

Introduction

- We'll take you through launching a rocket and landing it on a different planet
- Obviously: this is simplified

Technicalities

Jargon

Orbit An object rotating around another object is said to be in orbit

Inclined An orbit that is tilted relative to the plane of the parent body

Delta-v

- Your fuel budget
- How hard can I accelerate/decelerate my rocket
- Depends on thrust to mass ratio

Prepare for launch!

Stage	TWR (k)	Δv (atm)	Time
1	2.10	279 m/s	12.9 s
2	2.07	322 m/s	15.0 s
3	2.04	380 m/s	17.8 s

Total Δv : 981 m/s

Stage	TWR (k)	Δv (vac)	Time
4	4	198	3155 m/s
5	---	---	---

Stage	TWR (k)	Δv (vac)	Time
6	6	135	2523 m/s
7	---	---	---

Stage	TWR (k)	Δv (vac)	Time
8	8	115	2191 s

2 x Z-400 Rechargeable Battery

Mass = 20 kg
Capacity = 400 e

2 x OX-STAT Photovoltaic Panel

Mass = 5 kg
Electricity = 0.75 e/s

6 x 60° Liquid Fuel Booster

2-2-2 Asparagus Staging

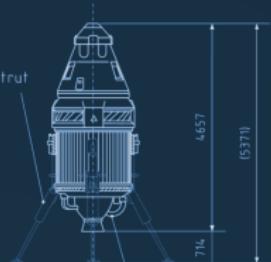
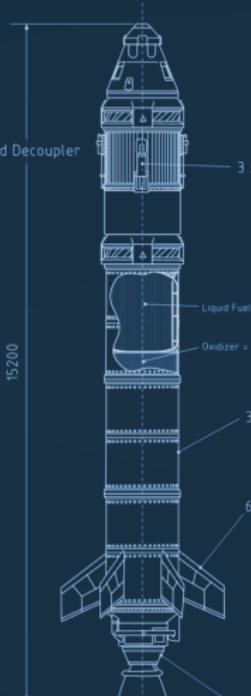
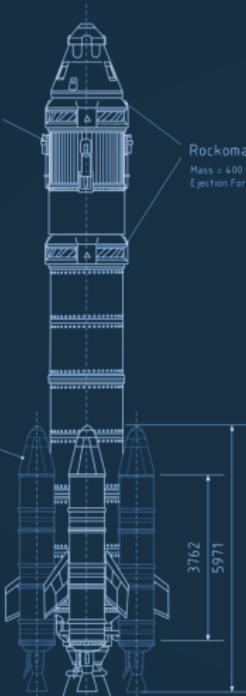
6 x NCS Adapter incl. Standard NC
Mass = 400 kg

18 x FL-T400 Liquid Fuel Tank
Mass = 250 kg
Liquid Fuel = 180 L
Oxidizer = 220 L

6 x LV-T45 Liquid Fuel Engine
Mass = 1500 kg
Thrust = 200 000 N
Isp (atm) = 320 s

6 x Hydraulic Detachment Manifold
Mass = 400 kg
Ejection Force = 100 N

Total Δv (atm) = 5 404 m/s
Total Δv (vac) = 6 826 m/s
Total Mass = 131 390 kg
Part Count = 78
Total Cost = 64 820 ₣



Rockomax „Poodle“ Liquid Engine
Mass = 2 500 kg
Thrust = 220 000 N
Isp (vac) = 390 s

Mk16-XL Parachute
Mass = 300 kg
Drag
- Stowed = 0.22
- Semi-Deployed = 1
- Deployed = 500

Mk1-2 Command Pod
Mass = 4 000 kg
Required Crew = 3

Staging



How do we do it

- Split the rocket in different parts (stages).
- Each stage carries its own fuel and engine, and can be separated from the rocket in sequence
- For instance: booster stage, landing stage, transfer stage, ...



Mk1-2 Command Pod
Mass = 4 000 kg
Required Crew = 3

Rockomax „Mainsail“ Liquid Engine
Mass = 6 000 kg
Thrust = 1 500 000 N
isp (atm) = 280 s
isp (vac) = 330 s

Total Δv (atm) = 5 404 m/s
Total Δv (vac) = 6 826 m/s
Total Mass = 131 390 kg
Part Count = 78
Total Cost = 64 820 ₣

Staging



Why do we do it

- Rocket efficiency is inversely proportional to its weight
- Delta-v goes up as total mass decreases, so we want to carry as little mass as possible
- Throw away excess weight of unused engines and empty fuel tanks

6 x 60° Liquid Fuel Tank
Mass = 200 kg
Volume = 600 l
Isp (atm) = 220 s
Isp (vac) = 240 s

6 x 18 x 60° Liquid Fuel Tank
Mass = 600 kg
Volume = 1800 l
Isp (atm) = 220 s
Isp (vac) = 240 s

6 x 18 x 120° Liquid Fuel Tank
Mass = 1200 kg
Volume = 1800 l
Isp (atm) = 220 s
Isp (vac) = 240 s

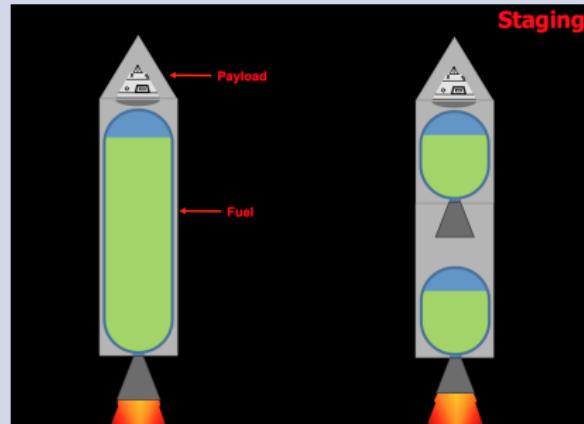
6 x Hydraulic Detachment Manifold
Mass = 400 kg
Ejection Force = 100 N



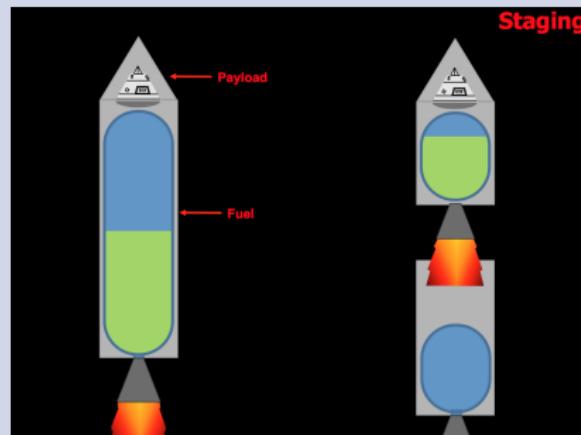
Rockomax „Mainsail“ Liquid Engine
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Isp (vac) = 330 s

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Staging

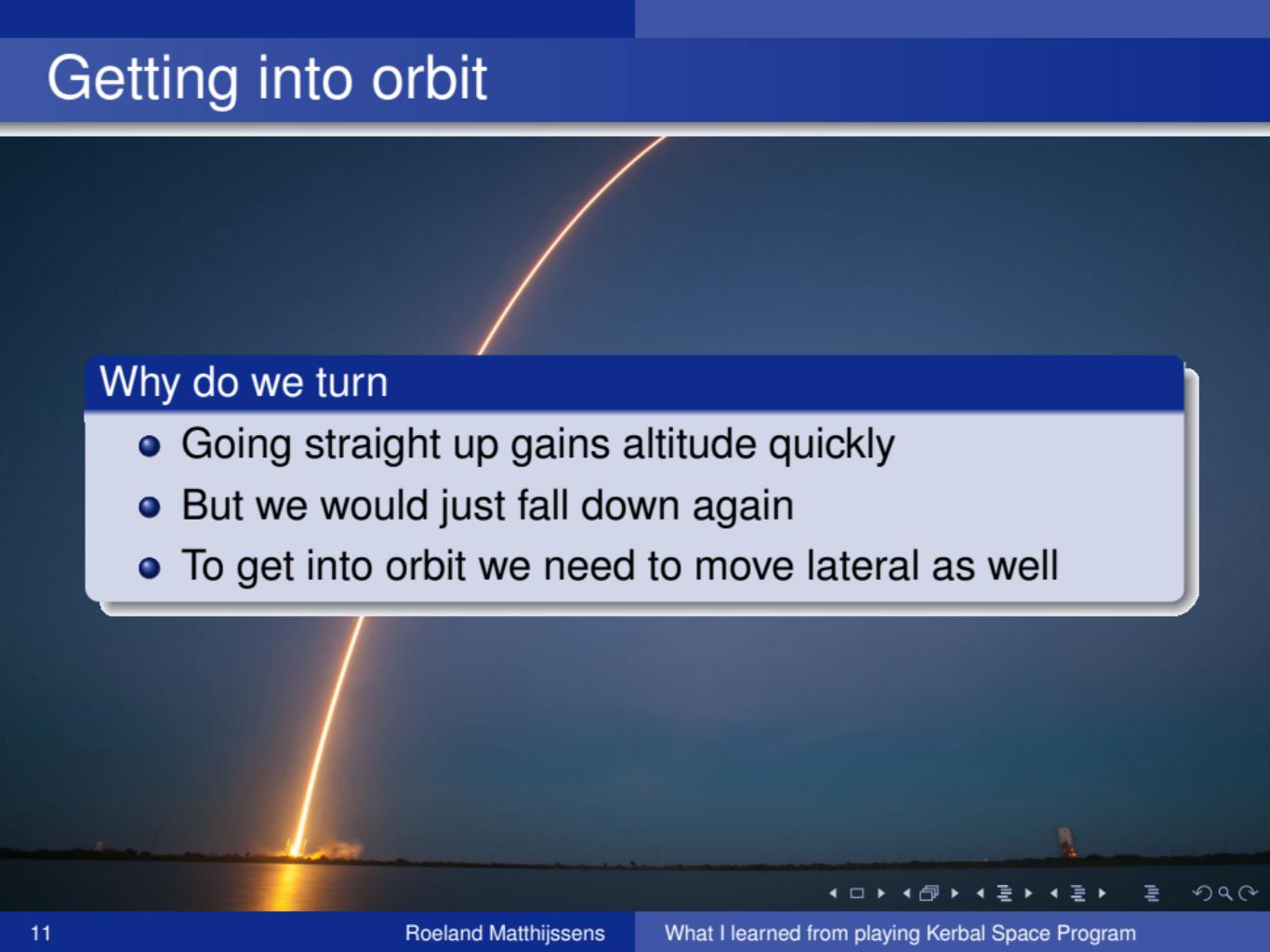


Staging



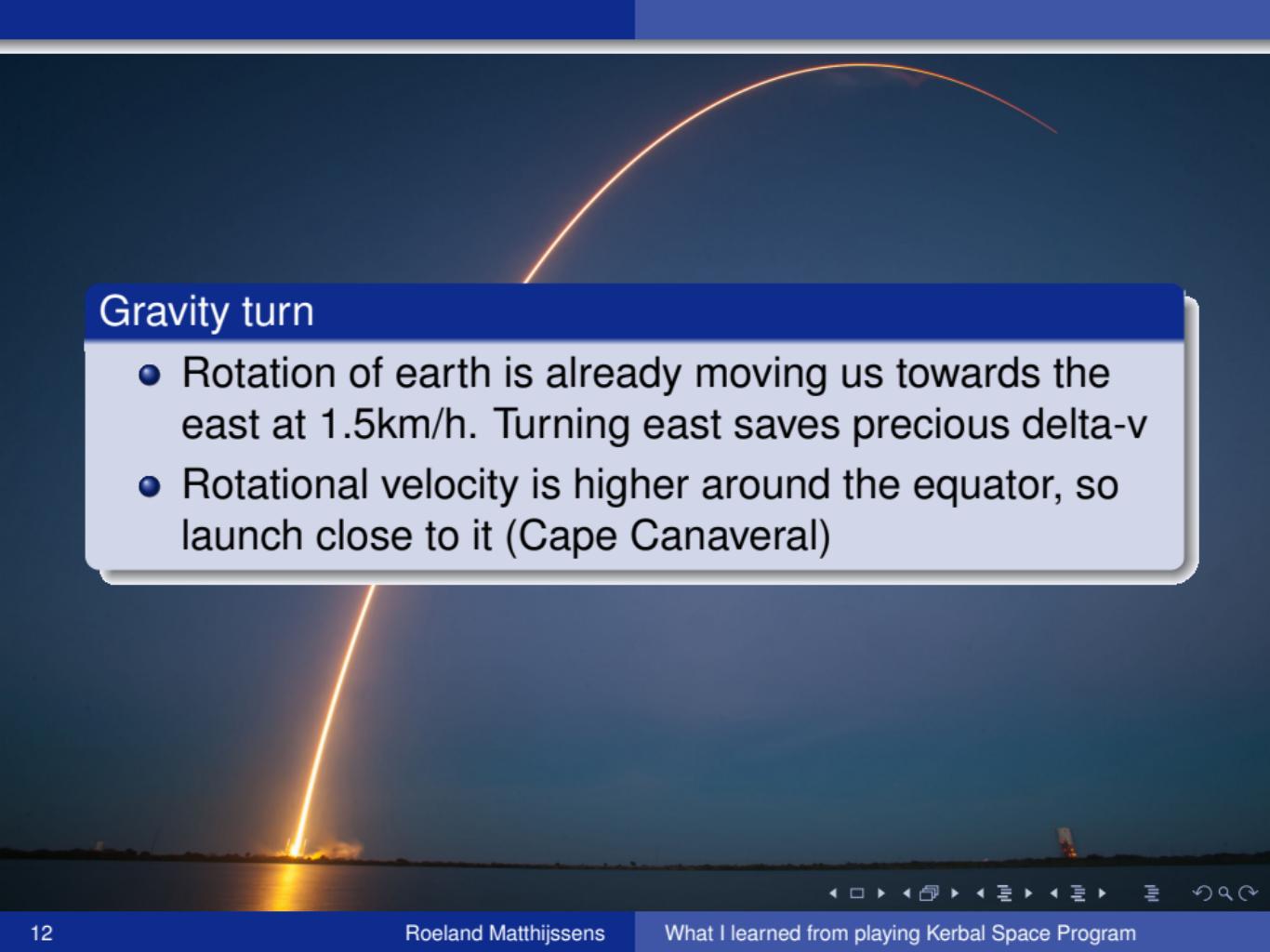


Getting into orbit



Why do we turn

- Going straight up gains altitude quickly
 - But we would just fall down again
 - To get into orbit we need to move lateral as well



Gravity turn

- Rotation of earth is already moving us towards the east at 1.5km/h. Turning east saves precious delta-v
- Rotational velocity is higher around the equator, so launch close to it (Cape Canaveral)

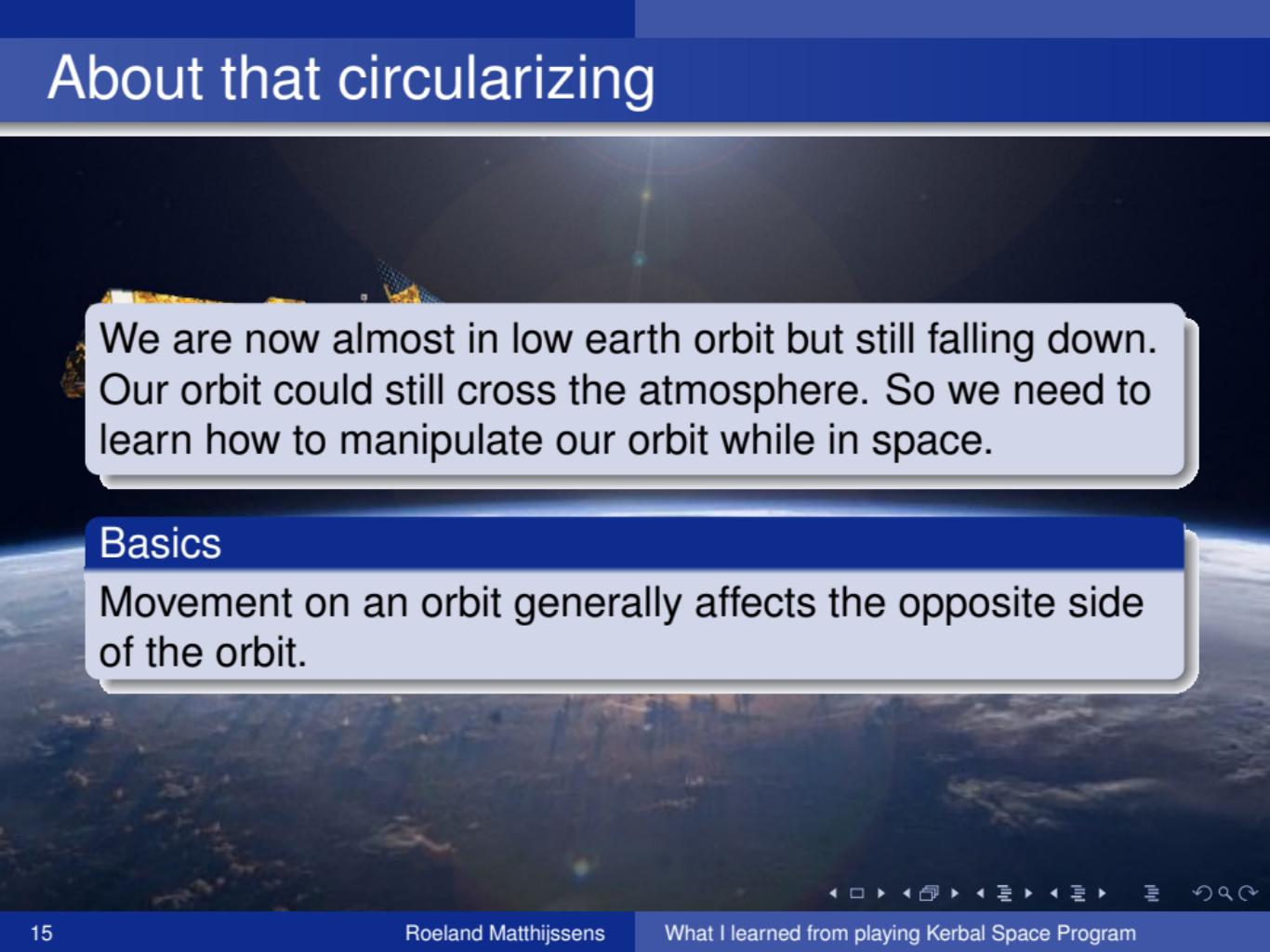
Let's Launch

How do we do it

- Point eastward (about 5 degrees)
- Fire rocket
- Don't. Touch. Anything. Let gravity do the work
- Activate staging at the appropriate times
- Circularize orbit
- It's easy, it's not ro... oh...



About that circularizing



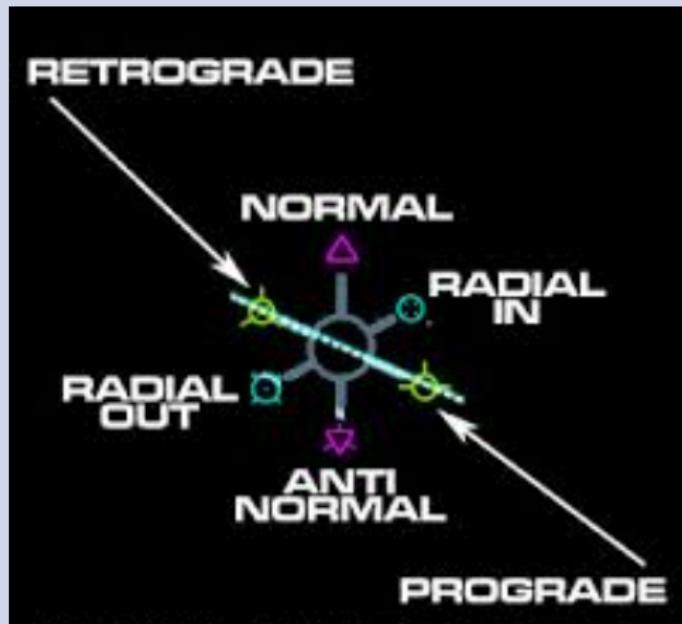
We are now almost in low earth orbit but still falling down. Our orbit could still cross the atmosphere. So we need to learn how to manipulate our orbit while in space.

Basics

Movement on an orbit generally affects the opposite side of the orbit.

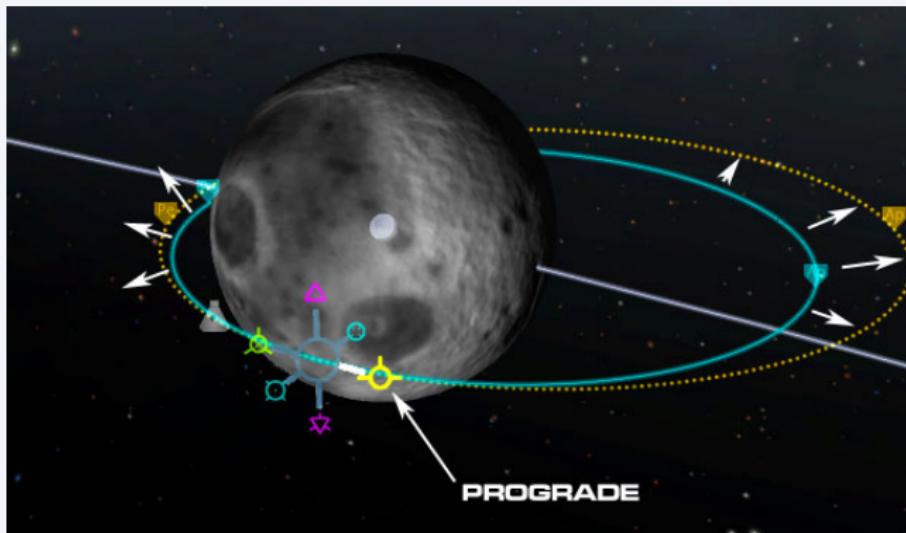
Adjusting orbits

At any point on the orbit you can burn in 3 directions:
up/down, left/right, forwards/backwards



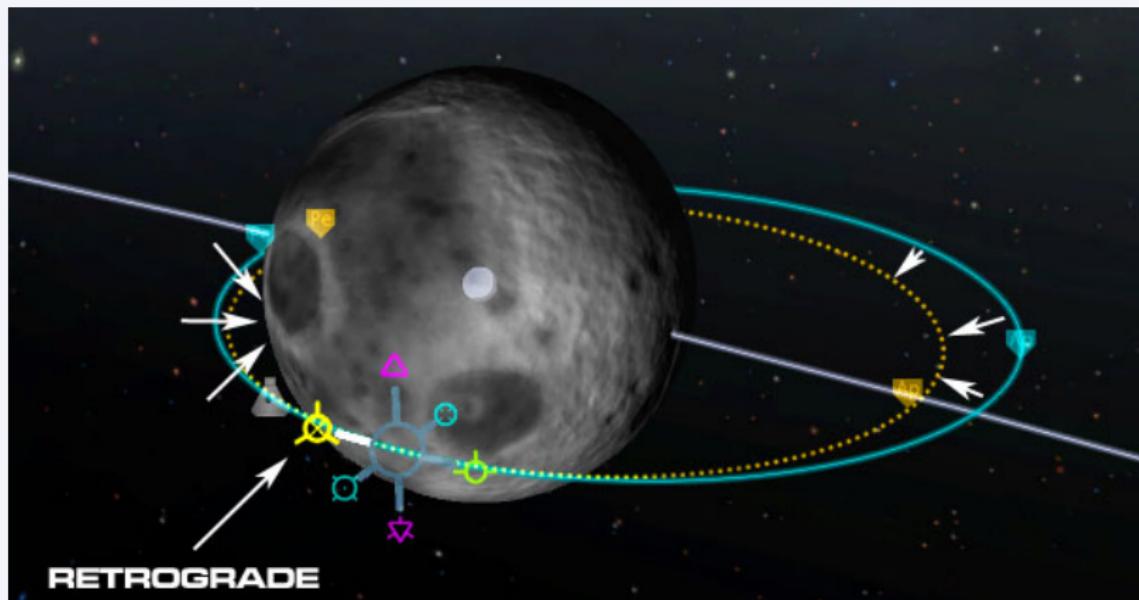
Prograde

Burning in the prograde direction expands the size of the orbit, with the center of expansion at the current position.



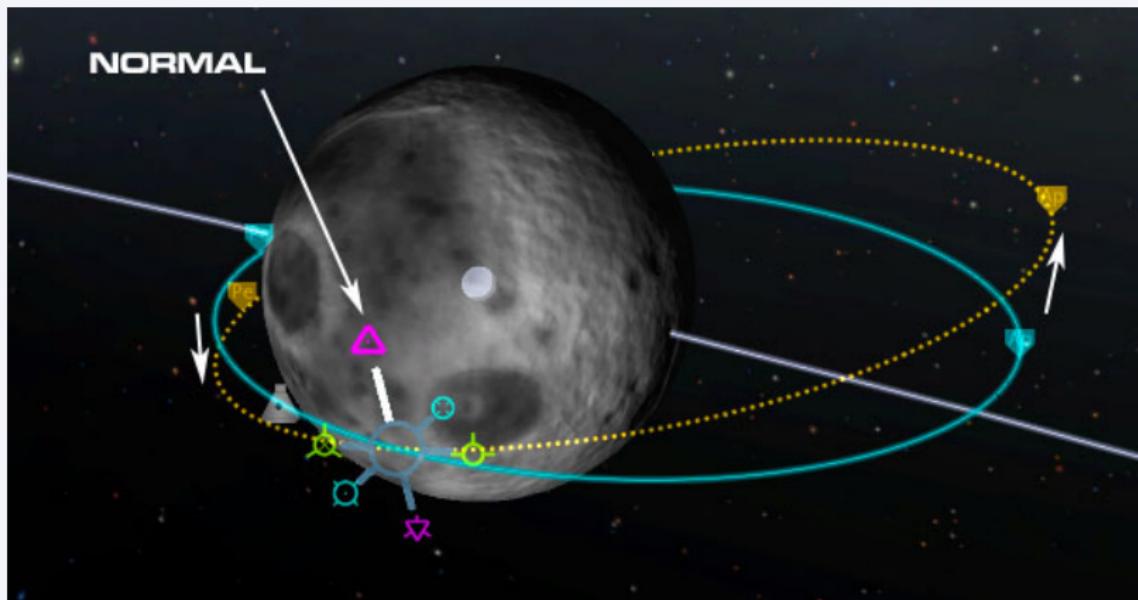
Retrograde

Burning in the retrograde direction contracts the size of the orbit, with the center of contraction at the current position.



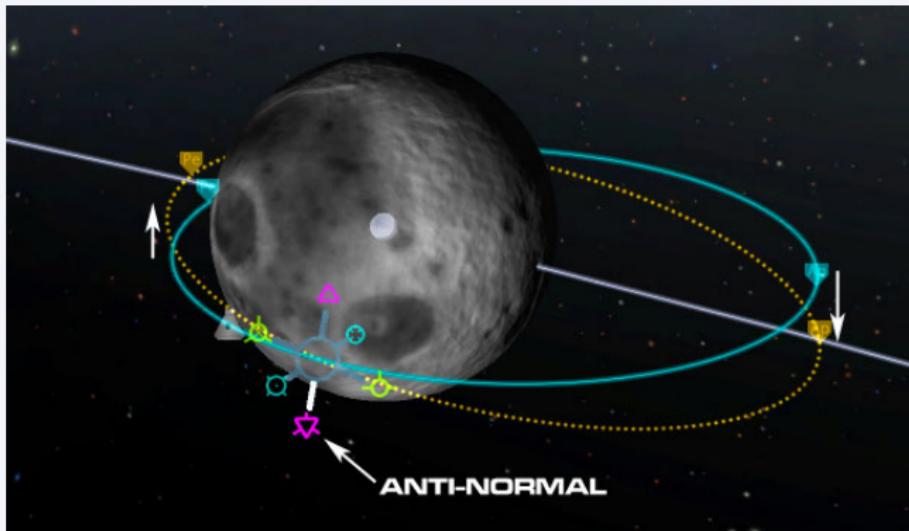
Normal

Burning in the normal direction tilts the orbital plane counterclockwise on the axis determined by the current position and the planet's center.



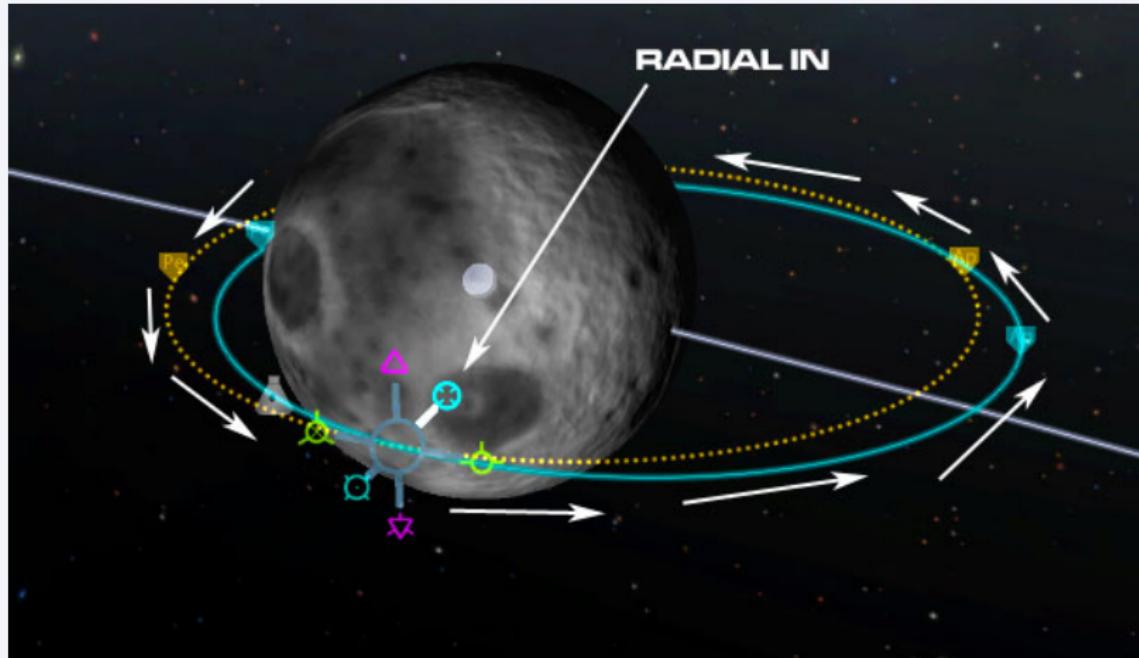
Anti-normal

Burning in the anti-normal direction tilts the orbital plane clockwise on the axis determined by the current position and the planet's center.



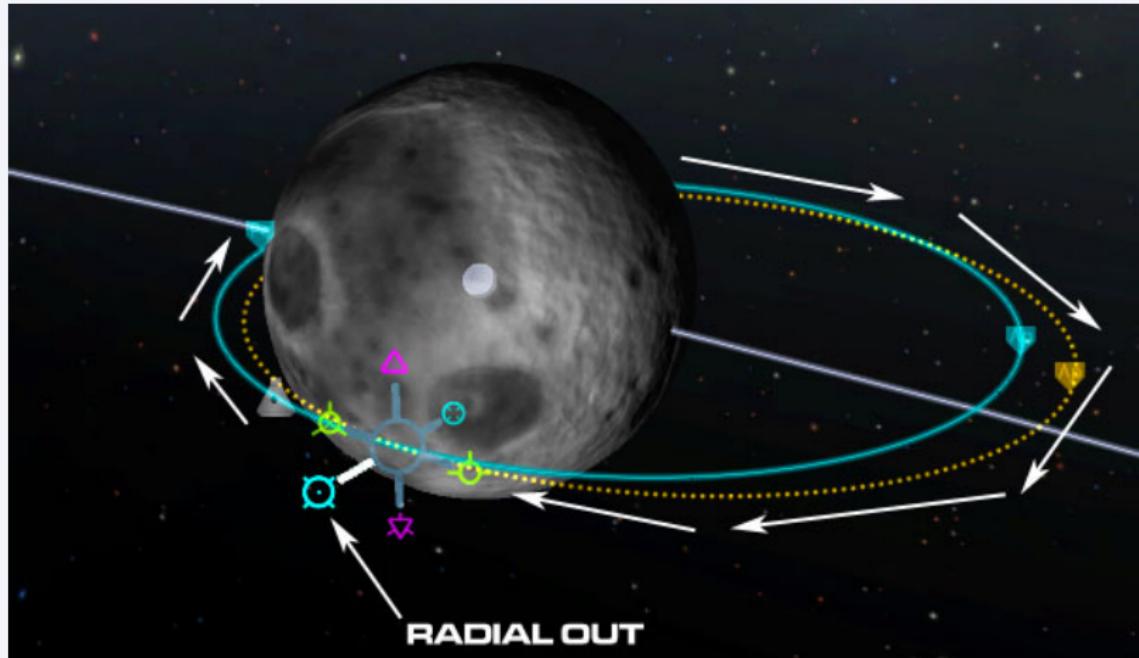
Radial in

Burning in the radial-in direction rotates the orbital plane counterclockwise.



Radial out

Burning in the radial-out direction rotates the orbital plane clockwise.



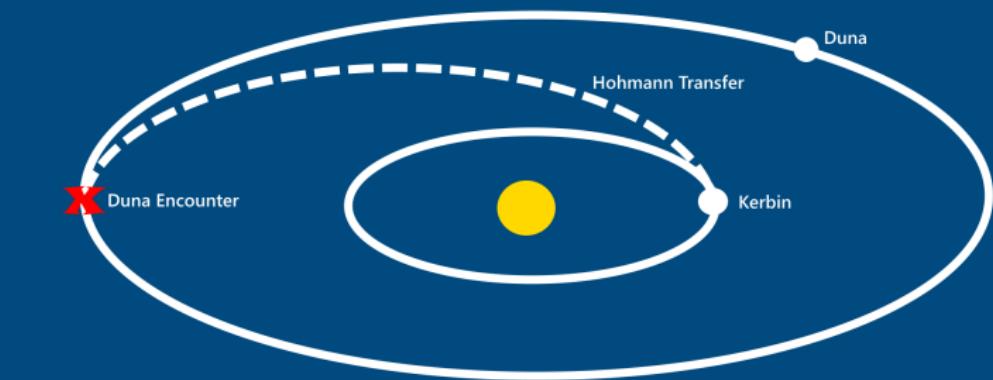
Planning transfer

We finally achieved a stable low earth orbit. We need to plan our transfer to our target body



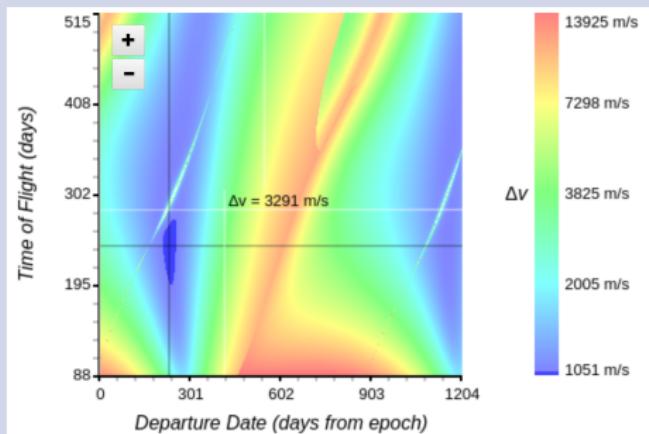
Hohmann transfer

Now that we know our target and know the delta-v required to get there we need to plan the encounter



Hohmann transfer window

To be able to transfer with the least amount of delta-v possible, we need to launch when the target body is at an optimal position. If we are on a strict delta-v budget we need to wait for a good window



Oberth effect

$$V = V_0 + at$$

$$X = X_0 + V_0 t + at^2/2$$

$$V^2 - V_0^2 = 2a(X - X_0)$$

$$\bar{V} = \frac{V_f + V_i}{2}$$

$$\Delta X = \bar{V} \Delta t$$

~~$$\vec{F}_{\text{tot}} = m \vec{a}$$~~

$$W = F d_{\parallel} = F_I d$$

$$W_{\text{tot}} = \Delta(\text{KE})$$

$$\Delta U = -W_{\text{if}}$$

$$\frac{1}{2} kx^2$$

$$\omega = \sqrt{\frac{k}{m}}$$

$$p = m v$$

$$\vec{P}_{\text{init}} = \vec{P}_{\text{final}}$$

$$\left(\sum_i m_i \vec{v}_j \right) = \left(\sum_i m_i \vec{v}_j \right)$$

$$v_x = v \cos(\theta)$$

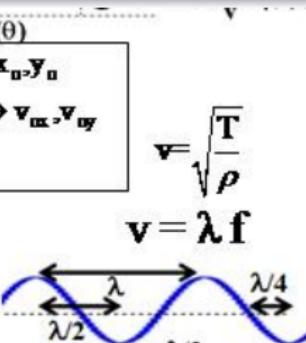
$$x \rightarrow x, y \quad x_0 \rightarrow x_0, y_0$$

$$v \rightarrow v_x, v_y \quad v_0 \rightarrow v_{0x}, v_{0y}$$

$$a \rightarrow a_x, a_y$$

$$\mu N$$

$$a = \frac{v^2}{R}$$



$$V = \omega r$$

$$a = \alpha r$$

$$\omega = \omega_0 + \alpha t$$

$$I = \sum_i m_i r_i^2$$

$$\theta = \theta_0 + \omega_0 t + \frac{1}{2} \alpha t^2$$

$$\omega^2 - \omega_0^2 = 2\alpha(\theta - \theta_0)$$

$$L = r_{\perp} p = m v r_{\perp}$$

$$L = I \omega$$

$$\tau = \frac{\Delta L}{\Delta t}$$

$$\tau = I \alpha$$

$$\frac{1}{2} I \omega^2$$

$$\sum_i \vec{F}_i = 0$$

$$\sum_i \vec{\tau}_i = 0$$

$$E = K + U$$

$$E_i = E_f$$

$$\frac{1}{2} m v^2$$

$$x = A \cos(\omega t) = \{ \text{or} \} A \sin(\omega t)$$

$$v = A \omega \sin(\omega t) = \{ \text{or} \} A \omega \cos(\omega t)$$

$$a = A \omega^2 \cos(\omega t) = \{ \text{or} \} -A \omega^2 \sin(\omega t)$$

$$\Delta Q = (\text{quant.}) C_{\text{cond.}} \Delta T$$

$$\Delta Q_{\text{intro}} = \Delta W_{\text{by}} + \Delta E$$

$$\frac{RT}{2} \mid_{\text{deg freedom}}$$

$$C_p = C_v + R$$

$$\Delta S \geq 0$$

$$\Delta Q = 1 \Delta (\text{quant.})$$

$$e = \frac{\Delta W}{\Delta Q}$$

$$e = 1 - \frac{T_L}{T_H}$$

$$P = \frac{F}{A}$$

$$PV = nRT$$

$$M = \rho V$$

$$P_1 = P_2$$

$$\Delta P = \rho g \Delta h$$

$$B = \rho_{\text{liqu}} V_{\text{disp}} g$$

$$A_1 v_1 = A_2 v_2$$

$$\frac{GM_e}{R_e} = g R_e$$

$$\frac{GMm}{r^2}$$

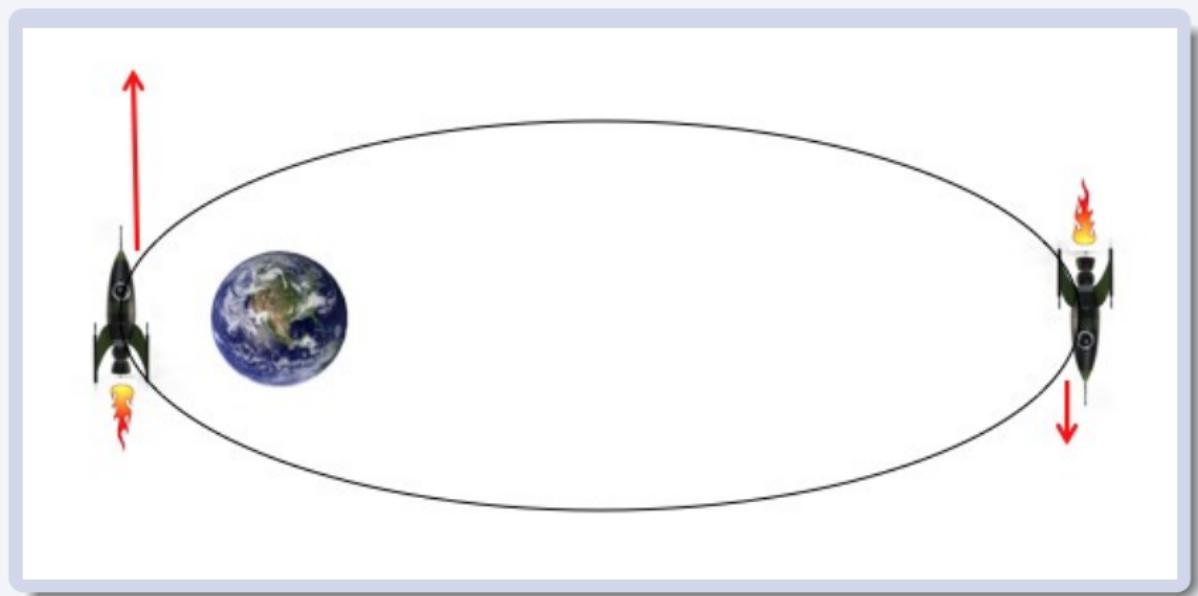
$$M_e = 5.97(10)^{24} \text{ Kg}$$

$$R_e = 6.37(10)^6 \text{ m}$$

$$\frac{GMm}{r}$$

$$p + \frac{1}{2} \rho v^2 = \text{const}$$

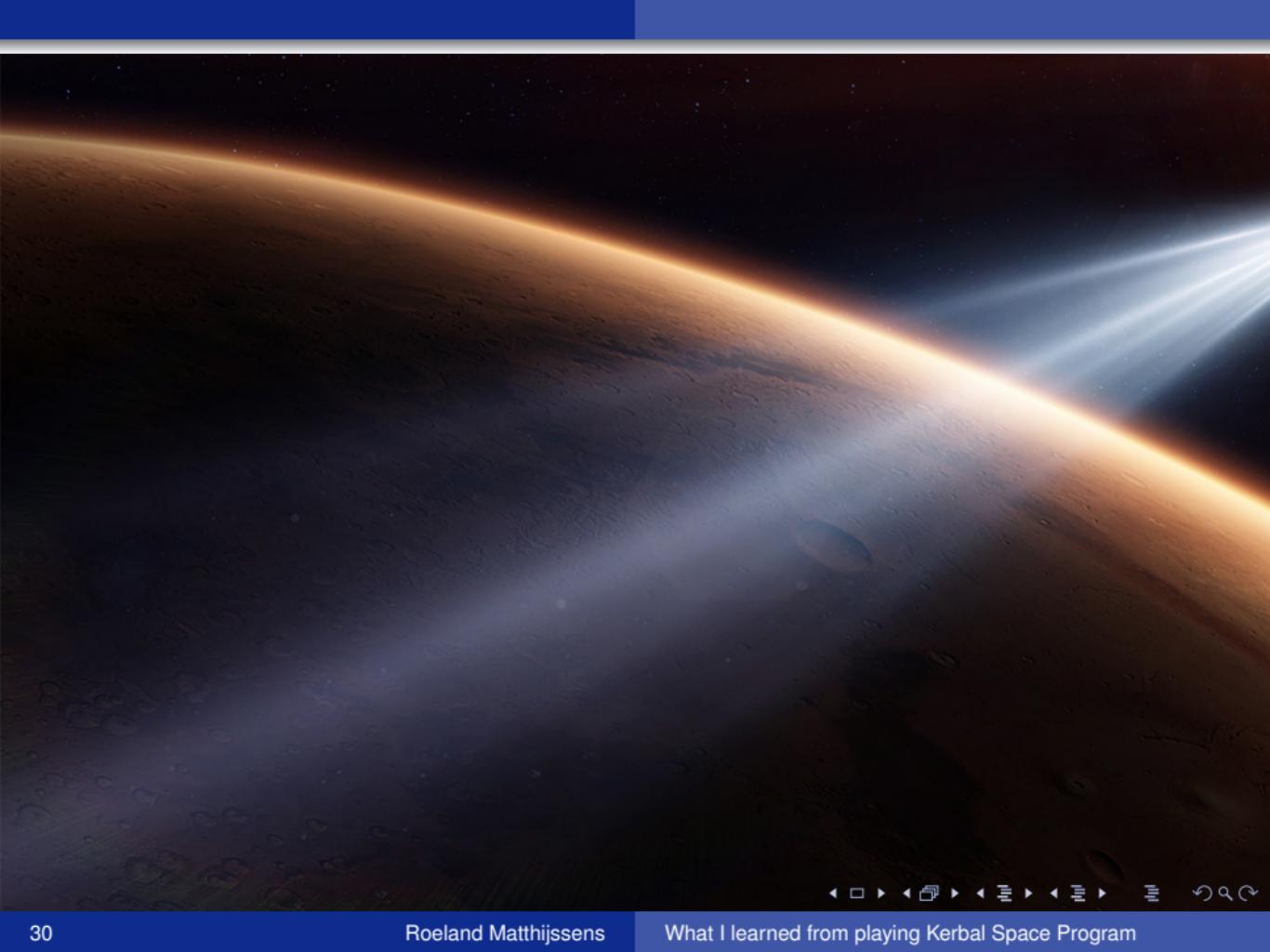
jk, Oberth effect



Escape

Now that we can make optimal use of the Oberth effect we can begin our escape burn and start our transfer orbit

- Decelerate to drop orbit deeper into the gravity well
 - Wait for lowest point of orbit (to make maximal use of Oberth effect)
 - Make transfer burn and coast to encounter



Sphere of Influence

Sphere of Influence(SOI) is the region around a planet in which the gravitational pull from the planet is larger than any other.

- We are in the SOI of our sun
- The moon is also influenced by the gravity of the sun, but it's close enough to earth to make its influence higher. So the moon is within the SOI of earth.

Intercept

When we are within the SOI of the target planet we need to slow down to be captured by the planet.

- Burn retrograde to slow down
- Keep slowing down until you fall into an highly elliptical orbit
- Wait for lowest point of orbit

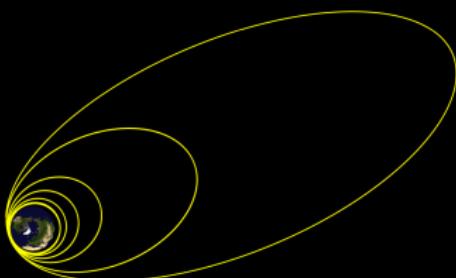
Intercept

Aerobreaking

Instead of having to circularize your orbit (which is expensive) we can make use of the body's atmosphere to slow down.

- Drop orbit just inside of the atmosphere
- Not too deep because you can burn up, not too shallow because it would take a very long time
- Wait a couple of orbits: you get closer every orbit because atmospheric drag slows you down

Aerobraking

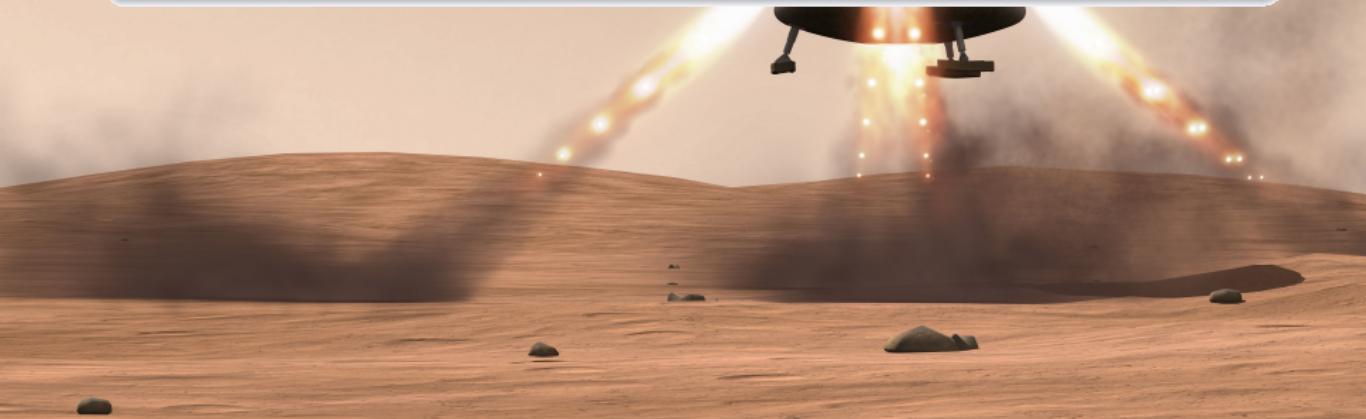




Landing

Determine entry angle

- Not to steep, because too much atmospheric drag means we'll burn up
- Not to shallow, because we won't slow down quickly enough and skip out of atmosphere



Landing

Suicide Burn

We need to slow down our descent before we hit the ground. At the last possible second we fire all engines at 100% throttle.

Why?

- Gravity is accelerating our fall, so decelerating earlier gives gravity more time to accelerate us again
- Slowing down quickly saves fuel.

