

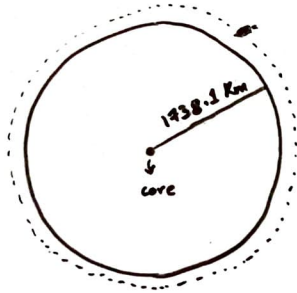
Wireless Power Transmission

⇒ Geosynchronous Orbit Calculation:

- F_1 persists for the complete Moon Orbital Time Period $T_m = 27.322$ days

Hence Rotation $\gamma_0 = T_m$ [Due to earth gravitation the F_1 is tidally locked]

* Gravitational force $g_m = 1.62 \text{ m/s}^2$; * Mass $M_m = 7.35 \times 10^{22} \text{ Kg}$



Radius $R_m = 1,738.1 \text{ Km}$ [Equatorial Radius]

Rotational Time Period $\Rightarrow \gamma_m = 27.3 \text{ days} = 655.2 \text{ hours}$

Semi-Major Axis $\Rightarrow a_m = 3.844 \times 10^8 \text{ m}$

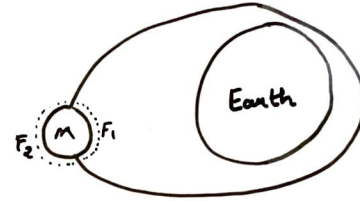
Universal Gravitational Constant = $G = 6.67 \times 10^{-11} \text{ N} \cdot \text{m}^2 / \text{Kg}^2$

$$F_g = m a_c \Rightarrow [a_c = r \omega^2] \Rightarrow F_g = m r \omega^2 \quad [\omega = \text{Angular Velocity of Cusat}]$$

$$\Rightarrow \omega = \frac{\text{change in Angular Position}}{\text{change in Time}} = \frac{\Delta \theta}{\Delta t} = \frac{2\pi}{T_m} = \frac{2\pi}{655.2 \times (3600)} = 2.663811435 \times 10^{-6} \text{ rad/sec}$$

$$\therefore \frac{G M_m m_s}{r^2} = m_s r \omega^2 \Rightarrow r = \sqrt[3]{\frac{G M_m}{\omega^2}} = \sqrt[3]{\frac{6.67 \times 10^{-11} \times 7.35 \times 10^{22}}{2.663811435 \times 10^{-6}}} = 88403353.2 \text{ m} = 88403.3532 \text{ Km}$$

$$\text{Altitude} = r - R_m = 88403.3532 - 1738.1 = \underline{\underline{86,665.2532 \text{ Km}}}$$

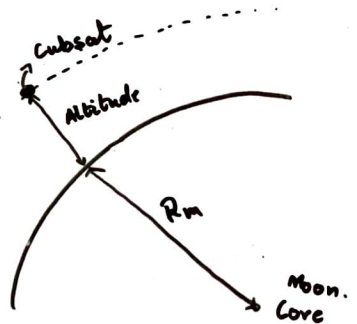


$F_1 = \text{near View}$

$F_2 = \text{Far View.}$

$F_1 \Rightarrow \text{Visible face for total time period } T_m$

$F_1 \Rightarrow \infty \text{ face visibility.}$



[$m_s = \text{Cusat mass}$]