

## A Low-Order System Frequency Response Model

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### Abstract

This tutorial paper presents the derivation of a simple, low order System Frequency Response (SFR) model that can be used for estimating the frequency behavior of a large power system, or islanded portion thereof, to sudden load disturbances. The SFR model is a simplification of other models used for this purpose, but it is believed to include the essential system dynamics.

The SFR model is based on neglecting nonlinearities and all but the largest time constants in the equations of the generating units of the power system, with the added assumption that the generation is dominated by reheat steam turbine generators. This means that the generating unit inertia and reheat time constants predominate the system average frequency response. Moreover, since only two time constants predominate, the resulting system response can be computed in closed form, thereby providing a simple, but fairly accurate, method of estimating the essential characteristics of the system frequency response.

### Key Words

Frequency, islanded operation, uniform frequency, reheat time constant, governor droop, regulation, inertia constant, damping, load-frequency behavior, frequency rate of change, frequency response.

### Introduction

Frequency response models have received limited treatment in the literature. The basic concept of the model derived here is based on the idea of uniform or average frequency, where synchronizing oscillations between generators are filtered out, but the average frequency behavior is retained. The synchronizing oscillations are illustrated in Figure 1, taken from the Florida simulations of reference [1]. We seek to average these individual machine responses with a smooth curve that can be used to represent the average frequency for the system. Such a filtered or average frequency is shown by the heavy line in Figure 1.

The concept of a uniform frequency model has been explored by numerous investigators dating back 50 years or more. Our approach is similar to that of Rudenberg [2], who provides many references on the subject as well as a mathematical derivation of the basic concept. Similar and related approaches have been pursued more recently [3,4] through work on energy functions. The basic ideas are also important in the work on system area control simulators [5,6], as well as the work on long

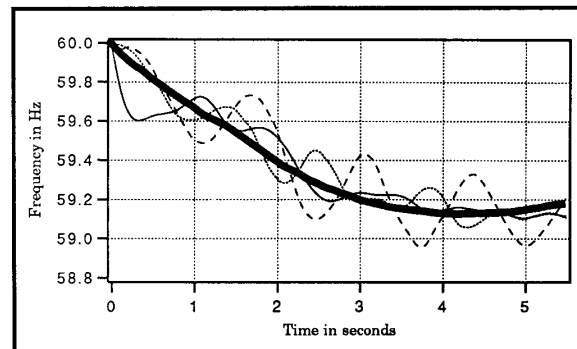


Figure 1 Simulated Unit Frequencies and System Average Frequency After Islanding

term dynamics [7,8]. In addition to these resources, certain ideas have also been adopted from the work on coherency based dynamic equivalents [9,10], as well as the work on transient energy stability analysis [11]. A related, but quite different approach, has been taken by in the work on emergency control [12], but this model is more complex than believed necessary and is more difficult to use than the method presented here. The analysis and results found here are similar to that of references 13 and 14, but our model is simpler. References 15-17 provide still other methods of analyzing the problem of frequency behavior, in varying degrees of complexity. Our approach is to provide the minimum order model that retains the essential average frequency response shape of a system with typical time constants and active speed governing.

The basic SFR model averages the machine dynamic behavior in a large system into an equivalent single machine. Topologically, we can think of the separate machines being replaced by a single large machine that is connected to the individual generator buses through ideal phase shifters. The result is a representation of only the average system dynamics, while ignoring the intermachine oscillations shown in Figure 1. There are theories that can be used to determine the total impedance to a so-called "inertial center" [3,4,18], but it is probably adequate to consider only the generator and transformer impedances.

The effect of these impedances places a limitation on the amount of load that any generating unit can absorb [19-20]. Our model neglects this limitation, although we acknowledge the validity of the concept. We assume here that the disturbance is small compared to the total rating of the island, and that the equivalent machine will be able to absorb this change. The model can be easily adapted, however, to limit the load change.

### The SFR Model

Consider a large system in which most of the generating units are reheat steam turbine units. We wish to reduce this system to one described by a minimum number of equations that will compute only the average frequency behavior. In this system, the dynamic performance of

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each rotating mass is controlled by a separate governor by integrating the individual accelerating power.

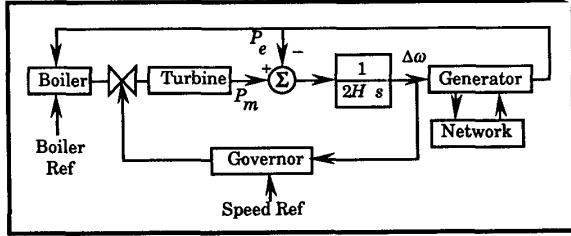


Figure 2 Generating Unit Frequency Controls

A general diagram of an individual power plant is shown in Figure 2. The SFR model examines only the midrange frequencies associated with changes in shaft speed. For this purpose, we may ignore the thermal system dynamics of the boiler as being too slow, and also the generator response as being too fast. This leaves a reduced system consisting of the governor servo motors, steam turbine, and inertia. A typical illustration of a typical governor-turbine model is shown in Figure 3 [21].

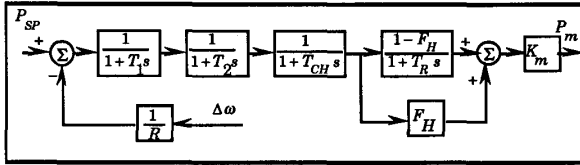


Figure 3 Typical Reheat Turbine Governor Model

The most significant time constant in this system is the reheater time constant, identified as  $T_R$  in Figure 3. This constant is usually in the range of about 6 to 12 seconds and tends to dominate the response of the largest fraction of turbine power output. Therefore, we ignore all the smaller time constants as insignificant compared to  $T_R$ . The second dominant time constant in the system is the inertia constant, called  $H$  in Figure 2. This constant is on the order of 3 to 6 seconds for a typical large unit and is always multiplied by two, which increases its effect. The third dominant constant is the  $1/R$  constant, the inverse of the governor regulation, which acts as a gain in the control diagram.

If we assume that the two time constants for the reheat and inertia dominate the response in the first few seconds, we have the reduced plant model to that shown in Figure 4.

We identify the following quantities in Figure 4:

- $P_{SP}$  = Incremental power set point, per unit
- $P_m$  = Turbine mechanical power, per unit
- $P_e$  = Generator electrical load power, per unit
- $P_a = P_m - P_e$  = Accelerating power, per unit
- $\Delta\omega$  = Incremental speed, per unit
- $F_H$  = Fraction of total power generated by the HP turbine
- $T_R$  = Reheat time constant, seconds
- $H$  = Inertia constant, seconds
- $D$  = Damping Factor
- $K_m$  = Mechanical Power Gain Factor

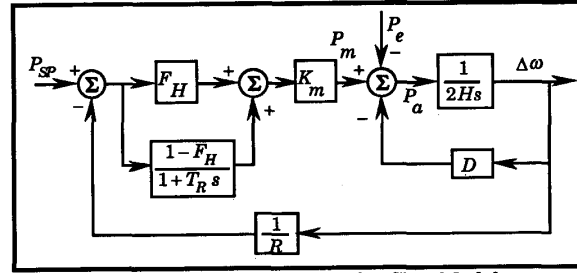


Figure 4 The Reduced Order SFR Model

By direct block diagram analysis of Figure 4, we compute

$$\Delta\omega = \left( \frac{R\omega_n^2}{DR + K_m} \right) \left( \frac{K_m(1 + F_H T_R s)P_{SP} - (1 + T_R s)P_e}{s^2 + 2\zeta\omega_n s + \omega_n^2} \right) \quad (1)$$

where

$$\omega_n^2 = \frac{DR + K_m}{2HRT_R}$$

$$\zeta = \left( \frac{2HR + (DR + K_m F_H)T_R}{2(DR + K_m)} \right) \omega_n \quad (2)$$

Obviously, from (1), the nature of the response to a change in either  $P_{SP}$  or  $P_e$  is of the same form, but of different sign and phase.

In the many studies, we are interested only in a change in  $P_e$  (with  $P_{SP} = 0$ ). For this problem, characterized by a sudden load upset, we can further simplify the system to that shown in Figure 5, where we define a "disturbance power"  $P_d$  as the new system input variable. Here we have chosen the sign of  $P_d$  such that

- $P_d > 0$  For a sudden increase in generation
- $P_d < 0$  For a sudden increase in load

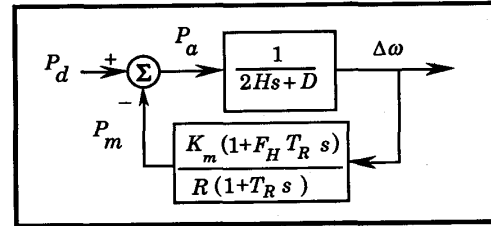


Figure 5 Simplified SFR Model with Disturbance Input

For this, the special problem of interest, we compute the frequency response in per unit to be

$$\Delta\omega = \left( \frac{R\omega_n^2}{DR + K_m} \right) \left( \frac{(1 + T_R s)P_d}{s^2 + 2\zeta\omega_n s + \omega_n^2} \right) \quad (4)$$

and the per unit speed or frequency can be computed for any  $P_d$ .

For sudden disturbances we are usually interested in  $P_d$  in the form of a step function, i.e.,

$$P_d(t) = P_{step} u(t) \quad (5)$$

where  $P_{Step}$  is the disturbance magnitude in per unit based on the system voltampere base  $S_{SB}$  and  $u(t)$  is the unit step function. In the Laplace domain, we write

$$P_d(s) = \frac{P_{Step}}{s} \quad (6)$$

and this expression can be substituted into (4) with the result

$$\Delta\omega = \left( \frac{R\omega_n^2}{DR + K_m} \right) \left( \frac{(1 + T_R s) P_{Step}}{s(s^2 + 2\zeta\omega_n s + \omega_n^2)} \right) \quad (7)$$

This equation can be solved directly to write, in the time domain,

$$\Delta\omega(t) = \frac{RP_{Step}}{DR + K_m} [1 + \alpha e^{-\omega_n t} \sin(\omega_n t + \phi)] \quad (8)$$

where

$$\alpha = \sqrt{\frac{1 - 2T_R\zeta\omega_n + T_R^2\omega_n^2}{1 - \zeta^2}} \quad (9)$$

and

$$\phi = \phi_1 - \phi_2 = \tan^{-1} \left( \frac{\omega_n T_R}{1 - \zeta\omega_n T_R} \right) - \tan^{-1} \left( \frac{\sqrt{1 - \zeta^2}}{-\zeta} \right) \quad (10)$$

The time response (8) is a damped sinusoidal frequency deviation, expressed in per unit.

#### Normalization

The equations shown above are typical of any reheat unit in the system, with all equations assumed to be in per unit on some base. Let us assume that all are in per unit on a common system base,  $S_B$ . We now combine all units into a single large unit that represents all generating units in the entire system. This can be done by adding the power equations, as follows:

$$\sum_i 2H_i s \Delta\omega = \sum_i P_{mi} - \sum_i P_{ei} \quad (11)$$

$$\sum_i P_{GVi} = \sum_i P_{SPi} - \sum_i \left( \frac{1}{R_i} \right) \Delta\omega \quad (12)$$

$$(1 + T_R s) \sum_i P_{mi} = K_m (1 + F_H T_R s) \sum_i P_{GVi} \quad (13)$$

where we assume that all equations are on a common system base  $S_B$ . We now renormalize (11) to (13) to the total system base  $S_{SB}$  which is equal to the sum of the ratings of all generating units in the system.

$$S_{SB} = \sum_{i=1}^n S_{Bi} \quad (14)$$

This change of base multiplies (11-13) by the ratio of the two bases with the result

$$\frac{S_B}{S_{SB}} \sum_i 2H_i s \Delta\omega = \frac{K_m (1 + F_H T_R s)}{(1 + T_R s)} \left[ \frac{S_B}{S_{SB}} \sum_i P_{SPi} - \frac{S_B}{S_{SB}} \sum_i \left( \frac{1}{R_i} \right) \Delta\omega \right] - \frac{S_B}{S_{SB}} \sum_i P_{ei} \quad (15)$$

which defines the equivalent generator parameters. The normalized value of these parameters on the total system base will be typical of those for a single unit on its own base.

The parameter  $K_m$  is an effective gain constant that expresses the total mechanical power in terms of the governing valve area. This gain is affected by the system power factor and by the spinning reserve, as follows.

$$K_m = \frac{P_m \text{ in MW}}{S_{SB}} = \frac{1}{S_{SB}} \sum_i S_{Bi} F_{Pi} (1 - f_{SR}) = F_P (1 - f_{SR}) \quad (16)$$

where

$F_P$  = Power Factor

$f_{SR}$  = Fraction of Units on Spinning Reserve

We assume a constant power factor for all units.

#### Example

As an example of the computation of the SFR consider a system with the following typical parameters.

$$\begin{array}{lll} R = 0.05 & H = 4.0 \text{ s} & K_m = 0.95 \\ F_H = 0.3 & T_R = 8.0 \text{ s} & D = 1.0 \end{array}$$

Then we compute

$$\begin{array}{lll} \omega_n = 0.559 & \zeta\omega_n = 0.438 & \phi_1 = 131.94^\circ \\ \zeta = 0.783 & \omega_r = 0.348 & \phi_2 = 141.54^\circ \\ \alpha = 6.011 & \sqrt{1 - \zeta^2} = 0.622 & \phi = -9.60^\circ \end{array}$$

The result, for values of  $P_{Step}$  from  $-0.05$  to  $-0.30$  is shown in Figure 6. Note that the response is underdamped since  $\zeta < 1$  and that the damping exponential  $\zeta\omega_n$  is fairly large, resulting in almost negligible oscillation beyond the first swing.

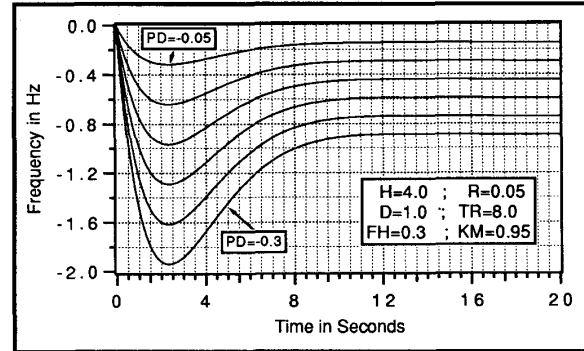


Figure 6 Frequency Response for Varying Values of  $P_d$

#### Characteristics of the Equivalent System Response

It is instructive to examine the effect on the transient response of the several important parameters in the system. Since the solution (8) is written in closed form we readily compute the slope of the response.

$$\frac{d\Delta\omega}{dt} = \frac{\alpha\omega_n RP_{Step}}{DR + K_m} e^{-\zeta\omega_n t} \sin(\omega_n t + \phi_1) \quad (17)$$

where all parameters are previously defined. We are particularly interested in the value of (17) at two points, when  $t = 0$ , which corresponds to the maximum rate of

change of slope, and when the slope is zero, which corresponds to the maximum frequency deviation.

$$1. \quad t = 0$$

$$\left. \frac{d\omega}{dt} \right|_{t=0} = \frac{aR\omega_n P_{Step}}{DR + K_m} \sin \phi_1 = \frac{P_{Step}}{2H} \quad (18)$$

$$2. \quad \frac{d\omega}{dt} = 0$$

$$0 = \frac{aR\omega_n P_{Step}}{DR + K_m} e^{-\omega_n t} \sin(\omega_n t + \phi_1) \quad (19)$$

Equation (19) is satisfied when  $\omega_n t + \phi_1 = n\pi$  for  $n$  an integer, including zero. If we call this time  $t_z$ , we compute

$$t_z = \frac{n\pi - \phi_1}{\omega_n} = \frac{1}{\omega_n} \tan^{-1} \left( \frac{\omega_n T_R}{\zeta \omega_n T_R - 1} \right) \quad (20)$$

These slope parameters are clearly observed in Figure 6. The initial slope depends only on  $P_{Step}$  and  $H$ , hence it changes for each run plotted in the figure. However,  $t_z$  is not a function of  $P_{Step}$ , so the maximum frequency deviation occurs at exactly the same time (2.35 s) for all disturbances. Note also that Figure 6 is plotted in hertz so all equations must have been multiplied by 60.

Another parameter that can be readily checked is the regulation. Governors are set with droop  $R$  to give a steady-state speed vs. power relationship of

$$\Delta\omega = \frac{RP_{Step}}{DR + K_m} \quad (21)$$

where both  $\omega$  and  $P_{Step}$  are incremental, normalized quantities. Note that  $P_{Step}$  must take the sign of  $P_d$ , which would be negative for an increase in load or loss of generation. Thus, in Figure 6, where we let  $P_{Step} = -0.3$  and  $R = 0.05$  we compute

$$\Delta\omega = \frac{-(0.05)(0.3)}{(1)(0.05) + 0.95} = -0.015 \text{ per unit} = -0.9 \text{ Hz}$$

which is clearly observed in Figure 6. This result can be also be verified mathematically by the final value theorem

$$\lim_{t \rightarrow \infty} \Delta\omega(t) = \lim_{s \rightarrow 0} s \Delta\omega(s) = \frac{RP_{Step}}{DR + K_m} \quad (22)$$

It is difficult to visualize the effect of each physical parameter of the SFR model without plotting the results. Therefore, each parameter is now varied in turn and the results plotted to illustrate the effect of that parameter.

#### The Effect of Governor Droop, $R$

The value of  $R$  is varied from 0.05 to 0.10 in increments of 0.01 per unit. The results are shown in Figure 7. Actual observed system responses in islanded situations have sometimes shown the net system regulation to be in the neighborhood of 0.1.

This indicates that some generating units are operating with governor valves blocked, giving a regulation of 1.0 per unit for these units. This increases the net regulation for the system in proportion to the size of the units with blocked valves. Note in Figure 7 that the steady state regulation is exactly as given by (21). Thus,

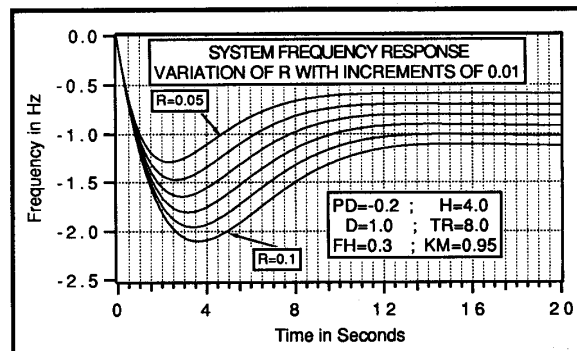


Figure 7 Frequency Response for Varying Values of  $R$

for the disturbance with  $P_{Step} = -0.2$  and  $DR + K_m = 1$ , as illustrated, we compute  $\Delta\omega = 60(0.2)R = 12R$  for each case. The simulation shown in Figure 7 must be extended for a rather long time to accurately observe the final value.

It is also important to note that the assumed droop setting  $R$  has absolutely no effect on the initial rate of frequency decline. This is important. Even if all governors are at the extreme valve-closing end of their individual backlash limits, as in following a gradual load decrease, a sudden loss of load would require a rapid change to a valve open condition. While traversing this backlash region, the system is operating essentially open loop and the natural 100% turbine regulation prevails. However, the initial rate of change is the same as for a tightly tuned system with no backlash. This effect can be noted in Figure 5, where the regulation term is seen to be affected by the lag of the feedback term, which is a "lag-lead" function. The difference regulation makes is in the recovery time, the maximum offset, and the final value. It is important that this recovery be fast in order to limit the time of exposure at frequencies below about 58 Hz.

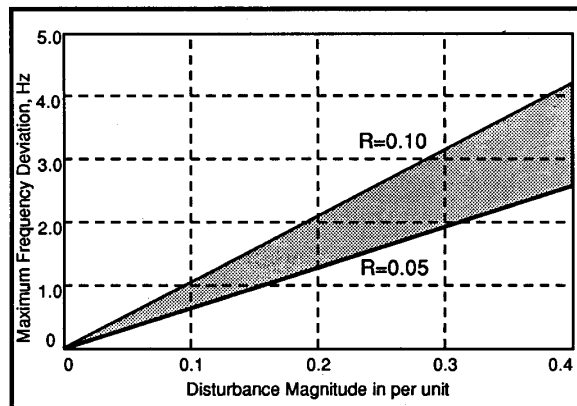


Figure 8 Max Frequency Deviation vs. Disturbance Size

The value of  $R$  is probably bounded between about 0.05 and 0.10 for most systems. This effectively bounds the maximum frequency deviation for a given disturbance to the shaded region shown in Figure 8. Here, the disturbance is the step function magnitude, as before. Figure 8 is plotted with nominal values of the other parameters ( $H = 4.0$  s,  $T_R = 8.0$  s,  $F_H = 0.3$ ,  $K_m = 0.95$ ,  $D = 1.0$ ). The maximum disturbance power is actually limited, as discussed in [20], but this is not considered in the calculation of values for Figure 8.

### The Effect of Inertia, $H$

The value of  $H$  has a direct effect on the initial slope, as shown in the simulations of Figure 9 for various  $H$  values.

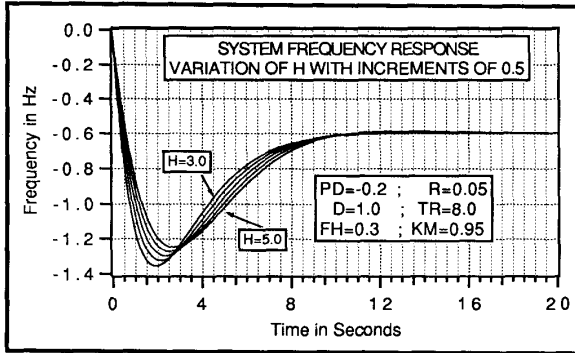


Figure 9 Frequency Response for Varying Values of  $H$

The inertia constant affects almost all response measures, including  $\omega_n$ , and  $\zeta$ , as given by equation (2), and thereby all other computed parameters as given by (9) and (10), including the initial slope and the time  $t_p$  of maximum frequency deviation as given by (18) and (19), respectively. It is also interesting to observe that  $(1/2H)$  is the forward loop gain in Figure 5. Note also that  $H$  is involved only in the forward loop. All other parameters are feedback parameters.

The most pronounced effect of changes in  $H$  is the change in the initial slope, and the time of the peak response. Note that  $H$  does not affect the final steady-state value of frequency.

### The Effect of Reheat Time Constant, $T_R$

The reheat time constant is an important system parameter. Usually 70% or more of the turbine output is delayed by the reheat time constant and the variation of this time lag has a pronounced effect on frequency performance. Figure 10 shows a range that is typical of large generating units.  $T_R$  has an effect on the damping ratio  $\zeta$  and the natural undamped frequency  $\omega_n$ , but does not effect the initial slope or the final value.

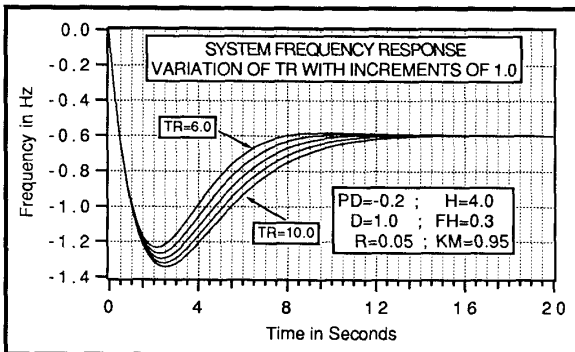


Figure 10 Frequency Response for Varying Values of  $T_R$

### The Effect of High Pressure Fraction, $F_H$

The constant  $F_H$  measures the fraction of shaft power developed by the high pressure turbine on a single reheat system. This is the fraction of shaft power that is not delayed by reheating. Figure 11 shows the effect of varying only  $F_H$ , with other parameters at their nominal

values. Large values of  $F_H$  have a pronounced effect on  $\zeta$  and can make the system overdamped ( $\zeta > 1$ ). When this occurs, the frequency response is not given by (8), but is a combination of exponentials. The frequency response equation for this condition is computed from (7) by factoring the quadratic to write

$$\Delta\omega(s) = \frac{T_1 T_2 R \omega_n^2 P_{Step}}{DR + K_m} \left( \frac{1 + T_R s}{s(1 + T_1 s)(1 + T_2 s)} \right) \quad (23)$$

Then

$$\Delta\omega(t) = \frac{T_1 T_2 R \omega_n^2 P_{Step}}{DR + K_m} \left( 1 + \frac{T_1 - T_R}{T_2 - T_1} e^{-t/T_1} - \frac{T_2 - T_R}{T_2 - T_1} e^{-t/T_2} \right) u(t) \quad (24)$$

In the numerical example plotted in Figure 11, when  $F_H$  is greater than 0.4, the system is overdamped. This corresponds to the three curves with the smallest frequency displacement in the figure.

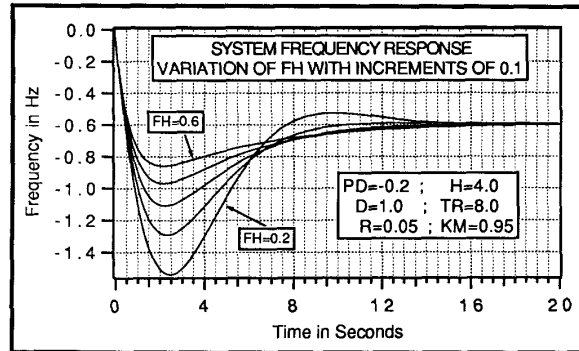


Figure 11 Frequency Response for Varying Values of  $F_H$

### Performance Analysis and Model Restrictions

Performance analysis for a given system using the SFR model is relatively easy, providing data from an actual system disturbance is available to tune the parameters. Actual system disturbance records are necessary since some system parameters are almost impossible to determine due to unknown operating conditions. Regulation or droop ( $R$ ), for example, depends on turbine droop adjustment, the effect of large nuclear units with blocked governor valves, and also the frequency dependence of the load. However, this effect is clearly revealed in the frequency record of an actual disturbance by observing the steady-state frequency error. This steady-state error can be computed from (21) to be

$$f_{ss} = \frac{60 R P_{Step}}{DR + K_m} \text{ Hz} \quad (25)$$

If the disturbance magnitude is known, and it usually is easily estimated, then  $R$  can be estimated. Note that the denominator of (25) is approximately unity in most cases, which helps in making the estimate.

### The Effect of Frequency Dependence of Load

Power system loads are known to be sensitive to system frequency. One way of characterizing this dependence is to model the load as having a constant component as well as a frequency dependent component.

$$P_L = P_{L0} (1 + k_f \Delta\omega) \quad (26)$$

Then the incremental change in load is a function of the incremental change in frequency. But this effect is already included in the model in the form of the damping constant  $D$ , where we may write

$$\Delta P_L = D \Delta \omega \quad (27)$$

where all values are in per unit. The damping constant is usually used to represent the damping of the load and other effects to rotor oscillations. From Figure 4, however, we note that the product (27) is of the same sign as the electrical power, which is exactly the incremental load of the system. Thus, we have both components of load represented, the constant portion and the frequency dependent portion.

From equations (2) and (8), we see that the system undamped natural frequency and damping are functions of  $D$ , but this dependence always appears as the product  $DR$ . Since  $R$  is small, nominally about 0.05 per unit, the product  $DR$  will also be small and the effect of  $D$  is diminished. We can illustrate the effect of  $D$  on the frequency response by plotting for various values of this parameter, with the results seen in Figure 12. Note the similarities to Figure 7, but with reduced effect.

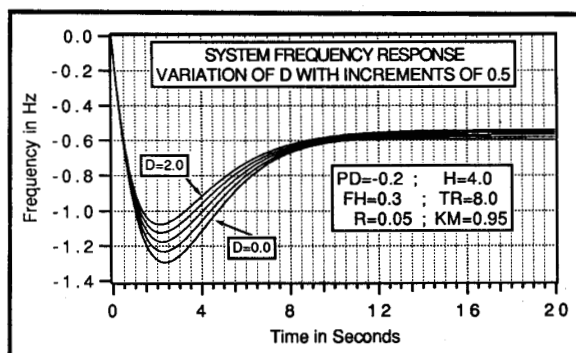


Figure 12 Frequency Response for Various Values of  $D$

**Use of the SFR Model For An Actual System Disturbance**  
Data for an actual system separation is used to validate the SFR model. The physical system, in this case, separated and islands were created, one with excess generation and one with excess load. The simulation presented is for the island with excess load. The size of the disturbance and steady state frequency error are known and  $H$  is easily estimated with confidence. Only the reheat time constant needs to be estimated by trial and error. The result is shown in Figure 13, where the discrete points are the actual system frequency at the point of measurement and the smooth line is the simulation of that frequency.

Note that the data from the physical system includes the effect of the oscillation of local machines with respect to other machines in the island. This is usually the case, and measurements from different parts of the island will show these local oscillations, similar to the machine plots in Figure 1. The local machines oscillate about the average system frequency, which is computed by the SFR model.

#### Comparison with Other Models of Frequency Behavior

One application for frequency response models is for the setting of underfrequency load shedding relays, where relays are used to shed portions of the load and thereby restore load-generation balance. One method of

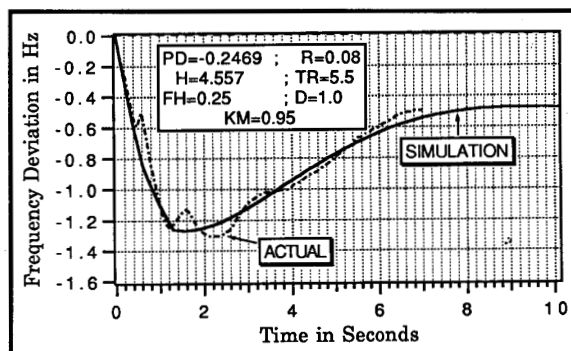


Figure 13 Model Validation for an Actual Disturbance  
estimating frequency behavior for this purpose is to model the frequency decline following the disturbance, using a model that reflects the load and generator behavior [21]. This model results in a first order differential equation that has the solution

$$\Delta \omega(t) = \frac{P_{Step}}{D} (1 - e^{-(D/2H)t}) \quad (28)$$

This model describes the frequency trajectory as it falls from its initial value in an exponential fashion. The initial rate of decline is as given in (18), but the subsequent rate of decay is faster than that predicted by (17) since the exponential model does not include the governor behavior. Still, this model is often used for determining the time at which load shedding relays should be employed and is attractive for its simplicity. Note that only the inertia and damping are required to find the frequency for a disturbance of any size. The exponential model (28) includes the effect of load frequency dependence  $D$ . As noted before, the  $D$  effect is similar to that of governor droop  $R$ , but  $R$  is much more important in its end result.

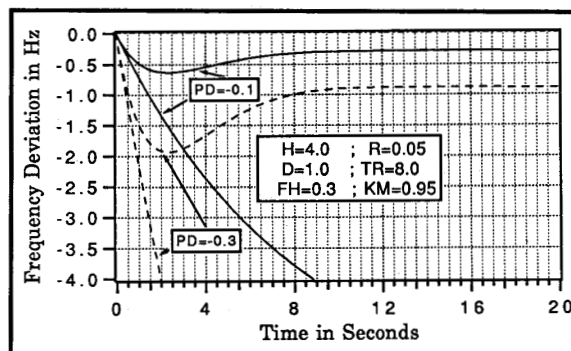


Figure 14 Comparison of SFR and Exponential Models

Other models described in the references are more complex and will not be described here in detail.

#### Conclusions

The SFR model of system frequency behavior following an islanding event or a large disturbance on the interconnected system is a greatly simplified model of system behavior. The model developed for this purpose omits many details and ignores small time constants in an effort to provide a model that may be useful in approximating the system frequency performance, including the essential behavior of speed governing and turbine response. However, in spite of the model

simplicity, the comparison with actual system disturbances and detailed stability simulations, as shown in Figures 13 and 1, respectively, are rather encouraging. Moreover, the model provides an understanding of the way in which important system parameters affect the frequency response. Such understanding is difficult to achieve in high order models, where performance is a very complex function of many system variables.

#### References

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## Discussion

**M. E. Connolly**, (Houston Lighting & Power Co., Houston, Texas): The authors are to be congratulated on their reduced order model of system frequency response. The matter of system frequency response has been a topic of major interest in the Electric Reliability Council of Texas since the region operates as an island not synchronously connected to the national grids. Various methods for simplified computation of system frequency response to loss of generation disturbances have been investigated.

One general observation that has come out of these investigations is that simplified methods that neglect the limits on turbine capability will yield unsatisfactory estimates of the maximum frequency excursion. Figure 8 as presented in the paper could be misleading in this respect. The typical scenario for unit commitment to minimize cost causes many units to bump against turbine limits when responding to generation imbalances of the magnitude examined. Could the authors comment on the appropriate method to introduce this constraint into their formulation?

Another observation is offered with respect to the timing of the maximum frequency excursion. The examples presented by the authors are restricted to cases in which the power imbalance does not exceed the available high pressure response  $|P_d| < |F_H|$ . For imbalances which exceed the available high pressure response the timing of the minimum frequency will be more dependent on the reheat time constant. System frequency will continue to decay until sufficient downstream turbine power is developed to cover the power deficit.

ERCOT has employed a very simple technique to estimate system minimum frequency for screening purposes. This method is based on a simple algebraic formulation of the power balance equation at the time of minimum frequency deviation. At this point  $dw/dt = 0$  and the power balance can be expressed as:

$$P_d = P_L + P_{GOV} \quad (a)$$

The power due to governor response can be estimated as:

$$P_{GOV} = (\Delta w/R)(S_{SB} * F_H)$$

The assumption here is that *only* the high pressure power is available for short term response.

This simple model can yield very good results when expanded to explicitly account for the parameters of each generating unit on line, including the maximum turbine unit:

$$P_{GOV} = \sum_{i=1}^n \text{Min} [((1/R_i)\Delta w S_{Bi} * K_{Hi}), (S_{Bi} - P_i)]$$

where  $P_i$  is the power output of generator immediately preceding this disturbance.

This expanded formulation was implemented in a spreadsheet program using equation (a) and with  $P_L = D\Delta w$ . The spreadsheet can be used to directly calculate  $P_d$  for a chosen  $\Delta w$ . A second option is to choose  $P_d$  and allow the spreadsheet to iterate a solution to  $\Delta w$ , given some initial estimate of  $\Delta w$ .

The "spreadsheet method" has been used in ERCOT studies of modifications to spinning reserve policy to accommodate generating units larger than 1000MW. Studies were made using both the simplified method and a full stability program to simulate the results of an instrumented trip of a 1250MW unit.

The performance of the "spreadsheet method" was compared to results from modeling the same disturbances using a full stability study. The stability studies included complex generator and excitation modeling with standard IEEE modeling of the turbine/governors. The "spreadsheet method" gave very good results for those cases to which it was suited:

Case No.	Total Load (MW)	$P_d$ (MW)	fmin1	fmin2
1	34177	1516	59.575	59.541
2	34177	2020	59.32	59.29
3	27780	1300	59.32	59.35

fmin1 is minimum by stability program

fmin2 is minimum by spreadsheet

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**M. S. Baldwin**, (1851 Green Meadow Lane, Orlando, Florida 32825): It is refreshing to read a paper in which the authors try to simplify life instead of overly complicating it.

As the authors state, this paper is a tutorial, in that the model highlights the most important parameters of the power system. It appears to me that the model, if relatively easy and inexpensive to use, can be a screening tool. There are many perturbations of unit frequencies around the average system frequency, as shown in Figure 1. Studies using more complex models can make definition of the average response difficult to define. The foregoing statement is not intended to suggest that, in some cases, the local variations from the average response should not be studied, but the use of this model could reduce the number of cases which would have to be studied with a more complex model.

It is stated that all of the generators of an islanded portion of a power system are represented by a single large unit connected to the generator buses through ideal phase shifters. How many generator and load buses were represented in the parametric study for which the results are shown in Figures 7 through 12? Were the load buses separate from the generator buses? For Figure 1, how closely did the system represented with the SFR model compare with that used for the simulated unit frequencies?

The parameter  $K$  is affected by the system power factor. Was the power factor used as a variable parameter? If so, what were the results? References 19 and 20 report the results of investigations made to determine maximum frequency decay rates. In follow-up investigations, it was noted that the initial drop in frequency was aggravated by generators operating in the underexcited condition, ie, absorbing reactive power from the system. In this case, the voltage regulator action, shortly after the imposition of the overload, aggravated the frequency decay. There is no mention of voltage regulator simulation in the SFR model, although the turbine governor is represented. It should be noted that present day digital governing systems sample and correct only once or twice a second. Would the authors comment on these points?

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The authors are grateful to the discussers for their genuine interest in the subject of the paper and for their valuable comments, observations, and questions.

Mr. Baldwin's comments are appreciated, particularly in view of his considerable past contributions to the modeling of system frequency response as noted in references 19, 20, and his many other papers. His question concerning the concept of reducing the system to a single unit through phase shifters is simply a conceptual method of system reduction to a single generator feeding a single load at a single bus. This concept was used in [10] and can be briefly reviewed as follows:

1. The  $n$  machines in the island to be studied are combined in a step-by-step process, beginning with each machine having its normal connection at the appropriate generator nodes.
2. The nodes to which the machines are connected are



now connected to a common bus through ideal phase shifting transformers of the appropriate tap setting and phase shift angle, with the result that the network is not perturbed in any way.

3. Next, a new generator of rating equal to the sum of all the original generators is connected to the new common bus and the original generators are removed. Again, the network is unperturbed from its original condition and all power flows are exactly the same as in the original network.
4. The original generator nodes are now eliminated by combining the phase shifting transformers with the generator step-up transformers at each of the former generator buses.
5. Since the network is entirely passive, all nodes can be reduced to a single load bus, with that load fed from the common generator.

The authors referred to this concept as a justification for taking the step of lumping all island load and generation to a common bus. It is a logical and appropriate concept for the purpose at hand, namely, to view only the average frequency performance of the island. The plots given in the paper are not for any particular system, but were simply the result of using the model presented with only one generator and one load characterizing an islanded system with a disturbance of a given per unit magnitude. The exceptions are the results shown in Figures 1 and 13, which show the model performance in simulating the behavior of a large system.

Mr. Baldwin also correctly noted that there is no result given in the paper that shows the effect of varying the parameter  $K_m$ . Figure 15 shows the effect of varying the gain parameter for a system with typical parameters. The result is very similar to varying the parameters  $R$  and  $D$ . This is expected since these parameters all occur in the denominator of the coefficient term in (8), in the form  $DR + K_m$ . Therefore, varying any of these parameters has a similar effect on the frequency.

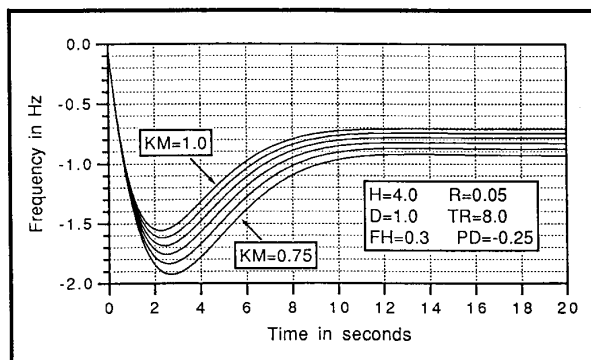


Figure 15 Frequency Response for Varying  $K_m$

Mr. Baldwin's comments regarding the effect of the generator voltage regulators is appreciated. The generator excitation is not represented in the SFR model and the level of excitation is assumed constant. He correctly notes that this may be important in some cases, especially where units are operating underexcited. It would seem that this would be more of a problem in the generation rich island than in the load rich island.

We also appreciate the comment regarding the slow sampling time of digital governors that make governing corrections only once or twice per second. On the initial downward swing in frequency, this means that the

system is operating open loop insofar as the speed governing control is concerned. This could be simulated, at least approximately, by setting the speed regulation to 100% for that period of time prior to the first governor correction. In our simulations, we assumed that the governors are all continuous controllers. If a large fraction of the total generation is controlled by digital governors with slow sampling rates, this could lead to large errors in the results that are difficult to estimate. The SFR model is linear, however, and could be made to simulate pulsed governing control, with 100% regulation between pulses.

The comments of Mr. Connolly are greatly appreciated, and his experience in making similar types of frequency studies in the Texas interconnection are valuable contributions to the discussion. His comment regarding the limits on turbine capability are interesting. Our turbine model is exactly the same as that used in most transient stability simulations. These models include no limitations, but are entirely linear. The governor models, however, usually include limiting both on valve position and valve velocity. These limitations were not included in the linear SFR model. Such limitations can be included, however, in a rather direct way. Again, since the model is linear, the response can be treated by linear superposition. Thus, if there is a loss of turbine output at a given time (or frequency) in the response, this can be included in the simulation by changing the turbine gain  $K_m$  by an appropriate amount at the correct time. This concept is illustrated in Figure 16, where a 10% loss of turbine power is assumed to occur at exactly 1.0 second following the initial disturbance. This is simulated by changing the turbine gain constant  $K_m$  from its original value of 0.9 to a final value of 0.8. This amounts to a change in the coefficient in equation (8) of just over 11%. This illustrates one of the advantages of a linear model, in which the output can be altered by linear superposition of several independent inputs. In this case, the first step input occurs in the islanding disturbance, and the second step occurs in the gain  $K_m$ , but both inputs are assumed to affect only the magnitude of the step. Actually, changing  $K_m$  also affects the undamped natural frequency and the damping factor, as given by (2), but the error in neglecting these effects will not be large for small changes in turbine gain.

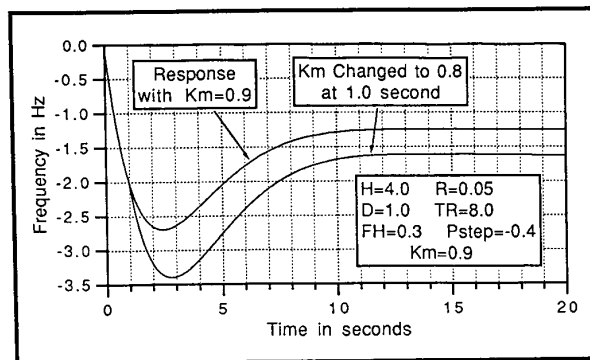


Figure 16 Frequency Response for a Sudden 10% Change in  $K_m$  at a time  $t = 1.0$  second

Mr. Connolly noted correctly that, in the examples presented in the paper, the size of the disturbance was restricted to cases where the response is relatively small. The model is linear, however, and the response for larger disturbances has the same shape, as noted in Figure 17, where the initial disturbance varies from -0.2 to -1.0 per unit.

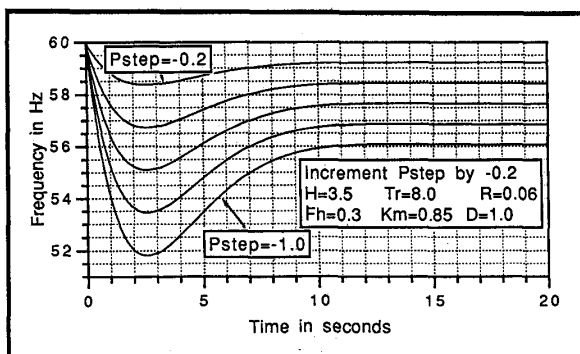


Figure 17 Frequency Response for Varying  $P_d$  from 0.2 to 1.0 per unit in Steps of 0.2 per unit

Mr. Connolly's description of the "spreadsheet method" used by ERCOT is very interesting indeed. We tried a simulation of the first of his three disturbances, with the results shown in Figure 18.

We believe that any such disturbance can easily be simulated by the SFR method, once the constants

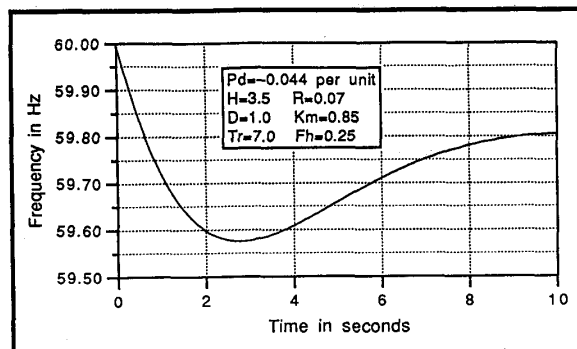


Figure 18 Simulation Results of the ERCOT Case 1 describing the system are known. The fact that ERCOT finds these simple models to be applicable to modifications in spinning reserve policy is encouraging for the application of approximate linear models. Indeed, the results given in the discussion are remarkably close to more detailed simulations.

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