



# The Data Incubator Capstone Project



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#### Introduction

- 1. Airfoil self noise is caused due to interaction of smooth non-turbulent flow with airfoil edges near wake.
- 2. At high speeds, noise can increase vibrations on the airplane wing and can cause the wing to break.
- 3. Reducing airfoil self-noise is an important problem in the aerospace industry.

How to predict airfoil self noise? Cross section \*Chord Length Leading Edge Relative Wind ower Camber - Chord

#### The Data Set

Noise measurements from different sizes of NACA 0012 airfoils placed at various wind tunnel speeds(v) and angles of attack( $\theta$ ).

(http://archive.ics.uci.edu/ml/datasets/Airfoil+Self-Noise#)

Number of measurements = 1503, Number of variables = 6 (Continuous)

Independent Variables (X)	Dependent Variable (Y)
Frequency (f) [Hz]	
Angle of Attack (θ) [deg]	
Chord Length (L) [m]	Scaled sound pressure level (P) [dB]
Free stream velocity (v) [m/s]	
Suction side displacement Thickness (t) [m]	Aim: Predict P

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# Exploratory Data Analysis

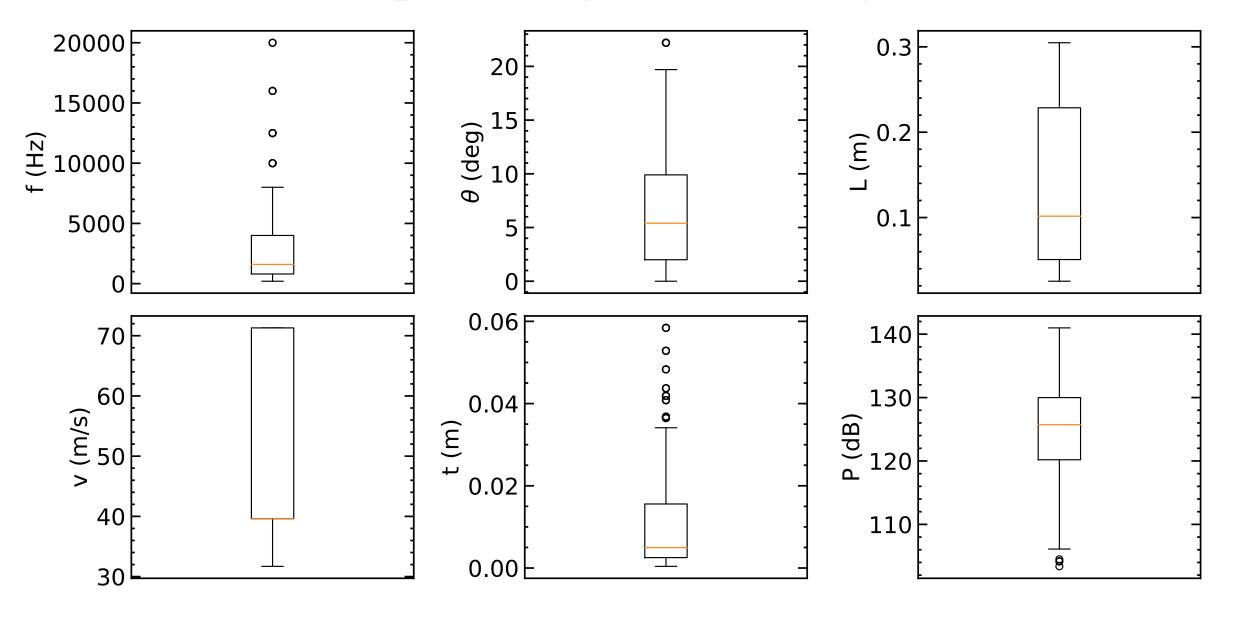
#### 1. Check for null values

2. Outlier Removal  $\Rightarrow$  For a variable q:

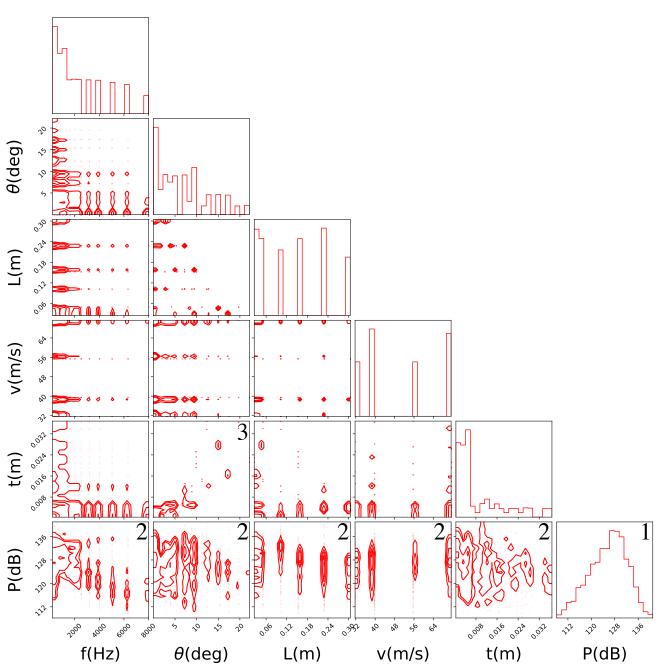
$$Outlier(q)_{Lower}$$
,  $Outlier(q)_{Upper} = q^{25th} - 1.5IQR$ ,  $q^{75th} + 1.5IQR$ 

 $q^{25th} = 25^{th}$  percentile of q distribution  $q^{25th} = 75^{th}$  percentile of q distribution IQR = Interquartile range

# Exploratory Data Analysis

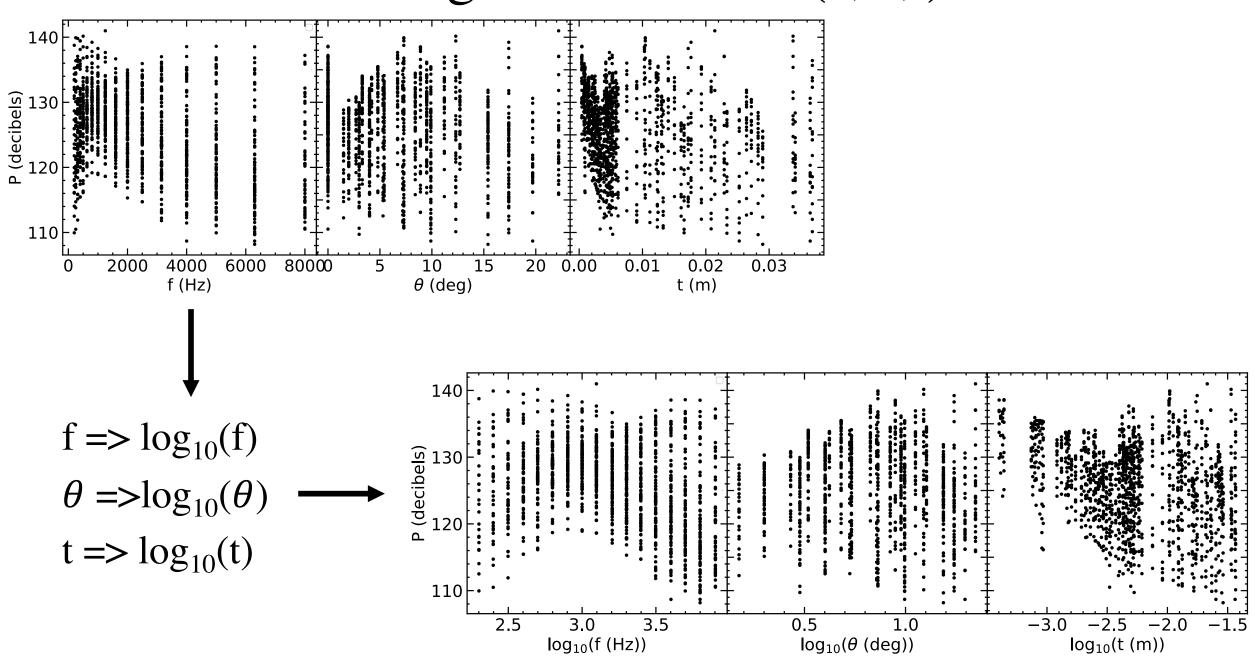


### Exploratory Data Analysis

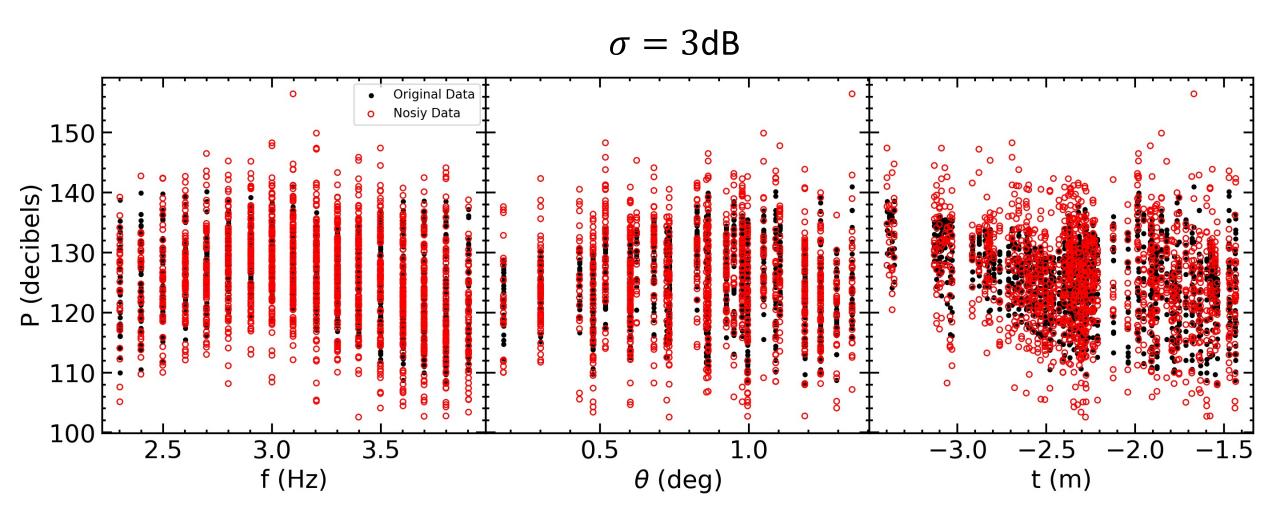


- 1. Distribution of *P* is approximately Gaussian.
- 2. *P* varies **non-linearly** w.r.t f, θ, L, v & t => **non-linear regression required**
- 3. Signs of multi-collinearity => regularization

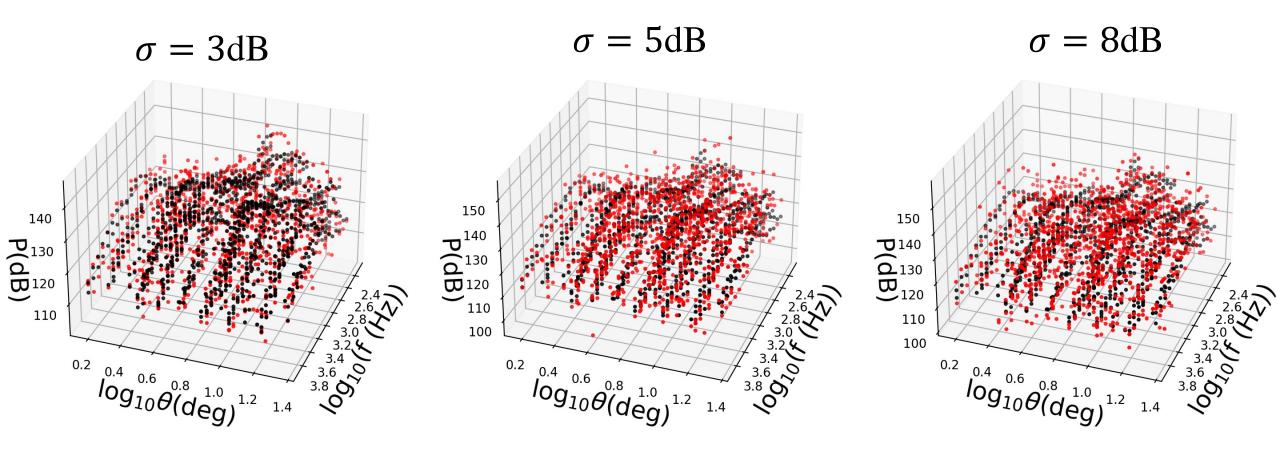
# Change of variables $(f,\theta,t)$



# Adding Gaussian Noise to dependent variable, P $P_{\text{noisy}} = P + N(0, \sigma); \ \sigma = Noise in pressure in dB$



#### 3D view of added noise



#### Future work

⇒ Predict P using different techniques like non-linear regression (OLS, SVR, Neural Networks etc) and compare the accuracies.

⇒ Use regularization techniques (Lasso/Ridge) to constrain the parameters of correlated independent variables

⇒ Perform the above two for different levels of added noise to P. Relate the accuracy of the selected model.

Thank you....

Questions?