1 State Models

Consider a system with

- 1. n, the order of the ODE
- 2. m, the number of inputs
- 3. p, the number of outputs

General procedure:

- 1. Create n state variables
- 2. Create n first order ODEs
- 3. Write $\dot{x} = f(x, u)$
- 4. Write y = h(x, u)

If the system is linear, then the state model is

$$\dot{x} = Ax + Bu$$
$$y = Cx + Du$$

2 Numerical Simulation with MATLAB

Example, some random order 4 system with $x_1(0) = 1$, $x_2(0) = 2$, $u_1 = \cos(t)$, $u_2 = \sin(t)$.

```
function x_dot = f(t, x)
x_dot = [
    x(3);
    x(4);
    -10*x(1) + 10*x(2) + cos(t)
    10*x(1) - 10*x(2) - sin(t)
];
end

[t, x] = ode45(@f, [0, 10], [1, 2, 0, 0]);
```

3 Linearization

Select a point (x_0, u_0) and to be the equilibrium point. That is,

$$f(x_0, u_0) = 0$$

From the set of equilibrium points, choose the appropriate one to linearize about. Then,

$$A = \frac{\partial f}{\partial x} \Big|_{x=x_0, u=u_0}$$

$$B = \frac{\partial f}{\partial u} \Big|_{x=x_0, u=u_0}$$

$$C = \frac{\partial h}{\partial x} \Big|_{x=x_0, u=u_0}$$

$$D = \frac{\partial h}{\partial u} \Big|_{x=x_0, u=u_0}$$

Example for inverted pendulum on a cart with equilibrium point $x_0 = (x_10, 0, 0, 0)$ and $u_0 = 0$.

```
% Declare symbolic variables
syms x1 x2 x3 x4 u
% Define the system
f = [
    x3;
    x4;
    (4*x4^2*\sin(x2) - 3*\cos(x2)*\sin(x2) + 4*u)/(4 - 3*\cos(x2)^2);
    (-3*x4^2*sin(x2)*cos(x2) + 3*sin(x2) - 3*u*cos(x2))/(4 - 3*cos(x2))
   )^2);
];
h = [x1; x2];
% Compute the Jacobian
dfdx = jacobian(f, [x1, x2, x3, x4])
dfdu = jacobian(f, u)
dhdx = jacobian(h, [x1, x2, x3, x4])
dhdu = jacobian(h, u)
A = subs(dfdx, [x1, x2, x3, x4, u], [x10, 0, 0, 0])
B = subs(dfdu, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
C = subs(dhdx, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
D = subs(dhdu, [x1, x2, x3, x4, u], [x10, 0, 0, 0, 0])
```

4 Solutions to Linear Systems

Split the system into two parts: the zero-input response and the zero-state response.

$$x(t) = x_{z-i}(t) + x_{z-s}(t)$$

 $y(t) = Cx_{z-i}(t) + Cx_{z-s}(t) + Du(t)$

4.1 Zero-Input Response

The zero-input problem is:

$$\dot{x} = Ax$$

$$y = Cx$$

$$x(0) = x_0$$

$$u = 0$$

The solution is

$$x_{z-i}(t) = e^{At}x_0$$
$$y_{z-i}(t) = Ce^{At}x_0$$

4.2 Matrix exponential properties

$$e^{At} = \sum_{k=0}^{\infty} \frac{A^k t^k}{k!}$$

$$e^{At}|_{t=0} = I$$

$$e^{At_1} e^{At_2} = e^{A(t_1 + t_2)}$$

$$e^{A_1 t} e^{A_2 t} = e^{(A_1 + A_2)t} \iff A_1 A_2 = A_2 A_1$$

$$(e^{At})^{-1} = e^{-At}$$

$$A e^{At} = e^{At} A$$

$$\frac{d}{dt} e^{At} = A e^{At} = e^{At} A$$

$$e^{At} = V e^{Dt} V^{-1} = \mathcal{L}^{-1} \{ sI - A \}^{-1}$$

4.3 Zero-State Response

The zero-state problem is:

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

$$x(0) = 0$$

$$u = u(t)$$

By integrating factor method, the solution is:

$$x_{\text{z-s}}(t) = \int_0^t e^{A(t-\tau)} Bu(\tau) d\tau$$
$$y_{\text{z-s}}(t) = \int_0^t Ce^{A(t-\tau)} Bu(\tau) d\tau + Du(t)$$

4.4 Total Response

The total trajectory and response respectively are:

$$x(t) = \underbrace{e^{At}x_0}_{x_{z-i}(t)} + \underbrace{\int_0^t e^{A(t-\tau)}Bu(\tau)d\tau}_{x_{z-s}(t)}$$
$$y(t) = \underbrace{Ce^{At}x_0}_{y_{z-i}(t)} + \underbrace{\int_0^t Ce^{A(t-\tau)}Bu(\tau)d\tau + Du(t)}_{y_{z-s}(t)}$$

5 Laplace Transform Method

5.1 Laplace Transform and Properties

$$\mathcal{L}f(t) = F(s) = \int_0^\infty f(t)e^{-st}dt$$

$$\mathcal{L}\dot{f}(t) = sF(s) - f(0)$$

$$\mathcal{L}\ddot{f}(t) = s^2F(s) - sf(0) - \dot{f}(0)$$

$$\mathcal{L}\left\{\int_0^t f(\tau)d\tau\right\} = \frac{F(s)}{s}$$

$$\mathcal{L}f(t - t_d) = e^{-st_d}F(s)$$

5.2 Poles and Convergence

- In general, the right most pole determines the region of convergence.
- Use the analogy that the real part of poles correspond to the expontential decay rate of the system and the imaginary part corresponds to the oscillation frequency.
- Repeated poles is correspond to te^{-at} or $t\sin(at)$.
- idk i might add fvt later

5.3 Solution to State Space Model

The trajectory and response respectively are:

$$x(t) = \underbrace{\mathcal{L}^{-1}\{(sI - A)^{-1}x_0\}}_{x_{z-i}(t)} + \underbrace{\mathcal{L}^{-1}\{(sI - A)^{-1}BU(s)\}}_{x_{z-s}(t)}$$
$$y(t) = \underbrace{\mathcal{L}^{-1}\{C(sI - A)^{-1}x_0\}}_{y_{z-i}(t)} + \underbrace{\mathcal{L}^{-1}\{[C(sI - A)^{-1}B + D]U(s)\}}_{y_{z-s}(t)}$$

5.4 Transfer Function

The transfer function is defined from:

$$Y(s) = \underbrace{C(sI - A)^{-1}x_0}_{Y_{z-i}(s)} + \underbrace{\left[C(sI - A)^{-1}B + D\right]U(s)}_{Y_{z-s}(s)}$$
$$G(s) = C(sI - A)^{-1}B + D$$

6 Step Response

6.1 First Order System

The standard first-order system is described by the following transfer function:

$$G(s) = \frac{K}{\tau s + 1}$$

For a step response $u = u_0(t) \implies U = 1/s$, the output is:

$$Y(s) = \frac{K}{\tau s + 1} \frac{1}{s}$$
$$y(t) = K(1 - e^{-t/\tau})$$

To find the time constant τ and the gain K, we can use some properties:

$$\lim_{t \to \infty} y(t) = K(1 - e^{-t/\tau}) = K$$
$$y(\tau) = K(1 - e^{-1}) = 0.632K = 63.2\%K$$

6.2 Second Order System

The standard second-order system is described by the following transfer function:

$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

For a step response $u = \alpha u_0(t) \implies U = 1/s$, the response can be obtained by using $\omega_d = \omega_n \sqrt{1 - \zeta^2}$:

$$y(t) = \alpha - \alpha e^{-\zeta \omega_n t} \left[\cos(\omega_d t) + \frac{\zeta \omega_n}{\omega_d} \sin(\omega_d t) \right]$$

To determine ζ and ω_n , define overshoot $M_p = (y_{\text{max}} - y_{\infty})/y_{\infty}$ and peak time $t_p \implies y(t_p) = y_{\text{max}}$.

$$\zeta = \frac{(\ln(M_p))^2}{\pi^2 + (\ln(M_p))^2}$$
$$\omega_n = \frac{\pi}{t_p \sqrt{1 - \zeta^2}}$$

7 Impulse Response

Recall the Dirac delta function $\delta(t)$ is defined as:

$$\int_{-\infty}^{\infty} \delta(t)dt = 1$$
$$\delta(t) = 0 \quad \forall t \neq 0$$
$$\int_{-\infty}^{\infty} f(t)\delta(t - t_0)dt = f(t_0)$$

The zero-state response to a impulse input $u = \delta(t)$ is:

$$g(t) = \int_0^t Ce^{A(t-\tau)}B\delta(\tau)d\tau + D\delta(t)$$

$$= Ce^{At}B + D\delta(t)$$

The Laplace transform of the impulse response is:

$$G(s) = C(sI - A)^{-1}B + D$$

This is no coincidence. The transfer function is the Laplace transform of the impulse response.

By convolution,

$$y_{\text{z-s}}(t) = (g * u)(t)$$

$$= \int_0^t g(t - \tau)u(\tau)d\tau$$

$$= \int_0^t Ce^{A(t - \tau)}Bu(\tau)d\tau + Du(t)$$

Which is exactly the zero-state response obtained using the time-domain integrating factor method.

8 Realization

to do later cause lazy

9 Stability

9.1 Internal Stability

Internal stability is concerned with unforced (u = 0) systems. This corresponds to the trajectory $x_{z-i}(t)$ and the response $y_{z-i}(t)$.

For x(0), the system is internally **stable** about an equilibrium point x_e if and only if for every $\epsilon > 0$, there exists a $\delta > 0$ such that:

$$||x(0) - x_e|| < \delta \implies ||x(t) - x_e|| < \epsilon, \quad \forall t \ge 0$$

A system is said to be **asymptotically stable** if it is stable and $\lim_{t\to\infty} x(t) = x_e$.

For linear systems, the system is internally stable if and only if every eigenvalue of A has a negative real part. Re(λ_i) < 0, $\forall i = 1, ... n$.

9.2 BIBO Stability

BIBO (bounded-input bounded-output) stability is concerned with zero state systems. This corresponds to the trajectory $x_{z-s}(t)$ and the response $y_{z-s}(t)$.

Assume G(s) is a rational and proper transfer function. Then, the system is BIBO stable if and only if every pole of G(s) has a negative real part. $Re(pole_i) < 0, \forall i = 1, ... n$.

If the system is not stable, then there exists a bounded input u(t) such that the output y(t) is unbounded. Not all bounded inputs will cause an unstable BIBO system to be unbounded.

9.3 Connection between Internal and BIBO Stability

For a linear system, internal stability implies BIBO stability. However, BIBO stability does not imply internal stability. This is because the poles of G(s) are a subset of the eigenvalues of A.

asymptotic stability \implies BIBO stability

9.4 Closed Loop Stability

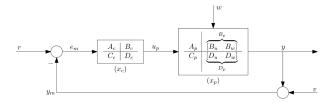


Figure 1: Closed loop system

Assume:

- Both controller (A_c, B_c, C_c, D_c) and plant (A_p, B_p, C_p, D_p) are minimal realizations.
- $D_c = 0$ or $D_p = [D_u, D_w] = 0$, such that $D_c D_u = D_u D_c = 0$ and $D_c D_w = D_w D_c = 0$.

Then, the state-space form of the closed loop system is:

$$\dot{x}_{c}l = \begin{bmatrix} \dot{x}_{c} \\ \dot{x}_{p} \end{bmatrix} = \underbrace{\begin{bmatrix} A_{c} - B_{c}D_{u}C_{c} & -B_{c}C_{p} \\ B_{u}C_{c} & A_{p} - B_{u}D_{c}C_{p} \end{bmatrix}}_{A_{cl}} \begin{bmatrix} x_{c} \\ x_{p} \end{bmatrix} + \underbrace{\begin{bmatrix} B_{c} & -B_{c}D_{w} & -B_{c} \\ B_{u}D_{c} & B_{w} & -B_{u}D_{c} \end{bmatrix}}_{B_{cl}} \begin{bmatrix} r \\ w \\ v \end{bmatrix}$$

$$y_{c}l = \begin{bmatrix} e_{m} \\ u_{p} \\ y \\ y_{m} \end{bmatrix} = \underbrace{\begin{bmatrix} -D_{u}C_{c} & -C_{p} \\ C_{c} & -D_{c}C_{p} \\ D_{u}C_{c} & C_{p} \\ D_{u}C_{c} & C_{p} \end{bmatrix}}_{C_{cl}} \begin{bmatrix} x_{c} \\ x_{p} \end{bmatrix} + \underbrace{\begin{bmatrix} 1 & -D_{w} & -1 \\ D_{c} & 0 & -D_{c} \\ 0 & D_{w} & 0 \\ 0 & D_{w} & 1 \end{bmatrix}}_{D_{cl}} \begin{bmatrix} r \\ w \\ v \end{bmatrix}$$