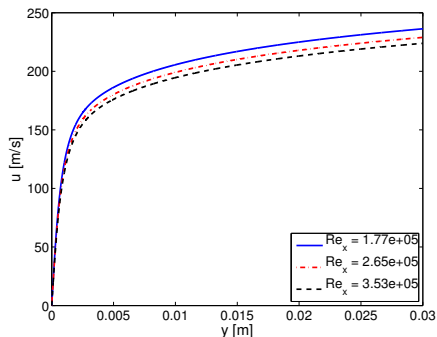
- Friction velocity can be determined from the skin friction coefficient
- The incompressible 1/7th power law is used for the skin friction coefficient. Thus,

$$c_{f,\text{inc}}(Re_x) = C_{cf} Re_x^{-1/7}$$

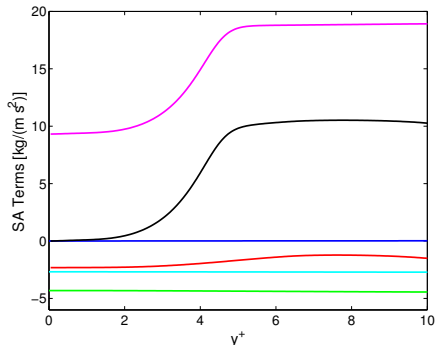
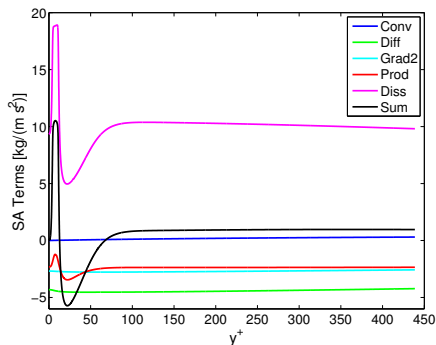
- To complete the manufactured solution,  $u_{eq}^+$  is set using the velocity profile model of Cebeci and Bradshaw (1980):

$$u_{eq}^+ = \frac{1}{\kappa} \log(1 + \kappa y^+) + C_1 \left[ 1 - e^{-y^+/\eta_1} - \frac{y^+}{\eta_1} e^{-y^+/\eta_1} \right]$$

# Manufactured Velocity Profiles

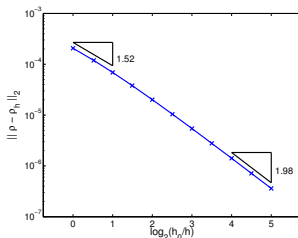


# SA Equation Budgets

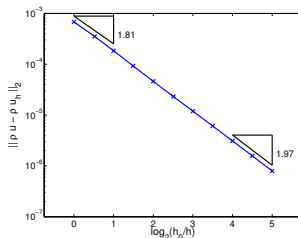


- Inside viscous sublayer, source term is small relative to other terms
- Production and dissipation terms go to constants at the wall
- Source term is largest in buffer region

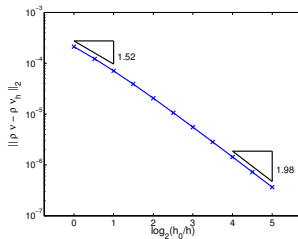
# Convergence Rates: Low $Re_x$



(a)  $\bar{\rho}$

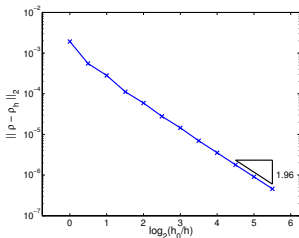


(b)  $\bar{\rho} \tilde{u}$

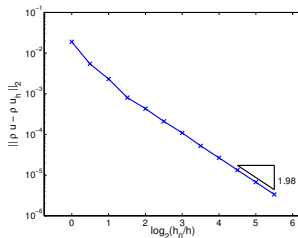


(c)  $\bar{\rho} \nu_{sa}$

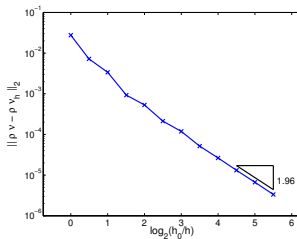
# Convergence Rates: $Re_x = 3.5 * 10^5$



(d)  $\bar{\rho}$



(e)  $\bar{\rho} \tilde{u}$



(f)  $\bar{\rho} \nu_{sa}$

# Summary

## What we have learned:

- Be skeptical of an MS like you would be skeptical of a model
  - ▶ Manufactured Solutions must be verified!
- Exercise each term in the PDE in a similar way to that of a real solution
- *More details:* “Manufactured Solutions for the Favre-Averaged Navier-Stokes Equations with Eddy-Viscosity Turbulence Models”
  - ▶ AIAA 50th Aerospace Sciences Meeting
  - ▶ AIAA Journal (In Revision)

# Manufactured Analytical Solutions Abstractions Library

**Goal:** Provide a repository and standardized interface for MMS usage

## High Priority:

- Extreme fidelity to generated MMS
- Portability
- Traceability
- Extensible

## Low Priority:

- Speed/Performance

# Verifying the “Verifier”

Precision is not negotiable.

## MASA Testing

- Error target  $< 1e-15$ 
  - ▶ Absolute error on local machines
  - ▶ Relative error (other)
  - ▶ On all supported compiler sets
- -O0 not sufficient
  - ▶ -fp-model precise (Intel)
  - ▶ -fno-unsafe-math-optimizations (GNU)
  - ▶ -Kieee -Mnofpapprox (PGI)
- “make check”
  - ▶ Run by Buildbot every two hours

```
[nick@magus trunk]$ make check
```

```
-----  
Initializing MASA Tests  
-----
```

```
PASS: init.sh  
PASS: misc  
PASS: fail_cond  
PASS: catch_exception  
PASS: register  
PASS: poly  
PASS: uninit  
PASS: vec  
PASS: purge  
PASS: heat_const_steady  
PASS: euler1d
```

```
:  
:  
:
```

```
-----  
Finalizing MASA Tests  
-----
```

```
=====  
All 60 tests passed  
=====
```



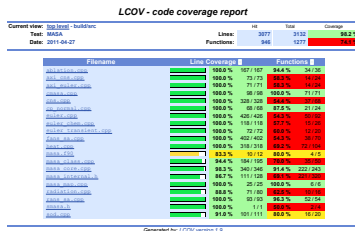
# Portability

## Software Environment

- Built with: Autotools, C++
- Supports Intel, GNU, **Portland Group** compilers
- C/C++ interfaces
- Fortran interfaces provided through **iso.c bindings**
  - ▶ Fortran 2003 Standard

## Testing

- SVN: version control
- Buildbot: automated testing
- GCOV: line coverage
  - ▶ 15,826 lines of code
  - ▶ 13,195 lines of testing
  - ▶ 98%+ line coverage



# Traceability

## Doxygen provides code and model documentation

### 3.2 Euler Equations

19

where  $\phi = \rho, u, v, w$  or  $p$ , and  $f_n(\cdot)$  functions denote either sine or cosine function. Note that in this case,  $\phi_x, \phi_y$  and  $\phi_z$  are constants and the subscripts do not denote differentiation.

Although ? provide the constants used in the manufactured solutions for the 2D supersonic and subsonic cases for Euler and Navier-Stokes equations, only the source term for the 2D mass conservation equation (3.20) is presented.

Source terms for mass conservation ( $Q_\rho$ ), momentum ( $Q_u, Q_v$  and  $Q_w$ ) and total energy ( $Q_e$ ) equations are obtained by symbolic manipulations of compressible steady Euler equations above using Maple 13 (?) and are presented in the following sections for the one, two and three-dimensional cases.

#### 3.2.2.1 1D Steady Euler

The manufactured analytical solutions (3.52) for each one of the variables in one-dimensional case of Euler equations are:

$$\begin{aligned}\rho(x) &= \rho_0 + \rho_s \sin\left(\frac{a_{\rho s} \pi x}{L}\right) \\ u(x) &= u_0 + u_s \sin\left(\frac{a_{us} \pi x}{L}\right) \\ p(x) &= p_0 + p_s \cos\left(\frac{a_{ps} \pi x}{L}\right)\end{aligned}\quad (3.26)$$

The MMS applied to Euler equations consists in modifying the 1D Euler equations (3.20) – (3.22) by adding a source term to the right-hand side of each equation:

$$\begin{aligned}\frac{\partial(\rho u)}{\partial x} &= Q_\rho \\ \frac{\partial(\rho u^2)}{\partial x} + \frac{\partial(p)}{\partial x} &= Q_u \\ \frac{\partial(\rho u v)}{\partial x} + \frac{\partial(p v)}{\partial x} &= Q_v\end{aligned}\quad (3.27)$$

so the modified set of equations (3.27) conveniently has the analytical solution given in Equation (3.53).

Source terms  $Q_\rho, Q_u$  and  $Q_v$ , are obtained by symbolic manipulations of equations above using Maple and are presented in the following sections. The following auxiliary variables have been included in order to improve readability and computational efficiency:

$$\begin{aligned}\text{rho}_1 &= \rho_0 + \rho_s \sin\left(\frac{a_{\rho s} \pi x}{L}\right) \\ U_1 &= u_0 + u_s \sin\left(\frac{a_{us} \pi x}{L}\right) \\ P_1 &= p_0 + p_s \cos\left(\frac{a_{ps} \pi x}{L}\right)\end{aligned}$$

where the subscripts refer to the 1D case.

The mass conservation equation written as an operator is:

$$\mathcal{L} = \frac{\partial(\rho u)}{\partial x}$$

Generated on Mon Apr 25 11:02:30 2011 for MASS-4.32.0 by Doxygen

### 3.2 Euler Equations

29

k	u_0	u_x	u_y	u_z	v_0	L
v_0	v_x	v_y	v_z	w_0	w_x	w_y
rho_0	rho_x	rho_y	rho_z	p_0	p_x	p_y
a_px	a_py	a_pz	a_rhox	a_rhoy	a_rhoz	a_us
a_uy	a_uz	a_vx	a_vy	a_vz	a_wx	a_wy
a_wz	mu	Gamma				

Table 3.6: Parameters used by the 3D Steady Euler

- `masa_eval_2d_exact_u()`
- `masa_eval_2d_exact_v()`
- `masa_eval_2d_exact_p()`
- `masa_eval_2d_exact_rho()`
- `masa_eval_2d_grad_u()`
- `masa_eval_2d_grad_v()`
- `masa_eval_2d_grad_p()`
- `masa_eval_2d_grad_rho()`

#### 3.2.3.3 3D Steady Euler

Initialization:

- `euler_3d`

Functions:

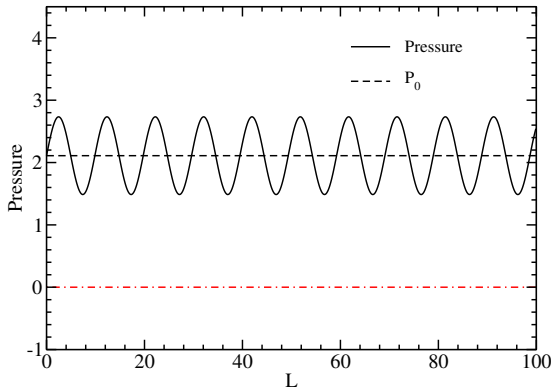
- `masa_init()`
- `masa_eval_3d_source_rho_u()`
- `masa_eval_3d_source_rho_v()`
- `masa_eval_3d_source_rho_w()`
- `masa_eval_3d_source_rho_e()`
- `masa_eval_3d_source_rho_x()`
- `masa_eval_3d_exact_u()`
- `masa_eval_3d_exact_v()`
- `masa_eval_3d_exact_w()`
- `masa_eval_3d_exact_p()`

Generated on Mon Apr 25 11:02:30 2011 for MASS-4.32.0 by Doxygen

# Providing Reasonable Defaults

## Requirements:

- Sufficient algebraic complexity to exercise all terms
- Physically consistent
  - ▶ e.g. solutions should not return negative densities, temperatures, etc.



## Available Solutions in MASA 0.40

Equations	Dimensions	Time
Euler	1,2,3	Transient, Steady
Heat	1,2,3	Transient, Steady
Navier-Stokes	1,2,3	Transient, Steady
N-S + Sutherland	3	Transient, Steady
N-S + ablation	1	Transient, Steady
Sod Shock Tube	1	Transient
Euler + chemistry	1	Steady
RANS: Spalart-Allmaras	1	Steady
FANS: SA	2	Steady
FANS: SA + wall	2	Steady
Radiation	1	Steady
SMASA: Gaussian distribution	1	Steady

# Future Solution Development

## Single Physics

- Additional RANS models ( $v^2$ -f, k- $\epsilon$ , etc.)
- Shocks

## Multiphysics

- Turbulence with chemistry
- Flow with improved transport

## Different physical systems

- Einstein's field equations (General Relativity)
- Schrodinger equation (Quantum Mechanics)
- Black-Scholes (Finance)
- etc.

# Importing New Solutions

## Requirements

- Latex documents can be loaded directly into MASA documentation
    - ▶ Model document detailing analytical solution and source terms
    - ▶ Interface documentation detailing parameters and functions
  - Source and analytical terms in C/C++/Fortran90
    - ▶ Can be integrated into your local MASA copy automatically using a perl script
    - ▶ Submit a patch
      - unit tests
  - **Willingness to share**
- 
- Automate the import process as much as possible
  - Success of MASA depends on use as a community tool

# Snapshot

## Release

- MASA 0.40.2 was released March 3rd, 2012
- <https://red.ices.utexas.edu/projects/software>
- Open source, LGPL V2.1, free

## Publication

- “MASA: a library for verification using manufactured analytical solutions”
  - ▶ Engineering With Computers
  - ▶ DOI: 10.1007/s00366-012-0267-9

# Combinatorial Explosion

## Explosion in source term size

- Complexity increases with more sophisticated mathematical models
  - Sutherland viscosity model has over 1,612,000 characters
- Large memory requirements (128 GB not sufficient for Sutherland)
- Computational intensity
- **segfaults**

## Hierarchic MMS

- Decompose each equation into sub-terms
- Each term is operated on individually to plug in a manufactured solution
- Resulting expressions are re-combined to regain source term



# The Hierarchic MMS

Consider the full 3D Navier-Stokes energy equation:

$$\mathcal{L} = \frac{\partial(\rho e_t)}{\partial t} + \nabla \cdot (\rho \mathbf{u} e_t) + \nabla \cdot \mathbf{q} + \nabla \cdot (p \mathbf{u}) - \nabla \cdot (\boldsymbol{\tau} \cdot \mathbf{u})$$

Decompose:

$$\mathcal{L}_1 = \frac{\partial(\rho e_t)}{\partial t}$$

$$\mathcal{L}_2 = \nabla \cdot (\rho \mathbf{u} e_t)$$

$$\mathcal{L}_3 = \nabla \cdot \mathbf{q}$$

$$\mathcal{L}_4 = \nabla \cdot (p \mathbf{u})$$

$$\mathcal{L}_5 = -\nabla \cdot (\boldsymbol{\tau} \cdot \mathbf{u})$$

## Hierarchic MMS extensions:

- Expand each component of divergence
- Transient and steady cases (  $\mathcal{L}_1 = 0$  )
- Sutherland model only requires altering  $\mathcal{L}_5$
- Navier-Stokes  $\rightarrow$  Euler and additional terms

# Hierarchical MMS Results

## Reductions

- Symbolic factorization tricks permit further simplification of these operators:

Term	Before	After	Reduction
$\mathcal{L}_1$	$70.1 \times 10^3$	$1.1 \times 10^3$	98.4%
$\mathcal{L}_2$	$292.8 \times 10^3$	$4.0 \times 10^3$	98.6%
$\mathcal{L}_3$	$11.3 \times 10^3$	$1.2 \times 10^3$	89.3%
$\mathcal{L}_4$	$1.5 \times 10^3$	$5.8 \times 10^2$	61.3%
$\mathcal{L}_5$	$3.1 \times 10^3$	$1.3 \times 10^3$	58.0%
$\mathcal{L}$	$378.8 \times 10^3$	$8.2 \times 10^3$	97.8%

[illegible]

## Original source term C-output

[illegible]

# Shortcomings

## Symbolic Shortcomings

- Hierarchic decomposition is an improvement, but is it enough?
  - ▶ Even with factorization, source terms still massive
  - ▶ Generating manufactured solutions was a **full time job** at PECOS
- Everything we have discussed has been generated outside of MASA
  - ▶ Not thrilled with Maple, Mathematica

## Enter Automatic Differentiation

- AD numerically evaluates the derivative of a function
  - ▶ applies chain rule repeatedly
- Superior error characteristics (round-off)
- Slow (but we barely care)
- Several libraries: NAG, Sacado, etc.

# Review: Complex Numbers

- A new element  $i$
- Take the quotient with  $i^2 \equiv -1$

$$\mathbb{C}[\mathbb{R}] \equiv \{a + bi : a, b \in \mathbb{R}\}$$

- Arithmetic:

$$(a + bi) + (c + di) = ((a + c) + (b + d)i)$$

$$(a + bi) - (c + di) = ((a + c) - (b + d)i)$$

$$\begin{aligned}(a + bi) \times (c + di) &= (ac) + adi + bci + bdi^2 \\ &= ((ac) + (ad + bc)i - (bd))\end{aligned}$$

# “Dual Numbers” - [Clifford 1873],[Study 1891]

- A new element  $\epsilon$
- Closed under addition and multiplication:

$$\left\{ \sum_{i=0}^m a_i \epsilon^i : a_i \in \mathbb{R}, m < \infty \right\}$$

- Take the quotient with  $\epsilon^2 \equiv 0$

$$\mathbb{D}[\mathbb{R}] \equiv \{a + b\epsilon : a, b \in \mathbb{R}\}$$

- Used with quaternions to represent rotations and translations
- Arithmetic:

$$(a + b\epsilon) + (c + d\epsilon) = ((a + c) + (b + d)\epsilon)$$

$$(a + b\epsilon) - (c + d\epsilon) = ((a + c) - (b + d)\epsilon)$$

$$\begin{aligned}(a + b\epsilon) \times (c + d\epsilon) &= (ac) + ad\epsilon + bc\epsilon + bd\epsilon^2 \\ &= ((ac) + (ad + bc)\epsilon)\end{aligned}$$

# “Hyper-dual Numbers” - [Fike 2009]

- Add two new elements  $\epsilon_1, \epsilon_2$  to  $\mathbb{R}$
- Take the quotient with  $\epsilon_1^2 \equiv \epsilon_2^2 \equiv 0$

$$\mathbb{H}[\mathbb{R}] \equiv \{a + b\epsilon_1 + c\epsilon_2 + d\epsilon_1\epsilon_2 : a, b, c, d \in \mathbb{R}\}$$

- Arithmetic:

$$\begin{aligned}(a + b\epsilon_1 + c\epsilon_2 + d\epsilon_1\epsilon_2) + (e + f\epsilon_1 + g\epsilon_2 + h\epsilon_1\epsilon_2) &= \\ ((a + e) + (b + f)\epsilon_1 + (c + g)\epsilon_2 + (d + h)\epsilon_1\epsilon_2) &= \\ (a + b\epsilon_1 + c\epsilon_2 + d\epsilon_1\epsilon_2) - (e + f\epsilon_1 + g\epsilon_2 + h\epsilon_1\epsilon_2) &= \\ ((a - e) + (b - f)\epsilon_1 + (c - g)\epsilon_2 + (d - h)\epsilon_1\epsilon_2) &= \\ (a + b\epsilon_1 + c\epsilon_2 + d\epsilon_1\epsilon_2) \times (e + f\epsilon_1 + g\epsilon_2 + h\epsilon_1\epsilon_2) &= \\ (ae) + (af + be)\epsilon_1 + (ag + ce)\epsilon_2 + (ah + de + bg + cf)\epsilon_1\epsilon_2 &= \end{aligned}$$



# “Hyper-Dual Numbers”

- With  $\epsilon_1^2 \equiv \epsilon_2^2 \equiv 0 \equiv (\epsilon_1\epsilon_2)^2 = 0$
- Where  $X \equiv x + \epsilon_1 + \epsilon_2$ , we find:

$$f(X) = f(x) + f'(x)\epsilon_1 + f'(x)\epsilon_2 + f''(x)\epsilon_1\epsilon_2$$

- The Taylor series truncates exactly at the second-derivative term
- Using hyper-dual numbers results in first- and second-derivative calculations that are exact, regardless of step size
- Methods for computing exact higher derivatives can be created by using more non-real parts ( $\epsilon_3$ , for instance)
  - ▶ Accomplished using Templates in C++

# MASA PDE Examples

## Manufactured Solution

```
// Arbitrary manufactured solutions
U.template get<0>() = u_0 + u_x * sin(a_ux * PI * x / L) +
                    u_y * cos(a_uy * PI * y / L);

U.template get<1>() = v_0 + v_x * cos(a_vx * PI * x / L) +
                    v_y * sin(a_vy * PI * y / L);

ADScalar RHO = rho_0 + rho_x * sin(a_rhox * PI * x / L) +
                rho_y * cos(a_rhoy * PI * y / L);

ADScalar P = p_0 + p_x * cos(a_px * PI * x / L) +
              p_y * sin(a_py * PI * y / L);
```

# MASA PDE Examples

## Source Terms: Euler

```
// Gas state
ADScalar T = P / RHO / R;
ADScalar E = 1. / (Gamma-1.) * P / RHO;
ADScalar ET = E + .5 * U.dot(U);

// Mass, momentum and energy
Scalar Q_rho = raw_value(divergence(RHO*U));
RawArray Q_rho_u = raw_value(divergence(RHO*U.outerproduct(U)) +
                               P.derivatives());
Scalar Q_rho_e = raw_value(divergence((RHO*ET+P)*U));
```

# Future AD work

## Future Work

- Automatic Latex Generation
- Latex Parser – MMS generator
- Will this work with complex multiphysics?
- Inverse Problems

# Stochastic MASA and QUESO Verification

## Initial Effort

- Conjugate Prior(s)
  - ▶ Posterior is in the same family as the prior
- Initial problem: Gaussian
- QUESO: multilevel Monte Carlo
  - ▶ {5k, 50k, 500k} samples
- Tricky: sampling posterior
  - ▶ Convergence rates can be bounded

5k Posterior Mean	= 2.19617
50k Posterior Mean	= 2.18629
500k Posterior Mean	= 2.1789
SMASA Posterior Mean	= 2.17749

5k Posterior Std.Dev.	= 0.31234
50k Posterior Std.Dev.	= 0.313286
500k Posterior Std.Dev.	= 0.303793
SMASA Posterior Std.Dev.	= 0.301511

Future work: expand AD to inverse problems.

# Fortran 90 Reminder: What you need from MASA

```
program main
  use masa
  implicit none

  dx = real(lx)/real(nx)
  dy = real(ly)/real(ny);

  ! initialize the problem
  call masa_init("laplace example","laplace_2d")

  ! evaluate source terms (2D)
  do i=0, nx
    do j=0, ny

      y = j*dy
      x = i*dx

      ! evaluate source term
      field = masa_eval_2d_source_f (x,y)

      ! evaluate analytical term
      exact_phi = masa_eval_2d_exact_phi (x,y)

    enddo
  enddo

end program main
```

# C Reminder: What you need from MASA

```
#include <masa.h>

int main()
{
    err += masa_init("laplace example", "laplace_2d");

    // grab / set parameter values
    Lx = masa_get_param("Lx");
    masa_set_param("Ly", 42.0);

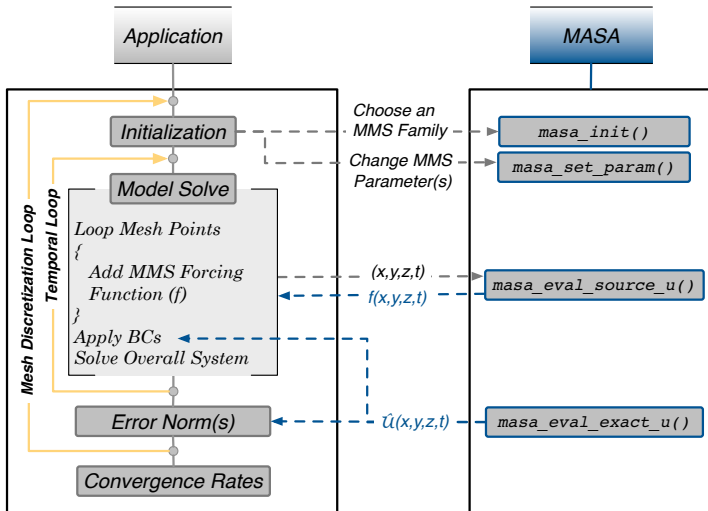
    for(int i=0; i<nx; i++)
        for(int j=0; j<ny; j++)
        {
            x=i*dx;
            y=j*dy;

            // source term
            ffield = masa_eval_2d_source_f (x,y);

            // manufactured solution
            phi_field = masa_eval_2d_exact_phi(x,y);

        } // finished iterating over space
    } //end program
```

# General Verification Approach Using MMS and MASA





# Conclusions

## Summary

- MMS is not a difficult concept, but can be tricky and time consuming
- Must have a high degree of confidence in your verification suite
- MASA is an open source library designed to:
  - ▶ Increase use of existing MMS in the community
  - ▶ Provide a standardized interface and toolset to the community
  - ▶ Serve as an example of high quality verification software
  - ▶ Available at: <https://red.ices.utexas.edu/projects/software>

# Conclusions

Thank you!

Have a well verified day.

[nick@ices.utexas.edu](mailto:nick@ices.utexas.edu)

Thank you for your attention.

Have a well verified day.

[nick@ices.utexas.edu](mailto:nick@ices.utexas.edu)