## AST 301 Problem Set 6

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a) By definition of the Schwarzschild metric in relativistic units, we have:

$$ds^2 = -(1-\frac{2M}{r})dt^2 + (1-\frac{2M}{r})^{-1}dr^2 + r^2(d\theta^2 + sin^2\theta d\phi^2)$$

Because we are looking for a proper "radial" distance, we set dt = 0,  $d\theta = 0$ , and  $d\phi = 0$ . Therefore, we have:

$$ds^2 = (1 - \frac{2M}{r})^{-1}dr^2$$

$$ds = \sqrt{(1 - \frac{2M}{r})^{-1}} dr$$

Integrating, and setting the lower bound to be the Schwarzschild radius and leaving a variable upper bound, we have:

$$s = D(r) = \int_{2M}^{r} \sqrt{(1 - \frac{2M}{r'})^{-1}} dr' = \int_{2M}^{r} \sqrt{g_{rr}(r')} dr'$$

b) we first make the substitution

$$r = 2M(1 + \epsilon)$$

$$dr = 2Md\epsilon$$

Our integral becomes, with adjusted bounds in epsilon coordinates:

$$D(r) = 2M \int_0^{\epsilon} \frac{1}{\sqrt{1 - \frac{1}{1 + \epsilon'}}} d\epsilon'$$

rearranging,

$$=2M\int_0^{\epsilon} \sqrt{1+\frac{1}{\epsilon'}}d\epsilon'$$

 $\epsilon$  is less than one, so we must add a factor of  $\epsilon'$  inside of the square root to use the binomial theorem.

$$=2M\int_0^{\epsilon} \epsilon'^{-1/2}\sqrt{1+\epsilon'}d\epsilon'$$

by the binomial theorem,

$$(1+\epsilon)^{1/2} \approx 1 + \frac{1}{2}\epsilon$$

the integral may now be approximated by:

$$=2M\int_0^{\epsilon} \epsilon'^{-1/2} + \frac{1}{2}\epsilon'^{1/2}d\epsilon'$$

Integrating and taking the first order term, we get:

$$D(r) = 4M\epsilon'^{\frac{1}{2}}\Big|_{0}^{\epsilon}$$

we find that

$$\frac{D}{M} = 4\epsilon^{\frac{1}{2}}$$

c) now we may approximate  $\sqrt{(1-\frac{2M}{r})}$  from the original integral with the binomial theorem because r>>2M.

$$\sqrt{(1-\frac{2M}{r})}\approx 1-\frac{M}{r}$$

our integral for D becomes:

$$D(r) = \int_{2M}^{r} \frac{r'dr'}{r' - M}$$

Using a substitution r - m = x, we write:

$$= \int_{M}^{r-M} \frac{(x+M)dx}{x}$$

integrating,

$$= (x + Mlog(x))\Big|_{M}^{r-M}$$

we find that

$$D(r) = r - 2M + M(\log(\frac{r}{M} - 1))$$

approximated to the leading order.

I redefine "curly  $\nu$ " as  $\beta$  for ease of typing. We define the impact parameter, b, as:

$$sin(\beta) = \frac{b}{D}$$

by small angle approximation,

$$sin(\beta) \approx \beta$$

$$b \approx \beta D$$

by definition of angular momentum, we have:

$$r \times p = L = bpsin(\beta + \frac{\pi}{2}) = bpcos(\beta)$$

Again, by small angle approximation, we have:

$$L \approx bp$$

Subbing in b,

$$L \approx \beta Dp$$

At  $r_{min}$ , we know that  $\frac{dr}{d\lambda}$  and  $\frac{d^2r}{d\lambda^2}$  must vanish. We have these equations for r's derivatives:

$$(\frac{dr}{d\lambda})^2 = E^2 - \frac{L^2}{r^2}(1 - \frac{2M}{r}) = 0$$
$$\frac{d^2r}{d\lambda^2} = \frac{L^2}{r^3}(1 - \frac{3M}{r}) = 0$$

From the second equation, we have:

$$r_{min} = 3M$$

Subbing into the first equation, we get:

$$E^2 = \frac{L^2 \beta^2 D^2}{3 M^2 c^2}$$

Using  $E_{\gamma} = h\nu_{\infty}$ ,  $p_{\gamma} = \frac{E}{c}$ , and  $L_{\gamma} = \beta Dp_{\gamma}$ , we get:

$$h^2 \nu_{\infty}^2 = \frac{h^2 \nu_{\infty}^2 \beta^2 D^2}{27c^2 M^2}$$

reducing,

$$1 = \frac{\beta^2 D^2}{27c^2 M^2}$$

solving for  $\beta$ , we get:

$$\beta = \frac{3\sqrt{3}cM}{D}$$

a) One of our constants of motion for a circular orbit is given by:

$$p_t = -(1 - \frac{2M}{r})m\frac{dt}{d\tau} = -E$$

rearranging,

$$\frac{E}{m} = \frac{dt}{d\tau} (1 - \frac{2M}{r})$$

the energy per unit mass for a circular orbit is given by:

$$\frac{E}{m} = \sqrt{\frac{(r_c - 2M)^2}{r_c(r_c - 3M)}}$$

Subbing the above equation into the one above it, (taking  $r = r_c$ ) we get:

$$\sqrt{\frac{r_c}{r_c - 3M}} = \frac{dt}{d\tau}$$

Separating and integrating, and setting the clocks to start at the same time, we get:

$$\sqrt{\frac{r_c}{r_c - 3M}}\tau = t$$

We now use dimensional analysis to insert our factors of G and c, knowing that the square root term must be dimensionless:

$$\sqrt{\frac{r_c}{r_c - \frac{3GM}{c^2}}} \tau = t$$

multiply the inside of the square root by  $\frac{\frac{1}{M}}{\frac{1}{M}}$  because we are interested in the ratio  $\mu = \frac{r_c}{M}$ 

$$\sqrt{\frac{\mu}{\mu - \frac{3G}{c^2}}}\tau = t$$

solving for  $\mu$ , we get:

$$\mu = \frac{3G}{c^2(1 - (\frac{\tau}{t})^2)}$$

we know that 1 hour on the planet corresponds to 7 years "at infinity", so  $\tau=1$  and t=7\*365\*24=61320 Solving explicitly for  $\mu$ ,

$$\mu = 2.2278 * 10^{-27} \frac{m}{kg}$$

In relativistic units, we get:

$$\mu = 3.0000000008$$

b) The equation for orbital energy per mass for a circular orbit is given by:

$$\epsilon = \frac{r_c - 2M}{\sqrt{r_c(r_c - 3M)}}$$

dividing num and denom by M, we get:

$$\epsilon = \frac{\mu - 2}{\sqrt{\mu(\mu - 3)}}$$

plugging in our value for  $\mu$  (relativistic units) we get:

$$\epsilon = 20440$$

c) No, the energy per unit mass for the marginally stable orbit is  $\frac{\sqrt{8}}{3}$ . Our  $\epsilon$  is much greater and its orbit is therefore unstable. (We know that  $\epsilon \to 1$  as  $r_c \to \infty$ )

## 4

We know that the energy input to the system is  $E_{rad}$ , so the efficiency of the heat engine may be given by:

$$\frac{E_{rad} - E_{min}}{E_{rad}}$$

The Carnot efficiency is the maximum efficiency of a heat engine:

$$\eta = 1 - \frac{T_{cold}}{T_{hot}}$$

We are given  $E_{rad} = aT_{hot}^4L^3$ .  $E_{min}$  may be calculated through the metric.