2

Contorol System Analysis in State Space

*State-Space Representations of Tononsfer-Function System

(1) State-Space Reporesentations in Canonical

=> Consider a system defined by:

g + a,y + -- any + any = bou + b,u + -- +bn, u+bnu

where is the Imput and y is the output.

$$\frac{y(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + \dots + b_{m-1} s + b_m}{s^n + a_1 s^{n-1} + \dots + a_{m-1} s + a_m}$$

Controllable Canonical Form

$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{1} \\ \vdots \\ \dot{x}_{m-1} \\ \dot{x}_{m} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & \cdots & 1 \\ \vdots & \vdots & \vdots & \vdots \\ 0 & 1 &$$

 $y = \begin{bmatrix} b_n - a_n b_0 \\ b_n - a_n b_0 \end{bmatrix} \begin{bmatrix} b_n - a_n b_0 \\ b_n - a_n b_0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_n \end{bmatrix} + b_0 u$

=> The Controllable Canonical form is important in discussing the pole-placement approach to Control System design.

Observable Canonical form



$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \vdots \\ \dot{x}_{n} \end{bmatrix} = \begin{bmatrix} 0 & 0 & --- & 0 & --- & 0 & --- & 0 \\ 1 & 0 & --- & 0 & --- & 0 \\ 1 & 0 & --- & 0 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ 1 & \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ 2 & 0 & 0 & --- & 1 & --- & 0 \\$$

$$y = [00 - - 01] \begin{bmatrix} x_1 \\ x_2 \\ x_n \end{bmatrix} + b_0 u$$



$$\frac{Y(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + - - - + b_{n-1} s + b_n}{(s + P_1)(s + P_2) - - - - (s + P_n)}$$

All distinct noots

= bo +
$$\frac{C_1}{S+P_1}$$
 + $\frac{C_2}{S+P_1}$ + $--\frac{C_m}{S+P_m}$
=> The diagond Cononical from of the state space
grapsesentation of this system is given by:

$$\begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \dot{x}_{n} \end{bmatrix} = \begin{bmatrix} -\rho_{1} \\ -\rho_{2} \\ \dot{x}_{n} \end{bmatrix} \begin{bmatrix} \chi_{1} \\ \chi_{2} \\ \vdots \\ \chi_{m} \end{bmatrix} + \begin{bmatrix} 1 \\ 1 \\ \vdots \\ \chi_{m} \end{bmatrix} U$$

$$y = [C, C_2 \cdots C_m] \begin{bmatrix} x_1 \\ x_2 \\ x_n \end{bmatrix} + b_0 u$$

Josedan Cononical Form

=> Case where denomination polynomia Contains multiple stoots.

Suppose,

$$\frac{Y(s)}{U(s)} = \frac{b_0 s^{n} + b_1 s^{n-1} + \dots + b_{m_1} s + b_m}{(s + p_1)^3 + (s + p_n) + (s + p_s) - \dots + (s + p_m)}$$

=> State-space graphesentation in the Jordan Cononical form is given by.

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_3 \\ \vdots \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} -\rho_1 & 0 & 0 & -\frac{1}{2} & 0 \\ 0 & -\rho_1 & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & -\frac{1}{2} & 0 & 0 \\ 0 & 0 & 0 & -\frac{1}{2} & 0 \\ 0 & 0 & 0 & -\frac{1$$

$$y = \begin{bmatrix} C, & C_{n} \end{bmatrix} \times_{1}^{x_{1}} + b_{0} U$$

* Eigenvalues of an non Matorix A

> The eigenvalues of an nxnmabir A are the moots of the Chanasteristic equalion

1 x I - A 1 = 0

* Diagonalization of non Matrix

=> If non methorix A with distinct eigenvolves is given by

$$A = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 1 \\ 0 & 0 & 0 & \cdots & 1 \\ -q_m - q_m & \cdots & -q_n \end{bmatrix}$$

the transformation X=PZ, where

$$\rho = \begin{bmatrix} 1 & 1 & --- & 1 \\ x_1 & x_2 & x_3 & --- & x_n \\ x_1^2 & x_2^2 & x_3^2 & --- & x_n^2 \\ \vdots & \vdots & \vdots & \vdots & \vdots \\ x_n^{n-1} & x_n^{n-1} & \dots & x_n^{n-1} \end{bmatrix}$$

Swhere >1, 12--->n = n distincts
Eigenvalues of A

Will transform P'AP into the diagond matrix.

$$P^{-1}AP = \begin{bmatrix} \lambda_1 & \lambda_2 \\ 0 & \lambda_3 \end{bmatrix}$$

当耳

= 4

l L

* 9

To

Chid

Scanned by CamScanner

- => If the metrix A involves multiple eigenvalues, then diagonalization is impossible.
- De shall prove this for a general case in what follows.

* Anvariance of Ligen Values

To prove the invariance of the eigenvalue under a linear transformation, we must show that the Characteristic polynomials INI-Al and INI-P-IAPI are identical.

$$\begin{aligned} |\lambda I - P^{-1}AP| &= |\lambda P^{-1}P - P^{-1}AP| \\ &= |P^{-1}(\lambda I - A)P| \\ &= |P^{-1}||\lambda I - A||P| \\ &= |P^{-1}||P||\lambda I - A| \\ &= |P^{-1}P||\lambda I - A| \\ &= |\lambda I - A| \end{aligned}$$

*Nonuniqueness of Set of State Variables

Let X, Xz--- Xn are a Set of State Variables.

=> Then we may take as another set of state Variables any set of functions.

$$\hat{x}_{1} = x_{1} (\alpha_{1} \alpha_{2} - - \alpha_{m})$$

$$\hat{x}_{2} = x_{2} (\alpha_{1} x_{2} - - \alpha_{m})$$

$$\hat{x}_{3} = x_{2} (\alpha_{1} x_{2} - - \alpha_{m})$$

$$\hat{x}_{4} = x_{4} (x_{1} x_{2} - - x_{m})$$

Kenovided that for every sot of value 2, , 22 --- In , there Cornes ponds to a wigner set of volues x, x2 ... Xn (and Vice-Vensa

=> Thus, if x is a state vactor, then &, where | x - Px | is also a state vector, provided the matrix Pis nonsingular. -> Different State Vectoors Convey the Same information don't the System * Lolving the time-invariant State # Solving Homogeneous State Equations × X=> (Sefore we solve Vector-matrix differential eguctions, let us or eview the solution of the scalar differential equation. $\times = \alpha \times$ => In solving this equalion, we may assume a solution ox (4) of the form. X(E) = bo + b, E + b2 +2 + . - ~ bktk + . -. => By substituting this assumed solution we obtain : b, +2 b, + + 3 b, +2+ + KbKtK-1+... = a (bo+b, t+bnt2+ -- ... bx tk+--) => If the assumed solution is to be true solution , the above equation must hold for any E. b1 = abo b2= 2ab1=2abo

$$b_{K} = \frac{1}{K!} A^{K} b_{0}$$

$$S_{0} \times (t) = \left(I + At + \frac{1}{2!} A^{2} t^{2} + \dots + \frac{1}{K!} A^{K} t^{K} + \dots \right) \times (0)$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow \left[\times (t) = e^{At} \times (0) \right]$$

$$\Rightarrow$$

⇒In particular if S=-t them eAteAt = e-AteAt = RA(t-t) = I => This the invest of eat is e-At. = (eat) = e-At => Since the inverse of eat duary exists, estion non singula. one-lid = e(A+B)t = eAt ent) if AD = BAS = e(AID) = eAleBE (4 AD = BA) # Laplace Tonansform apponoach to the Soldier
of Homogeneous State Randien $\dot{\times}$ (t) = $A \times (t)$ $8\times(s)-\infty(0)=A\times(s)$ $\left\{\times(s)=\frac{1}{2}\left[x(t)\right]\right\}$ \times (s) = (SI-A) $^{-1}$ \times (0) x (E) = 1-1 (SI-A)-1×(O) =) (SI-A) == == + A+ A+ + --> 2 (SI-A)-1) = I + A + ALL + ALL + ALL + ALL + ... * State-Tononsition matrix We can write solution of the homogeneous Stale candion State trasition $\dot{\times} = A \times$ $(as) \times (t) = \mathcal{O}(t) \times (0)$

× (0)

Where collis on nen motoris and to the U-ione solution of CP(U)=A(CP(U) =I $\times(o) = \mathcal{O}(o) \times (o) = \times(o)$ c((E) = ene = f-1/(SI-N)-1/ Q-1(t)= e-11= cp(-1) => Q(b) is called the state A are distinct, than CQ(6) will contain the n axponantido. exit, exit exit

Ether.

C((6) = eAt = [ent end of emb]

=> If there is a multiplicity in the eigenvalues then cold) will (on lain, in addition to the exponentials exil, eight, - eight, term like text ad their.

* Penoperties of State-Tenansition Matrices

2.
$$Q(t) = e^{At} = (e^{-At})^{-1} = [Q(-t)]^{-1}$$
 on $Q^{-1}(t) = Q(t)$

Example 9-5: Obtain the State-transition matrix Q(t) of the following System:

$$\begin{bmatrix} \dot{\alpha}_1 \\ \dot{\alpha}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \begin{bmatrix} \alpha_1 \\ \alpha_2 \end{bmatrix}$$

$$\Rightarrow$$
 A= $\begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix}$

$$P(t) = e^{At} = 1^{-1} [(SI - A)^{-1}]$$

$$SI-A = \begin{bmatrix} S & O \\ O & S \end{bmatrix} - \begin{bmatrix} O & 1 \\ -2 & -5 \end{bmatrix} = \begin{bmatrix} S & -1 \\ 2 & S+3 \end{bmatrix}$$

$$= \frac{1}{s(s+3)+2} \begin{bmatrix} s+3 & -2 \\ +1 & s \end{bmatrix}$$

$$= \frac{1}{(S+1)(S+2)} \begin{bmatrix} S+3 & 1 \\ -2 & 1 \end{bmatrix}$$

$$= \frac{(5+3)}{(5+1)(5+1)} \frac{1}{(5+1)(5+1)} \frac{1}{(5+1)(5+1)} \frac{2}{(5+1)(5+1)}$$

Scanned by CamScanner

* Some useful Results in Vector matrix Analysis # M # Cayley - Hamilton Theorem 17 Consider a non matrix A and its characteristic egudion: =7 | >I-A|= x + a, x + --- + a -, x + a = 0 The Chyley-Hamilton theorem states that the matrix A satisfies its own characteristic equation An + a. An-1+ -- + an-, A + an I = 0 => To prove this theorem note that adj (XI-A) is a Polynomid in i of degree in-1 adi (XI-A) = B, x-1 + Bzx-2 + -- -- + Dn-, x + Bn where B = I Since, (>I-A) ad; (XI-A) = [ad; (>I-A)] (>I-A) = |>I-A|I wo obtain 1xI-A)I= Ix +a, Ix + --- +a, Ix + an I = (xI-A) (B, xn-1+ Qxx-7+-- Bm, x + Bm) = (B, x-1+Bxx-2+--+Bm, x+Bm) (xI-A) ⇒ If A is substituted for > in this last equation , then Clearly NI-A become Zero. we obtain An-1 A+ an I = 0

Minimal polynomial



The least degree polynomid having A as a scot is colled the minimal polynomial

Alimind polynomid of a norm matrix A is defined as the polynomid $\varphi(n)$ of least dogice $\varphi(n) = \sum_{i=1}^{m} + \alpha_i \sum_{i=1}^{m} - \dots + \alpha_{m-1} \sum_{i=1}^{m} + \alpha_m \sum_{i=1}^{m} - \dots + \alpha_{m-1} \sum_{i=1}^{m}$

Such that of (A) = 0

The minimal polynomial plays an important sole in the Computation of polynomials in a non matrix.

=> Lot us suppose that d(x), a polynomid in x, is the gradest common divisor of all the elements of adi (xI-A).

Lowe can show that if the coefficient of the in highest - degree term in it of d(n) is chosen so I highest - degree term in it of d(n) is chosen so I , then the minimal polynomial al(n) is given by

 $Q(x) = \frac{|x - A|}{d(x)}$

Minimal polynomial Q(x) of an non matrix A can be determined by the following procedure:

1. Forum ads (NI-A) and won't e the elements of ads(NI-A) as fectored polynomid in ?

2. Determine d(r) on the greatest Common divisor of all the elements of celi (NI-A): Choose the coefficient of the highest degree tomin in it of d(r) to be 1. If the highest degree divisor, d(r) = 1.

There is no common divisor, d(r) = 1.

3. The minimal polonomidi

$$Q(x) = \frac{1 \times I - A}{d(x)}$$

* Materix Exponential eAt Co Matlab provides a simple way to compute est , where Tip a Constat. 00 #Computation of eAt: Method 1 If materix A can be transformed into a diagonal from, then et can be given by eAt = Peptp-1 = P Where Pis a diagonalizing metrix for A. ⇒If metrix A (a be transfoomed into a Joshdon canonical form, then eAt can be eAt _ Se^{It} S-1 S= transformation is matrix that transforms matrix A into a Joseph Canonical form J given by # Computation of eAt: Method? => The Second method of computing eAt uses the Laplace transform approach. eAt = 1-1 [(SI-A)-1]

Computation of eA+: Method3

=> This method is based on Sylvester's Interpolation method.

 \leq

Cose1: Minimal polynomial of A Involves Only Distinct

= We shall assume that the degree of the minimal Polynomid of A'sm.

=> By using Sylvester's interpolation formula, it can be shown that eAt can be obtained by solving the following

determinat equalion.

=> Solving above equation from et is the Same as:

Where dx (t) can be determined by solving the following Sct of megnations.

αο(t) + α,(t) λ2 +α2(1) λ2 + ---- + αm-1(t) λ2 = ext \(\lambda_0(t) \) + \(\lambda_1(t) \) \(\lambda_m + - \cdot \) + \(\lambda_m \) = \(\lambda_m \) = \(\lambda_m \) = \(\lambda_m \) = \(\lambda_m \) \(\lambda_m \) = \(\lambda_

Cose 2: Minimal Polynomial of A Involves Multiple Roots

- Delynomial of A involves three cand roots (>=> >> >>) that are delisting.
- > By applying Sylvester's interpolation formula it Can be shown that eat can be obtained from the following determinant equation.

=> Note that if the minimal polynomial of A is not found, it is possible to substitute the Characteristic Polynomial for the minimal polynomial-

* Linear Andependence of Vector

=> The vectors X, X2 --- Xn are said to be linearly independent if

$$C_{1} \times_{1} + C_{2} \times_{2} + C_{3} \times_{3} + \cdots + C_{m} \times_{m} = 0$$

Where C., Cz --- (m are Constat, implise that

#

- The Vectors X, X2 --- Xn are said to be linearly dependent if and Only if Xi Can be expensed as a linear Combined on of X; (j=1,2 --- nj±i)

 $X_i = \sum_{j=1}^{n} C_j X_j$ Some Set of Constatisting

* Controllability

- # Controllable > A System is said to be Controllable at time to if it is possible by meas of a Unconstrained Control Vector to transfer the System from any initial State × (to) to any other state in a finite intend of time.
- # Observable > A Signten is Said to be observable at time to if, with the system in state × (10), it is Possible to determine this state from the observation of the output over a finite time litterval.
- => The Concept of Controllability and Observability were introduced by Kalman;
- # Complete State Contorollability of Continuous
 -Time System
- => Consider the Continuous-time System

×=A×+BU

where, X = State Vector (m-Vector) U = Control Signal (Scalar) $A = n \times n matrix$ $C = n \times 1 matrix$

=> The System described above is said to be state Controllable at to to if it is possible to Construct on un constrained Control Signed that will traite on initial state to any find state in a finite line intend to St St. > If every State is Controllable, then the System is Said to be completely State Controllable. => Without any loss of generality, we can assume that the find State in the Origin of the State Space and the initial time is Zero => The Solution of dave equation is : $X(t) = e^{At} \times (0) + \int e^{A(t-\tau)} Bu(\tau) d\tau$ $\times(0) = -\int e^{-A\tau} Bu(\tau) d\tau$ $e^{-A \tau} = \sum_{\kappa=1}^{\infty-1} \alpha_{\kappa}(\tau) A^{\kappa}$ $\mathcal{S}_{0} \times (0) = -\sum_{\kappa=0}^{\infty-1} A^{\kappa} B \int_{0}^{\infty} d_{\kappa}(\tau) u(\tau) d\tau$ Let us put BK = JKK(T)U(T)dT 80 ×(0)= - = AKBBK =-[B; AB; ---; A~'B] | B.

le • • • •

10

The System is completely state Controllette.

Then given any indical state x(0) above aqualion

munt be soldisfied.

[B: AD: ... ; An'B] be n.

Mondition for State Controllability:

The System given by

X = AX+ Bu

is Completely State Controllable if ad only if the Vectors B, AB --- And one linearly independent , on themen madix

[B:AD: -- : AM'B] is of ona KM.

The mesult just obtained can be extended to the cone where the Control Vector u is on-dimensioned.

If the System is described by

×=A×+BU

then it can be proved that the Condition for Complete State Controllability is that the name media

[B:AB: --- ; A^-'B]

be of onakn, os Contain n linea independent Column Vectos

// The matrix

[B|AB|--- |AMB] is Commonly collect the Controllability matrix.

Alternative From of the Condition for Complete State Controllability

$$P^{-1}AP = D = \begin{bmatrix} \lambda_1 & 0 \\ \lambda_2 & \lambda_3 \end{bmatrix}$$

$$\Rightarrow (PZ) = A(PZ) + BU$$

$$\Rightarrow (PZ) = F(I)$$

$$\Rightarrow z = P^{-1}APZ + P^{-1}BU - 0$$

We can enchante \mathcal{E}_{2} @ as: $\dot{Z}_{1} = \lambda_{1}Z_{1} + f_{11}U_{1} + f_{12}U_{2} + \cdots + f_{13}U_{3}$ $\dot{Z}_{2} = \lambda_{2}Z_{2} + f_{2}U_{1} + f_{22}U_{2} + \cdots + f_{23}U_{3}$ $\dot{Z}_{n} = \lambda_{n}Z_{n} + f_{n}U_{n} + f_{n2}U_{2} + \cdots + f_{n4}U_{3}$

=> If the elements of any one onow of the mon matrix F are all zero, then the Cornesponding State variable cannot be Controlled by any of the U:

The Condition of complete State Controllability

is that if the eigenvectors of A are distinit

is that the System is completely State Controllable

then the System is completely State Controllable

if and only if no show of p-18 has all zero element"

=> If the A matrix does not possess distinct eigen Veltans, then diagonalization is impossible. Ly In Such Case we may tones form A into a Jardan Cararical form.

=> Suppose that we can find a transformation motion South that

S-IAS=J

If we define new state vector Zby X=SZ So, Z = s-'ASZ+5-'BU ⇒ Z=JZ+5-'BU

The System is Completely State Controllable if and only if -

- 1) No two Josedan blocks in J are associated with the same eigen values.
- 2) The elements of any now of S-1B that Cornes pond to the last now of each Jordan block are not all zero.
- 3) The claments of sow of 5-1B that Corresponds to distinct eigenvalues are not all zero.

Condition for Complete State Controllability in the Splane

Necessary and Sufficient Condition for Complete state Controllability is that no Concellation occur in the transfer function concellation occur on townsfer matrix. If Concellation occur the System Commot be Controlled in the items of the Concelled mode?

Output Controllability



The the penactical design of a Control System # Un , we may want to Control the output siether Si than the State of the System.

C

y

11

 \Rightarrow Consider the System defined by $\dot{x} = Ax + Bu$ $\dot{y} = Cx + Du$

Where, X = Stede Vactor (m-Vactor) V = Control Vactor (m-Vactor) Y = Output Vactor (m-Vactor) $A = m \times m matrix$ $C = m \times m matrix$ $C = m \times m matrix$

The System described above is Said to be Completely Output Controllable if it is possible to Construct an unconstrained control vectors u(t) that will transfer any given initial output Y(to) to any final output Y(to) in a finite time interval to It & to,

The System described above is completely output Controllable if and only if the mx(mil) on matrix.

[CB:CAD; CAZD; --- ; CAT-B; D]

Un Conterollable System => System which has a Subsystem that is physically disconnected from the imput

Stabilizability >> Foon a postially Controllable System, if the Un Controllable modes are stable and the estable unstable modes are Controllable, the System is Said to be Stabilizable.

* Observability

=> Consider on unforced System described by the following equations:

×=A× -0 Y = C× -0 Y = C× -0 A => n×n metrix C ⇒ m×n metrix

=> System is Completely observable if:

Frenz State X(ta) Can be determined forom the observation of y(t) over a finite time intend, to < t < t, "

=> In this Soution we treat only linear, time -invariant System. Therefor, without loss of generality, we can assume that to =0.

Importace of Concept of)
Observability

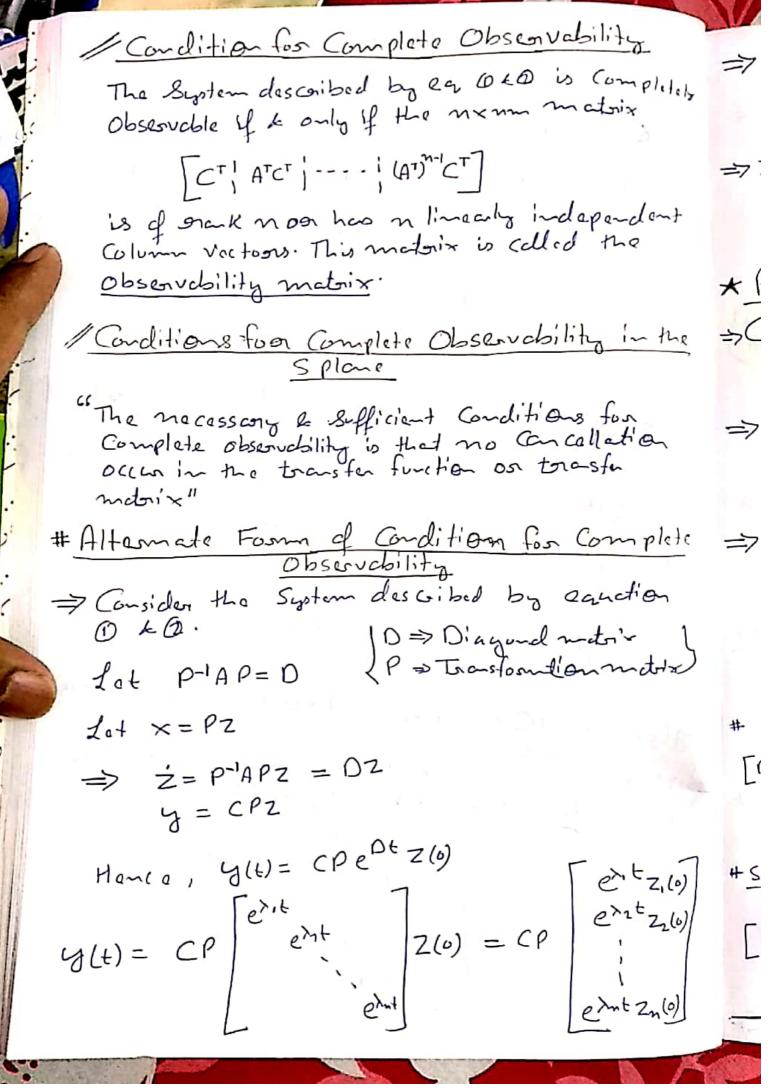
> In practice, the difficulty encountered with State feedback control is that some with State Variables are not assessible of the State Variables are not assessible for direct measurement, with the subset for direct measurement, with the subset it be come necessary to estimate the it be come necessary to estimate the unmeasurable State Variables in order to construct the Control signals.

&I

S

=7

If we describe Syptim byin = Ax + Bu => x(1) = eA6 x(0) + [eA(6-7) Bu (7)d7 Y= Ex+ DU => GLE)= CeAEX(0) + c geALE-T) BULTI OU Since the matrix A.B. C.k D are Known and U(4) is also known, the last two terms on the sight had side of this last equation and known quality. Henre, for investigating a necessary & sufficient Condition for Complete observability, it suffices to Consider the System described by Egud'an Oto # Complete Observability of Continus-time System => The output Vactor y(t) is:characteristics y(t) = ceAt×(0) We Know, eAt = STUKLED AK So, $y(t) = \sum_{k=n}^{m-1} \alpha_k(t) CA^k \times (0)$ Y(t) = <((t) C×(0) + <,(t) CA×(0) + -+ </--(t) CA⁻¹×(0) => At Can be Shown that for Complet observability this medrix of mon xn enequines the sick to - CA - Y bo n.



- The System is Completely observable yourse of the Column of the man matrix (P Consists of all Zeo clement.
- => If the metric A camet be transformed into a diagond metric, then by use of a Suitable transformation metric S, we can transform A transformation Canonical form.

* Painciple of Duality

=> Consider the System S. described bo ×= Ax + BU Y = Cx

=> And the dual system So defined by $\dot{Z} = A^T Z + c^T V$ $M = D^T Z$

The poinciple of duality States that the system Squis Completely State Controllable (observable) if and only if system Squis Completely observable (State Controllable).

Stelo Controllability

+ State observability

[cT; ATCT! --- |(AT) MCT]

Food Ond System 52

Stede Controllebitity

[CT | ATCT | -- | (AT) - CT)

> Raken

Stete Observebilits

Talance 1-- 1AMBT

[B: AB 1--- 1AMB]

> maka