#### 1 State Models

Consider a system with

- 1. n, the order of the ODE
- 2. m, the number of inputs
- 3. p, the number of outputs

General procedure:

- 1. Create n state variables
- 2. Create n first order ODEs
- 3. Write  $\dot{x} = f(x, u)$
- 4. Write y = h(x, u)

If the system is linear, then the state model is

$$\dot{x} = Ax + Bu$$
$$y = Cx + Du$$

### 2 Numerical Simulation with MATLAB

Example, some random order 4 system with  $x_1(0) = 1$ ,  $x_2(0) = 2$ ,  $u_1 = \cos(t)$ ,  $u_2 = \sin(t)$ .

```
function x_dot = f(t, x)
x_dot = [
    x(3);
    x(4);
    -10*x(1) + 10*x(2) + cos(t)
    10*x(1) - 10*x(2) - sin(t)
];
end

[t, x] = ode45(@f, [0, 10], [1, 2, 0, 0]);
```

### 3 Linearization

Select a point  $(x_0, u_0)$  and to be the equilibrium point. That is,

$$f(x_0, u_0) = 0$$

From the set of equilibrium points, choose the appropriate one to linearize about. Then,

$$A = \frac{\partial f}{\partial x} \Big|_{x=x_0, u=u_0}$$

$$B = \frac{\partial f}{\partial u} \Big|_{x=x_0, u=u_0}$$

$$C = \frac{\partial h}{\partial x} \Big|_{x=x_0, u=u_0}$$

$$D = \frac{\partial h}{\partial u} \Big|_{x=x_0, u=u_0}$$

Example for inverted pendulum on a cart with equilibrium point  $x_0 = (x_10, 0, 0, 0)$  and  $u_0 = 0$ .

```
% Declare symbolic variables
syms x1 x2 x3 x4 u
% Define the system
f = [
    x3;
    x4;
    (4*x4^2*\sin(x2) - 3*\cos(x2)*\sin(x2) + 4*u)/(4 - 3*\cos(x2)^2);
    (-3*x4^2*sin(x2)*cos(x2) + 3*sin(x2) - 3*u*cos(x2))/(4 - 3*cos(x2))
   )^2);
];
h = [x1; x2];
% Compute the Jacobian
dfdx = jacobian(f, [x1, x2, x3, x4])
dfdu = jacobian(f, u)
dhdx = jacobian(h, [x1, x2, x3, x4])
dhdu = jacobian(h, u)
A = subs(dfdx, [x1, x2, x3, x4, u], [x10, 0, 0, 0])
B = subs(dfdu, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
C = subs(dhdx, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
D = subs(dhdu, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
```

## 4 Solutions to Linear Systems

Split the system into two parts: the zero-input response and the zero-state response.

$$x(t) = x_{z-i}(t) + x_{z-s}(t)$$
  
 $y(t) = Cx_{z-i}(t) + Cx_{z-s}(t) + Du(t)$ 

#### 4.1 Zero-Input Response

The zero-input problem is:

$$\dot{x} = Ax$$

$$y = Cx$$

$$x(0) = x_0$$

$$u = 0$$

The solution is

$$x_{z-i}(t) = e^{At}x_0$$
$$y_{z-i}(t) = Ce^{At}x_0$$

#### 4.2 Matrix exponential properties

$$e^{At} = \sum_{k=0}^{\infty} \frac{A^k t^k}{k!}$$

$$e^{At}|_{t=0} = I$$

$$e^{At_1} e^{At_2} = e^{A(t_1 + t_2)}$$

$$e^{A_1 t} e^{A_2 t} = e^{(A_1 + A_2)t} \iff A_1 A_2 = A_2 A_1$$

$$(e^{At})^{-1} = e^{-At}$$

$$A e^{At} = e^{At} A$$

$$\frac{d}{dt} e^{At} = A e^{At} = e^{At} A$$

$$e^{At} = V e^{Dt} V^{-1} = \mathcal{L}^{-1} \{ sI - A \}^{-1}$$

#### 4.3 Zero-State Response

The zero-state problem is:

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

$$x(0) = 0$$

$$u = u(t)$$

By integrating factor method, the solution is:

$$x_{\text{z-s}}(t) = \int_0^t e^{A(t-\tau)} Bu(\tau) d\tau$$
$$y_{\text{z-s}}(t) = \int_0^t Ce^{A(t-\tau)} Bu(\tau) d\tau + Du(t)$$

#### 4.4 Total Response

The total trajectory and response respectively are:

$$x(t) = \underbrace{e^{At}x_0}_{x_{z-i}(t)} + \underbrace{\int_0^t e^{A(t-\tau)}Bu(\tau)d\tau}_{x_{z-s}(t)}$$
$$y(t) = \underbrace{Ce^{At}x_0}_{y_{z-i}(t)} + \underbrace{\int_0^t Ce^{A(t-\tau)}Bu(\tau)d\tau + Du(t)}_{y_{z-s}(t)}$$

# 5 Laplace Transform Method

### 5.1 Laplace Transform and Properties

$$\mathcal{L}f(t) = F(s) = \int_0^\infty f(t)e^{-st}dt$$

$$\mathcal{L}\dot{f}(t) = sF(s) - f(0)$$

$$\mathcal{L}\ddot{f}(t) = s^2F(s) - sf(0) - \dot{f}(0)$$

$$\mathcal{L}\left\{\int_0^t f(\tau)d\tau\right\} = \frac{F(s)}{s}$$

$$\mathcal{L}f(t - t_d) = e^{-st_d}F(s)$$

## 5.2 Poles and Convergence

- In general, the right most pole determines the region of convergence.
- Use the analogy that the real part of poles correspond to the expontential decay rate of the system and the imaginary part corresponds to the oscillation frequency.
- Repeated poles is correspond to  $te^{-at}$  or  $t\sin(at)$ .
- idk i might add fvt later

#### 5.3 Solution to State Space Model

The trajectory and response respectively are:

$$x(t) = \underbrace{\mathcal{L}^{-1}\{(sI - A)^{-1}x_0\}}_{x_{z-i}(t)} + \underbrace{\mathcal{L}^{-1}\{(sI - A)^{-1}BU(s)\}}_{x_{z-s}(t)}$$
$$y(t) = \underbrace{\mathcal{L}^{-1}\{C(sI - A)^{-1}x_0\}}_{y_{z-i}(t)} + \underbrace{\mathcal{L}^{-1}\{[C(sI - A)^{-1}B + D]U(s)\}}_{y_{z-s}(t)}$$

#### 5.4 Transfer Function

The transfer function is defined from:

$$Y(s) = \underbrace{C(sI - A)^{-1}x_0}_{Y_{z-i}(s)} + \underbrace{\left[C(sI - A)^{-1}B + D\right]U(s)}_{Y_{z-s}(s)}$$
$$G(s) = C(sI - A)^{-1}B + D$$

## 6 Step Response

#### 6.1 First Order System

The standard first-order system is described by the following transfer function:

$$G(s) = \frac{K}{\tau s + 1}$$

For a step response  $u = u_0(t) \implies U = 1/s$ , the output is:

$$Y(s) = \frac{K}{\tau s + 1} \frac{1}{s}$$
$$y(t) = K(1 - e^{-t/\tau})$$

To find the time constant  $\tau$  and the gain K, we can use some properties:

$$\lim_{t \to \infty} y(t) = K(1 - e^{-t/\tau}) = K$$
$$y(\tau) = K(1 - e^{-1}) = 0.632K = 63.2\%K$$

## 6.2 Second Order System

The standard second-order system is described by the following transfer function:

$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

For a step response  $u = \alpha u_0(t) \implies U = 1/s$ , the response can be obtained by using  $\omega_d = \omega_n \sqrt{1 - \zeta^2}$ :

$$y(t) = \alpha - \alpha e^{-\zeta \omega_n t} \left[ \cos(\omega_d t) + \frac{\zeta \omega_n}{\omega_d} \sin(\omega_d t) \right]$$

To determine  $\zeta$  and  $\omega_n$ , define overshoot  $M_p = (y_{\text{max}} - y_{\infty})/y_{\infty}$  and peak time  $t_p \implies y(t_p) = y_{\text{max}}$ .

$$\zeta = \frac{(\ln(M_p))^2}{\pi^2 + (\ln(M_p))^2}$$
$$\omega_n = \frac{\pi}{t_p \sqrt{1 - \zeta^2}}$$

# 7 Impulse Response

Recall the Dirac delta function  $\delta(t)$  is defined as:

$$\int_{-\infty}^{\infty} \delta(t)dt = 1$$
$$\delta(t) = 0 \quad \forall t \neq 0$$
$$\int_{-\infty}^{\infty} f(t)\delta(t - t_0)dt = f(t_0)$$

The zero-state response to a impulse input  $u = \delta(t)$  is:

$$g(t) = \int_0^t Ce^{A(t-\tau)}B\delta(\tau)d\tau + D\delta(t)$$
 
$$= Ce^{At}B + D\delta(t)$$

The Laplace transform of the impulse response is:

$$G(s) = C(sI - A)^{-1}B + D$$

This is no coincidence. The transfer function is the Laplace transform of the impulse response.

By convolution,

$$y_{\text{z-s}}(t) = (g * u)(t)$$

$$= \int_0^t g(t - \tau)u(\tau)d\tau$$

$$= \int_0^t Ce^{A(t - \tau)}Bu(\tau)d\tau + Du(t)$$

Which is exactly the zero-state response obtained using the time-domain integrating factor method.

#### 8 Realization

to do later cause lazy

# 9 Stability

## 9.1 Internal Stability

Internal stability is concerned with unforced (u = 0) systems. This corresponds to the trajectory  $x_{z-i}(t)$  and the response  $y_{z-i}(t)$ .

For x(0), the system is internally **stable** about an equilibrium point  $x_e$  if and only if for every  $\epsilon > 0$ , there exists a  $\delta > 0$  such that:

$$||x(0) - x_e|| < \delta \implies ||x(t) - x_e|| < \epsilon, \quad \forall t \ge 0$$

A system is said to be **asymptotically stable** if it is stable and  $\lim_{t\to\infty} x(t) = x_e$ .

For linear systems, the system is internally stable if and only if every eigenvalue of A has a negative real part. Re( $\lambda_i$ ) < 0,  $\forall i = 1, ... n$ .

#### 9.2 BIBO Stability

BIBO (bounded-input bounded-output) stability is concerned with zero state systems. This corresponds to the trajectory  $x_{z-s}(t)$  and the response  $y_{z-s}(t)$ .

Assume G(s) is a rational and proper transfer function. Then, the system is BIBO stable if and only if every pole of G(s) has a negative real part.  $Re(pole_i) < 0, \forall i = 1, ... n$ .

If the system is not stable, then there exists a bounded input u(t) such that the output y(t) is unbounded. Not all bounded inputs will cause an unstable BIBO system to be unbounded.

#### 9.3 Connection between Internal and BIBO Stability

For a linear system, internal stability implies BIBO stability. However, BIBO stability does not imply internal stability. This is because the poles of G(s) are a subset of the eigenvalues of A.

asymptotic stability  $\implies$  BIBO stability

#### 9.4 Closed Loop Systems

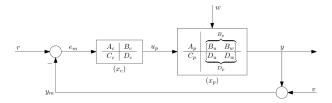


Figure 1: Closed loop system

Assume:

- Both controller  $(A_c, B_c, C_c, D_c)$  and plant  $(A_p, B_p, C_p, D_p)$  are minimal realizations.
- $D_c = 0$  or  $D_p = [D_u, D_w] = 0$ , such that  $D_c D_u = D_u D_c = 0$  and  $D_c D_w = D_w D_c = 0$ .

Then, the state-space form of the closed loop system is:

$$\dot{x}_{c}l = \begin{bmatrix} \dot{x}_{c} \\ \dot{x}_{p} \end{bmatrix} = \underbrace{\begin{bmatrix} A_{c} - B_{c}D_{u}C_{c} & -B_{c}C_{p} \\ B_{u}C_{c} & A_{p} - B_{u}D_{c}C_{p} \end{bmatrix}}_{A_{cl}} \begin{bmatrix} x_{c} \\ x_{p} \end{bmatrix} + \underbrace{\begin{bmatrix} B_{c} & -B_{c}D_{w} & -B_{c} \\ B_{u}D_{c} & B_{w} & -B_{u}D_{c} \end{bmatrix}}_{B_{cl}} \begin{bmatrix} r \\ w \\ v \end{bmatrix}$$

$$y_{c}l = \begin{bmatrix} e_{m} \\ u_{p} \\ y \\ y_{m} \end{bmatrix} = \underbrace{\begin{bmatrix} -D_{u}C_{c} & -C_{p} \\ C_{c} & -D_{c}C_{p} \\ D_{u}C_{c} & C_{p} \\ D_{u}C_{c} & C_{p} \end{bmatrix}}_{C_{cl}} \begin{bmatrix} x_{c} \\ x_{p} \end{bmatrix} + \underbrace{\begin{bmatrix} 1 & -D_{w} & -1 \\ D_{c} & 0 & -D_{c} \\ 0 & D_{w} & 0 \\ 0 & D_{w} & 1 \end{bmatrix}}_{D_{cl}} \begin{bmatrix} r \\ w \\ v \end{bmatrix}$$

Casting to transfer functions where

$$G_c(s) := C_c(sI - A_c)^{-1}B_c + D_c$$
  

$$G(s) := C_p(sI - A_p)^{-1}B_p + D_p := \begin{bmatrix} G_p(s) & G_d(s) \end{bmatrix}$$

Then the system becomes:

$$\begin{bmatrix} E_m \\ U_p \\ Y \\ Y_m \end{bmatrix} = \begin{bmatrix} \frac{1}{1+G_pG_c} & \frac{G_d}{1+G_pG_c} & \frac{-1}{1+G_pG_c} \\ \frac{G_c}{G_c} & \frac{-G_cG_d}{1+G_pG_c} & \frac{-G_c}{1+G_pG_c} \\ \frac{G_pG_c}{1+G_pG_c} & \frac{G_d}{1+G_pG_c} & \frac{-G_pG_c}{1+G_pG_c} \\ \frac{G_pG_c}{1+G_pG_c} & \frac{G_d}{1+G_pG_c} & \frac{-1}{1+G_pG_c} \end{bmatrix} \begin{bmatrix} R \\ W \\ V \end{bmatrix}$$

## 9.5 Unmodeled Disturbance Input

Most of the time we just assume that disturbance is added to the input  $u_p$ . In the transfer function matrix,  $G_d = G_p$ .

# 9.6 Internal Stability of Closed Loop Systems

The closed loop system is internally stable if and only if every eigenvalue of  $A_{cl}$  has a negative real part.

### 9.7 I/O Stability of Closed Loop Systems

The closed loop system is I/O stable if and only if every entry of the transfer function matrix is BIBO stable. There are 5 unique transfer functions to be considered:

$$\frac{1}{1 + G_p G_c}$$

$$\frac{G_c}{1 + G_p G_c}$$

$$\frac{G_p G_c}{1 + G_p G_c}$$

$$\frac{G_d}{1 + G_p G_c}$$

$$\frac{G_c G_d}{1 + G_p G_c}$$

If the disturbance is added to the input, then  $G_d = G_p$  and then we only need to check:

$$\frac{1}{1 + G_p G_c}$$

$$\frac{G_c}{1 + G_p G_c}$$

$$\frac{G_p G_c}{1 + G_p G_c}$$

$$\frac{G_p}{1 + G_p G_c}$$

#### 9.7.1 Characteristic Polynomial for Closed Loop Systems

Define the numerator and denominator of the transfer functions:

$$G_c := \frac{n_c}{d_c}, \quad G := [G_p, G_d] := [\frac{n_p}{d_n}, \frac{n_d}{d_d}]$$

The characteristic polynomial is:

$$P = n_p n_c + d_p d_c$$

By Theorem 4.4.1, the closed loop system is I/O stable if and only if all roots of P have a negative real part.

#### 9.7.2 Theorem 4.4.2

The closed loop system is I/O stable if and only if

- 1.  $1 + G_pG_c$  has only negative real part roots
- 2.  $G_pG_c$  has no pole-zero cancellations