

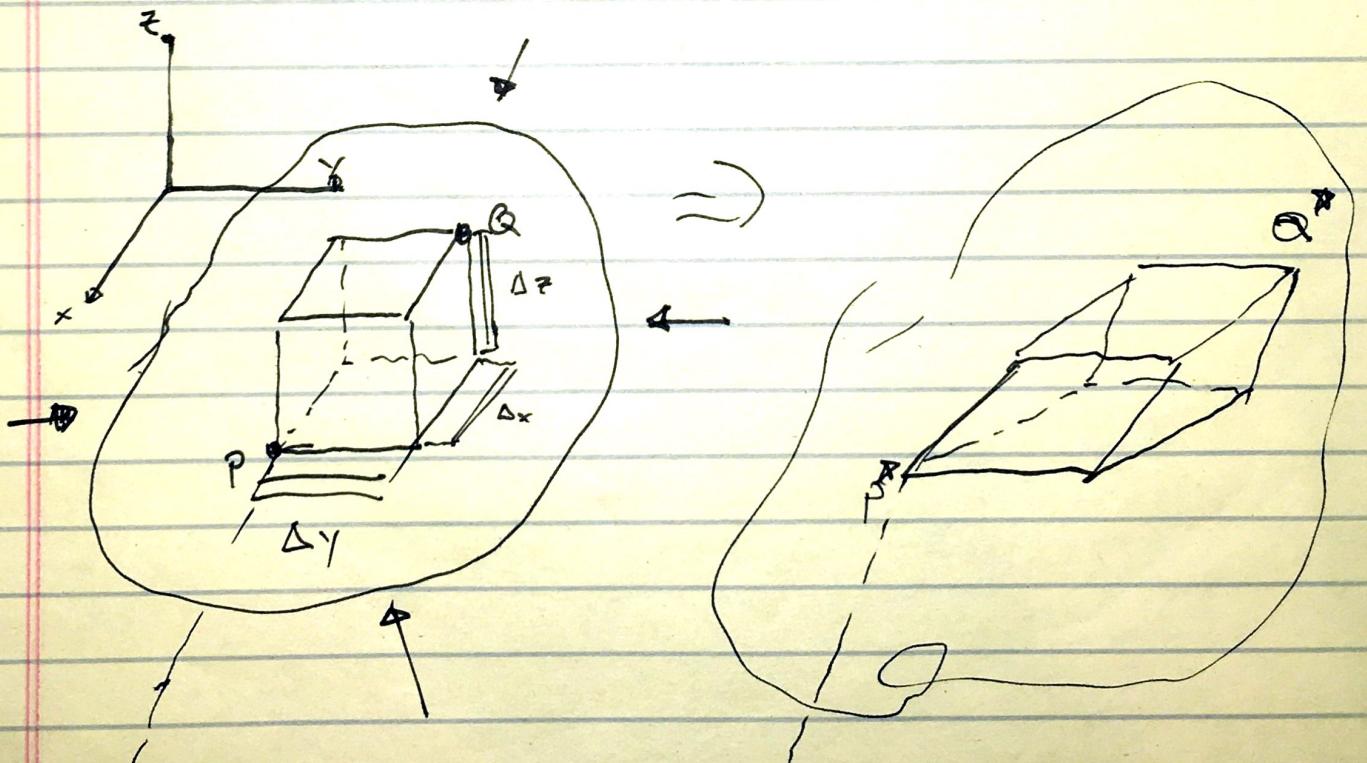
# MACBOOK CHARGER

## Lecture #3: Strain & Deformation

- Strain is purely geometric w/ no dependence on body material from deformation

### Displacement:

- Rigid Body Displacement: No Deformation
- Deformation  $\rightarrow$  strain: depends on relative motion in body
- Deformation can be broken into
- Shear:  $\underline{\underline{E}} \rightarrow \underline{\underline{E}}_0 \angle \theta^* \gamma = \theta(\pi/2 - \theta)$
- ~~Linear~~ Normal:  $\epsilon = \Delta l / l_0$



$$\vec{P} - \vec{P}^* = (u, v, w) \quad ; \quad \vec{Q} - \vec{Q}^* = (u^*, v^*, w^*)$$

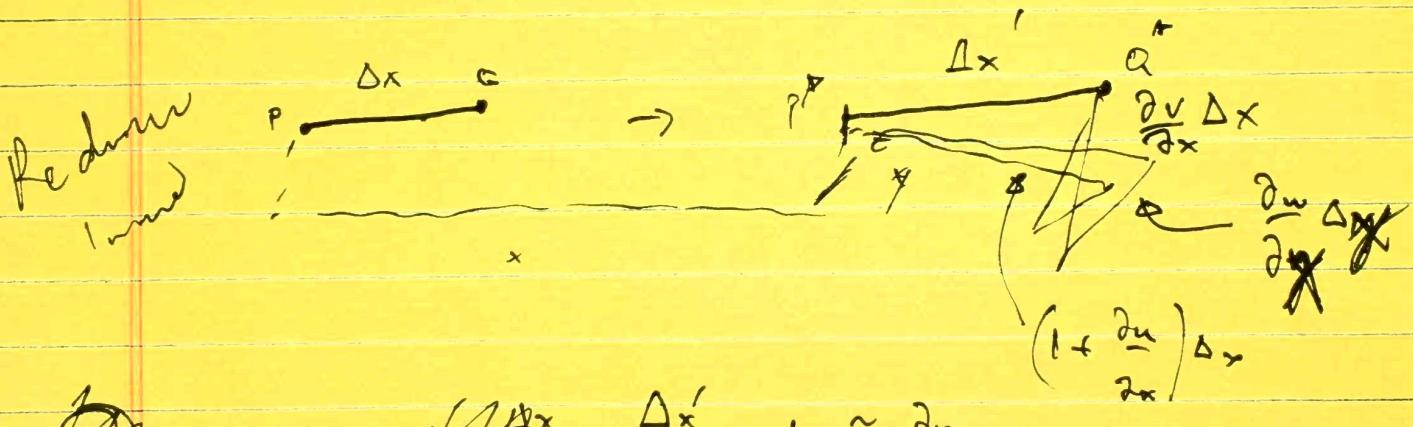
$$w^* = w + \frac{\partial u}{\partial x} \Delta x + \frac{\partial v}{\partial y} \Delta y + \frac{\partial w}{\partial z} \Delta z$$

H.J.T

Based on Taylor Series Expansion neglected

$$\left. \begin{aligned} u^* &= u + \frac{\partial u}{\partial x} \Delta x + \frac{\partial u}{\partial y} \Delta y + \frac{\partial u}{\partial z} \Delta z \\ v^* &= v + \frac{\partial v}{\partial x} \Delta x + \frac{\partial v}{\partial y} \Delta y + \frac{\partial v}{\partial z} \Delta z \\ w^* &= w + \frac{\partial w}{\partial x} \Delta x + \frac{\partial w}{\partial y} \Delta y + \frac{\partial w}{\partial z} \Delta z \end{aligned} \right\} \begin{array}{l} \text{H.O.T. neglected} \\ \text{if small deformation} \\ \text{and plane stay 2D.} \end{array}$$

Normal Strains:



$$\textcircled{1} \quad \epsilon_{xx} = \frac{\Delta x'}{\Delta x} - 1 \approx \frac{\partial u}{\partial x} + 1$$

$$\textcircled{2} \quad \Delta x' = (1 - \epsilon_{xx}) \Delta x$$

$$\textcircled{3} \quad (\Delta x')^2 = \left[ \left( \left( 1 + \frac{\partial u}{\partial x} \right) \Delta x \right)^2 + \left( \frac{\partial v}{\partial x} \Delta x \right)^2 + \left( \frac{\partial w}{\partial x} \Delta x \right)^2 \right]$$

square ②  $\rightarrow$  ③

$$\left( (1 - \varepsilon_{xx}) \frac{\Delta x}{\Delta x} \right)^2 = \left[ \left( 1 + \frac{\partial u}{\partial x} \right) \Delta x \right]^2 + \left( \frac{\partial w}{\partial x} \Delta x \right)^2 + \left( \frac{\partial v}{\partial x} \Delta x \right)^2$$



$$(1 + \varepsilon_{xx})^2 (\Delta x)^2 = \left[ 1 + 2 \frac{\partial u}{\partial x} + \left( \frac{\partial u}{\partial x} \right)^2 + \left( \frac{\partial w}{\partial x} \right)^2 + \left( \frac{\partial v}{\partial x} \right)^2 \right] \Delta x^2$$

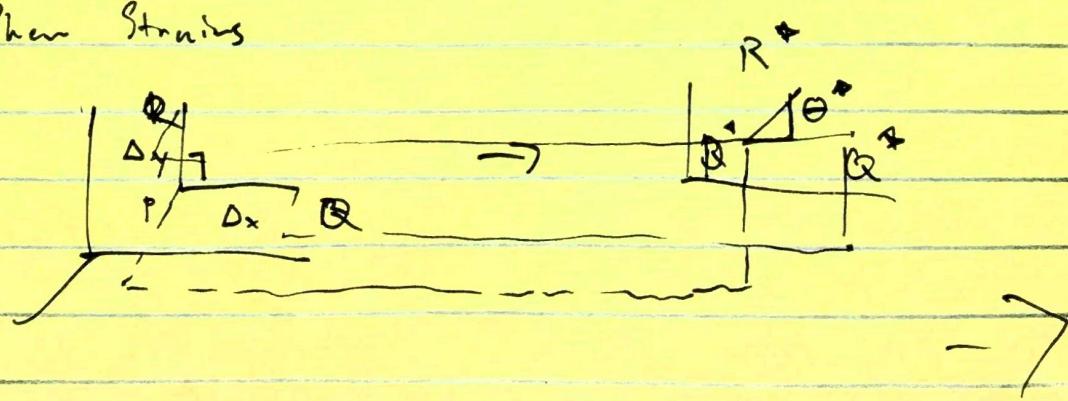
$$\Sigma_{xx} = \sqrt{1 + 2 \frac{\partial u}{\partial x} + \left( \frac{\partial u}{\partial x} \right)^2 + \left( \frac{\partial w}{\partial x} \right)^2 + \left( \frac{\partial v}{\partial x} \right)^2 - 1}$$

only works for small gauge length

would store b/u two points in 1 direction  
under arbitrary forces

$\Sigma_{yy}$  &  $\Sigma_{zz}$  are similar

Shear Strains



can define cosine of  $\theta^*$  in deflections of  
 $\bar{P}G \div \bar{PR}$

note that  $(\frac{\partial u}{\partial x})^2$  will likely be very small  
 and can often be neglected, leading to

$$\epsilon_{xx} = \sqrt{1 + 2 \frac{\partial u}{\partial x}} - 1$$

$$\epsilon_{yy} = \sqrt{(1 + \frac{\partial u}{\partial y})^2} - 1 = 1 + \frac{\partial u}{\partial y}$$

④

$\epsilon_{yy} \div \epsilon_{zz}$  are similar

$$\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}$$

Note: relatively simple for displacement field  $\rightarrow$  strain field but not reverse

Rate to measure entire strain field experimentally

Stress Transformations are similar to stress functions  
but replace

$$\sigma_{xx} \leftrightarrow \epsilon_{xx} \quad \frac{1}{4} \quad 2\gamma_{xy} \leftrightarrow \gamma_{xy}$$

must satisfy compatibility requirements

6 equations of  $\underline{\gamma}$  should be integrable  
ie strains must satisfy compatibility in order to  
satisfy displacements

### Stress Strain Relationships

! assume linear relationships

$$\begin{bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz} \\ \gamma_{xy} \\ \gamma_{yz} \\ \gamma_{zx} \end{bmatrix} = \begin{bmatrix} K_{11} & K_{12} & K_{13} & \dots & K_{16} \\ K_{21} & \ddots & & & \\ K_{31} & & \ddots & & \\ \vdots & & & \ddots & \\ \vdots & & & & K_{66} \end{bmatrix} \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \epsilon_{zz} \\ \epsilon_{xy} \\ \epsilon_{yz} \\ \epsilon_{xz} \end{bmatrix}$$

- !  $K$  are coefficients of elasticity ( $36 \rightarrow 21$ )
- strain energy arguments twist wedge to 21
- ! assume isotropic material (elastic constant in all directions) ( $21 \rightarrow 2$ )

## Elastic Constants Relationships (5 in total)

$\nu$  : Poisson's Ratio

$E$  : Elastic Young's Modulus

$G$  : (G) shear modulus (torsional)

$\lambda$  : Lamé's constant (inertial)

$K$  : Bulk modulus (used for volume pressure)

using  $\nu \neq E$ , can show 3D Hooke's Law

$$\epsilon_{xx} = \frac{1}{E} [ \sigma_{xx} - \nu (\sigma_{yy} + \sigma_{zz}) ]$$

$$\epsilon_{yy} = \frac{1}{E} [ \sigma_{yy} - \nu (\sigma_{xx} + \sigma_{zz}) ]$$

$$\epsilon_{zz} = \frac{1}{E} [ \sigma_{zz} - \nu (\sigma_{xx} + \sigma_{yy}) ]$$

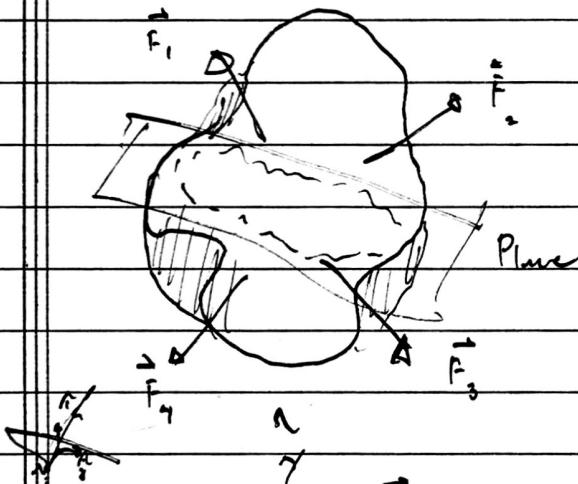
$$\gamma_{xy} = \frac{2(1+\nu)}{E} \gamma_{xy}$$

$$\gamma_{yz} = \frac{2(1+\nu)}{E} \gamma_{yz}$$

$$\gamma_{xz} = \frac{2(1+\nu)}{E} \gamma_{xz}$$

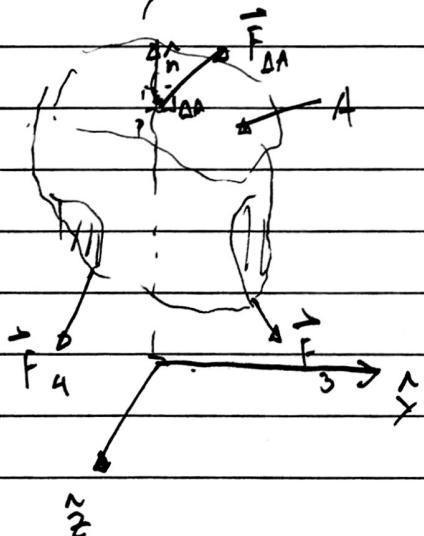
## Experimental Mechanics

- Midterm: probably going to be over KZ's stuff
- $\tilde{F}_{\text{final}}$ : actual design of possible experimental
- Hours: 10:00 AM - 11:20 AM (not 11:50)
- What the fuck is AFM
- Soft Schedule
- Review Stress from Shigley (Chapman #1)
  - Review Machine Design Notes



Under loading, find stress at point with some types:

→ Surface Force: contact of 2 bodies  
 → Body Forces: act on all elements in body (e.g., centrifugal, force field)

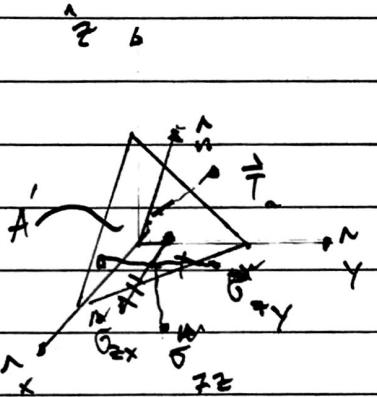


Point P is arbitrary

$$\vec{T}_P = \lim_{\Delta A \rightarrow 0} \frac{\vec{F}_{\text{on}}}{\Delta A}; \vec{T}_P \text{ is resultant stress}$$

$$\vec{\Delta F}_n = a \Delta F_{nx} \hat{i} + b \Delta F_{ny} \hat{j} + c \Delta F_{nz} \hat{k}$$

but depends on  $\hat{n}$



on elementary tributary

Notation for planes

$\sigma_{(plane)}$  (direction on plane)

$\vec{T}_a$  is resolved into 9  $\sigma$ 's

$$\sum F_x = 0 : \vec{T}_{nx} A - \sigma_{xx} A \cos(n, x) - \sigma_{zx} A \cos(n, z) = 0$$

$$- \sigma_{yx} A \cos(n, y) = 0$$

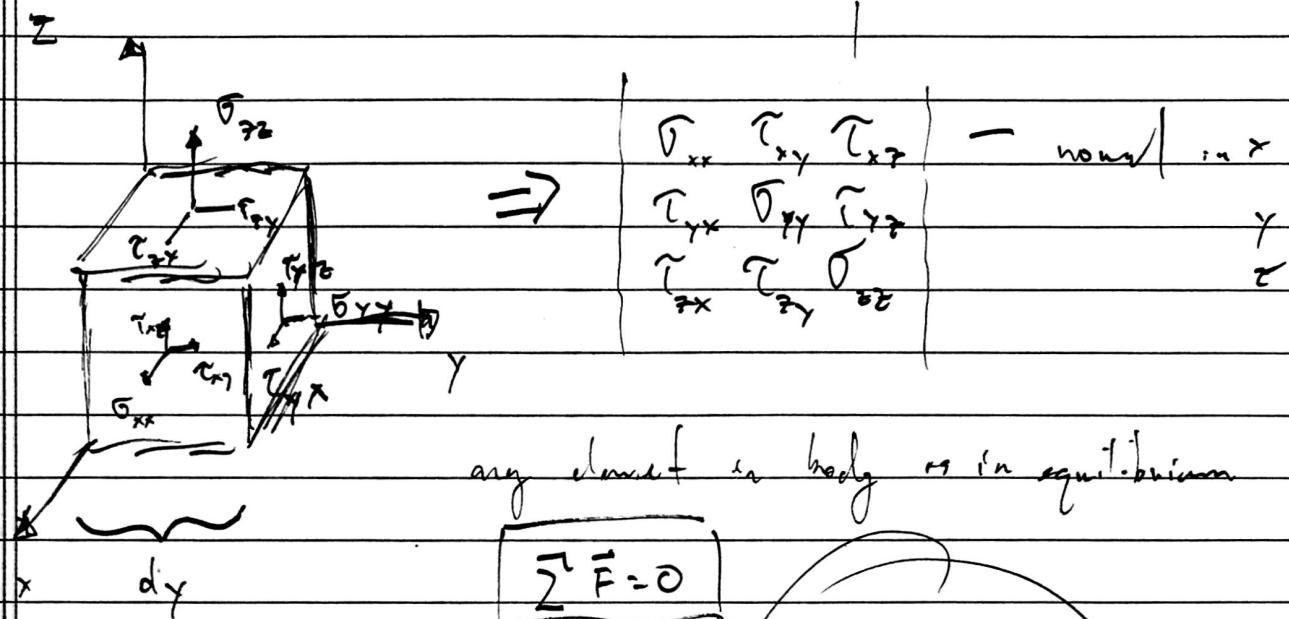
$$\therefore \vec{T}_{nx} = \sigma_{xx} \cos(n, x) + \sigma_{yx} \cos(n, y) + \sigma_{zx} \cos(n, z)$$

w/  $\cos(n, x)$  ... is projection of

$A$  on planes  $\{$  can be done

for  $\vec{T}_{ny}$  &  $\vec{T}_{nz}$

$$\boxed{\boxed{||\vec{T}_n|| = \sqrt{|\vec{T}_{nx}|^2 + |\vec{T}_{ny}|^2 + |\vec{T}_{nz}|^2}}}$$



any element in body is in equilibrium

$$\sum \vec{F} = 0$$

Stress Equations of Equilibrium

