Question 1

The distribution of stress in an aluminum machine component is given by:

$$\sigma = \begin{bmatrix} \sigma_x & \tau_{xy} & \tau_{xz} \\ \tau_{yx} & \sigma_y & \tau_{yz} \\ \tau_{zx} & \tau_{zy} & \sigma_z \end{bmatrix} = \begin{bmatrix} y + 2z^2 & 3z^2 & 2y^2 \\ 3z^2 & x + z & x^2 \\ 2y^2 & x^2 & 3x + y \end{bmatrix} MPa$$

Calculate the state of strain of a point positioned at (1, 2, 4). Use E = 70 GPa and $\nu = 0.3$.

Solution

From the generalized Hooke's law, the stress-strain relation is given by:

$$\epsilon_x = \frac{1}{E}(\sigma_x - \nu(\sigma_y + \sigma_z))$$

$$\epsilon_y = \frac{1}{E}(\sigma_y - \nu(\sigma_x + \sigma_z))$$

$$\epsilon_z = \frac{1}{E}(\sigma_z - \nu(\sigma_x + \sigma_y))$$

$$\gamma_{xy} = \frac{1}{2G}\tau_{xy}$$

$$\gamma_{yz} = \frac{1}{2G}\tau_{yz}$$

$$\gamma_{xz} = \frac{1}{2G}\tau_{xz}$$

Where

$$G = \frac{E}{2(1+\nu)} = \frac{70}{2(1+0.3)} = 26.923 \,\text{GPa}$$

Evaluate the stress state at (1, 2, 4):

$$\sigma = \begin{bmatrix} 2+2(4)^2 & 3(4)^2 & 2(2)^2 \\ 3(4)^2 & 1+4 & 1^2 \\ 2(2)^2 & 1^2 & 3(1)+2 \end{bmatrix} = \begin{bmatrix} 34 & 48 & 8 \\ 48 & 5 & 1 \\ 8 & 1 & 5 \end{bmatrix} MPa$$

Using the stress-strain relations, the strain state is given by:

$$\epsilon = \begin{bmatrix} \frac{1}{E}(\sigma_x - \nu(\sigma_y + \sigma_z)) & \frac{1}{2G}\tau_{xy} & \frac{1}{2G}\tau_{xz} \\ \frac{1}{2G}\tau_{yx} & \frac{1}{E}(\sigma_y - \nu(\sigma_x + \sigma_z)) & \frac{1}{2G}\tau_{yz} \\ \frac{1}{2G}\tau_{zx} & \frac{1}{2G}\tau_{zy} & \frac{1}{E}(\sigma_z - \nu(\sigma_x + \sigma_y)) \end{bmatrix}$$

$$= \begin{bmatrix} \frac{1}{70 \times 10^3}(34 - 0.3(5 + 5)) & \frac{1}{2(26.923) \times 10^3}48 & \frac{1}{2(26.923) \times 10^3}8 \\ \frac{1}{2(26.923) \times 10^3}48 & \frac{1}{70 \times 10^3}(5 - 0.3(34 + 5)) & \frac{1}{2(26.923) \times 10^3}1 \\ \frac{1}{2(26.923) \times 10^3}8 & \frac{1}{2(26.923) \times 10^3}1 & \frac{1}{70 \times 10^3}(5 - 0.3(34 + 5)) \end{bmatrix}$$

$$= \begin{bmatrix} 4.43 \times 10^{-4} & 8.91 \times 10^{-4} & 1.49 \times 10^{-4} \\ 8.91 \times 10^{-4} & -9.57 \times 10^{-5} & 1.86 \times 10^{-5} \\ 1.49 \times 10^{-4} & 1.86 \times 10^{-5} & -9.57 \times 10^{-5} \end{bmatrix}$$

Question 2

The aluminum rectangular parallelepiped (E=70 GPa and $\nu=\frac{1}{3}$) shown in Fig. 1 has dimensions $a=150\,\mathrm{mm}$, $b=100\,\mathrm{mm}$, and $c=75\,\mathrm{mm}$. It is subjected to tri-axial stresses $\sigma_x=70\,\mathrm{MPa}$, $\sigma_y=-30\,\mathrm{MPa}$, and $\sigma_z=-15\,\mathrm{MPa}$ acting on the x,y, and z faces, respectively. Determine,

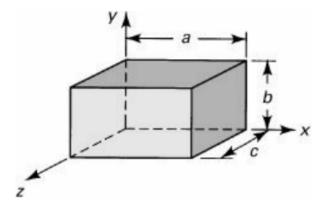


Figure 1: Rectangular parallelepiped subjected to tri-axial stresses.

- (a) the changes Δa , Δb , and Δc in the dimensions of the block.
- (b) the change ΔV in the volume.

Solution

(a)

From the generalized Hooke's law, the stress-strain relation is given by:

$$\epsilon_x = \frac{1}{E}(\sigma_x - \nu(\sigma_y + \sigma_z))$$

$$\epsilon_y = \frac{1}{E}(\sigma_y - \nu(\sigma_x + \sigma_z))$$

$$\epsilon_z = \frac{1}{E}(\sigma_z - \nu(\sigma_x + \sigma_y))$$

From the definition of strain,

$$\Delta x = \epsilon_x a$$
$$\Delta y = \epsilon_y b$$
$$\Delta z = \epsilon_z c$$

By direct substitution,

$$\Delta x = \frac{1}{E} (\sigma_x - \nu(\sigma_y + \sigma_z))a$$

$$= \frac{1}{70 \times 10^3} (70 - \frac{1}{3}(-30 - 15))150$$

$$= \boxed{0.182 \,\text{mm}}$$

$$\Delta b = \frac{1}{E} (\sigma_y - \nu(\sigma_x + \sigma_z))b$$

$$= \frac{1}{70 \times 10^3} (-30 - \frac{1}{3}(70 - 15))100$$

$$= \boxed{-0.0690 \,\text{mm}}$$

$$\Delta c = \frac{1}{E} (\sigma_z - \nu(\sigma_x + \sigma_y))c$$

$$= \frac{1}{70 \times 10^3} (-15 - \frac{1}{3}(70 - 30))75$$

$$= \boxed{-0.0304 \,\text{mm}}$$

(b)

The change in volume is given by:

$$\Delta V = eV_0$$

Where dilation e is given by:

$$e = \epsilon_x + \epsilon_y + \epsilon_z = \frac{1 - 2\nu}{E} (\sigma_x + \sigma_y + \sigma_z)$$
$$= \frac{1 - \frac{2}{3}}{70 \times 10^3} (70 - 30 - 15)$$
$$= 1.1905 \times 10^{-4}$$

Therefore,

$$\Delta V = eV_0$$
= (1.1905 × 10⁻⁴)(150)(100)(75)
= 134 mm³

Question 3

A rectangular plate is subjected to uniform tensile stress σ along its upper and lower edges as shown in Fig. 2. Determine the displacements u and v in terms of x, y, and material properties (E, ν) using Eqns. (1) and (2) and the appropriate conditions at the origin.

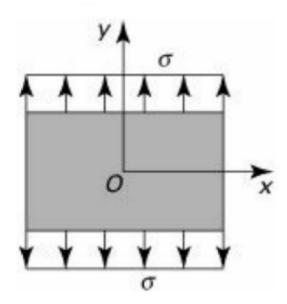


Figure 2: Rectangular plate subjected to uniform tensile stress.

$$\epsilon_x = \frac{\nabla u}{\nabla x}, \ \epsilon_y = \frac{\nabla v}{\nabla y}$$
 (1)

$$\gamma_{xy} = \alpha_x - \alpha_y = \frac{\nabla u}{\nabla y} + \frac{\nabla v}{\nabla x} \tag{2}$$