

# Introduction

- We will take you through successfully launching a rocket and landing it on a different body
- This is a simplified version based on our knowledge of KSP not real life per se

## Jargon

**Orbit** An object rotating around another object is said to be in orbit

**Inclined** An orbit that is tilted relative to the plane of the parent body

## delta-v

- aka: fuel budget
- how hard can i accelerate/decelerate my rocket
- dependent on:
  - fuel efficiency
  - thrust to weight ratio

# Technicalities

## Points on orbit

apoapsis : highest point on orbit

periapsis : Lowest point on orbit



Let's prepare for launch

1	2.10	278 m/s	12.6 %
2	2.07	322 m/s	15.6 %
3	2.04	380 m/s	17.6 %
Total $\Delta v$ : 981 m/s			

4	1.98	275 m/s	14.3 %
5	---	---	---

6	1.95	252 m/s	12.6 %
7	---	---	---

8	---	---	---
9	---	---	---

2 x Z-400 Rechargeable Battery  
Mass = 28 kg  
Capacity = 400 e

2 x OX-STAT Photovoltaic Panel  
Mass = 5 kg  
Electricity = 8.75 w/s

Rockomax Brand Decoupler  
Mass = 400 kg  
Ejection Force = 250

3 x 120° LT-2 Landing Strut  
Mass = 100 kg

Liquid Fuel = 1640 L

Decayzer = 1760 L

3 x Rockomax X200-32 Fuel Tank  
Mass = 2 300 kg  
Liquid Fuel = 1640 L  
Decayzer = 1760 L

6 x 60° AV-R8 Winglet  
Mass = 20 kg  
Lift = 0.6  
Deflection Range = 15°  
Surface Area = 0.95

Rockomax „Mainsail“ Liquid Engine  
Mass = 9 000 kg  
Thrust = 1 500 000 N  
Isp (atm) = 280 s  
Isp (vac) = 330 s

Rockomax „Poodle“ Liquid Engine  
Mass = 2 500 kg  
Thrust = 220 000 N  
Isp (vac) = 390 s

Mk16-XL Parachute  
Mass = 300 kg  
Drag  
- Stowed = 0.22  
- Semi-Deployed = 1  
- Deployed = 580

Mk1-2 Command Pod  
Mass = 4 000 kg  
Required Crew = 3

6 x 60° Liquid Fuel Booster  
2-2-2 Asparagus Staging  
6 x RCS Adapter incl. Standard RCS  
Mass = 400 kg

18 x FL-T400 Liquid Fuel Tank  
Mass = 250 kg  
Liquid Fuel = 360 L  
Decayzer = 220 L

6 x LV-T45 Liquid Fuel Engine  
Mass = 1 500 kg  
Thrust = 200 000 N  
Isp (atm) = 320 s

6 x Hydraulic Detachment Manifold  
Mass = 400 kg  
Ejection Force = 80 N

Total  $\Delta v$  (atm) = 5 404 m/s  
Total  $\Delta v$  (vac) = 6 826 m/s  
Total Mass = 131 390 kg  
Part Count = 78  
Total Cost = 64 820 ¢

## KERBAL X

The Kerbal X is one of the most successful rockets that can be ordered from a catalog. Despite the original design having been meant for a plastic model, it's proved itself quite dependable as a full-sized craft.

The X is capable of achieving orbit around Kerbin, and even features a very optimistic set of landing legs on its upper stage.

# Staging

2 x Z-400 Rechargeable Battery

Mass = 28 kg  
Capacity = 400 e

2 x OX-STAT Photovoltaic Panel

Mass = 5 kg  
Electricity = 0.75 w/s

Rockomax Brand Decoupler

Mass = 400 kg  
Explosion Force = 250

3 x 120° LT-Z Landing Strut

Mass = 100 kg

4457

(5371)

11500

## How do we do it

- Split the rocket in 2 or more parts.
- Each part carries own fuel and engine
- Each stage can be separated from the rocket in sequence
- e.g. booster stage, landing stage, transfer stage, ...

Total  $\Delta v$  (atm) = 5 404 m/s  
Total  $\Delta v$  (vac) = 6 826 m/s  
Total Mass = 131 390 kg  
Part Count = 78  
Total Cost = 64 820 €

Mass = 1 000 kg  
Thrust = 1500 000 N  
Isp (atm) = 280 s  
Isp (vac) = 330 s

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## Staging

Technical drawing of a crane hook. The drawing shows a cross-section of the hook with various dimensions labeled. The top part of the hook is labeled with '100' and '100'. The middle part of the hook is labeled with '100' and '100'. The bottom part of the hook is labeled with '100' and '100'. The drawing is a technical illustration of a crane hook, showing its internal structure and dimensions.

## Why do we do it

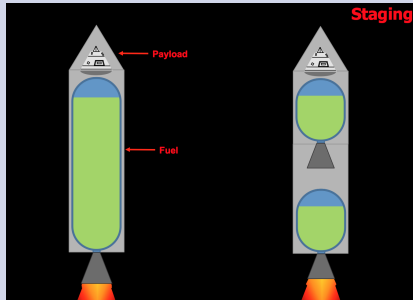
- Rocket efficiency is inversely proportional to its weight
- Delta-v goes up as total mass decreases (All else being equal)
- We want to carry as little mass as possible
- Throw away excess weight of unused engines and empty fuel tanks

- 6 x 60\* Liquid  
2-2-2  
6 x MCS  
18 x FL-  
6 x LV-  
6 x Hydr

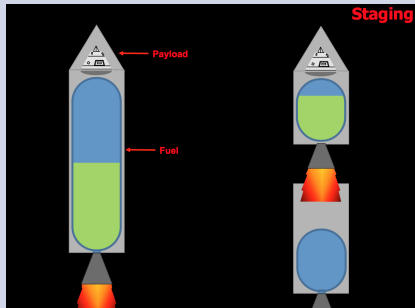
The X is capable of achieving orbit around Kerbin, and even features a very optimistic set of landing legs on its upper stage.



# Staging



# Staging





# Getting into orbit

## Why do we turn

- Going straight up gains altitude quickly
- But we would just fall down again
- To get into orbit we need to move lateral as well
- combining vectors is efficient:  $a^2 + b^2 = c^2$

A photograph of a rocket launch at night. A bright orange plume of fire and smoke is visible at the base of the rocket, which is launching from a coastal area. A large, curved orange arc is visible in the dark blue sky above the launch site. The background shows a body of water and a distant shoreline with some lights.

## Gravity turn

- Rotation of earth is already moving us towards the east at 1.5km/h. Turning east saves precious delta-v
- rotational velocity is higher around the equator that why we want to launch from Cape Canaveral

# Let's Launch

## How do we do it

- throttle up
- point eastwards (about 5degrees)
- don't. touch. anything. let gravity do it's work
- activate staging at the appropriate times
- keep apoapsis in front of you untill desired height
- circularize orbit
- It easy, it's not ro... oh...



# About that circularizing

We are now almost in low earth orbit but still falling down. Our periapsis may still be within the atmosphere. So we need to learn how to manipulate our orbit while in space.

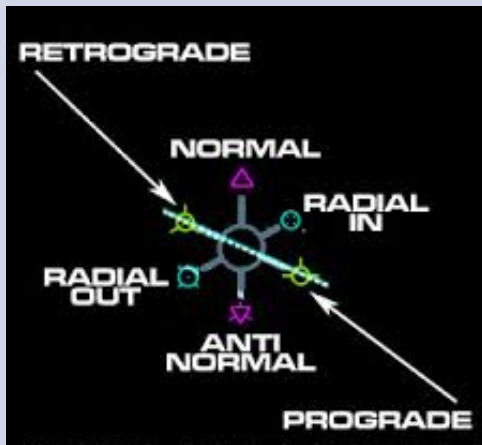
## Basics

Movement on an orbit generally affects the opposite side of the orbit.



# Adjusting Orbits

At any point on the orbit you can burn in 3 directions.  
up/down, left/right, forwards/backwards

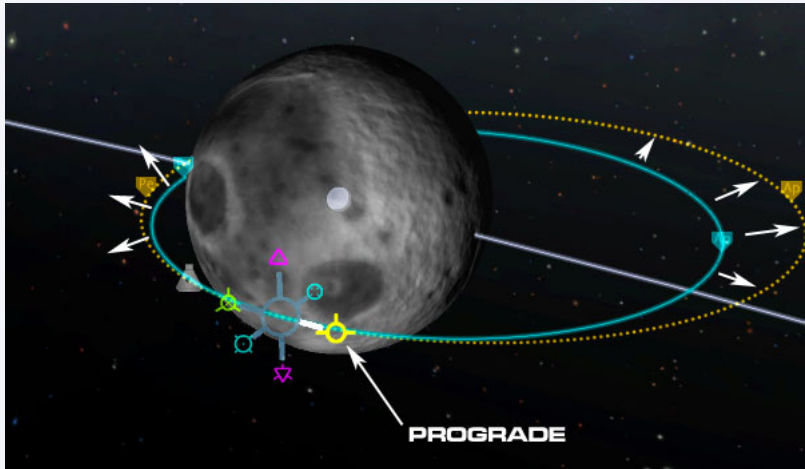


# Low Orbit

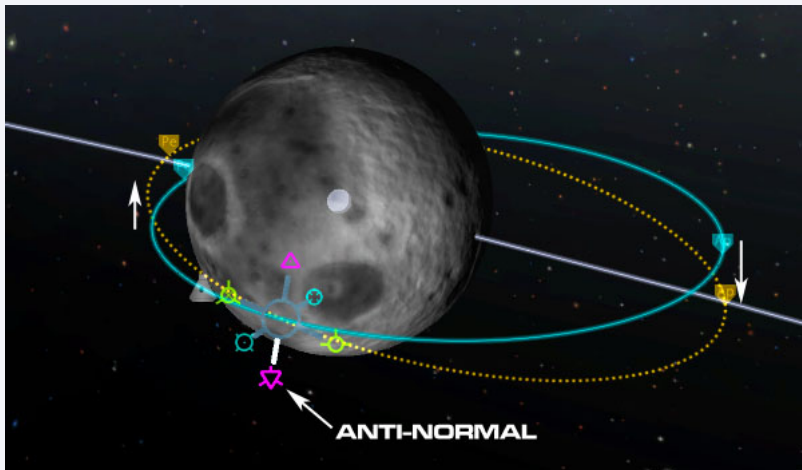
## Maveuvering in space

- prograde** Along your movement vector, used to increase orbit altitude
- retrograde** Oposite your movement vector, used to decrease orbit altitude
- normal** Perpendicular to orbit, used to increase/decrease inclination
- anti-normal** perpendicular to orbit, used to increase/decrease inclination
- radial** Towards parent body, Used to shift orbit around
- anti-radial** Away from parent body, used to shift orbit around

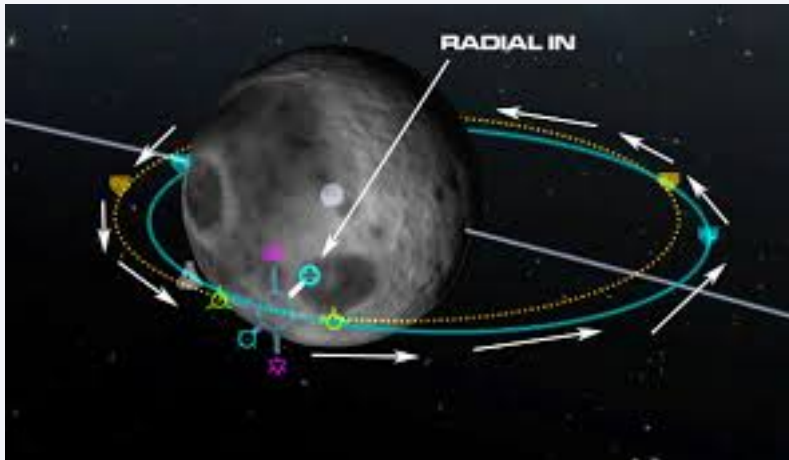
# Prograde



# Normal



# Radial



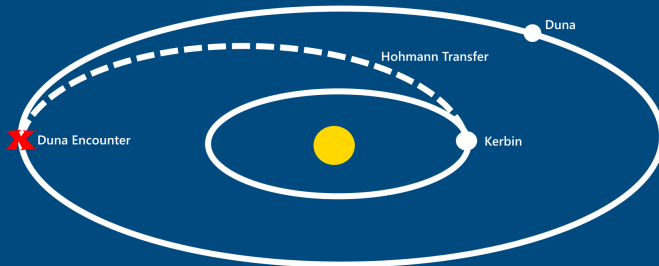
# Planning Transfe

Finally achieved a stable low earth orbit. We need to plan our tranfer to our target body



# Hohmann transfer

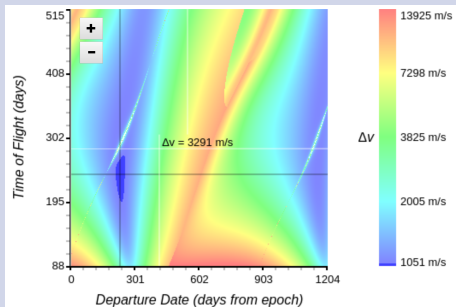
Now that we now our target and know the delta-v required to get there we need to plan the encounter





# Hohmann transfer window

To be able to transfer with the least amount of delta-v as possible we need to launch when the target body is at an exact position relative to us. If we are on a strict delta-v budget we need to wait for a good window



# Oberth Effect

$$\begin{aligned} \mathbf{v} &= \mathbf{v}_0 + \mathbf{a}t \\ \mathbf{x} &= \mathbf{x}_0 + \mathbf{v}_0 t + \mathbf{a}t^2/2 \\ v^2 - v_0^2 &= 2\mathbf{a}(\mathbf{x} - \mathbf{x}_0) \\ \bar{\mathbf{v}} &= \frac{\mathbf{v}_f + \mathbf{v}_i}{2} \end{aligned}$$

$$\Delta \mathbf{x} = \bar{\mathbf{v}} \Delta t$$

$$\vec{F}_{\text{tot}} = m \vec{a}$$



$$\begin{aligned} W &= F d_{\parallel} = F_{\parallel} d \\ W_{\text{tot}} &= \Delta(\text{KE}) \end{aligned}$$

$$\Delta U = -W_{\text{if}}$$

$$\frac{1}{2} kx^2 \quad \omega = \sqrt{\frac{k}{m}}$$

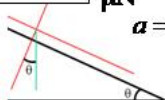
$$p = m v$$

$$\vec{P}_{\text{init}} = \vec{P}_{\text{final}}$$

$$\left( \sum_i m_i \vec{v}_i \right) = \left( \sum_i m_i \vec{v}_i \right)$$

$$\begin{aligned} v_x &= v \cos(\theta) \\ \mathbf{x} &\rightarrow x, y \quad \mathbf{x}_0 \rightarrow x_0, y_0 \\ \mathbf{v} &\rightarrow v_x, v_y \quad \mathbf{v}_0 \rightarrow v_{0x}, v_{0y} \\ \mathbf{a} &\rightarrow a_x, a_y \end{aligned}$$

$$\mu N \quad a = \frac{v^2}{R}$$



$$E = K + U$$

$$E_i = E_f$$

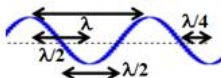
$$\frac{1}{2} m v^2$$

$$\begin{aligned} x &= A \cos(\omega t) \text{ [or] } A \sin(\omega t) \\ v &= A \omega \sin(\omega t) \text{ [or] } A \omega \cos(\omega t) \\ a &= A \omega^2 \cos(\omega t) \text{ [or] } -A \omega^2 \sin(\omega t) \end{aligned}$$

$$\begin{aligned} M_e &= 5.97(10)^{24} \text{ Kg} \\ R_e &= 6.37(10)^6 \text{ m} \end{aligned}$$

$$v = \sqrt{\frac{T}{\rho}}$$

$$v = \lambda f$$



$$\begin{aligned} v &= \omega r \\ a &= \alpha r \\ \omega &= \omega_0 + \alpha t \\ I &= \sum_i m_i r_i^2 \quad \theta = \theta_0 + \omega_0 t + \frac{1}{2} \alpha t^2 \\ \omega^2 - \omega_0^2 &= 2\alpha(\theta - \theta_0) \\ L &= \mathbf{r}_{\perp} p = m \mathbf{v} r_{\perp} \quad \tau = r_{\perp} F = r F_{\perp} \\ L &= I \omega \quad \tau = \frac{\Delta L}{\Delta t} \\ \frac{1}{2} I \omega^2 &= \tau I \alpha \\ \sum_i \vec{F}_i &= 0 \quad \sum_i \vec{\tau}_i = 0 \end{aligned}$$

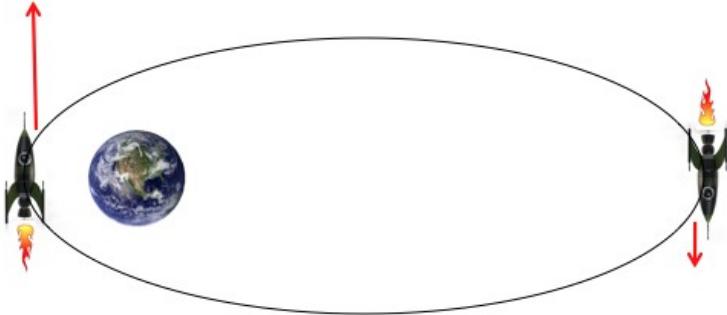
$$\begin{aligned} \Delta Q &= (\text{quant.}) C_{\text{cond}} \Delta T \quad \Delta S \geq 0 \\ \Delta Q_{\text{into}} &= \Delta W_{\text{by}} + \Delta E \\ \frac{RT}{2} &|_{\text{deg freedom}} \quad C_p = C_v + R \\ \Delta Q &= l \Delta(\text{quant.}) \quad PV = nRT \\ e &= \frac{\Delta W}{\Delta Q} \quad e = 1 - \frac{T_L}{T_H} \quad P = \frac{F}{A} \end{aligned}$$

$$\begin{aligned} M &= \rho V \quad P_1 = P_2 \\ \Delta P &= \rho g \Delta h \\ B &= \rho_{\text{liq}} V_{\text{disp}} g \end{aligned}$$

$$\frac{GMm}{r^2}$$

$$A_1 v_1 = A_2 v_2 \quad p + \frac{1}{2} \rho v^2 = \text{const}$$

# JK, Oberth Effect



# Oberth Effect explained

- things that go fast have higher energy

- $KE = 0.5 * m * v^2$

- mass is a form of energy

- $E = MC^2$

- Newton's third law

- $F_a = -F_b$

Dumping mass (fuel) while going fast(Deep in the gravity well) gives more kinetic energy to the ship

# Escape

Now that we can make optimal use of the Oberth effect we can begin our escape burn and start our transfer orbit

- decelerate to drop periapsis deeper into the gravity well
- wait for periapsis (to make maximal use of Oberth effect)
- make transfer burn and coast to encounter

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